

Research Article

Numerical Simulation and Response of Stiffened Plates Subjected to Noncontact Underwater Explosion

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A numerical simulation has been carried out to examine the response of steel plates with different arrangement of stiffeners and subjected to noncontact underwater explosion (UNDEX) with different shock loads. Numerical analysis of the underwater explosion phenomena is implemented in the nonlinear finite element code ABAQUS/Explicit. The aim of this work is to enhance the dynamic response to resist UNDEX. Special emphasis is focused on the evolution of mid-point displacements. Further investigations have been performed to study the effects of including material damping and the rate-dependant material properties at different shock loads. The results indicate that stiffeners configurations and shock loads affect greatly the overall performance of steel plates and sensitive to the materials data. Also, the numerical results can be used to obtain design guidelines of floating structures to enhance resistance of underwater shock damage, since explosive tests are costly and dangerous.

1. Introduction

The dynamic responses of floating and submerged structures subjected to noncontact underwater explosion have received attention as a concerned topic for a national defense since the 1950s. When a naval ship is attacked by an underwater explosion (UNDEX), the ship can be severely damaged by shock waves and gas bubble pulse. Predicting the shock response of ships to noncontact UNDEX from underwater weapon is of great importance for the warhead design of underwater weapon.

The design and analysis of structure subjected to UNDEX require a detailed understanding of explosion phenomena and the dynamic response of various structural elements [1]. UNDEX structural damage depends on standoff distance and explosion weight. The most important way to reduce the damages due to UNDEX is to provide sufficient standoff distance between the structure and explosion source in order to decrease the effect of the UNDEX wave so that the structure is not highly damaged. To accomplish these objectives, it is

necessary to do various scenarios to evaluate the behavior of the ship structure to blast loading. These scenarios should include studying such aspects (explosive magnitude, distance from source of explosion, structure scantling, complex fluid-structure interaction phenomena, structure geometry, etc.) [2].

Rajendran [3] carried a numerical simulation of underwater explosion experiments on plane plates using ANSYS/LS-DYNA for elastic and inelastic response of circular and rectangular plates. Gupta et al. [4] predicted the failure modes of stiffened and unstiffened square plates subjected to underwater explosion using an elastoplastic model with isotropic hardening, strain rate effects, and fracture criterion. Qiankun and Gangyi [5] predict the shock response of a ship section to noncontact UNDEX using ABAQUS. The results showed that the size of fluid mesh and the fluid thickness from the wetted surface to the outer boundary of fluid is of great importance for improving numerical accuracy. Rajendran and Narasimhan [6] studied the damage prediction of clamped circular plates subjected

to contact underwater explosion. Jacinto et al. [7] studied a linear dynamic analysis of the plate models under explosive loading using ABAQUS.

Zhang et al. [8] developed a procedure which employs the finite element method coupled with the DAA method to study the transient dynamic response of a surface ship model subjected to an underwater explosion bubble. And they found that the ship model's vertical response subjected to an underwater explosion bubble is mainly the global response, which is composed of two parts: rigid-body motion and elastic deformation.

When a charge of high explosive is detonated below the surface of water, the sudden release of energy from underwater explosions generates a shockwave propagate in a spherical pattern at very high speed. The explosive also forms highly compressed gas bubble in the surrounding water, which are very long compared with the times for shock passage. Of the total energy released from a 1500-lb TNT underwater explosion approximately 53% goes into the shockwave and 47% goes into bubble pulsation. Most cases demonstrate that the damage done to marine structures, such as the surface of ships and submarines, occurs early and is due to the primary shockwaves. This investigation only considers the effects of the shockwaves as demonstrated in [9].

Zhang et al. [10] calculated the dynamic bending moment of bubble acting on hulls based on the underwater explosion bubble theory. And he found that underwater explosion bubble will cause general longitudinal bending of ship and lead to hogging and sagging damage of ship.

Forghani et al. [11] studied the modeling of damage development in blast loaded composite laminates to blast loads. It was shown that the tie-break interface option in LS-DYNA can be used successfully in simulating cohesive cracks. Chirica and Boazu [2] studied the response of ship hull laminated plates to blast loads using finite-element computer code COSMOS/M. Various scenarios to evaluate the behavior of the ship structure laminated plate to blast loading are studied like explosive magnitude, distance from source of explosion, and plate thickness.

Jan [12] used finite element software ABAQUS to simulate and analyze the transient dynamic response of a midjet submersible vehicle (MSV) pressure hull that experiences loading by an acoustic pressure shock wave resulting from an underwater explosion (UNDEX). The results showed that, the local cavitations formed on the fluid-structure interface, after the shockwave hit the midjet submersible vehicle. The aim of this work is to enhance the dynamic response to resist UNDEX and obtain the optimal configuration for stiffened plates to resist underwater shock loading

2. The UNDEX Loading

Noncontact underwater explosion is the major source of threat to ships and submarines. The responses and damages of submerged structures are divided into two categories: near-field explosion and far-field explosion depending on the distance between the explosive charge and the target (standoff distance) [12].

