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NATURAL FIBRE REINFORCED COMPOSITE MATERIALS

by

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A thesis submitted to the University of Plymouth in partial fulfilment for the degree of PhD

DOCTOR OF PHILOSOPHY

School of Engineering

June 2019

Acknowledgements

Many people have greatly contributed to the development of this thesis over the years.

I want to express my most sincere thanks to the supervision team: John Summerscales (Director of Studies), Jasper Graham-Jones, Wayne Hall (latterly at Griffith University in Australia) and Mary Margaret (Miggy) Singh for their expert guidance, advice, support, and encouragement. Further the expert commentators have made valuable input: Professor M Neil James (Head of School of Marine Science & Engineering/ Associate Dean) Dr Shungi Pan and Dr Frank Abraham.

The work was mostly conducted in the Basque Country (Euskadi) in northern Spain using facilities in the University of the Basque Country (UPV/EHU), University of Zaragoza and Acciona (latterly Nordex) Windpower in Sarigurren.

Although these doctoral studies have been primarily self-funded, I am most grateful for the Santander Universities Scholarship funding which supported the experimental work in University of Zaragoza.

The author would especially like to thank Mr. Brendon Weager at Composites Evolution for reinforcement fabrics and Mr. Jaime Ferrer-Dalmau at Entropy Resins for the bio-epoxy.

I am also grateful to the anonymous referees for their respective comments on the manuscripts of the book chapter and the conference papers.

Last but not least important, I would like to thank Aurora for her unconditional love, making me believe that together nothing is impossible and giving me the strength to keep going at weakest moments. Love you Michelle for always keeping my heart happy. Aita, Ama, Xabi, Olatz, Maialen, Borja, Nora and Naia, mila esker for always believing in me and giving me the strength to continue, maite zaituztet. Thank you Iñigo for your invaluable help as I took the first steps. My Welsh family John, Irene, Lowri, Robs, Ceri, Cai, Elin, Mared and Meilyr diolch for making me feel part of the family. Max, Bea, David, Salam and Chamy, thank you for making me feel at home. Marta, Cecilia and Iker, thank you for your technical support and friendship throughout the experimental campaign. Asier, mila esker, hau posible egin dezulako. Thank you Alvaro and Elena for being good friends more than colleagues. Sorry Ion and Yeli not for attending your wedding, it was a real pity. Thank you family and friends for all your love and laughter that have made me happy every day.

Finally dedicate the work to my uncle Jose and aunty Maite who have recently passed away, thank them for making me feel proud of being part of the Mitxelena family.

Author's declaration

At no time during the registration for the degree of Doctor of Philosophy has the author has been registered for any other University award without prior agreement of the Doctoral College Quality Sub-Committee. Work submitted for this research degree at the University of Plymouth has not formed part of any other degree either at the University of Plymouth or at another establishment. This study was partially financed with the aid of a scholarship from Santander.

The following external institutions were visited for consultation purposes:

Green and Wood-Based CompositesPfintzal - GermanyFraunhofer ICT 30 September 2008CoMatComp Composites San-Sebastian - Euskadi (Spain)06-09 September 2009Resin CuringMadrid - SpainTA Instruments 01-04 April 2009Journées Europeennes des CompositesParis - FranceJEC29-31 March 2011

Publications (or public presentation of creative research outputs):

Book Chapter:

Aitor Hernandez Michelena, Jasper Graham-Jones, John Summerscales and Wayne Hall <u>Eco-friendly flax fibre/epoxy resin/composite system for surfboard production</u> Chapter 20 in Raul Fangueiro and Sohel Rana (editors): <u>Natural Fibres:</u> <u>Advances in Science and Technology Towards Industrial Applications: From Science To</u> <u>Market</u>,

Springer RILEM Book Series, March 2016, (12 Part III), 267-277. ISBN 978-94-017-7513-7.

DOI: <u>https://doi.org/10.1007/978-94-017-7515-1_20</u>

Conference papers:

- A Hernandez Michelena, J Graham-Jones, W Hall and J Summerscales <u>Resin transfer moulding (RTM) to produce surfboard fins with natural fibres</u> ICCST-9, Sorrento ~ ITALY, 24-26 April 2013/oral: Thursday 25 April 2013 at 15:10. DEStech Publications Inc, Lancaster PA, 2013. ISBN 978-1-60595-113-3, pages 577-583.
- 2. Aitor Hernandez Michelena, Jasper Graham-Jones, Wayne Hall and John Summerscales <u>Resin transfer moulding (RTM) fins production with natural fibres</u> 1st International Conference on Natural Fibers: sustainable materials for advanced applications (ICNF2013),

Guimarães ~ PORTUGAL, 09-11 June 2013, Session 8a, abstracts book pages 49-50.

3. Aitor Hernandez Michelena, Jasper Graham-Jones, Wayne Hall and John Summerscales <u>Ecoboard</u> <u>production</u> <u>with</u> <u>natural</u> <u>fibres</u> 1st International Conference on Natural Fibers: sustainable materials for advanced applications (ICNF2013),

Guimarães ~ PORTUGAL, 09-11 June 2013, poster, abstracts book pages 339-340.

- Aitor Hernandez Michelena, Jasper Graham-Jones, Wayne Hall and John Summerscales <u>Natural fibre treatments review</u>, 5th International Conference on Sustainable Materials, Polymers and Composites (Ecocomp 2013), Birmingham, 03-04 July 2013.
- 5. A Hernandez Michelena, J Graham-Jones, J Summerscales and W Hall Resin Transfer Moulding (RTM) production of surfboard fins with natural fibres. Part II: experimental work testing (abstract and 209) 2nd International Conference on Natural Fibers – From Nature to Market (ICNF2015), São Miguel – Azores ~ PORTUGAL, 27-29 April 2015, oral presentation. CD-ROM Session 13.

ISBN 978-989-98468-4-5. Book of Abstracts pages 265-266, ISBN 978-989-98468-5-2.

 A Hernandez Michelena, J Graham-Jones, J Summerscales and W Hall, <u>Eco-friendly epoxy resin/flax fibre composite system as a material for surf boards</u> <u>production</u>
 (abstract 210)
 2nd International Conference on Natural Fibers – From Nature to Market (ICNF2015), São Miguel – Azores ~ PORTUGAL, 27-29 April 2015, oral presentation. CD-ROM Session 13. ISBN 978-989-98468-4-5. Book of Abstracts pages 267-268, ISBN 978-989-98468-5-2.

- 7. A Hernandez Michelena, J Graham-Jones, J Summerscales and W Hall <u>Eco-friendly epoxy resin/flax fibre composite system interface improvement with the</u> <u>chemical treatments of the flax surface and resin</u> formulation 2nd International Conference on Natural Fibers – From Nature to Market (ICNF2015), São Miguel – Azores ~ PORTUGAL, 27-29 April 2015, no presentation. Book of Abstracts pp 269-270. ISBN 978-989-98468-5-2.
- A Hernandez Michelena, J Graham-Jones, J Summerscales and W Hall. *Silane Modification of the Flax/Epoxy System Interface*. Procedia Engineering, Volume 200, 2017, Pages 448-456.
 DOI: http://dx.doi.org/10.1016/j.proepg.2017.07.063

DOI: <u>http://dx.doi.org/10.1016/j.proeng.2017.07.063</u> PEARL: <u>http://hdl.handle.net/10026.1/10050</u>

Word count of main body of thesis: 39479 words

14 Signed.....

NATURAL FIBRE REINFORCED COMPOSITE MATERIALS

Abstract

Nowadays, due to the global warming and pollution issues, sustainable materials must be considered. Composites materials can offer excellent mechanical performance with low weight, hence saving fuel. However, most of the composites systems are based on petrochemicals, and natural materials may be a better option; for example, fibre reinforcements from plant stems (bast) and bio-based resins are becoming available commercially.

One of the major inconveniences of the NFRP is the moisture absorption in marine environments. This problem is usually solved increasing the fibre-matrix bonding. Commercially available fibre systems are chemically treated in order to tackle this issue, but those treatments produce environmental burdens. This thesis seeks to develop an environmentally-friendly, commercially competitive and easily performed treatment methodology for improving the NFRP mechanical properties.

The proposed silane-in-hardener method, adding coupling agent to the hardener, rather than direct treatment of the fibres in solvent, can eliminate solvent use, considerably reducing environmental burdens. This new proposal also reduces process time and improves the composite mechanical performance, resulting in commercial competitiveness. The primary research question in this thesis is can sensible NFRP properties be achieved with silane-in-hardener replacing prior treatment of reinforcements?

Flax fibre and epoxy resin were selected for the experimental campaign. First, flax fibre was mercerised in different immersion-time and concentrations conditions, and the resulting mechanical performance of composite systems evaluated; from the campaign the best mercerised system was selected. In a second stage, raw flax fibre and best mercerised flax fibre were silanised and resulting composite system mechanical properties evaluated. In a third stage, silane was directly added to the epoxy resin and the mechanical properties evaluated together with raw flax fibre. In the final stage, the developed silane in resin method was applied to flax/bio-epoxy system and compared with the basic system in order to evaluate its real improvement.

Additionally, water immersion tests were performed to the silanised flax fibre/epoxy resin system in order to evaluate whether the moisture resistance was increased or not.

From the experimental campaign, it was concluded that the fibre mercerisation process reduces the resulting composite mechanical performance. First, whenever the flax is immersed in the NaOH solution the fibre swells, impeding the flax fibre correct wetting, reducing as a result the composite mechanical performance. Second, because at long immersion-times and concentrations the flax fibre starts to degrade, reducing the composite performance. Similarly, when the flax was immersed in a silane solution, fibre swelling was also obtained, reducing mechanical performance. In contrast when the 1% w/w silane was added to the resin system, the swelling was avoided, and the objective interfacial properties enhanced, getting as a result static mechanical properties improvement.

However, when the enhanced and base systems moisture ingress resistance was evaluated, the values difference was not as representative as expected.

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Copyright statement Title page Acknowledgements Author's declaration Publications arising from this study (Book chapter & Conference papers) Word count

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Nomenclature (list of symbols)

| AB | Acciona Blades |
|------------|-----------------------------------------------------------------|
| ABS | Acrylonitrile/Butadiene/Styrene |
| AF | Aramid Fibres |
| Af | Fabric areal weight |
| AFM | Atomic Force Microscopy |
| Aii | Extensional stiffnesses |
| ANC | Acrylonitrile Copolymer |
| AP | Acidification Potential |
| Aramids | Aromatic polyamides |
| ASTM | American Society for Testing and Materials |
| АТР | Aquatic Toxicity Potential |
| BCM | Bulk Compression Moulding |
| BET-SSA | Brunauer-Emmett-Teller specific surface area |
| Bii | Bending-extension counting stiffnesses |
| BDA | Bichbenol A |
| | Specific heat |
| | Longitudinal Compression test |
| | Transverse Compression test |
| CE | Composite Evolution |
| CE | Composite Evolution |
| CEM | Carbon Fibres |
| | Continuous Fildment Mat |
| CFRP | Carbor Fibre Reinforced Polymer |
| | Constant parameter |
| CLSM | Confocal (monochromatic) Laser Scanning Microscope |
| CLI | |
| cm | Centimetre |
| CM | |
| CMC | Ceramic Matrix Composites |
| CNSL | Cashew Nut Shell Liquid |
| CNI | Carbon Nanotubes |
| CP-MAS NMR | Cross-Polarised Magic Angle Spinning Nuclear Magnetic Resonance |
| CPS | Centripoise |
| CRAG | Composite Research Advisory Group |
| C-RTM | Compression RTM |
| CSA | Cross-Sectional Area |
| CSM | Chopped Strand Mat |
| CT | Computer Tomography |
| CTFE | Polychlorotrifluoroethylene |
| CV | Coefficient of Variation |
| d | Diameter |
| D | Mass diffusivity |
| DCM | Dough Compression Moulding |
| deg | Degree |
| denier | g/9000m |
| DGEBA | Diglycidyl Ether of Bisphenol A |
| DIC | Digital Image Correlation |
| Dij | Bending stiffnesses |
| DIN | Deutsches Institut für Normung |
| DMA | Dynamic Mechanical Analysis |
| DMTA | Dynamic-Mechanical Thermal Analysis |
| dpf | Deniers per filament |
| DSC | Differential Scanning Calorimetry |
| E | Elastic modulus |
| E- | Electrical |
| E1 | Elastic modulus in 1 direction |

| ECF | Excel Conditional Formatting |
|------------------|-----------------------------------------------------|
| FCM | Extrusion Compression Moulding |
| ECH EFW/ | Enoxy Equivalent Weight |
| | The fibre modulue |
| | Fauilibrium Maisture Content |
| EMC | Equilibrium Moisture Content |
| EP | Eutrophication Potential |
| Ex | Elastic modulus in x direction |
| EZP | Electrokinetic zeta-potential |
| F | Formaldehyde |
| FEA | Finite Element Analysis |
| FFM | Finite Element Modelling |
| FE | Fibre Failure |
| | The force required to achieve dehending |
| | Fibre Civelling ratio |
| 15W | Fibre Swelling ratio |
| FTIR | Fourier Transform InfraRed |
| fx | Apparent diameter |
| g | Gram |
| G | Shear modulus |
| GC-MS | Chromatography-Mass Spectroscopy |
| GE | Glass fibres |
| CEDD | Class Fibre Deinferend Delumer |
| GFRP | Mada II fracture touchage |
| GII | Mode II fracture toughness |
| GPa | Giga Pascals |
| gsm | Grams per suquare meter |
| GVI | General Visual Inspection |
| GWP | Global Warming Potential |
| h | Hour |
| H ₂ O | Water |
| ΗΔΙΤ | Highly Accelerated Life Testing |
| | High Donsity Polyothylono |
| | High Impact DC |
| HIPS | High Impact PS |
| HIP | Human Toxicity Potential |
| НТРВ | Hydroxyl Terminated Polybutadiene Rubber |
| IBT | Isocyanate group back titration |
| ICM | Injection Compression Moulding |
| ICSTM | Imperial College in London |
| IFF | Inter Fibre Failure |
| IECC | Inter Fibre Fandre |
| | Illingia Institute of Technology Deserved Institute |
| | Ininois Institute of Technology Research Institute |
| ILSS | Interlaminar Shear Strength |
| IM | Injection Moulding |
| IR | Infra-Red |
| ISO | International Organization for Standardization |
| Iso | Isophthalic |
|] | Joule |
| ĸ | Kilo |
| ĸ | Permeability |
| K K | Thermal conductivity |
| | |
| K1/K2 | Fabric characteristic constants |
| kg | Kilogram |
| kJ | KiloJoule |
| Kth | Layer position |
| L | Fibre length |
| Lc | Fibre critical length is defined |
| | Life Cycle Assessment |
| LCM | Liquid Composite Moulding |
| | Liquid Composite Moulding |
| LUPE | Low Density Polyethylene |

| Le | The embedded length of the fibre |
|----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| LF | Lignocellulosic Fibres |
| LOI | Loss On Ignition |
| LTPT | Low Temperature Plasma Treatment |
| LV | Liquid/Vapour |
| m- | meta |
| m | Meter |
| М | Molar |
| MACO | Maleated Castor Oil |
| MAPP | Maleic Anhydride Polypropylene |
| MFKP | Methyl Ethyl Ketone Peroxide |
| min | Minute |
| Mm | Maximum moisture content |
| mm | Millimetre |
| MMC | Metal Matrix Composites |
| MPa | Mega Pascals |
| mDa c | Milli Pascal ner second |
| M. | Moisturo untako at timo t |
| N | Nowton |
| | Net Applicable |
| N/A N6 | Nylon type 6 |
| | Nylon type 6 |
| | Nyion type oo |
| | Soulull llyuloxide |
| | Non-Crimp Fabrics |
| | Natural Fibres |
| NFRP | Natural Fibre Reinforced Polymer |
| NOL | Naval Ordnance Laboratory |
| NRADP | Non-Renewable/Ablotic Resource Depletion |
| 0- | Ortho |
| 0 | |
| ODP | Ozone Depletion Potential |
| ODP OH | Ozone Depletion Potential Hydroxyl |
| ODP OH OM | Ozone Depletion Potential Hydroxyl Optical Microscope |
| ODP OH OM Ortho | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic |
| ODP OH OM Ortho P | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page |
| ODP OH OM Ortho P p- | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para |
| ODP OH OM Ortho P p- P | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol |
| ODP OH OM Ortho P p- P PAA | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) |
| ODP OH OM Ortho P P- P PAA PAN | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) Polyacrilonitrile |
| ODP OH OM Ortho P P- PAA PAN PBAT | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) Polyacrilonitrile Poly(Butylene Adipate-co-Terephthalate) |
| ODP OH OM Ortho P P- PAA PAN PBAT PBS | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) Polyacrilonitrile Poly(Butylene Adipate-co-Terephthalate) Poly (Butylene Succinate) |
| ODP OH OM Ortho P P- PAA PAN PBAT PBS PC | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) Polyacrilonitrile Poly(Butylene Adipate-co-Terephthalate) Poly (Butylene Succinate) Polycarbonate |
| ODP OH OM Ortho P P- PAA PAN PBAT PBS PC PCL | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) Polyacrilonitrile Poly(Butylene Adipate-co-Terephthalate) Poly (Butylene Succinate) Polycarbonate Polycaprolactone |
| ODP OH OM Ortho P P- P PAA PAN PBAT PBS PC PCL PEI | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) Polyacrilonitrile Poly(Butylene Adipate-co-Terephthalate) Poly (Butylene Succinate) Poly (Butylene Succinate) Polycarbonate Polycaprolactone Polyetherimide |
| ODP OH OM Ortho P P- PAA PAA PAN PBAT PBS PC PCL PEI PET | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) Polyacrilonitrile Poly(Butylene Adipate-co-Terephthalate) Poly (Butylene Succinate) Poly (Butylene Succinate) Polycarbonate Polycaprolactone Polycaprolactone Polyetherimide Polyethylene terephthalate |
| ODP OH OM Ortho P P PAA PAA PAA PAN PBAT PBS PC PCL PEI PET PET | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) Polyacrilonitrile Poly(Butylene Adipate-co-Terephthalate) Poly (Butylene Succinate) Poly (Butylene Succinate) Polycarbonate Polycaprolactone Polycaprolactone Polyetherimide Polyethylene terephthalate Polyethylene terephthalate |
| ODP OH OM Ortho P PAA PAN PBAT PBS PC PCL PEI PET PET PET PF | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) Polyacrilonitrile Poly(Butylene Adipate-co-Terephthalate) Poly (Butylene Succinate) Poly (Butylene Succinate) Polycarbonate Polycaprolactone Polycaprolactone Polyetherimide Polyethylene terephthalate Polyethylene terephthalate Perimeter of the natural fibre |
| ODP OH OM Ortho P PAA PAA PAN PBAT PBS PC PCL PEI PET PET PET PET PF | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) Polyacrilonitrile Poly(Butylene Adipate-co-Terephthalate) Poly (Butylene Succinate) Poly (Butylene Succinate) Polycarbonate Polycaprolactone Polycaprolactone Polyetherimide Polyethylene terephthalate Polyethylene terephthalate Perimeter of the natural fibre Phenolic resin |
| ODP OH OM Ortho P P- P PAA PAN PBAT PBS PC PCL PEI PET PET PET PET PET PF PHB | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) Polyacrilonitrile Poly(Butylene Adipate-co-Terephthalate) Poly (Butylene Succinate) Poly (Butylene Succinate) Polycarbonate Polycarbonate Polycaprolactone Polyetherimide Polyethylene terephthalate Polyethylene terephthalate Polyethylene terephthalate Perimeter of the natural fibre Phenolic resin Polyhydroxybutyrates |
| ODP OH OM Ortho P P- P PAA PAN PBAT PBS PC PCL PEI PET PET PET PET PF PF PHB PLA | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) Polyacrilonitrile Poly(Butylene Adipate-co-Terephthalate) Poly (Butylene Succinate) Poly (Butylene Succinate) Polycarbonate Polycaprolactone Polycaprolactone Polyetherimide Polyethylene terephthalate Polyethylene terephthalate Polyethylene terephthalate Perimeter of the natural fibre Phenolic resin Polyhydroxybutyrates Poly Lactic Acid |
| ODP OH OM Ortho P P PAA PAN PBAT PBS PC PCL PEI PET PET PET PET PF PF PHB PLA PMC | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) Polyacrilonitrile Poly(Butylene Adipate-co-Terephthalate) Poly (Butylene Succinate) Poly (Butylene Succinate) Polycarbonate Polycarbonate Polycaprolactone Polyetherimide Polyethylene terephthalate Polyethylene terephthalate |
| ODP OH OM Ortho P P- P PAA PAA PAN PBAT PBS PC PCL PEI PET PET PET PET PET PET PET PET PET PET | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) Polyacrilonitrile Poly(Butylene Adipate-co-Terephthalate) Poly (Butylene Succinate) Poly (Butylene Succinate) Polycarbonate Polycaprolactone Polycaprolactone Polyetherimide Polyethylene terephthalate Polyethylene terephthalate |
| ODP OH OM Ortho P P- P PAA PAA PAN PBAT PBS PC PCL PEI PET PET PET PET PET PET PET PET PET PET | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) Polyacrilonitrile Poly(Butylene Adipate-co-Terephthalate) Poly (Butylene Succinate) Poly (Butylene Succinate) Polycarbonate Polycaprolactone Polycaprolactone Polyetherimide Polyethylene terephthalate Polyethylene terephthalate Polyethylene terephthalate Perimeter of the natural fibre Phenolic resin Polyhydroxybutyrates Poly Lactic Acid Polymer Matrix Composite Poly Methyl MethAcrylate Polymethylmethacrylate |
| ODP OH OM Ortho P P- P PAA PAA PAN PBAT PBS PC PCL PEI PET PET PET PET PF PF PHB PLA PMC PMMA PMMA POCP | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) Polyacrilonitrile Poly(Butylene Adipate-co-Terephthalate) Poly (Butylene Succinate) Poly (Butylene Succinate) Polycarbonate Polycaprolactone Polycaprolactone Polyetherimide Polyethylene terephthalate Polyethylene terephthalate Polyethylene terephthalate Perimeter of the natural fibre Phenolic resin Polyhydroxybutyrates Poly Lactic Acid Polymer Matrix Composite Poly Methyl MethAcrylate Polymethylmethacrylate Photochemical Oxidants Creation Potential |
| ODP OH OM Ortho P P- P PAA PAN PBAT PBS PC PCL PEI PET PET PET PET PF PF PHB PLA PMC PMMA PMMA PMMA POCP POM | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) Polyacrilonitrile Poly(Butylene Adipate-co-Terephthalate) Poly (Butylene Succinate) Poly (Butylene Succinate) Polycarbonate Polycaprolactone Polycaprolactone Polyetherimide Polyethylene terephthalate Polyethylene terephthalate Polyethylene terephthalate Perimeter of the natural fibre Phenolic resin Polyhydroxybutyrates Poly Lactic Acid Polymer Matrix Composite Poly Methyl MethAcrylate Polymethylmethacrylate Photochemical Oxidants Creation Potential Acetal |
| ODP OH OM Ortho P PAA PAA PAN PBAT PBS PC PCL PEI PET PET PET PET PF PF PHB PLA PMC PMMA PMMA POCP POM PP | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) Polyacrilonitrile Poly (Butylene Adipate-co-Terephthalate) Poly (Butylene Succinate) Poly (Butylene Succinate) Polycarbonate Polycaprolactone Polycaprolactone Polyetherimide Polyethylene terephthalate Polyethylene terephthalate Polyethylene terephthalate Perimeter of the natural fibre Phenolic resin Polyhydroxybutyrates Poly Lactic Acid Polymer Matrix Composite Poly Methyl MethAcrylate Polymethylmethacrylate Photochemical Oxidants Creation Potential Acetal Polynropylene |
| 0 0DP 0H 0M 0rtho P p- P PAA PAA PAA PAA PBAT PBS PC PCL PEI PET PET PET PF PF PHB PLA PMC PMMA PMMA POCP POM PP PP PPS | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) Polyacrilonitrile Poly(Butylene Adipate-co-Terephthalate) Poly (Butylene Succinate) Poly (Butylene Succinate) Polycarbonate Polycaprolactone Polyetherimide Polyethylene terephthalate Polyethylene terephthalate Polyethylene terephthalate Polyethylene terephthalate Perimeter of the natural fibre Phenolic resin Polyhydroxybutyrates Poly Lactic Acid Polymer Matrix Composite Poly Methyl MethAcrylate Polymethylmethacrylate Photochemical Oxidants Creation Potential Acetal Polypropylene Polypropylene Polyphenylene sulphide |
| 0 0 0 0 0 0 0 0 0 0 0 0 0 0 | Ozone Depletion Potential Hydroxyl Optical Microscope Orthophthalic Page Para Phenol Poly (Acrylic Acid) Polyacrilonitrile Poly (Acrylic Acid) Polyacrilonitrile Poly (Butylene Adipate-co-Terephthalate) Poly (Butylene Adipate-co-Terephthalate) Poly (Butylene Succinate) Poly (Butylene Succinate) Polycarbonate Polycaprolactone Polycaprolactone Polycaprolactone Polyethylene terephthalate Polyethylene terephthalate Polyethylene terephthalate Polyethylene terephthalate Polyethylene terephthalate Polyethylene terephthalate Polyethylene terephthalate Polyethylene terephthalate Polyethylene terephthalate Polyethylene terephthalate Polyhydroxybutyrates Poly Lactic Acid Polymer Matrix Composite Poly Methyl MethAcrylate Photochemical Oxidants Creation Potential Acetal Polypropylene Polyphenylene sulphide Particle Reinforced Composites |

| Prepreg | Pre(im)preg(nated) |
|----------------|-----------------------------------------------------------------|
| PS | Polystyrene |
| PTFE | Polytetrafluoro Ethylene |
| PU | Polyurethane |
| PVA | Polyvynil Acetate |
| PVB | Polyvinyl Butyral |
| PVC | Polyvinyl Chloride |
| 0 | one-dimension volumetric flow rate |
| R | Radical |
| RH | Relative Humidity |
| RIFT | Infusion under Elexible Tooling |
| RIFT | Resin Infusion under Flexible Tooling |
| RRV | Resin-Rich Volumes |
| RT | Room Temperature |
| RTM | Resin Transfer Moulding |
| RTP | Reinforced Thermosetting Plastic |
| c | Second |
| S C | Shear strength |
| S | Shear Test |
| SAN | Sturene/Acrylonitrile |
| | Seemann Composite Perin Infusion Moulding Process |
| SD | Standard Deviation |
| SEM | Scanning Electron Microscony |
| | Scanning/Transmission Electron Microscopes |
| | Scalining/ Hansinission Liection Microscopes |
| | 4.4' (1.2 dipropul tetramothyldicilovano) bic 2 mothowynhonol |
| | 4,4 -(1,5-ulpropyr-tetrametryluisiloxarie)bis-2-metrioxyprienor |
| SL | Soliu/Liquiu Shaat Maulding Compound |
| SMC | |
| 5V T | Soliu/vapour |
| | Temperature |
| 1 • | Time |
| | lime |
| 100 | |
| 1900 | Transverse Tensile test |
| | roluene Dilsocyanate |
| tex | g/km |
| Ig | |
| TGA - | Inermo-Gravimetric Analysis |
| Im T T STAR | Melting Temperature |
| TOF-SIMS | lime-of-flight secondary ion mass spectrometry |
| | Inermoplastic |
| | Inermosetting |
| - | Inrough-Inickness |
| UD | Unidirectional |
| UP | Unsaturated Polyester resin |
| USA | United States of America |
| UI | Ultrasound lesting |
| UIS | Ultimate Tensile Strength |
| UV | UltraViolet |
| UΣ | Microstrain |
| V | Volume |
| VARIM | vacuum-Assisted Resin Transfer Moulding |
| VB | vacuum Bagging |
| VCXPS | X-ray Photoelectron Spectroscopy |
| VE | Vinyl Ester resin |
| Vf | fibre Volume fraction |
| W | Watt |

| w | Weight |
|----------------|---------------------------------------------|
| W ₀ | Weight of dry specimen |
| WEEE | Waste Electrical and Electronic Equipment |
| wt | Weight |
| Wt | Weight of wet specimen at time t |
| Х | Longitudinal strength |
| XPS | X-ray Spectroscopy |
| Y | Transverse strength |
| γ | Wenzel's contact angle |
| ε' | Strain at failure |
| ε _x | Strain in x direction |
| θγ | Young's contact angle |
| v | Poisson's Ratio |
| ρf | Fibre density |
| σ' | Strength |
| σ_1 | Tensile strength in 1 direction |
| σ _f | Fibre strength at the critical fibre length |
| σ _x | Tensile strength in x direction |
| т | Fibre-matrix shear strength |
| \$ | Dollar |
| % | Percent |
| € | Euro |
| 0 | Degree |
| °C | Celsius degree |
| μm | Microns |
| 2D | Two dimensions |
| 3D | Three dimension |
| a and β | Weibull shape and scale parameters |
| | |

Chapter 1: Introduction

This thesis reports a research study on natural fibre (NF) reinforced polymer (NFRP) thermoset matrix composite materials. The low wettability and poor fibre-matrix adhesion lead to the mechanical performance of the NF/thermoset composites being lower than expected from consideration of each component characteristic. The research project sought to improve the flow and fibre/matrix interfacial properties by adding silane to the NFRP system in order to obtain the maximum, or most balanced, performance from each component, with the ultimate aim of producing a commercially competitive product that can be utilised for eco-surfboard and eco-surfboard fin production. It was therefore most important to discover how the NF/epoxy/silane composite would respond under rigorous experimental testing. In previous studies, referenced in the literature, NF surfaces have been treated directly with silanes without modifying the epoxy matrix. It is suspected that direct NF treatment leads to fibre swelling thus decreasing the NFRP performance. The present study aims to show that when silane is added directly to the matrix an enhanced NFRP is obtained. This result is a positive contribution to knowledge.

In this study, flax fibre is the principal reinforcement component. Flax is selected because it is claimed to have the best mechanical performance. In this particular study, commercially available Biotex and Lineo woven fabrics have been utilised. For the production of the NFRP Huntsman petrochemical epoxy and SuperSap bio-epoxy resins were selected, because the epoxy resin has good mechanical performance. Huntsman petrochemical resin was selected for the first stage on the basis of availability and price. The SuperSap bio-content formulation was selected for the second stage, as a potentially sustainable matrix for NFRP. The composite was manufactured using either hand-lamination or infusion techniques.

The properties of the unmodified SuperSap/flax composite system were taken as the reference case. In this investigation, flax fibre chemical treatments and epoxy resin modification were performed in order to improve the NFRP interfacial and flow properties. Modified systems properties were compared with the reference case.

The experimental procedures were undertaken in the following order:

- 1. Undertake the fibre chemical treatment
- 2. Evaluate the flow behaviour in panel production/evaluation of mechanical properties
- 3. ANOVA analysis of the generated data in the mercerisation process
- 4. Utilise the optical microscope to establish the fibre swelling evaluation
- 5. Propose a flow **model** for application
- 6. Apply the developed knowledge in optimised system selection
- 7. Evaluate optimised system moisture aging properties

Mercerisation and silane treatments were selected for the flax fibre interfacial property improvement. In contrast, bio-epoxy formulations were modified with silane added directly to the epoxy hardener.

Apart from the flow properties characterisation, the mechanical properties of the different systems were evaluated. Different modifications were evaluated with the basic flax/epoxy system.

This study seeks to improve the interfacial properties of the NFRP. The improvement of the interfacial properties are directly correlated to the mechanical static, dynamic and long term performance properties. For this reason, the composite systems developed in the first part of the PhD study were tested at laboratory scale and in real use.

The main objective was to correlate the NF chemical treatment with the composite flow and mechanical properties. The following correlation scheme is proposed:



Figure 1.1: Correlation scheme of the different factors

A number of methods have been applied to optimise the fibre/matrix bond within NFRP. However, for reasons of cost, the extent of fibre processing should be limited to avoid a premium price, and for reasons of sustainability, the processes and the by-products of fibre processing should be minimised.

The research is documented in the following 11 Chapters and nine Appendices:

- **Chapter 1: Introduction** establishes the major research question of the thesis and details the chapter contents;
- Chapter 2: Conventional composites provides an overview of the composite materials;
- **Chapter 3: Bio-composites** provides an overview of the composites produced with bio components, paying special attention to flax fibre and bio-epoxies;
- **Chapter 4: The interface in natural fibre composites** discusses the different NF treatments present in the literature focusing on mercerisation and silanisation;
- Chapter 5: Composite design manufacturing testing image analysis The present chapter reviews composite design, manufacturing, testing techniques and procedures available in the market. Image analysis for materials characterisation were shortly described;
- Chapter 6: Composite durability in marine environment discusses the addition of silane to the flax/bio-epoxy system in order to improve marine aging resistance;
- **Chapter 7: Research methodology** describes the fibres, resins, process materials and selected chemicals; then the selected manufacturing techniques;
- Chapter 8: Results the results of the experimental tests are presented;
- Chapter 9: Discussion the results are analysed and discussed;
- Chapter 10: Conclusions establishes the principal outcomes of the research project;
- **Chapter 11: Future Research** evaluates guidelines for further research interests and projects for the potential commercialisation of the product.
- * **Appendix A** collects all product technical data sheets (TDS);
- * Appendix B collects all the experimental data;
- * **Appendix C** shows the swelling monitoring pictures;
- * Appendix D compiles all published articles;
- * Appendix E shows all the moisture aging campaign experimental data;
- Appendix F documents explains the procedure for the calculation of mechanical tests uncertaintly;
- * Appendix G contains all the numbers used in the mechanical test results ANOVA analysis.

2. Conventional composites

2.1. INTRODUCTION

Chapter 2 describes the components in conventional composite materials and their overall characteristics. Composite reinforcements, matrix systems and surface treatments for interface optimisation are each individually described. Additional fabric reinforcement forms are also considered, specifically fabric styles (woven, knitted, stitched or random) with particular reference to fibre orientation and achievable fibre volume fraction (V_f).

A polymer matrix composite structure is basically a matrix phase charged with a reinforcement which gives stiffness and strength to the structure. Stresses are transferred between the matrix and the reinforcement. The fibre/matrix interface is as important as the fibre and the matrix. The interface is the distinct boundary between the reinforcement and the matrix. This concept would be deeply studied in Chapter 4.

2.2. CONTINUOUS REINFORCEMENTS: GLASS

E-glass fibres are the principal reinforcement used in composite production, since the product is low in cost and is mechanically acceptable in both strength and stiffness (Mallick, 2010). Carbon fibre (CF) is used only in particular cases where the high elastic modulus attracts elevated price in comparison to glass fibre (GF). CF finds application in materials that require high stiffness, low thermal expansion, high thermal capacity, electrical conductivity and electromagnetic shielding properties. The comparative mechanical and physical properties of E-glass and CF are shown in Table 2.1.

| Table 2.1: Comparative table of the E-glass and (| CF pro | perties | (Mallick. | 2010) | 1 |
|---------------------------------------------------|--------|---------|-----------|-------|---|
| | | | (| ====; | |

| | E-glass fibre | PAN carbon fibre |
|-----------------------------------------------------------------------|---------------|---------------------------|
| Filament diameter (µm) | 10 | 7.2 |
| Relative density (vs H ₂ O) | 2.54 | 1.81 |
| Axial tensile modulus (GPa) | 72.4 | 228 |
| Axial tensile strength (MPa) | 1725 | 3800 |
| Coefficient of thermal expansion (10 ⁻⁶ °C ⁻¹) | 5 | -0.6 (axial), 10 (radial) |
| Axial thermal conductivity (W/m·°C) | 1.04 | 15 |
| Axial electrical resistivity (ohm cm) | 0.1 | 0.0017 |

GF is the most commonly used reinforcing inorganic fibre because of its high performance and very economical price (Khazanov, Kolesov, & Trofimov, 1995). GF is produced by melting oxides from sand together with other additives in large furnaces at very high temperatures, usually between 1000 and 1800°C (Dwight, 2000). After the melting process, this viscous glass is bushed to produce thousand-filament yarns.

In accordance to the components used in the production of each kind of glass fibre, they are denominated with a specific nomenclature. For example, the most common is designated "Electrical" E-glass, since at the beginning it was produced for electrical insulators and printed circuit boards. Other examples are A, D, S or AR glasses, all depending on the composition and manufacturing temperatures of production.

The general properties of the major GF are shown in Table 2.2.

| | E-glass | R-glass | S-2 glass |
|------------------------------|---------|---------|-----------|
| Relative density ASTM 1505 | 2.58 | 2.54 | 2.46 |
| Softening point ASTMC338(°C) | 846 | 952 | 1056 |
| Tensile strength 23°C (GPa) | 3445 | 4135 | 4890 |
| Tensile modulus 23°C (GPa) | 72 | 86 | 87 |
| Elongation (%) | 4.8 | 4.8 | 5.7 |

Table 2.2: Physical properties of reinforcement glasses (Fecko, 2006)

In real production when the filament diameter measurement is not practical, the weight of a bundle of fibre per unit length is used according to units of "tex" (g/km) or "denier" (g/9000m) (Thomason, 1995).

2.3. MATRICES - THERMOSET RESINS

Thermosetting (TS) resins are normally oligomers (low molecular weight polymers) which are usually liquid before mixing with a second component to crosslink the different polymer chains and hence solidify the mixture. This can be achieved by either an addition or a condensation/ring-opening reaction. In some cases, heat is applied to the mixture to accelerate the process. Process temperatures are typically between ambient and 200°C. T_g (Glass transition temperature) is the temperature where segmental motion is frozen out of the

polymer chain. This is one of the greatest advantages of using the thermoset resins. The T_g normally follows the cure temperature, so it should be possible to produce creep-free composites with much lower process energies than for a TP (thermoplastic) matrix.

Resin viscosity depends directly on the molecular weight of the polymer and the temperatures used in the process. Viscosity increases with the molecular weight (Ghijsels, Groesbeek & Raadsen, 1984). The temperature increment can produce the opposite effect in the flow index of the resin; under higher temperatures the resin would flow more easily, but as cure progresses the viscosity increases. Figure 2.1 illustrates the viscosity evolution of two different epoxy formulations along the curing process. In this case, the resin curing accelerates when the system reaches 40°C, as a result the viscosity increases with progress of cure. This viscosity rise is not shown in Figure 2.1.



Figure 2.1: Instantaneous viscosity evolution according to temperature (Courtesy of Acciona Blades)

The most common systems would use the following formulations:

- Unsaturated Polyester resin (UP) diluted with styrene and cured with $\sim 1\%$ catalyst
- Vinyl Ester resin (VE) diluted with styrene and cured with \sim 1% catalyst
- Epoxy resin cured with hardener in stoichiometric proportions typically a 3:1 to 5:1 mix ratio

2.3.1. Epoxy resin

Epoxy is considered to be a high standard resin for composite material production. The cured resin has great mechanical, thermal, corrosion resistance and chemical properties. Due to its internal structure, epoxy has good electrical properties (Park et al., 2007). Finally, epoxy based adhesives are very important in the current market because of their adhesive capabilities in different substrates (Bhuniya & Maiti, 2002).

The chemistry of the epoxy resin is studied in detail in this chapter since the research project is based on a flax fibre/bio-epoxy resin composite.

The characteristic epoxy group is based on the highly-strained "epoxy ring" (Figure 2.2) a highly reactive three-member ring composed of two carbon atoms and one oxygen atom (Lee, Jang, Hong, Hwang, & Kim, 1999).



Figure 2.2: Epoxy group basic structure

The first step in the production of the epoxy resin is based on the reaction between bisphenol-A and epichlorohydrin (Wang & Zhang, 2004). Bisphenol-A is the most commonly used diol. The result of this reaction is the formation of diglycidyl ether of bisphenol A (DGEBA) a product prepolymer (Figure 2.3). There can be other options in the resin backbone production, such as the production of Bisphenol F or S.



Figure 2.3: DGEBA product chart

Two great advantages in the manufacture of epoxy composites are the control of the curing cycle and low shrinkage of the resin once cured (Wan, Bu, Xu, Li, & Fan, 2011); the SuperSap resin used in the experimental part of the project shrinks by only 2% (linear shrinkage). The low shrinkage obtained in the curing of the composite part also reduces residual stresses and thus maintains dimensional stability (Nawab et al., 2012; Shah & Schubel, 2010; Li, Potter, Wisnom & Stringer, 2004).

In contrast, one disadvantage could be the elevated viscosity (800-900 mPa·s) of the epoxy prepolymer that might impede the processing and the correct fibre wetting; this disadvantage might be even higher in the flax fibre NF wetting process. Due to the elevated solid content of the epoxy formulation, its market price is also higher. Furthermore, the epoxy process is long and difficult to perform in hot and humid environments due respectively to exotherm, accelerators and side reactions, and the curing process has to be controlled (Montserrat, Flaque, Calafell, Andreu & Malek, 1995; Wan, Bu, Xu, Fan & Li, 2011; Huang et al., 2012).

The selection of the curing agent is dependent on the curing temperature, the desired Tg or the final mechanical properties. The calculation of the quantity of the curing agent depends on the concept of the Epoxy Equivalent Weight (EEW) (Ellis, 1993). This concept provides the information about the epoxy content in the resin, the number of epoxy equivalents in a kilogram of resin or number of epoxy groups grams per mole of resin.

$$EEW = \frac{Epoxy \ Equivalent}{1kg \ of \ resin} \ or \ \frac{Epoxy \ grams}{mol \ of \ resin}$$

Elevated EEW value means that there are many available active points in the epoxy backbone, giving as a result highly reticulated epoxy. In contrast, when the EEW is low the cross-linking (reticulation) is also low. According to this number, the stoichiometry between the first component and the curing agent is determined, thus when stoichiometry is followed the maximum epoxy performance is achieved. Normally, the proportions of the first component / curing agent are 3:1 or 5:1. Occasionally, epoxy commercial systems have more than two components, and then the proportion is referred to by the first component. The epoxy system

curing agents are normally divided into two groups, the amines and the anhydrides. The SuperSap system uses a trifunctional amine for the reticulation process.

In the transformation from liquid prepolymer to infusible solid resin, the material passes through different stages. One of the most important is the "gelation" stage (Li, Li & Meng, 2012; Lin & Wang, 1994). Viscosity increment considerably decreases resin processability influencing especially on the flow process, such as in infusion or pultrusion, but not greatly influencing manual lamination. The gelation or gel-point usually occurs in the range of 55-80% of the complete process. One of the problems of gelation is that auto-acceleration of the reaction can happen suddenly, thus increasing the reaction temperature (exotherm) which can lead to final piece degradation (Jin & Park, 2012). This energy release can feed into new covalent bond production, creating in the process more free energy and advancing the reaction without much control.

In each curing process, different conversion rates (degree of cure) (α) are obtained; this value is affected by many factors such as the curing temperature and time, although the principal factors will be the type of curing agent used and the quantity that is added. In any case, the full conversion is 100% (Kim & Lee, 2002; Min, Stachurski & Hodgkin, 1993). In certain examples, for example fabric prepregs, the curing process may be halted to allow for the process to be continued later. The formulation used in this technique is referred to as a B-stage resin.

In the serial production of a composite design, it is normal for the part product to be demoulded when the Tg of the resin has been achieved. The mould is then free to allow production to continue. The free-standing de-moulded part can then be post-cured as an extra step in the completion of the curing process in order to achieve the definitive Tg. In this PhD research project for example the panels were post-cured after the tabs had been glued, directly affecting the Tg value. In Table 2.3 some epoxy systems typical values are shown.

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| Resin | Tg(°C) |
|-----------------------|--------|
| DGEBA (Pure)/3,3'-DDS | 184 |
| EPN825/DDS | 222 |
| EPN828/DDS | 211 |
| EPN834/DDS | 186 |
| DGEBA (Pure)/DDM | 176 |
| DER332/DDS | 190 |
| Epon828/DDM | 170 |

Table 2.3: Different epoxy systems Tg temperatures (Varma & Gupta, 2000)

Table 2.4 presents the mechanical properties of a general amine cured epoxy system developed

by using the correct resin/hardener mix rate and processed by the correct curing cycle.

Table 2.4: Epoxy resin general properties (Pham & Marks, 2000).

| Properties | Values at RT (25°C) |
|-------------------------|---------------------|
| Tensile strength (MPa) | 48 |
| Tensile modulus (GPa) | 3.9 |
| Tensile elongation (%) | 1.3 |
| Flexural strength (MPa) | 127 |
| Flexural modulus (GPa) | 3.6 |

2.3.2. Unsaturated Polyester (UP) and Vinyl Ester (VE)

UP are linear poly-condensation based polymers (Mouritz, Gellert, Burchill & Challis, 2001). The poly-condensation reactions are undertaken between acids/anhydrides and diols/oxides in order to produce a prepolymer mixture. Figure 2.4 shows the general reaction between acid and diol to achieve a linear polyester molecule.



Figure 2.4: UP prepolymer general synthesis reaction, where A is a chemical group

The prepolymers are usually brittle solids or viscous liquids. When unsaturated oligomers are combined with reactive solvents, they produce UP resins. For example, the polyester prepolymer is usually formulated with a reactive diluent: styrene. This formulation is widely used since it produces great mechanical performance at a very economical price. The formulation viscosity of these oligomers is usually between 200 and 2000 mPa·s.

The oligomers have unsaturation in the backbone that are the bonding points for the formation of 3D networks (Figure 2.5).



Figure 2.5: Prepolymer structure with the unsaturated double bond

UP resins have been mainly used in marine and building constructions. For example, surfboards and RTM fins are currently manufactured with UP resin reinforced with GF. The atmospheric pressure using vacuum inside a bag allows for the compacting of the different fabric layers. In addition, the resin can cure at room temperature (RT), this formulation is therefore ideal for the construction of large structures such as water tanks or ship hulls (Mouritz, Gellert, Burchill & Challis, 2001). Commercial UP formulations can vary from slow to rapid in their curing programmes according to their formulation.

Whereas UP have unsaturation distributed along the monomer backbone, VE resins have an unsaturated core molecule with terminal (chain-end) unsaturation, similar to the reaction positions in epoxy resins.

The resultant product forms a very fluid resin (200-300 mPa·s) possessing good mechanical properties and at a price that falls between the epoxy and UP resins. The low viscosity of VE is suitable for quick composite production and the process allows for the correct wetting of all the fibres. The curing process is similar to that used for conventional UP, where a free radical producer (catalyst) is added to the prepolymer to start the polymerisation of the resin at RT.

For the VE resin, the reticulation density is higher than in UP resins, resulting in the mechanical properties and chemical resistance being correspondingly higher (Banna, Shirokoff & Molgaard, 2011). Resulting from the chemical resistance of VE resins, chemical container tanks and concrete reinforcements are possible commercial products.

2.4. SURFACE FINISHES ON FIBRES

The application of a sizing medium to the reinforcement fibre surface is absolutely necessary since without coating the fabric the fibres would either break in the manufacturing process or the properties of the fabric-matrix interface would not be sufficient to withstand the appropriate use of the composite (Thomason, 1995). After fibre formation and taking into account the effect of Loss On Ignition (LOI), the commercial fibre sizing content may be between 0.2 and 2% in weight. Ideally, the sizing film should be coated over the complete filament surface. However, it has been shown by using modern surface analytical techniques that the spread of the sizing over the filament surface is heterogeneous and varying from 1nm to 10nm in thickness.

In addition to the above analytical studies, there are references to model applications that attempt to understand the fibre/matrix interface performance in the presence of the sizing (Zhuang, Burghardt & Mäder, 2010). However, few can claim 100% accuracy since it is extremely difficult to determinate the exact performance nature of the sizing agent on any specific batch of reinforcement.

In respect of the present research project, it is vitally important to understand the interfacial properties of the conventional composites as well as the surface finishes of conventional fibres; additionally, the NFRP interfacial properties are deeply studied in Chapter 4. A variety of sizing agents are normally available to coat reinforcement fibres (anti-static, binder, lubricant, film-formers, acidity/pH regulators and coupling agents). The coupling agent is utilised to create a strong interface between the fibre and the matrix. Coupling agents are usually organosilanes which bond well with most (fibre) surfaces (Liu, Thomason & Jones, 2008). A good interface can enhance stress transfer between the matrix and the reinforcement material and improve

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the composite dynamic properties while minimising environmental degradation, especially in water.

In general, silanes have a different functionality at opposite ends of the molecule: the silane functional group reacts with the fibre surface while the organic moiety reacts to form covalent bonds with the matrix (Figure 2.6). The normal assumption is that the glass fibres are vinyl-sized for UP and the aramid or carbon are epoxy-sized for epoxy resins, unless it is specially stated otherwise.



Figure 2.6: General structure of organosilane where R is either vinyl or epoxy functionality (Liu et al., 2008)

2.5. SHORT FIBRES AND FABRIC PRODUCTION

Fibres can classified as short (less than the critical length defined by the Cox shear-lag model), long (discontinuous with length/diameter aspect ratio variously > 200-1000x) or continuous (as used e.g. in textiles/filament winding/pultrusion).

In the fibre manufacturing process, thousands of filaments will be treated and transformed into different fabric configurations. In the case of CF, once the filaments have been conveniently produced they are grouped in tows. Commercially the main sizes of CF tows contain between 3000 and 48000 filaments per tow with 6K and 12K being most popular (NOTE: although K = 1024 in computing and k = 1000 in SI units, capital K is used for thousands of filaments in designating carbon fibre tows). The heaviest tows result from a faster manufacturing process and therefore offer a price reduction; on the other hand, the finest tows are used for high performance applications particularly in the aeronautics and aerospace sector. Smaller tows, with lower areal weight fabrics, result in having smaller resin-rich volumes (RRV) and consequently have greater strength.

Woven fabrics might be easily manipulated and have excellent drapability. However, these fabrics are not usually utilised in high standard applications. Firstly, in woven fabrics the fibre orientation within the fabric is usually limited to 0° and 90° directions. This 0/90 conformation might be woven in different ways, for example, twills or satin (Gandhi & Sondhelm, 2016), surfboards are produced with GF 125g/m² and 190g/m² 0/90 twill woven fabric. Secondly, the crossovers between the tows cause mechanical properties reduction leading to fatigue and long-term strength decrease.

For high performance part production, stitched Non-Crimp Fabrics (NCF) are used. The main advantage is that the fibre can be oriented in the necessary direction depending of the external load requirements. The stitching technology gives additional through-thickness reinforcing capability to the fabric. The long-term properties of NCF are superior to woven fabrics (Adden & Horst, 2010). NCF are costly to produce and their drapability may be less suitable than in woven fabrics.

In the fabric manufacturing process, the first step is to twist the strand in order to obtain the "yarn" for short fibres, but unnecessary misorientation for continuous fibres (Clarke, 2010). Hundreds of yarns feed the loom in order to manufacture different characteristic fabrics with the aim of varying the angle, fabric weight or the weaving tension parameters.

Continuous fibres are selected for fabric production where they are usually arranged unidirectionally, randomly, cross-plied, woven (plain weave, twill and satin), knitted or stitched interlock. Figure 2.7 shows different fabric types in schematic arrangements.

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| unidirectional | plain weave | twill weave | satin weave |
|----------------|-------------|------------------|-------------|
| random | knitted | stitch interlock | cross ply |

Figure 2.7: Fabrics types

(drawn by John Summerscales and reproduced here with his permission)

2.6. SUMMARY

In Chapter 2, the principal types of reinforcements, matrices and interfaces have been described and their utilisation summarised. For the purposes of the present research project, it is clearly necessary to have complete understanding of the conventional composite materials that can be utilised in establishing an enhanced bio-composite system which is the main target of the research.

In respect of the project, it is vitally important to understand the interfacial properties of the conventional composites as well as to establish the most suitable surface finishes that can be applied to conventional fibres, although the NFRP particular case is deeply studied in Chapter 4. The principal aim is to establish a method for NFRP properties enhancement and the sizing performance of conventional composites.

In the current study, the same organosilane was used in two different approaches (directly in the NF and dissolved in the hardener) to improve the NFRP interfacial properties.

3. Bio-Composites

3.1. INTRODUCTION

The composites produced with bio-polymers (Section 3.9) or/and NF (Natural Fibre) are commonly described as bio-composites. In the present research project, a flax fibre/bio-epoxy bio-composite was selected for detailed experimental analysis. The first part of the chapter describes NF and in the second part defines bio-polymers.

3.2. NATURAL FIBRE DESCRIPTION

NF are fibrous materials that may be variously sourced from plant, animal or mineral matter (Pickering, 2008). The classification of NF matter is shown in Figure 3.1. Plant fibres are the result of a photosynthesis process and are normally defined as lignocellulosic fibres (LF) because of their elevated content in both components, lignin and cellulose. Cotton is considered to be a cellulosic fibre.



Figure 3.1: Natural fibres classification scheme (George, Chae & Bressler, 2016)

Living plants are renewable and sustainable entities, but the fibres processed from them may not be, and any claim for their green credentials should be supported by a life cycle assessment (LCA) (Gurunathan, Mohanty & Nayak, 2015), which will be further explained in Chapter 6. The environmental advantages of NF, in combination with bio-based matrices, are one of the main reasons for their use in composite manufacturing processes. The mechanical and physical properties of NF materials are dependent on their internal structures. LF are generally formed of cellulose, hemicellulose, lignin and other components (Figure 3.2). The cellulose component gives stiffness and reinforcement strength to the structure and the lignin matrix works as a bridge transferring the load from one microfibril to the next. In recent years there has been an increase interest in microfibril isolation and its utilisation as a nano-reinforcement because of its higher mechanical property (Malainine, Mahrouz & Dufresne, 2005).



Figure 3.2: Lignocellulosic fibre internal structure scheme (George et al., 2016)

The mechanical properties of NF will vary according to their different lignocellulosic fibre composition and in the context of the present study, will directly affect the interfacial properties. Generally, in most cases where the lignocellulosic fibres have been treated to improve their interaction with the polymer matrix, the oily component from the fibre is retained. The present study will aim to eliminate this component as well as the lignin when this is also retained (Cho et al., 2007; Shanmugam & Thiruchitrambalam, 2013). The first interaction would lead to the improvement of composite rigidity, but lignin elimination would cause a drastic drop in strength. As a result mild, rather than very aggressive, treatments are usually selected.

3.3. LIGNOCELLULOSIC FIBRES CLASSIFICATION AND TYPES

LF originate from plant source and are classified according to their extracted parts, whether from the outer stem (bast) or the inner stem (core). Bast fibres are located around the stem and under the external bark of dicotyledons plants forming bundles or strands that run parallel to the length of the

stem. They provide structural strength and rigidity to the plant. Strands may vary in length but may attain spans of up to 1000 mm with widths of up to 1 mm. In the case of flax, ramie, hemp, kenaf and jute ultimate fibres are very long with an "aspect ratio" of 1000 (ratio of length to width).

Leaf fibres may also be long, as for example when they are extracted from sisal or pineapple leaves. Their mechanical properties are, however, lower than in the bast fibres. Reinforcing fibres may also be extracted from plant fruit, seeds, core, roots or grass, with the limitation that they are considerably shorter and therefore more suitable for use as fillers in TP composites. LF classification is shown in Table 3.1.

| Deat | Last | | | Seed | | | 6 | Current la constante | Oth |
|---------|-----------|--------|-----------|------|----------|-------|-------|----------------------|--------|
| Dast | Leal | Fibres | Pod | Husk | Fruit | Hulls | Core | Glass/leeus | oulei |
| Hemp | Pineapple | Cotton | Kapok | Coir | Oil palm | Rice | Kenaf | Wheat | Wood |
| Ramie | Sisal | | Loofah | | | Oat | Jute | Oat | Roots |
| Flax | Agava | | Milk Weed | | | Wheat | Hemp | Barley | Galmpi |
| Kenaf | Henequen | | | | | Rye | Flax | Rice | |
| Jute | Curaua | | | | | | | Bamboo | |
| Mestra | Banana | | | | | | | Bagasse | |
| Urena | Abaca | | | | | | | Corn | |
| Roselle | Palm | | | | | | | Rape | |
| | Cabuja | | | | | | | Rye | |
| | Albardine | | | | | | | Esparto | |
| | Raphia | | | | | | | Sabai | |
| | Curauá | | | | | | | Canary grass | |

Table 3.1: Lignocellulosic NF classification (Pickering, 2008)

Bast fibres are the most commonly used reinforcements in NFRP composites because of their length and mechanical properties. The characteristic properties of the most commonly utilised Natural fibres are shown in Table 3.2.

Table 3.2: NF characteristics (Pickering, 2008)

| Fibre | Density (g/m ³) | Length (mm) | Diameter (µm) |
|--------------|-----------------------------|-------------|---------------|
| Cotton | 1.21 | 15-56 | 12-35 |
| Coir | - | 0.3-3 | 7-30 |
| Flax | 1.38 | 10-65 | 5-38 |
| Jute | 1.23 | 0.8-6 | 5-25 |
| Sisal | 1.2 | 0.8-8 | 7-47 |
| Hemp | 1.35 | 5-55 | 10-51 |
| Henequen | 1.4 | - | 8-33 |
| Ramie | 1.44 | 40-250 | 18-80 |
| Kenaf (bast) | 1.2 | 1.4-11 | 12-36 |
| Kenaf (core) | 0.31 | 0.4-1.1 | 18-37 |
| Pineapple | 1.5 | 3-8 | 8-41 |
| Bagasse | 1.2 | 0.8-2.8 | 10-34 |

3.3.1. Flax

Flax fibres have a particular relevance in the present research project. Flax belongs to the Linaceae family and it is one of the oldest commercial crops in the world used to produce Linen in the textile industry. The genus Linum has many species, and Linen flax L. usitatissimum is an annual plant with a plant stem diameter of 1.6-3.2mm. Flax is generally harvested after 100 days or when the base starts to turn yellow. Fibres are extracted from the plant stem and linseed/flax oil is produced from the seed. The fibres are separated from the bark by retting, and approximately 0.1-0.25 m of the stalk supply the bast fibres. Flax is grown widely in Ireland, France, China and Belarus (Heller et al., 2015).

The flax quality required for composite applications in an industrial capacity is not yet well established. Retting is one of the least well defined procedures in these applications, and is a critical parameter during flax production since it directly affects the final mechanical properties of the fibres (Martin, Mouret, Davies & Baley, 2013). Because of this problem many alternative methods have been proposed, such as radio frequency assisted retting, microwave assisted retting and enzymatic retting (Ruan, Du, Gariepy & Raghavan, 2015; Nair, Rho, Yaylayan & Raghavan, 2013; Evans, Akin & Foulk, 2002).

Besides retting quality questions, a number of companies now can offer high quality flax fibre and fabrics, for example Composite Evolution based in the UK, Lineo based in Belgium or Bcomp based in Switzerland. In the current research both Composite Evolution and Lineo fibre was used for composite manufacturing.

3.4. NATURAL FIBRES EXTRACTION TECHNIQUES

Lignocellulosic NF is extracted from the plant by using three main techniques - mechanical, biological and chemical - which may be applied singly or in combination depending on the quality of the fibre. The extraction techniques will directly affect the final mechanical properties of the fibre (Zeng, Mooney & Sturrock, 2015). They are described in detail below.

3.4.1. Mechanical extraction

Primitive fibre separation techniques allowed the plant to rot in the field before it was gathered and beaten against a hard surface to eliminate non-essential plant tissue. Today laborious manual methods have been replaced by machinery leading to an improved extraction process and a higher yield contribution.

When machines are applied to isolate the LF from the plant, the process is denominated as decortication (Hepworth, Hobson, Bruce & Farrent, 2000). In the process the plant is passed through slotted rollers in order to separate the NF from the waste material such as the core or bark. This isolation process is applied in the extraction of LF from sugar cane (bagasse), jute, hemp or kenaf plants. Although decortication is the most commonly applied technique for most plants, in the more specific example of cotton a gin machine is used which may also applied for the extraction of kenaf fibres.

3.4.2. Biological extraction

Basically, in the biological process enzymes from micro-organism such as moulds and bacteria are used to separate the LF from the plant. Retting or degumming is a most necessary part of the process (Shahid, Mohammad, Chen, Tang & Xing, 2016). The bast fibres are separated from the straw by retting in a slow-running river, pond or tank, typically for a 3 to 20 day period depending on the type of fibre, the humidity and temperature conditions or whether dew/ground-retting (straw left exposed in the field) or enzyme retting techniques are used. During retting the water content of the straw increases and this encourages the growth of bacteria or fungi which in turn causes the deterioration of the pectin binders that hold the cellulose fibres together in the plant stem.

It has been shown that the quality of the retting process is controlled by such factors as microorganism diversity, water quality and pH factors, parameters that require consistent monitoring to achieve the best results. Complications arise in the thicker parts of the stem since they take longer to ret causing the thinner end stems to over-ret. As a result, it has been proven that the use of less

damaging microorganism, clean water and a controlled pH below 4.9, will considerably improve the retting process and consequently increase the quality of the extracted LF.

The requirement for high water utilisation is one of the limiting factors in the retting process. For this reason retting is undertaken directly in the field, cutting the plant after the growing season and leaving it to ret by natural biological action in the absence of humidity control and identified microorganism content (Liu et al., 2015). Under such conditions the retting process may take up to a month longer than in pond retting.

3.4.3. Chemical extraction

This separation process usually involves retting by chemical action or pulping using alkaline or acid agents to extract NF from the plant.

Dew-retting is the preferred process used today since it is quicker, more efficient and creates less waste water than the water immersion methods; its disadvantage is it produces chemical pollution. A number of chemical methods have been devised to improve dew-retting. They include the use of chemical chelators (e.g. EDTA or oxalic acid) at high alkalinity or of detergents/surfactants (Adamsen, Akin & Rigsby, 2002). Chemical treatment permits a high rate of production although the quality of the fibres produced may be impaired and the cost of production may be high. Kessler has developed a flash explosion steam treatment to produce high quality natural fibres by decomposing the lignin and hemicellulose fractions at high processing temperatures (Kessler, Becker, Kohler & Goth, 1998).

Chemical "degumming" is another chemical option, whereby hot alkali solutions are used to dissolve the pectin binder and ultimately release the fibres from the bundles (Feng, Chen & Zhang, 2008). The normal reaction medium is a sodium hydroxide solution in water with sodium carbonate added for economic reasons. The process improves some characteristics of the fibres (e.g. fineness, softness and elongation-at-break) but at a cost of reducing fibre length and strength. Because of the loss of mechanical properties degumming is not normally used as a technique when fibres are required for reinforcement in composites.

3.5. NATURAL FIBRE CHEMISTRY

Most NF are composed of lignin, cellulose, hemicellulose and pectin with each represented by its respective component percentage, as illustrated in Table 3.3.

| | Bark/Steam fibre | | | | | | | | | |
|-----------|------------------|-----------|---------|----------------|----------|------|-------|--|--|--|
| Flax | 71-78 | 18.6-20.6 | 2.2 | 2.3 | 3.9-10.5 | 1.7 | 5-10 | | | |
| Hemp | 70.2-74.4 | 17.9-22.4 | 3.7-5.7 | 0.9 | 2.10 | 0.8 | 2-6.2 | | | |
| Jute | 61-71.5 | 13.6-20.4 | 12-13 | 0.2 | 1.2 | 0.5 | 8 | | | |
| Kenaf | 45-57 | 21.5 | 8-13 | 3-5 | N.S. | N.S. | N.S. | | | |
| Ramie | 68.6-76.2 | 13.1-16.7 | 0.6-0.7 | 1.9 | 6.1 | 0.3 | 7.5 | | | |
| Banana | 63-64 | 10 | 5 | N.S. | N.S. | N.S. | 11 | | | |
| Nettle | 86 | 4.0 | 5.4 | 0.6 | 2.1 | 3.1 | N.S. | | | |
| | Leaf fibre | | | | | | | | | |
| Sisal | 67-78 | 10-14 | 8-11 | 10 | 1.3 | 2 | 10-22 | | | |
| Curaua | 73.6 | 9.9 | 7.5 | N.S. | N.S. | N.S. | N.S. | | | |
| Pineapple | 80-83 | 15-20 | 8-12 | 2-4 | 1-3 | 4-7 | 8-15 | | | |
| Abaca | 56-63 | 21.7 | 12-13 | 1.0 | 1.6 | 0.2 | N.S. | | | |
| Henequen | 77.6 | 4-8 | 13.1 | N.S. | N.S. | N.S. | N.S. | | | |
| | | | Fr | uit/Seed fibre | | | | | | |
| Cotton | 85-90 | 5.70 | 0.7-1.6 | 0-1 | 1.0 | 0.6 | 20-30 | | | |
| Coir | 36-43 | 0.15-0.25 | 41-45 | 3-4 | 5.2-16.0 | N.S. | 30-49 | | | |
| Oil palm | 65 | 0-22 | 19 | N.S. | N.S. | N.S. | 46 | | | |
| | | | | Wood | | | | | | |
| Hardwood | 43-47 | 25-35 | 16-24 | N.S. | N.S. | N.S. | N.S. | | | |
| Softwood | 40-44 | 25-29 | 25-31 | N.S. | N.S. | N.S. | N.S. | | | |

 Table 3.3: Chemical composition and structural parameters of NF (Gurunathan et al., 2015)

 Fibre
 Cellulose (wt%)
 Hemicellulose (wt%)
 Lignin (wt%)
 Pectin (wt%)
 Water soluble (wt%)
 Wax (wt%)
 Microfibrillar angle (deg)

3.6. NATURAL FIBRE PROPERTIES

The mechanical and moisture properties of LF are described in the following section. Variability is one of the most complicated questions in their study arising from such factors as plant type, crop year and crop location and issues arising from the climatic cycle experienced or the extraction techniques employed. All such factors may directly influence the final chemical composition of the fibres as well as properties associated with the diameter, morphology and density of the fibres adding to the difficulty of measuring these properties accurately.

