

# Validation of boxwing models for GNSS satellites

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# Content

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- Background
- Approach
- About the scalable boxwing model
  - Scale Factors
  - Plate Groupings
- Results
- Conclusion

# Background

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## Background

- Radiation Pressure a significant source of error in orbit modeling
  - Solar Radiation
  - Earth Reflected/Emitted Radiation
- New Empirical CODE Orbit Model (ECOM2) [Arnold et al. 2015, J. Geod.]
  - Effective for GPS & GLONASS outside of eclipse season
  - Less effective for Galileo, etc.
- Adjustable boxwing model  
[Solano, 2014, PhD Thesis]; [Montenbruck et al. 2015, J. Geod.]; [Montenbruck et al. 2017, Adv. Space Research]
  - Semi-Analytical model for radiation pressure
  - Improved performance during eclipse season

# Approach

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## Approach

- Implement Scalable–boxwing model in development version of Bernese GNSS Software
- Compute scale factors per satellite, plate, etc.
  - Investigate various plate groupings
  - Identify long–term trends in scale factors per SVN
- Analyze improvements over ECOM2
  - ECOM2–only
  - ECOM2–plus–boxwing
  - ECOM2–plus–scaled–boxwing:
    - Satellite–specific, Yearly–average Scale Factors

# Scalable-Boxwing Model Definition

Radiation Pressure force calculation per plate:

Without immediate thermal re-radiation:  $\vec{F} = -\frac{\Phi}{c} \cdot A \cos \theta \cdot \left[ (\alpha + \delta) \vec{e}_{\odot} + \frac{2}{3} \delta \vec{e}_n + 2\rho \cos \theta \cdot \vec{e}_n \right]$

With immediate thermal re-radiation (MLI):  $\vec{F} = -\frac{\Phi}{c} \cdot A \cos \theta \cdot \left[ (\alpha + \delta) \left( \vec{e}_{\odot} + \frac{2}{3} \vec{e}_n \right) + 2\rho \cos \theta \cdot \vec{e}_n \right]$

$\Phi =$  solar flux

$c =$  speed of light

} Constants

$A =$  surface area of plate

$\alpha =$  absorptivity of plate

$\delta =$  diffuse reflectivity of plate

$\rho =$  specular reflectivity of plate

$\vec{e}_n =$  unit vector normal to plate

$\vec{e}_{\odot} =$  unit vector towards radiation source

$\theta =$  angle between  $\vec{e}_{\odot}$  and  $\vec{e}_n$

} Macromodel definition

$$\alpha + \delta + \rho = 1$$

} Attitude geometry

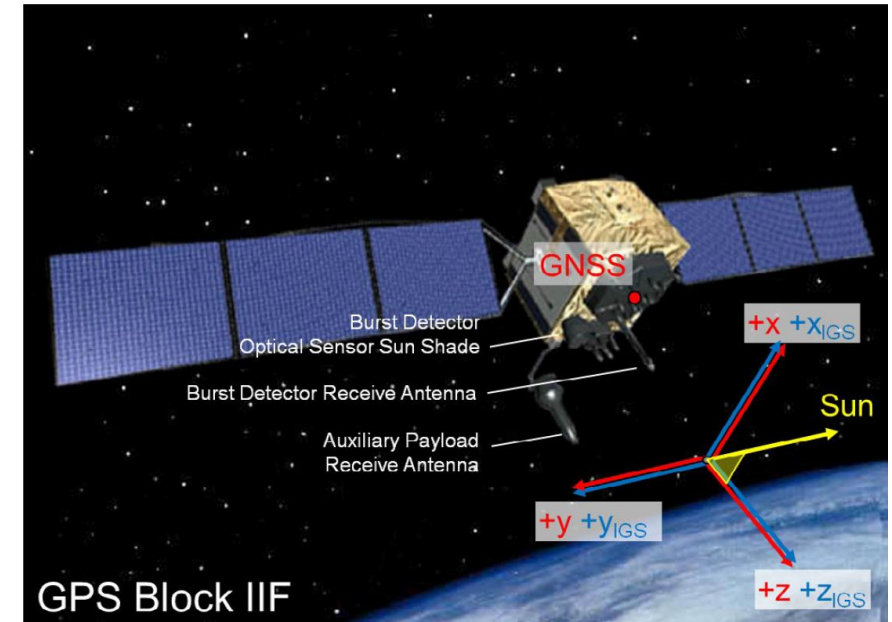
Exact geometry and material properties are not necessarily known. A single scale factor multiplied by the force on a given plate can compensate for uncertainties in all four plate properties.

Scale factors can be introduced as solveable parameters in the least squares model.

# Example Boxwing Macromodel & Plate Groupings: GPS

## Macromodel Definitions:

- By SV block
  - Example is GPS IIF
- As collection of plates
  - Geometrical and optical properties for each plate
  - Force calculated on any plate where  $\cos \theta > 0$  and summed together
  - Only specular and diffuse reflectivity are specified
  - Absorptivity is Calculated