Figure 1 shows the different events occurring during the UNDEX event in a pressure against time history plot [13]. The loading mechanisms due to underwater explosion include incident shock wave, free surface reflection wave, bottom reflection wave, gas bubble oscillation, bubble-pulse loading, and bulk and hull cavitations [14]. The reflection of the shock wave from the bottom of the ocean is a compression wave that adds additional load to the structure. The reflection of the shock wave from free ocean surface causes a reduction in the pressure produced by the shock wave [11, 15]. The incident wave is the shock wave produced by the UNDEX charge. The scattered wave is the acoustic field generated by the interaction of the incident wave and the submerged structure. The initial shock wave is modeled as a spherical incident shock wave applied as a transient load active on both the acoustic and structural meshes at their common surfaces (the wetted interface). The distribution of this shock wave onto the plate is obtained using the incident pressure wave equations as demonstrated in [16]. The pressure history $P(t)$ of the shockwave at a fixed location starts with an instantaneous pressure peak, P_{\max} , followed by a decline which is usually approximated by an exponential function. The empirically determined equation of the pressure profile [9, 12, 17] has the following form:

$$P(t) = P_{\max} e^{-(t-t_1)/\theta} \text{ (MPa)} \quad t \geq t_1, \quad (1)$$

where t is the time elapsed after the detonation of charge in (msec); t_1 is the arrival time of shock wave to the target after the detonation of charge in (msec); P_{\max} is the peak pressure magnitude (MPa) at the shock wave front, and θ is the time decay constant which describes the exponential decay in (msec) and is equal to the time taken by the peak pressure to fall to $1/e$ times its initial value. The peak pressure and decay constant depend upon the size of the explosion and the standoff distance from this charge at which pressure is measured. The peak pressure, P_{\max} , and decay constant, θ , can be expressed as follows [18]:

$$P_{\max} = K_1 \left(\frac{W^{1/3}}{R} \right)^{A_1} \text{ (MPa)} \quad (2)$$

$$\theta = K_2 W^{1/3} \left(\frac{W^{1/3}}{R} \right)^{A_2} \text{ (millisecond (msec))}.$$

In case of TNT charge, the maximum bubble radii of explosive gas, R_{\max} , and first pulsation periods, T , are expressed as [19]:

$$R_{\max} = 3.38 \left(\frac{W}{(D + 10.3)} \right)^{1/3} \text{ (m)} \quad (3)$$

$$T = 2.1 \frac{W^{1/3}}{(D + 10.3)^{5/6}} \text{ (sec)},$$

where K_1 , K_2 and A_1 , A_2 are constants that depend on explosive charge type when different explosives are used. These input constants are as stated in Table 1 [20]. W is the weight of the explosive charge in (Kg); D is a charge

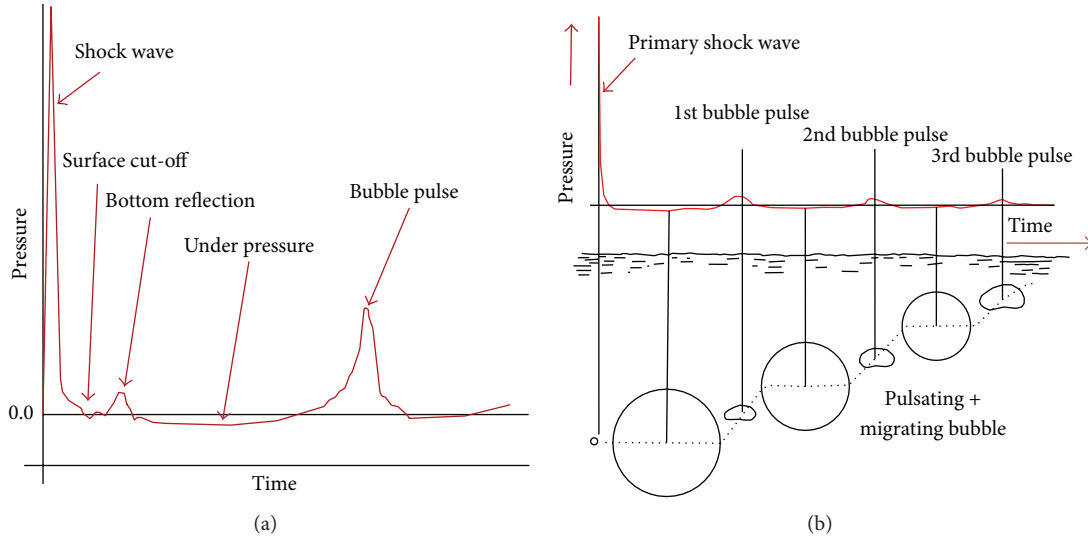


FIGURE 1: Phenomenon of the UNDEX: shock wave and high pressure bubble [13].

TABLE 1: Shock wave parameters for various explosive charges.

Constants	Explosive material type			
	HBX-1	TNT	PETN	Nuclear
K1	53.51	52.12	56.21	1.06E4
A1	1.144	1.180	1.194	1.13
K2	0.092	0.0895	0.086	3.627
A2	-0.247	-0.185	-0.257	-0.22

depth in (m), and R is the distance between the explosive charge and target in (m). Also Cole [9] provided further information regarding the systematic presentation of physical effects associated with underwater explosions.