3.6.1. Mechanical properties

Table 3.4 presents the general physical and mechanical properties represented in the most important LF and GF materials. The table illustrates the great variation in the test results of these properties. **The present research project aims to optimise the mechanical properties of NFRP, limiting as many factors as possible in order to reduce the dispersion in the results**. The mechanical tests performed will be reviewed in Chapter 5 and the results summarised in Chapter 8. One of the great advantages of LF is their high specific property values. The specific property is the

evaluated property divided by the density, but since the NF density is very low in comparison with the GF density, the value obtained is accordingly elevated.

| Fibre | Density (g cm-3) | Length (mm) | Diameter (µm) | Strain at break (%) | Tensile strength (MPa) | Young's modulus (GPa) | Specific strength (MPa) | Specific modulus (GPa) | Moisture content (%) |
|-----------|------------------|-------------|---------------|---------------------|------------------------|-----------------------|-------------------------|------------------------|----------------------|
| Cotton | 1.21 | 15-56 | 12-35 | 2-10 | 287-597 | 6-10 | 194-452 | 4-6.5 | 33-34 |
| Jute | 1.23 | 0.8-6 | 5-25 | 1.5-3.1 | 187-773 | 20-55 | 140-320 | 14-39 | 12 |
| Flax | 1.38 | 10-65 | 5-38 | 1.2-3 | 343-1035 | 50-70 | 345-620 | 34-48 | 7 |
| Sisal | 1.20 | 0.8-8 | 7-47 | 1.9-3 | 507-855 | 9-22 | 55-580 | 6-15 | 11 |
| Ramie | 1.44 | 40-250 | 18-80 | 2-4 | 400-938 | 61.4-128 | 590 | 29 | 12-17 |
| Hemp | 1.35 | 5-55 | 10-51 | 1.6-4.5 | 580-1110 | 30-60 | 210-510 | 20-41 | 8 |
| Coir | 1.2 | 0.3-3.0 | 7-30 | 15-25 | 175 | 6 | 92-152 | 5.2 | 10 |
| Kenaf | 1.2 | 1.4-11 | 12-36 | 2.7-6.9 | 295-930 | 22-60 | 246-993 | 18-50 | 6.2-12 |
| Banana | 1.35 | 0.9-0.4 | 12-30 | 5-6 | 529-914 | 27-32 | 392-677 | 20-24 | 10-11 |
| Pineapple | 1.5 | 3-8 | 8-41 | 1-3 | 170-1627 | 60-82 | 287-1130 | 42-57 | 10-13 |
| Abaca | 1.5 | 4.6-5.2 | 10-30 | 2.9 | 430-813 | 31.1-33.6 | N.S. | N.S. | 14 |
| Bamboo | 0.6-1.1 | 1.5-4 | 88-25 | 1.3-8 | 140-441 | 11-36 | 383 | 18 | N.S. |
| Nettle | 1.51 | 5.5 | 20-80 | 1.7 | 650 | 38 | N.S. | N.S. | 11-17 |
| Hardwood | 0.3-0.88 | 3.3 | 16 | N.S. | 51-120.7 | 5.2-15.6 | N.S. | N.S. | N.S. |
| Softwood | 0.30.59 | 1.0 | 30 | 4.4 | 45.5-11.7 | 3.6-14.3 | N.S. | N.S. | N.S. |
| E-glass | 2.5 | N.S. | 15-25 | 2.5 | 2000-3500 | 70-73 | 800-1400 | 29 | N.S. |
| S-glass | 2.5 | N.S. | N.S. | 2.8 | 3-3.5 | 63-67 | 1.8 | 34.4 | N.S. |

Table 3.4: Physical and mechanical properties of important LF and GF (Gurunathan et al., 2015)

3.6.2. Moisture properties

LF surfaces have high levels of polar hydroxyl (-OH) groups which provides the NF with its hydrophilic property. It is therefore very important to determine the water uptake of the NFRP since moisture absorption can result in fibre/matrix interface ageing and subsequently lead to a reduction in its composite mechanical property. These properties will be examined in the experimental tests described in Chapter 6. Table 3.5 presents the equilibrium moisture content (EMC) percentages of fibres derived from the principal plant materials at 65% relative humidity and a temperature of 21°C. It is noted in this table that the fibre composite EMC for flax is 7%.

Chapter 6 describes the composite aging tests in moist environments. The experimental campaign includes comparative bio-composite systems moisture aging test in Appendix E. See Table 3.5 where different fibres EMC at 65% Relative Humidity (RH) and 21°C. According to Pickering study; 7% of EMC value is compared with the PhD experimental results.

| Tuble | Table Sist Mi Equilibrium Holstare Contene (Ene) percentage (Hetering, 2000) | | | | | | | | | | | | |
|------------|------------------------------------------------------------------------------|------|------|------|--------|-----------|------|-------|--------|------|-------|-------|------|
| Fibre | Sisal | Hemp | Coir | Aloe | Banana | Pineapple | Wood | Abaca | Cotton | Jute | Kapok | Ramie | Flax |
| EMC (%) | 11 | 9 | 10 | 12 | 15 | 13 | 12 | 9,5 | 8 | 12 | 10 | 9 | 7 |

Table 3.5: NF Equilibrium Moisture Content (EMC) percentage (Pickering, 2008)

3.7. NATURAL FIBRE REINFORCEMENT FORMS AND FABRICS

In the earlier stages of producing NFRP, NF was added to the matrix in order to obtain a more economical final product without due consideration of its mechanical properties. However, following these initial stages, research in the development of NF composite materials has concentrated on introducing reinforcement tailoring into each specific final product. The matrix might be reinforced with short or continuous NF fabric depending on the final application. In composite design it would be a key factor in the selection of the correct NF, NF reinforcing form and V_f; always considering the interface properties.

3.7.1. Slivers and yarns production

With the utilisation of classical textile spinning techniques it is possible to produce sliver and fibre yarns.



Figure 3.3: Flax fibre yarn roll produced by spinning technique (Courtesy of Composite Evolution)

The first step replicates the mat production carding technique by which the fibre is opened, mixed and carded. The web obtained from the carding process is gathered into a sliver with the assistance of a can coiler and the sliver obtained from this process might be used directly in semi-finished product manufacturing, as for example in pultrusion. Additionally, if the textile requires to be produced, the slivers will have to be transformed into yarns as illustrated in Figure 3.3.

Sliver spinning strengthens the yarn to obtain a base material in fabric manufacturing which also facilitates in its handling. However, the twisting process considerably reduces the mechanical properties of the yarn because of fibre disorientation. The loss of performance has been studied in

many prediction models (Shah, Schubel & Clifford, 2013). Further, spinning is very energy intensive (Dissanayake, 2011).

3.7.2. Fabrics production

NF fabric manufacturing replicate the same procedures as those used in the production of GF or CF. In fabric manufacturing the first step is to twist the strand in order to obtain the yarn. This process aids the handling of the yarn in weaving the fabric. Hundreds of yarns may feed the loom in order to manufacture different fabrics, each characterised by different fibre angles, fabric weights and weaving tension parameters. In the case of NF fabrics lubricants are not applied because in comparison with synthetic fibres the wear resistance is elevated so the fibres will not be damaged. Figure 3.4 illustrates a flax fabric production method.



Figure 3.4: Stitched flax fabric production process (Courtesy of Composite Evolution)

Sizing has to be applied at a certain point along the manufacturing process and the various systems employed as surface treatments. Chapter 4 will discuss the different techniques for the NF fibres surface treatments used in order to improve the adhesion between fibre and matrix in the composite production.

In fabric production, the fibres are usually arranged either uni-directionally, randomly, cross-plied, woven (plain, twill or satin weave), knitted or stitched-interlock. Figure 3.5 illustrates two different woven fabric types in schematic arrangements.



Figure 3.5: Jute woven fabrics (plain left and twill right) (Courtesy of Easy Composites)

One of the greatest advantages of NF reinforcements is their low price in relation to their reinforcing capability. This advantage is most obvious in the case of short fibres, slivers or mats, however when the fibres are transformed into fabrics production costs and embodied energy are proportionally increased to reach levels similar to GF. This phenomenon has been studied by Dissanayake (Dissanayake, 2011) using flax fibre LCA. GF has a stronger reinforcing capability than any other NF although it may lack specific rigidity. As a result, when the economic issue is a fact, NF renewable source, wear and impact properties are a positive point. Basically, this would be the only option for competing with the GF advantages.

3.8. NATURAL FIBRE REINFORCED POLYMERS (NFRP)

NF composites might be produced using either TP or TS matrix systems (Summerscales & Grove, 2014). The mechanical properties of NFRP systems are illustrated in Tables 3.6-3.7. Normally NF and TP or TS matrices are modified in order to improve the adhesion between fibre and matrix and will be discussed in detail in Chapter 4.

Table 3.6 captions the mechanical properties of different NF/TS composite systems. The first system is a GF/UP that may be applied as a reference and has significant comparative value. Table 3.7 captions the mechanical properties of different NF/TP composite systems

| | | | | / = • • • / | |
|------------------------------------|------------------------------------|------------------------|-----------------------|---------------------------------------------|----------------------------------------|
| Fibre/resin | Volume fraction V _f (%) | Tensile strength (MPa) | Young's modulus (GPa) | Charpy impact strength (kJ/m ²) | Interlaminar shear strength ILSS (MPa) |
| GF Chopped strand mat/polyester | 30 | 95 | 8 | 40 | 25 |
| Jute/polyester | 45 | 60 | 7 | 29 | 10 |
| Hemp/low-OH polyurethane | 21 | 23 | 2 | 19 | 3 |
| Hemp/high-OH polyurethane | 19 | 27 | 3 | 7 | 3 |
| Alkalised Hemp/high-OH polyurthane | 20 | 35 | 3 | 9 | 5 |

Table 3.6: Shows the NF/TS composites examples (Pickering, 2008)

| Fibre/resin | Volume fraction V _f (%) | Tensile strength (MPa) | Young's modulus (GPa) |
|---------------------------------|------------------------------------|------------------------|-----------------------|
| GF Chopped strand mat/polyester | 30 | 95 | 8 |
| Kenaf/PLA | 60 | 50 | 5 |
| Flax/PLA | 20 | 66 | 5.7 |
| Bamboo/PP | 20 | 16 | 2.5 |
| Sugar cane/PP | 20 | 16 | 6 |
| Sisal/LDPE | 20 | 16.5 | 93 |
| Wood/PP | 30 | 38 | - |
| Oil palm fibre/HDPE | 45 | 8 | 0.5 |
| Hemp/starch | 50 | 153 | - |

Table 3.7: Shows the NF/TP composite examples (Pickering, 2008)

3.9. BIO-POLYMER DESCRIPTION

Bio-polymers can be bio-based and/or bio-degradable; the bio-based (bio-sourced) bio-polymer is derived from natural sources; bio-degradable bio-polymers can be returned to their natural origin (Reddy, Vivekanandhan, Misra, Bhatia & Mohanty, 2013). Bio-based and bio-degradable concepts are sometimes indifferently used, however this is not correct; Bio-based polymers can be totally or partially produced from renewable sources and may not be biodegradable; in contrast fossil-based polymers can be bio-degradable and may decompose and revert to nature.

Reddy has proposed grouping bio-polymers into three categories according to their production methods (Reddy, Misra & Mohanty, 2012)., as illustrated in Table 3.8. They are discussed as follows -

- Renewable resource based biopolymers: these products are entirely synthesised from renewable or waste resources, either from animal (e.g. chitin and chitosan, etc.) or from plant sources (e.g. starch, cellulose, lignin, etc.). The most commonly synthesised are poly(lactic acid) (PLA) and polyhydroxybutyrates (PHB) bio-plastics, although recent research has produced commodity plastics such as polyethylene (PE), polypropylene (PP) or nylon from biological sources.
- Petroleum-based bio-polymers: are products synthesized from petroleum and are biodegradable with the capacity to decompose back to nature. Polycaprolactone (PCL) and poly(butylene adipate-co-terephthalate) (PBAT) are two example of these products.

3. **Bio-polymers from mixed resources:** are formed from bio-based and petroleum source monomers in combination. Bio-thermoset resins or bio-based thermoplastic blends are possible options that have been investigated in the present PhD research project.

| Bio-polymer source | Туре | Example | |
|-------------------------------|-------------------------------|----------------------------------|--|
| | Poly (lactic acid) (PLA) | PDLA, PDLLA | |
| | Polyhydroxyalkanoates (PHAs) | PHB, PHBV | |
| Renewable based | Starch plastics | Wheat, potato, corn based | |
| | Cellulosics | Cellulose esters | |
| | Protein-based plastics | Plant and animal proteins based | |
| | Aliphatic polyesters | PCL, PBS | |
| Petroleum based biodegradable | Aliphatic-Aromatic polyesters | PBT | |
| | Poly (vinyl alcohol) (PVOH) | PVOH | |
| Mixed recourses | Polyesters | PTT | |
| Mixed resources | Thermosets | Bio-based epoxy and polyurethane | |

Table 3.8: Bio-polymers classification and examples (Reddy et al., 2012)

The world is turning from petrochemical based products to "green chemistry" based polymers with emphasis on "end of oil" alternatives to the conventional matrix options. Dwindling fossil resources, surging energy demand and global warming are factors stimulating the growing demand for renewable polymer products with a low carbon footprint. Natural polymers, the products of biomass conversion in bio-refineries and chemical carbon dioxide fixation open new fields of research. Mülhaupt has discussed the dreams and reality of producing "green polymers" regarding their potential application (Mülhaupt, 2013), cost-effectiveness and sustainable development. This chapter refers to the current market situation concerning the application of bio-matrix materials.

When a bio-composite is produced it is normal to reinforce the matrix using NF. In this context both the matrix and the fibre are obtained from renewable sources resulting in a high bio-content product as exemplified in a starch based biodegradable bio-polymer reinforced with coconut fibre (Lomelí-Ramírez et al., 2014).

3.10. THERMOSET RESINS

Today many resin companies are attempting to develop bio-based TS formulations containing different "bio" constituents. The most challenging question is to maintain a petrochemical resin

performance in such bio-based formulations. As such these have been as investigated by Raquez (Raquez, Deléglise, Lacrampe & Krawczak, 2010).

In the composite industries the most widely used formulations are epoxy and UP resins. In petrochemical-based matrices the mechanical performance is very high and the interaction with the reinforcement is also excellent, resulting in a high standard composite (Oral, Guzel & Ahmetli, 2013; Yi, Um, Byun, Lee & Lee, 2013). In contrast, generally the bio-resins are not searching the required mechanical neither adjusting to a commercially competitive price. However, there are a number of products being developed to meet market requirements, and there are some resin systems that are highly competitive both for mechanical and commercial reasons.

3.10.1. Bio-epoxy resins

In the Aouf study, SuperSap (PhD selected) commercial bio-epoxy system is used and other alternatives evaluated (Aouf et al., 2013). Examples of natural feedstock are itaconic acid, the cardanol (Cashew Nut Shell Liquid), seeds oil, or rosin diacid, soybean oil among other options (Ma et al., 2013; Rao & Palanisamy, 2013; Das & Karak, 2010; Huang, Zhang, Li, Xia & Zhou, 2013; Lu, Khot & Wool, 2005).

Epoxy resins are usually formed by reaction of Bisphenol A (BPA) and epichlorohydrin to formulate the prepolymer. Once the prepolymer has been formed, it is usually catalysed by an amine or anhydride. It has been proven that contact with the BPA monomer provokes an adverse human health problem and new epoxy resin formulations are being formulated from the options noted above to overcome the problem.

The isosorbide molecule is a bicyclic ring structure with chiral diols. This particular structure provides the isosorbide molecule with a monomer characteristic which is a good candidate base for epoxy, polyester or polyurethane resin formulations. This molecule can be obtained from the enzymatic hydrolysis of starch or the catalytic dehydration of sorbitol, a glucose derived material. This material

utilisation allows for a 50% bio-content resin formulation. The epichlorohydrin remains as a petrochemical based material.

Another option is epoxidised oil-based bio-epoxy resins. Triglyceride oils have been used as a base material for polymers production. In the bio-epoxy resins formulated by Entropy, triglyceride oils are extracted from pine trees (La Rosa et al., 2014). In this case the active epoxy groups are located in the middle of the aliphatic chain, impeding the correct cross-linking to form three dimensional networks. This results from steric hindrance and the autocatalysis of some of the polymer chains. The mechanical properties of bio-epoxy resins do not reach petrochemical epoxy values since the reticulation density is lower than expected.

Some epoxy systems use an amide catalyst instead of amine (Kocaman & Ahmetli, 2016). This molecule has lower reactivity than the amine to provide a lower cross-link density polymer. However, although the mechanical properties are reduced amides can be obtained from renewable resources, as from castor oil for example (Van der Steen, Bretz, Kabasci & Stevens, 2013). Unlike the aromatic amines which are toxic, the castor oils do not contain an aromatic ring thus forming a non-pollutant catalytic system. The introduction of these renewable components decreases the strength properties of the resin without significantly reducing the elastic modulus.

Some manufacturers have up-scaled their laboratory investigations into industrial production as detailed in Table 3.9 (Marrot, Bourmaud, Bono & Baley, 2014). The Sicomin (France) commercial product called Greenpoxy 55 has a bio-content determined under ASTM D6866 at 55% in weight. The epichlorohydrin is substituted by a bio component. Similarly Entropy (USA) developed the SuperSap resin system with a 51% input from renewable resources (average value dependent on the final formulation and solid content), the BPA substitution obtained from pine oil molecules and amide catalytic system bio-source. The bio-epoxy resin produced by Sandtech (France) has a bio-content grading from 55% to 85%. Part of the bio-source is again created by the substitution of the BPA molecules.

| Commercial name | Manufacturer | Bio-content (%) |
|--------------------|--------------|--------------------|
| Greenpoxy 55 | Sicomin | 55 |
| Supersap | Entropy | 51 |
| L' Epoxy lin | Sandtech | 55-85 |

Table 3.9: Commercial bio-epoxy resins bio-content (Marrot et al., 2014)

3.10.2. Bio-polyester resins

References to sustainable UP are numerous in the research literature (Cousinet et al., 2015; Dai et al., 2015; Sadler et al., 2014; Costa et al., 2016; Gobin, Loulergue, Audic & Lemiègre, 2015). The polyester prepolymer is dissolved in styrene, a dangerous solvent to human health, and a potent environmental pollutant because of its Volatile Organic Compound (VOC) content. Research to formulate resins with a lower solvent content are ongoing and there are some formulations that are solvent free.

An alternative method to increase the UP bio-content and avoid the use of the styrene solvent would be to utilise triglyceride oil that would transform into a fatty acid monomer and form a covalent bridge in the curing process.

There is the further option of formulating UP by using the isosorbide molecule which forms the polyester prepolymer when reacting with the ethylene glycol molecule. The isosorbide is a 100% bio source material that is obtained from glucose or starch.

The laboratory conducted approaches have to be up-scaled for commercial production as shown in Table 3.10.

| Commercial name | Manufacturer | Bio-content (%) |
|---------------------|----------------|--------------------|
| Enviroguard 93250 | Cray Valley | 53 |
| Norsodyne G703 | CCP composites | 23 |
| Eco-series products | Interplastic | 5-20 |

Table 3.10: Commercial bio-polyester resins bio-content (Marrot et al., 2014)

3.10.3. Other bio-resins

Apart from epoxy or polyester based thermoset resins, approaches have been made to utilise a range of different materials. Thermosetting resins have been produced based on cashew nut shell liquid (CNSL) used as a curing agent, or furan resins based on furfuryl alcohol (from sugar cane bagasse), or on soy oil derived polyurethanes (Kasemsiri, Neramittagapong & Chindaprasirt, 2015; Rivero, Fasce, Ceré & Manfredi, 2014; Arnold, Weager, Hoydonckx & Madsen, 2009; Luo, Mohanty & Misra, 2013). For example Transfurans is a commercialised furan based resin, and Santech is an elevated bio-content polyurethane product.

3.11. SUMMARY

Chapter 3 describes natural fibres characteristics and bio-polymers available in the market; focusing on the flax fibre and bio-epoxy resin. This is in this PhD.

As mentioned in the first part of the document, flax fibre was selected as the most appropriate reinforcing NF for the PhD bio-composite production. The main reason was that the flax fabrics are commercially available in high quality product, getting as a result high standard bio-composite, consequently reducing experimental campaign variability. Both Composite Evolution and Lineo offer different types of fabrics, for example UD NCF or woven multi-axials fabrics, for easier utilisation of such components in a potential commercial product.

Similarly Entropy's SuperSap bio-epoxy resin was selected. The main reasons for its selection were the commercial availability, price, 51% bio-content and different formulations availability. For example, Entropy offers a wide range of commercial formulated resins for the infusion process, RTM or hand-lamination.

The utilisation of these two components together can result in the easy manufacture of a high quality bio-composite; both at lab-scale and scaled-up for the manufacturing of surfboards and RTM fins.

4. The interface in natural fibre composites

4.1. INTRODUCTION

The interface is the distinct boundary between the reinforcement and the matrix (Hayes, Lane & Jones, 2001; Lane, Hayes & Jones, 2001; Pukánszky, 2005). The question is that for NFRP the fibre and matrix adhesion is usually very weak, for that reason the current PhD investigation tried to solve in different ways this question. In order to select the best approximation, an extensive literature review was performed, what it is reflected in Chapter 4.

The research study is primarily involved with the analysis of the NF composites interface enhancement. The anticipated contribution to knowledge is to produce a sustainable and economical interfacial property improvement method for a particular flax fibre/bio-epoxy system. The realisation of the full mechanical performance of the reinforcement is critically dependent on the effective load transfer by shear over the "half critical length" at each fibre end, which in turn is a function of the chemical and physical bonds between the fibre and the matrix.

Instead of a distinct interface, there may be an interphase, normally considered to be a transition volume adjacent to the reinforcement component (fibres or particles) with functionally-graded properties, see Figure 4.1. The interphase may involve preferential orientation of polymer molecules, or in CMC (Ceramic Matrix Composites) and MMC (Metal Matrix Composites) may result from inter-diffusion of atoms between the two phases (Naslain, 1998; Karger-Kocsis, Mahmood & Pegoretti, 2015).

In a good (first class) interface there is strong adhesion between the fibre and the matrix, obtaining (creating) a proper load/strain/stress transfer inside the composite structure, as it might be observed in the micrograph (Figure 4.2) showing a transverse section of a

homogeneous UD composite; best practice when the system is observed with SEM. In contrast, when the interface is weak, the external load causes debonding of the reinforcement from the matrix, with no corresponding transfer of the load. When the load is correctly transferred a fibre fracture limit may be achieved with the fibres broken into segments of a critical length and developing no interface fractures.



Figure 4.1: Composite interface and interphase explicative scheme (Acciona Blades courtesy)



Figure 4.2: Transverse section of a UD NCF fibre reinforced composite (Courtesy of Acciona Blades)

A satisfactory interface can be achieved either by mechanical keying or by applying an appropriate chemical system, or by combining both options. The bast fibres are composed primarily of cellulose (an anhydroglucose polysaccharide) and are thus hydrophilic. The flax fibre used in this investigation is classified as bast fibre. The matrix system may be hydrophilic (e.g. polyamide or phenolic), which would increase the degree of hydrogen bonding, or it may be hydrophobic (e.g. polypropylene, unsaturated polyester or epoxy), and hence would have low compatibility and would require some treatment to achieve a good interface. The hydrophilic nature of the system makes the composite vulnerable to moist environments leading to plasticisation and possibly fungal deterioration. Decreasing the polarity of the fibre by chemical reaction could produce a more hydrophobic nature in the fibre. Bast fibre composites are targeted at low-value applications (not aerospace/defence) so it would be inappropriate to use a high cost approach to interface optimisation.

In order to maximise the interfacial properties in reinforced materials there are several possible approaches:

- Addition of compatibilisers in the matrix
- Physical modification of the fibre surface
- Chemical modification (including grafting) of the fibre surface
- Coating the fibre surface with a coupling agent

In this chapter, these four compatibility development methods are reviewed generally for NFRP. The four methods are not mutually exclusive. Specific emphasis is placed on mercerisation or silanisation as NF chemical treatment techniques, and in addition of compatibilisers (silane) to the matrix. In addition to composite mechanical characterisation tests, specific compatibility tests, as well as newly developed tests, have been studied.

4.2. INTERFACE/INTERPHASE

Fibre dominated properties, such as longitudinal tensile, compressive and flexural moduli/strength, show moderate sensitivity to interfacial properties variation, whilst off-axis properties, such as transverse tensile and flexural strength, in-plane and interlaminar shear

strength, are highly sensitive to interfacial properties variation. Using the InterFacial Shear Strength (IFSS) values, Dai, et al. demonstrated that CF treatments directly affect the IFSS properties (Dai, Shi, Zhang, Li & Zhang, 2011).

The two key factors in the mechanical properties of the fibre-matrix have been identified in the literature as the fibre-matrix adhesion level and the interphase morphology. An optimum level of fibre-matrix adhesion has been identified for each particular composite system. The best adhesion may not always result from the most adequate system. For example, when interfacial adhesion is increased normally the stiffness of the resultant composite rises considerably, producing a rigid final product (Cañigueral, 2009). The fibre-matrix adhesion enhancement will also result in achieving higher fatigue properties (Harper & Hallett, 2010; Afaghi-Khatibi, Ye & Mai, 2001).

Many models, based on interfacial properties, have been developed in order to predict the final mechanical properties of the composite. However, these models are not accurate and have resulted in limited success. For example, Shia and Hui presented a simple interface model for the prediction of the Young's modulus of a silicate-elastomer nanocomposite system (Shia, Hui, Burnside & Giannelis, 1998). A discrepancy arose between the theoretical prediction and the experimental Young's modulus values because of the imperfect bonding between the fibre and the matrix. Kerans et al. presented a paper where the CF surface was oxidised in order to obtain the desired interface, increasing the fibre-matrix adhesion to produce composite tailored properties (Kerans, Hay, Parthasarathy & Cinibulk, 2002).

The correlation between the interfacial models and the experimental data is far from ideal, even when synthetic fibres are used. The inclusion of NF complicates the issue even further. Fotouh et al. presented a study where the TP/short NF composite system static and fatigue performance were predicted using different theoretical models (Fotouh, Wolodko & Lipsett, 2015). A number of models were presented in order to obtain the best performance prediction (micromechanical, energy method, rheology etc.), but it was concluded that models based on

empirical statistically treated values were best suited to obtain the mechanical performance prediction.

Some models are totally based on theoretical knowledge (phenomenological) whilst others are partially based on laboratory data (semi-empirical) obtained by direct methods. The lack of success of these models has resulted in a need to develop specific methods to predict multidirectional mechanical properties based on lamina methods in composites.

One of the main failings in the theoretical models is the underlying assumption that the prediction of composite mechanical and interfacial properties failure arises because the interrelationship between fibre interface and matrix have not been considered in the models. This may mean that a wider zone than the interface is necessary for the correct understanding of the composite behaviour, and for this reason interphase understanding was added to the experimental data when the model development was being considered. The interphase was defined as the 3D space between bulk fibre and bulk matrix. This complex region is schematically explained in Figure 4.3 (Drzal, Herrera-Franco & Ho, 2000).



Figure 4.3: Schematic representation of the Interphase region (Drzal et al., 2000)

Interphase properties are dependent on the following factors - fibre and matrix surface energy; the mechanical bonding between fibre and matrix; the interfacial material properties in charge of giving a proper stress transfer; the resin chemical and thermal shrinkage during the curing and cooling processes. These factors are important because the residual stresses developed in the interphase region affect the final mechanical properties of the composite. All the parameters considered above could be applied to the study of the interface/interphase in NF composites. Le Duigou, et al. studied the actual fibre surface involved in the practical adhesion of a flax/epoxy composite (Le Duigou, Kervoelen, Le Grand, Nardin & Baley, 2014). Two different flax fibre types were tested. The results showed:

- the surface chemistry studied with X-ray spectroscopy (XPS) did not affect interfacial adhesion;
- roughness observed with SEM lead to source defects which reduced composite adhesion;
- the biochemical composition of each fibre was directly related to the resin penetration, concluding that when the fibre wetting is higher the adhesion would be improved.

It is already clear that the interface/interphase concept is of vital importance in understanding the composite system. In the case of NFRP it has even greater importance to understand because:- a) reinforcing with a lignocellulosic material is very difficult in producing inhomogeneity related to the crop (Lefeuvre, Bourmaud, Lebrun, Morvan & Baley, 2013); b) inhomogeneity is difficult because of the transformation process in drying (Baley, Le Duigou, Bourmaud & Davies, 2012); c) the water uptake may affect the properties in the interphase region (Zhang, Milanovic, Zhang, Su & Miao, 2014; Masseteau, Michaud, Irle, Roy & Alise, 2014). All these parameters might affect the interphase property and subsequently the final performance of the composite. In order to reach interphase region performance quality and increase fibre/matrix compatibility there are different techniques which will be described in the following sections.

4.3. COMPATIBILISERS IN THE MATRIX

In the literature there are multiple matrix chemical modifications that are available for enhancing quality compatibility with NF reinforcements. For TS polymers the addition of compatibilisers might increase the fibre/matrix adhesion enhancing directly the final properties

of the composite. The most commonly referred to epoxy and TS matrix modifications will be discussed in this section.

4.3.1. Epoxy matrix modification

Generally, the epoxy resin is formulated subject to the mechanical, thermal or curing properties required in the final matrix. The main modifications in the resin formulation process may, possibly, be caused by the number of components used in prepolymer production. The number would include bisphenol, epiclorohydrin or the two in a mixed ratio, the curing agents, the reactive diluents and modifiers, the hardener selection and the selected curing cycle. Such properties will be dependent on the selection of the different mechanical, thermal, ageing, and electrical components. In the next section, the epoxy resin composite interface properties will be studied, with added reference to the most commonly used resin formulation modification techniques. The information will be classified according to the reinforcement type.

Chen, et al. (Chen et al., 2007) evaluated the newly developed T800 CF sizing technique for evaluating interface increment. The resin was reformulated in order to increase resin toughness and wetting capability. These two factors directly increase the interface properties - a tougher matrix allows this property to be transferred to the interface; increased wetting properties creates improved mechanical anchoring of the resin to the rough CF surface. If fibre sizing and resin reformulation effects are added, then the interfacial shear values might even double interface/interphase bonding. The above performance was evaluated using the NOL (Naval Ordnance Laboratory) ring innovative test method.

Since correct wetting is such an important factor in achieving a good interface many studies have been instigated to evaluate wetting methods for CF fibre. Xu (Xu et al., 2008) concluded that the utilisation of acrylic acid in the formulation of the epoxy resin substantially increases the wettability of the fibre tows and directly improves their interface properties.

The aim of any epoxy resin reformulation is to achieve a decisive performance and to increase its adhesion with the matrix. As demonstrated by Xu, the formation of chemical bonds in the

interface definitely increases the performance of the composite. For this reason, siloxane, silanes or silsequioxanes have been introduced during the epoxy resin backbone synthesis (Wang, Jiang, Zhang & Cheng, 2012; Chruściel & Leśniak, 2015), to achieve covalent chemical bonds between the resin and the fibre. The current PhD study aims to introduce silanes into the epoxy matrix formulation, differing from previously mentioned studies where the silanes were introduced into the synthesis process. In this particular case the silane will be added to the bio-epoxy resin in order to increase the adhesion between the flax fibre and the bio-epoxy resin.

Matrix modifications tend to focus on improving the static properties, but interfacial development specifically affects long term properties such as the fatigue or aging of the matrix (Koimtzoglou, Kostopoulos & Galiotis, 2001; Zafar, Bertocco, Schjødt-Thomsen & Rauhe, 2012). Damage produced by either dynamic stress or aging agents directly affects the interface. As a result, a strong interface would impede the advance of a crack through the interface debonding but would allow brittle fracture of the fibre.

4.4. PHYSICAL FIBRE SURFACE TREATMENTS

The NF surface can be modified using a wide variety of physical treatments. In general, surface roughness and composition can be changed to obtain better interfacial adhesion with the matrix. George, et al. have critically reviewed the physical and chemical treatments that improve the fibre-matrix adhesion in NFRP, identifying fibrillation and electric discharge techniques as useful NF treatments. Physical treatments by electric discharge are commonly used techniques in modifying the natural fibre surface. They can be divided into low-frequency corona treatment or high-frequency (low temperature) cold plasma treatment. In both cases, the surface roughness of the NF is increased resulting in improved mechanical adhesion in the composite. In Zheng, a tensile and impact strength increment was also applied after the electric treatment.

In both TP and TS matrices this modification is a totally physical phenomenon since the interaction forces are not dependent on the functional groups in the matrix. The interfacial shear strength values obtained are positive because of the mechanical anchoring; however, the

process is often very time consuming and cannot therefore be considered as commercially viable. For this reason, physical treatments were not undertaken in this research project.

4.5. CHEMICAL FIBRE SURFACE TREATMENTS

Cellulose is a hydrophilic polar molecule while many matrix materials are hydrophobic non-polar molecules. To optimise the interface the chemical nature of one of the respective surfaces can be modified. The study of fibre treatment is basically concerned with the issue of hydrophilic/polar vs. hydrophobic/non-polar bodies. It is noted that for TP matrix composites it is unlikely that there will be a chemical reaction between the polymer and the fibre while for TS matrix composites such a reaction may occur.

The NF treatments that can be applied to secure the best NFRP interfacial properties have been compared and the following section evaluates those that have most relevance to the present research project.

The review paper by Gurunathan, et al. (Gurunathan, Mohanty & Nayak, 2015) provides a clear scheme offering different options, as shown in Figure 4.4.



Figure 4.4: NF chemical treatment option scheme (Gurunathan et al., 2015)

4.5.1. Mercerisation (NaOH) and NaCl

Chemical treatment by Mercerisation has been widely applied for the elimination of undesired substances, most especially lignin, and for subsequent NF surface activation with the polymer matrix. The Mercerisation reaction path is shown in Figure 4.5 and Table 4.2 lists the most important literature references.

Fibre – OH + NaOH \rightarrow Fibre – O-Na+ + H₂O*

Figure 4.5: The reaction shows the surface activation by alkali treatment

Mercerisation is defined as the "treatment of cellulosic textiles in yarn or fabric form with a concentrated solution of caustic alkali [soda], whereby the fibres are swollen, the strength and dye affinity of the materials are increased, and their handle is modified. The process takes its name from its discoverer, John Mercer (1844) (Mercer, 1850). The treatment removes practically all non-cellulose components except waxes and changes the crystal structure of the cellulose. Mercerisation results in "hydrated cellulose" which is chemically identical to the precursor material but has different physical properties (Rashaduzzaman, 2013). For rapeseed straw, Paukszta (Paukszta, 2013) applied solution concentrations in the range 12.5-20% for >5 minutes to obtain optimal conditions for the process. Mercerisation converts cellulose I to 50±15% cellulose II with a consequent expansion of the crystal lattice (Table 4.1) (Guthrie, 1974).

 Table 4.1: Unit cell dimensions for cellulose crystals (Wada, Nishiyama, Chanzy, Forsyth & Langan, 2008)

| Cellulose | Crystal unit cell | а | b | С | a (°) | β(°) | γ(°) |
|-------------|-----------------------------|-------|-------|-------|--------|--------|--------|
| | | (nm) | (nm) | (nm) | | | |
| Ia | one-chain triclinic P1 | 67.17 | 59.62 | 104.0 | 118.08 | 114.80 | 80.37 |
| I_{β} | two-chain monoclinic P21 | 77.84 | 82.01 | 103.8 | 90 | 90 | 96.5 |
| II | monoclinic P21 | 81.0 | 90.3 | 103.1 | 90 | 90 | 117.10 |

Chattopadhyay (Chattopadhyay & Sarkar, 1945) reviewed different methods for the determination of NF cellulose quantity. NF has been bleached with sodium chlorite (NaClO₂) for many years. Sodium chloride reacts in an acid environment to produce chlorine dioxide (ClO₂) molecules, which in contact with the NF removes the lignin from the structure and reacts with

the hemicellulose OH groups. Both effects will reduce the hydrophilicity of the NF structure whilst increasing its elasticity, as indicated by Li, et al. (Li, Tabil, & Panigrahi, 2007).

References to the Mercerisation process can be divided in two main groups:

- a) NF Mercerisation in isolation, and
- b) NFRP composite properties after Mercerisation.

NF were divided according to the selected fibre type, with particular reference to jute. Roy, et al. proposed a mild alkali treatment of the jute fibre for upgrading its mechanical properties (Roy et al., 2012). Different combinations were tested and it was concluded that the most satisfactory results were obtained when jute was treated with a 0.5% NaOH solution for 24h, securing an improvement of 82% tensile strength and 45% elongation at break.

Gassan and Bledzki reported an approximate increase in yarn tensile strength and modulus of 120% and 150% in jute fibre after mercerisation treatment (Gassan & Bledzki, 1999). The jute fibre was treated with 20% NaOH solution for 20 min at RT. The effect of the NaOH treatment is great. There are other references in the literature that support similar mechanical properties increments, although such increments are not as high as the data reported by Gassan and Bledzki (Ray & Sarkar, 2001; Ray, Das & Mitra, 2012).

Saha, et al. reported the mercerisation process (0.5% NaOH for 30 min) is supported by an additional alkali-steam process (under pressure 30 min alkali treatment), which is claimed to be more efficient that the conventional immersion alkali treatment (Saha et al, 2010).

| Fibre | Reinforcement format | Treatment | Matrix | Gain in modulus | Gain in strength | Other effects | Reference |
|-------------------------|----------------------------------------------------|----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|--------|--------------------|-----------------------------------------------|------------------------------------------------------------------------------------------|--------------------------|
| | Tormat | | | (%) | (%) | | |
| Cotton and flax | Cloth | NaOH 3.25M at 15°C +neutralised with H ₂ SO ₄ +Wash with water | - | - | - | Improve colours in printing and dyeing | Mercer 1850 |
| Jute | Straw | NaOH 0.25%,0.5%,1% for 0.5,1,2,4,8,16,24,36 and 48 hours +wash with distilled water | - | - | 82 | Reduce Hydrophilicity 50.5%, increase ε _f 45% and increase diameter 37% | Roy 2012 |
| Jute | Yarn | 20 min at 20°C in 25% NaOH | - | 150 | 120 | - | Gassan 1999 |
| Jute | Straw | 0.5% NaOH for 30 min + alkali-steam treatment 30 min | - | - | 65 | - | Saha 2010 |
| Ramie Flax Cotton | Yarn | Pretreat 0.05% Triton QS-44 + NaOH 20% 5min+ neutralise 2%Ac-COOH | - | - | In tension Ramie 6 Flax 18 Cotton 27 | ε _f in tension Ramie 7 Flax 20 Cotton 44 | Cheek 1989a and 1989b |
| Ramie Flax Cotton | Yarn | Dyeing | - | - | - | Mercerisation affects the dyeing process | Cheek 1989b |
| Ramie | Straw | 15% NaOH with applied loads of 0.049 and 0.098 N | - | decrease | 4-18 | Increase ε _f 37,5%, | Goda 2006 |
| Flax | Pulp | Cook with sosa anthraquinone + freeze + Wash H_2SO_4 30 min until pH = 4 | - | - | - | Increase the ISO brightness | Fillat 2010 |
| Flax | Straw | Wash with water + dry 3h at 80°C + NaOH 5% at 55°C for 10 min + dry 7 days at 20°C + oven dry 3h at 70°C | - | - | - | Determination of the ideal mercerisation conditions = NaOH 5% at 55°C for 10 min | Aly 2012 |
| Kenaf | Hand-made UD fabric from straw | NaOH 5% for 24h at RT + wash with water + immerse in distilled water for 24h and dry in the oven for 24h | Ероху | - | - | Confirm the enhanced ROM | Mahjoub 2014 |
| Kenaf | Mat and UD | NaOH 6% at RT for 48h/144h+wash with distilled water + dry RT 48h + dry in over for 6h at 100° | Ероху | Mat 10 UD 3 | Mat 26 UD 10 | 48h treatment improves mechanical properties while 144h damages the fibres | Fiore 2015 |
| Kenaf | UD fabric | 6% NaOH 24h + water wash + over dry at 40°C for 24 h | Ероху | - | - | Increase flexural strength in 16% | Yousif 2012 |
| Flax | Hand-made UD fabric from yarn | NaOH 2 min 13% | Ероху | 0.7 | 11 | - | Van de Weyenberg 2006 |
| Jute | Yarn | NaOH 25% for 20min at 20°C | Ероху | 150 | 120 | - | Gassan 1999 |
| Cotton cellulose | Yarn | NaOH 13.5% for different times at 25°,50°C and 65°C | Ероху | - | 50-70 loss | - | McKelvey 1959 |
| Abaca | Straw | 5, 10 or 15 wt.% NaOH for 2 h | Epoxy | - | - | - | Cai 2016 |
| Flax | Yarn and biaxial weft-knitted | NaOH of 8 wt.% at 25 °C for 2 h + wash with water + neutralise with 5 wt.% Ac acid for 30 min + wash with boiled water for 30 min + wash with tap water + air dry +oven at 105 °C for 6 h | UP | 5 | 19.8 | Increase & 29.5% | Xue 2013 |
| Alfa fibre | Nonwoven sheet (Alfa fibre +wool +PET/PE) | NaOH+NaClO | UP | 5.2 | 5.2 | Increase the thermal properties | Triki 2013 |
| Kenaf | Straw | NaOH 6% for 48h at 19°C + dry 5 h at 110°C | UP | - | - | Increase the flexural modulus 63% | Aziz 2005 |
| Palmyra | Mat | NaOH 5% for 30min + neutralise HCl + dry 70°C | UP | 60 | 37 | Impact strength increment | Thiruchitrambalam |

Table 4.2: Mercerisation and NaCl treatments references

| palm and jute | | | | | | 55% | 2012 |
|------------------|--------------------------|------------------------------------------------------------------------------------------------------------|-------------------------|-------|------|-----------------------------------------------|--------------|
| Sisal | Aligned nonwoven mats | 5% NaOH for 1 h at 30°C + wash distilled water + vacuum dry | UP | - | 22 | 43% Flexural strength and 21% impact strength | Misra 2002 |
| Hemp | Mat | 5% NaOH for 1h at RT + wash with water + neutralize with 2% ac acid to pH 6 + dry 24 at RT + vacuum dry | UP | - | 34 | - | Mehta 2006 |
| Sisal | Straw | NaOH 2M at RT for 2h + washed with water + neutralised with Ac acid + dry at RT | Soy-resin | 110.7 | 34.5 | - | Kim 2010 |
| Sisal | Straw | 5% and 10% NaOH + bleached with NaClO/H2O (1:1) at 60–75 °C | Cardanol based resin | - | - | Improve the weight loss and thermal stability | Barreto 2011 |

Other (bast) fibres, such as ramie, may also be mercerised to modify their chemical and morphological properties. Cheek and Roussel have reported on two different mercerisation methods, namely slack and tension mercerisation (Cheek & Roussel, 1989a and 1989b). In both cases, the NF was pre-treated with 0.05% Triton QS-44 (a surfactant), followed by immersion in NaOH 20% for 5min, and finally neutralised with 2% acetic acid solution. The two treatments differ in that the slack fibres are treated directly over the fibre while tension fibres are wrapped over a steel frame in order to keep fibres equidistant. Following slack treatment, the ramie yarn strength was reduced by 45%, and for flax by 49%, while in cotton the value was increased by 26%. In contrast, using the tension method the increment for ramie was 7%, for flax 22% and for cotton 38%. The clear conclusion is that when the fibre is kept under tension the desired strength increment is achieved; in contrast when the fibres are loose the mercerisation effect may be negative. In all of the above cases, the elongation of the break property is improved. Cheek (Cheek & Roussel, 1989b) performed the NF NaOH treatment to improve the dyeing process.

Goda (Goda, Sreekala, Gomes, Kaji & Ohgi, 2006) mercerised ramie fibre under a certain tension value, obtaining an increase of up to 18% in tensile strength and 37.5% for the strain at break values.

Another option for the chemical modification treatment of the natural fibre would use sosa/anthraquinone in combination. This technique has been used for many years in the flax industry, particularly to bleach the flax fibres (Fillat, Pepió, Vidal & Roncero, 2010). The elimination of the lignin fraction achieves the desired aesthetic appearance of the product but contributes little to its mechanical properties since it does not directly increase the modulus and tensile strength. On the other hand, other properties are improved such as the flexural behaviour of the composite.

Aly et al. presented a study using Box–Behnken Method (Software) for the determination of the optimal mercerisation condition of flax fibre (Aly et al., 2012). The fibres were treated with a solution of NaOH 5% at 55 °C for 10 min.

NFRP references were classified according to TS matrix or TP matrix in order to facilitate comparative issues.

Mahjoub et al. have reported on a kenaf fibre/epoxy resin system using mercerisation to enhance mechanical properties (Mahjoub, Yatim, Sam & Raftari, 2014). The composite strain properties were generally increased when the kenaf fibre was treated with 5% NaOH solution for 3h, washed with water, kept in distilled water for 24h and finally dried in the oven for another 24h. Fiore et al. immersed kenaf fibres in NaOH 6% solution at RT for 48h/144h, thoroughly washed them with distilled water, dried at RT for 48h and finally dried in the oven for 6h at 100°C (Fiore, Di Bella & Valenza, 2015). For the epoxy composite three-point dynamic test, the 48h composites storage properties increased; however, in the case of 144h treatment the mechanical performance was substantially decreased because exposure to the excessive NaOH treatment caused fibre damage.

Yousif et al. presented a kenaf/epoxy system flexural properties study concluding that NaOH 6% treatment to the kenaf fibre gives 16% enhancement to the composite flexural strength (Yousif, Shalwan, Chin & Ming, 2012).

Van de Weyenberg et al. proposed a mercerisation treatment for flax fibre in which the mechanical properties were best improved when the flax fibre was immersed in a 4% NaOH solution for 45s in order to increase the flax/epoxy composite transverse strength by up to 30% (Van de Weyenberg, Truong, Vangrimde & Verpoest, 2006).

Gassan et al. proposed a similar method to improve the interfacial properties of a jute/epoxy system (Gassan & Bledzki, 1999). Resin shrinkage is a key factor in the mechanical properties

of the final composite. When jute fibre is treated with NaOH 25% solution for 20 min at 20°C the resulting mechanical properties of the composite rise considerably; the jute yarn tensile modulus and strength increase respectively by about 150% and 120%.

McKelvey et al. (McKelvey, Webre & Klein, 1959) added epoxy matrix to the cotton yarn to enhance the mechanical performance of the fibre with the fibre swelling when mercerised at different temperatures and timed immersion conditions. The conclusion was that the tensile strength of the composite was reduced by 50-70%.

Cai et al. studied the mercerisation effect in an abaca/epoxy composite system (Cai, Takagi, Nakagaito, Li & Waterhouse, 2016). The results showed that the mechanical properties developed in low concentrations at 5% to directly increase the composite performance. In contrast, higher concentrations damaged the fibre concluding, therefore, that mild mercerisation processes were more beneficial for this system.

Xue and Hu (Xue & Hu, 2013) reported mercerisation treatment in a biaxial weft-knitted flax fabric concluding that flax/UP composite tensile strength and strain at break numbers were substantially improved.

Triki et al. (Triki, Guicha, Ben Hassen & Arous, 2013) reported on improving the interfacial properties of an alfa fibre/UP composite system combining NaOH and NaClO treatments in order to improve the development of mechanical and thermal properties. Likewise, Aziz et al. (Aziz, Ansell, Clarke & Panteny, 2005) improved the interfacial properties of four different UP matrix composites that were reinforced with kenaf fibre. Each system performed differently, but in each one the mechanical and moisture resistance properties were improved following mechanical treatment of the fibre.

Shanmugam and Thiruchitrambalam (Thiruchitrambalam & Shanmugam, 2012) modified Palmyra palm leaf stalk with mercerisation using benzoylation and permanganate treatment,

which resulted in an increase in the tensile, impact and moisture absorption properties of the UP composite system.

Misra et al. (Misra, Misra, Tripathy, Nayak & Mohanty, 2002) dipped sisal fibres in a sodium chlorite solution with a liquor ratio of 25:1 at 75°C for 2h, washed with distilled water, neutralised with 2% solution of sodium sulphite (Na₂SO₃) for 15 minutes at RT using a liquor ratio of 25:1, and finally washed and vacuum dried to obtain the bleached fibres. The impact and flexural properties of the sisal/polyester composite were raised after the bleaching process because of the elimination of the interfacial and lignin properties respectively.

Mehta (Mehta, Drzal, Mohanty & Misra, 2006), investigated the tensile strength of a hemp/UP composite system which showed an increase of 34% following alkali treatment of the hemp.

An alternative method was proposed by Kim and Netravali (Kim & Netravali, 2010) to enhance a sisal fibre/soy resin composite system. The sisal fibre was immersed in a NaOH 2M solution at RT for 2 hours, washed with water, neutralised with acetic acid and finally dried at RT. The treatment was undertaken to slack and tension fibres, the latter technique achieving better results. Compared with the reference control sample, the best results were achieved when the fibre tension was 50g/fibre, with 111% modulus and a strength increment of 34.5%.

In alternative bio-composite option sisal fibre was reinforced with a cardanol based resin (Barreto, Rosa, Fechine & Mazzetto, 2011). The alkali treatment improved the thermal and weight loss properties of the system.

In conclusion, it would appear that NFRP mechanical properties are improved in certain cases by using mild treatments whilst in other examples the treatment selected are more aggressive; also the mercerisation process is improved when the fibre has tension. For this reason, the current research project will investigate a wide range of mercerisation conditions in order to determine which may be best suited for producing a flax/bio-epoxy composite. The research

question is if the interfacial properties increment achieved by mercerisation has real impact in flax/bio-epoxy composite macro-mechanical properties; increasing macro-mechanical properties, such as transverse tensile strength.

4.5.2. Silanisation

Silanisation is a chemical treatment that will increase the interfacial properties of NF composites based on obtaining covalent bonding over the NF surface (Figure 4.6). The concept is to secure a reaction between NF surface OH groups and silane OH groups. The reaction that occurs in the interface will improve the mechanical and aging properties of the NF composites.



Figure 4.6: Si-O-Si bonding scheme

It is clearly understood that fibre coating and treatment are different concepts. The GF coating is normally obtained by applying a thin coating of sizing to cover the fibre surface. A similar process is followed for NF coating. In the fibre manufacturing process silane sizing is sprayed over the GF filament surface (Thomason & Adzima, 2001). The correct application of the coating and its composition will directly affect the interfacial adhesion of the composite, and consequently decide its final mechanical properties. Dey et al. (Dey, Deitzel, Gillespie & Schweiger, 2014) evaluated the performance of a GF/epoxy composite with silane sizing and clearly found that the azamide-based film former produced a 47% increase in interfacial shear strength over amino silane sizings with different film formers. The relative reactivity and wettability of the film former can enhance the chemical bond formation as well as the roughness of the fibre surface.
The most popular additive is silane. Onjun (Onjun & Pearson, 2010) reported on the importance of adding silane to the epoxy matrix into the interface area of the composite. The silane creates a bridge between the fibre and the matrix and hence enhances the long term mechanical performance of the composite. The silane increases the GF/epoxy hygrothermal resistance due to an increase in the subcritical debonding energy (G) of the composite.

Table 4.3 compiles references to the most interesting NF silanisation literature. Many of the studies, however, involve arduous and expensive methods to reach attainment.

| Fibre | Reinforcement format | Treatment | Silane | Matrix | Gain in modulus (%) | Gain in strength (%) | Other effects | Reference |
|-------------------|-------------------------------|-------------------------------------------------------------------------------------------------------------------------|-------------------------------------------------------------------------------------------------------------------------------------------------------------------|--------|---------------------------|------------------------------------------------|-----------------------------------------------------------------------|-----------------------------|
| Glass | - | Silane | 3-aminopropyltriethoxysilane and glycidoxypropyltrimethoxysilane | Ероху | - | - | Increase hygrothermal resistance + increase debonding energy | Onjun 2010 |
| NF | - | Silane | General | TP/TS | - | - | Review of the literature | Xie 2010 |
| Flax and Ramie | Straw | Silane | A-1100 and A-1120 silanes | - | - | - | Measure water contact angle | Gliesche 1995 |
| Cellulose | Microcrystalline particles | 5w/w% silane suspension in 80/20 v/v ethanol/water mixture 2 h + dry 2 days at RT + oven 2h at 120°C | γ –Methacryloxypropyltrimethoxysilane (MPA) γ-Aminoproyltriethoxysilane (APS) Hexadecyltrimethoxysilane (HDS) γ-Mercaptopropytrimethoxylsilane (MRPS) | - | - | - | Silane cellulose adhesion + Enhance surface to polymer grafting | Abdelmouleh 2004 |
| Jute | Yarn | NaOH 5% for 1.5h+washed water+UV radiation + Silane | 3-(trimethoxysilyl)-propylmethacrylate | - | 265 | lesser water uptake and less weight loss | Increase ε _f 350% | Hassan 2003 |
| Hemp | Straw | Silane 3% methyl alcohol water (60:40) + Wash in distilled water + dry at RT for 8 h + oven dry at 100°C 6h | Oligomeric siloxane | - | - 61 | - 33 | - | Kabir 2013 |
| Sisal | Straw | 0.2M silane 80/20 (v/v) ethanol/water mixture at RT for 72 h + wash with mix + dry RT | 3-aminopropyltriethoxysilane (APS) and N- (2-aminoethyl)-3- aminopropyltrimethoxysilane (AAPTS) | - | - | - | Prove silane NF covalent bonds are formed | Zhou 2014 |
| Flax | Straw | 1-3% silane acetone and water 50/50 v/v for 2h + dry 8 h at 80 °C | 3-aminopropyl trimethoxy silane | Ероху | - | - | Flexural longitudinal modulus increased 46% and strength 4% | Van de Weyenberg 2003 |
| Hildegardia | Fabric | NaOH + Silane (Silane/acetone sprayed over the fibre) | Silicon-based silane | Ероху | - | - | Reduction of voids content + increase fibre-matrix bonding | Guduri 2007 |
| Abaca | UD fabric | NaOH+ 1wt% Silane in 1:1 alcohol/water for 24 h and pH 5.3 + wash in water + oven dry for 2 h at 100°C | γ-glycidoxypropyl-trimethoxy silane | Ероху | - | 80 transverse | - | Liu 2014 |
| Sisal | UD fabric | NaOH + 2% silane in alcohol for 5 min at pH 5 + oven dry 2 h at 100°C | KH550 g-amine propyl triethoxysilane | Ероху | - 1 | - 16 | Increase ε _f 47% | Rong 2001 |

Table 4.3: Silanisation methods references

Xie et al. (Xie, Hill, Xiao, Militz & Mai, 2010) reviewed the use of silane coupling agents in NF/polymer composites. In general, trialkoxysilanes have been selected to improve the interfacial properties of the NFRP. The silane coupling agents generic chemical structure is $A_{(4-n)}$ -Si-(R'X)_n (n = 1,2) where A is alkoxy, X represents an organofunctionality, and R' is an alkyl bridge. The alkoxy will react with the NF surface and the R' organofunctional group will react with the organic polymer matrix because of their similar polarities. Nonreactive alkyl groups in the silane increases compatibility with the matrix due to their similar polarities; however, the reactive organofunctionality may covalently bond with the polymer matrix. The organofunctional parts of the silane usually belong to amino, mercapto, glycidoxy, vinyl, or methacryloxy groups. Aminosilanes, especially γ -aminopropyltriethoxysilane (APS), are reported most extensively in the literature as coupling agents between natural fibres and thermoplastics or thermosets.

Gliesche and Mäder used silanes (A-1100, A-1120 from OSi-Specialties) as coupling agents in flax and ramie fibres (Gliesche & Mäder, 1995). Measurement of the contact angle of water using the capillary rise method showed no effects. Typical water contact angles for the untreated flax and ramie fibres were 87° and 77° respectively.

The silane treatment of the cellulose fibres enhances the fibre surface to produce polymer grafting and decrease its hydrophilicity (Abdelmouleh et al., 2004). The silane treatment resulted in O-Si-O and C-Si-C bridges being formed between the fibre surface and the silane groups. Fibres were treated with 5 w/o silane suspended in 80/20 v/v ethanol/water mixture stirring for two hours; the obtained fibres were dried for days at RT; and finally cured at 120°C for two hours in a nitrogen atmosphere.

Hassan et al. (Hassan, Islam, Shehrzade & Khan, 2003) reported improved tensile strength, elongation to break and durability in mercerised jute fibres which had been grafted with silanes and acrylamide under ultraviolet radiation. This conclusion predicts that polymer loading could be easily accepted in the future production of composites. The mercerisation might be applied to the NF under UV radiation in order to increase its effect. Kabir et al. (Kabir, Wang, Lau & Cardona, 2013) presented a study where the hemp fibres were subjected to alkali, acetyl and silane treatments. When hemp fibre is directly treated with the silane, the mechanical properties are drastically reduced: 61% for tensile modulus and 33% for strength. In contrast, when silane is applied to a previously mercerised fibre the properties are enhanced. However, in both of these cases the mechanical properties would not reach untreated fibre composite mechanical properties.

Zhou et al. (Zhou, Cheng & Jiang, 2014) reported that when sisal fibres are treated with silane covalent bonds are formed. Van de Weyenberg (Van de Weyenberg et al., 2003) used different fibre treatments to increase the base properties of the flax fibre. Table 4.4.

Table 4.4: Flax fibre base tensile and flexural properties

| | Tensile strength (MPa) | Flexural strength (MPa) | Tensile modulus (GPa) | Flexural modulus (GPa) |
|--------------|------------------------------|-------------------------------|-----------------------------|------------------------------|
| Longitudinal | 133 | 218 | 28 | 17.7 |
| Transverse | 4.5 | 8 | 2.7 | 0.36 |

In all methods applying mercerisation, the first step requires the fibres to be dipped in a NaOH solution of different concentration (1, 2 or 3%) for 20 min at RT. After removal, the fibres are washed very thoroughly in cold water, then placed in acidified water (20 drops of HCl 0.1 M in 1 l of water) to remove excessive NaOH. The fibres are again rinsed in cold water and dried in an oven at 80 °C for eight hours. Silanisation is undertaken by soaking the fibres for two hours in a solution of equal volumes of acetone and water containing 3-aminopropyl trimethoxy silane with concentration of 1%. The fibres are then dried in an oven for 8 h at 80 °C. Significant improvement is achieved when the flax fibres are treated in a 1% NaOH solution and covered with 3% epoxy resin solution. Flexural longitudinal tensile modulus and strength are increased by approximately 58% and 38% respectively. Treatment by 1% silane leads to increments of 46% and 4% in longitudinal modulus and strength properties. Additionally, it is reported that with silane the transverse flexural modulus and strength properties are increased 400% and 110% respectively.

Van de Weyenberg's treatments were considered to be the most effective, the easiest to perform, are economically viable as well as being the greenest.

In certain studies, e.g. Guduri et al. (Guduri, Rajulu, & Luyt, 2007) the Hildegardia NF reinforced epoxy matrix is toughened with polycarbonate polymer. Additionally, the NF is treated in a NaOH solution for one hour and afterwards the surface is sprayed with a 1% silane coupling agent. These treatments elevate the adhesion between the fibre and the matrix but water resistance is decreased. This might be caused by the fibre surface being more polar resulting in an increase in its hydrophilicity.

Liu et al. (Liu, Zhang, Takagi, Yang & Wang, 2014) proposed a combined chemical treatment for improving the transverse mechanical properties of a unidirectional abaca fibre/epoxy system. Mercerisation formed the first step in the process by the mild immersing the fibre for five minutes in a solution of 1.0 w/o NaOH, and hard immersing the abaca for 30 minutes in a 5.0 w/o % NaOH solution. After the RT immersion both of the treated bundles were thoroughly washed with water and dried in the oven at 70°C. In the second step silane is applied to the mercerised abaca fibre. 1.0 w/o acetic acid, 1.0 w/o y-glycidoxypropyl-trimethoxy silane, 49 w/o alcohol and 49 w/o water solution is prepared mixing all the components for 60 minutes in an opaque container. Next the fibre is immersed in the solution for 24h keeping the solution at 5.3 pH. The fibres are then washed with distilled water and kept at RT for 30 minutes. Finally, the abaca fibre is dried in the oven for 2h at 100°C in order to obtain covalent bonding between the fibres and the silanes. The authors concluded that thorough treatment of the fibres with silanes can increase the interfacial adhesion of the resulting composites and improve their mechanical and outdoor performance. For example, when the abaca fibre is mercerised with 5% NaOH for 30 min, followed with silane, the composite transverse tensile strength at $V_f = 0.3$ is increased by 80%. However, the Si–O–C bonds in natural fibre composites are less stable under hydrolysis than the Si-O-Si bonds in glass fibre composites (Shokoohi, Arefazar & Khosrokhavar, 2008).

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Rong et al. (Rong, Zhang, Liu, Yang & Zeng, 2001) showed that the silanisation process in a sisal/epoxy system did not produce the expected mechanical performance increment. There was, however, a 47% increment in the strain at break value, but the tensile modulus and strength values decreased.

The use of silanes will inevitably increase the proportion of silica in the ash if incineration is the chosen route for the disposal of a bast fibre composite. This ash, in the form of a respirable dust, may be a relatively harmless by-product (Normohammadi, Kakooei, Omidi, Yari & Alimi, 2016).

In the above paragraph the main NF silanisation treatments were generally explained. The research question is if any of these treatments might be applied to the flax/bio-epoxy composite in order to increase their interfacial, macroscopic mechanical properties and moisture absorption properties. The silanisation is made to the untreated and mercerised flax fibre and compare their performance.

4.5.3. Other treatments

Acetylation is the process of introducing an acetyl radical into an organic molecule, and in the process the hydroxyl groups in the cellulose molecule react with the acetic acid (or anhydride) to produce cellulose acetate. The replacement of the polar hydroxyl group with the nonpolar acetyl group increases the hydrophobicity of the cellulose making it more compatible with nonpolar matrix polymers, i.e. the fibre adhesion to the non-polar polymer matrix would be raised.

Another option would be the acrylation. In the literature, references to NF treatments with acrylic acid are limited. This may arise because acrylic acid is harmful to human health producing eye irritation and pulmonary edema among other negative issues. Acrylic acid is used to increase moisture resistant properties and does not enhance interfacial properties.

Benzoyl chloride is also harmful to human health and a "short exposure could cause serious temporary or moderate residual injury" as noted in the NFPA 704 standard (Standard System for the Identification of the Hazards of Materials for Emergency Response). Basically, the method replaces

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polar OH groups with non-polar benzoyl groups, it is suggested that hydrogen bridges are formed in the interface to justify mechanical and thermal property increment.

Cyanoethylation involves the reaction of alcohol groups on the cellulose molecule with 2propenenitrile (a.k.a. acrylonitrile, cyanoethylene or vinylcyanide) to form a nitrile derivative which can be catalytically reduced to the corresponding primary amine. This amine group might form a covalent bond with the polymer matrix, subsequently enhancing the mechanical properties.

The etherification of NF is another possible treatment that can modify the final properties of composites. Functionalisation of the fibre surface commences with the activation of the hydroxyl group to increase hydrophilicity; the formation of these charged groups will attack other species compounds to form a new covalent ether bond.

Graft copolymerisation is an effective method in the modification of natural fibre surfaces to enhance fibre/matrix interaction. Graft polymerisation is based on the introduction of a new polymeric chain, polymer B, onto a principal polymeric chain, polymer A. The method will first produce activated points by which the desired synthetic branch can be attached to the main polymer chain. The new polymer branches will work as new anchoring points, or increase the similarity between the fibre surface and the composite matrix and substantially improve the interface properties.