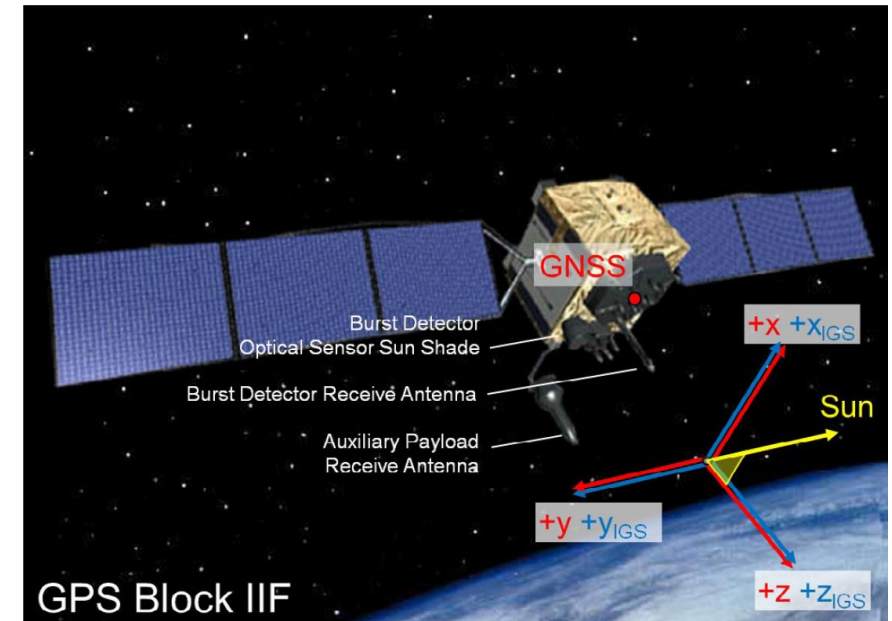
[Montenbruck et al, 2015. Adv. In Space Research]

Multiscale	Plate	Mod	Area (A) [m <sup>2</sup> ]	Normal ( $\vec{e}_n$ )	Specularity ( $\rho$ )	Diffusivity ( $\delta$ )	Rotation Sys.	Description
	1	1	5.720	[+1, 0, 0]	0.112	0.448		+X
	2	1	5.720	[-1, 0, 0]	0.112	0.448		-X
	3	1	7.010	[0, +1, 0]	0.112	0.448		+Y
	4	1	7.010	[0, -1, 0]	0.112	0.448		-Y
	5	1	5.400	[0, 0, +1]	0.112	0.448		+Z
	6	1	5.400	[0, 0, -1]	0.000	0.000		-Z
	7	0	22.250	[+1, 0, 0]	0.195	0.035	+SUN: [0, +1, 0]	Solar panels front
	8	0	22.250	[-1, 0, 0]	0.196	0.034	-SUN: [0, +1, 0]	Solar panels back

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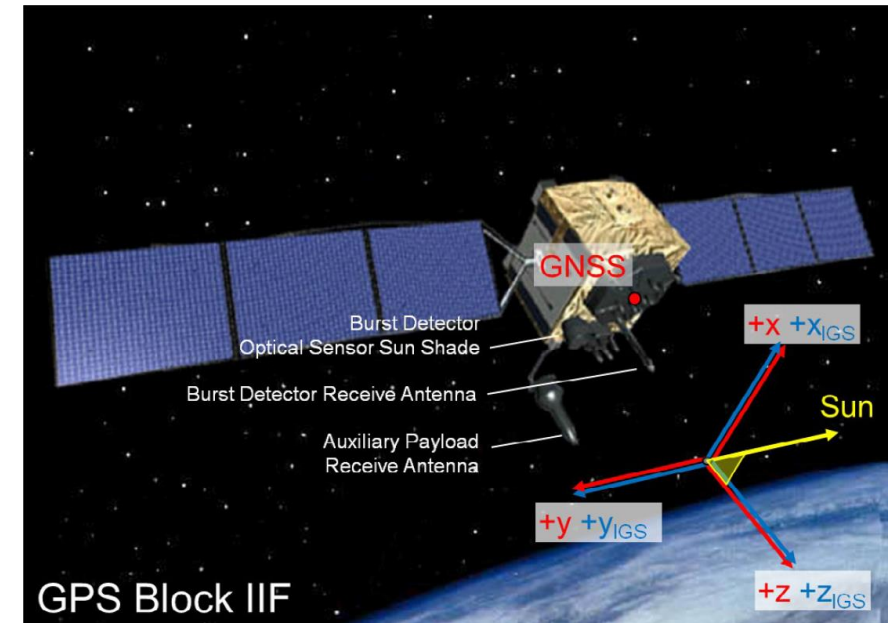
GPS Block IIF  
[Montenbruck et al, 2015. Adv. In Space Research]

	Plate	Mod	Area (A) [m <sup>2</sup> ]	Normal ( $\vec{e}_n$ )	Specularity ( $\rho$ )	Diffusivity ( $\delta$ )	Rotation Sys.	Description
Monoscale	1	1	5.720	[+1, 0, 0]	0.112	0.448		+X
	2	1	5.720	[-1, 0, 0]	0.112	0.448		-X
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GPS Block IIF  
[Montenbruck et al, 2015. Adv. In Space Research]

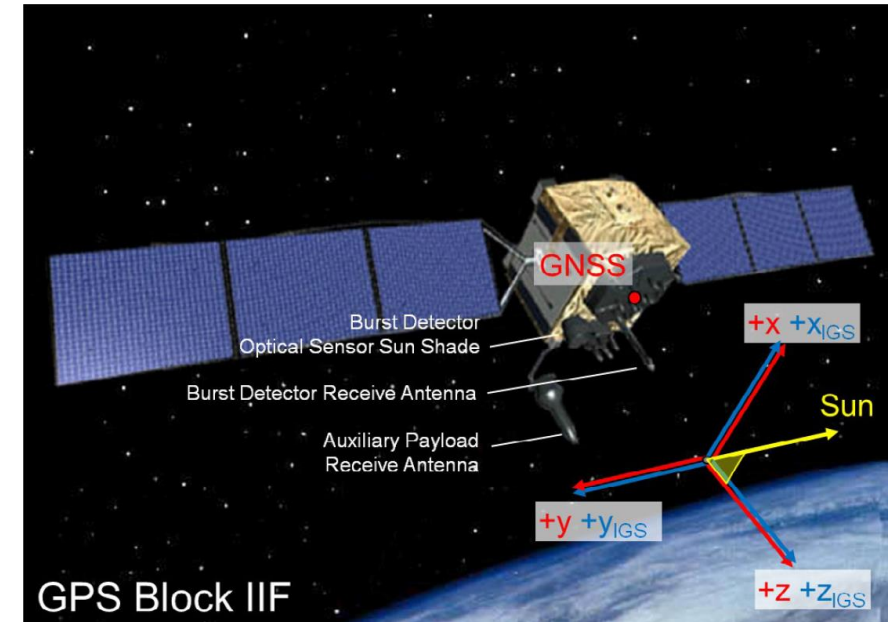
	Plate	Mod	Area (A) [m <sup>2</sup> ]	Normal ( $\vec{e}_n$ )	Specularity ( $\rho$ )	Diffusivity ( $\delta$ )	Rotation Sys.	Description
Smartscale-2	1	1	5.720	[+1, 0, 0]	0.112	0.448		+X
	2	1	5.720	[-1, 0, 0]	0.112	0.448		-X
	3	1	7.010	[0, +1, 0]	0.112	0.448		+Y
	4	1	7.010	[0, -1, 0]	0.112	0.448		-Y
	5	1	5.400	[0, 0, +1]	0.112	0.448		+Z
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GPS Block IIF  
[Montenbruck et al, 2015. Adv. In Space Research]