When pressure from an underwater explosion impinges upon a flexible surface, such as the hull of surface ship, the reflected pressure on the fluid-structure interaction surface can be reasonably and accurately predicted using Taylor's plate theory [12, 15]. For an air backed plate of mass per unit area (m) subjected to an incident plane shockwave, $P_i(t)$, a reflection wave of pressure, $P_r(t)$, leaves the plate which is moving at velocity $v_p(t)$. Using Newton's second law of motion,

$$m \cdot \frac{dv_p}{dt} = P_i + P_r. \quad (4)$$

The fluid particle velocities behind the incident and reflected shockwave are $v_i(t)$ and $v_r(t)$, respectively; thus, the velocity of the plate becomes

$$v_p(t) = v_i(t) - v_r(t). \quad (5)$$

The incidence and reflective shockwave pressures are $P_i = \rho_f c v_i$ and $P_r(t) = \rho_f c v_r$, respectively, where ρ_f is fluid density and c is the acoustic velocity. By substituting the pressure into (4) and solving with (1), $P_r(t)$ becomes

$$P_r(t) = P_i(t) - \rho_f c v_p = P_{\max} e^{-((t-t_1)/\theta)} - \rho_f c v_p. \quad (6)$$

Equation (4) can then be rewritten as

$$m \cdot \frac{dv_p}{dt} + \rho_f c v_p = 2P_{\max} e^{-((t-t_1)/\theta)}. \quad (7)$$

Differentiating (7) yields the following expression for plate velocity, where $\psi = \rho_f c \theta / m$ and $t > 0$. The total pressure on the plate is

$$P_t(t) = 2P_i(t) - \rho_f c v_p = \frac{2P_{\max}}{1-\psi} \left[e^{-((t-t_1)/\theta)} - \psi e^{-\psi(t-t_1)/\theta} \right]. \quad (8)$$

A shock factor (SF), which is proportional to the energy density of the shockwave arriving at a structure, due to various combinations of charge weight and standoff distance is derived by

$$SF = \frac{0.45\sqrt{w}}{R} \left(\frac{Kg^{1/2}}{m} \right). \quad (9)$$

3. Models Description

Six different steel plates are considered using three-dimensional parts with an extruded shell base feature available in ABAQUS/Explicit version 6.10. All the plates are 20 mm in thickness and $3000 \times 3000 \text{ mm}^2$ with rectangular stiffeners 30 mm in thickness and 100 mm in height. Stiffeners are created by adding an extruded shell feature which implies perfect connection without any additional constraint. Care is taken not to overlap the material of the stiffener by offsetting its reference surface from mid-surface, thus avoiding the possibility of additional stiffness at the junction in real fabrication. Figure 2 shows the different stiffener configurations used in the numerical study.

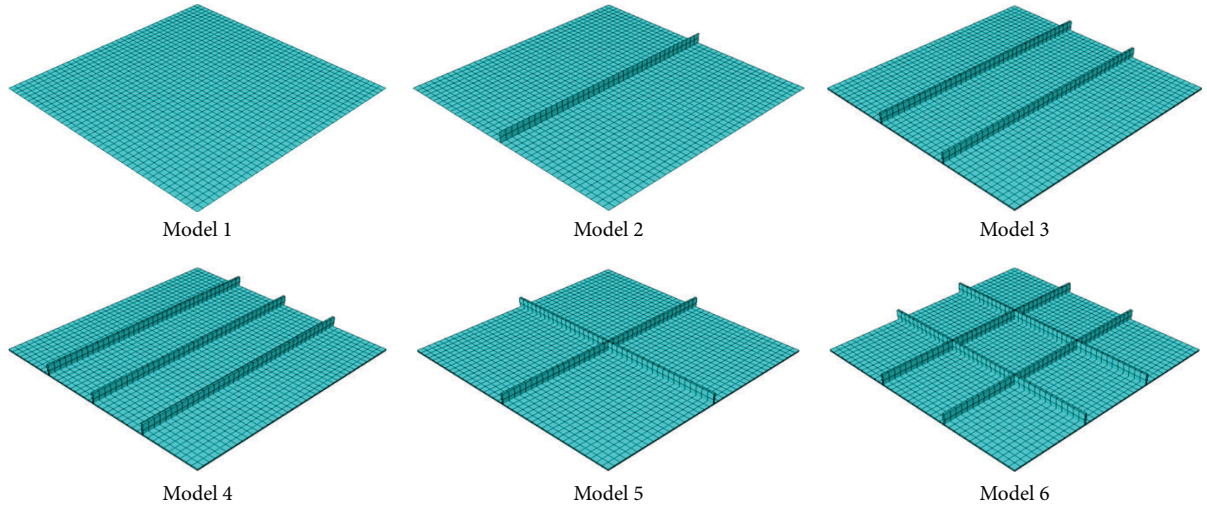


FIGURE 2: Geometrical configurations of stiffened plates.

4. Numerical Model of the Plates

The nonlinear finite element program ABAQUS/Explicit is used to undertake a three-dimensional (3D) analysis of the problem.