Oxidation of NF is another useful, low cost and simple process of increasing the mechanical properties in the interface, although oxidative substances may constitute health hazards. The peroxide group tends to decompose into two radicals which interact with the hydrogen group on the NF surface transferring the radical to the fibre surface. The fibre and matrix grow more compatible with equalisation of the polarities allowing for an improvement in the tensile properties of the composite.

This process could be a useful option for improving the interface of a specific polymer/matrix system but may not be applicable in all cases. However, the isocyanate group is implicated as a cause of cancer and is not environmentally green. Isocyanate treatment of NF creates covalent bonds between the fibre hydroxyl group and the isocyanate group (-N=C=O) producing an enhanced fibre/matrix interaction.

4.6. COUPLING AGENTS ON THE FIBRE SURFACE

Fibre surface coatings (FSC) for synthetic fibres may include antistatic agents, binders for easy processing, lubricants for textile processes, and coupling agents to promote good adhesion between the fibres and the resin lubricants. For the highest quality composites, the initial surface coating may be burnt off after the fabric has been produced and the textile is then coated with a high proportion of coupling agent. This is impractical for NF as the fibres would be incinerated along with the FSC. There are a variety of surface coatings that have been used to increase the strength of the bond between fibres and the matrix. They include silanes, titanates and zirconates, maleated polyolefins, isocyanates, maleimides and triazines. These agents normally have different chemical functionality at the two opposed ends such that the molecule bonds to the two different components of the composite. Figure 4.7 illustrates a coupling agent performance scheme.



Figure 4.7: Coupling agent performance scheme

After the treatment process, the fibre structure and performance of the composite are modified while in the fibre coating only the fibre surface is altered. Coatings can be applied to synthetic and NF composites, and in both cases a reaction between fibre and silane OH groups is desired in order to achieve optimum bonding properties.

4.6.1. Synthetic fibre coating

GF is one of the main reinforcements in the composite industry. The GF filaments are covered with the pertinent sizing at 0.2-0.3% of reinforcement weight. The coupling agents are the most important sizing component. The coupling agents improve the resin/fibre interaction appreciably, and enhance the mechanical properties of the composite both in the short term and after aging. Water may be one

of the most damaging agents to the interface with moisture absorption directly decreasing its mechanical properties (Plonka et al., 2004). This damage is considered to be greater in a marine environment because of the presence of Na⁺Cl⁻ ions in the seawater (Wei, Cao & Song, 2011; Wood & Bradley, 1997; Gellert & Turley, 1999). In GF, the coupling agent is usually the silane, forming a bond between the GF surface and the pertinent matrix. Figure 4.8 presents the silane formula showing its reaction with the GF surface.



Figure 4.8: Silane schematic figure

The coating is usually applied by wetting the target fibre with a sizing emulsion but there are critical emulsion compositions and application techniques; this technique would be replicated to the NF treatment with silanes.

4.6.2. Natural fibre coating

As with synthetic fibre coating, applying a polymer coating to the NF fibre surface will improve the interfacial properties of the composite. Although NF chemical or physical modifications are more popular techniques, fibre coating is also considered to be a most useful procedure. Jana and Prieto (Jana & Prieto, 2002) presented a wood fibre coating method to improve the interfacial properties. Three different coupling agents were proposed to secure the most complete adhesion of the epoxy resin coating to the wood particle surface. Once the coating had fixed the particles, the epoxy operated as a bridge between the wood layer and the thermoplastic matrix. This same concept may be applied to the flax/bio-epoxy system which is the subject of the current research project.

Xie et al. (Xie et al., 2010) reviewed the use of silane coupling agents in NF/polymer-based reinforced composites. This concept is basically replicated for the current study flax fibre treatment, in order to increment the flax/bio-epoxy adhesion.

4.7. MODELLING CHEMICAL TREATMENT

4.7.1. Mercerisation (NaOH) and NaCl

Bledzki (Bledzki, Fink & Specht, 2004) offered a direct model which related the mercerisation process to the final mechanical properties of a hemp fibre/epoxy resin system. For the formulation of an epoxy resin composite model, the considered parameters were listed as the alkali concentration, the temperature, and treatment time applied to the fibres and the tensile strength. In this example the model was based on the change in cellulose crystallinity from cellulose I to II under the mercerisation treatment. Similar conclusions were obtained by Borysiak (Borysiak, 2013) where the transcrystalline morphology progression in a wood fibre/PP composite was studied under different treatments, concluding that a direct relationship developed between treatments and the growth of the crystalline progression.

Another model related the change in cellulose morphology with moisture absorption after mercerisation, bleaching and thermal treatments (Stana-Kleinschek, Strnad & Ribitsch, 1999). The author concluded that after the mercerisation treatment moisture absorption increased because of wax elimination and the change in cellulose crystallinity. Zou et al. (Zou, Wang, Gan & Yi, 2012) similarly concluded that in a sisal/PLA composite moisture absorption increased after the mercerisation of the fibres. it was concluded that NF treatments increase the water absorption rate although after treatment the fibre surface is cleaned and activated thus improving the interaction with the matrix and, as a consequence, reducing fibre absorption.

Jandas et al. (Jandas, Mohanty, Nayak & Srivastava, 2011) studied the interfacial properties of a banana fibre/PLA composite with different models. The banana fibre was treated with different chemical reactants modifying the surface by mercerisation which subsequently increased the mechanical performance of the composite. Mechanical prediction models, such as Hirsch's, modified Bawyer and Bader's, and Brodnyan were then applied to study the theoretical prediction process of the composite.

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4.7.2. Silanisation

There are not many references in the literature where a prediction model has been applied to study the long term performance of the silanised fibres. Liu et al. (Liu, Jones, Liu & Jiang, 2014) for example, investigated the effect of air plasma and a silane coupling agent on polyurethane adhesion with human hair. The two systems were studied under immersion artificial sweat solution aging a 50°C for 24h, and micro-droplet technique was used to follow the IFSS loss in the aging. It was concluded that the air plasma technique was more efficient in maintaining the mechanical properties, performing better than the silane treated human hair.

However, the previous method evaluates the long term properties, but the model prediction is not matching correctly the experimental results. In contrast, Hidalgo-Salazar et al. matched experimental results to the Burger theoretical creep model curve (Hidalgo-Salazar, Mina & Herrera-Franco, 2013). Fique fibres were mercerised and silanised to improve their interfacial properties with recycled LDPE-Al bulk matrix. The model works best for the case of untreated and mercerised fibres composites.

4.8. CHARACTERISATION OF FIBRE SURFACES AND INTERFACES

This section describes the different techniques used to characterise fibre-matrix adhesion. All these techniques aim to quantify the fibre-matrix adhesion of a given composite system in order to predict its macroscopic performance. The characterisation techniques may be divided into the following categories:

- 1. contact angle study
- 2. reactive sites on fibre surface
- 3. mechanical interface characterisation
- 4. optical qualitative analysis.

4.8.1. Contact angle study

The technique is based on measuring the contact angle between the fibre and the composite matrix. Depending on the values achieved the fibre wettability would vary. In the NF surface treatments case, the fibre surface energy differs and is directly related to the NF wetting, related to the interfacial properties. The Wilhelmy technique involves dipping a single fibre into a non-penetrating liquid while measuring the force on the fibre due to wetting.

4.8.2. Reactive sites on fibre surface

The NF surface has many -OH reactive groups. Matrix adhesion is different depending on the treatment selected for the composite system. There are references in the literature that determine the hydroxyl number of the fibre surface, as for example in the method presented by Freire et al. (Freire, Silvestre, Neto & Rocha, 2005), where the –OH group number is determined by using a complex gas chromatography-mass spectroscopy (GC-MS) method.

4.8.3. Mechanical interface characterisation

The mechanical characterisation of the interface can be divided into three main groups - direct methods, indirect methods and composite laminate methods, as explained in Figure 4.9 (Drzal et al., 2000). Direct methods have been studied in greater detail because of their relevance in the process of evaluating NF composites.





4.8.3.1. Composite lamina methods

Composite laminate tests are used to measure fibre-matrix adhesion. Interface dominated tests are used such as 90° tensile test; 90° flexural test; four-point shear test, short-beam shear test, Mode I/Mode II delamination tests, and in-plane 10°/45° shear test. Many of these tests are formalised by international standards such as ASTM or ISO standards. These documents are the guidelines that govern test performance, the data to be achieved and the analysis of the results.

Composite testing in general are very useful techniques for the evaluation of interfacial properties and are particularly important in the evaluation process of NFRP interfacial properties.

4.8.4. Optical qualitative analysis

These complementary techniques provide information on surface chemical and mechanical characteristics. For example, yarn swelling could be studied by optical microscopy or wetting properties determined with X-ray photoelectron spectroscopy (XPS). Other important techniques, such as IR spectroscopy, SEM fractography, Raman spectroscopy and surface energy analysis are used for the evaluation of NFRP interfacial properties.

It is of fundamental importance to characterise the NF surface in order to identify the most relevant treatment that can be applied to its surface morphology and chemistry that will subsequently have a direct effect on its mechanical and flow properties.

4.9. SUMMARY AND FUTURE DIRECTION

Chapter 4 reviewed literature references to matrix and NF modifications with particular reference to composite mechanical systems and ageing property enhancement. This chapter gives the continuation to Chapter 2 where the conventional composites were revised and Chapter 3 where the natural fibres were studied.

Chapter 4 review focused on studies that enhance the understanding of interfacial properties, the application of epoxy resin silane additions, the use of NF mercerisation/silanisation treatments and

the evaluation of modified properties systems. The same silane has been selected and the amine is expected to participate in curing the epoxy matrix that has been selected for this study.

Similarly, the NF surface will be modified and the bio-epoxy resin formulation changed in order to increase the fibre matrix compatibility value. The addition of silane to the matrix and the chemical treatment of NF are the principal techniques which will be investigated. The aim is to increase the flax fibre/bio-epoxy system by tailoring the interphase region in a commercially competitive manner.

In the initial stages of the present experimental project the plan was to perform acetylation treatments on the flax fibres in order to increase their fibre-matrix adhesion properties. Acetylation was considered because it was both effective and economical to apply; however, time limitations prohibited its use. Mercerisation techniques were therefore applied which have the advantage of activating the NF surface while acetylation merely reduces fibre hydrophilicity.

The silane was added to the bio-epoxy in order to strengthen those properties impeding moisture ageing. In order to improve performance moisture tests under ASTM D5229 standards were undertaken. The aim is to discover whether or not the presence of silane in the interface will improve the moisture resistance of the composite as well as enhancing the mechanical properties.

The current study has proven that silane addition to the bio-epoxy matrix clearly improves the ILSS values by approximately 20% of the reference system. In-plane shear properties were also studied and improved; 90° tensile test and in-plane 10°/45° shear test were of particular relevance. These values can be checked in Chapter 8 which refers to the experiments undertaken in the research project.

The literature has been searched for investigations that refer to the modelling of NF treatments. The current PhD tried to set a treatment model based on the flax/bio-epoxy composites evaluation experimental results, Chapter 8; and model set in Chapter 9.

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The project surface modification was not studied in detail by advanced optical analysis, such as SEM fractography, but fibre swelling after the chemical treatment was studied with the aid of optical microscopy. See result Chapter 8 and Appendix C.

Based on the collected information the mercerisation, silanisation and silane addition to the matrix and indicative conditions were identified and will be the focus of the research undertaken in this doctoral study. The research methodology is described in detail in Chapter 7. The results obtained for treated systems are presented in Chapter 8 and Appendix C. The effect of different treatment factors (e.g. treatments temperature and concentration) upon the flax/bio-epoxy composite system were evaluated with the ANOVA technique and are discussed in Chapter 9 with conclusions drawn in Chapter 10.

5. Composite design, manufacturing, testing and image analysis

5.1. INTRODUCTION

In the production process of a composite optimised design, it would follow the steps described in this Chapter; it will concentrate on the material design of the composite, its manufacture, its testing and the image analysis of the prototype. The four subjects above are directly inter-related and the processes involved in the production of the composite prototype are shown in the chart below (Figure 5.1). The input data for the design of the composite will be obtained from mechanical scale testing in the laboratory; once the designed prototype has been tested it will be manufactured; running concurrently with the manufacturing procedure will be a rigorous process of checking the quality of the product by image analysis.

Figure 5.1: Composite design, test, manufacturing and image analysis relationship chart

All the sections would be focused on the techniques developed and used along the PhD study, such as resin infusion, development of flow models or optical microscopy.

5.2. COMPOSITE DESIGN

There are different approximations in the composite design process; for example Kedward (Kedward, 2000) (Figure 5.2) proposed a methodology and Potter (Potter, 1992) another. In general most of the methods are based on the same common point checklist, based on product function, geometry, environment, duty, cost issues and programme/contact issues.



Figure 5.2: General scheme for the design of the composite (Kedward, 2000)

The starting point of any design project as proposed by Kedward is the "**conceptual design**", which must be defined specifically and clearly to allow for the success of the project. There are a number of factors to be considered in setting the concept, and in the context of sustainable composites they can be simplified to seven main groups; ¹geometric envelope, ²material selection, ³loading conditions, ⁴environmental conditions, ⁵manufacturing processes, ⁶inspection methods, ⁷cost evaluation and life cycle assessment.

Following the "conceptual design" stage, the composite design is preliminary defined in plans and drawings and candidate material systems may also be evaluated. In a second step, referred to as "**trade-off studies**". Trade-off involves the following procedures: first select an approximate design, check how close it matches the final requirement, and if inappropriate exchange or trade in to achieve the main goal. The outcome is the preliminary design; its structural configuration, its manufacturing definition and tooling requirement in order to achieve an accurate cost estimate.

"**Detail Design and Evaluation**" is the third step. For the study of the more detailed structural design it is imperative that secondary stresses, which usually appear in section joints or in phase transition zones, are carefully studied. Global and local FEM software models may be applied to the study of such stresses with particular reference to the angle structures of joints.

In the **"Prototype Manufacture and Evaluation**" stage the prototype is produced in accordance with the previously defined manufacturing steps. Once the prototype has been produced, it will be subjected to real scale testing to establish its correct service behaviour.

Once the prototype has been developed and evaluated the design process enters the "**production phase**". If the prototype evaluation is accepted, then the design will then be up-scaled into an industrial product and absolute and definitive data on production costs and time will be sought.

5.3. MECHANICAL TESTS

The mechanical tests performed on composites are classified into two main groups; first, the characterisation of the composite system will be tested at laboratory scale to obtain the necessary data on the design (more detailed practical example in Chapter 7); second, the final composite will be tested in order to compare its theoretical performance with its real scale performance. See Figure 5.3 below.





5.4. COMPOSITE MANUFACTURING

The main manufacturing techniques used in TS composites production are, open mould, prepregging/autoclaving, compression moulding, liquid moulding and continuous techniques.

5.4.1. Resin infusion under flexible tooling (RIFT)

The resin infusion technique is a modified version of the Resin Transfer Moulding (RTM) technique. In RTM, dry fabrics are enclosed by two rigid mould parts for the subsequent wetting of the fibre and the formation of the final composite piece. In the Resin Infusion under Flexible Tooling (RIFT) technique, one of the rigid mould parts is substituted by a flexible film (Williams, 1977). The RIFT technique was selected for the manufacture of all the test plates prepared in the present research project, which will be described in detail in Chapter 7 and the results presented in Chapter 8.

In the RTM technique, the size of the final piece is controlled by the size of the mould cavity. In the RIFT technique the thickness of the product depends on the pressure applied throughout the whole manufacturing cycle. With RTM, the piece may be more strongly packed under pressure of 3 bar, whereas with RIFT the pressures are lower with typical values of 0.8-0.9 bar (and an absolute 1 bar limit). RIFT is easier to use when manufacturing pieces of different sizes, particularly when they are large. This versatility results from the flexibility of the film under vacuum whereas with RTM the contra-mould tool is rigid under internal pressure.

The RIFT technique should not be confused with the vacuum bagging technique. The principal difference is that, with RIFT, the dry fabric (not wetted fabric) is placed onto the tool and the peelply, release film, **flow media**, (no bleeder or breather), and vacuum bag are enclosed in the system by the sealant tape. The resin is fed directly into the vacuum system from an external source to wet the fibres. Unlike with vacuum bagging the resin is introduced into the system once the bagged dry reinforcement has been prepared. Figure 5.4 illustrates how the system is prepared and how the liquid resin progresses to wet all the fibres.



Figure 5.4: Schematic figure of RIFT manufacturing technique (Courtesy of Richard Pemberton under a Creative Commons licence)

5.4.2. Flow models

The properties of composite materials are produced by using different techniques, as, for example, by resin infusion. The flow mesh properties will vary according to the technique selected and the characteristics of the fibre/ resin. The flow properties of each particular composite system and the selected manufacturing technique will directly affect the component V_f. In the studies of Shah (Shah, 2014), the flow dynamics and mechanical properties of different plant fibre composites are related to the V_f, and the predicted absolute theoretical maximum fibre content coincides with the experimental values.

Francucci and Rodriguez (Francucci, Rodríguez & Vázquez, 2010), in a study relevant to the present research project, evaluated how each NF treatment affected the Vf of the composite and in a similar study by Cherif et al. the sorption and mechanical properties related to NF treatment was investigated.

Based on the references mentioned in the above paragraphs, the current PhD produced a simplistic prediction flow model; such model relates the laminate flow properties with the flax fibre swelling produced by different treatments. Fibre swelling data is shown in Appendix C and the model discussed in Chapter 9.

5.5. IMAGE ANALYSIS

Composite performance cannot be understood by mechanical property analysis alone and it is most important to discover the interface and interphase properties in the adhesion between fibre and matrix. SEM is therefore a most useful and easy to apply technique to study the above properties. Macroscopic and optical tests are complementary and best adapted techniques to study fibre-matrix adhesion. Lee, et al. (Lee, Jhan, & Chung, 2012) presented a study where the CF surface was treated to increase its roughness for better adhesion with the matrix, and subsequently to improve the shear strength of the fibre-matrix. SEM was used to relate macroscopic test values with the data from a microscopic image as exemplified in Figure 5.5.



Figure 5.5: SEM images of a) Sisal/PP; b) Cotton/PP resin systems (Amigó et al., 2009)

The 2-D characterisation of fibre distribution within the image plane has been reviewed by Summerscales et al. (Summerscales, Virk & Hall, 2013). This technique can obtain similar accuracy in fibre volume fraction as can be achieved with the densities technique. Another application is the "tow structure" capacity determination which defines every yarn, tow or filament diameter, shape and spatial orientation of the fibre. Similarly, it is capable of identifying the different fibre types in a hybrid composite material reinforced with, for example, GF and carbon fibres. In the case of short-fibre reinforced composites the identification of the fibres, their spatial orientation and length can be determined by using image analysis.

However, in the current PhD study, the SEM technique was not applied. The optical microscope was used for the fibre swelling monitoring. The proposed model tried to find a relationship between the fibre swelling and the laminate V_f , with the subsequently affecting the macroscopic mechanical properties. See Appendix C for the swelling data, Chapter 8 for the macroscopic mechanical tests results and Chapter 9 to see the relationship between swelling and mechanical properties.

5.6. SUMMARY

Manufacturing details are summarized in Chapter 7. The results obtained from the test campaigns (Appendix B) and from image analysis (Appendix C) are illustrated in Chapter 8 and further discussed in Chapter 9. The results from Chapter 8 and Appendix B were introduced in the proposed mechanical prediction models to check whether they were working or not. Additionally, a flow model was proposed, the proposed model tried to find a relationship between the fibre swelling and the laminate V_f. See Appendix C for the swelling data and Chapter 9 for the model discussion. The composite design, manufacturing, testing and image analysis processes that were applied to the research project are discussed in detail in Chapter 9.

6. Composite durability in marine environment

6.1. INTRODUCTION

In this doctoral research project, silane was added to the flax/bio-epoxy composite system in order to enhance the composite mechanical and aging properties. In this Chapter, tests to examine the durability of the composite will be briefly explained and the ageing of the composite under marine conditions will be examined.

6.2. COMPOSITE DURABILITY

One of the most difficult issues to consider in the design of a bio-composite will be its mechanical properties after long term environmental exposure and hence its life prediction. The most reliable prediction would result from testing its long term performance as a case study product. However, such a programme would be long and expensive to perform and so data from accelerated ageing tests must be collated in an attempt to establish a direct correlation with the in-use composite performance.

Chapter 6 identifies the most common ageing agents in service applications and the tests performed on their respective components. The doctoral study considers the interfacial degradation of the composite in a marine environment in order to fully understand and interpret the test procedures conducted under these conditions on the NF/bio-epoxy system ageing process.

Normally after different ageing processes the mechanical properties (modulus, strength and strain-tofailure) when tested are considerably decreased. This arises as a consequence of composite chemical structural modification, characterised by matrix swelling, fibre/resin interface debonding, matrix micro-cracking, polymer chain scission and the solution of soluble components. There are a number of options for measuring ageing processes in composite materials, but the most common procedures are -

- Chemical resistance to liquids test
- Natural weathering test
- Marine environments test

6.2.1. MARINE ENVIRONMENT TEST

Marine environments may be hazardous for composite materials with water, then salt concentration, causing the main questions. It is therefore imperative to create a composite system that maintains its structural integrity in a marine environment and to discover how to overcome the most damaging factors. In the present research project, the main aim is to evaluate whether the silanisation process increases the resistance of the flax/bio-epoxy system in a marine environment (Appendix E).

GF/UP systems are selected for marine use because of their elevated corrosion resistance properties and are used in such components as marine propellers, sea water pumps, floating piers, offshore platforms, ship hulls and the strengthening of steel structures (Summerscales, 2014). Epoxy resin, CF or aramid fibre could also be used in marine applications although their utilisation would be more complicated. In the case of the marine application of a NFRP composite, the hygroscopic properties of the NF complicate its use due to moisture induced swelling and plasticisation; giving as a result mechanical performance reduction.

The eight factors of marine degradation can be listed as follows - water, salt, sand, UV radiation, marine vegetation/micro-organism attachment, marine pollutants and wave action. In each case, the harshest of these ageing agents must be identified and the experimental campaign planned in accordance. For the best test design, the composite service environment must be studied; it is not truly representative to have the composite component close to the sea, partially or totally immersed in the water; so depending on the environment the test would differ.

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Based on the above facts a marine immersion test, to be discussed in the next section, and a salt spray test can be applied.; these tests are usually employed to simulate industrial and/or environmental degradation.

6.2.1.1. Water absorption

Organic resins usually absorb water, while GF and CF do not. In contrast aramid fibre and NF do absorb water leading to the swelling of the fibres. As a result laminates absorb water which is usually limited by diffusion and in turn is ruled by Fick's law (Equation 6.1) (Tsai, Bosze, Barjasteh & Nutt, 2009).

$$\frac{M_t}{M_m} = k \cdot t^n$$
 Equation 6.1

Where M_t is the moisture uptake at time t, M_m is its maximum moisture uptake, at equilibrium state, and k and n are the diffusion kinetic parameters. The diffusion exponent n indicates the mode of diffusion. When n is equal to 0.5, diffusion obeys Fick's law. The mechanism is non-Fickian when n =1 (or n > 1) while the diffusion is anomalous when n shows an intermediate value, between 0.5 and 1. Moisture uptake in vegetal fibre reinforced plastics usually follows Fickian behaviour. Fick's law, in the case of a one-dimensional approach, shows that the water uptake increases linearly with the square root of time, and then gradually slows until an equilibrium plateau is reached.

Fickian diffusion process depends on density (ρ), specific heat (C), mass diffusivity (D), thermal conductivity (K) and maximum moisture content (M_m), in addition to other factors such as the geometry and the initial condition of the samples. The application of the Fickian process may therefore be complex and may not always produce satisfactory results. For example, Fickian predictions may be invalid when there are voids in the sample, or cracks/delaminations are present, or when moisture propagates along the fibre-matrix interface.

Scida et al. (Scida, Assarar, Poilâne & Ayad, 2013) investigated a flax fibre/epoxy resin composite system with hygrothermal accelerated ageing and monitored the degradation process with Fick's law

(Equation 6.2) and water absorption uptake (Equation 6.1) and finally measuring the composite's Young's moduli and tensile strength.

$$M_t = \frac{W_t - W_0}{W_0} \times 100(\%)$$
 Equation 6.2

Where W_0 is the weight of dry specimen and W_t is the weight of wet specimen at time t. After a 38 day experimental campaign under hygrothermal ageing, the Young's modulus and tensile strength were seriously affected by the ageing process. This weight change was monitored using the formulas in ASTM D5229 standard, where liquid uptake, Tg and moisture diffusion were monitored to follow the degradation properties in the ageing process.

In the research project, the flax fibre/bio-epoxy resin composite was investigated under RT marine ageing conditions (Appendix E). Flexural samples were water age tested and their degradation monitored. Water absorption and loss of mechanical performances were evaluated in untreated and in silanised systems alike and the data compared. The test procedures are described in Chapter 7, evaluated in Chapter 8 (Appendix E) and the results discussed in Chapter 9.

6.2.1.2. Marine immersion test

In this test the samples are immersed in a saline liquid and tested for their ageing with the ASTM C581 standard. However, if a marine environment immersion is to be simulated, ASTM D1141 adds a pre-determined quantity of different ions to the water which Millero (Millero, Feistel, Wright & McDougall, 2008) has defined in detail "The composition of Standard Seawater and the definition of the Reference-Composition Salinity Scale".

6.2.1.3. Composite mechanical deterioration

The composite is aged in a marine environment to establish its destructive processes by monitoring its tensile, compression, shear and impact properties (Al-Bastaki & Al-Madani, 1995; Gu, 2009).

The flax/bio-epoxy system was subjected to the moisture test referred to above. The test records the measurement of water uptake, and evaluates the deterioration in flexural mechanical properties of the sample. This data can be compared with the deterioration data from the flexural properties tests which are presented in Appendix E.

6.2.2. REFERENCE STUDIES

There are many studies in the literature which refer to the effects of marine ageing on NF reinforced composites. Le Duigou, et al. (Le Duigou, Davies & Baley, 2013) has evaluated the interfacial properties of a flax fibre/epoxy system after marine ageing, which raises similar questions to those addressed in the present research project. In the micro-droplet test conducted by the above authors, the interface debonds according to the length of time the composite has been immersed. Macromechanical and micromechanical degradation in a similar marine environment over a time period will be addressed in the research project.

6.2.3. IN SERVICE CASE TESTING

Although in many studies accelerated testing can be a very useful tool, it is however better accomplished if studied under actual, authentic conditions as, for example, when US army personnel reported the different ageing properties of boats and vessel components (Fried & Graner, 1966; Cobb, 1963; Graner & Della Rocca, 1971).

6.3. SUMMARY

In the first part of Chapter 6, the composite marine environment tests were described, and the procedure and standard information was collected in order to be applied to the flax/bio-epoxy system. Test procedure is described in Chapter 7 with obtained results in Appendix E, and the subsequent discussion in Chapter 9. The PhD main idea was to study whether the silane addition to the matrix achieves increment in the aging resistance, apart from upscaling to the case studies.

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7. Research methodology

7.1. INTRODUCTION

The literature review in Chapter 4 identified NF treatment and matrix modification methods to improve the interfacial properties of the NFRP. Selection considered the following factors - process simplicity, procedure effectiveness, sustainability, and economic viability. The methods and procedures were applied to flax fibre/bio-epoxy and flax fibre/petrochemical resin composite systems.

Initially, all raw materials sources, their respective specifications and performance capabilities were confirmed, e.g.: "apparent" fibre diameter, modulus, strength; fabric weave style, areal weight, tows/metre in both warp and weft, crimp angle; the surface treatment of the natural fibres; and resin characteristics.

Second, the fibre modification process parameters were controlled to achieve the best NFRP performance; i.e. parameters such as the chemical treatment, solvent concentration and exposure time.

Finally, the composite infusion parameters, fibre volume fraction and fibre alignment were controlled.

In summary, consistent processes were developed to achieve quality control (QC) of the materials received and the treatment and conditioning of the fibre, to ensure the best outcome in manufacturing the composite.

After all the fibres had undergone their respective treatments and the epoxy resin system had been modified, 36 different laminates were produced which form the basis of the experimental research conducted in this project. The experiments involved a series of treatment procedures which included matrix modification, fibre mercerisation and silanisation, and most importantly the mechanical testing of the laminate composites. These procedures are discussed in the following sections of this chapter and are summarised in (Table 7.15).

The mechanical testing of the composites was strictly aligned to the standards relating to the calibration and procedures attendant to the equipment so that the most reliable results were obtained from the mechanical tests performed. The variability of the test results undertaken on any composite material is a significant factor, a problem that is magnified in NFRP composites because of the great variation in the morphology and properties of NF. Mechanical test reliability is desired in the current PhD. Appendix F presents the test uncertainty determination process.

After the experimental study had been undertaken in the laboratory, it paved the way for the real prototype to be designed and ultimately manufactured.

This research is focused on the enhancement of NFRP interfacial properties through modification of the polymer matrix and NF surface with chemical treatments. The study was divided into different sections to fulfil the research. First, the base mechanical properties of the flax fibre/epoxy were obtained; **Laminates 1-4 and 29**, used Biotex/SuperSap bio-resin matrix in Laminates 1-2, Biotex/Huntsman in laminates 3 and 29, Lineo/Huntsman in Laminate 4. Second, the most appropriate modification methods for the specific composite were identified; **Laminates 5-29** used Huntsman petrochemical resin reinforced with Biotex fabric. Third, best modification method was applied to a bio-composite; **Laminates 30-36** used SuperSap bio-resin reinforced with Biotex fabric. Fourth, the theoretical knowledge as researched was applied to producing various bio-composite types, modifying and testing their mechanical performance under moisture conditions; **Laminates 33-34** used SuperSap bio-resin reinforced with Biotex fabric.

7.2. RAW MATERIAL

The NF, bio-resin, petrochemical resin and chemical treatment characteristics as selected are described below.

7.2.1. Flax fibre characteristics

Two different flax fibre fabric suppliers were selected, namely Composite Evolution (CE) from Chesterfield - England and Lineo (NV) from St Martin du Tilleul - France.

7.2.1.1. Composite Evolution flax fibre

CE is dedicated to the production of flax and jute yarns and fabrics under the trade name Biotex ("Natural Fibres Reinforcements", 2017). For this study, unidirectional (0° UD) 275 gsm flax fabric was supplied (See Figures 7.1 and 7.2). CE fibre properties are shown in Table 7.1 and yarn properties are shown in Table 7.2.

Table 7.1: Mechanical properties of Biotex fibre (data from CE with no statistical information)

| Flax fibre average properties | | |
|-------------------------------|------------------------|--|
| Density | 1500 kg/m ³ | |
| Diameter** | 20 µm | |
| Tensile modulus | 50 GPa | |
| Tensile strength | 500 MPa | |
| Strain at failure | 2% | |

** The diameter was very roughly estimated by CE**

Table 7.2: Biotex flax fibre yarn properties (data from CE)

| Flax yarn properties | 250 tex | 1000 tex |
|---------------------------|---------|----------|
| Tenacity/tensile strength | 15N/tex | 38N/tex |
| Elongation at break | 15.5% | 4.4% |

CE also supplied data on the different tests performed on composites produced with their fabrics. The data provided one reference point for comparison with the data obtained in the present research project.



Figure 7.1: Tested CE flax fibre UD roll



Figure 7.2: Flax-ply woven and UD fabric (Courtesy of Easy Composites)

Table 7.3 presents company data for laminates made by CE from 30-33 v/o Biotex flax fabrics and unsaturated polyester resin manufactured by the vacuum infusion process. All composites were made from 250 tex Biotex flax yarn fabrics and tested at room temperature.

| Table 7.5. Nechanical property data for composites made nom blotex ob fabrics (data nom CL) | | | | | |
|---------------------------------------------------------------------------------------------|-----------------------|------------------------|------------------------|---------------|--|
| Property | UD flax-polyester | Biaxial flax-polyester | Woven flax-polyester | Test standard | |
| Density | 1.3 g/cm ³ | 1.3 g/cm ³ | 1.24 g/cm ³ | | |
| Tensile modulus | 18.8 GPa | 8.7 GPa | 7.2 GPa | ISO 527-4 | |
| Tensile strength | 174 MPa | 85 MPa | 68.3 MPa | ISO 527-4 | |
| Tensile elongation | 1.5% | 1.7% | 2.5% | ISO 527-4 | |
| Flexural modulus | 15.1 GPa | 6.8 GPa | 4.0 GPa | ISO 14125 | |
| Flexural strength | 196 MPa | 135 MPa | 97.4 MPa | ISO 14125 | |
| Charpy impact | TBC | TBC | 28.0 kJ/m ² | ISO 179 – 1 | |

Table 7.3: Mechanical property data for composites made from Biotex UD fabrics (data from CE)

The technical data sheet (TDS) is included in Appendix A1.

7.2.1.2. Lineo flax fibre

Lineo ("Ecotechnilin", 2017) is a dedicated company producing flax and jute yarns and fabrics under the trade name Flax-ply (Figure 7.3). For this study, unidirectional (0° UD) 150 gsm flax fabric was supplied.



Figure 7.3: Flax-ply woven and UD fabric

Table 7.4 presents mechanical test results supplied by Lineo. The results are obtained for 12 layers of a 180 gsm Flax-ply UD panel. The V_f is of 60%. The selected polymer matrix was not identified in the supplier information.

| Property | UD Flax-ply | Test standard |
|---------------------|------------------------|---------------|
| Density | 1.33 g/cm ³ | |
| Tensile modulus | 35 GPa | ISO 527-4 |
| Tensile strength | 330 MPa | ISO 527-4 |
| Tensile elongation | 1.8 % | ISO 527-4 |
| Flexural modulus | 22 GPa | ISO 14125 |
| Flexural strength | 300 MPa | ISO 14125 |
| Flexural Elongation | 2.4 % | ISO 14125 |

Table 7.4: Mechanical property data for composites made from Flax-ply UD fabric (data from Lineo)

The technical data sheet (TDS) is included in Appendix A2.

7.2.2. Bio-epoxy resin

The initial bio-epoxy matrix system selected was from Entropy Resins, a company established in California, USA. In 2012, Ferrer Dalmau, a Catalonian company, started production in Europe in collaboration with the parent USA Company, and this greatly facilitated the research project making the raw material more freely available.

Entropy ("Entropy Resins", 2017) offers a wide variety of formulations which were used as follows:

| • | Supersap CLR/INF or Supersap CLR/(CLX-INH) | Infusion panels |
|---|----------------------------------------------------|-----------------|
| • | Supersap CPM/(CPF-CPL) | RTM fins |
| • | SuperSap ONE/ONF or SuperSap BRT/(CLX – CLF – CLS) | Surfboards |

Supersap CLR epoxy is a clear modified liquid bio-epoxy resin. In contrast to traditional petroleumbased resins, this formulation contains bio-renewable materials either (a) sourced as co-products from other operations, or (b) from waste streams of other industrial processes, such as wood pulp and bio-fuels production.

Supersap CLR is water-clear and UV-stabilised to avoid the yellowing of the product. It is claimed to be ideal for outdoor applications such as surfboard production. The low viscosity affords the option of producing pieces in RTM or infusion. The resin can cure at RT, or be heated in the oven. Normally post-cure is recommended.

The wettability of natural fibres is one of the most important considerations in obtaining the highest quality mechanical properties in composites. A low viscosity resin is essential for application to RTM and RIFT processes. In RTM, the resin flows long distances in comparison with other processing techniques. Rudd, et al. (Rudd, Long, Kendall & Mangin, 1997) stated that viscosity is the most significant practical limitation in the production of a suitable resin system. Low viscosity indicates a high flow rate, whilst a high viscosity indicates the opposite low flow rate. Resins with extreme low viscosity may be unsuitable for LCM processes since they may lead to high porosity or gross voidage, i.e. unwetted volumes. Resins formulated for liquid composite moulding processes typically have an initial viscosity of around 200 mPa.s (1 mPa.s = 1 centipoise). Becker (Becker, 1991), quotes an upper limit of 800 mPa.s for viscosity in RTM. The non-injection point (NIP) is defined as a viscosity of 1000 mPa.s (Pearce, Guild & Summerscales, 1998). At this viscosity level and under the low pressures applied in infusion processes the flow front is effectively stationary. For this reason,

Supersap CLR/ (CLX-INH) formulation is the most appropriate system because of its low viscosity, although Supersap CLR/INF may also be an option with its higher viscosity (See Table 7.6).

Table 7.5 illustrates the properties of the Supersap CLR resin system (Figure 7.4); Table 7.6 presents the typical working properties that are available during manufacture with Supersap CLR resins; Table 7.7 presents the mechanical properties for cured resin systems.



Figure 7.4: Supersap CLR/CLX system used for the panel production (Courtesy of 5TX surf)

| CERTCOIL System properties (End op) | y Resill uutu j |
|-------------------------------------|-----------------------|
| Property | Supersap CLR |
| Visual appearance | White to light yellow |
| Gardener colour | 1-2 |
| Viscosity (cPs @ 25°C) | 2000-4000 |
| Density (SG at 25°C: water = 1) | 1.17 |
| Bio-carbon content by mass (%) | 18.2-25.4 |
| Bio-content by mass (%) | 30.8-45.2 |

 Table 7.5: Supersap CLR resin system properties (Entropy Resin data)

Table 7.6: Supersap CLR resins manufacturing data (Entropy Resin data)

| Property | Supersap CLR/INF | Supersap CLR/ (CLX-INH) |
|-------------------------------|------------------|-------------------------|
| Mix ratio by weight | 100:33 | 100:19:19 |
| Mix viscosity (cPs @ 25°C) | 500-1000 | 360 |
| System biocontent by mass | 21-30% | 17% |
| Gel-time (min, 150 g at 25°C) | 45 | 75 |
| Thin film set time (h @ 25°C) | 4 | |
| Tack free time (h @ 25°C) | 8 | 8-10 |
| Cure cycle (25°C) before | 7-10 days | 7-10 |
| post-cure | 2h @ 80°C | |

| Table 717 Theenamed properties for eared supersup resin systems (Endopy Resin add) | | | | |
|------------------------------------------------------------------------------------|--------------------------------|--------------------------|--|--|
| Property | Supersap CLR/INF/infusion fast | Supersap CLR/CLX/INH | | |
| | hardener | | | |
| Cure cycle | 24h @ 25°C the 2h @ 48°C | 24h @ 25°C the 2h @ 48°C | | |
| Tg (°C) | 115 | 56 | | |
| Tensile modulus (GPa) | 4.27 | 3.29 | | |
| Tensile strength (MPa) | 69 | 58 | | |
| Elongation at break (%) | 2 | 2.5 | | |
| Flexural modulus (GPa) | 3.79 | 3.16 | | |
| Flexural strength (MPa) | 110 | 99 | | |
| Compression strength (MPa) | - | 92 | | |

Table 7.7: Mechanical properties for cured Supersap resin systems (Entropy Resin data)

The values in Tables 7.5-7 were used as reference points for comparison with values obtained from the literature review and in the present research project. They were also used to inform the calculations within the different models.

The technical data sheet (TDS) is appended in Appendix A3.

7.2.3. Petrochemical resin

The initial experiments for NFRP production were undertaken with petrochemical epoxy resin from Huntsman LLC. (USA). Araldite LY 1569 CH/Aradur 3489 CH epoxy infusion system was selected for the production of the laminate ("Huntsman LLC", 2017).

| Resin | Araldite LY 1569 CH | |
|----------------------------------------------|------------------------------------|--|
| Visual appearance | Clear liquid | |
| Viscosity (ISO 12058 cPs at 25°C) | 1300-1500 | |
| Density (ISO 1675 g/cm ³ at 25°C) | 1.1-1.2 | |
| Hardener | Aradur 3489 CH | |
| Visual appearance | Clear liquid | |
| Viscosity (ISO 12058 cPs at 25°C) | 5-20 | |
| Density (ISO 1675 g/cm ³ at 25°C) | 0.92-0.93 | |
| Mixture (100:28 by weight) | Araldite LY 1569 CH/Aradur 3489 CH | |
| Viscosity (ISO 12058 cPs at 25°C) | 200-300 | |

Table 7.8: Araldite LY 1569 CH/Aradur 3489 CH system properties (Huntsman LLC data)

Table 7.9: Typical working properties available during manufacture with Araldite LY 1569 CH/Aradur 3489 CH (Huntsman LLC data)

| Property | Araldite LY 1569 CH/Aradur 3489 CH |
|-------------------------------------|------------------------------------|
| Mix ratio by weight | 100:28 |
| Mix viscosity (cPs @ 25°C) | 200-300 |
| Gel-time (min, at 80°C/100°C/120°C) | 43-46/15-16/7-8 |
| Tg (°C, 4h 80°C/8h 80°C) | 76-79/77-80 |

Table 7.10: Mechanical properties for cured Araldite LY 1569 CH/Aradur 3489 CH resin systems (Huntsman LLC data)

| Property | Araldite LY 1569 CH/Aradur 3489 CH |
|-------------------------|------------------------------------|
| Cure cycle | 8h 80°C |
| Tg (°C) | 77-80 |
| Tensile modulus (GPa) | 2.85-3 |
| Tensile strength (MPa) | 67-71 |
| Elongation at break (%) | 4.4-5 |
| Flexural modulus (GPa) | 2.91-3 |
| Flexural strength (MPa) | 120-130 |
| Elongation at break (%) | 5.5-6.5 |

The values in Tables 7.8-10 were used as reference points for comparison with values obtained from the literature review and in the present research project. They were also used to inform the calculations applied to the different models.

The technical data sheet (TDS) is presented in Appendix A4.

7.2.4. Chemical products

7.2.4.1. Sodium hydroxide (NaOH)

CAS Number: 1310-73-2

Molecular Weight: 40

Form: granules

Supplier: Sigma-Aldrich ("Sigma-Aldrich NaOH", 2017)
7.2.4.2. 3-(trimethoxysilyl) propylamine (BYK-C 8001)

CAS Number: 82985-35-1

Molecular Weight: 179.29 (C₆H₁₇NO₃Si)

Methanol

Form: viscous liquid

Supplier/Code: BYK-Chemie GmbH / BYK-C 8001 Coupling Agent for Epoxy Composites (BYK C-8001)

CAS-No. Classification Classification Concentratio EC-No. (REGULATION **Chemical name** (67/548/EEC n Registration (EC) No (%)) Number 1272/2008) 13822-56-5 Xi; R38 Skin Irrit. 2; H315 237-511-5 3-(trimethoxysilyl)propylamine >= 50 - < 100 Xi; R41 Eye Dam. 1; H318 01-2119510159-45 Flam. Liq. 2; H225 67-56-1

200-659-6

01-

2119433307-44

Table 7.11: Information from BYK-C 8001 Safety Data Sheet (MSDS)



F; R11

T; R23/24/25-

R39/23/24/25

Acute Tox. 3; H331

Acute Tox. 3; H311

Acute Tox. 3; H301

STOT SE 1; H370

>= 1 - < 3

Figure 7.5: 3-(trimethoxysilyl) propylamine structure

7.2.4.3. Ethanol (Ethyl alcohol CH₃CH₂OH)

CAS Number: 64-17-5

Molecular Weight: 46.07

Form: thin clear liquid

Supplier: Sigma-Aldrich ("Sigma-Aldrich Ethanol", 2017)

7.2.5. Infusion consumables ("Easy Composites", 2017)
Vacuum bag: AeroFilm® VB160
Peel-ply: AeroFilm® PP230
Flow mesh: AeroFilm® FM105
Pipes: 1/2" I.D. Wire Reinforced Vacuum Hose
Spiral: Resin Infusion Spiral Medium Flow 10m Coil
VAP membrane: AB production material (commercially sensitive data not released for this thesis)
Tacky tape: Vacuum Bagging Gum Sealant Tape 15m

7.3. TREATMENT PROCEDURE

Experimental procedures for modifying the matrix, and the chemical treatment for the NF were selected from the options in the literature review in Chapter 4. The most appropriate method was selected for each procedure with emphasis upon the economic viability, the operational validity and the effectiveness of each selection.

In the Laminates from 1 to 29, for each fibre/matrix combination, one laminate was produced and two sets of samples are extracted: (i) for the evaluation of longitudinal tensile properties; and (ii) for the evaluation of transverse tensile properties. Laminates 1, 2, 3, 4 and 29 were tested in order to get the base properties, **untreated systems mechanical properties**, in longitudinal and transverse directions. For the **modified systems**, from Laminate 5 to 28, the longitudinal test was used to check if the fibre had been damaged during the test, and the transverse test to check the evolution of the fibre/matrix interface.

After undertaking the following processes – treatment of the fibres, production of the laminate, and the cutting and testing of the samples - the data recorded was used to determine which treatments had the greatest value/effectiveness to undertake the research. Once this process was completed, the know-how information was applied to **Supersap CLR-INF**/ **Biotex UD composite system**; Laminates 30 to 36 were manufactured and tested to complete the characterisation process.

The laminate samples were investigated for the following procedures - mercerisation, matrix modification and silanisation

7.3.1. Mercerisation

In the research Project, a similar method to that proposed by Kim and Netravali (KN) [194] to increase the sisal fibre/soy resin in the composite system was applied (See Chapter 4 section 4.1). The KN sisal fibre was immersed in 2M NaOH solution for 2 hours at RT, washed with water, neutralised with acetic acid and finally dried at RT. This method was selected because of its low energy consumption, its effectiveness in the improvement of mechanical properties and its potential for a low pollutant product. In the KN study, in comparison with the reference control sample, the highest quality results were achieved when the fibre tension was at 50g/fibre, reaching strength of 35% and showing a modulus increase of 111%.

The method developed for treating the NF had the following major procedures:

- The fibres were immersed in five different solutions, **0.25M**, **0.5M**, **1M**, **1.5M** and **2M** solutions, to avoid damaging the mechanical performance of the fibres since it is implied in the literature that the application of a 2M solution might be damaging, although Kim and Netravali [194] have stated otherwise.
- An alternative option was to increase the concentration of the solution and reduce the immersion time; however, the intention was to proceed in the opposite direction, by extending the immersion time and maintaining or reducing the solution concentration. This procedure was undertaken to minimise/eliminate the production of pollutant substances. Based on this strategy the fibres were immersed for **2h**, **4h**, **12h** and **24h**; -the fibres were

dried for 24h at RT; the fibres were not washed with water, but neutralised with acetic acid and dried in the oven. The objective was to minimise both pollutant production and the consumption of energy.

Based on the procedures above, twenty (20) different options were defined for the mercerisation of the NF as presented in Table 7.12.

| Mercerisation | 1h | 3h | 12h | 24h |
|---------------|-------------|-------------|-------------|-------------|
| 0.25M | Laminate 5 | Laminate 6 | Laminate 7 | Laminate 8 |
| 0.5M | Laminate 9 | Laminate 10 | Laminate 11 | Laminate 12 |
| 1M | Laminate 13 | Laminate 14 | Laminate 15 | Laminate 16 |
| 1.5M | Laminate 17 | Laminate 18 | Laminate 19 | Laminate 20 |
| 2M | Laminate 21 | Laminate 22 | Laminate 23 | Laminate 24 |

Table 7.12: Different options in the Mercerisation process

7.3.2. Matrix modification

The matrix was modified with **1%** (in resin/hardener system weight) of 3-(trimethoxysilyl) propylamine. The silane was added directly to the epoxy system hardener and stirred; once it had been fully mixed it was added to the epoxy base resin. This procedure has been described as "Epoxy matrix modification" in Chapter 4, section 3.1.1. Of equal relevance the studies presented by Wang, et al. and Chruściel et al. suggested that the interfacial properties of the composite fibre/matrix might be improved by adding silanes to the petrochemical matrix for use with synthetic fibres. To evaluate this matrix modification method in this PhD, two laminates were manufactured; Laminate 25 and Laminate 26. From the Laminate 5-24 tests, the best mercerisation treatment was identified: **1M NaOH for 3h**, the results from **Laminate 14**. *Laminate 25* was manufactured with untreated flax fibre/ 1% silane modified epoxy matrix and *Laminate 26* was manufactured with 1M NaOH for 3h mercerised flax fibre/ 1% silane modified epoxy matrix (Table 7.13).

| Table 7.13: Matrix modification different obt | ions |
|-----------------------------------------------|------|
|-----------------------------------------------|------|

| Epoxy system | Untreated | 1M NaOH 3h |
|------------------|-------------|-------------|
| Modified (1%) | Laminate 25 | Laminate 26 |

7.3.3. Silanisation

The silanisation process was undertaken by combining the most relevant factors identified in the literature review (Chapter 4, section 5.2) with due reference to sustainable issues. The methodology adopted the following procedure – the NF laminates were immersed in a 1-5% silane solution in water/solvent, washed with water and dried in the oven for some hours at 80-100°C. However, the Van de Weyenberg [160] treatments were considered to be the most effective, the easiest to perform, economically viable, and were deemed to be the greenest; additionally, silanisation is performed once the NF has been mercerized. One change from the Van de Weyenberg treatment was that the drying was at RT for 48 h, thus reducing energy consumption.

The previously activated, mercerised and aligned fibres were immersed in **1%** (in solution weight) 3-(trimethoxysilyl) propylamine solution (**50%/50% by volume** of **ethanol/water**) then **dried at RT for 48h**. Laminate 27 was produced with 1% silanised flax fibre and epoxy resin but was not mercerised; laminate 28 was produced with both mercerised (1M NaOH 3h) and silanised flax fibre and epoxy resin. See Table 7.14.

Table 7.14: Different options for the silanisation process

| Solution | Untreated | 1M NaOH 3h | | |
|--------------|-------------|-------------|--|--|
| 1% Silanised | Laminate 27 | Laminate 28 | | |

7.4. DEFINITION OF THE MANUFACTURING EQUIPMENT

7.4.1. Lamination table

Laminates 1 and 2 were produced at UniZar. The laminates were formed on a steel plate then cured in an oven preheated for 30min before introducing the laminates for cure at 80°C for 2h. In contrast, laminates 3 to 36 were produced in AB, and manufactured on an AMOND lamination table (Figure 7.6). The process was monitored with AB in-house software and cured on an electrically heated mould tool (Figure 7.7).



Figure 7.6: Lamination table controlled by thermocouples



Figure 7.7: Control software - (Left) Lamination table (Right) computer interface display

7.4.2. Cutting machine

The AB Mutronic DIADISC 5200R was selected for ease of sample manufacturing (Figure 7.8).



Figure 7.8: DIADISC 5200R cutting machine

7.4.3. Universal testing machine

Two Universal machines were used as follows:

- Unizar: Universal testing machine Ibertest STIB 200 W, computer coupled, accuracy class 1 according to UNE-EN-ISO-7500-1:2006, load cell 200kN.
- AB: Universal testing machine Shimadzu AG-X PLUS 250kN, computer coupled, TRAPEZIUM X software, accuracy class 0.5 according to UNE-EN-ISO-7500-1:2006, load cell 100kN (Figure

7.9).



Figure 7.9: Shimadzu AG-X PLUS 250kN Universal testing machine

7.5. LAMINATE MANUFACTURE

Laminate manufacturing was divided in three different blocks, Laminates 1, 2, 3, 4 and 29 to acquire base properties, Laminates 5 to 28 for the best treatment selection, Laminates 30 to 36 for optimised system complete mechanical characterisation. Table 7.15 summarises all the manufactured laminates.

To obtain reference properties, Supersap CLR-INF/ Biotex UD composite system was completely characterised. For this aim two different laminates were manufactured: (i) laminate 1 for longitudinal tensile test (T0°), transverse tensile (T90°), 10° in-plane shear test (S10°); and (ii) laminate 2 for the longitudinal compression (C0°) and transverse compression (C90°) tests. The standards used for testing are given in §7.5.1 below.

The Laminate 3 samples were produced and tested at AB laboratory to obtain longitudinal and transverse reference properties for Huntsman/Biotex UD composite system. Additionally, Laminate 29 was manufactured using VAP membrane in the process, this was done to establish any effect of the membrane.

The Laminate 4 samples were produced and tested at AB laboratory to obtain longitudinal and transverse reference properties for Lineo/ Biotex UD composite system. Lineo commercially available fabric is already pre-treated with a patented treatment method to enhance the fibre/matrix interfacial properties.

For Laminates 5-28, the fibres were treated in the 5'TX" surf workshop and subsequent panel manufacturing and testing were undertaken at AB. Optimised system Laminates 30 to 36 were manufactured and tested in AB.

Note that Laminates 3-29 used petrochemical **Araldite LY 1569 CH/Aradur 3489 CH** Huntsman epoxy resin, because it is (i) more readily available, (ii) relatively inexpensive and (iii) the formulation manufacturing parameters are well-known.

Flax fabric wettability was modified by the treatments, so it proved necessary to modify the infusion strategy as the experimental campaign progressed.

All the experimental tests undertaken on the laminates in the project are summarised in Table 7.15.

| ID | Matrix | Fibre | Layers | | Treatment | t | Method | Where |
|----|------------------|--------|--------|--------------|--------------|--------------|-------------------|----------|
| | | | * | None | NaOH | Silane | | |
| 1 | Supersap CLR/INF | Biotex | 4 | ✓ | × | × | No | Unizar |
| 2 | Supersap CLR/INF | Biotex | 8 | ✓ | × | × | No | Unizar |
| 3 | LY 1569 / 3489 | Biotex | 4 | \checkmark | × | × | No | AB |
| 4 | LY 1569 / 3489 | Lineo | 8 | √ | × | × | No | AB |
| 5 | LY 1569 / 3489 | Biotex | 4 | × | ✓ | × | 0.25M/1h | 5'TX"/AB |
| 6 | LY 1569 / 3489 | Biotex | 4 | × | √ | × | 0.25M/3h | 5'TX"/AB |
| 7 | LY 1569 / 3489 | Biotex | 4 | × | √ | × | 0.25M/12h | 5'TX"/AB |
| 8 | LY 1569 / 3489 | Biotex | 4 | × | √ | × | 0.25M/24h | 5′TX″/AB |
| 9 | LY 1569 / 3489 | Biotex | 4 | × | \checkmark | × | 0.5M/1h | 5'TX"/AB |
| 10 | LY 1569 / 3489 | Biotex | 4 | × | ✓ | × | 0.5M/3h | 5'TX"/AB |
| 11 | LY 1569 / 3489 | Biotex | 4 | × | √ | × | 0.5M/12h | 5'TX"/AB |
| 12 | LY 1569 / 3489 | Biotex | 4 | × | √ | × | 0.5M/24h | 5′TX″/AB |
| 13 | LY 1569 / 3489 | Biotex | 4 | × | √ | × | 1M/1h | 5′TX″/AB |
| 14 | LY 1569 / 3489 | Biotex | 4 | × | √ | × | 1M/3h | 5'TX"/AB |
| 15 | LY 1569 / 3489 | Biotex | 4 | × | ~ | × | 1M/12h | 5′TX″/AB |
| 16 | LY 1569 / 3489 | Biotex | 4 | × | ✓ | × | 1M/24h | 5′TX″/AB |
| 17 | LY 1569 / 3489 | Biotex | 4 | × | ✓ | × | 1.5M/1h | 5′TX″/AB |
| 18 | LY 1569 / 3489 | Biotex | 4 | × | ~ | × | 1.5M/3h | 5′TX″/AB |
| 19 | LY 1569 / 3489 | Biotex | 4 | × | ✓ | × | 1.5M/12h | 5′TX″/AB |
| 20 | LY 1569 / 3489 | Biotex | 4 | × | ✓ | × | 1.5M/24h | 5′TX″/AB |
| 21 | LY 1569 / 3489 | Biotex | 4 | × | ✓ | × | 2M/1h | 5′TX″/AB |
| 22 | LY 1569 / 3489 | Biotex | 4 | × | ~ | × | 2M/3h | 5′TX″/AB |
| 23 | LY 1569 / 3489 | Biotex | 4 | × | ✓ | × | 2M/12h | 5′TX″/AB |
| 24 | LY 1569 / 3489 | Biotex | 4 | × | ✓ | × | 2M/24h | 5′TX"/AB |
| 25 | LY 1569 / 3489 | Biotex | 4 | × | × | √ | 1% hardener | AB |
| 26 | LY 1569 / 3489 | Biotex | 4 | × | √ | ~ | 1M/3h+1% hardener | 5′TX″/AB |
| 27 | LY 1569 / 3489 | Biotex | 4 | × | × | √ | 1% fibre | AB |
| 28 | LY 1569 / 3489 | Biotex | 4 | × | √ | \checkmark | 1M/3h+1% fibre | 5′TX″/AB |
| 29 | LY 1569 / 3489 | Biotex | 4 | ✓ | × | × | No | AB |
| 30 | Supersap CLR/INF | Biotex | 8 | ✓ | × | × | No | AB |
| 31 | Supersap CLR/INF | Biotex | 4 | × | × | ✓ | 1% hardener | AB |
| 32 | Supersap CLR/INF | Biotex | 8 | × | × | ✓ | 1% hardener | AB |
| 33 | Supersap CLR/INF | Biotex | 4 | × | × | \checkmark | 1% hardener | AB |
| 34 | Supersap CLR/INF | Biotex | 4 | ✓ | × | × | No | AB |
| 35 | Supersap CLR/INF | Biotex | 4 | × | × | ✓ | 1% hardener | AB |
| 36 | Supersap CLR/INF | Biotex | 4 | \checkmark | × | × | No | AB |

Table 7.15: List of UD Laminate manufacturing tests undertaken with Biotex or Lineo flax fibra and Supersap CLR/INF or Araldite LY 1569 /Aradur 3489 epoxy matrix

7.5.1 Laminates 1 and 2

The manufacturing parameters of the UniZar reference laminates 1 and 2 are summarised in Table 7.16. The infusion process is illustrated in Figure 7.10 and the lamination scheme for the two laminates is shown in Figure 7.11.

Table 7.16: Manufacturing characteristics of laminate 1 and 2

| | Laminate 1 | Laminate 2 |
|-----------------------------|---------------------|--------------------|
| Test samples | T0°, T90°, Sh10° | C0°, C90° |
| Reinforcement | 4 layers of fabric | 8 layers of fabric |
| Fabric weight (gsm) | 275 | 275 |
| Fabric dimensions | 450mm x 330 mm | 450mm x 330 mm |
| Mass of fabric (g) | 163.35 | 326.70 |
| Resin used (g) | 177 | 354.7 |
| Resin preheat | 40°C for 30 min | 40°C for 30 min |
| Infusion vacuum (atm gauge) | 0.9 | 0.9 |
| Infusion time (minutes) | 17 | 23 |
| Cure cycle | 5min RT + 2h @ 80°C | 2h @ 80°C |



Figure 7.10: Manufacturing process for laminates 1 and 2



Figure 7.11: Lamination scheme for laminates 1-2

7.5.2. Laminates 3 and 4

The manufacturing parameters of the AB reference laminates, 3 Biotex and 4 Lineo, are summarised in Table 7.17. Both laminates were infused in one shot under the same infusion strategy as illustrated in Figure 7.12. Figure 7.13 shows the lamination scheme for laminates 3 and 4.

| | Laminate 3 | Laminate 4 | |
|-----------------------------|------------------------------|----------------------------|--|
| Test samples | T0°, T90° | T0°, T90° | |
| Reinforcement | 4 layers of fabric | 8 layers of fabric | |
| Fabric weight (gsm) | 275 | 150 | |
| Fabric dimensions | 600mm x 330 mm | 600mm x 330 mm | |
| Mass of fabric (g) | 218 | 120 | |
| Resin used (g) | 1000 | 1000 | |
| Infusion temperature (°C) | 40 | 40°C for 30 min | |
| Infusion vacuum (atm gauge) | 0.93 | 0.93 | |
| Infusion time (minutes) | 12 | 12 | |
| Cure cycle | 2h @ 65°C + 1h @ 80°C + 1h @ | 2h @ 65°C + 1h @ 80°C + 1h | |
| | 100°C | @ 100°C | |
| Tg (°C) | 65 | 65 | |

Table 7.17: Manufacturing characteristics of laminate 3-4



Figure 7.12: manufacturing process of laminates 3 and 4



Figure 7.13: Lamination scheme for laminates 3 and 4

7.5.3. Laminate 5 to 20

Once the base properties were clear, a different mercerisation treatments were applied to the Biotex/Huntsman composites system. Four laminates were manufactured at one shot as it is shown in Figure 7.14. The laminates manufacturing parameters are summarised in Table 7.18. Figure 7.15 shows the scheme for each of laminate 5 to 20 characteristics.

Table 7.18: Manufacturing characteristics of laminate 5 to 20

| | Laminate 5 to 20 |
|-----------------------------|------------------------------------|
| Test samples | T0°, T90° |
| Reinforcement | 4 layers of fabric |
| Fabric weight (gsm) | 275 |
| Fabric dimensions | 600mm x 330 mm |
| Mass of fabric (g) | 218 |
| Resin used (g) | 1000 |
| Resin preheat | 40°C for 30 min |
| Infusion vacuum (atm gauge) | 0.93-5 |
| Infusion time (minutes) | 12-52 |
| Cure cycle | 2h @ 65°C + 1h @ 80°C + 1h @ 100°C |
| Tg (°C) | 65 |



Figure 7.14: Manufacturing process for laminates 5 to 8



Figure 7.15: Lamination scheme for laminates 5 to 20

7.5.4. Laminate 21 to 24

Once the base properties were clear, different mercerisation treatments were applied to Biotex/Huntsman composites system. Four laminates were manufactured at one shot (Figure 7.16). On this occasion, flow media was placed covering all the laminate surface and VAP membrane introduced to the laminate scheme (Figure 7.17). The infusion strategy was modified since the mercerisation treatment provoked fibre swelling (as previously observed by other researchers [Francucci 13, Masoodi 14 and Nguyen 15]) with the consequent flow reduction in the infusion process. The laminate manufacturing parameters are summarised in Table 7.19. Figure 7.17 shows the scheme for a single laminate with 21 to 24 laminates characteristics.

| | Laminate 21 to 24 | |
|-----------------------------|------------------------------------|--|
| Test samples | T0°, T90° | |
| Reinforcement | 4 layers of fabric | |
| Fabric weight (gsm) | 275 | |
| Fabric dimensions | 600mm x 330 mm | |
| Mass of fabric (g) | 218 | |
| Resin used (g) | 1000 | |
| Resin preheat | 40°C for 30 min | |
| Infusion vacuum (atm gauge) | 0.93-5 | |
| Infusion time (minutes) | - | |
| Cure cycle | 2h @ 65°C + 1h @ 80°C + 1h @ 100°C | |
| Tg (°C) | 65 | |

Table 7.19: Manufacturing characteristics of laminates 21 to 24



Figure 7.16: Manufacturing process for laminates 21-24



Figure 7.17: Lamination scheme for laminates 21-24

For laminates 25-36, the flow medium was placed covering all the laminate surface and VAP membrane introduced to the laminate scheme.

7.5.5. Laminate 25 to 28

Once the best mercerisation treatment was selected, silanisation process was performed on the Biotex flax UD fabric. For laminates 25-26, the resin system was modified. For laminates 27-28, the fibre was modified. For all of these cases, the manufacturing characteristics (Table 7.19) and lamination scheme (Figure 7.17) were common. On this occasion, two laminates were manufactured at one shot (Figure 7.18).



Figure 7.18: manufacturing process of laminates 27 and 28

7.5.6. Laminate 29

Laminate 29 was manufactured to determine whether the mechanical properties differ from Laminate 3 properties when the manufacturing conditions were modified; both were manufactured with Biotex/Huntsman combination. The manufacturing characteristics were described in Table 7.19 and lamination scheme in Figure 7.17.

7.5.7. Laminate 30 and 32

Laminates 30 and 32 were manufactured to evaluate the silane addition to the matrix in compression and interlaminar shear properties. Both laminates were manufactured with Biotex/SuperSap composite system, and in the Laminate 32 production 1% silane was added to the matrix. The manufacturing characteristics were described in Table 7.19 and lamination scheme in Figure 7.17.

7.5.8. Laminate 31

Laminate 31 was manufactured in order to get optimised system mechanical properties. The samples were tested using strain gauges in order to get the more accurate mechanical properties. The manufacturing characteristics were described in Table 7.19 and lamination scheme in Figure 7.17.

7.5.9. Laminate 33 and 34

Laminates 33 and 34 were manufactured in order to evaluate the silane addition to the matrix in flexural properties. Both laminates were manufactured with Biotex/SuperSap composite system, and

in the Laminate 33 production 1% silane was added to the matrix. The manufacturing characteristics were described in Table 7.19 and lamination scheme in Figure 7.17. In this occasion two laminates were manufactured at one shot (Figure 7.19).



Figure 7.19: manufacturing process of laminates 33 and 34

7.5.10. Laminate 35 and 36

Laminates 35 and 36 were manufactured in order to evaluate the silane addition to the matrix in inplane shear properties. Both laminates were manufactured with Biotex/SuperSap composite system, and in the Laminate 35 production 1% silane was added to the matrix. The manufacturing characteristics were described in Table 7.19 and lamination scheme in Figure 7.17. In this occasion two laminates were manufactured at one shot (Figure 7.20).