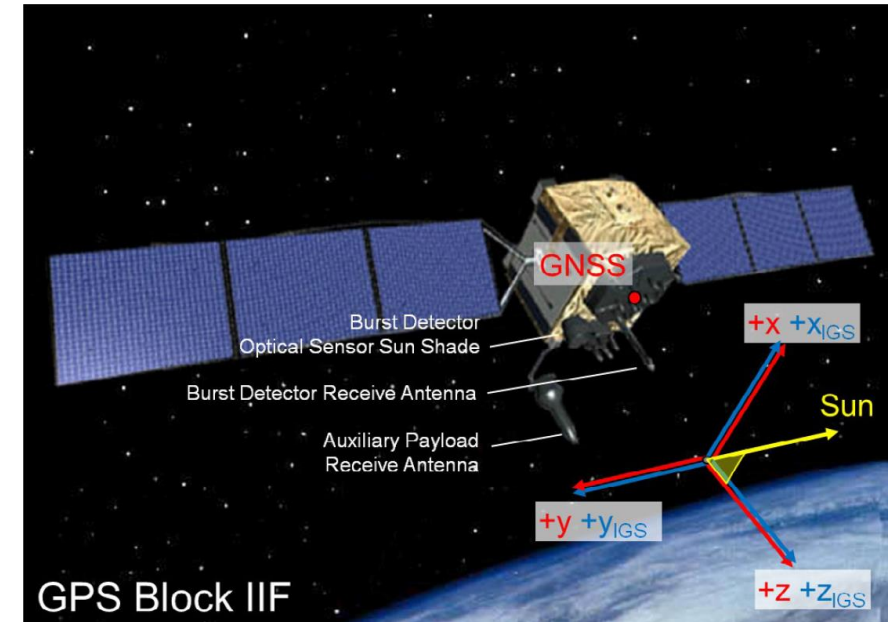
Smartscale-4

Plate	Mod	Area (A) [m <sup>2</sup> ]	Normal ( $\vec{e}_n$ )	Specularity ( $\rho$ )	Diffusivity ( $\delta$ )	Rotation Sys.	Description
1	1	5.720	[+1, 0, 0]	0.112	0.448		+X
2	1	5.720	[-1, 0, 0]	0.112	0.448		-X
3	1	7.010	[0, +1, 0]	0.112	0.448		+Y
4	1	7.010	[0, -1, 0]	0.112	0.448		-Y
5	1	5.400	[0, 0, +1]	0.112	0.448		+Z
6	1	5.400	[0, 0, -1]	0.000	0.000		-Z
7	0	22.250	[+1, 0, 0]	0.195	0.035	+SUN: [0, +1, 0]	Solar panels front
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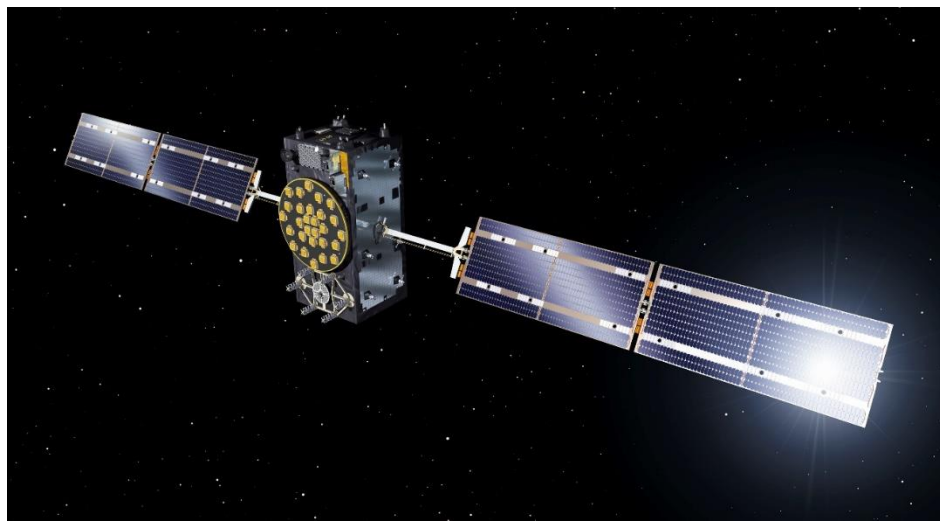


GPS Block IIF  
[Montenbruck et al, 2015. Adv. In Space Research]