4.1. Models Geometry. ABAQUS/Explicit offers an element library for a wide range of geometric models. In the present study, the fourth node shell element (S4R-6 DOF at each node) with reduced integration and hourglass control was used to model the geometry of the plates and stiffeners. Six different models consisting of grids of shell elements of size 0.075 m were used as shown in the Figure 2. The fluid region of the model is represented by an assemblage of 4-node acoustic tetrahedral elements (AC3D4). In explicit dynamic analysis, the shortest wavelength can be estimated using the highest frequency in the UNDEX load or prescribed boundary conditions [21]. Since the acoustic wave length decreases with increasing frequency, there exists a highest frequency for a given acoustic domain mesh. Based on inequality:

$$f_{\max} < \frac{c}{n_{\min} L_{\max}}, \quad (10)$$

where f_{\max} is the upper frequency range of the shockwave; L_{\max} represent the maximum internodal interval of an element in a mesh; n_{\min} is the number of internodal intervals per acoustic wavelength ($n_{\min} \geq 10$ is recommended); $C = \sqrt{K_f/\rho_f}$ is the acoustic speed of the fluid; k_f and ρ_f are the bulk modulus of the acoustic medium and its density.

Artificial bulk viscosity is activated to properly represent propagation of the induced compressive stress wave by employing quadratic and linear functions of volumetric strain rates with default values of 1.2 and 0.06, respectively [21]. The outer boundary of the external fluid is represented by half cylindrical surface as shown in Figure 3.

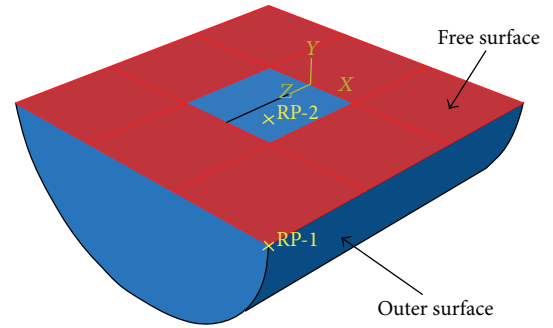


FIGURE 3: The outer and free surface used in fluid model.

TABLE 2: Plastic material properties for steel [27].

True stress (Pa)	True plastic strain
300×106	0.000
350×106	0.025
375×106	0.100
394×106	0.200
400×106	0.350

4.2. Boundary Conditions and Fluid-Structure Coupling. The panel on the ship's frame is typically stiffened by beams or stringers; thus, the panel can be divided into many small panels. The restraining moment of the borders of these panels is the torsion rigidity of a girder of stringer. During analysis, fully clamped boundary conditions are imposed on the four sides of the panels.

The boundaries of the fluid may cause shockwave refraction or reflection, resulting in its superposition or cancellation by the incident wave [21, 22]. To prevent this phenomenon, the boundary condition of the fluid element is set as a nonreflective boundary during analysis except

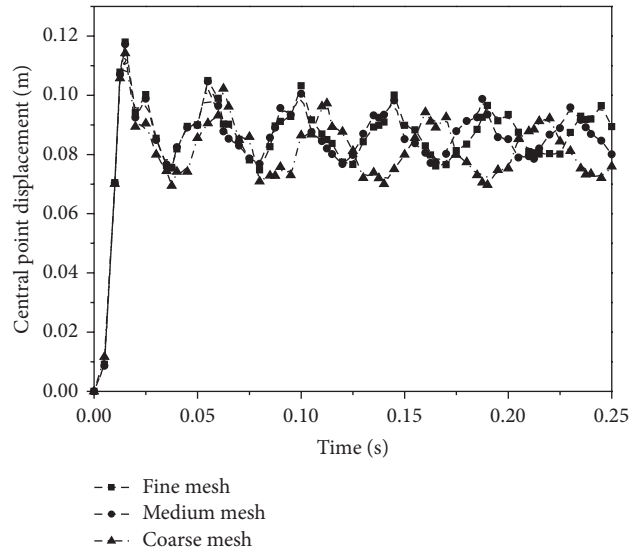


FIGURE 4: Displacement distribution at the plate central point.

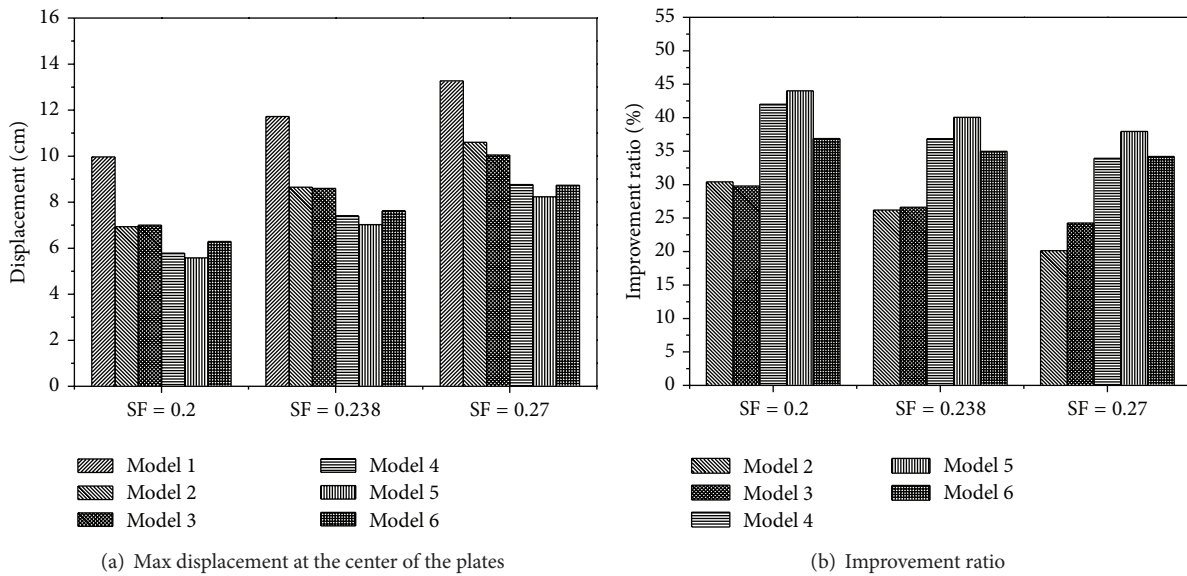


FIGURE 5: Max displacement and improvement ratio for all models.