Figure 7.20: manufacturing process of laminates 35 and 36

7.6. TEST

In the research process, between six and ten samples were tested for each laminate in every specified type of test. The procedures were defined by international (ISO) standards indicated below.

7.6.1. Test standards

The following standard tests were conducted:

- Tension (ISO527-5 for unidirectional or ISO527-4 for multiaxial composites)
- Compression (ISO 14126:1999)
- In-plane shear 10° off-axis
- In-plane shear ±45° (BS EN ISO 14129:1998) for unidirectional composites
- ILSS (ASTM D2344)
- Flexural test (ASTM D 790-03)

In all the tests performed, the thickness measurement was the arithmetic mean from three different positions. Test data for the standards included - stress-strain curves, maximum load, Young's modulus, stress at break (strength), and strain at break.

7.6.1.1. Tensile test

For the **ISO 527-4/-5** tensile test standard, longitudinal and transverse tests were performed on the samples as shown in Table 7.20.

| | ISO 527-4/-5 longitudinal tensile test conditions | ISO 527-4/-5 Transverse tensile test conditions |
|--------------------------|------------------------------------------------------|----------------------------------------------------|
| Number of specimens | 10 | 10 |
| Specimen dimensions | 250mm x 15mm x a(mm) | 250mm x 25mm x a(mm) |
| Tabs type | Double bonded GF/epoxy ±45° | Double bonded GF/epoxy ±45° |
| | tabs | tabs |
| Tabs dimensions | 50mm x 15mm x a(mm) | 50mm x 25mm x a(mm) |
| Free length between tabs | 150mm | 150mm |
| Test speed | 2mm/min | 2mm/min |
| Strain measurement | Video extensometers | Video extensometers |
| Test climate | ISO291 class 2 | ISO291 class 2 |

Table 7.20: Longitudinal tensile test conditions

7.6.1.2. Compression test

For the **ISO 14126** compression test standard, longitudinal and transverse tests were performed on the samples as shown in Table 7.21:

| | ISO 14126 longitudinal compression test conditions | ISO 14126 Transverse compression test conditions |
|--------------------------|----------------------------------------------------------|--------------------------------------------------------|
| Number of specimens | 10 | 10 |
| Specimen dimensions | 110mm x 10mm x a(mm) | 110mm x 25mm x a(mm) |
| Tabs type | Double bonded GF/epoxy ±45° tabs | Double bonded GF/epoxy ±45° tabs |
| Tabs dimensions | 50mm x 10mm x a(mm) | 50mm x 25mm x a(mm) |
| Free length between tabs | 10mm | 10mm |
| Test speed | 2mm/min | 2mm/min |
| Strain measurement | Strain Gauges | Strain Gauges |
| Test climate | ISO291 class 2 | ISO291 class 2 |

Table 7.21: Longitudinal compression test conditions



Figure 7.21: UniZar longitudinal and transverse compression test samples

7.6.1.3. In-plane shear test

For the in-plane shear properties, both 45° (**ISO 14129** standard) and 10° (no International standard) shear tests were adopted. Samples and test characteristics were identical except for the 45° or 10° angle between the reinforcement and the test axis (Table 7.22).

| ē | and 10° snear test conditions | | | | | |
|---------------------|--------------------------------------------------------|----------------------------------|--|--|--|--|
| | 45° (ISO 14129) and 10° in-plane shear test conditions | | | | | |
| Number of specimens | | 10 | | | | |
| | Specimen dimensions | 250mm x 25mm x a(mm) | | | | |
| | Tabs type | Double bonded GF/epoxy ±45° tabs | | | | |
| | Tabs dimensions | 50mm x 25mm x a(mm) | | | | |
| | Free length between tabs | 150mm | | | | |
| | Test speed | 2mm/min | | | | |
| | Strain measurement | Video extensometers | | | | |
| | Test climate | ISO291 class 2 | | | | |

Table 7.22: 45° and 10° shear test conditions



Figure 7.22: UniZar 10° shear test sample

7.6.1.4. Interlaminar shear test (ILSS)

The composite interlaminar shear properties were tested under **ASTM D2344** standard, "*Short-Beam Strength of Polymer Matrix Composite Materials and Their Laminates*". The samples were tested in three-point bending with a short span to promote interlaminar delamination rather than invalid tension, compression or flexure failure modes. Tests conditions are shown in Table 7.23 and test example in Figure 7.23.

Table 7.23: Short-beam test conditions

| | ASTM D2344 short-beam test conditions |
|---------------------|---------------------------------------|
| Number of specimens | 10 |
| Specimen dimensions | 6 a x 2 a x a (thickness, mm) |
| Tabs type | No tabs |
| Span | 4 a |
| Test speed | 1mm/min |
| Test climate | ASTM D 5229 |



Figure 7.23: Short-beam test sample

7.6.1.5. Three-points bending test

The composite flexural properties were tested under **UN EN ISO14125** standard, "*Fibre-reinforced plastic composites -- Determination of flexural properties*". The samples were tested in three-point bending with a long span designed to get flexural failure. Tests conditions are shown in Table 7.24 and an example tested sample in Figure 7.24.

Table 7.24: Flexural test conditions

| UN EN ISO14125 flexural test conditions | | | | | | | |
|-----------------------------------------|----------------------|--|--|--|--|--|--|
| Number of specimens | 10 | | | | | | |
| Specimen dimensions | 100mm x 15mm x 2(mm) | | | | | | |
| Spam | 80mm | | | | | | |
| Tabs type | No tabs | | | | | | |
| Test speed | 2mm/min | | | | | | |
| Strain measurement | Bridge movement | | | | | | |
| Test climate | ISO291 class 2 | | | | | | |



Figure 7.24: Short-beam test sample

7.7. TEST PROCEDURE

Tests performed by AB were undertaken under DNV-GL accredited procedures. The reliability study is attached in Appendix F to show such tests high quality.

7.8. TEST DEVIATIONS

There were three major inconveniences during the testing campaign:

First, in the cutting process the diamond saw blade was burning the surface of the flax/ epoxy composite panels. The cut was precise and clean but friction from the saw was igniting the dust released in the process. To overcome this problem hard steel saw blades were substituted for the diamond blades and they produced equally precise cuts, without burning (Figure 7.25).



Figure 7.25: Diamond (left) and hard steel (right) saw blades

Second, the material supply times were unreliable. A further constraint was that CE stopped the production of the Biotex UD275 format fabric. This was substituted for a new Biotex UD150 product. In the construction of the Biotex UD275 fabric the longitudinal flax rovings are wrapped by a polyester yarn in order to provide stability to the fabric (Figure 7.26). The wrapping reduces the mechanical properties of the fabric. For this reason, CE has commercialised the Biotex UD150 alternative where the rovings are kept aligned with a binder. Any future work would be limited to use the new Biotex UD150 or any sensible alternative



Figure 7.26: Biotex UD275 fabric microscopy image

Third, 2M NaOH produced flax fibre swelling impeding the complete panel infusion. A new trial infusion strategy incorporating VAP was devised to produce laminates 21 to 24 with minimal defects.

7.9. SUMMARY

In Chapter 7 the materials and the equipment used to undertake the research project have been stated and described. The definition of the flax fibre treatment procedures has been reviewed in detail and the selection of the most appropriate procedures addressed based upon the appraisal conducted in Chapter 4.

8. Results

8.1. INTRODUCTION

Chapter 8 presents the data obtained from the experimental campaign.

8.2. MECHANICAL TESTS

Table 8.1 summarises the treatment of the respective laminates. The mechanical testing data is contained in Appendix B, and the page number as indicated in the final column of the Table 8.1. The experimental campaign covers different mechanical tests and all of them give as a result stress-strain curves; an example is shown in Figure 8.1.



Figure 8.1: Tensile 0° test example (Laminate 29)

During the experimental campaign there are some deviations from the expected data from mechanical tests. In the following point those deviations would be clarified.

- For transverse tensile test usually some noise is always detected in the deformation signal, this happens because the interfibre cracking; example Laminate 1 Tensile 90° test (Figure 8.2. left) or Laminate 25 Tensile 90° test
- 2. Depending on the software configuration, the negative strains might be plot as negative or positive; example **Laminate 2 Compression 0° test** (Figure 8.2. right)
- 3. Along all the experimental testing campaign it was seen that sometimes the deformation signal initially goes negative and next it changes to positive, e.g. **Laminate 3 Tensile 0° test** (Figure 8.3. left). This happen because of two factors, initial coupon angle and video-extensometry. The testing coupons have got initial curvature, that angle provokes initial negative deformation record by the extensometers. This phenomenon might be solved using more accurate data acquisition camera or testing with strain gauges. **Laminate 6 Tensile 90° test** (Figure 8.3. right) example is very exaggerated case, where the laminate initial bending provokes negative records
- Sometimes the video-extensionetry stickers release from the testing coupons, getting as a result strange records like in Laminate 11 Tensile 90° test, samples 6 and 8. (Figure 8.4. left)
- 5. When the camera lens is not correctly adjusted the signal line tends to be thicker, example **Laminate 21 Tensile 90° test** (Figure 8.4. right)

6. Some laminates have been tested with strain gauges in order to obtain more accurate modulus and strain values, example **Laminate 29 Tensile 0° test** (Figure 8.5)











Figure 8.4: 90° tensile test-Laminate 11 (left) and 90° tensile test-Laminate 21 (right)



Figure 8.5:0° tensile test measure with strain gauges-Laminate 29

| ID | Matrix | Fibre | Layers | Treatment | Method | Where | APPENDIX C |
|----|-------------------------------|-----------|--------|-------------|----------------------|----------|---------------|
| 1 | Supersap CLR/INF | Biotex UD | 4 | No | No | Unizar | Pages 190-191 |
| 2 | Supersap CLR/INF | Biotex UD | 8 | No | No | Unizar | Pages 192-193 |
| 3 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | No | No | AB | Pages 194-195 |
| 4 | Araldite LY 1569 /Aradur 3489 | Lineo UD | 8 | No | No | AB | Pages 196-197 |
| 5 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH | 0.25M/1h | 5′TX″/AB | Pages 198-199 |
| 6 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH | 0.25M/3h | 5′TX″/AB | Pages 200-201 |
| 7 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH | 0.25M/12h | 5'TX"/AB | Pages 202-203 |
| 8 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH | 0.25M/24h | 5'TX"/AB | Pages 204-205 |
| 9 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH | 0.5M/1h | 5′TX″/AB | Pages 206-207 |
| 10 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH | 0.5M/3h | 5'TX"/AB | Pages 208-209 |
| 11 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH | 0.5M/12h | 5′TX″/AB | Pages 210-211 |
| 12 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH | 0.5M/24h | 5′TX″/AB | Pages 212-213 |
| 13 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH | 1M/1h | 5′TX″/AB | Pages 214-215 |
| 14 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH | 1M/3h | 5′TX″/AB | Pages 216-217 |
| 15 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH | 1M/12h | 5′TX″/AB | Pages 218-219 |
| 16 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH | 1M/24h | 5'TX"/AB | Pages 220-221 |
| 17 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH | 1.5M/1h | 5′TX″/AB | Pages 222-223 |
| 18 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH | 1.5M/3h | 5′TX″/AB | Pages 224-225 |
| 19 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH | 1.5M/12h | 5'TX"/AB | Pages 226-227 |
| 20 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH | 1.5M/24h | 5′TX″/AB | Pages 228-229 |
| 21 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH | 2M/1h | 5′TX″/AB | Pages 230-231 |
| 22 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH | 2M/3h | 5′TX″/AB | Pages 232-233 |
| 23 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH | 2M/12h | 5′TX″/AB | Pages 234-235 |
| 24 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH | 2M/24h | 5′TX″/AB | Pages 236-237 |
| 25 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | Silane | 1% hardener | 5′TX″/AB | Pages 238-239 |
| 26 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | NaOH+Silane | 1M/3h+1% hardener | 5′TX″/AB | Pages 240-241 |
| 27 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | Silane | 1% fibre | 5′TX″/AB | Pages 242-243 |
| 28 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | Silane | 1M/3h+1% fibre | 5′TX″/AB | Pages 244-245 |
| 29 | Araldite LY 1569 /Aradur 3489 | Biotex UD | 4 | No | No | 5′TX″/AB | Pages 246-247 |
| 30 | SuperSap CLR/INF | Biotex UD | 8 | No | No | 5′TX"/AB | Pages 248-250 |
| 31 | SuperSap CLR/INF | Biotex UD | 4 | Silane | 1% hardener | 5′TX"/AB | Pages 251-252 |
| 32 | SuperSap CLR/INF | Biotex UD | 8 | Silane | 1% hardener | 5'TX"/AB | Pages 253-255 |
| 33 | SuperSap CLR/INF | Biotex UD | 4 | Silane | 1% hardener | 5′TX″/AB | Pages 255 |
| 34 | SuperSap CLR/INF | Biotex UD | 4 | No | No | 5′TX″/AB | Pages 255 |
| 35 | SuperSap CLR/INF | Biotex UD | 4 | Silane | 1% hardener | 5′TX″/AB | Pages 256 |
| 36 | SuperSap CLR/INF | Biotex UD | 4 | No | No | 5′TX″/AB | Pages 256 |

Table 8.1: Experimental campaign laminates numbering

8.3. FLOW TESTS

In the panels production process different flow properties were observed. Laminates infusion times have been monitored and the results are shown in Table 8.2 and Figure 8.6. In Figure 8.6. it is clearly appreciated a change in the fibre colour, suggesting a degradation of the flax apart from the swelling process.

| Laminates | Concentration (M) | Infusion t (min) |
|-----------|----------------------|---------------------|
| 5-8 | 0.25 | 14 |
| 9-12 | 0.50 | 13 |
| 13-16 | 1.00 | 12 |
| 17-20 | 1.50 | 52 |
| 21-24 | 2.00 | None |

Table 8.2: Laminates infusion times



Figure 8.6: Laminates 5-8 (left) and laminates 21-24 right

Additionally, three different fibres swelling were studied. Table 8.3 shows fibre diameter measurement results obtained for Untreated, 1M NaOH 3h treated and 2M NaOH 72h treated systems. In the Figure 8.7 technical fibre (> 40 μ m) diameter evolution with the mercerisation process was graph. X-axis lists the measurement points (Appendix C) and Y-axis gives the diameter values in mm.

Table 8.3: measured technical fibre diameters in mm (Appendix C)

| Number | Untreated | 1M NaOH 3h | 2M NaOH 72h |
|--------|-----------|------------|-------------|
| 1 | | | 0,69 |
| 2 | | | 0,71 |
| 3 | | | 0,9 |
| 4 | | | 0,92 |
| 5 | | | 0,94 |
| 6 | | | 1,04 |
| 7 | | 0,56 | 1,05 |
| 8 | 0,67 | 0,71 | 1,06 |
| 9 | 0,82 | 0,73 | 1,06 |
| 10 | 0,84 | 0,74 | 1,06 |
| 11 | 0,85 | 0,76 | 1,07 |
| 12 | 0,85 | 0,78 | 1,12 |
| 13 | 0,89 | 0,78 | 1,12 |
| 14 | 0,9 | 0,79 | 1,12 |
| 15 | 0,91 | 0,83 | 1,13 |
| 16 | 0,93 | 0,89 | 1,13 |
| 17 | 0,93 | 0,91 | 1,14 |
| 18 | 0,93 | 0,93 | 1,15 |
| 19 | 0,97 | 0,94 | 1,15 |
| 20 | 1,01 | 0,97 | 1,16 |
| 21 | 1,04 | 1 | 1,18 |
| 22 | 1,24 | 1,01 | 1,19 |
| 23 | | 1,04 | 1,21 |
| 24 | | 1,27 | 1,21 |
| 25 | | 1,28 | 1,22 |
| 26 | | | 1,22 |
| 27 | | | 1,23 |
| 28 | | | 1,24 |
| 29 | | | 1,27 |
| 30 | | | 1,4 |
| | | | |



Figure 8.7: Fibre diameter evolution with the mercerisation process

8.4. MOISTURE TESTS

Two different samples were immersed in tap water for a certain days in order to monitor those coupons moisture absorption percentage. The idea was to compare the water uptake percentage for 33 system with silane in the hardener and 34 base system. The sample aging was performed under ASTM D5229 standard. The resulting data after the flexural test was shown in Appendix E.

Chapter 9: Discussion

9.1. INTRODUCTION

The main issue when NFRP are manufactured is the poor adhesion between the NF and the polymer matrix. For that reason many methods have been proposed to improve the NFRP interfacial properties; and the most important methods were reviewed in Chapter 4. The review process indentified that mercerisation and silanisation were the most promising methods, given their (i) interfacial properties improvement, (ii) ease of implementation, (iii) sustainability, and (iv) cost effectiveness.

The following sections aim to prove that the selected methods enhance the selected flax/bio-resin composite system interfacial properties in a simple, sustainable and economical way; and thus address the key research question posed for this doctoral study.

Interfacial properties improvement is also believed to enhance the moisture resistance and fatigue properties. Moisture resistance tests were performed in the current study, but fatigue tests were outside the scope and timescale of the current study.

A series of 36 panels were manufactured as described in Chapter 7. Samples were extracted for mechanical testing as described in Chapter 8. The results of individual tests are presented in Appendix B. The panels were divided to have different characteristics:

- · 35 with Biotex and 1 with Lineo flax fabric
- · 26 with synthetic resin and 10 with bio-based resin
- 13 with untreated, 21 with mercerised fabric, 1 with silanised fabric and 1 with combined mercerisation/silanisation treatment
- 30 with untreated resin system and 6 with coupling agent in the resin hardener

Samples were extracted from the panels for laminate characterisation and mechanical testing, including:

Plate thickness, fibre volume and mass fraction for every panel

- Longitudinal tension (modulus, strengths, strains at break and Poisson's ratios)
- Transverse tension (modulus and strengths, strains at break and Poisson's ratios)
- · Longitudinal compression (modulus, strengths and strains at break)
- Transverse compression (modulus, strengths and strains at break)
- · In-plane shear test (modulus, strengths and strains at break)
- · ILSS (strength and displacement)
- Flexural tests (displacement, modulus and strength)

A total of 746 laminate thickness measurements, 73 volume fraction determinations, and 746 mechanical tests were conducted.

Additionally, ANOVA (Analysis of Variance) statistical method was applied to the mercerisation data in order to get clearer conclusion from the set of results. ANOVA is a method for testing the hypothesis that there is no difference between two or more population means. The ANOVA analysis process data is shown in Appendix G and the graphs obtained in the §9.2. MERCERISATION.

9.2. MERCERISATION

In this subsection it is studied whether mercerisation process increases flax/epoxy composite interfacial properties, as a part of the PhD research question. In order to make data strong and scientific analysis, ANOVA method was selected. ANOVA method was then applied to the mercerisation data. In the following subsections, ANOVA analysis was complemented with different properties plots for better understanding of the property evolution through the mercerisation process; two plots are obtained, first the property vs. immersion time and second the property vs. concentration. Each property was individually studied and conclusions were obtained in §**9.2.8**. "**Definitive mercerisation process selection**".

9.2.1. Longitudinal tensile modulus E1 ANOVA

In this particular case there are three hypotheses to prove. For the null hypothesis, it must be checked whether the mean values of all groups are equal under immersion time, concentration and combinative effect. For the case of the longitudinal modulus, the **ANOVA analysis confirms that there is significant effect due to immersion time, concentration or combination of both factors**; this was confirmed because after the data analysis the **p-value** is lower than the a = 0.05 for the 95% level of significance, see Table 9.1.

| / | | | | | |
|-------------------------|----------------|-----|---------------|---------|-------------------------|
| Source | SS | df | MS | Fo | p-value |
| Effect A:Immersion time | 40975430,507 | 3 | 13658476,836 | 16,076 | 0,000000002653552740 |
| Effect B:Concentration | 1126725565,783 | 4 | 281681391,446 | 331,538 | 0,000000000000000000000 |
| Effect AB: Combined | 110512596,819 | 12 | 9209383,068 | 10,839 | 0,000000000000000450 |
| Error | 152931675,247 | 180 | 849620,418 | | |
| Total | 1431145268,356 | 199 | | | |

Table 9.1. ANOVA analysis for the E1

The posterior Fisher's analysis says that when the box is red, the difference between mean values is significant, and when the box is unaltered the difference is not significant. The value is significant when the difference is higher than the calculated **LSD (Least Significant Difference)**, in this case **813.402**. See the Fisher's analysis in Tables 9.2 and 9.3.

| Table 9.2. Fisher ana | ysis of the | immersion | time | effect | in the | e Eı |
|-----------------------|-------------|-----------|------|--------|--------|------|
|-----------------------|-------------|-----------|------|--------|--------|------|

| 0.25M | Value | 0.5M | Value | 1M | Value | 1.5M | Value | 2M | Value |
|-----------------------|--------|---------------------|---------|-----------------|---------|---------------------|---------|-----------------|---------|
| 0.25M 1h - 0.25M 3h | 320,29 | 0.5M 1h - 0.5M 3h | 831,79 | 1M 1h - 1M 3h | 78,20 | 1.5M 1h - 1.5M 3h | 2316,65 | 2M 1h - 2M 3h | 1004,57 |
| 0.25M 1h - 0.25M 12h | 71,28 | 0.5M 1h - 0.5M 12h | 77,80 | 1M 1h - 1M 12h | 2303,38 | 1.5M 1h - 1.5M 12h | 2696,26 | 2M 1h - 2M 12h | 889,64 |
| 0.25M 1h - 0.25M 24h | 540,21 | 0.5M 1h - 0.5M 24h | 1516,87 | 1M 1h - 1M 24h | 1387,82 | 1.5M 1h - 1.5M 24h | 821,33 | 2M 1h - 2M 24h | 789,56 |
| 0.25M 3h - 0.25M 12h | 249,01 | 0.5M 3h - 0.5M 12h | 753,99 | 1M 3h - 1M 12h | 2225,18 | 1.5M 3h - 1.5M 12h | 379,61 | 2M 3h - 2M 12h | 114,93 |
| 0.25M 3h - 0.25M 24h | 860,50 | 0.5M 3h - 0.5M 24h | 685,09 | 1M 3h - 1M 24h | 1309,61 | 1.5M 3h - 1.5M 24h | 3137,98 | 2M 3h - 2M 24h | 215,01 |
| 0.25M 12h - 0.25M 24h | 611,49 | 0.5M 12h - 0.5M 24h | 1439,07 | 1M 12h - 1M 24h | 915,57 | 1.5M 12h - 1.5M 24h | 3517,59 | 2M 12h - 2M 24h | 100,08 |

Table 9.2 studies the immersion time effect in the E_1 values. It is initially observed that for the lowest concentration of 0.25M the mean values difference in the set of measurements it is not significant, except for the difference between 3h and 24h. With increasing concentration, the difference between the mean values becomes more significant; for 0.5M the difference is lower than for 1M and 1.5M cases. For the case of 2M concentration, the set of values looks to be homogeneous, except at short immersion times.

Figure 9.1 plots the E_1 variation vs. the immersion time. This graph supports Fisher's analysis in the paragraph above. The variations for 0.25M and 2M concentrations tends to be a line with no fluctuations; for 0.5M the variations is bigger; when for the case of 1M and 1.5M the values variation is more pronounced.

In comparison to the untreated fibre, the mercerisation process provokes an increment of the E_1 value, above the basic system, except for the case of the 2M solution case. For longer immersion times, the difference is lower.



Figure 9.1. E₁ evolution vs. immersion time

Table 9.3. The Fisher analysis of the concentration effect in the E_1

| | 1110 110 | ner anar | <i>y</i> 515 61 chie | | ne ación | chiece in e | | | | |
|------------|----------|----------|----------------------|---------|----------|-------------|----------|---------|----------------------|---------|
| 1h | | Value | 3h | | Value | 12h | | Value | 24h | Value |
| 0.25M 1h - | 0.5M 1h | 405,72 | 0.25M 3h - | 0.5M 3h | 746,35 | 0.25M 12h - | 0.5M 12h | 256,65 | 0.25M 24h - 0.5M 24h | 570,94 |
| 0.25M 1h - | 1M 1h | 41,96 | 0.25M 3h - | 1M 3h | 356,53 | 0.25M 12h - | 1M 12h | 2332,70 | 0.25M 24h - 1M 24h | 805,65 |
| 0.25M 1h - | 1.5M 1h | 2484,39 | 0.25M 3h - | 1.5M 3h | 488,03 | 0.25M 12h - | 1.5M 12h | 140,59 | 0.25M 24h - 1.5M 24h | 2765,52 |
| 0.25M 1h - | 2M 1h | 5772,55 | 0.25M 3h - | 2M 3h | 7097,41 | 0.25M 12h - | 2M 12h | 6733,47 | 0.25M 24h - 2M 24h | 6021,90 |
| 0.5M 1h - | 1M 1h | 363,77 | 0.5M 3h - | 1M 3h | 389,82 | 0.5M 12h - | 1M 12h | 2589,35 | 0.5M 24h - 1M 24h | 234,71 |
| 0.5M 1h - | 1.5M 1h | 2890,12 | 0.5M 3h - | 1.5M 3h | 258,32 | 0.5M 12h - | 1.5M 12h | 116,06 | 0.5M 24h - 1.5M 24h | 2194,58 |
| 0.5M 1h - | 2M 1h | 6178,27 | 0.5M 3h - | 2M 3h | 6351,06 | 0.5M 12h - | 2M 12h | 6990,12 | 0.5M 24h - 2M 24h | 5450,96 |
| 1M 1h - | 1.5M 1h | 2526,35 | 1M 3h - | 1.5M 3h | 131,50 | 1M 12h - | 1.5M 12h | 2473,29 | 1M 24h - 1.5M 24h | 1959,87 |
| 1M 1h - | 2M 1h | 5814,51 | 1M 3h - | 2M 3h | 6740,87 | 1M 12h - | 2M 12h | 4400,77 | 1M 24h - 2M 24h | 5216,25 |
| 1.5M 1h - | 2M 1h | 3288,16 | 1.5M 3h - | 2M 3h | 6609,38 | 1.5M 12h - | 2M 12h | 6874,06 | 1.5M 24h - 2M 24h | 3256,38 |

Table 9.3 studies the concentration effect in the E_1 values. In general, the E_1 values are not greatly affected by low concentration variation. In contrast, for higher solution concentrations, the E_1 value differences tend to be higher. This analysis is supported by the Figure 9.2, where a plateau tendency at lower concentration values, and higher variance for **1M** in advance.

The E_1 values for 0.25M, 0.5M and 1M concentrations are higher than those for the untreated system. For 1.5M concentration, the treated systems mechanical performance starts to decrease, and for 2M concentration there is a definite great loss of properties.



Figure 9.2. E₁ evolution vs. concentration

Summarising the analysis above and the absolute numbers in Table 9.4, the following conclusions are drawn:

- The ANOVA analysis confirms that there is significant effect due to immersion time, concentration or combination of both factors
- Fisher's analysis, that studied the data variation between individual set of values, confirms that E_1 values observed for 0.25M and 2M are more similar than for the rest of concentrations along different immersion times, although the E_1 value drop is substantial at 2M concentration. In addition, when the concentration effect is studied for a specific immersion time, the data variation is less significant for low concentrations
- Table 9.4 suggests that long immersion times (24h) and high concentrations (2M) reduce the flax/epoxy composite performance. The 1h to 12h immersion times and 1.0M to 1.5M concentrations appear to be the most convenient combinations

| CORRECTED E1 modulus (MPa) | B: Concentration | | | | | | | | |
|----------------------------|------------------|----------|----------|----------|---------|--|--|--|--|
| A: Immersion time | 0,25M | 0,5M | 1M | 1,5M | 2M | | | | |
| 1h | 11675,83 | 11563,63 | 12669,16 | 9378,23 | 4884,59 | | | | |
| 3h | 12147,91 | 10176,85 | 12897,66 | 12195,89 | 4141,23 | | | | |
| 12h | 12091,61 | 10791,98 | 10217,37 | 12311,32 | 4027,48 | | | | |
| 24h | 11538,00 | 9865,64 | 11238,33 | 8747,22 | 4292,35 | | | | |

Table 9.4. E₁ values variation with immersion time and concentrations

*NOTE: The longitudinal properties are corrected to V_f=0.3 according to CLT

9.2.2. Longitudinal strength σ₁ ANOVA

There are three hypotheses to prove here. For the case of the null hypothesis, it must be checked whether the means values of all groups are equal under immersion time, concentration and combinative effect. For the case of the longitudinal strength, the ANOVA analysis confirms that there is significant effect due to immersion time, concentration or combination of both **factors**; this was confirmed because after the data analysis the **p-value** is lower than the a = 0.05for the 95% level of significance, see Table 9.5.

| Table 3.J. ANOVA analysis i | | | | | |
|-----------------------------|-----------|-----|-----------|---------|------------------|
| Source | SS | df | MS | Fo | p-value |
| Effect A:Immersion time | 2077,831 | 3 | 692,610 | 16,649 | 0,00014149319847 |
| Effect B:Concentration | 84579,465 | 4 | 21144,866 | 508,285 | 0,00000000000028 |
| Effect AB: Combined | 3740,495 | 12 | 311,708 | 7,493 | 0,00072518551999 |
| Error | 7488,078 | 180 | 41,600 | | |
| Total | 97885.868 | 199 | | | |

Table 0.5 ANOVA analysis for the σ_1

When the box is red in the posterior Fisher's analysis, the difference between mean values is significant, and when the box is unaltered the difference is not significant. The value is significant when the difference is higher than the calculated **LSD**, in this case **5.692**. See the Fisher's analysis in Tables 9.6 and 9.7.

| 0.25M | Value | 0.5M | Value | 1M | Value | 1.5M | Value | 2M | Value |
|-----------------------|-------|---------------------|-------|-----------------|-------|---------------------|-------|-----------------|-------|
| 0.25M 1h - 0.25M 3h | 1,15 | 0.5M 1h - 0.5M 3h | 7,85 | 1M1h - 1M3h | 8,71 | 1.5M 1h - 1.5M 3h | 9,96 | 2M1h - 2M3h | 14,13 |
| 0.25M 1h - 0.25M 12h | 5,61 | 0.5M 1h - 0.5M 12h | 10,26 | 1M 1h - 1M 12h | 0,25 | 1.5M 1h - 1.5M 12h | 16,51 | 2M 1h - 2M 12h | 8,52 |
| 0.25M 1h - 0.25M 24h | 11,60 | 0.5M 1h - 0.5M 24h | 13,04 | 1M 1h - 1M 24h | 3,00 | 1.5M 1h - 1.5M 24h | 1,60 | 2M 1h - 2M 24h | 13,88 |
| 0.25M 3h - 0.25M 12h | 6,76 | 0.5M 3h - 0.5M 12h | 2,41 | 1M 3h - 1M 12h | 8,46 | 1.5M 3h - 1.5M 12h | 6,55 | 2M 3h - 2M 12h | 5,60 |
| 0.25M 3h - 0.25M 24h | 12,75 | 0.5M 3h - 0.5M 24h | 5,19 | 1M 3h - 1M 24h | 11,71 | 1.5M 3h - 1.5M 24h | 8,36 | 2M 3h - 2M 24h | 0,24 |
| 0.25M 12h - 0.25M 24h | 5,99 | 0.5M 12h - 0.5M 24h | 2,78 | 1M 12h - 1M 24h | 3,25 | 1.5M 12h - 1.5M 24h | 14,90 | 2M 12h - 2M 24h | 5,36 |

Table 9.6. Fisher analysis of the immersion time effect in the σ_1

Table 9.6 studies the immersion time effect in the σ_1 values. Excepting for the 0.25M concentration, there is a great change of properties from 1h immersion time to 3h. For the 1M and 1.5M cases the strength rises. From the Fisher's analysis, it might be assumed that after the initial peak, the strength values get to a plateau, where the properties tend to be lower but not drastically. This tendency is clearly supported by the Table data, for the 0.5M and 2M cases.

Figure 9.3 plots the σ_1 variation *vs*. the immersion time. This graph supports the Fisher's analysis above. After the initial fluctuation, the data tends to vary less with the time evolution.

In comparison to the untreated system, most of the treated systems longitudinal strengths are lower, excepting for the case of 0.25M 1h and 1M 3h systems. The mercerisation process appears to decrease the system longitudinal strength properties.



Figure 9.3. σ_1 evolution vs. immersion time

| 1h | | Value | 3h | | Value | 12h | Value | 24h | Value | |
|----|------------|---------|-------|------------|---------|-------|----------------------|-------|----------------------|-------|
| | 0.25M 1h - | 0.5M 1h | 1,22 | 0.25M 3h - | 0.5M 3h | 7,79 | 0.25M 12h - 0.5M 12h | 3,44 | 0.25M 24h - 0.5M 24h | 0,23 |
| | 0.25M 1h - | 1M 1h | 17,37 | 0.25M 3h - | 1M 3h | 9,81 | 0.25M 12h - 1M 12h | 11,51 | 0.25M 24h - 1M 24h | 8,77 |
| | 0.25M 1h - | 1.5M 1h | 33,79 | 0.25M 3h - | 1.5M 3h | 24,98 | 0.25M 12h - 1.5M 12h | 11,68 | 0.25M 24h - 1.5M 24h | 20,58 |
| | 0.25M 1h - | 2M 1h | 51,54 | 0.25M 3h - | 2M 3h | 66,82 | 0.25M 12h - 2M 12h | 54,46 | 0.25M 24h - 2M 24h | 53,83 |
| l | 0.5M 1h - | 1M 1h | 18,59 | 0.5M 3h - | 1M 3h | 2,02 | 0.5M 12h - 1M 12h | 8,07 | 0.5M 24h - 1M 24h | 8,55 |
| ſ | 0.5M 1h - | 1.5M 1h | 35,00 | 0.5M 3h - | 1.5M 3h | 17,19 | 0.5M 12h - 1.5M 12h | 8,24 | 0.5M 24h - 1.5M 24h | 20,36 |
| ſ | 0.5M 1h - | 2M 1h | 52,76 | 0.5M 3h - | 2M 3h | 59,03 | 0.5M 12h - 2M 12h | 51,02 | 0.5M 24h - 2M 24h | 53,60 |
| | 1M 1h - | 1.5M 1h | 16,42 | 1M 3h - | 1.5M 3h | 15,17 | 1M 12h - 1.5M 12h | 0,16 | 1M 24h - 1.5M 24h | 11,81 |
| | 1M 1h - | 2M 1h | 34,17 | 1M 3h - | 2M 3h | 57,01 | 1M 12h - 2M 12h | 42,95 | 1M 24h - 2M 24h | 45,05 |
| ſ | 1 5M 1h - | 2M 1h | 17 75 | 1 5M 3h - | 2M 3h | 41 84 | 15M12h - 2M12h | 42 78 | 1 5M 24h - 2M 24h | 33 24 |

Table 9.7. The Fisher analysis of the concentration effect in the σ_1

Table 9.7 studies the concentration effect in the σ_1 values. In line with the conclusions obtained from the previous analysis (Table 9.6), the maximum mean value fluctuations in different concentrations are obtained for the 3h and 12h immersion times. For the case of the 24h, the values also fluctuate but in a lower proportion. Finally, for the 1h immersion time the concentration produces a progressive loss of strength properties; in the first row of Table 9.7 the difference between the mean values follows a progressive evolution.

This analysis is supported by the Figure 9.4, where there is a clear difference between mean values for 3h and 12h with a lower difference for the 24h case; and the 1h case has progressive loss of properties.

In comparison to the untreated system, the properties are generally lower because of the mercerisation effect, independent of the concentration; being the higher difference for 1.5M and 2M concentrations.



Figure 9.4. σ_1 evolution vs. concentration

Summarising the analysis above and the absolute numbers in Table 9.8, the following conclusions are drawn:

- The ANOVA analysis confirms that there is significant effect due to immersion time, concentration or combination of both factors

- Fisher's analysis, that studied the data variation between individual set of values, suggests that σ_1 values considerably fluctuates from 1h immersion time to 3h, staying similar for the rest of times. In regards to the concentrations, there are values fluctuation from 0.25M to 1M, getting highest values for 1M 3h case. Above 1M the mechanical performance decreased
- Table 9.8 suggests that long immersion times (24h) and high concentrations (2M) reduce the flax/epoxy composite performance. In addition, suggesting 1-12h 0.25M and 3h 1.0M combinations give best mechanical performance

| CORRECTED s1 strength (MPa) | B: Concentration | | | | | | |
|-----------------------------|------------------|--------|--------|--------|-------|--|--|
| A: Immersion time | 0,25M | 0,5M | 1M | 1,5M | 2M | | |
| 1h | 138,81 | 133,44 | 127,91 | 106,14 | 78,18 | | |
| 3h | 141,27 | 119,31 | 142,61 | 117,70 | 68,85 | | |
| 12h | 135,95 | 114,68 | 134,50 | 122,32 | 71,51 | | |
| 24h | 130,81 | 118,84 | 127,89 | 113,40 | 67,11 | | |

Table 9.8. σ_1 values variation with immersion time and concentrations

*NOTE: The longitudinal properties are corrected to V_f=0.3 according to CLT

9.2.3. Longitudinal strain ε₁ ANOVA

There are three hypotheses to prove. For the case of the null hypothesis it must be checked whether the mean values of all groups are equal under immersion time, concentration and combinative effect. For the case of the longitudinal strain, the **ANOVA analysis confirms that there is significant effect due to immersion time, concentration or combination of both factors**; this was confirmed because after the data analysis the **p-value** is lower than the a = 0.05 for the 95% level of significance, see Table 9.9.

| Source | SS | df | MS | Fo | p-value | | |
|-------------------------|---------|-----|--------|---------|------------------|--|--|
| Effect A:Immersion time | 1,864 | 3 | 0,621 | 8,805 | 0,00233067302800 | | |
| Effect B:Concentration | 228,586 | 4 | 57,146 | 809,802 | 0,00000000000002 | | |
| Effect AB: Combined | 7,828 | 12 | 0,652 | 9,244 | 0,00025871237142 | | |
| Error | 12,702 | 180 | 0,071 | | | | |
| Total | 250,980 | 199 | | | | | |

Table 9.9. ANOVA analysis for the ε_1

When the box is red, the posterior Fisher's analysis says the difference between mean values is significant, and when the box is unaltered the difference is not significant. The value is significant
when the difference is higher than the calculated **LSD**, in this case **0.234**. See the Fisher's analysis in Tables 9.10 and 9.11.

| 0.25M | Value | 0.5M | Value | 1M | Value | 1.5M | Value | 2M | Value |
|-----------------------|-------|---------------------|-------|-----------------|-------|---------------------|-------|-----------------|-------|
| 0.25M 1h - 0.25M 3h | 0,09 | 0.5M 1h - 0.5M 3h | 0,03 | 1M 1h - 1M 3h | 0,08 | 1.5M 1h - 1.5M 3h | 0,08 | 2M1h - 2M3h | 0,37 |
| 0.25M 1h - 0.25M 12h | 0,03 | 0.5M 1h - 0.5M 12h | 0,12 | 1M 1h - 1M 12h | 0,26 | 1.5M 1h - 1.5M 12h | 0,13 | 2M1h - 2M12h | 1,24 |
| 0.25M 1h - 0.25M 24h | 0,15 | 0.5M 1h - 0.5M 24h | 0,04 | 1M 1h - 1M 24h | 0,36 | 1.5M 1h - 1.5M 24h | 0,11 | 2M1h - 2M24h | 0,72 |
| 0.25M 3h - 0.25M 12h | 0,07 | 0.5M 3h - 0.5M 12h | 0,09 | 1M 3h - 1M 12h | 0,18 | 1.5M 3h - 1.5M 12h | 0,21 | 2M 3h - 2M 12h | 0,87 |
| 0.25M 3h - 0.25M 24h | 0,06 | 0.5M 3h - 0.5M 24h | 0,01 | 1M 3h - 1M 24h | 0,28 | 1.5M 3h - 1.5M 24h | 0,03 | 2M 3h - 2M 24h | 0,35 |
| 0.25M 12h - 0.25M 24h | 0,13 | 0.5M 12h - 0.5M 24h | 0,07 | 1M 12h - 1M 24h | 0,10 | 1.5M 12h - 1.5M 24h | 0,24 | 2M 12h - 2M 24h | 0,52 |

Table 9.10. Fisher analysis of the immersion time effect in the ε_1

The Table 9.10 studies the immersion time effect in the ε_1 values. According to the table above, most of the differences are below the LSD value, in this case 0.234. There are some red values for 1M and 1.5M columns, however it is considered that there is no significant fluctuation in the number since they are close to LSD number. In contrast, for 2M case there is a great fluctuation between all the values from the same set; according to all red boxes in Fisher.

The Figure 9.5 plots the ε_1 variation *vs*. the immersion time. This graph supports what confirmed in the paragraph above, Fisher's analysis. 0.25M and 0.5M lines are practically unchanged; 1M and 1.5M lines slightly fluctuate; and the 2M values considerably vary with the concentration effect.

In most cases, the longitudinal strain values are higher than the untreated system, and there is a great increment for the case of the 2M concentration.



Figure 9.5. ϵ_1 evolution vs. immersion time

1h Value 3h Value 12h Value 24h Value 0.5M 1h 0.5M 3h 0.25M 1h -0.25M 3h -0.25M 12h - 0.5M 12h 0.25M 24h - 0.5M 24h 0,10 0,16 0,00 0,20 0.25M 1h -1M 1h 0.25M 3h -1M 3h 0.25M 12h -1M 12h 0.25M 24h -1M 24h 0,28 0,11 0,00 0,23 1.5M 1h 1.5M 3h 0.25M 12h - 1.5M 12h 0.25M 1h -0,12 0.25M 3h -0,30 0.25M 24h - 1.5M 24h 0,02 0,39 0.25M 1h -2M 1h 0.25M 3h -2M 3h 2,55 0.25M 12h -2M 12h 0.25M 24h -2M 24h 2,09 2,95 3,35 1M 1h 0,38 0.5M 1h -0.5M 3h -1M 3h 0,27 0.5M 12h -1M 12h 0,00 0.5M 24h -1M 24h 0,03 0.5M 1h -1.5M 1h 0,03 0.5M 3h -1.5M 3h 0,14 0.5M 12h - 1.5M 12h 0,02 0.5M 24h - 1.5M 24h 0,19 0.5M 1h -2M 1h 1,99 0.5M 3h -2M 3h 0.5M 12h -2M 12h 0.5M 24h -2M 24h 2,39 3,34 2,75 1M 1h -1.5M 1h 0,40 1M 3h -1.5M 3h 0,41 1M 12h - 1.5M 12h 0,02 1M 24h - 1.5M 24h 0,16 1M 3h -1M 24h -1M 1h -2M 3h 1M 12h -2M 12h 2M 24h 2M 1h 2,37 2,66 3,34 2,72 1.5M 1h -1.5M 3h -1.5M 12h -2M 12h 3,33 1.5M 24h -2M 24h 2M 1h 1,96 2M 3h 2,25 2.57

Table 9.11. The Fisher analysis of the concentration effect in the ε_1

The Table 9.11 studies the concentration effect in the ε_1 values. The clearest conclusion obtained from Fisher's analysis is that the boxes are in red when compared with 2M data, much higher than the rest of the data population. In a lower scale, something similar happens for the 1.5M concentration, but the difference is not significant for all the cases. This behaviour is common for all the immersion times.

This analysis is supported by the Figure 9.6, where a clear rise in the values for 2M concentration is observed, and in a lower scale for the 1.5M concentration.

The strain values suffer a great increment for 2M case in comparison to the untreated system, and for the rest of concentrations the values are similar.



Figure 9.6. ϵ_1 evolution vs. concentration

Summarising the analysis above and the absolute numbers in Table 9.12, the following conclusions are drawn:

- The ANOVA analysis confirms that there is significant effect due to immersion time, concentration or combination of both factors
- In regards to the immersion times, the 2M values are the ones fluctuating most. Fisher's analysis, that studied the data variation between individual set of values, suggests that ϵ_1

values considerably rise when the concentration increases, a little for 1.5M and drastically for

2M

Table 9.12 suggests that long immersion times from 1h to 12h and high concentrations (2M) increase the strain values. In addition, 24h immersion time and 2M concentration suggest that the fibre was degraded during the treatment because of the low values

| CORRECTED ɛ1 strain (%) | B: Concentration | | | | | | | |
|-------------------------|------------------|------|------|------|------|--|--|--|
| A: Immersion time | 0,25M | 0,5M | 1M | 1,5M | 2M | | | |
| 1h | 1,67 | 1,70 | 1,45 | 1,94 | 4,13 | | | |
| 3h | 1,58 | 1,59 | 1,59 | 2,03 | 5,10 | | | |
| 12h | 1,69 | 1,47 | 1,87 | 1,72 | 5,86 | | | |
| 24h | 1,56 | 1,65 | 1,97 | 2,19 | 5,39 | | | |
| | | | | | | | | |

Table 9.12. ε_1 values variation with immersion time and concentrations

*NOTE: The longitudinal properties are corrected to V_f=0.3 according to CLT

9.2.4. Transverse tensile modulus E₂ ANOVA

There are three hypotheses to prove. For the case of the null hypothesis it must be checked whether the mean values of all groups are equal under immersion time, concentration and combinative effect. For the case of the transverse modulus, the **ANOVA analysis confirms that there is significant effect due to immersion time, concentration or combination of both factors**; this was confirmed because after the data analysis the **p-value** is lower than the a = 0.05 for the 95% level of significance, see Table 9.13.

| Source | SS | df | MS | FO | p-value |
|--------------------------|---------------|-----|--------------|---------|-------------------|
| Effect A: Immersion time | 3724535,905 | 3 | 1241511,968 | 8,286 | 0,00003709505623 |
| Effect B: Concentration | 191631888,940 | 4 | 47907972,235 | 319,736 | 0,000000000000000 |
| Effect AB: Combined | 14084428,872 | 12 | 1173702,406 | 7,833 | 0,00000000002225 |
| Error | 23973742,731 | 160 | 149835,892 | | |
| Total | 233414596,448 | 179 | | | |

Table 9.13. ANOVA analysis for the E₂

When the box is red, the posterior Fisher's analysis says the difference between mean values is significant, and when the box is unaltered the difference is not significant. The value is significant when the difference is higher than the calculated **LSD**, in this case **341.876**. See the data in Tables 9.14 and 9.15 where Fisher's analysis was done.

Table 9.14. Fisher analysis of the immersion time effect in the E₂

| 0.25M | Value | 0.5M | Value | 1M | Value | 1.5M | Value | 2M | Value |
|-----------------------|--------|---------------------|--------|-----------------|--------|---------------------|---------|-----------------|--------|
| 0.25M 1h - 0.25M 3h | 294,38 | 0.5M 1h - 0.5M 3h | 259,11 | 1M 1h - 1M 3h | 192,68 | 1.5M 1h - 1.5M 3h | 532,83 | 2M 1h - 2M 3h | 8,33 |
| 0.25M 1h - 0.25M 12h | 723,53 | 0.5M1h - 0.5M12h | 367,20 | 1M 1h - 1M 12h | 440,43 | 1.5M 1h - 1.5M 12h | 544,25 | 2M 1h - 2M 12h | 218,00 |
| 0.25M 1h - 0.25M 24h | 294,16 | 0.5M 1h - 0.5M 24h | 87,03 | 1M 1h - 1M 24h | 1,78 | 1.5M 1h - 1.5M 24h | 711,47 | 2M 1h - 2M 24h | 78,04 |
| 0.25M 3h - 0.25M 12h | 429,15 | 0.5M 3h - 0.5M 12h | 626,31 | 1M 3h - 1M 12h | 247,75 | 1.5M 3h - 1.5M 12h | 1077,07 | 2M 3h - 2M 12h | 226,33 |
| 0.25M 3h - 0.25M 24h | 0,23 | 0.5M 3h - 0.5M 24h | 346,14 | 1M 3h - 1M 24h | 194,46 | 1.5M 3h - 1.5M 24h | 1244,30 | 2M 3h - 2M 24h | 69,72 |
| 0.25M 12h - 0.25M 24h | 429,38 | 0.5M 12h - 0.5M 24h | 280,18 | 1M 12h - 1M 24h | 442,21 | 1.5M 12h - 1.5M 24h | 167,23 | 2M 12h - 2M 24h | 296,04 |

Table 9.14 studies the immersion time effect in the E₂ values according to Fisher analysis. For the first three concentrations, 0.25M, 0.5M and 1M, the immersion time effect for 1h and 24h values is the same, getting similar values for both cases; for middle way immersion times, 3h and 12h, the E₂ values are fluctuating. For the 1.5M concentration, the values are varying excepting from 12h to 24h that the values are keeping stable. Finally for the case of the 2M concentration, the value difference is lower than the **LSD=341.876** for all immersion times.

The Figure 9.7 plots the E_2 variation *vs*. the immersion time. This graph supports Fisher's analysis above. The lowest variation was obtained for 2M case, followed by the 0.25M, 0.5M and 1M cases, being 1.5M data the set of results fluctuating the most.

Transverse modulus properties are higher that the untreated system, excepting for the case of the 2M concentration. Independently from the immersion time.



Figure 9.7. E₂ evolution vs. immersion time

Table 9.15. Fisher analysis of the concentration effect in the E₂

| 1h | | Value | 3h | | Value | 12 | h | Value | 24 | า | Value |
|------------|---------|---------|------------|---------|---------|-------------|----------|---------|-------------|----------|---------|
| 0.25M 1h - | 0.5M 1h | 241,07 | 0.25M 3h - | 0.5M 3h | 205,80 | 0.25M 12h - | 0.5M 12h | 849,66 | 0.25M 24h - | 0.5M 24h | 140,11 |
| 0.25M 1h - | 1M 1h | 95,01 | 0.25M 3h - | 1M 3h | 582,07 | 0.25M 12h - | 1M 12h | 1258,97 | 0.25M 24h - | 1M 24h | 387,38 |
| 0.25M 1h - | 1.5M 1h | 742,31 | 0.25M 3h - | 1.5M 3h | 503,87 | 0.25M 12h - | 1.5M 12h | 2010,09 | 0.25M 24h - | 1.5M 24h | 1747,94 |
| 0.25M 1h - | 2M 1h | 1918,00 | 0.25M 3h - | 2M 3h | 1631,95 | 0.25M 12h - | 2M 12h | 976,47 | 0.25M 24h - | 2M 24h | 1701,89 |
| 0.5M 1h - | 1M 1h | 336,08 | 0.5M 3h - | 1M 3h | 787,87 | 0.5M 12h - | 1M 12h | 409,31 | 0.5M 24h - | 1M 24h | 247,27 |
| 0.5M 1h - | 1.5M 1h | 983,38 | 0.5M 3h - | 1.5M 3h | 709,66 | 0.5M 12h - | 1.5M 12h | 1160,42 | 0.5M 24h - | 1.5M 24h | 1607,82 |
| 0.5M 1h - | 2M 1h | 1676,93 | 0.5M 3h - | 2M 3h | 1426,15 | 0.5M 12h - | 2M 12h | 1826,13 | 0.5M 24h - | 2M 24h | 1842,00 |
| 1M 1h - | 1.5M 1h | 647,30 | 1M 3h - | 1.5M 3h | 78,21 | 1M 12h - | 1.5M 12h | 751,11 | 1M 24h - | 1.5M 24h | 1360,55 |
| 1M 1h - | 2M 1h | 2013,01 | 1M 3h - | 2M 3h | 2214,02 | 1M 12h - | 2M 12h | 2235,44 | 1M 24h - | 2M 24h | 2089,27 |
| 1.5M 1h - | 2M 1h | 2660 31 | 1.5M 3h - | 2M 3h | 2135.81 | 1.5M 12h - | 2M 12h | 2986 56 | 1.5M 24h - | 2M 24h | 3449 82 |

The Table 9.15 studies the concentration effect in the E_2 values. The clearest conclusion obtained in the Fisher analysis is shown by the last line of the table. The value drop is huge when increasing the solution concentration from 1.5M to 2M. Similarly, when the concentration is increased from 1M to 1.5M the properties increment is significant, excepting for the case of the 3h immersion time. Finally confirm that the value fluctuation looks to be proportional to the immersion time, highest fluctuation was obtained for the longest immersion times.

Figure 9.8 confirms that highest fluctuation was obtained for immersion times between 1M and 2M concentrations.

The transverse modulus drastically drops to values lower than the untreated systems, when the treatment concentration is 2M.



Figure 9.8. E₂ evolution vs. concentrations

Summarising the analysis above and the absolute numbers in Table 9.16, the following conclusions are drawn:

- The ANOVA analysis confirms that there is significant effect due to immersion time, concentration or combination of both factors
- In regards to the immersion times, it looks like in general for 12h the E₂ values are increasing, and 24h immersion time is too long, excepting for the 1.5M concentration.
 In regards to the concentrations it looks like the E₂ values are increasing for 1M and 1.5M cases. However, for 2M there is a dramatic loss of properties
- Table 9.16 suggests that 2M concentration is clearly damaging the fibre/matrix interface, getting a drastic decrease in the mechanical properties. Similarly, it looks like the 24h long

immersion times are generally reducing the E_2 value, excepting for the 1.5M case where optimised 4.67 MPa value is obtained.

| | | oron and and a | | | |
|-------------------|---------|----------------|---------------|---------|---------|
| E2 modulus (MPa) | | B: C | Concentration | | |
| A: Immersion time | 0,25M | 0,5M | 1M | 1,5M | 2M |
| 1h | 3058,53 | 2790,68 | 3164,10 | 3883,32 | 927,42 |
| 3h | 2731,44 | 2502,78 | 3378,19 | 3291,29 | 918,17 |
| 12h | 2254,61 | 3198,68 | 3653,47 | 4488,04 | 1169,64 |
| 24h | 2731,69 | 2887,37 | 3162,12 | 4673,84 | 840,71 |

Table 9.16. E₂ values variation with immersion time and concentrations

9.2.5. Transverse strength σ₂ ANOVA

There are three hypotheses to prove. For the case of the null hypothesis, it must be checked whether the means values of all groups are equal under immersion time, concentration and combinative effect. For the case of the transverse strength, the **ANOVA analysis confirms that there is significant effect due to immersion time, concentration or combination of both factors**; this was confirmed because after the data analysis the **p-value** is lower than the a = 0.05 for the 95% level of significance, see Table 9.17.

Table 9.17. ANOVA analysis for the σ_2

| Source | SS | df | MS | Fo | p-value |
|-------------------------|----------|-----|----------|---------|-------------------|
| Effect A:Immersion time | 75,379 | 3 | 25,126 | 8,005 | 0,00005269874953 |
| Effect B:Concentration | 5678,963 | 4 | 1419,741 | 452,340 | 0,000000000000000 |
| Effect AB: Combined | 290,418 | 12 | 24,202 | 7,711 | 0,0000000003374 |
| Error | 502,185 | 160 | 3,139 | | |
| Total | 6546,946 | 179 | | | |

When the box is red, the posterior Fisher's analysis says the difference between mean values is significant, and when the box is unaltered the difference is not significant enough. The value is significant when the difference is higher than the calculated **LSD**, in this case **1.565**. See the data in Tables 9.18 and 9.19 where Fisher's analysis was done.

| | Table 9.18. | Fisher | analysis | of | the | immersion | time | effect in | the σ_2 |
|--|-------------|--------|----------|----|-----|-----------|------|-----------|----------------|
|--|-------------|--------|----------|----|-----|-----------|------|-----------|----------------|

| 0.25M | Value | 0.5M | Value | 1M | Value | 1.5M | Value | 2M | Value | |
|-----------------------|-------|---------------------|-------|-----------------|-------|---------------------|-------|-----------------|-------|--|
| 0.25M 1h - 0.25M 3h | 1,22 | 0.5M 1h - 0.5M 3h | 1,22 | 1M 1h - 1M 3h | 1,64 | 1.5M 1h - 1.5M 3h | 3,98 | 2M 1h - 2M 3h | 1,28 | |
| 0.25M 1h - 0.25M 12h | 1,94 | 0.5M 1h - 0.5M 12h | 0,52 | 1M 1h - 1M 12h | 1,33 | 1.5M 1h - 1.5M 12h | 0,84 | 2M 1h - 2M 12h | 1,70 | |
| 0.25M 1h - 0.25M 24h | 0,59 | 0.5M 1h - 0.5M 24h | 2,95 | 1M 1h - 1M 24h | 1,78 | 1.5M 1h - 1.5M 24h | 1,04 | 2M 1h - 2M 24h | 0,07 | |
| 0.25M 3h - 0.25M 12h | 3,16 | 0.5M 3h - 0.5M 12h | 0,70 | 1M 3h - 1M 12h | 0,30 | 1.5M 3h - 1.5M 12h | 4,83 | 2M 3h - 2M 12h | 0,42 | |
| 0.25M 3h - 0.25M 24h | 1,81 | 0.5M 3h - 0.5M 24h | 1,73 | 1M 3h - 1M 24h | 3,42 | 1.5M 3h - 1.5M 24h | 2,95 | 2M 3h - 2M 24h | 1,21 | |
| 0.25M 12h - 0.25M 24h | 1,35 | 0.5M 12h - 0.5M 24h | 2,43 | 1M 12h - 1M 24h | 3,11 | 1.5M 12h - 1.5M 24h | 1,88 | 2M 12h - 2M 24h | 1,63 | |

Table 9.18 studies the immersion time effect in the σ_2 values according to Fisher analysis. For the concentrations from 0.25M and 0.5M, the σ_2 values fluctuate along the four immersion times; the 1M and 1.5M fluctuate more according to Fisher; finally the 2M measurements tend to be more stable.

Figure 9.9 plots the σ_2 variation vs. the immersion time. This graph supports Fisher's analysis above. The lowest variation was obtained for 2M case, followed by the 0.25M and 0.5M, being 1M and 1.5M cases the set of results fluctuating the most.

Transverse strength values are higher and lower than the untreated system depending on the immersion times, excepting for the case of the 2M concentration that there is a great drop of properties.



Figure 9.9. σ_2 evolution vs. Immersion time

| 1h | | Value | 3h | | Value | 12 | h | Value | 24h | 1 | Value |
|------------|---------|-------|------------|---------|-------|-------------|----------|-------|-------------|----------|-------|
| 0.25M 1h - | 0.5M 1h | 2,31 | 0.25M 3h - | 0.5M 3h | 0,13 | 0.25M 12h - | 0.5M 12h | 3,73 | 0.25M 24h - | 0.5M 24h | 0,05 |
| 0.25M 1h - | 1M 1h | 3,89 | 0.25M 3h - | 1M 3h | 4,30 | 0.25M 12h - | 1M 12h | 7,16 | 0.25M 24h - | 1M 24h | 2,69 |
| 0.25M 1h - | 1.5M 1h | 6,47 | 0.25M 3h - | 1.5M 3h | 1,27 | 0.25M 12h - | 1.5M 12h | 9,26 | 0.25M 24h - | 1.5M 24h | 6,02 |
| 0.25M 1h - | 2M 1h | 9,70 | 0.25M 3h - | 2M 3h | 9,64 | 0.25M 12h - | 2M 12h | 6,06 | 0.25M 24h - | 2M 24h | 9,04 |
| 0.5M 1h - | 1M 1h | 1,58 | 0.5M 3h - | 1M 3h | 4,43 | 0.5M 12h - | 1M 12h | 3,43 | 0.5M 24h - | 1M 24h | 2,75 |
| 0.5M 1h - | 1.5M 1h | 4,16 | 0.5M 3h - | 1.5M 3h | 1,40 | 0.5M 12h - | 1.5M 12h | 5,53 | 0.5M 24h - | 1.5M 24h | 6,08 |
| 0.5M 1h - | 2M 1h | 12,01 | 0.5M 3h - | 2M 3h | 9,51 | 0.5M 12h - | 2M 12h | 9,79 | 0.5M 24h - | 2M 24h | 8,99 |
| 1M 1h - | 1.5M 1h | 2,59 | 1M 3h - | 1.5M 3h | 3,03 | 1M 12h - | 1.5M 12h | 2,10 | 1M 24h - | 1.5M 24h | 3,33 |
| 1M 1h - | 2M 1h | 13,58 | 1M 3h - | 2M 3h | 13,94 | 1M 12h - | 2M 12h | 13,21 | 1M 24h - | 2M 24h | 11,73 |
| 1.5M 1h - | 2M 1h | 16,17 | 1.5M 3h - | 2M 3h | 10.91 | 1.5M 12h - | 2M 12h | 15.31 | 1.5M 24h - | 2M 24h | 15,07 |

Table 9.19. Fisher analysis of the concentration effect in the σ_2

Table 9.19 studies the concentration effect in the σ_2 values. The clearest conclusion obtained in the Fisher analysis is shown by the last line of the table. The value drop is huge when increasing the solution concentration from 1.5M to 2M. For 1h, 12h and 24h cases, there is a clear strength improvement with along the concentration, until 1.5M concentration is reached with the highest value and drop is obtained for 2M. 3h case is the exception, where the property drop happens from 1M to 2M concentrations.

Figure 9.10 confirms that 2M concentration is too elevated and provokes a drop in the mechanical performance of the flax/epoxy composite. At the same way, 1M and 1.5M would be the most optimum concentrations.

According to the Figure 9.10, the properties are lower that the untreated system for 0.25M, 0.50M systems and higher for 1M, 1.5M systems. For the case of 2M the documents are much lower than the basic system.



Figure 9.10. σ_2 evolution vs. concentrations

Summarising the analysis above and the absolute numbers in Table 9.20, the following conclusions are drawn:

- The ANOVA analysis confirms that there is significant effect due to immersion time, concentration or combination of both factors
- It is confirmed that 24h immersion time and 2M concentration are too long/high respectively and the properties are clearly reduced. Fluctuation was obtained for immersion times between 1M and 2M concentrations, but 1M and 1.5M might be considered the optimum concentrations.
- Table 9.20 suggests that 2M concentration is clearly damaging the flax fibre, getting a drastic decrease in the mechanical properties. Similarly it looks like the 24h long immersion times are generally reducing the E₂ value, excepting for the 1.5M case where the severity is lower. According to the table 1.5M concentration is working better in all immersion times.

| s2 Strength(MPa) | B: Concentration | | | | | | |
|-------------------|------------------|-------|-------|-------|------|--|--|
| A: Immersion time | 0,25M | 0,5M | 1M | 1,5M | 2M | | |
| 1h | 17,03 | 19,60 | 21,35 | 24,23 | 6,26 | | |
| 3h | 18,39 | 18,25 | 23,17 | 19,80 | 7,68 | | |
| 12h | 14,88 | 19,02 | 22,83 | 25,16 | 8,15 | | |
| 24h | 16,38 | 16,32 | 19,37 | 23,07 | 6,33 | | |

Table 9.20. σ_2 values variation with immersion time and concentrations

9.2.6. Transverse strain ε₂ ANOVA

There are three hypotheses to prove. For the case of the null hypothesis it must be checked whether the means values of all groups are equal under immersion time, concentration and combinative effect. For the case of the duration ANOVA analysis confirms that **there is NO main effect due to immersion time**, since the **p-value** is higher than the a = 0.05 for the 95%, 0.059. For the concentration or combination of both factors; ANOVA analysis confirms that **there is main effect due to concentration or combination of both factors**, since the **p-value** is lower than the a=0.05 for the 95% level of significance, see Table 9.21.

| Table 9.21. ANOVA analy | ysis for | the ε_2 |
|-------------------------|----------|---------------------|
|-------------------------|----------|---------------------|

| Source | SS | df | MS | F_{0} | p-value |
|--------------------------|--------|-----|-------|---------|-------------------|
| Effect A: Immersion time | 0,331 | 3 | 0,110 | 2,527 | 0,05940115382354 |
| Effect B: Concentration | 14,118 | 4 | 3,529 | 80,740 | 0,000000000000000 |
| Effect AB: Combined | 1,184 | 12 | 0,099 | 2,256 | 0,01152343425999 |
| Error | 6,994 | 160 | 0,044 | | |
| Total | 22,627 | 179 | | | |

When the box is red, the posterior Fisher's analysis says the difference between mean values is significant, and when the box is unaltered the difference is not significant. The value is significant when the difference is higher than the calculated **LSD**, in this case **0.185**. The Fisher's analysis data is presented in Tables 9.22 and 9.23.

| 0.25M | Value | 0.5M | Value | 1M | Value | 1.5M | Value | 2M | |
|----------------------|-------|--------------------|-------|----------------|-------|--------------------|-------|----------------|--|
| 0.25M 1h - 0.25M 3h | 0,06 | 0.5M 1h - 0.5M 3h | 0,09 | 1M 1h - 1M 3h | 0,08 | 1.5M 1h - 1.5M 3h | 0,01 | 2M 1h - 2M 3h | |
| 0.25M 1h - 0.25M 12h | 0,05 | 0.5M 1h - 0.5M 12h | 0,05 | 1M 1h - 1M 12h | 0,08 | 1.5M 1h - 1.5M 12h | 0,01 | 2M1h - 2M12h | |
| 0.25M 1h - 0.25M 24h | 0,01 | 0.5M 1h - 0.5M 24h | 0,13 | 1M 1h - 1M 24h | 0,25 | 1.5M 1h - 1.5M 24h | 0,04 | 2M 1h - 2M 24h | |
| 0.25M 3h - 0.25M 12h | 0,01 | 0.5M 3h - 0.5M 12h | 0,14 | 1M 3h - 1M 12h | 0,00 | 1.5M 3h - 1.5M 12h | 0,02 | 2M 3h - 2M 12h | |
| 0 25M 3h - 0 25M 24h | 0.05 | 05M3b - 05M24b | 0.22 | 1M3h - 1M24h | 0.17 | 15M3h - 15M24h | 0.05 | 2M3h - 2M24h | |

0.25M 12h - 0.25M 24h 0.64 0.5M 12h - 0.5M 24h 0.09 1M 12h - 1M 24h 0.17 1.5M 12h - 1.5M 24h 0.02 2M 12h -

0,00

Table 9.22. Fisher analysis of the immersion time effect in the ε_2

The Table 9.22 studies the immersion time effect in the ε_2 values according to Fisher analysis. Excepting for the 2M concentration, the transverse strain values are not changing significantly with the immersion time variation. In contrast, for the 2M data it is seen that the values are fluctuating, more in detail for the 12h immersion time test, where the value is lower in comparison to the rest of the set.