Smartscale-3

Plate	Mod	Area (A) [m <sup>2</sup> ]	Normal ( $\vec{e}_n$ )	Specularity ( $\rho$ )	Diffusivity ( $\delta$ )	Rotation Sys.	Description
1	1	5.720	[+1, 0, 0]	0.112	0.448		+X
2	1	5.720	[-1, 0, 0]	0.112	0.448		-X
3	1	7.010	[0, +1, 0]	0.112	0.448		+Y
4	1	7.010	[0, -1, 0]	0.112	0.448		-Y
5	1	5.400	[0, 0, +1]	0.112	0.448		+Z
6	1	5.400	[0, 0, -1]	0.000	0.000		-Z
7	0	22.250	[+1, 0, 0]	0.195	0.035	+SUN: [0, +1, 0]	Solar panels front
8	0	22.250	[-1, 0, 0]	0.196	0.034	-SUN: [0, +1, 0]	Solar panels back

# Discussion on Thermal re-radiation: Galileo

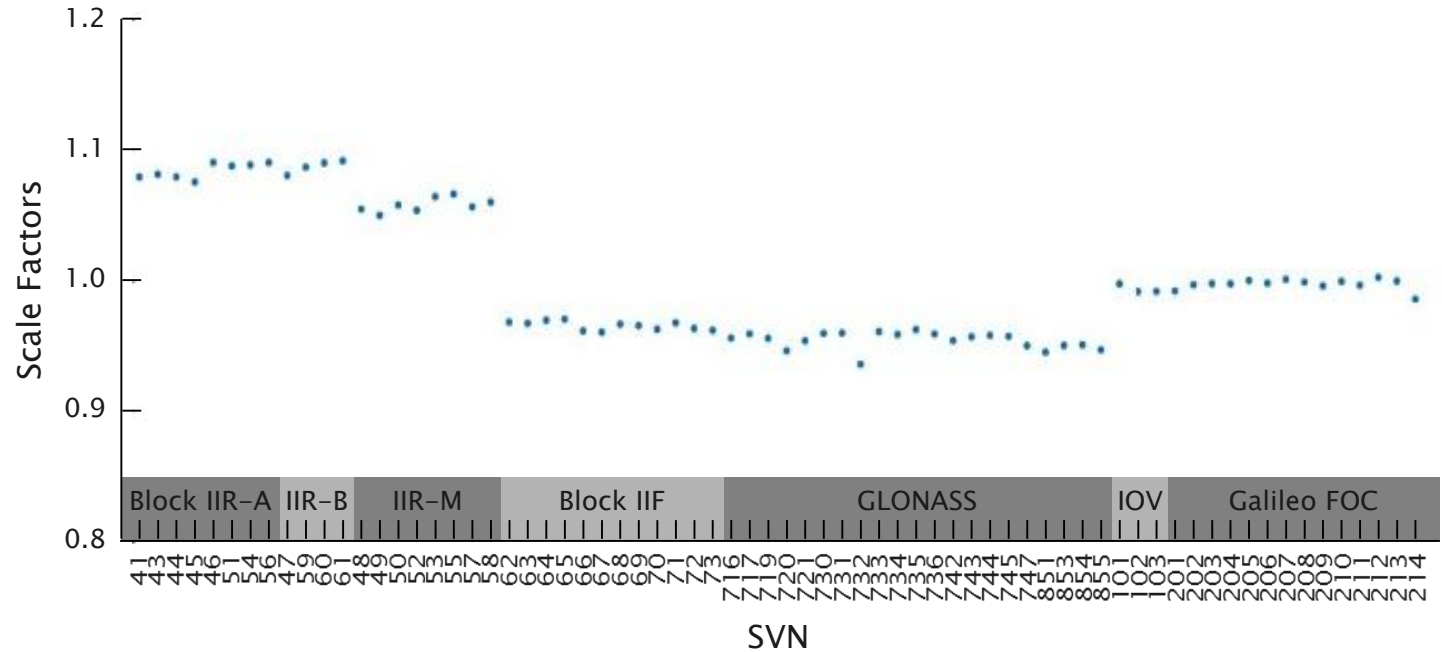
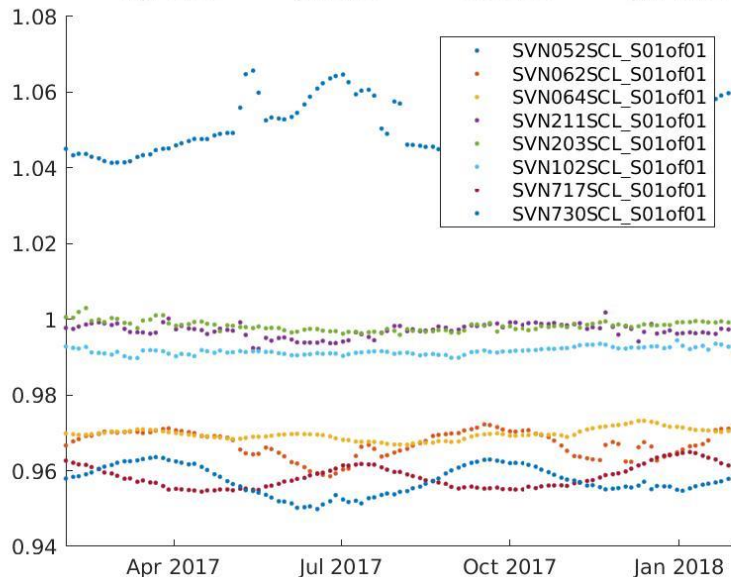
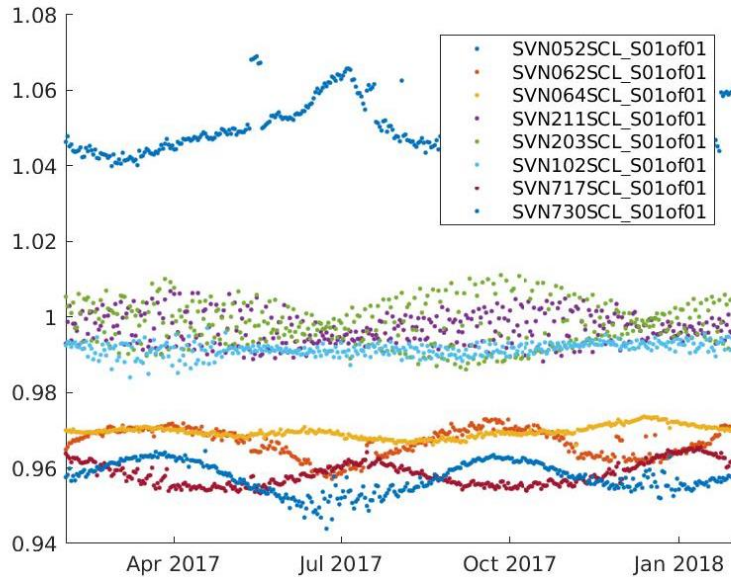