for the free surface where zero pressure boundary condition was applied to it as shown in Figure 3. The acoustic-structural interaction between the wet surfaces of the plat and the acoustic interaction surfaces (the wetted interface) was implemented by use of a surface-based “tie” constraint. The location of the charge and the stand-off point defined as reference points, prior to the interaction, and the effect of the UNDEX event are transferred to the structure by means of the incident wave loading feature [16].

4.3. Material Properties. The stiffened panel is made of mild steel. The numerical model uses the constitutive law for

elastic/plastic materials to model the stiffened panel. Isotropic hardening rules are adopted in the hardening model. The parameters of steel used in the numerical model are as follows: Poisson ratio is equal to 0.3 and mass density is equal to 7800.0 kg/m³. The initial yield stress is 300 MPa and the yield stress increases to 400 MPa at a plastic strain of 35%. Table 2 shows the plastic material properties for steel used in this study.

When the material sustains momentary dynamic loading the effect of the strain rate causes the material’s dynamic strength to exceed the strength during a static experiment. Thus, the effect of strain rate must be considered during the

analysis to match actual situations. As recommended by Jones [23], this study adopts the Cowper-Symonds strain rate mode as follows:

$$\sigma_{dy} = \sigma_y \left[1 + \left| \frac{\dot{\epsilon}}{D} \right|^{1/q} \right], \quad (11)$$

where σ_{dy} is the material's dynamic yielding stress; σ_y is the material's static yielding stress; $\dot{\epsilon}$ is strain rate, and D and q are material parameters whose values are normally $D = 40 \text{ s}^{-1}$ and $q = 5$ accepted values for steel, respectively.

The fluid region of the model is represented by the acoustic fluid domain. Its properties are the bulk modulus and density. In this numerical investigation, commonly accepted values for the sea water were stated in [16]. The bulk modulus is 2140.4 MPa and the density of the sea water is 1000 kg/m^3 .

4.4. Convergence Study. To verify the accuracy of analytical findings a convergence study has been carried out to choose the optimum mesh for the model. The finite element mesh contains one variable (n) (number of divisions along the plate side) that affects the number of elements in the model. Varying the number (n) affects the accuracy of the results. The elements throughout the convergence study have aspect ratio 1:1.

The steel plate of model (1) has been used to perform the convergence study on it. This plate is a square steel plate with fixed sides of side length 3000 mm and thickness 20 mm. Three different models consisting of grids of shell elements of size 0.0375 m, 0.075 m, and 0.15 m representing fine, medium, and coarse meshes, respectively, were used to verify the accuracy of the finite element models of the plates. The plate is loaded due to the UNDEX with charge mass 7 kg TNT at standoff distance of 5 m from the central point of the plate. For each mesh considered, the maximum displacement in the middle is compared.

The results of the convergence study show that the maximum displacement for the model consisting of grids of shell elements size of 0.15 m is 11.42 cm. For the model consisting of grids of shell elements size of 0.075 m, the maximum displacement is 11.72 cm. It can be noted that the difference is 2.56%. Also, the maximum displacement for model consisting of grids of shell elements size of 0.0375 m is 11.8 cm and the difference is only 0.67%. Consequently, the mesh with elements size of 0.075 m is chosen to perform the whole analysis since all displacement are less than 1% different from the results for the case of elements size 0.0375 m and give minimum possible element size and analysis time. Figure 4 shows the displacement distribution at the central point of the plate of the three different models with time.

5. Results and Discussions

This study is based on the plate's performance improvement ratio I_R , which can be expressed as the ratio between the performances of unstiffened to stiffened plate for a specified case. All models are subjected to UNDEX with different amounts of explosives (5, 7, and 9 kg) at standoff distance that is equal to ($R = 5 \text{ m}$) representing SF (0.2, 0.238, and 0.27).

The parameters that play an effective influence on improving the plate's performance include the stiffeners configurations, damping effect, strain rate sensitivity and the structural integrity.

5.1. Effect of Stiffeners Configurations. The submerged or floating structure within the vicinity of an UNDEX will be subjected to loading from both the shock wave and the bubble pulse pressure waves and its performance according to the strength of these waves and the resilience of the structure.

Figure 5(a) shows the maximum central-point displacement under SF of (0.2, 0.238, and 0.27). Generally, it can be noticed that the maximum central-point displacement for the models increase with the increase of SF for each model. This is axiomatic because of increasing the explosive charge weight with fixed standoff distance.