The analysis in the paragraph above it is supported by the Figure 9.11, where it is clearly seen that 2M data set is fluctuating most along the immersion times.

The transverse strain in general is slightly higher than the untreated systems for most of the cases, excepting the 2M case that obtains much higher values.



Figure 9.11. ε_2 evolution vs. immersion time

| 1h | | Value | 3h | | Value | 12h | Value | 24h | Value |
|-----------------|------|-------|------------|---------|-------|----------------------|-------|----------------------|-------|
| 0.25M 1h - 0.5M | 4 1h | 0,13 | 0.25M 3h - | 0.5M 3h | 0,16 | 0.25M 12h - 0.5M 12h | 0,03 | 0.25M 24h - 0.5M 24h | 0,02 |
| 0.25M 1h - 11 | 4 1h | 0,16 | 0.25M 3h - | 1M 3h | 0,03 | 0.25M 12h - 1M 12h | 0,04 | 0.25M 24h - 1M 24h | 0,09 |
| 0.25M 1h - 1.5M | 4 1h | 0,03 | 0.25M 3h - | 1.5M 3h | 0,02 | 0.25M 12h - 1.5M 12h | 0,03 | 0.25M 24h - 1.5M 24h | 0,02 |
| 0.25M 1h - 21 | 4 1h | 0,72 | 0.25M 3h - | 2M 3h | 0,67 | 0.25M 12h - 2M 12h | 0,41 | 0.25M 24h - 2M 24h | 0,81 |
| 0.5M 1h - 11 | 4 1h | 0,04 | 0.5M 3h - | 1M 3h | 0,13 | 0.5M 12h - 1M 12h | 0,01 | 0.5M 24h - 1M 24h | 0,07 |
| 0.5M 1h - 1.5M | 4 1h | 0,10 | 0.5M 3h - | 1.5M 3h | 0,18 | 0.5M 12h - 1.5M 12h | 0,06 | 0.5M 24h - 1.5M 24h | 0,00 |
| 0.5M 1h - 21 | 4 1h | 0,60 | 0.5M 3h - | 2M 3h | 0,51 | 0.5M 12h - 2M 12h | 0,37 | 0.5M 24h - 2M 24h | 0,83 |
| 1M 1h - 1.5M | 4 1h | 0,13 | 1M 3h - | 1.5M 3h | 0,05 | 1M 12h - 1.5M 12h | 0,07 | 1M 24h - 1.5M 24h | 0,08 |
| 1M 1h - 21 | 4 1h | 0,56 | 1M 3h - | 2M 3h | 0,64 | 1M 12h - 2M 12h | 0,37 | 1M 24h - 2M 24h | 0,90 |
| 1.5M 1h - 21 | 4 1h | 0,69 | 1.5M 3h - | 2M 3h | 0,69 | 1.5M 12h - 2M 12h | 0,44 | 1.5M 24h - 2M 24h | 0,82 |

Table 9.23. Fisher analysis of the concentration effect in the ε_2

Table 9.23 studies the concentration effect in the ϵ_2 values. According to the table above it is seen same tendency for all the four immersion times: the transverse strain is almost a plateau between 0.25M and 1.5M concentration, however the strain values drastically raise from 1.5M to 2M concentrations.

Figure 9.12 supports what is described in the paragraph above. The transverse strain values drastically increase for 2M concentration.

The great difference happens for 2M system in comparison to the untreated system, where the transverse strain values increment is visible.



Figure 9.12. ε_2 evolution vs. concentrations

Summarising the analysis above and the absolute numbers in Table 9.24, the following conclusions are drawn:

- The ANOVA analysis confirms that there is NO significant effect due to immersion time, there is significant effect due to concentration or combination of both factors
- It is confirmed that only for 2M set of values the concentrations are affecting the strain values. Similarly, when the fibres are treated at 2M the strain values suffer a great value rise
- Table 9.24 suggests that 2M concentration is clearly affecting the composite performance, getting a drastic increase in the strain values. It is suspected that the aggressive treatment might be damaging the fibre/matrix interface

| ε2 Strain (%) | B: Concentration | | | | |
|-------------------|------------------|------|------|------|------|
| A: Immersion time | 0,25M | 0,5M | 1M | 1,5M | 2M |
| 1h | 0,59 | 0,73 | 0,77 | 0,62 | 1,39 |
| 3h | 0,65 | 0,83 | 0,68 | 0,63 | 1,39 |
| 12h | 0,64 | 0,68 | 0,68 | 0,60 | 1,09 |
| 24h | 0,60 | 0,58 | 0,49 | 0,58 | 1,49 |

Table 9.24. ε_2 values variation with immersion time and concentrations

9.2.7. Poisson's ratio v21 ANOVA

The Poisson's ratio was not evaluated by ANOVA for two reasons.

- First, low signal-to-noise ratio of the video-extensometer camera resolution means the system is not able to resolve small deformations. This problem might be solved either by measuring the deformations with strain gauges or by increasing the video-extensometer camera resolution. The greater longitudinal strain is sensibly resolved by the video extensometer. However, the Poisson's ratio data generated is not reliable because of the uncertainty in the measured transverse strain.
- Second, it is not clear what the optimum magnitude should be for Poisson's ratio. Normally, each fabric has got its own natural value; Craig and Summerscales (Craig & Summerscales, 1988) studied two glass-fibre laminates (unidirectional and bidirectional) experimentally and presented a full set of axial moduli and Poisson's ratios for the three orthogonal planes in the respective composites. The values obtained were shown to satisfy Lempriere's criteria (Lempriere, 1968) which impose a thermodynamic constraint. A full study of this characteristic for the materials in the doctoral study is beyond the scope and duration of the project given that available systems cannot resolve the data.

In this case the UD fabric value must be around 0.25 and 0.35. Away from this range it might be confirmed that the immersion time and the concentration is affecting the Poisson's ratio, but no more conclusion might be obtained from ANOVA analysis

9.2.8. Definitive mercerisation process selection

Based on all the properties obtained from the mercerisation process and the ANOVA study, an optimum mercerisation process was selected. ANOVA analysis was performed in §9.2.1-9.2.6; with the mechanical data plotted in Figures from 9.13 to 9.16:

- Figure 9.13. Longitudinal tensile properties vs concentration
- Figure 9.14. Longitudinal tensile properties vs immersion time
- Figure 9.15. Transverse tensile properties vs immersion time
- Figure 9.16. Transverse tensile properties vs concentration

The main conclusions obtained from the analysis were:

- The ANOVA analysis confirms that for most of the properties there is significant effect due to immersion time and concentration or combination of both factors. Excepting for the transverse strain values where there is NO significant effect due to immersion time and there is significant effect due to concentration or combination of both factors The posterior Fisher's analysis studied the data variations between individual set of values and confirms their variations in different tables and plots
- Table 9.2 presents that analysis. For a sensible balance of these four properties (E₁, E₂, σ'₁ and σ'₂), the caustic soda concentration is in the range 0.5M to 1.5M with treatment times of 3h or 12h giving the best balance of mechanical properties. In regards to the longitudinal properties, 3h immersion time in a 1M concentration was the optimised combination. Additionally, making the **balance between the transverse modulus and strength**, the best system selected would be the **12h immersion time in a 1.5M concentration**. For **best longitudinal properties** while maintaining sensible transverse properties, the **3h and 1M combination was selected.** Appropriate transverse properties means that there is a great locking between fibre and matrix after the treatment. The idea is to get even better transverse properties after the silane addition to the interface
- It is certainly clear that 2.0M NaOH is not realising the potential of the reinforcement fibres.
 Clearly 2.0M NaOH is not the appropriate choice given the high proportion of red cells for E₁,
 E₂, σ'₁ and σ'₂. **24h immersion time** and **2M concentration** is damaging to the composite system, reducing drastically its performance

From the paragraph above it is concluded that the 1M/3h combination would be obtaining the best mechanical performance from all the options; however, the immersion produced during the mercerisation process looks to be decreasing the mechanical performance in comparison to the untreated system. The reply to the research question would be that the mercerisation process is decreasing the flax/epoxy system mechanical performance.







Figure 9.14. Longitudinal tensile properties vs immersion time







Figure 9.16. Transverse tensile properties vs concentration

9.3. PLATE MANUFACTURE

A series of 600 x 330 mm composite plates were manufactured by resin infusion under flexible tooling with a flow medium (RIFT II). The flow direction was parallel to the long axis of the plates with the flow medium covering the first quarter (150 mm) of the length. Up to 1.0M NaOH treatment, the reinforcement pack filled in ~10-12 minutes. The 1.5M NaOH panel took ~30 minutes to fill. The 2.0M NaOH treated reinforcement did not fill until the flow medium was extended to within 50 mm of the downstream edge. The selected 3h and 1M combination treatment (§9.2) is thus compatible with short process times.

In the plate manufacturing it was detected that the fibre degradation (§9.2) and swelling (§9.3) due to the mercerisation process, was affecting the plate manufacturing. For that reason different solutions were developed along the research project.

9.4. EFFECT OF FIBRE SWELLING

The mercerisation process provokes the fibre swelling, directly affecting the composite mechanical and flow properties. This phenomenon is studied in this subsection as a part of the overall research question.

The fibre swelling may provoke two effects: the resin may be delayed if the flow front constrains liquid moving forwards or may have a favourable effect by forcing resin forwards. These two effects must be included in the mass conservation equation using sink and source terms. As a result, the permeability value may be varied with the exposure time and position. Nguyen applied this variation in his models, varying permeability (mass source/sink terms) which led to better agreement with the experimental flow measurements than the constant permeability model. Mass sink became higher when the fibre volume increased.

Optical microscopy to determine the apparent fibre diameter revealed that fibres exposed to mercerisation were swollen up to 20% linearly (i.e. change in the diameter, Table 8.3) during the most aggressive treatments. The lignocellulosic material thus occupies a greater volume for the same areal weight of reinforcement fabric. Cripps et al. proposed that the dependence of permeability on

fibre volume fraction (assuming no contact between the individual filaments) would be (Cripps, Searle & Summerscales, 2000):

K α
$$(1 - V_f)^3/V_f^2$$
 or $ε^3/(1 - ε)^2$ Equation 9.1
Where the porosity is $ε = 1 - V_f$

For a reduction of permeability of one third for over-treated fibres relative to untreated fibres (1.5 M *vs* 1.0M NaOH), the above proportionality can be recalculated; because the swollen fibre would increase the cavity size as the flow progresses during RIFT process. Using the volume fraction of the swollen fibre derived from the apparent diameters (assuming circular fibre cross-section), the equation becomes $V_{ft} = \phi t^2 V_{fu}/\phi u^2$, where ϕ_x is the apparent diameter of the treated (subscript t) or untreated (subscript u) fibre. Subject to the foregoing assumptions, we might expect the new permeability to be predicted by Equation 9.1a:

$$\frac{K_t}{K_u} = \frac{1}{3} = \frac{(1 - V_{ft})^3}{(1 - V_{fu})^3} \frac{V_{fu}^2}{V_{ft}^2}$$
 Equation 9.1a

While appropriate quantitative data for permeability is not available due to the complex deformation history during the fabric compression and flow stages of resin infusion (and the limited monitoring capability in the manufacturing laboratory), the qualitative data are consistent with the expectations of the Cripps et al proportionality. See Figure 9.17:



Figure 9.17: Swelling model representation graph

Further, Francucci et al. (Francucci, Rodríguez & Vázquez, 2010) and Masoodi et al. (Masoodi, Pillai, Grahl & Tan, 2012) have studied the absorption, and consequent swelling, of natural fibre reinforcements by permeant fluids. The increased fibre diameter was indicated to be the principal reason for inconsistencies in permeability measurements for these reinforcements. In a study by Nguyen et al. on the influence of liquid absorption and fibre swelling during RTM resin impregnation of flax fibre reinforcements, the authors suggested a relationship between fibre swelling and permeability as given by Equation 9.2 (Nguyen, Lagardère, Cosson & Park, 2014):

$$K = (1 - f_{SW^2} V_f)^{n+1} / A (f_{SW^2} V_f)^n$$
 Equation 9.2

where K is the permeability, f_{sw} is the fibre swelling ratio (wet diameter/dry diameter), V_f is the fibre volume fraction and A and n are empirically derived constants. Note that Equation 9.2 has a similar format to Equation 9.1, now with exponent n (instead of 2) and a new parameter, f_{sw} .

9.5. SILANE COUPLING AGENT

Having established that 1.0M NaOH treatment for 3 hours (See §9.2.8.) should provide a sensible balance of mechanical properties without extended treatment and resin infusion times, this combination was used for the **definitive tests** to confirm or refute the hypothesis that silane treatment would enhance the properties and might be achieved by adding the silane to the resin hardener rather than extending fibre treatment times by further processing the reinforcement before composites manufacture.

The silane molecule (coupling agent) is normally an oleophilic (aliphatic) chain with an oleophobic reactive entity (e.g. silane) as a terminal group. To minimise the chemical energy of the system, the oleophobic entity should be expelled from the liquid resin to the surfaces, or in bulk systems to any available interface where it may react, dependent on compatibility of the reactive entity with the reinforcement fibre. Mixing the coupling agent into the resin system hardener will eliminate the need for fibre/fabric treatment prior to composite manufacture. Should this approach be demonstrated to work, then the laminate manufacture will be more "sustainable": the solvents used for fabric

treatments are eliminated (the hardener becomes the necessary solvent) with benefits to the environment and to worker health.

A series of panels were manufactured as described in Chapter 7 with samples extracted for mechanical testing as described in Chapter 8. The results of individual tests are presented in Appendix C. In Table 9.25, four different treatments results are shown (by reference to the relative heights of the bar graphs in the above Appendix using three colours: white = low, orange = middle and green = high). In Table 9.26, the data are grouped according to the seven mechanical properties (E₁, E₂, σ'_1 , σ'_2 , ε'_1 , ε'_2 , v_{12}).

The selected four systems using Biotex flax fibre and Huntsman epoxy resin have got the following characteristics:

- 25-untreated fibre/ resin + 1% silane in the hardener
- 26- fibre treatment in 1M NaOH solution for 3h immersion time/ resin + 1% silane in the hardener
- 27- fibre treatment in 1% silane solution / resin
- 28- fibre treatment in 1M NaOH solution for 3h immersion time + 1%silane solution/ resin

| | 25 Biotex/Huntsman Silane in hardener | 26 Biotex 3h 1M NaOH fibre treatment/Huntsnan- Silane in Hardener | 27 Biotex 1% silane fibre treatment/Huntsman | 28 Biotex 3h 1M NaOH + 1% silane fibre treatment/Huntsman |
|-----------------|---------------------------------------------|----------------------------------------------------------------------------|----------------------------------------------------|--------------------------------------------------------------------|
| E1 p57 | Н | М | Μ | M |
| E2 p60 | L | Н | М | M |
| S1 p58 | Н | L | Μ | L |
| S2 p61 | Μ | L | M | Н |
| ε 1 p58 | L | н | М | L |
| ε 2 p61 | L | L | н | н |
| v 12 p59 | М | Н | L | Н |

Table 9.25: Silanisation level effect is colour coded

Table 9.26: Seven mechanical properties

| | 25 Biotex/Huntsman Silane in hardener | 26 Biotex 3h 1M NaOH fibre treatment/Huntsman Silane in Hardener | 27 Biotex 1% silane fibre treatment/Huntsman | 28 Biotex 3h 1M NaOH + 1% silane fibre treatment/Huntsman |
|-----------------|---------------------------------------------|------------------------------------------------------------------------------|-------------------------------------------------------|--------------------------------------------------------------------|
| E1 p57 | 13,08 | 10,08 | 10,05 | 10,88 |
| E2 p60 | 3,66 | 4,08 | 3,72 | 3,87 |
| S1 p58 | 136,46 | 105,24 | 120,50 | 95,49 |
| S2 p61 | 16,65 | 14,11 | 17,53 | 19,28 |
| ε 1 p58 | 1,29 | 1,92 | 1,57 | 1,44 |
| ε 2 p61 | 0,36 | 0,33 | 0,44 | 0,50 |
| v 12 p59 | 0,52 | 0,73 | 0,31 | 0,68 |

Note that v_{12} data is included in the table but is not considered in the evaluation since Poisson's ratio(s) is (are) not considered to be a "performance" parameter in the same way as the other six properties.

Table 9.25 has

- 2 green cells for 1% silane in the hardener,
- 3 green cells for 1M NaOH+1% silane in the hardener,
- 3 green cells for 1M NaOH+1% silane in the fibre,
- 5 orange cells for 1% silane in the fibre, and
- 2 orange cells for 1% silane in the hardener and for 1M NaOH+1% silane in the fibre.

Panels 25, 26 and 27 had the greatest proportion of white (low) cells, indicating this case as a nonoptimal treatment.

The analysis in Table 9.25 considers only relative values of the respective mechanical properties. Table 9.26 presents the analysis for absolute properties. In regards to the interfacial properties, the 1M NaOH mercerisation followed by 1% silane treatment increases the transverse strength σ_2 and strain ϵ_2 , keeping the modulus E_2 in an acceptable value. In contrast, the longitudinal properties E_1 , σ'_1 , ϵ'_1 are clearly damaged by the chemical treatment. Similarly, when the fibres were treated with 1% silane, the longitudinal properties E_1 , σ'_1 , ϵ'_1 , were also reduced, but at lower scale; while the transverse properties E_2 , σ'_2 , ϵ'_2 , were slightly improved. These two systems comparison suggests that flax fibre silane treatment reduces the longitudinal properties while the 1M NaOH 3h + 1% silane treatments works to increase fibre-matrix adhesion.

When 1% **silane was added to the hardener, the highest E₁ and \sigma'_1 values** were achieved. In contrast, E₂, σ'_2 , ε'_2 values were reduced in comparison with the flax fibre direct silane treatment. When the fibre was mercerised and silane added to the hardener, longitudinal and transverse properties clearly decreased. This decrement suggested that the mercerisation process reduced the longitudinal properties; and the silane was not reacting with the mercerised flax surface when dissolved in the hardener.

In Table 9.27, all the values have been converted to percentages relative to the highest value achieved on each row (**hence one of the four values must be 100%**). The final row of the table gives the sum of the values in each column to identify the treatment with the best performance. This initial analysis indicates that the performance of each of the systems is broadly identical. For this reason, a new classification scheme was introduced to give weight to the importance of each property according to its relevance to composite design.

| | 25 | 26 | 27 | 28 |
|------------------|--------------------|---------------------|--------------------|---------------------|
| Percentage | Biotex/Huntsman | Biotex 3h 1M NaOH | Biotex 1% silane | Biotex 3h 1M NaOH + |
| (%) | Silane in hardener | fibre | fibre | 1% silane fibre |
| (/0) | | treatment/Huntsman- | treatment/Huntsman | treatment/Huntsman |
| | | Silane in Hardener | | |
| E1 p57 | 100,00 | 77,04 | 76,80 | 83,14 |
| E2 p60 | 89,71 | 100,00 | 91,27 | 94,92 |
| S1 p58 | 100,00 | 77,12 | 88,30 | 69,98 |
| S2 p61 | 86,34 | 84,77 | 90,92 | 100,00 |
| ε 1 p58 | 67,13 | 100,00 | 81,68 | 74,93 |
| ε 2 p61 | 71,85 | 65,44 | 87,83 | 100,00 |
| v 12 p59* | 71,75 | 100,00 | 43,03 | 93,66 |
| Total | 515,03 | 504,38 | 516,81 | 522,96 |

Table 9.27: Seven mechanical properties **percentage**

* v₁₂ is not considered in the subtotal value

Note: v_{12} was not considered in the subtotal value because the highest value was not considered as the best.

For orthotropic materials, elastic moduli values (E_1 , E_2 and v_{12}) are fundamental in the design process based in the rigidity matrix (Equation 9.3).

$$\begin{cases} \varepsilon_1 \\ \varepsilon_2 \\ \gamma_{12} \end{cases} = \begin{bmatrix} \frac{1}{E_1} & -\frac{\vartheta_{12}}{E_2} & 0 \\ -\frac{\vartheta_{12}}{E_2} & \frac{1}{E_2} & 0 \\ 0 & 0 & \frac{1}{G_{12}} \end{bmatrix} \begin{pmatrix} \sigma_1 \\ \sigma_2 \\ \sigma_3 \end{pmatrix}$$

Equation 9.3

Additionally there are different composite failure criteria. Puck says that the laminate can fail because of Fibre Failure (FF) and Inter Fibre Failure (IFF). The criteria distinguish between FF and IFF; and in the IFF there are three modes of failure, giving the angle of the crack in reference to the laminate plane. Depending on the IFF mode, the failure might be critical or acceptable for the design of the composite.

Puck bases the FF criteria on the fact that the fibre failure of a UD lamina under particular tensile conditions will occur when the tensile stress in the fibre σ_{1f} reaches the same level of a uniaxial tensile strength σ_{1t} or uniaxial compression strength σ_{1c} . The criteria are shown in the Equation 9.4:

$$f_{E,FF} = \frac{1}{\pm R_{II}^{t,c}} \left[\sigma_1 - \left(v_{\perp \parallel} - v_{\perp \parallel f} m_{\sigma f} \frac{E_{\parallel}}{E_{\parallel f}} \right) (\sigma_2 + \sigma_3) \right]$$
 Equation 9.4
$$R_{\parallel}^t for [...] \ge 0$$
$$-R_{\parallel}^c for [...] < 0$$

Finally, since the design idea would be based in the strength values and not in the strain values, the strain would be considered as the weakest factor.

Based on the justification above, the next design factors are considered in Table 9.17:

| Property | Design factor |
|---------------------------------------------------|---------------|
| E ₁ , E ₂ , V ₁₂ | 3 |
| σ ₁ , σ ₂ | 2 |
| ε ₁ , ε ₂ | 1 |

Based in the values in Table 9.28, Table 9.29 was produced. Table 9.27 percentage values were multiplied with the design factors in Table 9.28 in order to get Table 9.29.

| able 9.29: | Seven mechanical p | properties weighted | i percentage (max | 100, 200 | or 300) |
|-------------------|---------------------------------------------|-------------------------------------------------------------------------------|----------------------------------------------------|--------------------------------------------------------------------|-------------------|
| Percentage (%) | 25 Biotex/Huntsman Silane in hardener | 26 Biotex 3h 1M NaOH fibre treatment/Huntsman- Silane in Hardener | 27 Biotex 1% silane fibre treatment/Huntsman | 28 Biotex 3h 1M NaOH + 1% silane fibre treatment/Huntsman | Design factors |
| E1 p57 | 300,00 | 231,12 | 230,41 | 249,43 | 3 |
| E2 p60 | 269,12 | 300,00 | 273,81 | 284,75 | 3 |
| S1 p58 | 200,00 | 154,24 | 176,60 | 139,95 | 2 |
| S2 p61 | 172,69 | 169,55 | 181,84 | 200,00 | 2 |
| ε 1 p58 | 67,13 | 100,00 | 81,68 | 74,93 | 1 |
| ε 2 p61 | 71,85 | 65,44 | 87,83 | 100,00 | 1 |
| v 12 p59* | 215,25 | 300,00 | 129,09 | 280,97 | 3 |
| Total | 1080,78 | 1020,36 | 1032,18 | 1049,06 | |

Table 0.20: Soven mechanical properties weighted percentage (maximum 100 200 -- 200)

* v₁₂ is not considered in the subtotal value

Table 9.29 quantifies different laminate modification effectiveness for the measured mechanical properties; supporting the idea that the **laminate 25 was performing the best**.

In conclusion, for the silane in hardener series of tests, the mechanical properties conclusions were:

- Silane-in-hardener without mercerisation achieves the best composite system, this would be one clear answers for the PhD research question
- The silane reacts with the mercerised surface only when the fibre is directly treated with the silane but not when the silane is added to the hardener
- Mercerisation clearly reduce the composite longitudinal properties -
- Fibre silane treatment slightly reduce the composite mechanical properties -
- All the assumptions were supported with simplistic method in Table 9.29

9.6. BEST COMPOSITE SYSTEM

In Table 9.30, the data are grouped according to the seven mechanical properties (E₁, E₂, σ'_1 , σ'_2 , ϵ'_1 ,

 ε'_{2} , v_{12}). In order to determine the best system from the experimental campaign, the following four combinations were selected:

- 1-Untreated Biotex flax fibre/SuperSap bio-epoxy resin
- 3-Untreated Biotex flax fibre/Huntsman epoxy resin
- 4-Commercially treated Lineo flax fibre/Huntsman epoxy resin
- 25-Untreated Biotex flax fibre/Huntsman epoxy resin + 1% silane in the hardener -

| | 1 Biotex/SuperSap | 3 Biotex/Huntsman | 4 Lineo/Huntsman | 25 Biotex/Huntsman Silane in hardener |
|-----------------|----------------------|----------------------|---------------------|---------------------------------------------|
| E1 p57 | 15,62 | 10,21 | 8,49 | 14,18 |
| E2 p60 | 3,37 | 2,57 | 3,21 | 3,66 |
| S1 p58 | 175,04 | 136,66 | 168,02 | 147,88 |
| S2 p61 | 3,80 | 16,22 | 29,15 | 16,65 |
| ε 1 p58 | 1,36 | 0,96 | 1,52 | 1,40 |
| ε 2 p61 | 0,11 | 0,53 | 0,88 | 0,36 |
| v 12 p59 | 0,44 | 0,17 | 0,25 | 0,56 |

Table 9.30: Seven mechanical properties

Note that v_{12} data is included in the table, but is not considered in the evaluation since Poisson's ratio(s) is (are) not considered to be a "performance" parameter in the same way as the other six properties.

Table 9.30 shows the results for the four best laminates of the experimental campaign. When comparing the 3rd and 25th laminates, untreated flax and 1% silane in the hardener system respectively, it was clearly shown that six mechanical properties out of seven were improved. This supports the utilisation of the silane in the hardener for the enhancement of flax/epoxy composites.

When comparing the Biotex enhanced Laminate 25 system with the Lineo reinforced composite, panel 4, these were the conclusions: The Biotex panel longitudinal rigidity was higher comparing the E_1 and ε'_1 values, being very low E_1 value for Lineo system. In contrast longitudinal strength value was higher for Lineo, suggesting a better fibre-matrix adhesion. In regards to the transverse properties, E_2 and ε'_2 values suggest that the Biotex interface was more rigid/brittle than the Lineo. Additionally, σ'_2 confirms that the Lineo system had stronger fibre-matrix adhesion than the Biotex system. It is true that the patented Lineo fibre treatment is properly increasing the fibre matrix adhesion, however the longitudinal rigidity loss is too pronounced in regards to design issues; for that reason the enhanced Biotex/Huntsman 25th system is preferred at this occasion.

Finally, the 1st laminate manufactured with Biotex and SuperSap resin confirms that high quality composite system can be produced with a Bio-epoxy system. The data suggests that although high quality longitudinal properties are obtained, the fibre/matrix adhesion is too low.

Resuming:

- 1% addition to the hardener clearly improved the unmercerised Biotex system transverse mechanical properties, suggesting a great improvement of the fibre/matrix adhesion. This would be one clear answers for the PhD research question
- Lineo system still has better fibre-matrix interfacial properties than the enhanced Biotex, however longitudinal rigidity properties are drastically reduced with the Lineo's patented chemical treatment
- SuperSap bio-epoxy may be used for high standard composite manufacturing. The PhD experimental campaign was completed with the mechanical characterisation of the Biotex/SuperSap + 1% silane in the hardener optimised system in §9.14

9.7. VOLUME FRACTION VARIATION WITH TREATMENT

In previous sections, the relation between the flax chemical treatment with the fibre swelling and with the V_f variation was evaluated. V_f measurement was determined by CRAG 1000 "Density measurement" method; for all cases, the untreated fabric areal weight was considered, because it did not vary with the chemical treatments. A model was proposed in \$9.4 for the mercerisation process and a tendency described for the silane treatment in the \$9.5, in order to give an answer to the research question.

However, it was not mentioned that each treatment changed the laminate V_f ; i.e. each laminate has a characteristic V_f depending on the received chemical treatment. Although, in this doctoral study, the longitudinal properties were corrected for the best comparison, in the case studies and real manufacturing processes, higher volume fractions are preferred. As a result, ideally good mechanical properties balance must be accompanied with high V_f value. In Table 9.31, the V_f data are shown.

From the experimental campaign, the main conclusion obtained was that the fibre chemical treatment provokes V_f values decrement; the chemical treatment provokes fibre swelling and subsequent laminate thickness increment. Panels from 1 to 3 were manufactured with untreated fibres getting as a result the highest V_f values. When instead of treating fibres the **silane was added to the**

hardener, the V_f reduction was not observed, supporting with another the panel 25 enhanced method.

In section §9.4 the fibre swelling was related to the panel manufacturing flow properties. The fibre swelling must be directly correlated to the laminate thickness, for that reason Figure 9.18 represents the laminate thickness variation with the treatment and compare to the unswollen laminate thickness (Lamina #3#).

| Laminate | Vf longitudinal | Vf transverse |
|------------------------------------------------------------------|-----------------|---------------|
| 1-Biotex/SuperSap | 0,300 | 0,300 |
| 3-Biotex/Huntsman | 0,340 | 0,350 |
| 4-Lineo/Huntsman | 0,369 | 0,352 |
| 5-Biotex 1h 0.25M NaOH treament/Huntsman | 0,255 | 0,266 |
| 6-Biotex 3h 0.25M NaOH treament/Huntsman | 0,252 | 0,241 |
| 7-Biotex 12h 0.25M NaOH treament/Huntsman | 0,248 | 0,251 |
| 8-Biotex 24h 0.25M NaOH treament/Huntsman | 0,244 | 0,246 |
| 9-Biotex 1h 0.50M NaOH treament/Huntsman | 0,268 | 0,266 |
| 10-Biotex 3h 0.50M NaOH treament/Huntsman | 0,280 | 0,277 |
| 11-Biotex 12h 0.50M NaOH treament/Huntsman | 0,285 | 0,273 |
| 12-Biotex 24h 0.50M NaOH treament/Huntsman | 0,272 | 0,259 |
| 13-Biotex 1h 1.00M NaOH treament/Huntsman | 0,236 | 0,233 |
| 14-Biotex 3h 1.00M NaOH treament/Huntsman | 0,230 | 0,230 |
| 15-Biotex 12h 1.00M NaOH treament/Huntsman | 0,226 | 0,242 |
| 16-Biotex 24h 1.00M NaOH treament/Huntsman | 0,229 | 0,224 |
| 17-Biotex 1h 1.50M NaOH treament/Huntsman | 0,238 | 0,245 |
| 18-Biotex 3h 1.50M NaOH treament/Huntsman | 0,259 | 0,250 |
| 19-Biotex 12h 1.50M NaOH treament/Huntsman | 0,247 | 0,238 |
| 20-Biotex 24h 1.50M NaOH treament/Huntsman | 0,227 | 0,219 |
| 21-Biotex 1h 2.00M NaOH treament/Huntsman | 0,255 | 0,251 |
| 22-Biotex 3h 2.00M NaOH treament/Huntsman | 0,228 | 0,235 |
| 23-Biotex 12h 2.00M NaOH treament/Huntsman | 0,243 | 0,243 |
| 24-Biotex 24h 2.00M NaOH treament/Huntsman | 0,235 | 0,244 |
| 25-Biotex/Huntsman 1% silane in hardener | 0,325 | 0,316 |
| 26-Biotex 3h 1.00M NaOH treatment/Huntsman 1% silane in hardener | 0,234 | 0,221 |
| 27-Biotex 1% silane fibre treatment/Huntsman | 0,276 | 0,269 |
| 28-Biotex 3h 1M NaOH + 1% silane fibre treatment/Huntsman | 0,251 | 0,230 |

Table 9.31: Volume fraction values for laminates 1 to 28



Figure 9.18. Laminates thickness values vs the chemical treatment

*NOTE: The values are the average numbers from longitudinal and transverse plates

According to the Figure 9.18, the chemical treatment is provoking a swelling of the fibres and as a result an increment in the laminate thickness. Chapter 8 Table 8.3, shows data for the fibre swelling, however this data is insufficient for a quality analysis. An easy prediction method to determinate the laminate thickness increment and V_f decrement after any treatment, might be to measure the fibre swelling variation.



Figure 9.19. Laminates thickness increment vs the chemical treatment

Additionally Figure 9.19 represents different panels thickness differences in comparison to the unswollen system. For the case of the mercerisation process the values can reach values of 1.55, with the lowest values for the addition of the silane to the hardener and untreated fibres composite system.





When the data is represented the treatment vs. the percentage improvement, see Figure 9.20. This representation shows more definitive results. For the best of the cases just 7% is incremented the laminate thickness, and for the worse of the cases 55%, elevated value.

The reflection is that the fibre quantity (weight or volume) introduced in the system is the same, however the resin amount is varying from one sample to another. The first supposition is that fibre treatment provokes fibre swelling, getting as a result increased laminate thickness, resulting in a lower V_f value.

For the untreated fibre with silane in the hardener, the time for the fibre to swell in the liquid matrix will be limited by the cure time of the resin during plate manufacture, and hence shorter than the swelling duration during optimal mercerisation (albeit that the different liquids may also influence swelling).

9.8. VOLUME FRACTION MEASUREMENT DEVIATION

It was reported in Chapter 8 that the flax fibre swells after chemical treatment. The swelling altered the fabric homogeneity, giving as a result thickness variation along the laminate, subsequently increasing the deviation in the V_f measurement.

In Table 9.32, the V_f data Coefficient of Variation (CV) across all samples are represented, this was done in order to evaluate panel homogeneity.

| Laminate | CV (%) Vf Longitudinal | CV (%) Vf Transverse | |
|------------------------------------------------------------------|------------------------|----------------------|--|
| 3-Biotex/Huntsman | 3,82 | 3,43 | |
| 4-Lineo/Huntsman | 2,71 | 1,99 | |
| 5-Biotex 1h 0.25M NaOH treament/Huntsman | 2,75 | 2,63 | |
| 6-Biotex 3h 0.25M NaOH treament/Huntsman | 4,37 | 2,07 | |
| 7-Biotex 12h 0.25M NaOH treament/Huntsman | 4,03 | 2,79 | |
| 8-Biotex 24h 0.25M NaOH treament/Huntsman | 4,10 | 4,47 | |
| 9-Biotex 1h 0.50M NaOH treament/Huntsman | 2,24 | 3,01 | |
| 10-Biotex 3h 0.50M NaOH treament/Huntsman | 3,21 | 3,61 | |
| 11-Biotex 12h 0.50M NaOH treament/Huntsman | 4,91 | 3,66 | |
| 12-Biotex 24h 0.50M NaOH treament/Huntsman | 6,62 | 3,86 | |
| 13-Biotex 1h 1.00M NaOH treament/Huntsman | 4,24 | 4,77 | |
| 14-Biotex 3h 1.00M NaOH treament/Huntsman | 4,35 | 3,48 | |
| 15-Biotex 12h 1.00M NaOH treament/Huntsman | 3,98 | 3,72 | |
| 16-Biotex 24h 1.00M NaOH treament/Huntsman | 5,68 | 4,46 | |
| 17-Biotex 1h 1.50M NaOH treament/Huntsman | 5,04 | 2,86 | |
| 18-Biotex 3h 1.50M NaOH treament/Huntsman | 5,79 | 5,60 | |
| 19-Biotex 12h 1.50M NaOH treament/Huntsman | 3,24 | 4,20 | |
| 20-Biotex 24h 1.50M NaOH treament/Huntsman | 4,41 | 4,11 | |
| 21-Biotex 1h 2.00M NaOH treament/Huntsman | 4,31 | 2,79 | |
| 22-Biotex 3h 2.00M NaOH treament/Huntsman | 6,14 | 5,11 | |
| 23-Biotex 12h 2.00M NaOH treament/Huntsman | 4,53 | 4,53 | |
| 24-Biotex 24h 2.00M NaOH treament/Huntsman | 4,26 | 3,69 | |
| 25-Biotex/Huntsman 1% silane in hardener | 3,08 | 1,58 | |
| 26-Biotex 3h 1.00M NaOH treatment/Huntsman 1% silane in hardener | 4,70 | 2,71 | |
| 27-Biotex 1% silane fibre treatment/Huntsman | 3,26 | 3,72 | |
| 28-Biotex 3h 1M NaOH + 1% silane fibre treatment/Huntsman | 5,58 | 2,61 | |

Table 9.32: Volume fraction CV percentage for laminates 1 to 28

Although the results are not as clear as in the other Chapter 9 sections, there may be a tendency for lower V_f CV deviations for untreated/low treated fibre systems (from 3 to 9 laminates) and panels manufactured with whole flow mesh plus VAP membrane infusion strategy (from 25 to 28 laminates).



Figure 9.21: Longitudinal (Blue) and transverse (Red) V_f measurement CV (%)

Additionally Figure 9.21 graphically represents the CV variation, and compares the values difference between longitudinal and transverse panels data. The bar graphs confirms that in 20 out of 26 (threequarters) cases, the CV values for the longitudinal (Blue) measurements are higher than for the transverse (Red); this phenomenon suggests that transverse panels are more homogeneous than the longitudinal, supporting the "laminate position effect" studied in §9.10.

9.9. PANEL MANUFACTURING HOMOGENEITY

Table 9.33 resumes mechanical test standard deviations grouped according to the seven mechanical properties (E₁, E₂, σ'_1 , σ'_2 , ϵ'_1 , ϵ'_2 , ν_{12}).

| SD | F1 (kN) | E1 (GPa) | σı(MPa) | ε1(%) | V12 | F2(kN) | E ₂ (GPa) | σ ₂ (MPa) | ε2(%) |
|------------------------------------------------------------------|---------|----------|---------|-------|------|--------|----------------------|----------------------|-------|
| 3-Biotex/Huntsman | 0,45 | 0,59 | 7,76 | 0,15 | 0,02 | 0,05 | 0,19 | 0,84 | 0,08 |
| 4-Lineo/Huntsman | 0,20 | 0,32 | 8,37 | 0,08 | 0,07 | 0,06 | 0,23 | 1,01 | 0,08 |
| 5-Biotex 1h 0.25M NaOH treament/Huntsman | 0,23 | 0,83 | 6,20 | 0,13 | 0,08 | 0,16 | 0,38 | 2,62 | 0,08 |
| 6-Biotex 3h 0.25M NaOH treament/Huntsman | 0,38 | 0,93 | 6,82 | 0,12 | 0,04 | 0,13 | 0,45 | 1,80 | 0,21 |
| 7-Biotex 12h 0.25M NaOH treament/Huntsman | 0,18 | 0,82 | 5,07 | 0,09 | 0,08 | 0,21 | 0,34 | 3,23 | 0,06 |
| 8-Biotex 24h 0.25M NaOH treament/Huntsman | 0,39 | 0,89 | 7,46 | 0,12 | 0,09 | 0,19 | 0,33 | 3,11 | 0,14 |
| 9-Biotex 1h 0.50M NaOH treament/Huntsman | 0,23 | 1,17 | 8,01 | 0,07 | 0,06 | 0,06 | 0,39 | 1,11 | 0,22 |
| 10-Biotex 3h 0.50M NaOH treament/Huntsman | 0,35 | 0,64 | 9,68 | 0,13 | 0,04 | 0,10 | 0,08 | 1,98 | 0,14 |
| 11-Biotex 12h 0.50M NaOH treament/Huntsman | 0,37 | 0,89 | 10,30 | 0,16 | 0,08 | 0,07 | 0,58 | 0,97 | 0,12 |
| 12-Biotex 24h 0.50M NaOH treament/Huntsman | 0,30 | 1,12 | 7,46 | 0,13 | 0,08 | 0,07 | 0,29 | 1,11 | 0,06 |
| 13-Biotex 1h 1.00M NaOH treament/Huntsman | 0,37 | 0,70 | 9,27 | 0,11 | 0,07 | 0,13 | 0,35 | 2,01 | 0,12 |
| 14-Biotex 3h 1.00M NaOH treament/Huntsman | 0,35 | 1,46 | 5,57 | 0,12 | 0,04 | 0,11 | 0,47 | 1,74 | 0,20 |
| 15-Biotex 12h 1.00M NaOH treament/Huntsman | 0,45 | 1,98 | 5,58 | 0,14 | 0,06 | 0,22 | 0,74 | 2,59 | 0,13 |
| 16-Biotex 24h 1.00M NaOH treament/Huntsman | 0,39 | 1,23 | 6,74 | 0,30 | 0,05 | 0,15 | 0,19 | 2,13 | 0,05 |
| 17-Biotex 1h 1.50M NaOH treament/Huntsman | 0,17 | 1,18 | 5,89 | 0,16 | 0,17 | 0,05 | 0,39 | 1,18 | 0,14 |
| 18-Biotex 3h 1.50M NaOH treament/Huntsman | 0,17 | 1,20 | 5,25 | 0,17 | 0,06 | 0,28 | 0,47 | 1,20 | 0,11 |
| 19-Biotex 12h 1.50M NaOH treament/Huntsman | 0,17 | 1,06 | 2,78 | 0,11 | 0,06 | 0,11 | 0,55 | 1,64 | 0,10 |
| 20-Biotex 24h 1.50M NaOH treament/Huntsman | 0,22 | 0,87 | 6,12 | 0,27 | 0,05 | 0,18 | 0,43 | 2,27 | 0,11 |
| 21-Biotex 1h 2.00M NaOH treament/Huntsman | 0,11 | 0,32 | 3,43 | 0,41 | 0,06 | 0,03 | 0,29 | 0,44 | 0,55 |
| 22-Biotex 3h 2.00M NaOH treament/Huntsman | 0,14 | 0,24 | 3,03 | 0,71 | 0,08 | 0,05 | 0,16 | 0,66 | 0,38 |
| 23-Biotex 12h 2.00M NaOH treament/Huntsman | 0,08 | 0,23 | 2,64 | 0,44 | 0,16 | 0,06 | 0,27 | 0,74 | 0,32 |
| 24-Biotex 24h 2.00M NaOH treament/Huntsman | 0,11 | 0,29 | 3,02 | 0,61 | 0,16 | 0,04 | 0,09 | 0,66 | 0,24 |
| 25-Biotex/Huntsman 1% silane in hardener | 0,17 | 2,04 | 4,66 | 0,07 | 0,17 | 0,04 | 0,51 | 0,74 | 0,09 |
| 26-Biotex 3h 1.00M NaOH treatment/Huntsman 1% silane in hardener | 0,15 | 1,19 | 1,88 | 0,12 | 0,14 | 0,07 | 0,45 | 1,08 | 0,05 |
| 27-Biotex 1% silane fibre treatment/Huntsman | 0,15 | 0,68 | 2,74 | 0,08 | 0,07 | 0,08 | 0,54 | 1,58 | 0,07 |
| 28-Biotex 3h 1M NaOH + 1% silane fibre treatment/Huntsman | 0.19 | 0.81 | 5.50 | 0.17 | 0.15 | 0.09 | 0.30 | 1.35 | 0.07 |

Table 9.33: Seven mechanical properties standard deviation for laminates 1 to 28

Laminates 3 and 4 have low deviations because the fibres were untreated, resulting in a more homogeneous reinforcement. In the panels from 5 to 20, there were more red and orange cells because the chemical treatment provoked fibre swelling and inhomogeneous reinforcement. Panels from 21 to 24 were also deeply treated, however since the infusion strategy was modified covering whole panel surface with the flow media, the panel quality subsequently improved, contributing in the green cells achievement.

Finally, 25-28 group panels were manufactured with whole flow media and VAP membrane. This group general tendency was to obtain low CV measurement, especially for the strain values; however, there were slight differences depending on the treatment. It may be assumed that the laminates 25

and 26 where the silane was added to the hardener had more homogeneous values. In contrast, for laminate 27 and 28 where the fibre were directly treated with the silane the CV was higher; getting for the 28th laminate the highest deviation because of the sum of silanisation and mercerisation treatments.

9.10. LAMINATE POSITION EFFECT

In all the laminates, the longitudinal and transverse coupons were obtained from the areas shown in Figure 9.22.



Figure 9.22: Laminates 9-12 samples extraction areas

Based in the Table 9.32, V_f CV values and Figure 9.22, it was assumed that the laminates produced for transverse (T90°) tests were more homogeneous panels; obtaining lower CV values in general. Additionally this homogeneity effect is supported with lower CV in the mechanical tests results (Table 9.33).

It is believed that this effect happens because of these two factors.

- The resin inlet area is always "resin richer" than the rest of the panel, increasing the laminate thickness. This effect led to lower V_f in the first part of the T0° panel, getting as a result inhomogeneous panel
- The panel under the flow mesh and the rest of the panel have different V_f values

This effect was not solved even using the VAP membrane; the general panel homogeneity was increased, however the difference between the T0° and T90° was still obvious.

9.11. METHOD SUSTAINABILITY

The current study has used experimental measurements of a range of mechanical properties to consider the effect of mercerisation in combination with fibre or in-hardener matrix silanisation. In Chapter 4, different literature references and methods have been studied for the execution of the treatments. The main idea was to develop the most sustainable method keeping at the same time the treatment effectiveness while not compromising mechanical performance. These were the sustainable steps in the treatment:

- All the fibres were dried at RT
- The treatments were all applied at RT
- The fibres were not washed after the treatment (the washing was proved not to have an effect)
- Acid neutralisation was avoided
- In the case of the silane addition to the hardener, all solvent consumption was avoided

It was concluded that most of the chemical treatments proposed in Chapter 4 review are neither environmentally nor healthy treatments, at least in regards to the materials selected, see Figure 9.23. However, among all the treatments it is suggested that the mercerisation and silanisation are one of the less hazardous treatments.



Figure 9.23: Figure produced based on the Hazardous Materials Identification System (HMIS)

The data from the experimental campaign suggest that **optimum mechanical performance can be obtained using unmercerised fibre with silane-in-hardener**. The elimination of both the mercerisation and pre-manufacture silanisation processes obviously reduces environmental burdens by not generating pollutant wastes.

9.12. MECHANICAL PROPERTY PREDICTION

The values obtained for the longitudinal modulus and strength should correspond to the predictions from the rules-of-mixture presented by Virk et al. (Virk, Hall & Summerscales, 2012) (Equations 9.5, 9.6):

$$E_c = \kappa \eta_d \eta_l \eta_o V_f E_f + V_m E_m$$
 Equation 9.5

$$\sigma'_{c} = \kappa \sigma'_{f} V_{f} + \sigma_{m^{*}} (1-V_{f})$$
 Equation 9.6

Equation 9.6 is only valid for the axial strength of unidirectional composites although Potter has suggested that, for small misalignments, a correction factor can be included such that the first term of the equation becomes:

$$\sigma'_{c} = \kappa \sigma'_{f} V_{f} \sec^{2} \theta$$
 Equation 9.7

In the above Equations, the nomenclature and data appropriate to each parameter are:

- Ex is the Young's modulus,
- σ'_x is the strength,

 σ'_{f} = 500 MPa from Composite Evolution data

• σ_{m^*} is the stress in the matrix at the failure strain of the fibre,

 σ_{m^*} = 17 MPa from Huntsman data

- V_x is the component volume fraction ($V_f+V_m+V_v = 1$): derived from laminate measurements using CRAG 1000 thickness method
- κ is the fibre area correction factor:

for flax, $\kappa = 1.12$ (Brierley ultimates) (Brierley, 2014), $\kappa = 2.55$ (Thomason et al. technical fibres) (Thomason, Carruthers, Kelly & Johnson, 2011) or $\kappa = 2.70$ (Soatthiyanon technical fibres) (Soatthiyanon, 2014). The boundary between fibre scales may be ultimate fibre "apparent diameter" < 40 μ m or technical fibre "apparent diameter" > 40 μ m.

- η_d is the fibre diameter distribution factor,
- η_l is the fibre length distribution factor,
- η_0 is the fibre orientation distribution factor, and
- subscript *x* being *c*, *f*, *m* or *v* for composite, fibre, matrix or void respectively.

Data for the mechanical properties of fibre or resin were obtained from the reinforcement supplier, Composites Evolution, or the resin supplier, Huntsman, respectively.

Equation 9.5 was used to compare the predicted values for the composite Young's modulus and real values obtained from the investigation.

• Ec is the composite Young's modulus, calculated Ec=17100MPa =17,1GPa
- E_f is the fibre Young's modulus from Composite Evolution, E_f =50000MPa=50GPa
- E_m is the matrix Young's modulus from Huntsman, E_m =3000MPa
- Vf=0.3
- Vm=0.7
- κ assumed 1
- η_d assumed 1
- η_l assumed 1
- η₀ assumed 1



Figure 9.24 shows the results for longitudinal modulus properties.

Figure 9.24: Longitudinal modulus experimental and modelled properties data

Experimental modulus (Figure 9.24) is not expected to exceed theoretical predictions. In order to justify the behaviour, the key assumptions include:

- Supplier data for fibre and matrix properties is optimistic
- Orthotropic, macroscopically homogeneous, linearly elastic materials
- Both the fibre and the matrix are free of voids
- Each lamina has uniform thickness across the layer
- Poisson's strains are neglected
- Fibres are continuous with uniform cross-section (but these are polygonal natural fibres)
- Fibres lie parallel to each other
- Fibres are actually spun at ~30° maximum angle to yarn axis ($\eta_0 = \cos^4\theta = 0.5625$)
- Perfect bonding between the fibre and the matrix and hence Cox shear-lag theory applies

The strength (Figure 9.25) was predicted with the Equation 9.6, using:

- σ'_c is the composite strength, $\sigma'_c = 158.75$ MPa
- σ'_{f} is the fibre strength, σ'_{f} = 500 MPa from Composite Evolution data
- σ_{m^*} is the stress in the matrix at the failure strain of the fibre. The fibre strain at failure is $\epsilon_{f}^*=1.25\%$ and for the matrix $\epsilon_{m}^*=5.0\%$. For that reason, the 50MPa strength of the matrix is recalculated to **12.5MPa** at strain of fibre failure.
- V_f=0.3
- V_m=0.7
- κ assumed 1



(blue is experimental data; orange line is the extended rule-of-mixture prediction)

Experimental strengths are not expected to exceed theoretical predictions

- Fibre volume fractions are not consistent at 30%
- Virk (Virk et al., 2012) reported ~100 MPa for jute quasi-unidirectional NFRP, but jute is normally considered to be weaker than flax
- Virk (Virk et al., 2012) jute fibre composite strengths with CV 40%

Failure to achieve the predicted strength may arise from:

- Poor fibre-matrix adhesion
- Defects and voids in composite
- Fibre misalignment

9.13. WATER RESISTANCE TESTS

The #33 system was produced with the optimised combination, silanisation process adding silane to the hardener in the composite production. The silanised system was compared to the basic unmodified system, #34. Both composite systems were aged in water at same conditions. Figure 9.26 shows the water absorption percentage evolution against time in days. The samples were degraded for **81 days** under the same conditions, and the conclusions obtained from the study were that there is no great different between the two systems. However, when the samples were dried at RT during 19 days, the conclusion was that the silanised system (33) had absorbed marginally less than the base system (34).



Figure 9.26: Silanised (blue) and not silanised (red) systems water absorption vs time (days)

Although the longitudinal flexural test is not the most representative for the interfacial properties evaluation; this test was selected because the samples were prepared and available to be degraded in the immersion test; additionally as was observed for the tensile or compression tests, the silane is improving the composite longitudinal properties. For that reason, to complete the study, transverse flexural samples were degraded for **ten days** in water at RT then tested. The resulting data is shown in Table 9.34.

Table 9.34: Flexural properties comparison for silane and untreated systems

| | | | • | | | | |
|-------------|--------------------|---------------|---------------------------------------|-----------|-----------------|------------|-------------|
| | | Initial flexu | exural propeties Degraded flexural pr | | ural properties | Percentage | |
| | | Force (N) | Modulus (N/mm2) | Force (N) | Modulus (N/mm2) | Force (%) | Modulus (%) |
| Logitudinal | Silane in hardener | 103,12 | 18900 | 65,77 | 7877 | 36,22 | 58,32 |
| Logituumai | No treated | 84,47 | 16222 | 62,28 | 12483 | 26,27 | 23,05 |
| Transverse | Silane in hardener | 23,00 | 5720 | 15,03 | 3700 | 34,65 | 35,31 |
| Transverse | No treated | 21,70 | 5620 | 12,10 | 2985 | 44,24 | 46,89 |

For the case of the longitudinal tests, it was observed that the silanised system flexural initial performance is greater, in both, force at failure and modulus. However, when degraded it appears that the silanised system is performing worse that the basic system; especially for the modulus data, degrading up to 58,32%. In contrast, when comparing the transverse properties, the silanised system is performing better than the basic one.

The silanised system interfacial properties look to be stronger that the untreated system. However, in the mechanical properties comparison they are not as strong as expected. The selected flexural test is not the most appropriate; the transverse tensile test would be a better alternative.

Apart from this, it is believed that the silane addition to the system will improve the fatigue properties, but this was outside the scope and duration of the PhD study.

9.14. OPTIMISED SYSTEM PROPERTIES

It was determined that an optimised system (balance of mechanical properties, composite manufacture times and minimisation of environmental burdens) treatment would be to add 1% silane to the hardener and use untreated fibres. To work with the most sustainable combination, **Biotex/SuperSap + 1% silane in the hardener optimised system was selected**. The selected system was consciously studied in order to obtain the most reliable design properties from the mechanical tests; for example the optimum laminates were manufactured with VAP membrane and the strain values during the test were measured with strain gauges. For the complete characterisation of the composite systems, the following tests were performed:

- Tensile test with gauges
- Compression with gauges
- Shear with gauges
- Flexural test

The mechanical properties were obtained in order to perform the case studies design in §9.16.1, and in order to make a comparative study with the basic system, described in the following lines.

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From all the tests it was concluded that the silanised system is performing better that the untreated system, suggesting an interfacial properties improvement between the fibre and the resin. This might be possible because the silane migration to the interfacial region and creation of new bond between the fibre surface and epoxy active groups.

9.14.1. Tensile test

Two laminates were manufactured in order to get the most reliable properties measurements:

- 29-Biotex fabric/Huntsman resin
- 31-Biotex fabric/SuperSap bio-epoxy resin + 1% silane added to the hardener

Laminate 29 was manufactured in order to get reliable base properties for the best comparison to the enhanced system, Laminate 31 (VAP membrane, and silane added to the bio-epoxy system hardener). Both systems were characterised with strain gauges. Results are shown in Table 9.35.

| System | 1 Biotex/SuperSap No treated | 3 Biotex/Huntsman No treated | 29 Biotex/Huntsman No treated+VAP | 31 Biote/SuperSap Silane in hardener+VAP | 4 Lineo/Huntsman No treated | 25 Biotex/Huntsman Silane in hardener |
|-----------------|------------------------------------|------------------------------------|-----------------------------------------|---------------------------------------------------|-----------------------------------|---------------------------------------------|
| E1 | 15,62 | 10,21 | 10,95 | 13,82 | 8,49 | 14,18 |
| E ₂ | 3,37 | 2,57 | 2,07 | 2,94 | 3,21 | 3,66 |
| S ₁ | 175,04 | 136,66 | 152,25 | 153,46 | 168,02 | 147,88 |
| S ₂ | 3,80 | 16,22 | 20,41 | 18,54 | 29,15 | 16,65 |
| ε ₁ | 1,36 | 0,96 | 1,59 | 1,51 | 1,52 | 1,40 |
| ε2 | 0,11 | 0,53 | 0,53 | 0,47 | 0,88 | 0,36 |
| v ₁₂ | 0,44 | 0,17 | 0,40 | 0,52 | 0,25 | 0,56 |

Table 9.35. Comparative table with the experimental campaign best systems

Apart from the 29 and 31 systems data, the balance of the data was added in order to have overall view of the whole experimental campaign. The main conclusions were listed in the following points:

- **New VAP** has been used for the manufacturing
- Silane generally has better strength values
- SuperSap generally better than Huntsman
- Slight penalisation in moduli values

9.14.2. Compression test

Both systems compression properties were tested using strain gauges, and the obtained properties were compared in Table 9.36.

- 31-Biotex fabric/SuperSap bio-epoxy resin + 1% silane added to the hardener

30-Biotex fabric/ SuperSap bio-epoxy resin

-

| System | 31 Biotex/SuperSap Silane in hardener+VAP | 30 Biotex/SuperSap No treated+VAP |
|-----------------|----------------------------------------------------|-----------------------------------------|
| E _{c1} | 11,50 | 10,50 |
| E _{c2} | 2,60 | 3,10 |
| S _{c1} | 110,00 | 107,00 |
| S _{c2} | 91,00 | 86,00 |
| ε _{c1} | 4,30 | 3,50 |
| ε _{c2} | 4,20 | 4,35 |

Table 9.36: Silanised and untreated systems compression properties comparison

The conclusion was that the mechanical properties are generally improved adding silane to the hardener. The exception was that the transverse modulus was higher for the case of the untreated system; however in regards to design question strength values are more important than modulus values, especially for the case of the transverse properties. Similarly, the transverse strain properties difference is not considered as substantial.

9.14.3. In-plane shear test

Both systems in-plane shear properties were tested using strain gauges, and the obtained properties were compared in Table 9.37.

- **35-Biotex fabric/SuperSap bio-epoxy resin + 1% silane added to the hardener**
- 36-Biotex fabric/ SuperSap bio-epoxy resin

| System | 35 Biotex/SuperSap Silane in hardener+VAP | 36 Biotex/SuperSap No treated+VAP | |
|------------------------|-------------------------------------------------|-----------------------------------------|--|
| G ₁₂ | 1,64 | 1,54 | |
| τ ₁₂ | 25,60 | 24,90 | |
| γ ₁₂ | 2,90 | 3,80 | |

Table 9.37: Silanised and untreated systems in-plane shear properties comparison

The conclusion is that the silane addition to the hardener is increasing NFRP in-plane shear properties in regards to modulus and strength. It was also seen the in-plane shear properties were decreased.

9.14.4. Interlaminar Shear Strength (ILSS) test

Both systems ILSS properties were tested, and the obtained properties were compared in Table 9.38.

- 31-Biotex fabric/SuperSap bio-epoxy resin + 1% silane added to the hardener
- 30-Biotex fabric/ SuperSap bio-epoxy resin

| | 31 | 30 |
|-----------------|-------------------------------------------|-----------------------------------|
| System | Biotex/SuperSap Silane in hardener+VAP | Biotex/SuperSap No treated+VAP |
| G ₁₂ | 992,00 | 964,00 |

Table 9.38: Silanised and untreated systems ILSS properties comparison

It was proved that the adhesion of the silane to the hardener increases the laminate ILSS values considerably, in this occasion almost 50MPa.

9.14.5. Optimised system flexural test

For tensile, compression, flexural, in-plane and interlaminar shear tests, the silanised system mechanical properties were better than the untreated system. In contrast, according to the §9.13, the interfacial properties against the moisture absorption are not increasing as much as expected.

9. 15. MECHANICAL TESTS UNCERTAINTY ESTIMATION

Every mechanical test has got its own level of uncertainty: it is vital to determine every test uncertainly level in order to determine the **test quality** or reliability. In the Appendix H, mechanical tests uncertainty levels were calculated.

This point has got special importance for the case of the NFRP mechanical properties determination process since the NF inhomogeneity produces a great impact in the tests variability. The question is that the literature is full of NF composites characterisation documents and data, however in general they are not mentioning any quality standard for the test campaign. For that reason, the current PhD obtained reliable mechanical properties for flax/bio-epoxy system, enabling consistent design process with truthful mechanical properties.

Chapter 10: Conclusions

Flax fibre has been mercerised at different immersion times and concentrations conditions, and it was concluded that 3h 1M combination produces the best mechanical performance (§ 9.2). Apart from that, it was concluded that 2M concentrations damage the flax fibre and consequently also decreased the composite performance.

The mercerisation treatment concentration drastically affects the infusion speed, while 0.25M and 0.5M concentrations are not affected, 1M starts to change speed, at 1.5M concentration effect is higher and 2M impedes the infusion process. (§ 9.3). The infusion speed mentioned can be related to the fibre swelling and V_f with a flow model (§ 9.4).

When silane is introduced in the composite system, either in the fibre or in the resin (§ 9.5); Silanein-hardener 1% without mercerisation achieves the best composite system; and the silane reacts with the mercerised surface only when the fibre is directly treated with the silane but not when the silane is added to the hardener

Although optimised system was obtained, Lineo system still has better fibre-matrix interfacial properties than the enhanced Biotex, however longitudinal rigidity properties are drastically reduced with the Lineo's patented chemical treatment, supporting as a result the idea that SuperSap bioepoxy may be used for high standard composite manufacturing.

In regards to the treatments sustainability, according to the chemical products MSDS (Material Safety Data Sheet) of the products, it was concluded that the Mercerisation and silanisation treatments are less polluting than the other treatments reviewed in Chapter 4. (§ 9.11)

In the literature, there are many references where the NFRP systems are mechanically characterised, but most of the tests quality is usually very poor with elevated CV values. One of the current PhD's objectives was to get reliable design properties, and the conclusion is that high quality mechanical properties can be achieved if the process is performed under controlled parameters. (§ 9.15) As a summary the following three conclusions might be assumed as the most important PhD's outcoming conclusions:

- Mercerisation process may work in order to increase flax/epoxy interfacial properties in particular immersion time/concentration combinations, however; mercerisation would definitely reduce NFRP general mechanical performance
- Addition of silane in 1% concentration to the epoxy system would substantially increment flax/epoxy system static mechanical properties, although not as much as expected the moisture resistance properties
- Finally say that it may be manufactured a high standard flax/bio-epoxy composite with the contribution to knowledge developed along the PhD, in regards to process, mechanical testing and fibre treatments developed

Chapter 11: Recommendations and future work

The insights gained during this doctoral study have identified a sensible route to manufacture composites with low environmental burdens yet a reasonable balance of static mechanical properties. Because of lack of time, some experimental work was not completed. For that reason, it would be very interesting to complete the following work in next steps.

The optimised eco-composite system has been developed, mechanically tested and immersion properties studied; however, the immersion tests results were not as definitive as expected, see §9.13. For that reason, it would be very interesting to age again some samples and tests the interfacial evolution using different test couple, for example making **transverse tensile test to the aged samples**. From this test is expected to get more information about the interfacial properties.

Increasing the interfacial properties, it is also expected that the **fatigue resistance properties** would be improved. There are many references in the literature where the fatigue property enhancement is reported for NFRP.

Completing these two experiments the mechanical properties improvement would be confirmed.

Related to the system optimization (§9.14), it would be interesting to test different silanes to be added to the hardener to get most from the selected flax/bio-epoxy. **Different silanes structures** might perform differently an increase the composite interfacial properties due to the increment of the covalent bonds.

Apart from the laboratory scale tests it would be very interesting to scale up the knowledge achieved at laboratory scale into the real production.

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Appendix A

APPENDIX A1



Biotex Flax Yarn Technical Data Sheet March 2012

Introduction

Biotex Flax Yarn is a continuous reinforcement yarn based on natural flax fibre and designed for fibrereinforced polymer composite applications. The yarn can be used directly in processes include filament winding and pultrusion or can be converted into woven or non-woven textiles. The yarn can be processed in a similar way to glass or carbon fibre.