[[https://www.esa.int/spaceinimages/Images/2014/07/Galileo\\_satellite](https://www.esa.int/spaceinimages/Images/2014/07/Galileo_satellite)]

- Satellite geometry and optical properties given by GSA
- Front side of solar panel has two different “materials”
  - ~28% of solar panel surface area
  - With immediate thermal re-radiation yields scale factors closer to 1.

Plate	Mod	Area (A) [m <sup>2</sup> ]	Normal ( $\vec{e}_n$ )	Specularity ( $\rho$ )	Diffusivity ( $\delta$ )	Rotation Sys.	Description
1	1	1.320	[+1, 0, 0]	0.000	0.070		-X Material A
2	1	0.440	[-1, 0, 0]	0.000	0.070		+X Material A
3	1	0.880	[-1, 0, 0]	0.730	0.190		+X Material C
4	1	1.244	[0, +1, 0]	0.000	0.070		-Y Material A
5	1	1.539	[0, +1, 0]	0.730	0.190		-Y Material C
6	1	1.129	[0, -1, 0]	0.000	0.070		+Y Material A
7	1	1.654	[0, -1, 0]	0.730	0.190		+Y Material C
8	1	1.053	[0, 0, +1]	0.000	0.070		+Z Material A
9	1	1.969	[0, 0, +1]	0.220	0.210		+Z Material B
10	1	2.077	[0, 0, -1]	0.000	0.070		-Z Material A
11	1	0.959	[0, 0, -1]	0.730	0.190		-Z Material C
12	0	7.760	[+1, 0, 0]	0.080	0.000	+SUN: [0, +1, 0]	Solar Panels Material E
13	?	3.060	[+1, 0, 0]	0.100	0.000	+SUN: [0, +1, 0]	Solar Panels Material D
14	0	10.820	[-1, 0, 0]	0.196	0.034	-SUN: [0, +1, 0]	Solar Panels back