As shown in Figure 5(a), at shock factor equal to 0.2, the maximum displacement for model 1 is 99.69 mm. The introduction of one simple stiffener results in decreasing the peak central point displacement to 69.37 as in model 2, while for models 3, 4, 5, and 6 the maximum central-point displacement is 70 mm, 57.82 mm, 55.8 mm, and 62.93 mm, respectively. Similar trend of results is observed for all other models at SF equal to 0.238. The maximum displacement for all models is greater than the maximum displacement at shock factor equal to 0.2. The maximum displacement for model 1 is 11.72 mm while for models 2, 3, 4, 5, and 6 the maximum central-point displacement is 8.65 mm, 8.6 mm, 7.4 mm, 7.02 mm, and 7.62 mm, respectively.

At SF equal to 0.27, the maximum displacement for model 1 is 13.27 mm while for models 2, 3, 4, 5, and 6 the maximum central-point displacement is 10.6 mm, 10.05 mm, 8.77 mm, 8.24 mm, and 8.73 mm, respectively. From the above result, it can be found that at low SF the maximum central-point displacement of model 2 is better than model 3 and, by increasing the SF until 0.27, the maximum central-point displacement of model 3 becomes better than model 2. Also, the minimum displacement occurs at model 5. Therefore, the configurations of stiffeners have an important influence on the response of the stiffened plates and leads to better response of the plates to different shock waves; this result is matching to the results reported in [24–26].

Figure 5(b) shows the improvement ratio (I_R) for all models. As shock factor (SF) increases the I_R decreases, at SF equal to 0.2, I_R for model 2 is 30.41%, while for models 3, 4, 5, and 6 the I_R is 29.78%, 42%, 44.03%, and 36.87%, respectively.

The best I_R is for model 5 and the lowest I_R is for model 3. By increasing the SF to 0.238, it can be found that the best I_R is 40.07% for model 5 and the lowest I_R is 26.19% for model 2. When the SF increases to 0.27, I_R is 20.12%, 24.27%, 33.93%, 37.94%, and 34.21% for models 2, 3, 4, 5, and 6, respectively. This indicates that the stiffened steel plate with one cross stiffener configuration (model 5) exhibits the highest response reduction. Also, the lowest I_R is 20.12% for model 2.

For better understanding and evaluating the performance of the plates to the shock loading, the displacement of the center point of all plate's models is monitored with time under

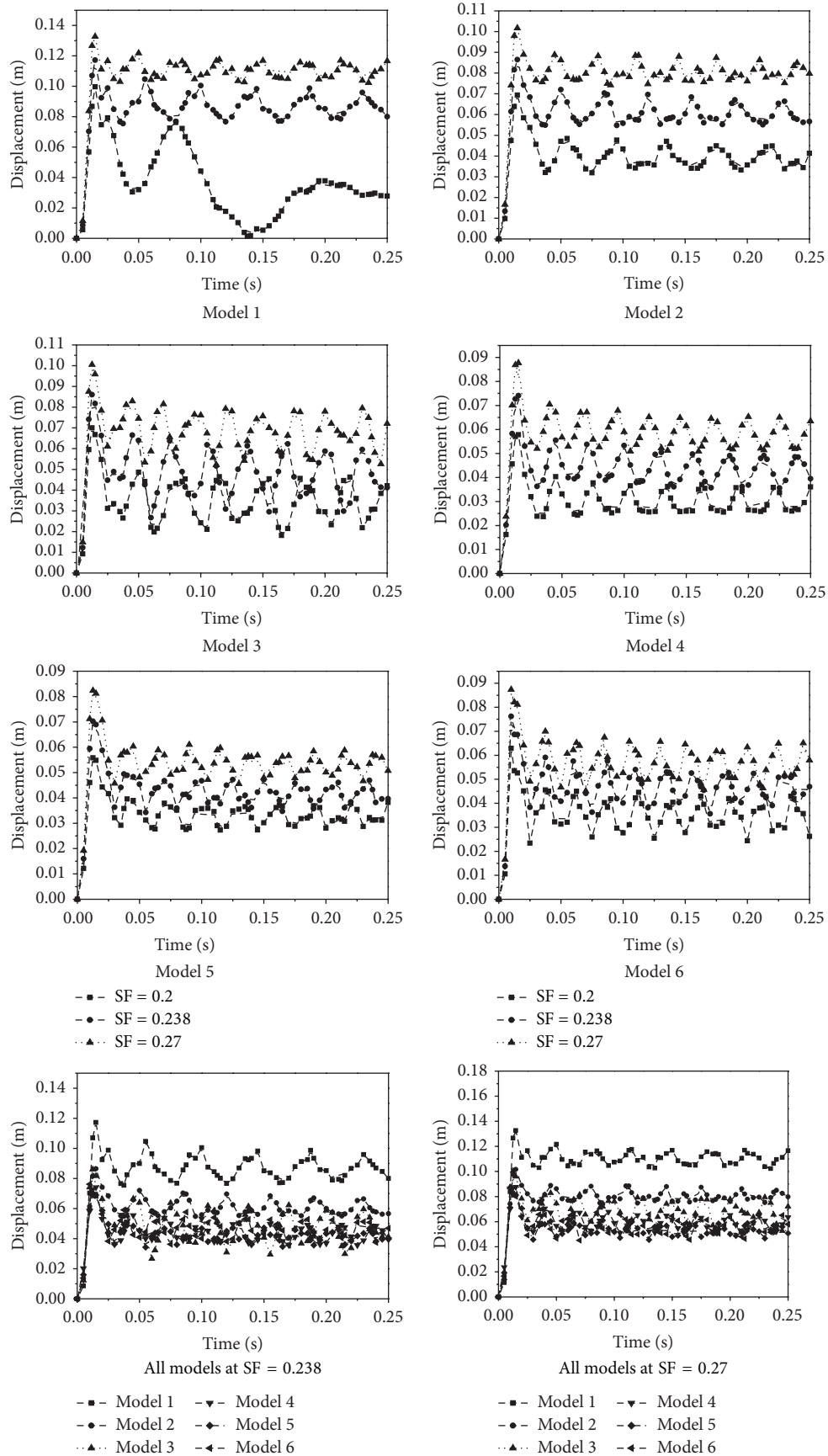
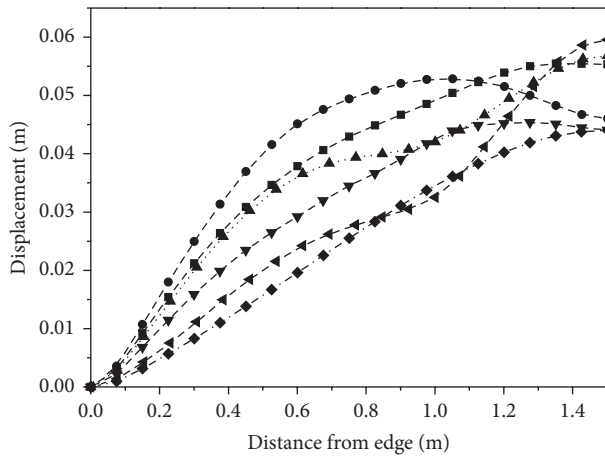
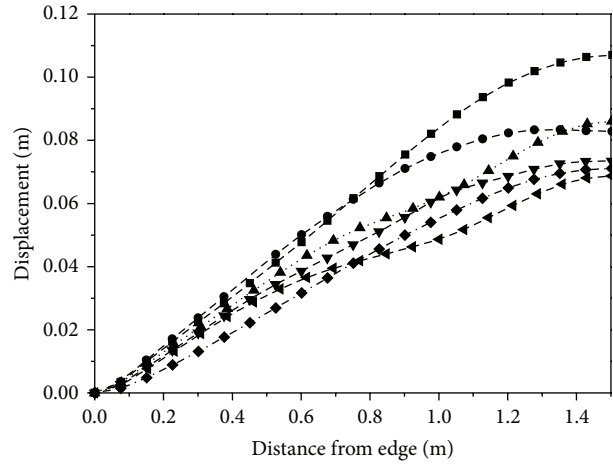