Biotex natural reinforcements and intermediates provide the high performance and easy processing normally associated with glass fibre composites but with lower weight and environmental impact. They are suitable for semi-structural and decorative applications in sectors such as automotive, construction, marine, sports and consumer goods. Biotex uses a unique Twistless Technology to ensure a high degree of fibre alignment, impregnation and performance.

Linear Density

250tex Standard 250-2000tex On request

Fibre Properties

Typical average properties for flax fibres:

| Flax fibre properties* | |
|------------------------|----------------------|
| Density | 1.5g/cm ³ |
| Diameter** | 20µm |
| Tensile modulus | 50GPa |
| Tensile strength | 500MPa |
| Strain | 2% |

*Flax fibre is a natural product and a certain amount of variation should be expected. **Flax fibre has a non-circular cross-section.

Yarn Properties

| Flax yarn properties | 250tex | 1000tex |
|---------------------------|--------|---------|
| Tenacity/tensile strength | 15N | 38N |
| Elongation at break | 15.5% | 4.4% |



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Composite Properties

| Property | UD flax- polyester laminate (0 dir) | Biaxial flax- polyester laminate (0 dir) | Woven flax- polyester laminate (0 dir) | Test Method |
|--------------------|----------------------------------------------|---------------------------------------------------|-------------------------------------------------|-------------|
| Density | 1.30g/cm ³ | 1.30g/cm ³ | 1.24g/cm ³ | |
| Tensile modulus | 18.8GPa | 8.7GPa | 7.2GPa | ISO 527-4 |
| Tensile strength | 174MPa | 85MPa | 68.3MPa | ISO 527-4 |
| Tensile elongation | 1.5% | 1.7% | 2.5% | ISO 527-4 |
| Flexural modulus | 15.1GPa | 6.8GPa | 4.0GPa | ISO 14125 |
| Flexural strength | 196MPa | 135MPa | 97.4MPa | ISO 14125 |
| Charpy impact | TBC | TBC | 28.0kJ/m ² | ISO 179-1 U |

Data for laminates made from 30-33vol% Biotex flax fabrics and unsaturated polyester resin by the vacuum infusion process and tested at ambient temperature. All fabrics made from 250tex Biotex flax yarn.

Sizing

Biotex Flax Yarn is supplied unsized as standard. Sizings are possible on request.

Packaging

Standard package options are shown in the table. Other packaging is possible on request. Packages are typically wrapped in plastic and delivered in cardboard cartons.

| Package type | Length (mm) | Internal diameter (mm) | Base (mm) | Tip (mm) | Standard weight (kg) | Possible weight range (kg) |
|----------------|----------------|------------------------------|-----------|----------|-------------------------|----------------------------------|
| Cardboard cone | 228 | N/A | 62 | 25 | 5 | 1-6 |
| Cardboard tube | 228 | 76 | N/A | N/A | 5 | 1-6 |

Storage

Biotex Flax yarn should be stored in a cool dry place away from direct sunlight. Flax fibre can absorb moisture from the atmosphere so drying may be required before use, especially if exposed to excessive humidity.

Safety

Flax fibre is a naturally occurring, non-hazardous material, but typical precautions should be taken when handling the material including using appropriate PPE and adequate ventilation. See MSDS for details.

Disclaimer

The information provided here is believed to be accurate but should be considered indicative only. It is the responsibility of the customer to check the suitability of the product for their specific application prior to use.



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APPENDIX A2



TECHNICAL DATA SHEET - FLAXTAPE™

Products:

FlaxTape[™] is the brand used for our patented tapes of unidirectional flax fibers. FlaxTape[™] is ready to be used for all conventional processes (infusion, RTM, film stacking, prepreg...)







Pictures of FlaxTape[™] production

| Description | Weight Wi | |
|------------------------------|-------------------------------------------------------------------------------------------------------------|------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| | x 50gr of flax/m ² 40 0 70gr of flax/m ² 40 0 110gr of flax/m ² 40 0 | |
| Tapes of unidirectional flax | | |
| fibers | | |
| | 200gr of flax/m ² | 40 cm |
| | Description Tapes of unidirectional flax fibers | Description Weight Tapes of unidirectional flax fibers 50gr of flax/m² 70gr of flax/m² 110gr of flax/m² 200gr of flax/m² 200gr of flax/m² |

| Main markets: | | Advantages | |
|-------------------|-------------------------------------------------------------------------|---------------------------------------------------------------------------|--|
| Sport and leisure | rackets, bicycle frames, skis, boards, | Dampening properties | |
| Transportation | Already used in aeronautic, automotive, boat manufacturing, railway. | Weight reduction, Mechanical & Acoustic properties, Bio-based material | |

Mechanical properties:

Composite made with 12 layers of "FlaxTape™ 110" and an epoxy resin processed by RTM:

| RATE OF FIBRES | By Volume | 50% | |
|-----------------|------------------------|---------|--|
| TRACTION | Modulus | 35 GPa | |
| (ISO E27) | Tensile strength | 365 MPa | |
| (130 327) | Failure Strain | 1,35 % | |
| C DVION | Modulus | 31 GPa | |
| FLEXION | Max Strength | 294 MPa | |
| (150 14 125) | Failure Strain | 2,6 % | |
| THEORIC DENSITY | 1.31gr/cm ³ | | |

Composite made with 12 layers of "FlaxTape™ 110" and 20 PP films by film stacking and compression moulding:

| RATE OF FIBRES | By Volume | 43% | |
|-----------------|------------------|------------------------|--|
| TRACTION | Modulus | 33 GPa | |
| (ISO 527) | Tensile strength | 275 MPa | |
| | Failure Strain | 1,22% | |
| FIFMON | Modulus | 23 GPa | |
| FLEXION | Max Strength | 210 MPa | |
| (ISO 14 125) | Failure Strain | 2,1 % | |
| THEORIC DENSITY | | 1.13gr/cm ³ | |

Damping properties:

Low frequency dampening (flexion/mode 2):

| Product | Dampening ratio |
|-----------|-----------------|
| UD Flax | 1.47% |
| UD Carbon | 0.18% |
| UD Glass | 0.15% |



<u>Contact:</u> François Vanfleteren – <u>sales@lineo.eu</u> - +33.232.43.13.67 TDS FlaxTape™

APPENDIX A3



Technical Data Sheet

SUPER SAP® CLR System Clear, UV Stable Epoxy Resin for high color work laminations, coatings and marine epoxy applications.

Product Overview SUPER SAP® CLR Epoxy Resin is our flagship, high-performance clear coating epoxy resin system. The CLR System is a water clear, UV stabilized epoxy system for applications that require a low color, low yellowing epoxy resin such as high color applications. This system features a quick air-release and a world-class UV resistance package. SUPER SAP® CLR works with multiple hardeners to match your working parameters. It has an ideal viscosity for a wide range of applications that use hand layup techniques with fast room temperature cures. Super Sap – Make Things Better.

| | CLX | CLF | CLS |
|----------------------------------|-----------------------|-----------------------|------------------------|
| MECHANICAL DATA | EXTRA FAST | FAST | SLOW |
| Tensile Modulus (ASTM D638) | 450,000 psi (3.1 GPa) | 440,000 psi (3.0 GPa) | 468,000 psi (3.2 GPa) |
| Tensile Strength (ASTM D638) | 9,500 psi (65.5 MPa) | 9,500 psi (65.5 MPa) | 9,800 psi (67.6 MPa) |
| Elongation (ASTM D638) | 6% | 5% | 6% |
| Flexural Modulus (ASTM D790) | 440,000 psi (3. GPa) | 440,000 psi (3. GPa) | 430,000 psi (3. gpa) |
| Flexural Strength (ASTM D790) | 14,000 psi (96.5 MPa) | 13,500 psi (93.1 MPa) | 14,580 psi (100.5 MPa) |
| Compression Strength (ASTM D695) | 11,330 psi (78.1 MPa) | 11,330 psi (78.1 MPa) | 12,520 psi (86.3 MPa) |
| Tg Ultimate (DSC, midpoint) | 170°F/62°C | 149°F/51°C | 171°F/63°C |
| Hardness (Shore D) | 70-80 | 70-80 | 70-80 |

| | | | |
|-----|------|------|--|
| PRO | LE33 | DAIA | |

| Mix Ratio (by volume) | 2:1 | 2:1 | 2:1 |
|--------------------------------------------------|-------------------------------|-------------------------------|----------------------------------------------|
| Mix Ratio (by weight) | 100:47 | 100:47 | 100:47 |
| Viscocity (A/B/Mixed @ 77 *F/25 *C) | 1850/100/580 | 1850/265/725 | 1850/500/800 |
| Component Density (specific density @ 77*F/25*C) | 1.12 (epoxy), 1.02 (hardener) | 1.12 (epoxy), 1.01 (hardener) | 1.12 (epoxy), 0.98 (hardener) |
| Mixed Density (specific density ∈ 77"F/25"C) | 1.09 | 1.08 | 1.08 |
| Pot Life (@ 77"F/25"C) | 18 min | 21 min | 43 min |
| Tack Free Time (@ 95°F/35°C) | 2 hrs | 4 hrs | 8 hrs |
| Recommended Full Cure | 7 days @ 77°F/25°C | 7 days @ 77°F/25°C | 7 days @ 77°F/25°C, Post cure recommended |

ENVIRONMENT DATA

| VOC Content (ASTM D2369) | 0.16 lbs/gal (19.5 g/L) | 0.26 lbs/gal (31.5 g/L) | 0.13 lbs/gai (16.1 g/L) | |
|--------------------------------------|-------------------------|-------------------------|-------------------------|--|
| Biobased Carbon Content (ASTM D6866) | 21% | 21% | 29% | |

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Advanced Materials

Araldite[®] LY 1568 CH* Aradur[®] 3489 CH*

STRUCTURAL COMPOSITES

Technical Data Sheet

| APPLICATIONS | Industrial composites | | | | | |
|--------------|--------------------------------------------------------------------------------------------------------------------------------------------|---------------------------------------------------------------------------------------------------|--------------------------------------|--|--|--|
| PROPERTIES | Laminating system with low viscosity. | | | | | |
| | The long pot life of the system facilit parts. | ates the production of very | y large industrial | | | |
| PROCESSING | Infusion Resin Transfer Moulding Wet lay-up Filament Winding | | | | | |
| KEY DATA | Resin Araldite [®] LY 1568 CH | | | | | |
| | Aspect (visual) | clear liquid | | | | |
| | Viscosity at 25 °C (ISO 12058-1) | 1300 - 1500 | [mPa.s] | | | |
| | Density at 25 °C (ISO 1675) | 1.1-1.2 | [g/cm ³] | | | |
| | Storage temperature (see expiry date on original container) | 2 - 40 | [°C] | | | |
| | Hardener Aradur [®] 3489 CH | | | | | |
| | Aspect (visual) | clear liquid | | | | |
| | Viscosity at 25 °C (ISO 12058-1) | 5 - 20 | [mPa.s] | | | |
| | Density at 25 °C (ISO 1675) | 0.92-0.93 | [g/cm ³] | | | |
| | Storage temperature (see expiry date on original container) | 2 - 40 | [°C] | | | |
| STORAGE | Provided that Araldite [®] LY 1568 CH and in their original, properly closed con temperatures they will have the shelf live | d Aradur [®] 3489 CH are ston tainers at the above mer es indicated on the labels. | ed in a dry place tioned strorage | | | |

Partly emptied containers should be closed immediately after use.

In addition to the brand name product denomination may show different appendices, which allows us to differentiate between our production sites: a.g., BD = Germany, US = United States, IN = India, CI = China, etc., These appendices are in use on packaging, transport and invoicing Unsummits. Generally the same specifications apply for all versions. Please address any additional need for clarification to the appropriate Huntsman contact.

Araldite[®] LY 1568 CH / Aradur[®] 3489 CH

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Enriching lives through innovation

| PROCESSING DATA | | | | | | |
|----------------------------------------|--------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|--------------------------------------------|-------------------|------------------|--|--|
| MIX RATIO | Components | | Parts by weight | Parts by volume | | |
| | Araldite® LY 1568 CH | | 100 | 100 | | |
| | Aradur" 3489 CH | | 28 | 38 | | |
| | We recommend that the components are weighed with an accurate balance to preven mixing inaccuracies which can affect the properties of the matrix system. The components should be mixed thoroughly to ensure homogeneity. It is important that the side and the bottom of the vessel are incorporated into the mixing process. When precedening largo quantitics of mixture the pot life will decrease due to exothermic reaction. It is advisable to divide large mixes into several smalle | | | | | |
| INITIAL MIX | | PC1 | | ImPa s | | |
| VISCOSITY (ISO 12058-1) | Araldite [®] LY 1568 CH/Aradur [®] 3489 | CH at 25 | | 200 - 300 | | |
| VISCOSITY BUILT UP | 6 | | | | | |
| (Rhemoeter AGR2, plate/plate 25 mm, | 5 | | | • | | |
| 100 rpm) | | | | | | |
| | 4 | | , | • | | |
| | 35 | | | | | |
| | 3 3 | | | | | |
| | 5 25 | • | • • | | | |
| | 54 z | | | | | |
| | | | •• | ×39/G | | |
| | | ×.* | ° | +15% | | |
| | ************************************** | 111*********************************** | | 0.3510 | | |
| | 55550000000000000000000000000000000000 | | | ▲40°G | | |
| | 0 50 YAS | 14n Svin Tiene (misi) | 310 | 257 821 | | |
| POTLIFE | | <i>la</i> 1 | | 6-1-1 | | |
| (Tecam, 23°C, 65 % RH) | Araldite [®] LY 1568 CH/Aradur [®] 3489 | CH 100 | | 850 - 950 | | |
| GEL TIME | | /°G1 | | (min1 | | |
| (Hot plate) | 1.07 [000] | | | | | |
| | Araldite [®] LY 1568 CH/Aradur [®] 3489 | CH at 80 | | 43-46 | | |
| | | at 100 | | 15-10 | | |
| | The values shown are for amall as | at 120 | en sin (hand sur | 7-8 | | |
| | structures the gel time can differ sig fibre content and the laminate thickn | indunis of pure inificantly from these. | ne given values r | lepending on the | | |

Araldite[®] LY 1568 CH / Aradur[®] 3489 CH

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| PROPERTIES OF THE | CURED, NEAT FORMULATION | | |
|---------------------------------------------------|---------------------------------|---------------------|--------------------------------------------------------------|
| GLASS TRANSITION TEMPERATURE (T _G) | Cure: | Т ₆ [°С] | Araldite [®] LY 1568 CH/Aradur [®] 3489 CH |
| (IEC 1006,10 K/min) | 15 h 50 °C | | 61.65 |
| | 6 h 70 °C | | 60-10 |
| | 4 h 80 °C | | /5-/8 |
| | 8 h 80°C | | 76-79 |
| | 5 h 100°C | | 77-80 |
| | | | 79-83 |
| (ISO 527) | | Cure: | 8 h 80 °C |
| | Tensile strength | [MPa] | 67-71 |
| | Elongation at tensile strength | [%] | 4.4-5.0 |
| | Ultimate strength | [MPa] | 54-58 |
| | Ultimate elongation | [%] | 10.5-11 |
| | Tensile modulus | [MPa] | 2850-3000 |
| FLEXURAL TEST (ISO 178) | | Cure: | 8 h 80 °C |
| | Flexural strength | [MPa] | 120-130 |
| | Elongation at flexural strength | [%] | 5.5-6.5 |
| | Ultimate strength | [MPa] | 100-115 |
| | Elexural modulus | [%] | 9-10 |
| | r lova a modulao | [MPa] | 2910-3010 |
| FRACTURE | | Cure: | 8 h 80 °C |
| BEND NOTCH TEST | Fracture toughness Kirc | [MPav | 0.7 - 0.8 |
| (PM 258-0/90) | Fracture energy G _{1C} | ml | 170 - 210 |
| | 0, 10 | $[J/m^2]$ | |

Araldite[®] LY 1568 CH / Aradur[®] 3489 CH

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| HANDLING | | |
|-----------|-------------------------------------------------------------------------------------------------------------------|----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| | Personal hygiene | |
| | Safety precautions at work | place |
| | protective clothing | yes |
| | gloves | essential |
| | arm protectors | recommended when skin contact likely |
| | goggles/safety glasses | yes |
| | Skin protection | |
| | before starting work | Apply barrier cream to exposed skin |
| | after washing | Apply barrier or nourishing cream |
| | Cleansing of contaminated | skin |
| | | Dab off with absorbent paper, wash with warm water and alkali-free soap, then dry with disposable towels. Do not use solvents |
| | Disposal of spillage | |
| | | Soak up with sawdust or cotton waste and deposit in plastic-lined bin |
| | Ventilation | |
| | of workshop | Renew air 3 to 5 times an hour |
| | of workplaces | Exhaust fans. Operatives should avoid inhaling vapours |
| FIRST AID | Contamination of the eyes by flushing with clean, runr consulted. | by resin, hardener or mix should be treated immediately ning water for 10 to 15 minutes. A doctor should then be |
| | Material smeared or spla contaminated area then wa doctor should be consulted clothing should be changed | ashed on the <i>skin</i> should be dabbed off, and the shed and treated with a cleansing cream (see above). A in the event of severe irritation or burns. Contaminated immediately. |
| | Anyone taken ill after inhalir | ng vapours should be moved out of doors immediately. |
| | In all cases of doubt call for | medical assistance. |

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APPENDIX B

1. TEST DEVIATIONS

During the experimental campaign there are some deviations from the expected data from mechanical tests. In the following point those deviations would be clarified.

- For transverse tensile test usually some noise is always detected in the deformation signal, this happens because the interfibre cracking; example Laminate 1 Tensile 90° test or Laminate 25 Tensile 90° test
- 2. Depending on the software configuration, the negative strains might be plot as negative or positive; example **Laminate 2 Compression 0° test**
- 3. Along all the experimental testing campaign it was seen that sometimes the deformation signal initially goes negative and next it changes to positive, e.g. Laminate 3 Tensile 0° test. This happen because of two factors, initial coupon angle and video-extensometry. The testing coupons have got initial curvature, that angle provokes initial negative deformation record by the extensometers. This phenomenon might be solved using more accurate data acquisition camera or testing with strain gauges. Laminate 6 Tensile 90° test example is very exaggerated case, where the laminate initial bending provokes negative records
- 4. Sometimes the video-extensometry stickers release from the testing coupons, getting as a result strange records like in **Laminate 11 Tensile 90° test, samples 6 and 8**.
- 5. When the camera lens is not correctly adjusted the signal line tends to be thicker, example Laminate 21 Tensile 90° test
- 6. Some laminates has been tested with strain gauges in order to obtain more accurate modulus and strain values, example **Laminate 29 Tensile 0° test**

2. MECHANICAL TESTS DATA

Laminates 1 and 2_Supersap CLR/INF-Biotex UD-No treatment

| | Width | Thicknoss | Tensile | Tensile | Tensile | Tensile | Failure | Poisson |
|--------------|-------|-----------|---------|---------------|------------------------|------------------|-----------------|-----------------|
| | width | THICKNESS | load | strength | modulus (Extensometry) | modulus (Gauges) | elongation | coeficient |
| Sample | а | e | F | σ_{xt} | E _{xt} | E _{xt} | ٤ _{xt} | v _{xy} |
| | mm | mm | kN | MPa | GPa | GPa | def (%) | |
| LINO_T0º_P01 | 15,05 | 2,03 | 5,358 | 175,039 | 15,617 | 15,203 | 1,36 | 0,436 |

Laminate 1 Tensile 0° test



Laminate 1 Tensile 90° test

| | Width | Thicknoss | Tensile | Tensile | Tensile | Tensile | Failure | Poisson |
|---------------|--------|-----------|---------|---------------|------------------------|---------------------------|-----------------|-----------------|
| | within | THICKNESS | load | strength | modulus (Extensometry) | modulus (Gauges) | elongation | coeficient |
| Sample | а | e | F | σ_{xt} | E _{xt} | E _{xt} | ٤ _{xt} | v _{xy} |
| | mm | mm | kN | MPa | GPa | GPa | def (%) | |
| LINO_T90º_P01 | 25,13 | 2,01 | 0,192 | 3,809 | 3,37 | Lectura nula por ruido | 0,11 | - |


Laminate 1 In-plane shear 10° test

| | Width | Thickness | In plane shear | In plane shear modulus | Maximum shear elongation |
|----------------------|-------|-----------|-------------------|---------------------------|--------------------------|
| Sample | а | е | IPS | G | γ |
| | mm | mm | MPa | GPa | % |
| E-IPSP01 | 25,81 | 3,78 | 22,650 | 4,409 | 0,51 |
| E-IPSP02 | 25,28 | 3,96 | 21,952 | 6,572 | Gage failure |
| E-IPSP03 | 25,95 | 3,90 | 23,306 | 2,973 | Gage failure |
| E-IPSP04 | 25,01 | 3,89 | 22,957 | 3,861 | 0,28 |
| E-IPSP05 | 25,79 | 3,87 | 21,514 | 10,406 | Gage failure |
| Average | 25,57 | 3,88 | 22,48 | 5,64 | 0,39 |
| Standard deviation | 0,40 | 0,06 | 0,73 | 2,97 | 0,17 |
| cv (%) | 1,57% | 1,63% | 3,26% | 52,69% | 42,34% |
| Characteristic value | | | 20,888 | -0,795 | 0,03 |



Laminate 2 Compression 0° test

| | Width Thickness | | Failure Ioad | Compression strength | Compression modulus | Failure elongation |
|----------------------|-----------------|-------|-----------------|-------------------------|------------------------|-----------------------|
| Sample | а | е | F | σ _{xc} | E _{xc} | ε _{xc} |
| | mm | mm | kN | MPa | GPa | def (%) |
| COP01 | 14,77 | 3,89 | 6,471 | 112,647 | 13,716 | -2,59 |
| COPO2 | 14,77 | 3,86 | 6,231 | 109,223 | 13,002 | -2,84 |
| СОРОЗ | 14,80 | 3,82 | 6,192 | 109,649 | 18,293 | -1,90 |
| COPO4 | 15,13 | 3,85 | 6,416 | 110,210 | 22,017 | -3,01 |
| COP05 | 14,67 | 3,79 | 5,253 | 94,401 | 16,024 | -1,47 |
| Average | 14,83 | 3,84 | 6,11 | 107,226 | 16,610 | -2,36 |
| Standard deviation | 0,18 | 0,04 | 0,49 | 7,29 | 3,67 | 0,65 |
| cv (%) | 1,20% | 0,99% | 8,09% | 6,80% | 22,08% | -27,68% |
| Characteristic value | | | | 89,868 | 7,879 | -0,947 |



Laminate 2 Compression 90° test

| | Width | Thickness | Failure Ioad | Compression strength | Compression modulus | Failure elongation |
|----------------------|-------|-----------|-----------------|-------------------------|------------------------|-----------------------|
| Sample | а | e | F | σ _{γc} | E _{yc} | ε _{yc} |
| | mm | mm | kN | MPa | GPa | def (%) |
| E-C90P01 | 25,95 | 3,96 | 7,201 | 70,029 | 4,010 | -1,99 |
| E-C90P02 | 25,90 | 3,92 | 6,240 | 61,450 | 3,600 | -3,51 |
| E-C90P03 | 25,76 | 3,95 | 7,841 | 77,115 | 3,124 | -2,97 |
| E-C90P04 | 25,68 | 3,93 | 6,789 | 67,220 | 2,834 | -3,32 |
| E-C90P05 | 25,67 | 3,96 | 7,694 | 75,771 | 3,265 | -2,53 |
| Average | 25,79 | 3,94 | 7,15 | 70,317 | 3,367 | -2 <i>,</i> 86 |
| Standard deviation | 0,13 | 0,02 | 0,66 | 6,41 | 0,45 | 0,61 |
| cv (%) | 0,50% | 0,44% | 9,21% | 9,11% | 13,46% | -21,47% |
| Characteristic value | | | | 55,061 | 2,288 | -1,533 |



Laminate 3_Araldite LY 1569 /Aradur 3489-Biotex UD-No treatment

Laminate 3 Tensile 0° test

| | | | | | Longitud | inal tensile te | est | | | Vf | N AF |
|----------------------|---------------|------|-------|------|----------|-----------------|------|------|--------------|------------------------|-------------|
| Cample | ID | а | b | F | Ext | σxt | εxt | vxy | Failura modo | VI | IVII |
| Sample | 10 | mm | mm | kN | Mpa | Mpa | % | | Failure moue | (%) | (%) |
| 1 | 1_Bio_T0_#3# | 2,24 | 17,90 | 6,70 | 12087,10 | 167,11 | 1,24 | 0,21 | Тор | | |
| 2 | 2_Bio_T0_#3# | 2,16 | 14,54 | 5,26 | 11129,20 | 167,63 | 1,21 | 0,22 | Тор | 34 | 41,3 |
| 3 | 3_Bio_T0_#3# | 2,2 | 15,54 | 5,58 | 10662,60 | 163,23 | 1,33 | 0,18 | Тор | | |
| 4 | 6_Bio_T0_#3# | 2,1 | 14,35 | 5,46 | 12128,80 | 181,13 | 1,53 | 0,16 | Bottom | | |
| 5 | 7_Bio_T0_#3# | 2,04 | 14,74 | 5,38 | 11776,20 | 179,08 | 1,55 | 0,20 | Тор | Parameters | Value |
| 6 | 8_Bio_T0_#3# | 2,04 | 14,95 | 5,09 | 11373,50 | 166,86 | 1,66 | 0,18 | Bottom | | |
| 7 | 9_Bio_T0_#3# | 2,06 | 15,13 | 5,68 | 11902,40 | 182,16 | 1,46 | 0,19 | Тор | Testing speed (mm/min) | 2 |
| 8 | 10_Bio_T0_#3# | 2,29 | 14,88 | 5,44 | 11177,10 | 159,69 | 1,52 | 0,18 | Bottom | Extensometry | Video |
| 9 | 11_Bio_T0_#3# | 2,08 | 14,65 | 5,14 | 12559,20 | 168,54 | 1,26 | 0,17 | Bottom | Reference points | Stickers |
| 10 | 12_Bio_T0_#3# | 2,19 | 15,40 | 5,59 | 11067,10 | 165,71 | 1,44 | 0,19 | Bottom | Norma | EN ISO527-5 |
| 11 | 13_Bio_T0_#3# | 2,22 | 15,33 | 5,92 | 12065,70 | 173,98 | 1,22 | 0,22 | Bottom | Precharge (kN) | 5 |
| 12 | 14_Bio_T0_#3# | 2,17 | 15,43 | 6,05 | 10974,30 | 180,81 | 1,40 | 0,17 | Тор | Modulus (ɛ %) | 0,05-0,25 |
| Average value | | 2,15 | 15,24 | 5,61 | 11575,27 | 171,33 | 1,40 | 0,19 | | | |
| Standard deviation | | | | 0,45 | 585,82 | 7,76 | 0,15 | 0,02 | | | |
| CV(%) | | | | 0,08 | 0,05 | 0,05 | 0,11 | 0,10 | | | |
| Characteristic value | | | | | 10333,41 | 154,88 | 1,08 | 0,15 | | | |



Laminate 3 Tensile 90° test



Laminate 4_Araldite LY 1569 /Aradur 3489-Lineo UD-No treatment

Laminate 4 Tensile 0° test



Laminate 4 Tensile 90° test

| | | | | Transver | se tensile test | : | | | Vf | Mf |
|----------------------|---------------|------|-------|----------|-----------------|-------|------|--------------|------------------------|-------------|
| Sampla | ID | а | b | F | Eyt | σyt | εyt | Epiluro modo | VI | IVII |
| Sample | 10 | mm | mm | kN | Mpa | Mpa | % | ranure moue | (%) | (%) |
| 1 | 1_Li_T90_#4# | 2,04 | 24,4 | 1,52 | 2978,59 | 30,51 | 1,10 | Tab | | |
| 2 | 2_Li_T90_#4# | 2,04 | 25,4 | 1,67 | 3092,73 | 32,32 | 1,19 | Middle | 32,5 | 36,8 |
| 3 | 3_Li_T90_#4# | 2,08 | 25,2 | 1,55 | 2880,90 | 29,48 | 1,12 | Middle | | |
| 4 | 4_Li_T90_#4# | 2,06 | 25,3 | 1,60 | 3159,33 | 30,70 | 0,98 | Tab | Doromotors | Value |
| 5 | 5_Li_T90_#4# | 2,11 | 24,81 | 1,62 | 3117,08 | 30,92 | 1,05 | Tab | Parameters | value |
| 6 | 6_Li_T90_#4# | 2,06 | 24,97 | 1,59 | 3114,91 | 30,83 | 1,02 | Tab | Testing speed (mm/min) | 2 |
| 7 | 7_Li_T90_#4# | 2,12 | 24,94 | 1,65 | 3296,60 | 31,17 | 0,98 | Tab | Extensometry | Video |
| 8 | 8_Li_T90_#4# | 2,02 | 24,84 | 1,66 | 3636,07 | 33,08 | 1,01 | Tab | Reference points | Stickers |
| 9 | 9_Li_T90_#4# | 2,11 | 24,9 | 1,70 | 3206,99 | 32,39 | 1,11 | Middle | Norma | EN ISO527-5 |
| 10 | 10_Li_T90_#4# | 2,15 | 25,02 | 1,70 | 3274,36 | 31,66 | 0,98 | Tab | Precharge (kN) | 1 |
| 11 | 11_Li_T90_#4# | 2,13 | 24,96 | 1,66 | 3614,16 | 31,31 | 0,93 | Tab | Modulus (ε%) | 0,05-0,25 |
| Average value | | 2,08 | 24,98 | 1,63 | 3215,61 | 31,31 | 1,04 | | | |
| Standard deviation | | | | 0,06 | 234,96 | 1,01 | 0,08 | | | |
| CV(%) | | | | 0,04 | 0,07 | 0,03 | 0,07 | | | |
| Characteristic value | | | | | 2712,58 | 29,15 | 0,88 | | | |





Laminate 5_Araldite LY 1569 / Aradur 3489-Biotex UD-NaOH 0,25M_1h

Laminate 5 Tensile 0° test



Laminate 5 Tensile 90° test



Laminate 6_Araldite LY 1569 / Aradur 3489-Biotex UD-NaOH 0,25M_3h

Laminate 6 Tensile 0° test



Laminate 6 Tensile 90° test

| | | | | Transver | rse tensile test | | | | Vf | N AF |
|----------------------|---------------|------|---------|----------|------------------|--------------|------|--------------|------------------------|-------------|
| Samplo | ID | а | b F Eyt | σyt | εyt | Epiluro modo | VI | IVII | | |
| Sample | 10 | mm | mm | kN | Мра | Mpa | % | Failure mode | (%) | (%) |
| 1 | 1_Bio_T90_#6# | 3,02 | 22,94 | 1,47 | 3299,92 | 21,21 | 0,38 | Bottom | 24,1 | 30,2 |
| 2 | 2_Bio_T90_#6# | 3,14 | 25,26 | 1,64 | 3261,10 | 20,72 | 0,63 | Middle | Baramotors | Value |
| 3 | 3_Bio_T90_#6# | 3,04 | 25,34 | 1,28 | 3116,33 | 16,60 | 0,35 | Middle | Farameters | value |
| 4 | 4_Bio_T90_#6# | 3,06 | 25,30 | 1,35 | 2531,06 | 17,42 | 0,58 | Middle | Testing speed (mm/min) | 2 |
| 5 | 5_Bio_T90_#6# | 2,98 | 25,98 | 1,48 | 2155,53 | 19,18 | 0,98 | Middle | Extensometry | Video |
| 6 | 6_Bio_T90_#6# | 3,08 | 25,35 | 1,35 | 2096,07 | 17,24 | 0,85 | Middle | Reference points | Stickers |
| 7 | 7_Bio_T90_#6# | 2,96 | 25,41 | 1,45 | 2685,28 | 19,28 | 0,70 | Bottom | Norma | EN ISO527-5 |
| 8 | 8_Bio_T90_#6# | 2,95 | 25,36 | 1,20 | 2949,79 | 16,10 | 0,58 | Middle | Precharge (kN) | 1 |
| 9 | 9_Bio_T90_#6# | 3,12 | 26,92 | 1,49 | 2487,90 | 17,74 | 0,81 | Middle | Modulus (ε %) | 0,05-0,25 |
| Average value | | 3,04 | 25,32 | 1,41 | 2731,44 | 18,39 | 0,65 | | | |
| Standard deviation | | | | 0,13 | 452,28 | 1,80 | 0,21 | | | |
| CV(%) | | | | 0,09 | 0,17 | 0,10 | 0,32 | | | |
| Characteristic value | | | | | 1739,43 | 14,44 | 0,19 | | | |





Laminate 7_Araldite LY 1569 / Aradur 3489-Biotex UD-NaOH 0,25M_12h

Laminate 7 Tensile 0° test



Laminate 7 Tensile 90° test

| | | | | Transve | rse tensile test | | | | Vf | Mf |
|----------------------|---------------|------|-------|---------|------------------|-------|------|--------------|------------------------|-------------|
| Samplo | ID | а | b | F | Eyt | σyt | εyt | Epiluro modo | VI | IVII |
| Sample | | mm | mm | kN | Mpa | Mpa | % | ranule mode | (%) | (%) |
| 1 | 1_Bio_T90_#7# | 3,06 | 23,71 | 0,95 | 1965,90 | 13,07 | 0,65 | Middle | 25,1 | 31,4 |
| 2 | 2_Bio_T90_#7# | 2,94 | 24,89 | 0,96 | 1954,17 | 13,10 | 0,65 | Bottom | Baramotors | Value |
| 3 | 3_Bio_T90_#7# | 2,90 | 24,88 | 0,90 | 1995,96 | 12,51 | 0,58 | Bottom | Falameters | |
| 4 | 4_Bio_T90_#7# | 2,89 | 24,78 | 0,94 | 2244,50 | 13,13 | 0,60 | Middle | Testing speed (mm/min) | 2 |
| 5 | 5_Bio_T90_#7# | 2,97 | 24,92 | 0,92 | 2157,22 | 12,39 | 0,56 | Middle | Extensometry | Video |
| 6 | 6_Bio_T90_#7# | 2,97 | 24,75 | 0,94 | 2003,56 | 12,75 | 0,60 | Middle | Reference points | Stickers |
| 7 | 7_Bio_T90_#7# | 2,85 | 24,84 | 1,19 | 2391,17 | 16,83 | 0,68 | Middle | Norma | EN ISO527-5 |
| 8 | 8_Bio_T90_#7# | 2,77 | 24,90 | 1,41 | 2819,02 | 20,50 | 0,72 | Middle | Precharge (kN) | 1 |
| 9 | 9_Bio_T90_#7# | 2,93 | 24,69 | 1,42 | 2759,98 | 19,63 | 0,70 | Middle | Modulus (ε %) | 0,05-0,25 |
| Average value | | 2,92 | 24,71 | 1,07 | 2254,61 | 14,88 | 0,64 | | | |
| Standard deviation | | | | 0,21 | 336,48 | 3,23 | 0,06 | | | |
| CV(%) | | | | 0,20 | 0,15 | 0,22 | 0,09 | | | |
| Characteristic value | | | | | 1516,60 | 7,78 | 0,51 | | | |





Laminate 8_Araldite LY 1569 / Aradur 3489-Biotex UD-NaOH 0,25M_24h

Laminate 8 Tensile 0° test



Laminate 8 Tensile 90° test



Laminate 9_Araldite LY 1569 / Aradur 3489-Biotex UD-NaOH 0,5M_1h

Laminate 9 Tensile 0° test



Laminate 9 Tensile 90° test

| | Transverse tensile test | | | | | | | | VE | N AF |
|----------------------|-------------------------|------|-------|------|---------|-------|------|--------------|------------------------|-------------|
| Sampla | ID | а | b | F | Eyt | σyt | εyt | Epiluro modo | VI | IVII |
| Sample | 10 | mm | mm | kN | Mpa | Mpa | % | Failure mode | (%) | (%) |
| 1 | 1_Bio_T90_#9# | 2,74 | 22,17 | 1,24 | 3290,28 | 20,48 | 0,56 | Bottom | 26,6 | 33,1 |
| 2 | 2_Bio_T90_#9# | 2,68 | 25,14 | 1,41 | 3091,16 | 20,98 | 0,50 | Bottom | | |
| 3 | 3_Bio_T90_#9# | 2,77 | 25,07 | 1,26 | 3124,89 | 18,09 | 0,57 | Middle | Parameters | Value |
| 4 | 4_Bio_T90_#9# | 2,92 | 25,12 | 1,36 | 2667,47 | 18,50 | 0,69 | Тор | | |
| 5 | 5_Bio_T90_#9# | 2,86 | 25,12 | 1,34 | 2242,68 | 18,70 | 0,97 | Тор | Testing speed (mm/min) | 2 |
| 6 | 6_Bio_T90_#9# | 2,65 | 25,06 | 1,31 | 3252,59 | 19,68 | 0,62 | Middle | Extensometry | Video |
| 7 | 7_Bio_T90_#9# | 2,71 | 25,15 | 1,32 | 2396,06 | 19,35 | 0,83 | Middle | Reference points | Stickers |
| 8 | 8_Bio_T90_#9# | 2,70 | 24,96 | 1,32 | 2507,79 | 19,52 | 0,85 | Middle | Norma | EN ISO527-5 |
| 9 | 9_Bio_T90_#9# | 2,70 | 24,87 | 1,42 | 2543,20 | 21,11 | 0,93 | Bottom | Precharge (kN) | 1 |
| 10 | 10_Bio_T90_#9# | 2,77 | 22,66 | 1,32 | 2435,95 | 21,01 | 1,18 | Middle | Modulus (ε %) | 0,05-0,25 |
| Average value | | 2,75 | 24,53 | 1,33 | 2755,21 | 19,74 | 0,77 | | | |
| Standard deviation | | | | 0,06 | 393,09 | 1,11 | 0,22 | | | |
| CV(%) | | | | 0,04 | 0,14 | 0,06 | 0,28 | | | |
| Characteristic value | | | | | 1904,09 | 17,33 | 0,30 | | | |





Laminate 10_Araldite LY 1569 / Aradur 3489-Biotex UD-NaOH 0,5M_3h

Laminate 10 Tensile 0° test



Laminate 10 Tensile 90° test



Laminate 11_Araldite LY 1569 /Aradur 3489-Biotex UD-NaOH 0,5M_12h

Laminate 11 Tensile 0° test



Laminate 11 Tensile 90° test



Laminate 12_Araldite LY 1569 / Aradur 3489-Biotex UD-NaOH 0,5M_24h

Laminate 12 Tensile 0° test



Laminate 12 Tensile 90° test



4 5

2

Laminate 13_Araldite LY 1569 / Aradur 3489-Biotex UD-NaOH 1M_1h

Laminate 13 Tensile 0° test



Laminate 13 Tensile 90° test



Laminate 14_Araldite LY 1569 / Aradur 3489-Biotex UD-NaOH 1M_3h

Laminate 14 Tensile 0° test



Laminate 14 Tensile 90° test



Laminate 15_Araldite LY 1569 / Aradur 3489-Biotex UD-NaOH 1M_12h

Laminate 15 Tensile 0° test

| | | | | L | ongitudinal te | nsile test | | | | Vf | Mf |
|----------------------|----------------|------|-------|------|----------------|------------|------|------|--------------|------------------------|-------------|
| Samplo | ID | а | b | F | Ext | σxt | εxt | vxy | Failura modo | VI | IVII |
| Sample | 10 | mm | mm | kN | Mpa | Mpa | % | | Failure moue | (%) | (%) |
| 1 | 1_Bio_T0_#15# | 3,36 | 17,22 | 5,33 | 6358,32 | 92,10 | 1,28 | 0,30 | Bottom | | |
| 2 | 2_Bio_T0_#15# | 3,37 | 18,00 | 6,50 | 6959,77 | 107,19 | 1,47 | 0,37 | Bottom | 22,6 | 28,5 |
| 3 | 3_Bio_T0_#15# | 3,26 | 18,19 | 5,49 | 7476,57 | 92,60 | 1,25 | 0,26 | Middle | | |
| 4 | 4_Bio_T0_#15# | 3,40 | 15,03 | 5,19 | 7969,94 | 101,66 | 1,38 | 0,44 | Bottom | | Value |
| 5 | 5_Bio_T0_#15# | 3,25 | 15,17 | 4,75 | 7351,32 | 96,39 | 1,25 | 0,26 | Middle | | |
| 6 | 6_Bio_T0_#15# | 3,27 | 15,15 | 4,80 | 6819,72 | 96,85 | 1,50 | 0,36 | Middle | Parameters | |
| 7 | 7_Bio_T0_#15# | 2,96 | 15,13 | 4,93 | 6794,41 | 110,12 | 1,55 | 0,38 | Bottom | | |
| 8 | 8_Bio_T0_#15# | 3,16 | 15,20 | 4,81 | 7283,29 | 100,15 | 1,43 | 0,31 | Bottom | | |
| 9 | 9_Bio_T0_#15# | 3,13 | 15,44 | 5,07 | 9987,64 | 104,96 | 1,44 | 0,42 | Middle | | |
| 10 | 10_Bio_T0_#15# | 3,15 | 15,28 | 5,14 | 9629,29 | 106,72 | 1,45 | | Bottom | | |
| 11 | 11_Bio_T0_#15# | 3,25 | 15,30 | 4,90 | 9720,20 | 98,46 | 1,27 | 0,35 | Bottom | | |
| 12 | 12_Bio_T0_#15# | 3,20 | 15,21 | 4,85 | 11226,10 | 99,57 | 1,17 | 0,27 | Middle | Testing speed (mm/min) | 2 |
| 13 | 13_Bio_T0_#15# | 3,38 | 15,21 | 5,56 | 10985,80 | 108,24 | 1,34 | 0,41 | Тор | Extensometry | Video |
| 14 | 14_Bio_T0_#15# | 3,35 | 15,25 | 5,34 | 12612,00 | 104,54 | 1,15 | 0,31 | Middle | Reference points | Stickers |
| 15 | 15_Bio_T0_#15# | 3,30 | 15,20 | 4,97 | 10481,90 | 99,07 | 1,06 | | Middle | Standard | EN ISO527-5 |
| Average value | | 3,25 | 15,73 | 5,18 | 8777,08 | 101,24 | 1,33 | 0,34 | | Precharge (kN) | 5 |
| Standard deviation | | | | 0,45 | 1983,73 | 5,58 | 0,14 | 0,06 | | Modulus (ε %) | 0,05-0,25 |
| CV(%) | | | | 0,09 | 0,23 | 0,06 | 0,11 | 0,18 | | | |
| Characteristic value | | | | | 4671,28 | 89,70 | 1,04 | 0,21 | | | |



Laminate 15 Tensile 90° test



Laminate 16_Araldite LY 1569 / Aradur 3489-Biotex UD-NaOH 1M_24h

Laminate 16 Tensile 0° test





Laminate 17_Araldite LY 1569 / Aradur 3489-Biotex UD-NaOH 1.5M_1h

Laminate 17 Tensile 0° test



Laminate 17 Tensile 90° test



Laminate 18_Araldite LY 1569 / Aradur 3489-Biotex UD-NaOH 1.5M_3h

Laminate 18 Tensile 0° test



12

18 18 18 18 18 18 18 18 18 18 18 18 18

3 4 5 6 7 8 9 10 11 12 13 14

123456789101121314



Laminate 19_Araldite LY 1569 /Aradur 3489-Biotex UD-NaOH 1.5M_12h

Laminate 19 Tensile 0° test



Stress (MPa)

10

0 ∟ -0,1

0,2

0,4

0,6

8,0

Strain (%)



1,4

1,6

1.2


Laminate 19 Tensile 90° test

Laminate 20_Araldite LY 1569 / Aradur 3489-Biotex UD-NaOH 1.5M_24h

Laminate 20 Tensile 0° test

| | Longitudinal tensile test | | | | | | | | | Vf | Mf |
|----------------------|---------------------------|------|-------|------|---------|-------|------|------|--------------|------------------------|-------------|
| fample | 10 | а | b | F | Ext | σxt | εxt | vxy | Failura moda | VI | IVII |
| Sample | U | mm | mm | kN | Mpa | Mpa | % | | Failure moue | (%) | (%) |
| 1 | 1_Bio_T0_#20# | 3,10 | 15,59 | 4,24 | 6513,80 | 87,73 | 1,68 | 0,33 | Tab | | |
| 2 | 2_Bio_T0_#20# | 3,02 | 15,56 | 4,50 | 6126,86 | 95,68 | 2,11 | 0,23 | Тор | 22,7 | 28,6 |
| 3 | 3_Bio_T0_#20# | 3,13 | 15,60 | 4,23 | 6093,36 | 86,58 | 1,66 | 0,28 | Bottom | | |
| 4 | 4_Bio_T0_#20# | 3,04 | 15,57 | 4,40 | 6647,37 | 93,00 | 1,83 | 0,38 | Bottom | | |
| 5 | 5_Bio_T0_#20# | 3,22 | 15,59 | 4,56 | 6416,69 | 90,83 | 1,76 | 0,24 | Тор | | |
| 6 | 6_Bio_T0_#20# | 3,25 | 15,70 | 4,31 | 5666,63 | 84,40 | 1,75 | 0,35 | Bottom | | |
| 7 | 7_Bio_T0_#20# | 3,20 | 15,64 | 3,83 | 6736,91 | 76,62 | 1,16 | 0,29 | Tab | Demonstern | Malua |
| 8 | 8_Bio_T0_#20# | 3,33 | 15,69 | 4,37 | 6385,00 | 83,73 | 1,74 | | Tab | Parameters | value |
| 9 | 9_Bio_T0_#20# | 3,46 | 15,67 | 4,33 | 8007,56 | 79,81 | 1,40 | 0,39 | Тор | | |
| 10 | 10_Bio_T0_#20# | 3,49 | 15,63 | 4,35 | 7593,09 | 79,69 | 1,49 | 0,39 | Middle | | |
| 11 | 11_Bio_T0_#20# | 3,37 | 15,66 | 4,21 | 7754,43 | 79,81 | 1,48 | 0,30 | Bottom | | |
| 12 | 12_Bio_T0_#20# | 3,20 | 15,72 | 4,04 | 8032,32 | 80,40 | 1,24 | 0,36 | Bottom | Testing speed (mm/min) | 2 |
| 13 | 13_Bio_T0_#20# | 3,30 | 15,60 | 4,06 | 8214,09 | 78,78 | 1,31 | 0,33 | Tab | Extensometry | Video |
| 14 | 14_Bio_T0_#20# | 3,14 | 15,70 | 4,12 | 8244,93 | 83,59 | 1,36 | 0,36 | Middle | Reference points | Stickers |
| 15 | 15_Bio_T0_#20# | 3,26 | 15,77 | 3,82 | 7477,96 | 74,37 | 1,28 | | Bottom | Standard | EN ISO527-5 |
| Average value | | 3,23 | 15,65 | 4,22 | 7060,73 | 83,67 | 1,55 | 0,33 | | Precharge (kN) | 5 |
| Standard deviation | | | | 0,22 | 874,62 | 6,12 | 0,27 | 0,05 | | Modulus (ε %) | 0,05-0,25 |
| CV(%) | | | | 0,05 | 0,12 | 0,07 | 0,17 | 0,16 | | | |
| Characteristic value | | | | | 5250,50 | 71,00 | 0,99 | 0,21 | | | |





Laminate 21_Araldite LY 1569 / Aradur 3489-Biotex UD-NaOH 2M_1h

Laminate 21 Tensile 0° test



Laminate 21 Tensile 90° test

| | | | | Vf | N AF | | | | | |
|----------------------|---------------------|------|-------|-------|----------|-------|-------|--------------|------------------------|-------------|
| Comolo | ID | а | b | F | Eyt | σyt | εyt | Failura mada | VI | IVII |
| Sample | U | mm | mm | kN | Mpa | Mpa | % | ranure mode | (%) | (%) |
| 1 | 1_Bio_T90_#21# | 3,06 | 25,10 | 0,498 | | 6,490 | | Middle | 25,12 | 30,4 |
| 2 | 2_Bio_T90_#21# | 2,96 | 25,12 | 0,467 | 903,744 | 6,285 | 1,212 | Bottom | Baramotors | Value |
| 3 | 3_Bio_T90_#21# | 2,97 | 25,08 | 0,457 | 820,166 | 6,132 | 1,459 | Tab | Farameters | value |
| 4 | 4_Bio_T90_#21# | 2,94 | 25,24 | 0,439 | 826,064 | 5,912 | 1,274 | Bottom | Testing speed (mm/min) | 2 |
| 5 | 5_Bio_T90_#21# | 2,92 | 25,03 | 0,432 | 791,317 | 5,917 | 1,610 | Bottom | Extensometry | Video |
| 6 | 6_Bio_T90_#21# | 2,80 | 24,87 | 0,471 | 1500,500 | 6,767 | 0,454 | Bottom | Reference points | Stickers |
| 7 | 7_Bio_T90_#21# | 2,84 | 24,74 | 0,496 | 1224,410 | 7,063 | 1,109 | Middle | Norma | EN ISO527-5 |
| 8 | 8_Bio_T90_#21# | 2,90 | 24,88 | 0,418 | 717,076 | 5,787 | 2,315 | Middle | Precharge (kN) | 1 |
| 9 | 9_Bio_T90_#21# | 2,88 | 24,64 | 0,424 | 659,774 | 5,971 | 1,843 | Middle | Modulus (ε %) | 0,05-0,25 |
| Average value | | 2,92 | 24,97 | 0,46 | 930,38 | 6,26 | 1,41 | | | |
| Standard deviation | | | | 0,03 | 286,18 | 0,44 | 0,55 | | | |
| CV(%) | | | | 0,07 | 0,31 | 0,07 | 0,39 | | | |
| Characteristic value | haracteristic value | | | | | | 0,19 | | | |





Laminate 22_Araldite LY 1569 / Aradur 3489-Biotex UD-NaOH 2M_3h

Laminate 22 Tensile 0° test



Laminate 22 Tensile 90° test



Laminate 23_Araldite LY 1569 /Aradur 3489-Biotex UD-NaOH 2M_12h

Laminate 23 Tensile 0° test



Laminate 23 Tensile 90° test

| | | | | Vf | N AF | | | | | |
|----------------------|----------------|------|-------|-------|----------|-------|-------|--------------|------------------------|-------------|
| Comple | ID | а | b | F | Eyt | σyt | εyt | Failura mada | VI | IVII |
| Sample | U | mm | mm | kN | Mpa | Mpa | % | Fallure mode | (%) | (%) |
| 1 | 1_Bio_T90_#23# | 3,17 | 24,71 | 0,613 | 1069,300 | 7,828 | 1,221 | Тор | 24,29 | 29,5 |
| 2 | 2_Bio_T90_#23# | 3,08 | 24,80 | 0,600 | 1115,820 | 7,852 | 1,211 | Тор | Parameters | Value |
| 3 | 3_Bio_T90_#23# | 3,00 | 24,78 | 0,609 | 1135,650 | 8,186 | 1,490 | Тор | Farameters | value |
| 4 | 4_Bio_T90_#23# | 3,19 | 24,72 | 0,618 | 1032,370 | 7,837 | 1,379 | Тор | Testing speed (mm/min) | 2 |
| 5 | 5_Bio_T90_#23# | 3,04 | 24,71 | 0,606 | 1102,800 | 8,071 | 1,516 | Тор | Extensometry | Video |
| 6 | 6_Bio_T90_#23# | 3,01 | 24,68 | 0,582 | 1357,830 | 7,831 | 0,785 | Тор | Reference points | Stickers |
| 7 | 7_Bio_T90_#23# | 3,04 | 24,80 | 0,753 | 1796,980 | 9,994 | 0,737 | Тор | Norma | EN ISO527-5 |
| 8 | 8_Bio_T90_#23# | 2,80 | 24,82 | 0,547 | 958,025 | 7,873 | | Middle | Precharge (kN) | 1 |
| 9 | 9_Bio_T90_#23# | 2,84 | 24,67 | | | | | Middle | Modulus (ε %) | 0,05-0,25 |
| Average value | | 3,02 | 24,74 | 0,62 | 1196,10 | 8,18 | 1,19 | | | |
| Standard deviation | | | | 0,06 | 268,75 | 0,74 | 0,32 | | | |
| CV(%) 0,10 | | | 0,10 | 0,22 | 0,09 | 0,27 | | | | |
| Characteristic value | stic value | | | | 597,70 | 6,53 | 0,49 | | | |



Laminate 24 Araldite LY 1569 / Aradur 3489-Biotex UD-NaOH 2M_24h

Laminate 24 Tensile 0° test



Laminate 24 Tensile 90° test

| | | Transverse tensile test | | | | | | | | N AF |
|----------------------|----------------|-------------------------|-------|-------|---------|-------|-------|--------------|------------------------|-------------|
| Comple | ID | а | b | F | Eyt | σyt | εyt | Failura mada | VI | IVII |
| Sample | U | mm | mm | kN | Mpa | Мра | % | Fallure mode | (%) | (%) |
| 1 | 1_Bio_T90_#24# | 2,92 | 24,80 | 0,509 | 905,295 | 7,035 | 1,433 | Тор | 24,39 | 29,6 |
| 2 | 2_Bio_T90_#24# | 2,92 | 24,84 | 0,475 | 855,473 | 6,543 | 1,401 | Тор | Baramotors | Value |
| 3 | 3_Bio_T90_#24# | 2,98 | 24,84 | 0,480 | 804,154 | 6,480 | 1,713 | Тор | Farameters | varue |
| 4 | 4_Bio_T90_#24# | 3,04 | 24,87 | 0,485 | 867,100 | 6,414 | 1,577 | Тор | Testing speed (mm/min) | 2 |
| 5 | 5_Bio_T90_#24# | 3,04 | 24,85 | 0,500 | 924,193 | 6,623 | 1,277 | Тор | Extensometry | Video |
| 6 | 6_Bio_T90_#24# | 2,98 | 24,77 | | | | | Middle | Reference points | Stickers |
| 7 | 7_Bio_T90_#24# | 2,88 | 24,95 | 0,483 | 906,686 | 6,723 | 1,726 | Middle | Norma | EN ISO527-5 |
| 8 | 8_Bio_T90_#24# | 3,04 | 24,87 | 0,393 | 699,823 | 5,198 | 1,176 | Bottom | Precharge (kN) | 1 |
| 9 | 9_Bio_T90_#24# | 3,25 | 24,80 | 0,432 | 679,443 | 5,358 | 1,866 | Middle | Modulus (ε %) | 0,05-0,25 |
| Average value | | 3,01 | 24,84 | 0,47 | 830,27 | 6,30 | 1,52 | | | |
| Standard deviation | | | | 0,04 | 94,66 | 0,66 | 0,24 | | | |
| CV(%) | | | | 0,08 | 0,11 | 0,10 | 0,16 | | | |
| Characteristic value | stic value | | | | | 4,83 | 0,99 | | | |



Laminate 25 Araldite LY 1569 / Aradur 3489 (1% silane)-Biotex UD-Untreated

Laminate 25 Tensile 0° test



Laminate 25 Tensile 90° test







25 25 25 4 5 6

2.5 25

25

25

25

25 25 4 5

25

25

25 25

Laminate 26 Araldite LY 1569 / Aradur 3489 (1% silane)-Biotex UD-NaOH 1M _3h

Laminate 26 Tensile 0° test



Laminate 26 Tensile 90° test



Laminate 27 Araldite LY 1569 / Aradur 3489-Biotex UD-1% silane

Laminate 27 Tensile 0° test



Laminate 27 Tensile 90° test



Laminate 28 Araldite LY 1569 / Aradur 3489-Biotex UD-1% silane + NaOH 1M_3h

Laminate 28 Tensile 0° test



Laminate 28 Tensile 90° test



Laminate 29 Araldite LY 1569 / Aradur 3489-Biotex UD

Laminate 29 Tensile 0° test

| | | | | Vf | NAF | | | | |
|----------------------|----------------|------|-------|-------|-----------|---------|-------|------------------------|-------------|
| Sample | ID | а | b | F | Ext | σxt | εxt | VI | IVII |
| Sample | | mm | mm | kN | Mpa | Mpa | % | (%) | (%) |
| 1 | 1_Bio_T0_#29# | 2,19 | 14,50 | 4,831 | 10295,900 | 152,147 | 1,669 | | |
| 2 | 2_Bio_T0_#29# | 2,14 | 15,48 | 5,212 | 10016,500 | 157,342 | 1,699 | 32 | 45 |
| 3 | 3_Bio_T0_#29# | 2,18 | 15,38 | 5,311 | 11051,100 | 158,391 | 1,701 | | |
| 4 | 4_Bio_T0_#29# | 2,19 | 14,67 | 4,610 | 263,631 | 143,484 | 1,992 | | |
| 5 | 5_Bio_T0_#29# | 2,24 | 14,85 | 4,947 | 10041,100 | 148,707 | 1,611 | Parameters | Value |
| 6 | 6_Bio_T0_#29# | 2,54 | 15,05 | 5,395 | 9186,350 | 141,131 | 1,719 | | |
| 7 | 7_Bio_T0_#29# | 2,27 | 15,00 | 5,122 | 13848,200 | 150,425 | 1,476 | Testing speed (mm/min) | 2 |
| 8 | 8_Bio_T0_#29# | 2,26 | 14,96 | | | | | Extensometry | Video |
| 9 | 9_Bio_T0_#29# | 2,26 | 14,98 | 4,635 | 11985,800 | 136,902 | 1,268 | Reference points | Stickers |
| 10 | 10_Bio_T0_#29# | 2,37 | 14,72 | 5,232 | 9601,120 | 149,960 | 1,636 | Norma | EN ISO527-5 |
| 11 | 11_Bio_T0_#29# | 2,31 | 14,93 | 5,174 | 11847,500 | 150,012 | 1,379 | Precharge (kN) | 5 |
| 12 | 12_Bio_T0_#29# | 2,27 | 14,93 | 5,533 | 12799,600 | 163,252 | 1,561 | Modulus (ε %) | 0,05-0,25 |
| 13 | 13_Bio_T0_#29# | 2,22 | 14,97 | 5,365 | 11239,800 | 161,426 | 1,655 | | |
| 14 | 14_Bio_T0_#29# | 2,31 | 15,00 | 5,450 | 9524,930 | 157,292 | 1,742 | | |
| Average value | | 2,27 | 14,96 | 5,14 | 10130,89 | 151,57 | 1,62 | | |
| Standard deviation | | 0,10 | 0,25 | 0,30 | 3270,52 | 7,91 | 0,18 | | |
| CV(%) | | | | 0,06 | 0,32 | 0,05 | 0,11 | | |
| Characteristic value | | | | | 3197,81 | 134,81 | 1,24 | | |



Laminate 29 Tensile 0° test –gauges

| | | | L | | Vf | Mf | | | | |
|----------------------|----------------|------|-------|------|----------|--------|------|------|------------|-------|
| Samplo | Sample | | b | F | Ext | σxt | εxt | vxy | | |
| Sample | ID. | mm | mm | kN | Mpa | Mpa | % | | (%) | (%) |
| 1 | 1_Bio_T0g_#29# | 2,13 | 14,88 | 5,00 | | 158,00 | | | 33,4 | 46,5 |
| 2 | 2_Bio_T0g_#29# | 2,26 | 14,62 | 5,01 | 13100,00 | 152,00 | 1,70 | 0,43 | Baramotors | Value |
| 3 | 3_Bio_T0g_#29# | 2,10 | 14,95 | 4,76 | 13800,00 | 152,00 | 1,40 | 0,49 | Parameters | value |
| Average value | | 2,16 | 14,82 | 4,92 | 13450,00 | 154,00 | 1,55 | 0,46 | | |
| Standard deviation | | | | 0,14 | 494,97 | 3,46 | 0,21 | 0,04 | | |
| CV(%) | | | | 0,03 | 0,04 | 0,02 | 0,14 | 0,09 | | |
| Characteristic value | | | | | 12364,36 | 146,40 | 1,08 | 0,37 | | |





Laminate 29 Tensile 90° test

Laminate 30 SuperSap CLR/INF-Biotex UD

Laminate 30 Compression 0° test

| | | Longitudinal compression test | | | | | | | | | | |
|----------------------|---------------|-------------------------------|-------|---------|----------|--------|------|------------|-------|--|--|--|
| Campila | Samplo ID a b | | | | Exc | σχς | EXC | | | | | |
| Sample | | mm | mm | kN | MPa | MPa | % | (%) | (%) | | | |
| 1 | 1_Bio_C0_#30# | 3,62 | 24,79 | 9730,00 | 11512,75 | 108,00 | 2,80 | | | | | |
| 2 | 2_Bio_C0_#30# | 3,66 | 24,84 | 9600,00 | 10575,00 | 106,00 | 3,60 | 0,31 | 43,8 | | | |
| 3 | 3_Bio_C0_#30# | 3,40 | 24,90 | 9130,00 | 10436,00 | 108,00 | 4,80 | | | | | |
| 4 | 4_Bio_C0_#30# | 3,40 | 25,07 | 8890,00 | | 104,00 | | Doromotors | Value | | | |
| 5 | 5_Bio_C0_#30# | 3,51 | 25,29 | 9700,00 | 10183,50 | 109,00 | 3,90 | Parameters | value | | | |
| Average value | | 3,52 | 24,98 | 9410,00 | 10676,81 | 107,00 | 3,78 | | | | | |
| Standard deviation | | 0,12 | 0,20 | 377,96 | 580,37 | 2,00 | 0,83 | | | | | |
| CV(%) | | | | 0,04 | 0,05 | 0,02 | 0,22 | | | | | |
| Characteristic value | | | | | 9446,49 | 102,76 | 2,02 | | | | | |

| | Module | | | Gaug | e 1 | | Gauge 2 | |
|---|---------------|----------|----------|---------|---------|----------|---------|---------|
| | | Exc | Exc1 | σxc 0,1 | σxc 0,3 | Exc2 | σxc 0,1 | σхс 0,3 |
| | 1_Bio_C0_#30# | 11512,75 | 12570,50 | 15,67 | 40,81 | 10455,00 | 12,81 | 33,72 |
| 2 | 2_Bio_C0_#30# | 10575,00 | 10575,00 | 13,61 | 34,76 | 0,00 | | |
| | 3_Bio_C0_#30# | 10436,00 | 10436,00 | 11,53 | 32,40 | 0,00 | | |
| 4 | 4_Bio_C0_#30# | 0,00 | 0,00 | | | 0,00 | | |
| 5 | 5_Bio_C0_#30# | 10183,50 | 9791,50 | 13,34 | 32,92 | 10575,50 | 14,13 | 35,28 |



Laminate 30 Compression 90° test

| | | Transverse compression test | | | | | | | | | | |
|----------------------|----------------|-----------------------------|-------|---------|---------|-------|------|------------|-------|--|--|--|
| Samplo | П | а | b | F | Eyc | σус | εγς | | | | | |
| Sample | 10 | mm | mm | kN | MPa | MPa | % | (%) | (%) | | | |
| 1 | 1_Bio_C90_#30# | 3,60 | 24,93 | 7720 | 2980,00 | 86,00 | | | | | | |
| 2 | 2_Bio_C90_#30# | 3,48 | 24,98 | 7660 | 3315,00 | 88,10 | 4,35 | 0,31 | 43,8 | | | |
| 3 | 3_Bio_C90_#30# | 3,52 | 24,91 | 7520 | 4475,00 | 85,70 | | | | | | |
| 4 | 4_Bio_C90_#30# | 3,65 | 25,27 | 7870 | 2884,00 | 85,30 | 5,00 | Daramotors | Value | | | |
| Average value | | 3,56 | 25,02 | 7692,50 | 3413,50 | 86,28 | 4,68 | Farameters | value | | | |
| Standard deviation | | 0,08 | 0,17 | 145,00 | 731,39 | 1,25 | 0,46 | | | | | |
| CV(%) | | | | 0,02 | 0,21 | 0,01 | 0,10 | | | | | |
| Characteristic value | | | | | 1863,06 | 83,63 | 3,70 | | | | | |

| Module | | | Gau | ge 1 | | Gau | ge 2 |
|----------------|---------|---------|---------|---------|---------|---------|---------|
| | Exc | Exc1 | σxc 0,1 | σxc 0,3 | Exc2 | σxc 0,1 | σxc 0,3 |
| 1_Bio_C90_#30# | 2980,00 | 2980,00 | 3,40 | 9,36 | | | |
| 2_Bio_C90_#30# | 3315,00 | 3315,00 | 3,40 | 10,03 | 3555,00 | 3,60 | 10,71 |
| 3_Bio_C90_#30# | 4475,00 | 4475,00 | 4,30 | 13,25 | 0,00 | | |
| 4_Bio_C90_#30# | 2884,00 | 2884,00 | 3,35 | 9,12 | 0,00 | | |



| Laminate | 30 | ILSS | test |
|----------|----|------|------|
| | | | |

| | Fmax | Short-beam Stregth | Max strength | Max displacement |
|-----------------|---------|--------------------|--------------|------------------|
| ILSS Sample | Ν | N/mm2 | N/mm2 | mm |
| 1_Bio_ILSS_#30# | 1750 | 18,52 | 1060,00 | 0,98 |
| 2_Bio_ILSS_#30# | 1860 | 18,65 | 1000,00 | 0,94 |
| 3_Bio_ILSS_#30# | 1920 | 18,42 | 943,00 | 1,10 |
| 4_Bio_ILSS_#30# | 1970 | 18,49 | 933,00 | 0,94 |
| 5_Bio_ILSS_#30# | 2020 | 19,84 | 1040,00 | 1,00 |
| 6_Bio_ILSS_#30# | 2150 | 19,99 | 996,00 | 0,96 |
| 7_Bio_ILSS_#30# | 1710 | 17,71 | 982,00 | 1,30 |
| 8_Bio_ILSS_#30# | 1820 | 18,19 | 975,00 | 0,98 |
| 9_Bio_ILSS_#30# | 1740 | 16,18 | 807,00 | 1,00 |
| 10_Bio_ILSS_#30 | 1840 | 17,66 | 907,00 | 0,94 |
| Average | 1880 | 18,3631 | 964 | 1,00 |
| SD | 138,71 | 1,09 | 72,31 | 0,11 |
| Max | 2150,00 | 19,99 | 1060,00 | 1,30 |
| Min | 1710,00 | 16,18 | 807,00 | 0,94 |
| CV | 0,07 | 0,06 | 0,07 | 0,11 |