# Long-term Trends in Scale Factors: Monoscale

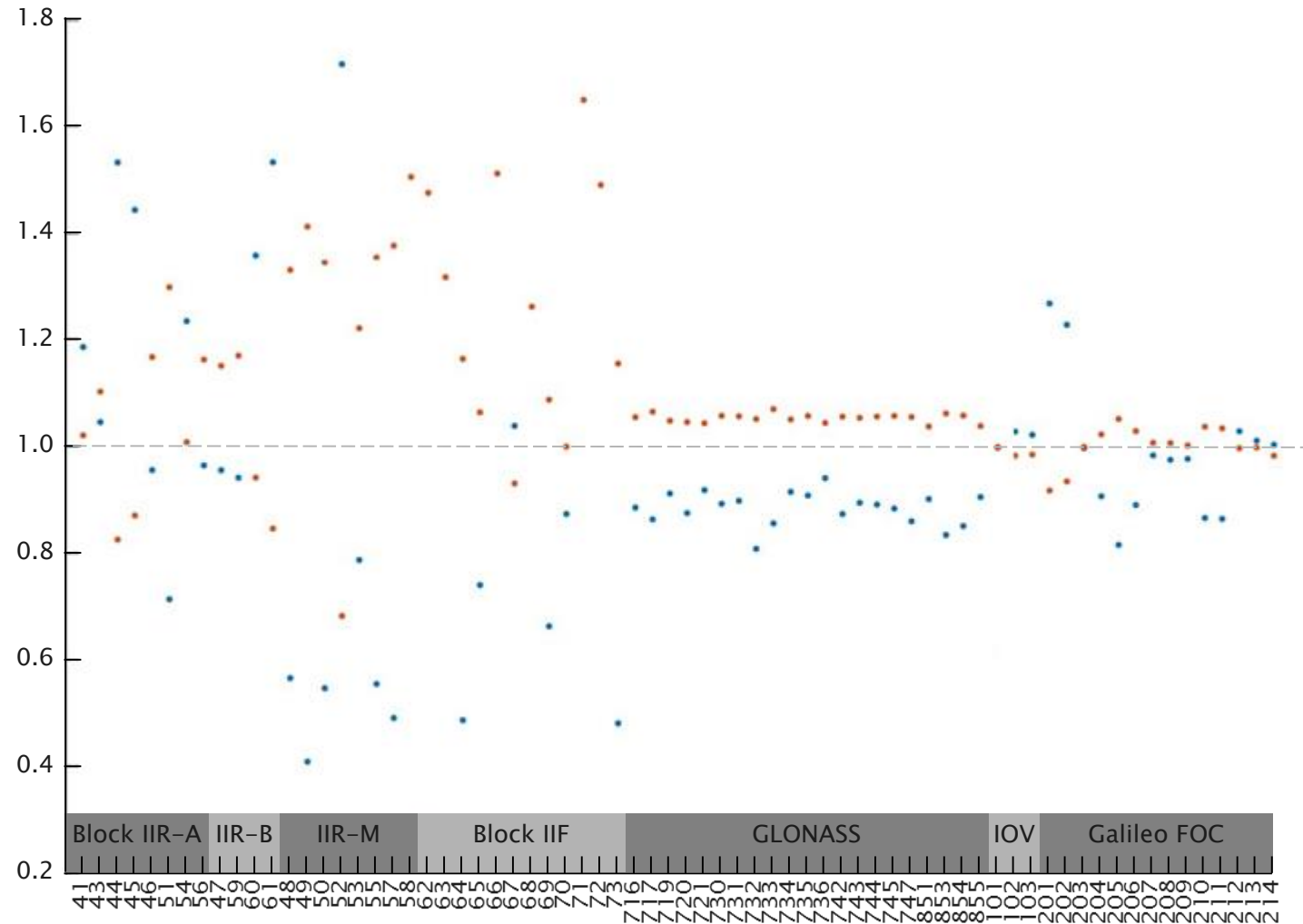


**Top Left:** Scale factors for selected satellites:  
Monoscale model, daily solution, 1-year

**Bottom Left:** Scale factors for selected satellites:  
Monoscale model, 7-day stack, 1-year

**Top Right:** Scale factors for all satellites:  
Monoscale model, 1-year stack

# Yearly Scale Factors: Smartscale-2



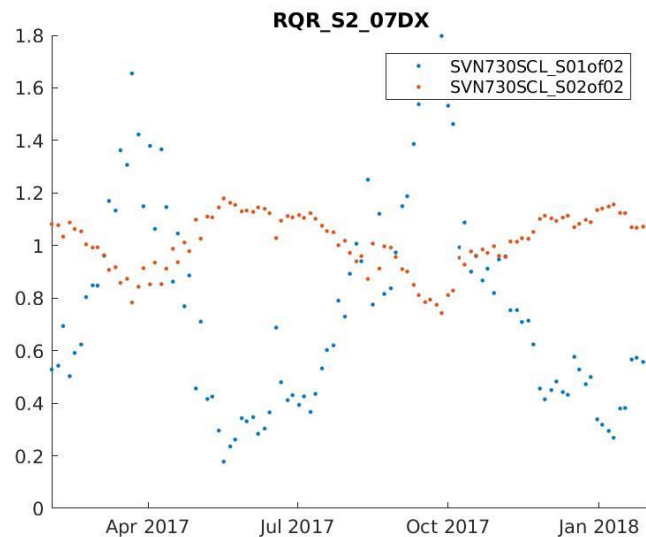
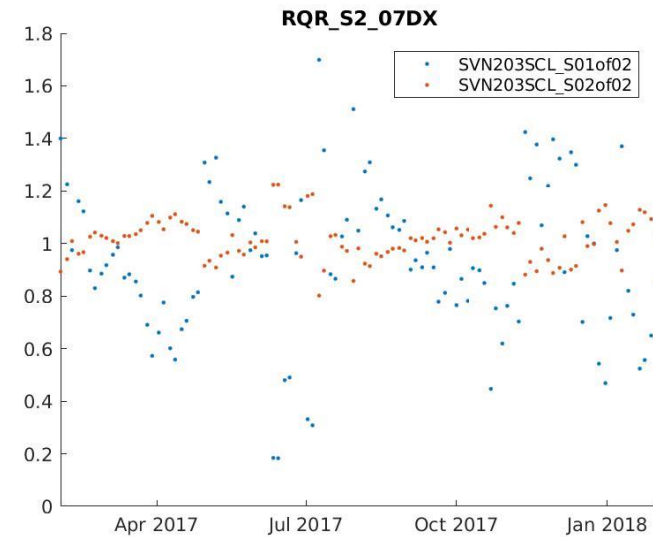
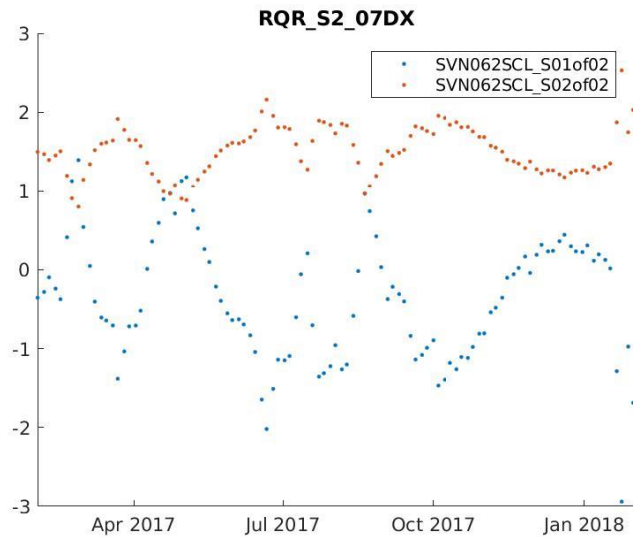
**GLONASS & Galileo:**  
similar scale factors for all  
satellites in same block

Scale factors close to 1.

**GPS:**  
more variation between  
satellites in same block

Scale factors farther away  
from 1.