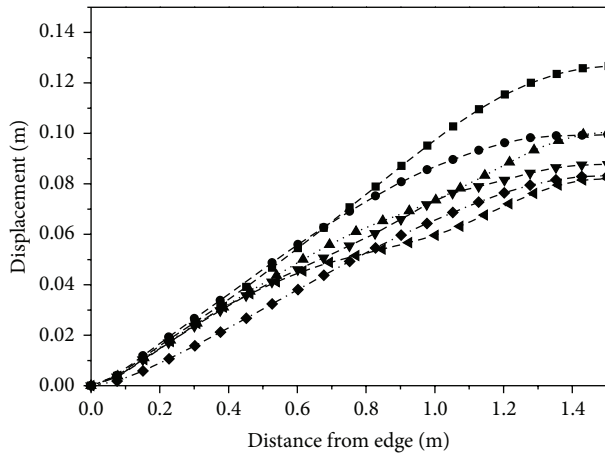
FIGURE 6: Displacement-time history of the center point of all models at different SFs.



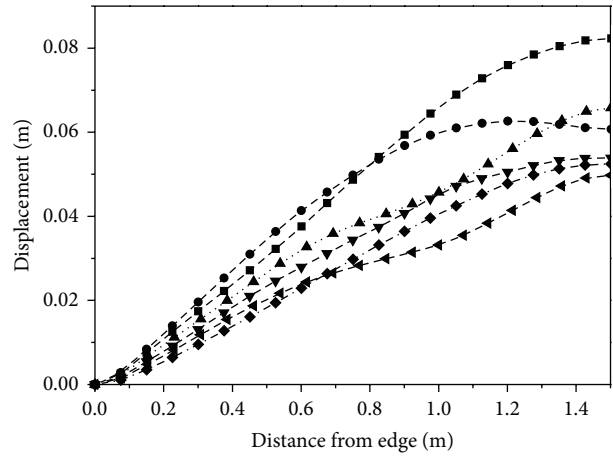
(a) All models at SF = 0.2 (12.5 msec)



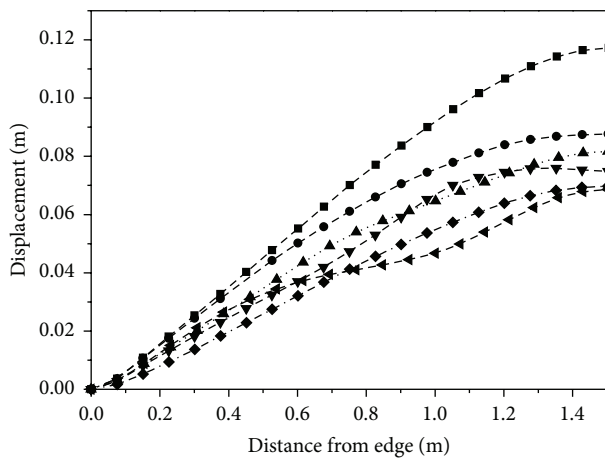
(b) All models at SF = 0.237 (12.5 msec)