Laminate 31 SuperSap CLR/INF (1% silane in hardener)-Biotex UD

Max_Disp_Ext.1(Deformacion) Е Rm I.Poisson Fm Sample kΝ N/mm2 N/mm2 % 1_Bio_T0_#31# True 4,48 12061,50 142,88 1,45 0,39 2_Bio_T0_#31# True 4,82 12576,90 149,83 1,61 0,41 3_Bio_T0_#31# True 5,08 15068,80 151,07 1,47 0,84 4_Bio_T0_#31# True 4,82 14735,50 150,17 1,28 0,45 5_Bio_T0_#31# True 5,03 14452,90 153,63 1,43 0,22 6_Bio_T0_#31# 16611,40 True 5,25 158,46 1,54 1,78 7_Bio_T0_#31# True 5,38 13907,20 163,85 1,62 0,48 8 Bio TO #31# True 5,62 11539,10 162,78 1.64 0.18 9 Bio TO #31# 5,33 15321,40 152,63 1,37 0,51 True 10_Bio_T0_#31# True 5,40 14555,50 145,98 1,42 0,60 11_Bio_T0_#31# 14111,70 160,94 True 5,70 1,61 0,45 12_Bio_T0_#31# True 5,46 13699,00 152,07 1,49 0,51 13_Bio_T0_#31# True 5,28 11617,00 148,62 1,56 0,30 14_Bio_T0_#31# True 5,44 13183,90 155,60 1,57 0,23 5,22 13817,30 153,47 1,50 0,53 Average SD 0,34 1482,99 6,23 0,11 0,40 Max 5,70 16611,40 163,85 1,64 1,78 Min 11539,10 1,28 4,48 142,88 0,18 CV 0,06 0,11 0,04 0,07 0,76





Laminate 31 Tensile 0° test - gauges

| | Fm | E | Rm | Max_Disp_Ext.1(Deformacion) | I.Poisson |
|-----------------|------|-------|-------|-----------------------------|-----------|
| Sample | kN | N/mm2 | N/mm2 | % | |
| 15_Bio_T0g_#31# | 5,3 | | 160 | | |
| 16_Bio_T0g_#31# | 5,22 | 13800 | 159 | 1,7 | 0,46 |
| 17_Bio_T0g_#31# | 5,38 | 14100 | 170 | 1,7 | 0,50 |



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| Editinate SE ic | | LUGL | | |
|-----------------|------|---------|-------|----------------|
| | Fm | E | Rm | Max_Disp_Ext.1 |
| Sample | kN | N/mm2 | N/mm2 | % |
| 1_Bio_T90_#31# | 1,02 | 2927,78 | 18,74 | 0,49 |
| 2_Bio_T90_#31# | 1,14 | 3377,61 | 21,41 | 0,54 |
| 3_Bio_T90_#31# | 0,95 | 2933,55 | 17,22 | 0,43 |
| 4_Bio_T90_#31# | 1,03 | 2868,68 | 19,24 | 0,51 |
| 5_Bio_T90_#31# | 1,08 | 2795,11 | 19,15 | 0,54 |
| 6_Bio_T90_#31# | 0,92 | 2877,51 | 16,47 | 0,40 |
| 7_Bio_T90_#31# | 0,99 | 2938,75 | 18,10 | 0,45 |
| 8_Bio_T90_#31# | 0,96 | 2637,13 | 17,04 | 0,47 |
| 9_Bio_T90_#31# | 1,04 | 3121,89 | 19,51 | 0,49 |
| Average | 1,02 | 2942,00 | 18,54 | 0,48 |
| SD | 0,07 | 208,10 | 1,52 | 0,05 |
| Max | 1,14 | 3377,61 | 21,41 | 0,54 |
| Min | 0,92 | 2637,13 | 16,47 | 0,40 |
| Coef.Variacion | 0,07 | 0,07 | 0,08 | 0,10 |

Laminate 31 Tensile 90° test



Laminate 31 Tensile 90° test – gauge



Laminate 32 SuperSap CLR/INF (1% silane in hardener)-Biotex UD

Laminate 32 Compression 0° test

| | | Longitudinal compression test | | | | | | | | | | | | |
|----------------------|---------------|-------------------------------|-------|----------|----------|--------|------|------------|-------|--|--|--|--|--|
| Samplo | ID | а | b | F | Exc | σχς | εхс | | | | | | | |
| Sample | | mm | mm | kN | MPa | MPa | % | (%) | (%) | | | | | |
| 1 | 1_Bio_C0_#32# | 3,62 | 24,79 | 9820 | 11615,00 | 113 | 3,65 | | | | | | | |
| 2 | 2_Bio_C0_#32# | 3,66 | 24,84 | 10100 | 10635,00 | 110 | 4,25 | 31,00 | 43,8 | | | | | |
| 3 | 3_Bio_C0_#32# | 3,40 | 24,90 | 10100 | 10100,00 | 113 | 5,00 | | | | | | | |
| 4 | 4_Bio_C0_#32# | 3,40 | 25,07 | 10500 | 9250,00 | 111 | 4,70 | Daramatars | Value | | | | | |
| Average value | | 3,52 | 24,90 | 10130,00 | 10400,00 | 111,75 | 4,40 | Parameters | value | | | | | |
| Standard deviation | | 0,14 | 0,12 | 279,76 | 990,61 | 1,50 | 0,59 | | | | | | | |
| CV(%) | | | | 0,03 | 0,10 | 0,01 | 0,13 | | | | | | | |
| Characteristic value |] | | | | 8300,03 | 108,57 | 3,15 | | | | | | | |

| | | | Gau | ge 1 | | Gauge 2 | | | | |
|---------------|----------|----------|---------|---------|----------|---------|---------|--|--|--|
| Modulus | Exc | Exc1 | σxc 0,1 | σxc 0,3 | Exc2 | σxc 0,1 | σхс 0,3 | | | |
| 1_Bio_C0_#32# | 11615,00 | 9335,00 | 12,73 | 31,40 | 13895,00 | 20,26 | 48,05 | | | |
| 2_Bio_C0_#32# | 10635,00 | 11940,00 | 13,38 | 37,26 | 9330,00 | 12,40 | 31,06 | | | |
| 3_Bio_C0_#32# | 10100,00 | 10100,00 | 13,02 | 33,22 | 0,00 | | | | | |
| 4_Bio_C0_#32# | 9250,00 | 9250,00 | 12,66 | 31,16 | 0,00 | | | | | |



Laminate 32 Compression 90° test

| | | Transverse compression test | | | | | | | | | | | |
|----------------------|----------------|-----------------------------|-------|---------|---------|-------|------|------------|-------|--|--|--|--|
| Samplo | П | а | b | F | Eyc | σγς | εγς | | | | | | |
| Sample | 10 | mm | mm | kN | MPa | MPa | % | (%) | (%) | | | | |
| 1 | 1_Bio_C90_#32# | 3,79 | 24,89 | 7890 | 3307,50 | 83,60 | 3,55 | | | | | | |
| 2 | 2_Bio_C90_#32# | 3,71 | 24,90 | 8500 | 3785,00 | 92,00 | 2,40 | 0,30 | 42,6 | | | | |
| 3 | 3_Bio_C90_#32# | 3,64 | 24,94 | 8410 | 3425,00 | 92,60 | 5,00 | | | | | | |
| 4 | 4_Bio_C90_#32# | 3,67 | 25,04 | 8420 | 3775,00 | 91,60 | 4,20 | Daramotors | Value | | | | |
| Average value | | 3,70 | 24,94 | 8305,00 | 3573,13 | 89,95 | 3,79 | Parameters | value | | | | |
| Standard deviation | | 0,06 | 0,07 | 279,58 | 243,68 | 4,25 | 1,10 | | | | | | |
| CV(%) | | | | 0,03 | 0,07 | 0,05 | 0,29 | | | | | | |
| Characteristic value | | | | | 3056,55 | 80,93 | 1,46 | | | | | | |

| | | | | Gau | ge 1 | | Gauge 2 | | | | | |
|---|----------------|---------|---------|---------|---------|---------|---------|---------|--|--|--|--|
| | Modulus | Exc | Exc1 | σxc 0,1 | σxc 0,3 | Exc2 | σxc 0,1 | σxc 0,3 | | | | |
| | 1_Bio_C90_#32# | 3307,50 | 2855,00 | 2,96 | 8,67 | 3760,00 | 3,43 | 10,95 | | | | |
| | 2_Bio_C90_#32# | 3785,00 | 3785,00 | 3,40 | 10,97 | | | | | | | |
| | 3_Bio_C90_#32# | 3425,00 | 3425,00 | 3,42 | 10,27 | 0,00 | | | | | | |
| ſ | 4_Bio_C90_#32# | 3775,00 | 3775,00 | 4,55 | 12,10 | 0,00 | | | | | | |



| | Max F | Short-beam Stregth | MaxTension | Displacement |
|-----------------|-------------------|--------------------|------------|--------------|
| Sample | N | N/mm2 | N/mm2 | mm |
| 1_Bio_ILSS_#32# | 2010 | 18,84 | 975 | 0,91 |
| 2_Bio_ILSS_#32# | 2170 | 19,85 | 1000 | 0,82 |
| 3_Bio_ILSS_#32# | 2130 | 20,06 | 1040 | 0,93 |
| 4_Bio_ILSS_#32# | 2000 | 19,29 | 1030 | 0,88 |
| 5_Bio_ILSS_#32# | 1900 | 18,14 | 948 | 0,87 |
| 6_Bio_ILSS_#32# | 2020 | 18,32 | 917 | 0,90 |
| 7_Bio_ILSS_#32# | 2140 | 20,03 | 1040 | 0,92 |
| 8_Bio_ILSS_#32# | 2040 | 19,26 | 1000 | 0,94 |
| 9_Bio_ILSS_#32# | 1910 | 18,02 | 936 | 0,88 |
| 10_Bio_ILSS_#32 | 1980 | 19,27 | 1030 | 0,90 |
| Average | 2030,00 | 19,11 | 992,00 | 0,90 |
| SD | 92,50 | 0,76 | 45,47 | 0,03 |
| Max | 2170,00 | 20,06 | 1040,00 | 0,94 |
| Min | Min 1900,00 18,02 | | 917,00 | 0,82 |
| CV | 0,05 | 0,04 | 0,05 | 0,04 |

Laminate 32 ILSS test



Laminate 33 SuperSap CLR/INF (1% silane in hardener)-Biotex UD Laminate 34 SuperSap CLR/INF -Biotex UD Flexural tests

| | | Initial flexu | ral propeties | Degraded flex | ural properties | Percentage | | | |
|-------------|-----------------------|---------------|--------------------------------------------------|---------------|-----------------|------------|-------------|--|--|
| | | Force (N) | Force (N) Modulus (N/mm2) Force (N) Modulus (N/m | | | | Modulus (%) | | |
| Logitudinal | 33-Silane in hardener | 103,12 | 18900 | 65,77 | 7877 | 36,22 | 58,32 | | |
| Logituumai | 34-No treated | 84,47 | 16222 | 62,28 | 12483 | 26,27 | 23,05 | | |
| T | 33-Silane in hardener | 23,00 | 5720 | 15,03 | 3700 | 34,65 | 35,31 | | |
| Transverse | 34-No treated | 21,70 | 5620 | 12,10 | 2985 | 44,24 | 46,89 | | |

Laminate 35 SuperSap CLR/INF (1% silane in hardener)-Biotex UD

| | Fmax | Shear strain | Shear strength | Modulus (0.2-0.6) |
|-----------------|------|--------------|----------------|-------------------|
| | N | % | N/mm2 | N/mm2 |
| 1_Bio_\$45_#35# | 3010 | 2,5 | 26,6 | 1720 |
| 2_Bio_\$45_#35# | 3000 | 3 | 25,1 | 1540 |
| 3_Bio_\$45_#35# | 2870 | 2,4 | 25,7 | 1630 |
| 4_Bio_S45_#35# | 2890 | 3,7 | 24,9 | 1500 |





Laminate 36 SuperSap CLR/INF -Biotex UD

Laminate 36 In-plane shear test S45

| | Fmax | Shear strain | Shear strength | Shear modulus |
|-----------------|------|--------------|----------------|---------------|
| Sample | Ν | % | N/mm2 | N/mm2 |
| 1_Bio_S45_#36# | 2970 | 3,9 | 25,7 | 1580 |
| 2_Bio_S45_#36# | 2950 | 3,7 | 26,2 | 1560 |
| 3_Bio_\$45_#36# | 3000 | 3,5 | 25,7 | 1550 |
| 4_Bio_\$45_#36# | 2940 | 3,9 | 24,9 | 1470 |



APPENDIX C



Figure D.1: Untreated flax fabric panoramic picture (left) and measurement 1(right)



Figure D.2: Untreated flax measurement 2 (left) and 3 (right)



Figure D.3: Untreated flax measurement 4 (left) and 5 (right)



Figure D.4: 1M NaOH 3h flax treatment measurement 1 (left) and 2 (right)



Figure D.5: 1M NaOH 3h flax treatment measurement 3 (left) and 4 (right)



Figure D. 6: 1M NaOH 3h flax treatment measurement 5



Figure D.7: 2M NaOH 72h flax treatment measurement 1(left) and 2 (right)



Figure D.8: 2M NaOH 72h flax treatment measurement 3(left) and 4 (right)



Figure D .9: 2M NaOH 72h flax treatment measurement 5 (left) and 6 (right)



Figure D.10: 2M NaOH 72h flax treatment measurement 7

Appendix E

Immersion tests

| | Date | Start | | 1 day | | | 2 days | | | 3 days | | | 7 days | | | 14 day | | | 20 day | | 35 day | | | 44 day | | | 81 day | | | 100 day-DRY | | Y | | |
|--------|--------|--------|-------|--------|-------|-------|--------|-------|-------|--------|-------|-------|--------|-------|-------|--------|-------|-------|--------|-------|--------|--------|-------|--------|-------|-------|--------|-------|-------|-------------|-------|-------|--------|--|
| | | 29-oct | | 30-oct | | | 31-oct | | | 01-nov | | | 05-nov | | | 12-nov | - | | 18-nov | | | 03-dic | | 03-dic | | | 12-dic | | | 18-ene | | | 20-feb | |
| | Sample | g | g | % | x | g | % | х | g | % | х | g | % | х | g | % | x | g | % | х | g | % | х | g | % | х | g | % | | g | % | x | | |
| | 1 | 4,120 | 4,161 | 0,995 | | 4,189 | 1,675 | | 4,189 | 1,675 | | 4,238 | 2,864 | | 4,275 | 3,762 | | 4,292 | 4,175 | | 4,314 | 4,709 | | 4,314 | 4,709 | | 4,367 | 5,995 | | 4,173 | 1,291 | | | |
| | 2 | 3,824 | 3,861 | 0,968 | | 3,891 | 1,752 | | 3,882 | 1,517 | | 3,929 | 2,746 | | 3,967 | 3,740 | | 3,979 | 4,053 | | 4,002 | 4,655 | | 4,008 | 4,812 | | 4,019 | 5,099 | | 3,868 | 1,145 | | | |
| 33 | 3 | 3,875 | 3,917 | 1,084 | | 3,939 | 1,652 | | 3,944 | 1,781 | | 3,986 | 2,865 | | 4,030 | 4,000 | | 4,045 | 4,387 | | 4,075 | 5,161 | | 4,075 | 5,161 | | 4,098 | 5,755 | | 3,931 | 1,432 | | | |
| Silane | 4 | 3,675 | 3,717 | 1,143 | 1,057 | 3,746 | 1,932 | 1,745 | 3,735 | 1,633 | 1,721 | 3,783 | 2,939 | 2,922 | 3,824 | 4,054 | 3,923 | 3,832 | 4,272 | 4,282 | 3,860 | 5,034 | 4,913 | 3,858 | 4,980 | 5,067 | 3,882 | 5,633 | 5,828 | 3,719 | 1,189 | 1,316 | | |
| | 5 | 4,160 | 4,213 | 1,274 | | 4,239 | 1,899 | | 4,245 | 2,043 | | 4,289 | 3,101 | | 4,331 | 4,111 | | 4,348 | 4,519 | | 4,375 | 5,168 | | 4,390 | 5,529 | | 4,421 | 6,274 | | 4,227 | 1,608 | | | |
| | 6 | 4.106 | 4.142 | 0.877 | | 4.170 | 1.559 | | 4.175 | 1.680 | | 4.230 | 3.020 | | 4,265 | 3.872 | | 4.282 | 4.286 | | 4.301 | 4,749 | | 4.320 | 5.212 | | 4.361 | 6.210 | | 4.156 | 1.227 | | | |
| | 1 | 4.377 | 4.416 | 0.891 | | 4,447 | 1.599 | | 4.447 | 1.599 | | 4,492 | 2.627 | | 4.553 | 4.021 | | 4,559 | 4.158 | | 4.591 | 4.889 | | 4.607 | 5.255 | | 4.647 | 6.169 | | 4,446 | 1.572 | | | |
| | 2 | 4.226 | 4.261 | 0.828 | | 4.304 | 1.846 | | 4.295 | 1.633 | | 4.344 | 2.792 | | 4.390 | 3.881 | | 4.401 | 4.141 | | 4.426 | 4.733 | | 4,428 | 4.780 | | 4,464 | 5.632 | | 4.300 | 1.749 | | | |
| 34 | 3 | 4.095 | 4.132 | 0.904 | | 4.164 | 1.685 | | 4.164 | 1.685 | | 4.215 | 2.930 | | 4.254 | 3,883 | | 4,269 | 4.249 | | 4,289 | 4.737 | | 4.294 | 4.860 | | 4.357 | 6.398 | | 4.166 | 1.729 | | | |
| Untre | 4 | 4.134 | 4.175 | 0.992 | 0,878 | 4.191 | 1.379 | 1,554 | 4.208 | 1.790 | 1,661 | 4.253 | 2.879 | 2,755 | 4.294 | 3.870 | 3,942 | 4.309 | 4.233 | 4,191 | 4.335 | 4.862 | 4,771 | 4.357 | 5.394 | 5,087 | 4.393 | 6,265 | 6,086 | 4.208 | 1.792 | 1,709 | | |
| | 5 | 4.169 | 4.205 | 0.864 | | 4.238 | 1.655 | | 4.239 | 1.679 | | 4.278 | 2.615 | | 4.340 | 4.102 | | 4.344 | 4.198 | | 4.368 | 4.773 | | 4.385 | 5.181 | | 4.441 | 6.529 | | 4.242 | 1.739 | | | |
| | 6 | 4,057 | 4,089 | 0,789 | | 4,104 | 1,158 | | 4,121 | 1,578 | | 4,166 | 2,687 | | 4,215 | 3,895 | | 4,226 | 4,166 | | 4,245 | 4,634 | | 4,262 | 5,053 | | 4,281 | 5,521 | | 4,125 | 1,676 | | | |

Appendix F

UNCERTAINTY ESTIMATION IN MECHANICAL PROPERTIES TESTS

1. Abstract

The current report defined Acciona Blades (AB) laboratory tests uncertainty type B values based on the ISO Guide [1]. While type A uncertainties are determined analyzing a series of independent repeated measurements, type B uncertainties are based on scientific judgment combining all available information on the possible variability of the quantity. The uncertainty data might come from properties of relevant materials, measurement instruments, manufacturer's specifications or calibration certificate data.

Most of the experimental campaign of the PhD was performed in AB laboratory, as a result the test uncertainties come from the instrumentation and machinery used in the referred laboratory.

2. Introduction

AB different tests uncertainties determination was performed following Lekou et al. study [2]. Based on this document the first step was to identify the parameters for which uncertainty is determined. Table 1 established the uncertainty measurands.

| Measurand | Unit | Symbol |
|--------------|------|--------|
| Thickness | mm | Н |
| Width | mm | W |
| Applied load | kN | F |

| able 1 | : Uncertainty | measurands |
|--------|---------------|------------|
| | | |

т

| Exhibited strain | % | ٤ |
|-------------------------|-----------------|------------------|
| Cross-sectional area | mm ² | А |
| Elastic modulus | MPa | Е |
| Poisson's ratio | | V ₂₁ |
| Strength | MPa | σ _F |
| Strain at failure | % | ε _{max} |
| Shear modulus | MPa | G12 |
| Shear strength | MPa | T12F |
| Shear strain at failure | % | Ý 12F |

The following relationships are used for the measurands calculation (Equations 1-8):

| $A = H \times W$ | Equation 1 |
|-----------------------------------------------------------------|------------|
| $\sigma = \frac{F}{A}$ | Equation 2 |
| $E = \frac{\Delta\sigma}{\Delta\varepsilon}$ | Equation 3 |
| $\vartheta = rac{-\Delta \varepsilon_y}{\Delta \varepsilon_x}$ | Equation 4 |
| $	au = \frac{F}{2A}$ | Equation 5 |
| $\gamma = \varepsilon_x + \varepsilon_y$ | Equation 6 |
| $G_{12} = \frac{\Delta \tau}{\Delta \gamma}$ | Equation 7 |
| $F^{sbs} = 0.75 \frac{F_m}{A}$ | Equation 8 |
The measurands are based in the tests performed in AB laboratory. All the tests are supported by the DNVGL-SE-0436:2016-03 Shop approval in renewable energy certificate. This certification contemplates the methods in Table 2.

| Standard | Description | | | | |
|------------|--------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|--|--|--|--|
| ISO 527-1 | Plastics-Determination of tensile properties – Part 1: General principles | | | | |
| ISO 527-4 | Plastics-Determination of tensile properties-Part 4: Test conditions for isotropic and anisotropic fibre-reinforced plastic composites | | | | |
| ISO 527-5 | Plastics-Determination of tensile properties-Part 5: Test conditions for unidirectional fibre-reinforced plastic composites | | | | |
| ISO 14126 | Fiber-reinforced plastic composites – Determination of compressive properties in the in-plane direction | | | | |
| ISO 14129 | Fiber-reinforced plastic composites – Determination of the in-plane shear stress/strain response, including the in-plane shear modulus and strength, by the $\pm 45^{\circ}$ tension test method | | | | |
| ASTM D2344 | Standard Test Method for Short-Beam Strength of Polymer Matrix Composite Materials and Their Laminates | | | | |

Table 2: Certificated mechanical tests

3. Uncertainty analysis

The uncertainty type B values based in the ISO Guide were calculated for different mechanical tests measurands [1]. The combined standard uncertainty $u_c(\gamma)$ is the positive square root of the combined variance, which is given by Equation 9.

$$u_c^2(\gamma) = \sum_{j=1}^N \left(\frac{\partial f}{\partial x_j}\right)^2 u_{x_j}^2 + 2\sum_{j=1}^{N-1} \sum_{k=j+1}^N \frac{\partial f}{\partial x_j} \frac{\partial f}{\partial x_k} u_{x_j} u_{x_k} r(x_j, x_k)$$

Equation 9

Where $\gamma = f(x_1, x_2, ..., x_N)$ are the input quantities, u_{xj} is a standard uncertainty of quantity x_j evaluated as either type A or B uncertainty and $r(x_j, x_i)$ is the relative coefficient between x_j and x_k .

3.1 Uncertainty of width and thickness

Based on Equation 1 the cross section area was calculated by measuring the width and the thickness of the specimen. The use of a measuring instrument (caliper) has an associated uncertainty.

The maximum uncertainty allowed for the caliper is 0.02 mm and it assumes a rectangular distribution. When only one measurement is made, we are assuming that it is going to be the contribution of uncertainty. The type B uncertainty for measuring the thickness and width is calculated with Equation 10:

$$u_{H,instr} = \frac{0.02}{\sqrt{3}}$$
 Equation 10

According to the ISO 527-5 standard, ISO 14129 and ASTM D2344 standard, the mean values are calculated making at least three measurements along the coupon length and such value is used to define the width and thickness of each coupon. These measurements were taken into account to define Type A total uncertainty of the thickness and width measurement. Type A uncertainty values for $u_{W,A}$ and $u_{H,A}$ are defined in Equations 11 and 12. The assessment of the mean value of the measured width W, and thickness H, respectively are used in Equations 11 and 12.

$$u_{W,A} = \frac{S_W}{\sqrt{n_A}}$$
 Equation 11

$$u_{H,A} = \frac{S_H}{\sqrt{n_A}}$$
 Equation 12

Where S_W and S_H is the standard deviation of the measurements of the width and the thickness, respectively and n_A is the number of measured for each specimen. All sourced are used in Equations 13 and 14 in order to determinate type B uncertainty of the width and thickness respectively:

$$u_W = \sqrt{u_{W,A}^2 + u_{W,instr}^2}$$
 Equation 13
 $u_H = \sqrt{u_{H,A}^2 + u_{H,instr}^2}$ Equation 14

3.2 Uncertainty of cross-sectional area

The measurement uncertainty of the cross-section u_A was obtained combining Equations 1 and 8, assuming thickness and width measurement are independents, see Equation 15.

$$u_A = \sqrt{\left(\frac{u_W}{W}\right)^2 + \left(\frac{u_H}{H}\right)^2}$$
 Equation 15

3.3 Uncertainty of the applied force

As the calibration certificate defines the load reading device accuracy must be within $\pm 0.5\%$ of the actual value, assuming a rectangular distribution, getting as a result Equation 16 for the u_F calculation.

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Equation 19

Equation 17

Equation 16

3.4 Uncertainty of the applied force in stress

The axial stress type B uncertainty calculation is based on Equations 1 and 2; obtaining as a result Equation 17.

$$u_{\sigma} = \sigma \sqrt{\left(\frac{u_F}{F}\right)^2 + \left(\frac{u_A}{A}\right)^2}$$

Where u_A was calculated using Equation 15 and u_F Equation 16.

3.5 Uncertainty on strain

According to the ISO standard [1] the strain gauge accuracy must be at least of ±1%, because of this reason Equation 18 uncertainty might be assumed.

The elastic modulus is calculated with Equation 19.

 $E = \frac{\sigma_2 - \sigma_1}{\varepsilon_2 - \varepsilon_1}$

 $u_{\varepsilon} = \frac{1\varepsilon}{100\sqrt{3}}$

Equation 18

The combination for the Equation 9 and 18 leads to get the Equation 20.

$$u_{E} = E \sqrt{\frac{u_{\sigma_{1}}^{2} + u_{\sigma_{2}}^{2}}{(\sigma_{2} - \sigma_{1})^{2}} + \frac{u_{\varepsilon_{1}}^{2} + u_{\varepsilon_{2}}^{2}}{(\varepsilon_{2} - \varepsilon_{1})^{2}}}$$
Equation 20

If calculation the elastic modulus on the average module of both sides of the specimen, the uncertainty result as will be:

$$u_E = \sqrt{u_{E_1}^2 + u_{E_2}^2}$$
 Equation 21

3.7 Uncertainty on Poisson's ratio

The type B Poisson's ration uncertainty was given by the Equation 22.

$$u_{\vartheta} = \vartheta \sqrt{\frac{u_{\varepsilon_{y_1}}^2 + u_{\varepsilon_{y_2}}^2}{\left(\varepsilon_{y_2} - \varepsilon_{y_1}\right)^2} + \frac{u_{\varepsilon_{x_1}}^2 + u_{\varepsilon_{x_2}}^2}{\left(\varepsilon_{x_2} - \varepsilon_{x_1}\right)^2}}$$

Equation 22

3.8 Uncertainty in shear parameters: Shear strength

Equation 16 might be applied in order to calculate shear strength uncertainty, getting as a result Equation 23.

$$u_{\tau} = \frac{1}{2}\tau \sqrt{\left(\frac{u_F}{F}\right)^2 + \left(\frac{u_A}{A}\right)^2}$$

Equation 23

3.9 Uncertainty in shear parameters: Shear strain

The shear strain uncertainty was obtained applying Equation 24.

$$u_{\gamma} = \sqrt{u_{\varepsilon_{\mathcal{X}}}^2 + u_{\varepsilon_{\mathcal{Y}}}^2}$$

Equation 24

3.10 Uncertainty in shear parameters: Shear modulus

For the shear modulus the Equation 25 was used.

$$u_G = G_{\sqrt{\frac{u_{\tau_1}^2 + u_{\tau_2}^2}{(\tau_2 - \tau_1)^2} + \frac{u_{\gamma_1}^2 + u_{\gamma_2}^2}{(\gamma_2 - \gamma_1)^2}}}$$
Equation 25

3.11 Uncertainty in short beam strength

For the short beam strength the Equation 26 was used.

$$u_{F^{sbs}} = 0.75F^{sbs} \sqrt{\left(\frac{u_F}{F}\right)^2 + \left(\frac{u_A}{A}\right)^2}$$
 Equation 26

The above discussed uncertainties refer to the uncertainty of the property measured by a single (individual) specimen. However, we analysed a series of test (10 specimens), whereas the test

report should also include the mean values of properties measured, along with an indication of standard deviation and Coefficient of variation (CV).

4. Experimental calculation

Section 3 described the equations to apply for the type B uncertainties calculation. Section used such equations for the uncertainty values determination.

4.1. ISO527 Tensile test

Longitudinal tensile test uncertainties calculation has been done using data from **3_Bio_T0_#3#** and **2_Bio_T0g_#29#** samples.

| Measurand | Unit | Symbol | Data Formula | | Result |
|----------------------|-------------------|-------------------|--------------|--------------------------------------------------------------------------------------------|---------|
| Thickness | mm | н | 2,15/0,08/3 | $u_H = \sqrt{u_{H,A}^2 + u_{H,instr}^2}$ | 0.047 |
| Width | mm | W | 15.24/0.92/3 | $u_W = \sqrt{u_{W,A}^2 + u_{W,instr}^2}$ | 0,093 |
| Cross-sectional area | mm ² | A | 34.19 | $u_A = \sqrt{\left(\frac{u_W}{W}\right)^2 + \left(\frac{u_H}{H}\right)^2}$ | 0.531 |
| | N | F _{max} | 5580 | $u_F = \frac{0.5F}{100\sqrt{3}}$ | 16.11 |
| Force | | F0.05 | 457 | | 1.32 |
| | | F _{0.25} | 1606 | | 4.63 |
| | | Omax | 163.23 | | 6.5 |
| Applied load | N/mm ² | σ _{0.05} | 13.46 | $u_{\sigma} = \sigma_{\sqrt{\left(\frac{u_F}{F}\right)^2 + \left(\frac{u_A}{A}\right)^2}}$ | 0.54 |
| | | σ _{0.25} | 47.25 | | 1.89 |
| | | Emax | 1.32 | $u_{r} = \frac{1\varepsilon}{1}$ | 0.0076 |
| Strain | % | ٤0.05 | 0.05 | $u_{\varepsilon} = \frac{1}{100\sqrt{3}}$ | 0.00029 |

| | | £ 0,25 | 0.25 | | 0.0014 |
|--------------------|-----|------------------|----------|------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|---------|
| Elastic modulus | MPa | E | 10662.60 | $u_{E} = E \sqrt{\frac{u_{\sigma_{1}}^{2} + u_{\sigma_{2}}^{2}}{(\sigma_{2} - \sigma_{1})^{2}} + \frac{u_{\varepsilon_{1}}^{2} + u_{\varepsilon_{2}}^{2}}{(\varepsilon_{2} - \varepsilon_{1})^{2}}}$ | 620.49 |
| Exhibited strain y | % | ε γ0.05 | 0.021 | $u_{\varepsilon} = \frac{1\varepsilon}{100\sqrt{3}}$ | 0.00012 |
| | | £ y0.25 | 0.108 | 10043 | 0.00062 |
| Poisson's ratio | | ϑ_{21} | 0.43 | $u_{\vartheta} = \vartheta \sqrt{\frac{u_{\varepsilon y1}^2 + u_{\varepsilon y2}^2}{(\varepsilon_{y2} - \varepsilon_{y1})^2} + \frac{u_{\varepsilon x1}^2 + u_{\varepsilon x2}^2}{(\varepsilon_{x2} - \varepsilon_{x1})^2}}$ | 0.0032 |

4.2. ISO14126 Compression test

Longitudinal compression test uncertainties calculation has been done using data from samples

1_Bio_C0_#30#.

| Measurand | Unit | Symbol | Data | Formula | Result |
|--------------------------|------|---------------|-------|----------------------------------------------------------------------------|---------|
| Thickness | mm | Н | 0.02 | $u_{H,inst} = \frac{0.02}{\sqrt{3}}$ | 0.012 |
| Width | mm | w | 0.02 | $u_{W,inst} = \frac{0.02}{\sqrt{3}}$ | 0.012 |
| Cross- sectional area | mm² | A | 89.74 | $u_A = \sqrt{\left(\frac{u_W}{W}\right)^2 + \left(\frac{u_H}{H}\right)^2}$ | 0.30 |
| | | Emax | 2.50 | | 0.014 |
| Strain in side A | % | E 0.05 | 0.05 | $u_{\varepsilon} = \frac{1\varepsilon}{100\sqrt{3}}$ | 0.00029 |
| | | £ 0.25 | 0.25 | | 0.0014 |
| Strain in side B | % | Emax | 3.10 | $u_{\varepsilon} = \frac{1\varepsilon}{100\sqrt{3}}$ | 0.018 |

| | | E 0.05 | 0.05 | | 0.00029 |
|---------------------------------|-----------------------|-------------------|----------------|----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|---------|
| | | E 0.25 | 0.25 | | 0.0014 |
| | | F _{max} | 9730 | | 28.09 |
| Force in side A | N | F0.05 | 1405 | $u_F = \frac{0.5F}{100\sqrt{3}}$ | 4.06 |
| | | F _{0.25} | 3662 | | 10.57 |
| | | σ _{max} | 108 | | 0.0010 |
| Applied load in side A | N/mm ² | σ0.05 | 8.87 | $u_{\sigma} = \sigma \sqrt{\left(\frac{u_F}{F}\right)^2 + \left(\frac{u_A}{A}\right)^2}$ | 0.00009 |
| | | O 0.25 | 35.82 | | 0.00034 |
| Elastic modulus side A | MPa | Е | 12570 | $u_{E} = E \sqrt{\frac{u_{\sigma 1}^{2} + u_{\sigma 2}^{2}}{(\sigma_{2} - \sigma_{1})^{2}} + \frac{u_{\varepsilon 1}^{2} + u_{\varepsilon 2}^{2}}{(\varepsilon_{2} - \varepsilon_{1})^{2}}}$ | 89.86 |
| Force in side | N | F _{0.05} | 1149 | 0.5 <i>F</i> | 3.31 |
| В | N F _{0.2} | F _{0.25} | 3026 | $u_F = \frac{1}{100\sqrt{3}}$ | 8.73 |
| Applied load | N/mm ² | 0 0.05 | 6.27 | $\left(\mathcal{U}_{\mathrm{E}} \right)^2 = \left(\mathcal{U}_{\mathrm{A}} \right)^2$ | 0.00006 |
| in side B | 11/11/11 | σ _{0.25} | 29.27 | $u_{\sigma} = \sigma_{\sqrt{\left(\frac{F}{F}\right)}} + \left(\frac{1}{A}\right)$ | 0.00028 |
| Elastic modulus in side B | MPa | E | 10455 | $u_{E} = E \sqrt{\frac{u_{\sigma 1}^{2} + u_{\sigma 2}^{2}}{(\sigma_{2} - \sigma_{1})^{2}} + \frac{u_{\varepsilon 1}^{2} + u_{\varepsilon 2}^{2}}{(\varepsilon_{2} - \varepsilon_{1})^{2}}}$ | 74.74 |
| Elastic modulus | MPa | E | 89.86 74.74 | $u_{E}=\sqrt{u_{E_{1}}^{2}+u_{E_{2}}^{2}}$ | 116.88 |

4.3. ISO 14129 In plane Shear by \pm 45° tensile test

In-plane shear test uncertainties calculation has been done using data from samples **1_Bio_S45_#35#**.

| Measurand | Unit | Symbol | Data | Formula | Result |
|-------------------------|-------------------|---------------------------|-------------|-------------------------------------------------------------------------------------------------|--------|
| Thickness | mm | Н | 4.52/0.08/3 | $u_H = \sqrt{u_{H,A}^2 + u_{H,instr}^2}$ | 0.047 |
| Width | mm | w | 25.05/0.1/3 | $u_W = \sqrt{u_{W,A}^2 + u_{W,instr}^2}$ | 0.13 |
| Cross-sectional area | mm ² | A | 113.22 | $u_A = \sqrt{\left(\frac{u_W}{W}\right)^2 + \left(\frac{u_H}{H}\right)^2}$ | 1.32 |
| | | F _{max} | 3010 | | 8.69 |
| Force | N | F _{0.2} | 212 | $u_F = \frac{0.5F}{100\sqrt{3}}$ | 0.61 |
| | | F0.6 | 999 | | 2.88 |
| | | Tmax | 40.7 | | 0.093 |
| Applied load | N/mm ² | T 0,1 | 5.0 | $u_{\tau} = \frac{1}{2}\tau \sqrt{\left(\frac{u_F}{F}\right)^2 + \left(\frac{u_A}{A}\right)^2}$ | 0.012 |
| | | T 0,5 | 18.8 | | 0.041 |
| | | Ex-max | 1.20 | | 0.0069 |
| Strain x | % | ε _{x-0.2} | 0.07 | $u_{\varepsilon} = \frac{1\varepsilon}{100\sqrt{3}}$ | 0.0004 |
| | | E x-0.6 | 0.27 | | 0.0016 |
| | | ε _{y-max} | 1.30 | | 0.0075 |
| Strain y | % | ε _{y-0.2} | 0.13 | $u_{\varepsilon} = \frac{1\varepsilon}{100\sqrt{3}}$ | 0.0008 |
| | | ε γ-0.6 | 0.33 | | 0.0019 |

| Shear strain | | Υ_{max} | 1.5 | | 0.16 |
|--------------------|-----|------------------|------|--------------------------------------------------------------------------------------------------------------------------------------------|-------|
| | % | Υ _{0.2} | 0.2 | $u_{\gamma} = \sqrt{u_{\varepsilon x}^2 + u_{\varepsilon y}^2}$ | 0.010 |
| | | Y 0.6 | 0.6 | | 0.052 |
| Elastic modulus | MPa | G12 | 1720 | $u_G = G \sqrt{\frac{u_{\tau_1}^2 + u_{\tau_2}^2}{(\tau_2 - \tau_1)^2} + \frac{u_{\gamma_1}^2 + u_{\gamma_2}^2}{(\gamma_2 - \gamma_1)^2}}$ | 18.51 |

4.4. ASTM D2344 Short-beam test

ILSS test uncertainties calculation has been done using data from samples **1_Bio_ILSS_#32#.**

| Measurand | Unit | Symbol | Data | Formula | Result |
|-------------------------|-------------------|------------------|--------------|-----------------------------------------------------------------------------------------------------|--------|
| Thickness | mm | Н | 3.87/0.02/3 | $u_H = \sqrt{u_{H,A}^2 + u_{H,instr}^2}$ | 0.017 |
| Width | mm | W | 13.58/0.16/3 | $u_W = \sqrt{u_{W,A}^2 + u_{W,instr}^2}$ | 0.093 |
| Cross-sectional area | mm² | A | 32.77 | $u_A = A \sqrt{\left(\frac{u_W}{W}\right)^2 + \left(\frac{u_H}{H}\right)^2}$ | 0.20 |
| Force | N | F _{max} | 2010 | $u_{Fm} = \frac{0.5F}{100\sqrt{3}}$ | 17.41 |
| Applied load | N/mm ² | F ^{sbs} | 18.84 | $u_{FSBS} = 0.75 \times F^{SBS} \sqrt{\left(\frac{u_F}{F}\right)^2 + \left(\frac{u_A}{A}\right)^2}$ | 0.15 |

4.5. ASTM D2344 Short-beam test- Flexural tests

ILSS test uncertainties calculation has been done using data from samples **1_Bio_ILSS_#32#.**

| Measurand | Unit | Symbol | Data | Formula | Result |
|-------------------------|-------------------|------------------|--------------|------------------------------------------------------------------------------------------|--------|
| Thickness | mm | Н | 3.87/0.02/3 | $u_H = \sqrt{u_{H,A}^2 + u_{H,instr}^2}$ | 0.017 |
| Width | mm | W | 13.58/0.16/3 | $u_W = \sqrt{u_{W,A}^2 + u_{W,instr}^2}$ | 0.093 |
| Cross-sectional area | mm² | A | 32.77 | $u_A = A \sqrt{\left(\frac{u_W}{W}\right)^2 + \left(\frac{u_H}{H}\right)^2}$ | 0.20 |
| Force | N | F _{max} | 2010 | $u_{Fm} = \frac{0.5F}{100\sqrt{3}}$ | 17.41 |
| Applied load | N/mm ² | σ | 18.84 | $u_{\sigma} = \sigma \sqrt{\left(\frac{u_F}{F}\right)^2 + \left(\frac{u_A}{A}\right)^2}$ | 0.15 |

5. References

- 1. ISO (1993) Guide to the Expression of Uncertainty in Measurement. International Organization for Standardization. Geneva, Switzerland.
- Lekou, D.J., Assimakopoulou, T.T. and Philippidis, T.P. "Estimation of the Uncertainty in Measurement of Composite Material Mechanical Properties During Static Testing". 2011. *Strain*, 47, 430–438.

Annex 1: Tensile test uncertainties calculations

Longitudinal tensile test uncertainties calculation has been done using data from samples **3_Bio_T0_#3#** and **2_Bio_T0g_#29#**.

1. Uncertainty of Thickness

| | Value 1 | Value 2 | Value 3 | Average | SD |
|-----------|---------|---------|---------|---------|------|
| Thickness | 2.14 | 2.17 | 2.13 | 2.15 | 0.08 |

$$u_{H} = \sqrt{u_{H,A}^{2} + u_{H,instr}^{2}}$$
$$u_{H,A} = \frac{s_{H}}{\sqrt{n_{A}}} \qquad u_{H,instr} = \frac{0.02}{\sqrt{3}}$$
$$u_{H,instr} = \frac{0.02}{\sqrt{3}}$$

$$u_H = \sqrt{\left(\frac{0.08}{\sqrt{3}}\right)^2 + \left(\frac{0.02}{\sqrt{3}}\right)^2} = 0.047$$

2. Uncertainty of Width

| | Value 1 | Value 2 | Value 3 | Average | SD |
|-------|---------|---------|---------|---------|------|
| Width | 15.33 | 15.22 | 15.16 | 15.24 | 0.92 |

$$u_{W} = \sqrt{u_{W,A}^{2} + u_{W,instr}^{2}}$$

$$u_{W,inst} = \frac{0.02}{\sqrt{3}}$$

$$u_{W,A} = \frac{s_{W}}{\sqrt{n_{A}}}$$

$$u_{W,inst} = \frac{0.02}{\sqrt{3}}$$

$$u_{W,A} = \frac{0.92}{\sqrt{3}}$$

$$u_{W,A} = \frac{0.92}{\sqrt{3}}$$

$$u_{W,A} = \frac{0.92}{\sqrt{3}}$$

3. Uncertainty of cross-sectional area

| Specimen | Н | w | Α | Uн | Uw |
|--------------|------|-------|-----------------|-----------------|-----------------|
| Specifici | mm | mm | mm ² | mm ² | mm ² |
| 3_Bio_T0_#3# | 2.20 | 15.54 | 34.19 | 0.047 | 0.531 |

$$u_A = \sqrt{\left(\frac{u_W}{W}\right)^2 + \left(\frac{u_H}{H}\right)^2}$$
$$u_A = 34.19\sqrt{\left(\frac{0.531}{15.54}\right)^2 + \left(\frac{0.047}{2.20}\right)^2} = 1.38$$

4. Uncertainty of the applied force

| Specimen | Fm | F0.05 | F 0.25 | |
|--------------|------|-------|---------------|--|
| | N | N | N | |
| 3_Bio_T0_#3# | 5580 | 457 | 1606 | |

$$u_F = \frac{0.5F}{100\sqrt{3}}$$

$$u_{F_m} = \frac{0.5 \ x \ 5580}{100 \ x \ \sqrt{3}} = 16.11$$

$$u_{F_{0.05}} = \frac{0.5 \ x \ 457}{100 \ x \ \sqrt{3}} = 1.32$$

$$u_{F_{m0.25}} = \frac{0.5 \ x \ 1606}{100 \ x \ \sqrt{3}} = 4.63$$

5. Uncertainty on the applied load

| σ _{max} | σ _{0.05} | 0 0.25 | A | UA | Fm | UFm | F _{0.05} | U F0.05 | F _{0.25} | U F0.25 |
|-------------------|-------------------|-------------------|-----------------|-----------------|---------|-------|-------------------|----------------|-------------------|----------------|
| N/mm ² | N/mm ² | N/mm ² | mm ² | mm ² | N | Ν | N | Ν | N | Ν |
| 163.23 | 13.46 | 47.25 | 34.19 | 1.38 | 5580.38 | 16.11 | 457.00 | 1.32 | 1606.00 | 4.63 |

* Sample 3_Bio_T0_#3#

$$u_{\sigma} = \sigma \sqrt{\left(\frac{u_F}{F}\right)^2 + \left(\frac{u_A}{A}\right)^2}$$
$$u_{\sigma max} = 163.23 \sqrt{\left(\frac{16.11}{5580.38}\right)^2 + \left(\frac{1.38}{34.19}\right)^2} = 6.5$$
$$u_{\sigma 0.05} = 13.46 \sqrt{\left(\frac{1.32}{457.00}\right)^2 + \left(\frac{1.38}{34.19}\right)^2} = 0.54$$
$$u_{\sigma 0.25} = 47.25 \sqrt{\left(\frac{4.63}{1606.00}\right)^2 + \left(\frac{1.38}{34.19}\right)^2} = 1.89$$

6. Uncertainty on tensile strain in x

| Emax | E 0.05 | £ 0.25 |
|------|---------------|---------------|
| % | % | % |
| 1.32 | 0.05 | 0.25 |

* Sample **3_Bio_T0_#3#**

$$u_{\varepsilon} = \frac{1\varepsilon}{100\sqrt{3}}$$
$$u_{\varepsilon_m} = \frac{1 x \ 1.32}{100 \ x \ \sqrt{3}} = 0.0076$$
$$u_{\varepsilon_{0.05}} = \frac{1 \ x \ 0.05}{100 \ x \ \sqrt{3}} = 0.00029$$
$$u_{\varepsilon_{0.25}} = \frac{1 \ x \ 0.25}{100 \ x \ \sqrt{3}} = 0.0014$$

7. Uncertainty on elastic modulus

| E | σ1-0.05 | U σ1-0.05 | σ2-0.25 | U σ2-0.25 | E 0.05 | U ε0.05 | E 0.25 | U ε0.25 |
|-------------------|-------------------|-------------------|-------------------|-------------------|---------------|----------------|---------------|----------------|
| N/mm ² | % | % | % | % |
| 10662.60 | 13.46 | 0.54 | 47.25 | 1.89 | 0.05 | 0.00029 | 0.25 | 0.0014 |

* Sample **3_Bio_T0_#3#**

$$u_E = E \sqrt{\frac{u_{\sigma 1}^2 + u_{\sigma 2}^2}{(\sigma_2 - \sigma_1)^2} + \frac{u_{\varepsilon 1}^2 + u_{\varepsilon 2}^2}{(\varepsilon_2 - \varepsilon_1)^2}}$$
$$u_E = 10662.60 \sqrt{\frac{0.54^2 + 1.89^2}{(47.25 - 13.46)^2} + \frac{0.00029^2 + 0.00014^2}{(0.25 - 0.05)^2}} = 620.49$$

8. Uncertainty on tensile strain in y

| ε y1-0.05 | £ y2-0.25 |
|------------------|------------------|
| % | % |
| 0.021 | 0.108 |

*Sample 2_Bio_T0g_#29# tested with gauges

$$u_{\varepsilon} = \frac{1\varepsilon}{100\sqrt{3}}$$
$$u_{\varepsilon_{y_1-0.05}} = \frac{1 \times 0.021}{100 \times \sqrt{3}} = 0.00012$$
$$u_{\varepsilon_{y_2-0.25}} = \frac{1 \times 0.108}{100 \times \sqrt{3}} = 0.00062$$

9. Uncertainty on Poisson's ratio

| ϑ_{21} | £ 0.05 | U ε1-0.05 | E 0.25 | U ε2-0.25 | £ y1-0.05 | U y1-0.05 | E y2-0.25 | Uy2-0.25 |
|------------------|---------------|------------------|---------------|------------------|------------------|------------------|------------------|----------|
| | % | % | % | % | % | % | % | % |
| 0.43 | 0.05 | 0.00029 | 0.25 | 0.0014 | 0.021 | 0.00012 | 0.108 | 0.00062 |

*Sample 2_Bio_T0g_#29# tested with gauges

$$u_{\vartheta} = \vartheta \sqrt{\frac{u_{\varepsilon y1}^{2} + u_{\varepsilon y2}^{2}}{(\varepsilon_{y2} - \varepsilon_{y1})^{2}} + \frac{u_{\varepsilon x1}^{2} + u_{\varepsilon x2}^{2}}{(\varepsilon_{x2} - \varepsilon_{x1})^{2}}}$$
$$u_{\vartheta} = 0.43 \sqrt{\frac{0.00012^{2} + 0.00062^{2}}{(0.108 - 0.021)^{2}} + \frac{0.00029^{2} + 0.0014^{2}}{(0.25 - 0.05)^{2}}} = 0.0032$$

Annex 2: Compression test uncertainties calculations

Longitudinal compression test uncertainties calculation has been done using data from samples

1_Bio_C0_#30#.

1. Uncertainty of thickness

$$u_{H,instr} = \frac{0.02}{\sqrt{3}}$$
$$u_{H} = \frac{0.02}{\sqrt{3}} = 0.012$$

*NOTE-For the compression samples the width has been measured in a single point, as a result $u_{H,A}$ is equal to zero.

2. Uncertainty of width

$$u_{W,inst} = \frac{0.02}{\sqrt{3}}$$
$$u_W = \frac{0.02}{\sqrt{3}} = 0.012$$

*NOTE-For the compression samples the width has been measured in a single point, as a result $u_{W,A}$ is equal to zero.

3. Uncertainty of cross-sectional area

| Specimen | Н | w | Α | Uw | Uн |
|---------------|------|-------|-----------------|-----------------|-----------------|
| opeenien | mm | mm | mm ² | mm ² | mm ² |
| 1_Bio_C0_#30# | 3,62 | 24,79 | 89.74 | 0.012 | 0.012 |

$$u_A = A \sqrt{\left(\frac{u_W}{W}\right)^2 + \left(\frac{u_H}{H}\right)^2}$$
$$u_A = 89.74 \sqrt{\left(\frac{0.012}{24.79}\right)^2 + \left(\frac{0.012}{3.62}\right)^2} = 0.30$$

4. Uncertainty on strain

4.a. Uncertainty on strain in side A

| Emax | E 0.05 | £ 0.25 |
|------|---------------|---------------|
| % | % | % |
| 2.50 | 0.05 | 0.25 |

*Sample **1_Bio_C0_#30#**

$$u_{\varepsilon} = \frac{1\varepsilon}{100\sqrt{3}}$$

$$u_{\varepsilon_m} = \frac{1 \ x \ 2.50}{100 \ x \ \sqrt{3}} = 0.014$$

$$u_{\varepsilon_{0.05}} = \frac{1 \ x \ 0.05}{100 \ x \ \sqrt{3}} = 0.00029$$

$$u_{\varepsilon_{0.25}} = \frac{1 \ x \ 0.25}{100 \ x \ \sqrt{3}} = 0.0014$$

4.b. Uncertainty on strain in side B

| Emax | E 0.05 | E 0.25 |
|------|---------------|---------------|
| % | % | % |
| 3.10 | 0.05 | 0.25 |

*Sample **1_Bio_C0_#30#**

$$u_{\varepsilon} = \frac{1\varepsilon}{100\sqrt{3}}$$
$$u_{\varepsilon_m} = \frac{1 x \ 3.10}{100 \ x \ \sqrt{3}} = 0.018$$
$$u_{\varepsilon_{0.05}} = \frac{1 \ x \ 0.05}{100 \ x \ \sqrt{3}} = 0.00029$$

$$u_{\varepsilon_{0.25}} = \frac{1 \, x \, 0.25}{100 \, x \, \sqrt{3}} = 0.0014$$

5. Uncertainty on applied force

5.a. Uncertainty of the applied force in side A

| Specimen | F 0.05 | F0.25 | |
|---------------|---------------|---------|--|
| | N | N | |
| 1_Bio_C0_#30# | 1405.95 | 3662.10 | |

$$u_F = \frac{0.5F}{100\sqrt{3}}$$

$$u_{F_{0.05}} = \frac{0.5 \ x \ 1405.95}{100 \ x \ \sqrt{3}} = 4.06$$

$$u_{F_{0.25}} = \frac{0.5 \ x \ 3662.10}{100 \ x \ \sqrt{3}} = 10.57$$

5.b. Uncertainty of the applied force in side B

| Specimen | F 0.05 | F _{0.25} | |
|---------------|---------------|--------------------------|--|
| • | N | N | |
| 1_Bio_C0_#30# | 1149.57 | 3026.03 | |

$$u_F = \frac{0.5F}{100\sqrt{3}}$$
$$u_{F_{0.05}} = \frac{0.5 x \, 1149.57}{100 \, x \, \sqrt{3}} = 3.31$$
$$u_{F_{0.25}} = \frac{0.5 \, x \, 3026.03}{100 \, x \, \sqrt{3}} = 8.73$$

5.c. Uncertainty of the maximum applied force

| Specimen | Fm |
|----------|----|
| • | N |

| 1_Bio_C0_#30# | 9730.00 |
|---------------|---------|
|---------------|---------|

$$u_F = \frac{0.5F}{100\sqrt{3}}$$
$$u_{F_m} = \frac{0.5 \, x \, 9730.00}{100 \, x \, \sqrt{3}} = 28.09$$

6. Uncertainty on the applied load

6.a. Uncertainty on the applied load in side A

| Α | UA | F _{0.05} | U F0.05 | σ _{0.05} | F _{0.25} | U F0.25 | σ _{0.25} |
|-----------------|-----------------|-------------------|----------------|-------------------|-------------------|----------------|--------------------------|
| mm ² | mm ² | N | Ν | N/mm ² | N | Ν | N/mm ² |
| 89.74 | 0.30 | 1405.95 | 4.06 | 8.87 | 3662.10 | 10.57 | 35.82 |

*Sample **1_Bio_C0_#30#**

$$u_{\sigma} = \sigma \sqrt{\left(\frac{u_F}{F}\right)^2 + \left(\frac{u_A}{A}\right)^2}$$
$$u_{\sigma 0.05} = 8.87 \sqrt{\left(\frac{4.06}{1405.95}\right)^2 + \left(\frac{0.30}{89.74}\right)^2} = 0.00009$$
$$u_{\sigma 0.25} = 35.82 \sqrt{\left(\frac{10.57}{3662.10}\right)^2 + \left(\frac{0.30}{89.74}\right)^2} = 0.00034$$

6.b. Uncertainty on the applied load in side B

| A | UA | F _{0.05} | U F0.05 | σ _{0.05} | F _{0.25} | U F0.25 | 0 0.25 |
|-----------------|-----------------|-------------------|----------------|-------------------|-------------------|----------------|-------------------|
| mm ² | mm ² | N | Ν | N/mm ² | N | Ν | N/mm ² |
| 89.74 | 0.30 | 1149.57 | 3.31 | 6.27 | 3026.03 | 8.73 | 29.27 |

*Sample **1_Bio_C0_#30#**

$$u_{\sigma} = \sigma \sqrt{\left(\frac{u_F}{F}\right)^2 + \left(\frac{u_A}{A}\right)^2}$$

$$u_{\sigma 0.05} = 6.27 \sqrt{\left(\frac{3.31}{1149.57}\right)^2 + \left(\frac{0.30}{89.74}\right)^2} = 0.00006$$
$$u_{\sigma 0.25} = 29.27 \sqrt{\left(\frac{8.73}{3026.03}\right)^2 + \left(\frac{0.30}{89.74}\right)^2} = 0.00028$$

6.c. Uncertainty of the maximum applied load

| Specimen | Fm | U _{Fm} | σ _{max} | | | | |
|------------------------------|---------|-----------------|-------------------|--|--|--|--|
| | N | N | N/mm ² | | | | |
| 1_Bio_C0_#30# | 9730.00 | 28.09 | 108 | | | | |
| *Sample 1 Bio C0 #30# | | | | | | | |

| *Sample | 1_ | Bio | CO | #30 |
|---------|----|-----|----|-----|
| • | | | | |

$$u_{\sigma} = \sigma \sqrt{\left(\frac{u_F}{F}\right)^2 + \left(\frac{u_A}{A}\right)^2}$$
$$u_{\sigma max} = 108 \sqrt{\left(\frac{28.09}{9730.00}\right)^2 + \left(\frac{0.30}{89.74}\right)^2} = 0.0010$$

7. Uncertainty on elastic modulus

7. a. Uncertainty on elastic modulus in side A

| EA | U _{00.05} | U σ0.25 | σ0.05 | 0 0.25 | E 0.05 | E 0.25 | U ε0.05 | U ε0.25 |
|-------------------|--------------------|-------------------|-------------------|-------------------|---------------|---------------|----------------|----------------|
| N/mm ² | N/mm ² | N/mm ² | N/mm ² | N/mm ² | % | % | % | % |
| 12570,50 | 0.00009 | 0.00034 | 8.87 | 35.82 | 0.05 | 0.25 | 0.00029 | 0.0014 |

```
*Sample 1_Bio_C0_#30#
```

$$u_{E_A} = E_A \sqrt{\frac{u_{\sigma 0.05}^2 + u_{\sigma 0.25}^2}{(\sigma_{0.05} - \sigma_{0.25})^2} + \frac{u_{\varepsilon 0.05}^2 + u_{\varepsilon 0.25}^2}{(\varepsilon_{0.25} - \varepsilon_{0.05})^2}}$$
$$u_{E_A} = 12570.50 \sqrt{\frac{0.00009^2 + 0.00034^2}{(35.82 - 8.87)^2} + \frac{0.00029^2 + 0.0014^2}{(0.25 - 0.05)^2}} = 89.86$$

7.b. Uncertainty on elastic modulus in side B

| Ев | U σ0.05 | U σ0.25 | σ 0.05 | 0 0.25 | E 0.05 | E 0.25 | U ε0.05 | U ε0.25 |
|-------------------|-------------------|-------------------|-------------------|-------------------|---------------|---------------|----------------|----------------|
| N/mm ² | % | % | % | % |
| 10455.00 | 0.00006 | 0.00028 | 6.27 | 29.27 | 0.05 | 0.25 | 0.00029 | 0.0014 |

*Sample **1_Bio_C0_#30#**

$$u_{E_B} = E_B \sqrt{\frac{u_{\sigma 0.05}^2 + u_{\sigma 0.25}^2}{(\sigma_{0.05} - \sigma_{0.25})^2} + \frac{u_{\varepsilon 0.05}^2 + u_{\varepsilon 0.25}^2}{(\varepsilon_{0.25} - \varepsilon_{0.05})^2}}$$

$$u_{E_B} = 10455.00 \sqrt{\frac{0.00006^2 + 0.00028^2}{(29.27 - 6.27)^2} + \frac{0.00029^2 + 0.0014^2}{(0.25 - 0.05)^2}} = 74.74$$

7.c. Uncertainty on elastic modulus

| Specimen | UEA | U _{EB} |
|---------------|-------------------|-------------------|
| Specimen | N/mm ² | N/mm ² |
| 1_Bio_C0_#30# | 89.86 | 74.74 |

$$u_E = \sqrt{u_{E_A}^2 + u_{E_B}^2}$$

 $u_E = \sqrt{89.86^2 + 74.74^2} = 116.88$

Annex 3: In-plane shear test uncertainties calculations

In-plane shear test uncertainties calculation has been done using data from samples **1_Bio_S45_#35#**.