# Long-term Trends in Scale Factors: Smartscale-2



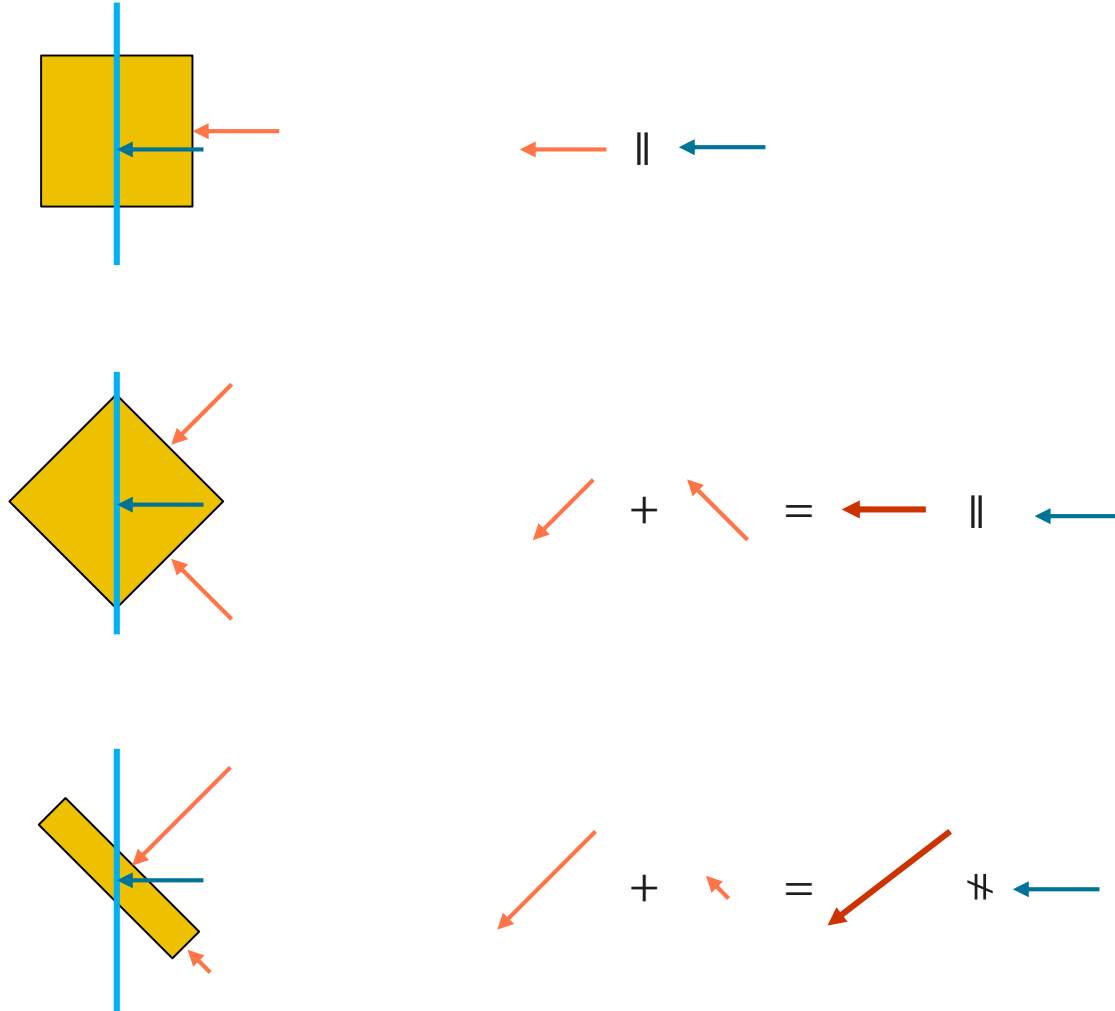
Scale factors for selected satellites:  
Smartscale-2 model, 7-day stack, 1-year  
Top Left: GPS SVN 62  
Bottom Left: GLONASS SVN 730  
Top Right: Galileo SVN 203

**\*Note different scale for GPS**

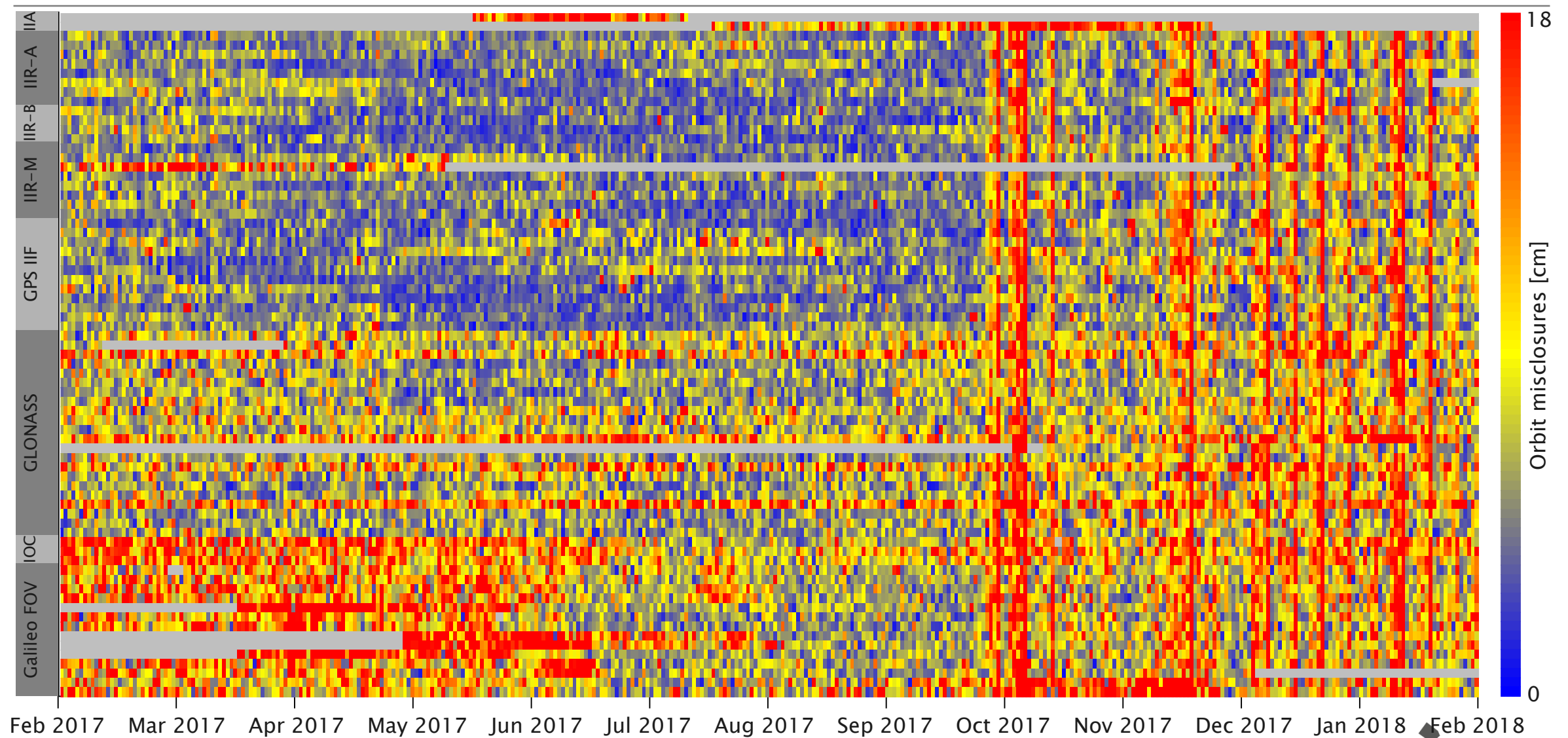
# Monoscale vs. Smartscale/Multiscale

Co-variances between scale factors due to:

- Similar optical properties
- Parallel plates
- Attitude geometry
- Parallel resultant force

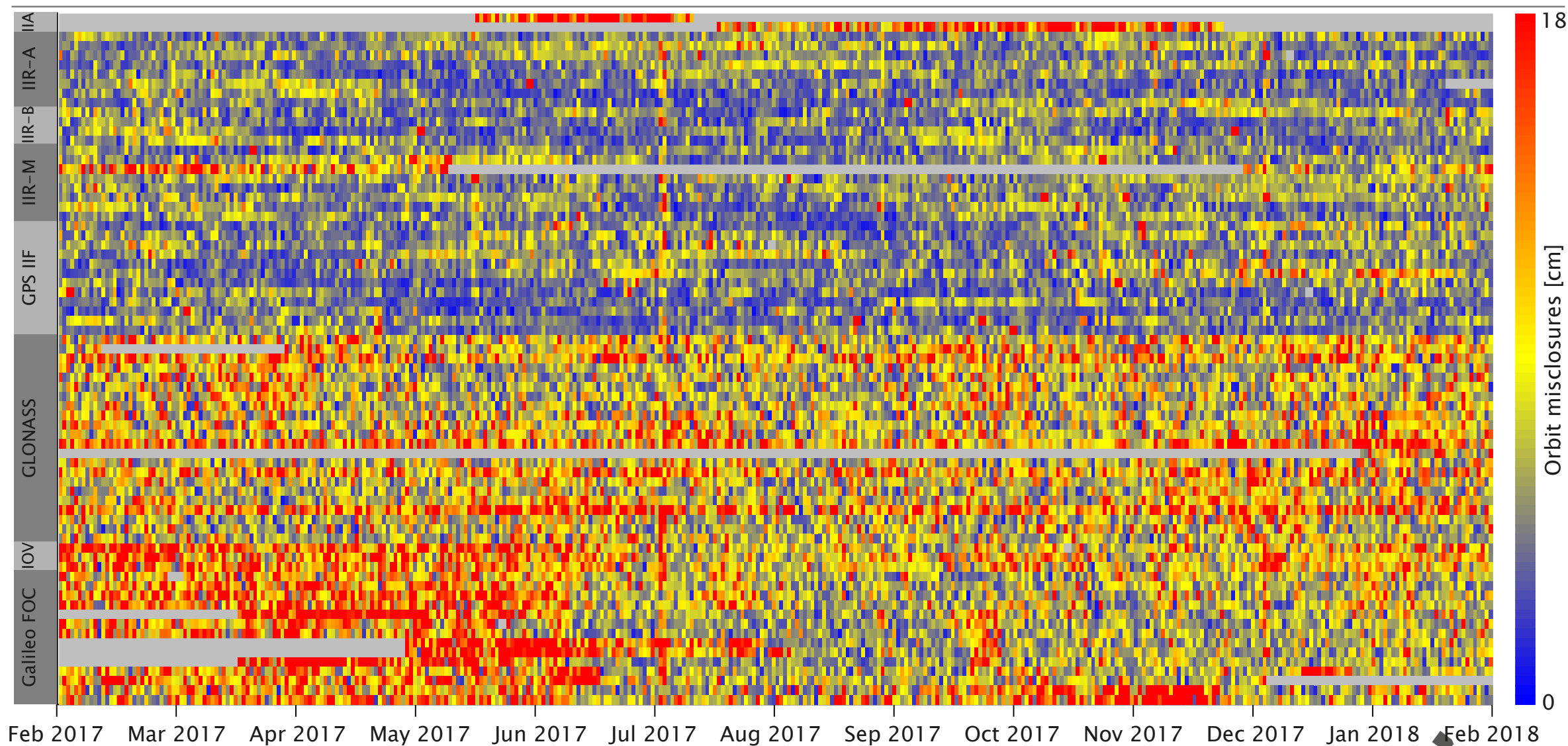


# Orbit Misclosures: ECOM-only (1-day solutions)





# Orbit Misclosures: ECOM-plus-boxwing (1-day solutions)



# Orbit Misclosures: ECOM-plus-scaled-boxwing

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# Conclusions & Future Work

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## Conclusions

- Able to stack scale factors for long periods of time
- Able to distinguish/validate thermal re-radiation
- Number of scale factors per satellite depends on characteristics
- Improvements at the daily solution level

## Forward Work



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# QUESTIONS?