1. Uncertainty of Thickness

| | Value 1 | Value 2 | Value 3 | Average | SD |
|-----------|---------|---------|---------|---------|------|
| Thickness | 4.46 | 4.61 | 4.50 | 4.52 | 0.08 |

| *Sample | 1_ | Bio | _S45_ | #35# |
|---------|----|-----|-------|------|
|---------|----|-----|-------|------|

$$u_H = \sqrt{u_{H,A}^2 + u_{H,instr}^2}$$
$$u_{H,A} = \frac{s_H}{\sqrt{n_A}} \qquad \qquad u_{H,instr} = \frac{0.02}{\sqrt{3}}$$
$$u_{H,A} = \frac{0.08}{\sqrt{3}} \qquad \qquad u_{H,instr} = \frac{0.02}{\sqrt{3}}$$

$$u_H = \sqrt{\left(\frac{0.08}{\sqrt{3}}\right)^2 + \left(\frac{0.02}{\sqrt{3}}\right)^2} = 0.047$$

2. Uncertainty of Width

| | Value 1 | Value 2 | Value 3 | Average | SD |
|-------|---------|---------|---------|---------|------|
| Width | 25.02 | 24.96 | 25.16 | 25.05 | 0.10 |

*Sample **1_Bio_S45_#35#**

$$\begin{split} u_W &= \sqrt{u_{W,A}^2 + u_{W,instr}^2} \\ u_{W,inst} &= \frac{0.02}{\sqrt{3}} \\ u_{W,inst} &= \frac{0.02}{\sqrt{3}} \\ \end{split}$$

$$u_W = \sqrt{\left(\frac{0.10}{\sqrt{3}}\right)^2 + \left(\frac{0.02}{\sqrt{3}}\right)^2} = 0.13$$

3. Uncertainty of cross-sectional area

| Specimen | н | w | A | ЦΗ | Uw |
|----------------|------|-------|-----------------|-----------------|-----------------|
| Specifien | mm | mm | mm ² | mm ² | mm ² |
| 1_Bio_S45_#35# | 4.52 | 25.05 | 113.22 | 0.047 | 0.13 |

$$u_{A} = \sqrt{\left(\frac{u_{W}}{W}\right)^{2} + \left(\frac{u_{H}}{H}\right)^{2}}$$
$$u_{A} = 113.22\sqrt{\left(\frac{0.13}{25.05}\right)^{2} + \left(\frac{0.047}{4.52}\right)^{2}} = 1.32$$

4. Uncertainty of the applied force

| Specimen | Fm | F 0.2 | F0.6 |
|----------------|------|--------------|------|
| Specimen | N | N | N |
| 1_Bio_S45_#35# | 3010 | 212 | 999 |

$$u_F = \frac{0.5F}{100\sqrt{3}}$$
$$u_{Fm} = \frac{0.5 \times 3010}{100 \times \sqrt{3}} = 8.69$$
$$u_{F_{0.2}} = \frac{0.5 \times 212}{100 \times \sqrt{3}} = 0.61$$
$$u_{F_{0.6}} = \frac{0.5 \times 999}{100 \times \sqrt{3}} = 2.88$$

5. Uncertainty on strain in x/y

| Emax-x | Emax-y | E x-0.2 | ε γ-0.2 | E x-0.6 | £ y-0.6 |
|--------|--------|----------------|----------------|----------------|----------------|
| % | % | % | % | % | % |

| 1.20 | 1.30 | 0.07 | 0.13 | 0.27 | 0.33 |
|------|------|------|------|------|------|
|------|------|------|------|------|------|

*Sample 1_Bio_S45_#35#

$$u_{\varepsilon} = \frac{1\varepsilon}{100\sqrt{3}}$$
$$u_{\varepsilon_{x}} = \frac{1 \times 1.20}{100 \times \sqrt{3}} = 0.0069$$
$$u_{\varepsilon_{y}} = \frac{1 \times 1.30}{100 \times \sqrt{3}} = 0.0075$$
$$u_{\varepsilon_{x-0.2}} = \frac{1 \times 0.07}{100 \times \sqrt{3}} = 0.0004$$
$$u_{\varepsilon_{y-0.2}} = \frac{1 \times 0.13}{100 \times \sqrt{3}} = 0.0008$$
$$u_{\varepsilon_{x-0.6}} = \frac{1 \times 0.27}{100 \times \sqrt{3}} = 0.0016$$
$$u_{\varepsilon_{y-0.6}} = \frac{1 \times 0.33}{100 \times \sqrt{3}} = 0.0019$$

6. Uncertainty on shear strain

| Uɛmax-x | Uɛmax-y | U εx-0.2 | U εy-0.2 | U εx-0.6 | U εy-0.6 |
|---------|---------|-----------------|-----------------|-----------------|-----------------|
| % | % | % | % | % | % |
| 0.0069 | 0.0075 | 0.0004 | 0.0008 | 0.0016 | 0.0019 |

$$u_{\gamma} = \sqrt{u_{\varepsilon_{\chi}}^2 + u_{\varepsilon_{y}}^2}$$
$$u_{\gamma \max} = \sqrt{0.0069^2 + 0.0075^2} = 0.010$$
$$u_{\gamma 0.2} = \sqrt{0.0004^2 + 0.0008^2} = 0.00085$$
$$u_{\gamma 0.6} = \sqrt{0.0016^2 + 0.0019^2} = 0.0024$$

7. Uncertainty on the applied load

| A | UA | Fm | UFm | Tmax | F 0.2 | U F0.2 | T 0.2 | F0.6 | U F0.6 | T 0.6 |
|-----------------|-----------------|----|-----|-------------------|--------------|---------------|-------------------|------|---------------|-------------------|
| mm ² | mm ² | Ν | N | N/mm ² | Ν | N | N/mm ² | Ν | N | N/mm ² |

| 113.22 | 1.32 | 3010 | 8.69 | 26.6 | 212 | 0.61 | 1.87 | 999 | 2.88 | 8.82 |
|--------|------|------|------|------|-----|------|------|-----|------|------|
|--------|------|------|------|------|-----|------|------|-----|------|------|

*Sample **1_Bio_S45_#35#**

$$u_{\tau} = \frac{1}{2}\tau \sqrt{\left(\frac{u_F}{F}\right)^2 + \left(\frac{u_A}{A}\right)^2}$$

$$u_{\text{tmax}} = \frac{1}{2} 26.6 \sqrt{\left(\frac{8.69}{3010}\right)^2 + \left(\frac{1.32}{113.22}\right)^2} = 0.16$$

$$u_{\tau 0.2} = \frac{1}{2} 1.87 \sqrt{\left(\frac{0.61}{212}\right)^2 + \left(\frac{1.32}{113.22}\right)^2} = 0.010$$

$$u_{\tau 0.6} = \frac{1}{2} 8.82 \sqrt{\left(\frac{2.88}{999}\right)^2 + \left(\frac{1.32}{113.22}\right)^2} = 0.052$$

8. Uncertainty on elastic modulus

| G 12 | U т0.2 | U т0.6 | T 0.2 | T0.6 | γ0.2 | γ0.6 | U γ0.2 | U γ0.6 |
|-------------------|-------------------|-------------------|-------------------|-------------------|------|------|---------------|---------------|
| N/mm ² | % | % | % | % |
| 1720 | 0.010 | 0.052 | 3.1 | 9.2 | 0.2 | 0.6 | 0.00085 | 0.0024 |

*Sample **1_Bio_S45_#35#**

$$u_{G12} = G_{12} \sqrt{\frac{u_{\tau 1}^2 + u_{\tau 2}^2}{(\tau_2 - \tau_1)^2} + \frac{u_{\gamma 1}^2 + u_{\gamma 2}^2}{(\gamma_2 - \gamma_1)^2}}$$

$$u_{G12} = 1720 \sqrt{\frac{0.010^2 + 0.052^2}{(9.2 - 3.1)^2} + \frac{0.00085^2 + 0.0024^2}{(0.60 - 0.20)^2}} = 18.51$$

Annex 4: ILSS test uncertainties calculations

ILSS test uncertainties calculation has been done using data from samples **1_Bio_ILSS_#32#.**

1. Uncertainty of thickness

| | Value 1 | Value 2 | Value 3 | Average | SD |
|-----------|---------|---------|---------|---------|------|
| Thickness | 3.86 | 3.85 | 3.89 | 3.87 | 0.02 |

^{*}Sample 1_Bio_ILSS_#32#

$$u_{H} = \sqrt{u_{H,A}^{2} + u_{H,instr}^{2}}$$
$$u_{H,A} = \frac{s_{H}}{\sqrt{n_{A}}} \qquad u_{H,inst} = \frac{0.02}{\sqrt{3}}$$
$$u_{H,A} = \frac{0.02}{\sqrt{3}} \qquad u_{H,inst} = \frac{0.02}{\sqrt{3}}$$
$$u_{H} = \sqrt{\left(\frac{0.02}{\sqrt{3}}\right)^{2} + \left(\frac{0.02}{\sqrt{3}}\right)^{2}} = 0.017$$

2. Uncertainty of width

| | Value 1 | Value 2 | Value 3 | Average | SD |
|-------|---------|---------|---------|---------|------|
| Width | 20.49 | 20.75 | 20.79 | 13.58 | 0.16 |

*Sample 1_Bio_ILSS_#32#

$$u_{W} = \sqrt{u_{W,A}^{2} + u_{W,instr}^{2}}$$
$$u_{W,A} = \frac{s_{W}}{\sqrt{n_{A}}} \qquad u_{W,inst} = \frac{0.02}{\sqrt{3}}$$
$$u_{W,A} = \frac{0.16}{\sqrt{3}} \qquad u_{W,inst} = \frac{0.02}{\sqrt{3}}$$
$$u_{W} = \sqrt{\left(\frac{0.16}{\sqrt{3}}\right)^{2} + \left(\frac{0.02}{\sqrt{3}}\right)^{2}} = 0.093$$

3. Uncertainty of cross-sectional area

| Specimen | Н | w | Α | Uн | Uw |
|-----------------|------|-------|-----------------|-------|-------|
| Specifien | mm | mm | mm ² | mm | mm |
| 1_Bio_ILSS_#32# | 2.15 | 15.24 | 32.77 | 0.017 | 0.093 |

$$u_A = A \sqrt{\left(\frac{u_W}{W}\right)^2 + \left(\frac{u_H}{H}\right)^2}$$
$$u_A = 32.77 \sqrt{\left(\frac{0.093}{15.24}\right)^2 + \left(\frac{0.017}{2.15}\right)^2} = 0.20$$

4. Uncertainty of force

| Specimen | Fsbs | |
|-----------------|------|--|
| opeenien | N | |
| 1_Bio_ILSS_#32# | 2010 | |

$$u_{Fm} = \frac{0.5F}{100\sqrt{3}} \qquad \qquad u_{Fm} = \frac{0.5 \times 2010}{100 \times \sqrt{3}} = 17.41$$

5. Uncertainty on the applied load

| Specimen | F ^{max} | UFmax | Α | UA |
|-----------------|-------------------------|-------|-----------------|-----------------|
| opeeniei | N | N | mm ² | mm ² |
| 1_Bio_ILSS_#32# | 2010 | 17.41 | 32.77 | 0.20 |

$$u_{F^{SBS}} = 0.75 \times F^{SBS} \sqrt{\left(\frac{u_{F}}{F}\right)^{2} + \left(\frac{u_{A}}{A}\right)^{2}}$$
$$u_{F^{SDS}} = 0.75 \times 18.84 \sqrt{\left(\frac{17.41}{2010}\right)^{2} + \left(\frac{0.2}{32.77}\right)^{2}} = 0.15$$

Appendix G

The two-way **ANOVA** was selected for the data study because there are two factors affecting the final property, the immersion time (A factor) and the concentration (B factor). Two-way ANOVA allows to compare population means when the populations are classified according to two (categorical) factors. Below is the outline of a two-way ANOVA table, with factors A and B, having **I** and **J** groups, respectively:

| Source | <u>df</u> | SS | MS | F | p-value |
|--------------------------------|------------|------|------|----------|---------|
| Α | I-1 | SSA | MSA | MSA/MSE | |
| В | J-1 | SSB | MSB | MSB/MSE | |
| $\mathbf{A} \times \mathbf{B}$ | (I-1)(J-1) | SSAB | MSAB | MSAB/MSE | |
| Error | n - IJ | SSE | MSE | | |
| Total | n-1 | SST | | | |

Where **n** is the sample size; **df** is degree of freedom; **SS** is sum of squares; **MS** is Mean Square; **F** is used for the F statistics; and **p-value** is the calculated evaluate if the null hypothesis (H_0) or the alternate hypothesis (H_a) are true. The hypothesis definitions are as follow:

H₀: The means of all groups under consideration are equal

H_a: The means are not all equal

In this particular case there are three hypotheses to prove. For the case of the null hypothesis it must be checked whether the means values of all groups are equal under immersion time, concentration and combinative effect.

I.

 $H_0:$ There is no main effect due to immersion time

 H_a : There is a main effect due to immersion time

II.

 H_0 : There is no main effect due to concentration

 $H_{\mbox{\scriptsize a}}$: There is a main effect due to concentration

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III.

H₀: There is no an interaction effect between the two variables

H_a: There is an interaction effect between the two variables

According to this, three different p-values are calculated and compared with the a value at a determinate level of significance. In this case for a 95% level of significance the a = 0.05. If the p-value>a, the H₀ is true, and if the p-value<a, the H_a is true. Additionally if the null hypothesis is true, the F statistic has an F distribution with k-1 and n-k degrees of freedom in the numerator/denominator respectively. If the alternate hypothesis is true, then F tends to be large. We reject H₀ in favour of H_a if the F statistic is sufficiently large.

Once the data was analysed with ANOVA, **Fisher's F test** was used in order to determinate if the average values difference is representative or not. This procedure contemplates to rest two values in order to get a number, and the number must be compared to the LSD value: a rule was introduced in the Excel spreadsheet to colour in red the boxes where **result>LSD**. According to Fisher's method the red boxes mean that the difference between the data is significant enough. In this particular case when the boxes are unmodified means that the modification by immersion time or concentration is not significant enough to affect the mechanical property; and in the other way round, when the box is turned red, the immersion time or concentration is significantly affecting the mechanical property.

Note that for the best statistical treatment, the same amount of measurements is desired, for that reason when there is not that number, it is copied the number just above for the completion of the set of data. See the red numbers in each section first table. In the mechanical tests experimental campaigns it is very common to miss some numbers because of different reasons such as software mistake, extensometer data recording stop or load-cell data recording problems.

In the following sections the calculations resume for all the properties are shown.

Longitudinal modulus E₁

| _ | | | | | B: Concentration | | |
|--------------|-----------|--------------------|--------------------|-------------------|------------------|--------------------|----------------------|
| E moo | dulus LTT | 0,25 | 0,5 | | 1,5 | | Total Y _i |
| A: Inmersion | 1 | 8961,39 | 12602,70 | 10581,00 | 9685,35 | 4301,69 | |
| time | | 10620,20 | 9259,02 | 9275,18 | 8122,03 | 5172,65 | |
| | | 10279,60 | 9191,33 | 10005,20 | 9049,65 | 4362,05 | |
| | | 9291,24 | 10096,00 | 10949,40 | 6278,69 | 3972,95 | |
| | | 8935,08 | 11667,90 | 10150,70 | 7453,10 | 3947,60 | |
| | | 9014,91 99244,52 | 10190,30 103301,76 | 9962,64 99664,09 | 6385,00 74400,5 | 9 3840,76 41519,02 | |
| | | 10648,80 | 9419,04 | 9741,82 | 7358,69 | 3876,28 | |
| | | 10253,10 | 95/9,96 | 8906,97 | /011,46 | 4066,06 | 110120.00 |
| | | 11093,50 | 9905,71 | 10578,00 | /054,89 | 3/58,05 | 418129,98 |
| | 2 | 10140,70 | 10792.90 | 9513,10 | 11720.10 | 4220,93 | |
| | 3 | 10779,70 | 10763,00 | 10720.10 | 9705 90 | 2120.40 | |
| | | 1115730 | 9367.06 | 8300 13 | 10617.80 | 3176.77 | |
| | | 10047.80 | 9371 11 | 9561.48 | 8261 29 | 3257.43 | |
| | | 1111540 | 8288.80 | 10529.80 | 9030.97 | 316617 | |
| | | 10032.20 102447.39 | 8990.58 94983.90 | 11246.10 98882.06 | 9712.83 97567.1 | 3026.69 31473.32 | |
| | | 10414.00 | 9164.77 | 10906.70 | 9803.28 | 2974.85 | |
| | | 10754,10 | 9529,20 | 12360,80 | 9587,87 | 3119,62 | |
| | | 8479,27 | 10042,40 | 8577,66 | 9708,06 | 3442,36 | 425353,77 |
| | | 9282,72 | 9181,88 | 8267,36 | 10400,10 | 3292,76 | |
| | 12 | 10291,10 | 10250,70 | 6358,32 | 9015,98 | 3063,85 | |
| | | 10888,00 | 9672,13 | 6959,77 | 10591,10 | 3692,20 | |
| | | 11765,40 | 11265,80 | 7476,57 | 9423,22 | 3641,65 | |
| | | 10101,10 | 10422,60 | 7969,94 | 8872,58 | 3450,86 | |
| | | 10025,30 | 10827,80 | 7351,32 | 10723,90 | 2888,33 | |
| | | 9638,86 99957,31 | 10288,10 102523,79 | 6819,72 76630,27 | 12220,00 101363, | 0 3102,64 32622,60 | |
| | | 10185,40 | 10740,00 | 6794,41 | 11125,00 | 3065,34 | |
| | | 9465,76 | 8572,76 | 7283,29 | 9709,50 | 3391,70 | |
| | | 8618,24 | 10059,60 | 9987,64 | 9747,39 | 3111,82 | 413097,17 |
| | 24 | 8978,15 | 10424,30 | 9629,29 | 9934,53 | 3214,21 | |
| | 24 | 9233,32 | 7605.99 | 11206 20 | 6126.96 | 2402.02 | |
| | | 9020,97 | 8214.20 | 8023.67 | 6093 36 | 3403,03 | |
| | | 10214.00 | 9690 30 | 7021.02 | 6647 37 | 3143 20 | |
| | | 930913 9384242 | 9522.93 88133.05 | 8917.43 85785.94 | 6416.69 66187.7 | 7 345743 3362343 | |
| | | 8310.75 | 7544.60 | 10047.90 | 5666.63 | 3328.32 | |
| | | 8564,69 | 9181,14 | 8815,23 | 6736,91 | 3559,90 | |
| | | 10950,60 | 7259,32 | 6907,00 | 6385,00 | 3456,91 | |
| | | 8826,16 | 8229,67 | 8611,93 | 8007,56 | 3066,12 | 367572,11 |
| | | 9204,06 | 9958,11 | 9074,64 | 7593,09 | 2848,04 | |
| | Total Y. | 395491,64 | 388942,5 | 360962,36 | 339518,16 | 139238,37 | Y = 1624153,03 |

| Source | SS | df | MS | F ₀ | p-value |
|-----------|---------------|-----|--------------|----------------|-------------------------|
| Efecto A | 40975430,51 | 3 | 13658476,84 | 16,08 | 0,0000000026535527399 |
| Efecto B | 1126725565,78 | 4 | 281681391,45 | 331,54 | 0,000000000000000000000 |
| Efecto AB | 110512596,82 | 12 | 9209383,07 | 10,84 | 0,0000000000000004499 |
| Error | 152931675,25 | 180 | 849620,42 | | |
| Total | 1431145268,4 | 199 | | | |

n= 10

| t _(0,05;180) | 1,973 |
|-------------------------|---------|
| LSD | 813,402 |

Immersion time effect (A)

| B ₁ A ₁ . B ₁ A ₂ | 320,29 | B ₂ A ₁ . B ₂ A ₂ | 831,79 | B ₃ A ₁ . B ₃ A ₂ | 78,20 | B ₄ A ₁ . B ₄ A ₂ | 2316,65 | B ₅ A ₁ . B ₅ A ₂ | 1004,57 |
|---------------------------------------------------------------|--------|---------------------------------------------------------------|---------|---------------------------------------------------------------|---------|---------------------------------------------------------------|---------|---------------------------------------------------------------|---------|
| B ₁ A ₁ _ B ₁ A ₃ | 71,28 | B ₂ A ₁ _ B ₂ A ₃ | 77,80 | B ₃ A ₁ _ B ₃ A ₃ | 2303,38 | B ₄ A ₁ B ₄ A ₃ | 2696,26 | B ₅ A ₁ _ B ₅ A ₃ | 889,64 |
| B ₁ A ₁ _ B ₁ A ₄ | 540,21 | B ₂ A ₁ _ B ₂ A ₄ | 1516,87 | B ₃ A ₁ _ B ₃ A ₄ | 1387,82 | B ₄ A ₁ B ₄ A ₄ | 821,33 | B ₅ A ₁ _ B ₅ A ₄ | 789,56 |
| B ₁ A ₂ _ B ₁ A ₃ | 249,01 | B ₂ A ₂ _ B ₂ A ₃ | 753,99 | B ₃ A ₂ _ B ₃ A ₃ | 2225,18 | B ₄ A ₂ _ B ₄ A ₃ | 379,61 | B ₅ A ₂ _ B ₅ A ₃ | 114,93 |
| B ₁ A ₂ . B ₁ A ₄ | 860,50 | B ₂ A ₂ _ B ₂ A ₄ | 685,09 | B ₃ A ₂ . B ₃ A ₄ | 1309,61 | B ₄ A ₂ _ B ₄ A ₄ | 3137,98 | B ₅ A ₂ . B ₅ A ₄ | 215,01 |
| B ₁ A ₃ _ B ₁ A ₄ | 611,49 | B ₂ A ₃ _ B ₂ A ₄ | 1439,07 | B ₃ A ₃ _ B ₃ A ₄ | 915,57 | B ₄ A ₃ B ₄ A ₄ | 3517,59 | B ₅ A ₃ _ B ₅ A ₄ | 100,08 |

| $B_1A_1 = B_2A_1$ | 405,72 | $B_1A_2 = B_2A_2$ | 746,35 | B ₁ A ₃ _ B ₂ A ₃ | 256,65 | $B_1A_4 = B_2A_4$ | 570,94 |
|-------------------|---------|---------------------------------------------------------------|---------|---------------------------------------------------------------|---------|---------------------------------------------------------------|---------|
| $B_1A_1 = B_3A_1$ | 41,96 | $B_1A_2 - B_3A_2$ | 356,53 | B ₁ A ₃ _ B ₃ A ₃ | 2332,70 | $B_1A_4 - B_3A_4$ | 805,65 |
| $B_1A_1 = B_4A_1$ | 2484,39 | $B_1A_2 - B_4A_2$ | 488,03 | B_1A_3 _ B_4A_3 | 140,59 | $B_1A_4 - B_4A_4$ | 2765,52 |
| $B_1A_1 = B_5A_1$ | 5772,55 | $B_1A_2 - B_5A_2$ | 7097,41 | B ₁ A _{3 -} B ₅ A ₃ | 6733,47 | $B_1A_4 - B_5A_4$ | 6021,90 |
| $B_2A_1 = B_3A_1$ | 363,77 | $B_2A_2 - B_3A_2$ | 389,82 | B ₂ A ₃ _ B ₃ A ₃ | 2589,35 | $B_2A_4 - B_3A_4$ | 234,71 |
| $B_2A_1 = B_4A_1$ | 2890,12 | B_2A_2 _ B_4A_2 | 258,32 | $B_2A_3 = B_4A_3$ | 116,06 | $B_2A_4 - B_4A_4$ | 2194,58 |
| $B_2A_1 = B_5A_1$ | 6178,27 | $B_2A_2 - B_5A_2$ | 6351,06 | $B_2A_3 - B_5A_3$ | 6990,12 | $B_2A_4 - B_5A_4$ | 5450,96 |
| $B_3A_1 = B_4A_1$ | 2526,35 | $B_3A_2 - B_4A_2$ | 131,50 | B ₃ A _{3 -} B ₄ A ₃ | 2473,29 | $B_3A_4 - B_4A_4$ | 1959,87 |
| $B_3A_1 = B_5A_1$ | 5814,51 | B ₃ A ₂ _ B ₅ A ₂ | 6740,87 | B ₃ A ₃ _ B ₅ A ₃ | 4400,77 | B ₃ A ₄ _ B ₅ A ₄ | 5216,25 |
| $B_4A_1 = B_5A_1$ | 3288,16 | B ₄ A ₂ _ B ₅ A ₂ | 6609,38 | B ₄ A ₃ _ B ₅ A ₃ | 6874,06 | $B_4A_4 = B_5A_4$ | 3256,38 |

Longitudinal strength σ_1

| S4- | enable TT | | | | | | |
|-----------|------------------------|----------------|----------------|----------------|----------------|--------------|----------------------|
| Su | engui LTT | 0,25 | 0,5 | 1 | 1,5 | 2 | Total Y ₁ |
| A: | 1 | 121,40 | 136,10 | 118,18 | 93,84 | 74,66 | |
| Inmersion | | 110,26 | 127,18 | 87,36 | 88,90 | 65,10 | |
| time | | 114,07 | 109,97 | 103,14 | 91,05 | 67,49 | |
| | | 112,47 | 122,63 | 110,84 | 79,67 | 68,02 | |
| | | 111,26 | 116,70 | 94,16 | 83,17 | 65,56 | |
| | | 114,85 1179,91 | 117,02 1192,08 | 95,55 1006,22 | 74,63 842,03 | 64,80 664,49 | |
| | | 128,46 | 108,99 | 103,80 | 79,29 | 65,41 | |
| | | 116,44 | 115,61 | 99,12 | 80,78 | 60,94 | |
| | | 125,57 | 113,29 | 91,43 | 80,45 | 61,64 | 4884,74 |
| | | 125,13 | 124,59 | 102,64 | 90,26 | 70,89 | |
| | 3 | 122,72 | 132,61 | 115,72 | 99,84 | 52,75 | |
| | | 127,86 | 114,36 | 113,74 | 92,07 | 53,89 | |
| | | 122,85 | 120,63 | 96,03 | 100,05 | 57,85 | |
| | | 114,51 | 113,70 | 111,97 | 93,31 | 49,14 | |
| | | 117,99 | 99,62 | 114,86 | 91,66 | 53,14 | |
| | | 107,70 1191,42 | 100,56 1113,55 | 107,64 1093,34 | 89,72 941,62 | 53,41 523,22 | |
| | | 129,00 | 100,67 | 114,26 | 91,21 | 48,90 | |
| | | 114,10 | 114,57 | 108,19 | 86,64 | 53,63 | |
| | | 120,83 | 112,50 | 104,11 | 100,56 | 50,81 | 4863,15 |
| | | 113,87 | 104,33 | 106,81 | 96,57 | 49,71 | |
| | 12 | 111,78 | 105,43 | 92,10 | 105,28 | 60,63 | |
| | | 113,39 | 101,35 | 107,19 | 102,43 | 60,56 | |
| | | 112,32 | 113,10 | 92,60 | 97,03 | 56,68 | |
| | | 114,03 | 111,41 | 101,66 | 98,82 | 61,51 | |
| | | 105,34 | 108,51 | 96,39 | 103,56 | 56,30 | |
| | | 107,64 1123,85 | 122,91 1089,48 | 96,85 1008,73 | 100,18 1007,09 | 60,39 579,27 | |
| | | 119,26 | 102,02 | 110,12 | 97,46 | 54,37 | |
| | | 121,12 | 99,26 | 100,15 | 97,40 | 55,87 | |
| | | 110,93 | 106,35 | 104,96 | 100,53 | 58,11 | 4808,41 |
| | | 108,04 | 119,14 | 106,72 | 104,40 | 54,86 | |
| | 24 | 101,75 | 106,72 | 108,13 | 87,73 | 52,58 | |
| | | 107,18 | 97,84 | 104,73 | 95,68 | 55,41 | |
| | | 111,47 | 108,54 | 92,67 | 86,58 | 55,39 | |
| | | 125,58 | 116,41 | 91,88 | 93,00 | 53,23 | |
| | | 105,38 1063,92 | 108,33 1061,66 | 101,63 976,20 | 90,83 858,07 | 48,85 525,66 | |
| | | 101,82 | 102,05 | 104,97 | 84,40 | 58,32 | |
| | | 92,07 | 103,05 | 94,67 | 76,62 | 50,73 | |
| | | 109,45 | 113,14 | 84,89 | 83,73 | 53,46 | |
| | | 99,31 | 102,50 | 95,01 | 79,81 | 47,48 | 4485,50 |
| | | 109,91 | 103,08 | 97,63 | 79,69 | 50,21 | |
| | Total Y _{*/*} | 4559,10 | 4456,77 | 4084,49 | 3648,81 | 2292,63 | Y= 19041,80 |

| FV | SC | GDL | СМ | F ₀ | p-value |
|-----------|----------|-----|----------|----------------|---------|
| Efecto A | 2077,83 | 3 | 692,61 | 16,65 | 0,0001 |
| Efecto B | 84579,46 | 4 | 21144,87 | 508,28 | 0,0000 |
| Efecto AB | 3740,49 | 12 | 311,71 | 7,49 | 0,0007 |
| Error | 7488,08 | 180 | 41,60 | | |
| Total | 97885,9 | 199 | | | |

n= 10

| t _(0,05;180) | 1,973 |
|-------------------------|-------|
| LSD | 5,692 |

Immersion time effect (A)

| B ₁ A ₁ . B ₁ A ₂ | 1,15 | B ₂ A ₁ . B ₂ A ₂ | 7,85 | B ₃ A ₁ . B ₃ A ₂ | 8,71 | B ₄ A ₁ . B ₄ A ₂ | 9,96 | B ₅ A ₁ . B ₅ A ₂ | 14,13 |
|---------------------------------------------------------------|-------|---------------------------------------------------------------|-------|---------------------------------------------------------------|-------|---------------------------------------------------------------|-------|---------------------------------------------------------------|-------|
| B ₁ A ₁ _ B ₁ A ₃ | 5,61 | B ₂ A ₁ _ B ₂ A ₃ | 10,26 | B ₃ A ₁ _ B ₃ A ₃ | 0,25 | B ₄ A ₁ _ B ₄ A ₃ | 16,51 | B ₅ A ₁ . B ₅ A ₃ | 8,52 |
| B ₁ A ₁ . B ₁ A ₄ | 11,60 | B ₂ A ₁ . B ₂ A ₄ | 13,04 | B ₃ A ₁ _ B ₃ A ₄ | 3,00 | B ₄ A ₁ . B ₄ A ₄ | 1,60 | B ₅ A ₁ . B ₅ A ₄ | 13,88 |
| $B_1A_2 - B_1A_3$ | 6,76 | B ₂ A ₂ _ B ₂ A ₃ | 2,41 | B ₃ A ₂ _ B ₃ A ₃ | 8,46 | B ₄ A ₂ _ B ₄ A ₃ | 6,55 | B ₅ A ₂ . B ₅ A ₃ | 5,60 |
| $B_1A_2 = B_1A_4$ | 12,75 | B ₂ A ₂ _ B ₂ A ₄ | 5,19 | B ₃ A ₂ _ B ₃ A ₄ | 11,71 | B ₄ A ₂ _ B ₄ A ₄ | 8,36 | B ₅ A ₂ _ B ₅ A ₄ | 0,24 |
| B ₁ A ₃ _ B ₁ A ₄ | 5,99 | B ₂ A ₃ . B ₂ A ₄ | 2,78 | B ₃ A ₃ . B ₃ A ₄ | 3,25 | B ₄ A ₃ . B ₄ A ₄ | 14,90 | B ₅ A ₃ . B ₅ A ₄ | 57,93 |

| $B_1A_1 = B_2A_1$ | 1,22 | $B_1A_2 = B_2A_2$ | 7,79 | B ₁ A ₃ _ B ₂ A ₃ | 3,44 | $B_1A_4 = B_2A_4$ | 0,23 |
|-------------------|-------|---------------------|-------|---------------------------------------------------------------|-------|---------------------------------------------------------------|-------|
| $B_1A_1 = B_3A_1$ | 17,37 | B_1A_2 . B_3A_2 | 9,81 | $B_1A_3 - B_3A_3$ | 11,51 | B ₁ A ₄ _ B ₃ A ₄ | 8,77 |
| $B_1A_1 = B_4A_1$ | 33,79 | B_1A_2 . B_4A_2 | 24,98 | $B_1A_3 = B_4A_3$ | 11,68 | $B_1A_4 = B_4A_4$ | 20,58 |
| $B_1A_1 = B_5A_1$ | 51,54 | B_1A_2 _ B_5A_2 | 66,82 | $B_1A_3 = B_5A_3$ | 54,46 | $B_1A_4 - B_5A_4$ | 53,83 |
| $B_2A_1 = B_3A_1$ | 18,59 | B_2A_2 _ B_3A_2 | 2,02 | $B_2A_3 - B_3A_3$ | 8,07 | $B_2A_4 - B_3A_4$ | 8,55 |
| $B_2A_1 - B_4A_1$ | 35,00 | B_2A_2 _ B_4A_2 | 17,19 | B ₂ A ₃ _ B ₄ A ₃ | 17,19 | $B_2A_4 - B_4A_4$ | 20,36 |
| $B_2A_1 = B_5A_1$ | 52,76 | B_2A_2 _ B_5A_2 | 59,03 | B ₂ A ₃ _ B ₅ A ₃ | 51,02 | $B_2A_4 - B_5A_4$ | 53,60 |
| $B_3A_1 = B_4A_1$ | 16,42 | $B_3A_2 - B_4A_2$ | 15,17 | $B_3A_3 - B_4A_3$ | 0,16 | $B_3A_4 - B_4A_4$ | 11,81 |
| $B_3A_1 = B_5A_1$ | 34,17 | B_3A_2 _ B_5A_2 | 57,01 | B ₃ A ₃ _ B ₅ A ₃ | 42,95 | $B_3A_4 - B_5A_4$ | 45,05 |
| $B_4A_1 = B_5A_1$ | 17,75 | $B_4A_2 - B_5A_2$ | 41,84 | B ₄ A ₃ _ B ₅ A ₃ | 42,78 | $B_4A_4 = B_5A_4$ | 33,24 |

Longitudinal strain ϵ_1

| | tenin ITT | | B: Concentration | | | | |
|-----------|------------------------|------------|------------------|------------|------------|------------|----------------------|
| 3 | | 0,25 | | | | | Total Y ₁ |
| A: | 1 | 1,46 | 1,54 | 1,29 | 1,62 | 3,59 | |
| Inmersion | | 1,13 | 1,58 | 1,03 | 1,55 | 2,84 | |
| time | | 1,32 | 1,48 | 1,19 | 1,45 | 3,11 | |
| | | 1,47 | 1,51 | 1,15 | 1,34 | 3,53 | |
| | | 1,49 | 1,52 | 1,00 | 1,39 | 3,59 | |
| | | 1,54 14,22 | 1,42 15,17 | 1,04 11,38 | 1,56 15,43 | 3,54 35,08 | |
| | | 1,53 | 1,53 | 1,13 | 1,64 | 3,54 | |
| | | 1,36 | 1,60 | 1,21 | 1,42 | 3,43 | |
| | | 1,38 | 1,38 | 1,21 | 1,52 | 3,79 | 91,27 |
| | | 1,52 | 1,62 | 1,14 | 1,93 | 4,12 | |
| | 3 | 1,27 | 1,59 | 1,26 | 1,77 | 3,84 | |
| | | 1,28 | 1,58 | 1,16 | 1,62 | 3,48 | |
| | | 1,32 | 1,66 | 1,10 | 1,62 | 4,26 | |
| | | 1,34 | 1,48 | 1,26 | 1,87 | 3,27 | |
| | | 1,36 | 1,45 | 1,18 | 1,72 | 4,15 | |
| | | 1,22 13,30 | 1,50 14,86 | 1,06 12,17 | 1,55 16,27 | 5,17 38,77 | |
| | | 1,46 | 1,15 | 1,23 | 1,54 | 3,78 | |
| | | 1,30 | 1,54 | 1,12 | 1,32 | 3,78 | |
| | | 1,51 | 1,46 | 1,30 | 1,79 | 3,72 | 95,37 |
| | | 1,24 | 1,44 | 1,50 | 1,47 | 3,31 | |
| | 12 | 1,30 | 1,25 | 1,28 | 1,52 | 5,40 | |
| | | 1,36 | 1,26 | 1,47 | 1,42 | 5,03 | |
| | | 1,31 | 1,34 | 1,25 | 1,35 | 3,79 | |
| | | 1,42 | 1,44 | 1,38 | 1,40 | 4,51 | |
| | | 1,33 | 1,40 | 1,25 | 1,46 | 4,72 | |
| | | 1,36 13,96 | 1,63 13,99 | 1,50 13,99 | 1,52 14,15 | 4,98 47,43 | |
| | | 1,46 | 1,19 | 1,55 | 1,30 | 4,65 | |
| | | 1,63 | 1,54 | 1,43 | 1,37 | 4,41 | |
| | | 1,42 | 1,39 | 1,44 | 1,39 | 5,00 | 103,52 |
| | | 1,38 | 1,55 | 1,45 | 1,42 | 4,95 | |
| | 24 | 1,19 | 1,34 | 1,32 | 1,68 | 3,32 | |
| | | 1,16 | 1,36 | 1,35 | 2,11 | 3,92 | |
| | | 1,23 | 1,34 | 1,46 | 1,66 | 3,60 | |
| | | 1,36 | 1,55 | 2,52 | 1,83 | 4,94 | |
| | | 1,24 12,70 | 1,52 14,73 | 1,43 15,00 | 1,76 16,58 | 4,25 42,25 | |
| | | 1,10 | 1,68 | 1,43 | 1,75 | 5,25 | |
| | | 1,13 | 1,52 | 1,40 | 1,16 | 4,11 | |
| | | 1,46 | 1,52 | 1,33 | 1,74 | 4,11 | |
| | | 1,36 | 1,51 | 1,34 | 1,40 | 3,86 | 101,27 |
| | | 1,46 | 1,37 | 1,43 | 1,49 | 4,88 | |
| | Total Y _{*/*} | 54,18 | 58,74 | 52,55 | 62,42 | 163,53 | Y= 391,43 |
| | | | | | | | |

| FV | | | | F ₀ | p-value |
|-----------|--------|-----|-------|----------------|---------|
| Efecto A | 1,86 | 3 | 0,62 | 8,81 | 0,0023 |
| Efecto B | 228,59 | 4 | 57,15 | 809,80 | 0,0000 |
| Efecto AB | 7,83 | 12 | 0,65 | 9,24 | 0,0003 |
| Error | 12,70 | 180 | 0,07 | | |
| Total | 251.0 | 199 | | | |

n= 10

| t _(0,05;180) | 1,973 |
|-------------------------|-------|
| LSD | 0,234 |

Immersion time effect (A)

| B ₁ A ₁ . B ₁ A ₂ | 0,09 B ₂ A ₁ B ₂ A ₂ | 0,03 B ₃ A ₁ B ₃ A ₂ | 0,08 B ₄ A ₁ B ₄ A ₂ | 0,08 B ₅ A ₁ . B ₅ A ₂ | 0,37 |
|---------------------------------------------------------------|------------------------------------------------------------------|------------------------------------------------------------------|--------------------------------------------------------------------|--------------------------------------------------------------------|------|
| B_1A_1 . B_1A_3 | 0,03 B ₂ A ₁ B ₂ A ₃ | 0,12 B ₃ A ₁ B ₃ A ₃ | 0,26 B ₄ A ₁ . B ₄ A ₃ | 0,13 B ₅ A ₁ . B ₅ A ₃ | 1,24 |
| B ₁ A ₁ _ B ₁ A ₄ | 0,15 B ₂ A ₁ B ₂ A ₄ | 0,04 B ₃ A ₁ B ₃ A ₄ | 0,36 B ₄ A ₁ B ₄ A ₄ | 0,11 B ₅ A ₁ B ₅ A ₄ | 0,72 |
| B ₁ A ₂ _ B ₁ A ₃ | 0,07 B ₂ A ₂ B ₂ A ₃ | 0,09 B ₃ A ₂ B ₃ A ₃ | 0,18 B ₄ A ₂ B ₄ A ₃ | 0,21 B ₅ A ₂ B ₅ A ₃ | 0,87 |
| B ₁ A ₂ _ B ₁ A ₄ | 0,06 B ₂ A ₂ B ₂ A ₄ | 0,01 B ₃ A ₂ B ₃ A ₄ | 0,28 B ₄ A ₂ B ₄ A ₄ | 0,03 B ₅ A ₂ B ₅ A ₄ | 0,35 |
| B ₁ A ₃ _ B ₁ A ₄ | 0,13 B ₂ A ₃ B ₂ A ₄ | 0,07 B ₃ A ₃ B ₃ A ₄ | 0.10 B ₄ A ₃ B ₄ A ₄ | 0,24 B5A3 B5A4 | 0.52 |

| $B_1A_1 = B_2A_1$ | 0,10 | $B_1A_2 - B_2A_2$ | 0,16 | B ₁ A ₃ _ B ₂ A ₃ | 0,00 | $B_1A_4 - B_2A_4$ | 0,20 |
|-------------------|------|---------------------|------|---------------------------------------------------------------|------|---------------------------------------------------------------|------|
| $B_1A_1 = B_3A_1$ | 0,28 | $B_1A_2 - B_3A_2$ | 0,11 | B ₁ A ₃ _ B ₃ A ₃ | 0,00 | $B_1A_4 - B_3A_4$ | 0,23 |
| $B_1A_1 = B_4A_1$ | 0,12 | B_1A_2 _ B_4A_2 | 0,30 | $B_1A_3 = B_4A_3$ | 0,02 | $B_1A_4 - B_4A_4$ | 0,39 |
| $B_1A_1 = B_5A_1$ | 2,09 | $B_1A_2 - B_5A_2$ | 2,55 | B ₁ A ₃ _ B ₅ A ₃ | 3,35 | $B_1A_4 - B_5A_4$ | 2,95 |
| $B_2A_1 = B_3A_1$ | 0,38 | $B_2A_2 - B_3A_2$ | 0,27 | $B_2A_3 = B_3A_3$ | 0,00 | $B_2A_4 - B_3A_4$ | 0,03 |
| $B_2A_1 - B_4A_1$ | 0,03 | B_2A_2 _ B_4A_2 | 0,14 | B ₂ A ₃ _ B ₄ A ₃ | 0,02 | $B_2A_4 - B_4A_4$ | 0,19 |
| $B_2A_1 = B_5A_1$ | 1,99 | $B_2A_2 - B_5A_2$ | 2,39 | B ₂ A ₃ _ B ₅ A ₃ | 3,34 | $B_2A_4 - B_5A_4$ | 2,75 |
| $B_3A_1 - B_4A_1$ | 0,40 | B_3A_2 . B_4A_2 | 0,41 | B ₃ A ₃ _ B ₄ A ₃ | 0,02 | $B_3A_4 - B_4A_4$ | 0,16 |
| $B_3A_1 = B_5A_1$ | 2,37 | $B_3A_2 - B_5A_2$ | 2,66 | B ₃ A ₃ _ B ₅ A ₃ | 3,34 | $B_3A_4 - B_5A_4$ | 2,72 |
| $B_4A_1 - B_5A_1$ | 1,96 | B_4A_2 _ B_5A_2 | 2,25 | B_4A_3 . B_5A_3 | 3,33 | B ₄ A ₄ _ B ₅ A ₄ | 2,57 |

Transverse modulus E₂

| _ | B: Concentration | | | | | | |
|----------------------|------------------|------------------------------------------------------------------------------------------------------------------------|--------------------------------------------------------------------------------------------------|-------------------------------------------------------------------------------------------------------------|-----------------------------------------------------------------------------------------------------------------------|-----------------------------------------------------------------------------------------------------------|----------------------|
| E moo | dulus LTT | 0,25 | 0,5 | 1 | 1,5 | | Total Y ₁ |
| A: Inmersion time | 1 | 3759,37 3381,06 3409,10 3016,79 2810,50 2785,04 2785,04 27262,43 2984,97 2727,53 | 3290,28 3091,16 3124,89 2667,47 2242,68 3252,59 2396,06 2507,79 2543,20 | 2839,04 2833,20 3059,49 3067,19 2826,85 3263,50 3178,15 3874,40 3535,07 | 3284,11 4181,58 3930,83 3697,35 3689,78 3563,20 3806,91 4102,67 4486,46 | 903,74 903,74 820,17 826,06 791,32 1500,50 8346,80 1224,41 717,08 659,77 | 124416,49 |
| | 3 | 3299,92 3261,10 3116,33 2531,06 2155,53 2096,07 24582,98 2685,28 2949,79 2487,90 | 2431,92 2524,89 2385,04 2483,01 2578,59 2661,60 2333,97 2418,49 2507,49 | 3143,40 2824,21 2596,01 3093,05 3006,11 3695,65 30403,69 3791,45 3844,45 3809,36 | 2962,27 3204,95 3284,36 2758,34 3754,58 2923,68 2923,68 2923,68 2923,68 2923,69 4113,44 | 1040,47 1009,84 1045,03 1027,43 844,43 1028,20 8263,52 1066,15 600,99 600,99 | 115396,83 |
| | 12 | 1965,90 1954,17 1995,96 2244,50 2157,22 2003,56 2003,56 20291,48 2391,17 2819,02 2759,98 | 4100,54 3132,84 2919,23 3215,11 2956,644 4109,66 2942,66 2942,66 2468,98 | 3007,81 3736,54 3600,49 4319,40 4869,34 4466,06 32881,22 3104,42 2888,68 2888,68 | 4303,16 4303,16 4200,54 4200,54 4337,65 4241,07 4576,94 4576,94 5636,35 | 1069,30 1115,82 1135,65 1032,37 1102,80 1357,83 10526,80 1796,98 958,03 958,03 | 132879,97 |
| | 24 | 3163,61 3004,76 2681,53 2561,01 2116,07 24585,24 3779,44 3120,60 2604,24 2553,98 | 2986,58 2483,84 2589,97 2879,95 2717,42 2945,75 3036,87 2855,23 3490,76 | 3015,66 3015,66 3090,58 3462,95 3384,38 2914,80 3282,63 3035,85 3256,58 | 4342,90 4632,38 5727,21 4682,86 4787,59 4458,65 4453,18 4503,68 4276,15 | 905,30 855,47 804,15 867,10 924,19 906,69 906,69 909,82 679,44 | 128661,66 |
| | I otal Y | 96986,49 | 102415,61 | 120220,89 | 14/028,48 | 34/03,4/3 | Y = 501354,943 |

| Source | SS | df | MS | F ₀ | p-value |
|-----------|--------------|-----|-------------|----------------|-------------------------|
| Efecto A | 3352082,31 | 3 | 1117360,77 | 3,98 | 0,0090888141792004200 |
| Efecto B | 32826045,12 | 4 | 8206511,28 | 29,23 | 0,00000000000000000030 |
| Efecto AB | 152318640,91 | 12 | 12693220,08 | 45,21 | 0,000000000000000000000 |
| Error | 44917828,10 | 160 | 280736,43 | | |
| Total | 233414596,4 | 179 | | | |

n= 9

| t _(0,05;180) | 1,975 |
|-------------------------|---------|
| LSD | 467,961 |

Immersion time effect (A)

| B ₁ A ₁ _ B ₁ A ₂ | 294,38 | B ₂ A ₁ _ B ₂ A ₂ | 259,11 | B ₃ A ₁ . B ₃ A ₂ | 192,68 | B ₄ A ₁ _ B ₄ A ₂ | 532,83 | B ₅ A ₁ . B ₅ A ₂ | 8,33 |
|---------------------------------------------------------------|--------|---------------------------------------------------------------|--------|---------------------------------------------------------------|--------|---------------------------------------------------------------|---------|---------------------------------------------------------------|--------|
| B ₁ A ₁ _ B ₁ A ₃ | 723,53 | B ₂ A ₁ _ B ₂ A ₃ | 367,20 | B ₃ A ₁ _ B ₃ A ₃ | 440,43 | B ₄ A ₁ B ₄ A ₃ | 544,25 | B ₅ A ₁ _ B ₅ A ₃ | 218,00 |
| B ₁ A ₁ _ B ₁ A ₄ | 294,16 | B ₂ A ₁ B ₂ A ₄ | 87,03 | B ₃ A ₁ _ B ₃ A ₄ | 1,78 | B ₄ A ₁ B ₄ A ₄ | 711,47 | B ₅ A ₁ _ B ₅ A ₄ | 78,04 |
| B ₁ A ₂ _ B ₁ A ₃ | 429,15 | B ₂ A ₂ _ B ₂ A ₃ | 626,31 | B ₃ A ₂ _ B ₃ A ₃ | 247,75 | B ₄ A ₂ B ₄ A ₃ | 1077,07 | B ₅ A ₂ _ B ₅ A ₃ | 226,33 |
| B1A2 . B1A4 | 0,23 | B ₂ A ₂ . B ₂ A ₄ | 346,14 | B3A2 . B3A4 | 194,46 | B ₄ A ₂ . B ₄ A ₄ | 1244,30 | B ₅ A ₂ . B ₅ A ₄ | 69,72 |
| B ₁ A ₃ _ B ₁ A ₄ | 429,38 | B2A3 . B2A4 | 280,18 | B ₃ A ₃ _ B ₃ A ₄ | 442,21 | B ₄ A ₃ B ₄ A ₄ | 167,23 | B ₅ A ₃ _ B ₅ A ₄ | 296,04 |

| $B_1A_1 = B_2A_1$ | 241,07 | $B_1A_2 = B_2A_2$ | 205,80 | B ₁ A ₃ _ B ₂ A ₃ | 849,66 | B_1A_4 . B_2A_4 | 140,11 |
|-------------------|---------|---------------------------------------------------------------|---------|---------------------------------------------------------------|---------|---------------------------------------------------------------|---------|
| $B_1A_1 = B_3A_1$ | 95,01 | $B_1A_2 = B_3A_2$ | 582,07 | B ₁ A ₃ _ B ₃ A ₃ | 1258,97 | $B_1A_4 = B_3A_4$ | 387,38 |
| $B_1A_1 = B_4A_1$ | 742,31 | $B_1A_2 = B_4A_2$ | 503,87 | $B_1A_3 = B_4A_3$ | 2010,09 | $B_1A_4 = B_4A_4$ | 1747,94 |
| $B_1A_1 = B_5A_1$ | 1918,00 | B ₁ A ₂ _ B ₅ A ₂ | 1631,95 | B ₁ A ₃ _ B ₅ A ₃ | 976,47 | B ₁ A ₄ _ B ₅ A ₄ | 1701,89 |
| $B_2A_1 = B_3A_1$ | 336,08 | $B_2A_2 = B_3A_2$ | 787,87 | $B_2A_3 - B_3A_3$ | 409,31 | $B_2A_4 = B_3A_4$ | 247,27 |
| $B_2A_1 - B_4A_1$ | 983,38 | $B_2A_2 - B_4A_2$ | 709,66 | B ₂ A ₃ _ B ₄ A ₃ | 1160,42 | B_2A_4 . B_4A_4 | 1607,82 |
| $B_2A_1 = B_5A_1$ | 1676,93 | $B_2A_2 = B_5A_2$ | 1426,15 | B ₂ A ₃ _ B ₅ A ₃ | 1826,13 | $B_2A_4 = B_5A_4$ | 1842,00 |
| $B_3A_1 = B_4A_1$ | 647,30 | B ₃ A ₂ _ B ₄ A ₂ | 78,21 | B ₃ A _{3 -} B ₄ A ₃ | 751,11 | $B_3A_4 = B_4A_4$ | 1360,55 |
| $B_3A_1 = B_5A_1$ | 2013,01 | $B_3A_2 = B_5A_2$ | 2214,02 | B ₃ A ₃ _ B ₅ A ₃ | 2235,44 | B ₃ A ₄ _ B ₅ A ₄ | 2089,27 |
| $B_4A_1 = B_5A_1$ | 2660,31 | B_4A_2 _ B_5A_2 | 2135,81 | B ₄ A ₃ _ B ₅ A ₃ | 2986,56 | B ₄ A ₄ _ B ₅ A ₄ | 3449,82 |

Transverse strength σ_2

| | | | B: Concentration | | | | |
|----------------------|-------------|-----------------------------------------------------------------------------------------|-----------------------------------------------------------------------------------------|---------------------------------------------------------------------------------------------------------------------------------------|----------------------------------------------------------------------------------------|--------------------------------------------------------------------------------|----------------------|
| E mo | auius L i i | 0,25 | | | | | Total Y ₁ |
| A: Inmersion time | 1 | 21,34 21,31 17,05 14,59 16,44 15,85 16,42 15,85 | 20,48 20,98 18,09 18,50 18,70 19,68 19,35 19,52 21,11 | 18,16 21,72 22,35 24,06 23,13 20,45 21,10 18,55 22,65 | 23,87 23,46 24,42 23,59 24,20 25,08 24,14 25,06 24,23 | 6,49 6,29 6,13 5,91 5,92 6,77 5,6,32 7,06 5,79 5,97 | 796,20 |
| | 3 | 21,21 20,72 16,60 17,42 19,18 17,24 16,19 19,28 16,10 17,74 | 19,33 18,82 17,55 16,23 19,42 19,64 164,22 19,62 16,72 16,89 | 24,56 25,49 21,85 24,84 22,86 20,82 21,75 21,64 24,71 208,51 208,51 | 19,48 18,80 19,24 20,11 19,66 19,29 20,24 20,52 20,81 | 7,49 7,68 7,50 8,12 7,80 8,06 69,10 9,01 6,72 6,72 | 785,50 |
| | 12 | 13,07 13,10 12,51 13,13 12,39 12,75 133,91 16,83 20,50 19,63 | 20,13 19,34 17,44 20,61 19,12 18,92 171,22 19,08 18,09 18,49 | 21,91 20,03 23,88 24,44 27,49 20,36 23,51 23,51 23,51 | 2651 26,44 26,98 22,58 25,79 25,23 26,21 23,61 23,12 | 7,83 7,85 8,19 7,84 8,07 7,83 7,87 7,87 7,87 | 810,43 |
| | 24 | 15,31 14,66 14,65 13,34 13,58 147,42 16,10 16,91 22,33 20,54 | 14,48 16,37 15,09 17,74 15,41 17,05 17,13 17,26 16,35 | 18,34 18,34 16,78 20,56 20,40 17,55 19,66 19,05 23,65 | 23,76 22,16 25,25 24,56 24,50 24,50 24,50 22,28 18,98 20,49 | 7,04 6,54 6,41 6,62 57,00 6,62 6,72 5,20 5,36 | 733,30 |
| | Total Y. | 600,1102 | 658,7206 | 780,4863 | 830,3569 | 255,75934 | Y = 3125,43334 |

| Source | SS | df | MS | F ₀ | p-value |
|-----------|---------|-----|--------|----------------|-------------------------|
| Efecto A | 67,84 | 3 | 22,61 | 3,27 | 0,0228149855637693000 |
| Efecto B | -315,79 | 4 | -78,95 | -11,41 | #iNUM! |
| Efecto AB | 5688,23 | 12 | 474,02 | 68,53 | 0,000000000000000000000 |
| Error | 1106,66 | 160 | 6,92 | | |
| Total | 6546,9 | 179 | | | |

n= 9

| t _(0,05;180) | 1,975 |
|-------------------------|-------|
| LSD | 2,323 |

Immersion time effect (A)

| B ₁ A ₁ . B ₁ A ₂ | 1,22 | B ₂ A ₁ . B ₂ A ₂ | 1,22 | B ₃ A ₁ . B ₃ A ₂ | 1,64 | B ₄ A ₁ . B ₄ A ₂ | 3,98 | B ₅ A ₁ . B ₅ A ₂ | 1,28 |
|---------------------------------------------------------------|------|---------------------------------------------------------------|------|---------------------------------------------------------------|------|---------------------------------------------------------------|------|---------------------------------------------------------------|------|
| B ₁ A ₁ _ B ₁ A ₃ | 1,94 | B ₂ A ₁ _ B ₂ A ₃ | 0,52 | B ₃ A ₁ _ B ₃ A ₃ | 1,33 | B ₄ A ₁ _ B ₄ A ₃ | 0,84 | B ₅ A ₁ _ B ₅ A ₃ | 1,70 |
| B ₁ A ₁ B ₁ A ₄ | 0,59 | B ₂ A ₁ _ B ₂ A ₄ | 2,95 | B ₃ A ₁ B ₃ A ₄ | 1,78 | B ₄ A ₁ B ₄ A ₄ | 1,04 | B ₅ A ₁ _ B ₅ A ₄ | 0,07 |
| $B_1A_2 = B_1A_3$ | 3,16 | B ₂ A ₂ _ B ₂ A ₃ | 0,70 | B ₃ A ₂ _ B ₃ A ₃ | 0,30 | B ₄ A ₂ _ B ₄ A ₃ | 4,83 | B ₅ A ₂ _ B ₅ A ₃ | 0,42 |
| B ₁ A ₂ . B ₁ A ₄ | 1,81 | B ₂ A ₂ . B ₂ A ₄ | 1,73 | B ₃ A ₂ . B ₃ A ₄ | 3,42 | B ₄ A ₂ . B ₄ A ₄ | 2,95 | B5A2 . B5A4 | 1,21 |
| B ₁ A ₃ _ B ₁ A ₄ | 1,35 | B2A3 . B2A4 | 2,43 | B3A3 _ B3A4 | 3,11 | B ₄ A ₃ . B ₄ A ₄ | 1,88 | B5A3 _ B5A4 | 1,63 |

| $B_1A_1 = B_2A_1$ | 2,31 | $B_1A_2 = B_2A_2$ | 0,13 | B ₁ A ₃ _ B ₂ A ₃ | 3,73 | $B_1A_4 = B_2A_4$ | 0,05 |
|---------------------|-------|---------------------------------------------------------------|-------|---------------------------------------------------------------|-------|---------------------------------------------------------------|-------|
| $B_1A_1 = B_3A_1$ | 3,89 | $B_1A_2 = B_3A_2$ | 4,30 | B ₁ A ₃ _ B ₃ A ₃ | 7,16 | $B_1A_4 = B_3A_4$ | 2,69 |
| $B_1A_1 = B_4A_1$ | 6,47 | $B_1A_2 - B_4A_2$ | 1,27 | $B_1A_3 = B_4A_3$ | 9,26 | $B_1A_4 = B_4A_4$ | 6,02 |
| $B_1A_1 = B_5A_1$ | 9,70 | B ₁ A ₂ _ B ₅ A ₂ | 9,64 | B ₁ A ₃ _ B ₅ A ₃ | 6,06 | B ₁ A ₄ _ B ₅ A ₄ | 9,04 |
| $B_2A_1 - B_3A_1$ | 1,58 | $B_2A_2 - B_3A_2$ | 4,43 | B ₂ A ₃ _ B ₃ A ₃ | 3,43 | $B_2A_4 - B_3A_4$ | 2,75 |
| $B_2A_1 - B_4A_1$ | 4,16 | $B_2A_2 - B_4A_2$ | 1,40 | $B_2A_3 - B_4A_3$ | 5,53 | $B_2A_4 - B_4A_4$ | 6,08 |
| B_2A_1 _ B_5A_1 | 12,01 | B_2A_2 _ B_5A_2 | 9,51 | B ₂ A ₃ _ B ₅ A ₃ | 9,79 | B ₂ A ₄ _ B ₅ A ₄ | 8,99 |
| $B_3A_1 - B_4A_1$ | 2,59 | B ₃ A ₂ _ B ₄ A ₂ | 3,03 | B ₃ A _{3 -} B ₄ A ₃ | 2,10 | $B_3A_4 - B_4A_4$ | 3,33 |
| $B_3A_1 - B_5A_1$ | 13,58 | B ₃ A ₂ _ B ₅ A ₂ | 13,94 | B ₃ A ₃ _ B ₅ A ₃ | 13,21 | B ₃ A ₄ _ B ₅ A ₄ | 11,73 |
| $B_4A_1 = B_5A_1$ | 16,17 | $B_4A_2 - B_5A_2$ | 10,91 | B ₄ A ₃ _ B ₅ A ₃ | 15,31 | B ₄ A ₄ _ B ₅ A ₄ | 15,07 |
Transverse strain effect ϵ_2

| Strain TTT | | B: Concentration | | | | | | | |
|----------------------|----------|--------------------------------------------------------------------------------------|------------------------------------------------------------------------------|------------------------------------------------------------------------------|------------------------------------------------------------------------------|-------------------------------------------------------------------------------|----------------------|--|--|
| | | 0,25 | 0,5 | | | | Total Y ₁ | | |
| A: Inmersion time | 1 | 0,58 0,69 0,52 0,50 0,51 0,65 5,28 0,69 0,54 0,59 | 0,56 0,50 0,57 0,69 0,97 0,62 6,53 0,83 0,85 0,93 | 0,67 0,82 0,76 0,95 0,95 0,65 6,91 0,76 0,76 0,58 | 0,34 0,62 0,39 0,68 0,69 0,81 0,69 0,69 0,69 0,70 | 1,21 1,21 1,46 1,27 1,61 0,45 1,11 2,31 1,84 | 36,77 | | |
| | 3 | 0,38 0,63 0,35 0,58 0,98 0,85 5,86 0,70 0,58 0,81 | 0,98 0,81 0,89 0,67 0,89 0,80 7,44 0,88 0,73 0,78 | 0,82 0,97 0,89 0,83 0,68 0,57 6,15 0,47 0,45 0,47 | 0,67 0,61 0,78 0,57 0,70 0,70 0,57 0,57 0,49 | 1,12 1,28 1,04 1,45 1,75 1,13 12,54 0,82 1,97 1,97 | 37,67 | | |
| | 12 | 0,65 0,65 0,58 0,60 0,56 0,60 5,74 0,68 0,72 0,70 | 0,61 0,59 0,63 0,68 0,48 0,81 0,81 0,81 | 0,75 0,63 0,76 0,60 0,69 0,45 0,55 0,86 0,86 | 0,71 0,67 0,46 0,68 0,59 0,66 0,53 0,47 | 1,22 1,21 1,49 1,38 1,52 0,78 9,81 0,74 0,74 0,74 | 33,20 | | |
| | 24 | 0,47 0,50 0,52 0,48 0,68 5,38 0,53 0,53 0,53 0,84 0,77 | 0,47 0,64 0,60 0,64 0,54 5,19 0,62 0,58 0,60 0,50 | 0,51 0,51 0,42 0,51 0,48 4,44 0,45 0,48 0,47 0,60 | 0,61 0,53 0,66 0,70 5,20 0,68 0,53 0,39 0,44 | 1,43 1,40 1,71 1,58 1,28 1,28 1,73 1,18 1,87 | 33,66 | | |
| | Total Y. | 22,25284 | 25,23722 | 23,63523 | 21,88288 | 48,28409 | Y = 141,29226 | | |

| Source | | df | MS | F ₀ | p-value |
|-----------|-------|-----|------|----------------|-------------------------|
| Efecto A | 0,30 | 3 | 0,10 | 1,86 | 0,1387478409993170000 |
| Efecto B | 1,62 | 4 | 0,40 | 7,55 | 0,0000134737562869246 |
| Efecto AB | 12,16 | 12 | 1,01 | 18,94 | 0,000000000000000000000 |
| Error | 8,56 | 160 | 0,05 | | |
| Total | 22,6 | 179 | | | |

n= 9

| t _(0,05;180) | 1,975 |
|-------------------------|-------|
| LSD | 0,204 |

Immersion time effect (A)

| B ₁ A ₁ . B ₁ A ₂ | 0,06 | B ₂ A ₁ . B ₂ A ₂ | 0,09 | B ₃ A ₁ . B ₃ A ₂ | 0,08 | B ₄ A ₁ . B ₄ A ₂ | 0,01 B ₅ A ₁ B ₅ A ₂ | 0,00 |
|---------------------------------------------------------------|------|---------------------------------------------------------------|------|---------------------------------------------------------------|------|---------------------------------------------------------------|------------------------------------------------------------------|------|
| $B_1A_1 = B_1A_3$ | 0,05 | B ₂ A ₁ _ B ₂ A ₃ | 0,05 | B_3A_1 , B_3A_3 | 0,08 | B ₄ A ₁ _ B ₄ A ₃ | 0,01 B ₅ A ₁ B ₅ A ₃ | 0,27 |
| B ₁ A ₁ _ B ₁ A ₄ | 0,01 | B ₂ A ₁ _ B ₂ A ₄ | 0,13 | B ₃ A ₁ _ B ₃ A ₄ | 0,25 | B ₄ A ₁ B ₄ A ₄ | 0,04 B ₅ A ₁ B ₅ A ₄ | 0,10 |
| B ₁ A ₂ _ B ₁ A ₃ | 0,01 | B_2A_2 , B_2A_3 | 0,14 | B_3A_2 , B_3A_3 | 0,00 | B ₄ A ₂ _ B ₄ A ₃ | 0,02 B ₅ A ₂ B ₅ A ₃ | 0,27 |
| B ₁ A ₂ . B ₁ A ₄ | 0,05 | B_2A_2 . B_2A_4 | 0,23 | B_3A_2 . B_3A_4 | 0,17 | B ₄ A ₂ . B ₄ A ₄ | 0,05 B ₅ A ₂ B ₅ A ₄ | 0,09 |
| B ₁ A ₃ _ B ₁ A ₄ | 0,04 | B ₂ A ₃ _ B ₂ A ₄ | 0,09 | B ₃ A ₃ _ B ₃ A ₄ | 0,17 | B ₄ A ₃ _ B ₄ A ₄ | 0,02 B ₅ A ₃ B ₅ A ₄ | 0,36 |

Concentration effect (B)

| $B_1A_1 = B_2A_1$ | 0,13 | $B_1A_2 - B_2A_2$ | 0,16 | B ₁ A ₃ _ B ₂ A ₃ | 0,03 | B_1A_4 . B_2A_4 | 0,02 |
|-------------------|------|---------------------------------------------------------------|------|---------------------------------------------------------------|------|---------------------------------------------------------------|------|
| $B_1A_1 = B_3A_1$ | 0,16 | $B_1A_2 - B_3A_2$ | 0,03 | $B_1A_3 - B_3A_3$ | 0,04 | B_1A_4 . B_3A_4 | 0,09 |
| $B_1A_1 = B_4A_1$ | 0,03 | $B_1A_2 - B_4A_2$ | 0,02 | $B_1A_3 - B_4A_3$ | 0,03 | B_1A_4 . B_4A_4 | 0,02 |
| $B_1A_1 = B_5A_1$ | 0,72 | $B_1A_2 - B_5A_2$ | 0,67 | $B_1A_3 - B_5A_3$ | 0,41 | $B_1A_4 - B_5A_4$ | 0,81 |
| $B_2A_1 = B_3A_1$ | 0,04 | $B_2A_2 - B_3A_2$ | 0,13 | B ₂ A ₃ _ B ₃ A ₃ | 0,01 | $B_2A_4 - B_3A_4$ | 0,07 |
| $B_2A_1 = B_4A_1$ | 0,10 | $B_2A_2 - B_4A_2$ | 0,18 | $B_2A_3 - B_4A_3$ | 0,06 | $B_2A_4 - B_4A_4$ | 0,00 |
| $B_2A_1 - B_5A_1$ | 0,60 | $B_2A_2 - B_5A_2$ | 0,51 | B ₂ A ₃ _ B ₅ A ₃ | 0,37 | B_2A_4 . B_5A_4 | 0,83 |
| $B_3A_1 - B_4A_1$ | 0,13 | $B_3A_2 - B_4A_2$ | 0,05 | B ₃ A _{3 -} B ₄ A ₃ | 0,07 | $B_3A_4 - B_4A_4$ | 0,08 |
| $B_3A_1 - B_5A_1$ | 0,56 | $B_3A_2 - B_5A_2$ | 0,64 | B ₃ A ₃ _ B ₅ A ₃ | 0,37 | B ₃ A ₄ . B ₅ A ₄ | 0,90 |
| $B_4A_1 = B_5A_1$ | 0,69 | B ₄ A ₂ _ B ₅ A ₂ | 0,69 | B ₄ A ₃ _ B ₅ A ₃ | 0,44 | B ₄ A ₄ _ B ₅ A ₄ | 0,82 |