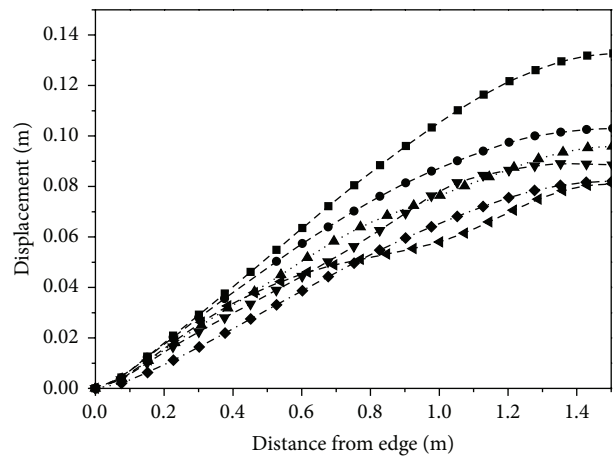
(c) All models at SF = 0.27 (12.5 msec)



(d) All models at SF = 0.2 (15 msec)



(e) All models at SF = 0.238 (15 msec)



(f) All models at SF = 0.27 (15 msec)



FIGURE 7: The transverse center-line deformation.

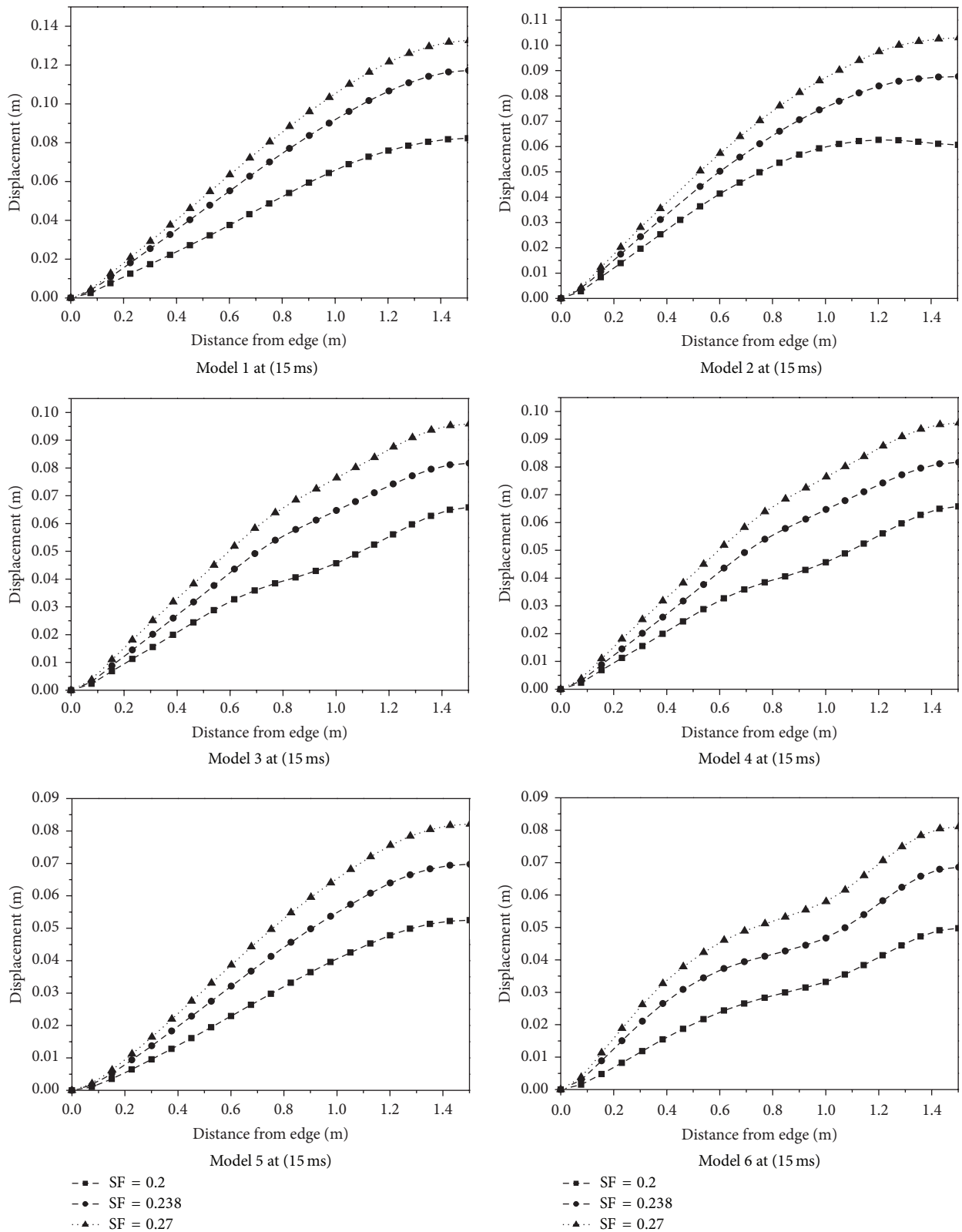


FIGURE 8: The transverse center-line deformation for all models at different SFs.

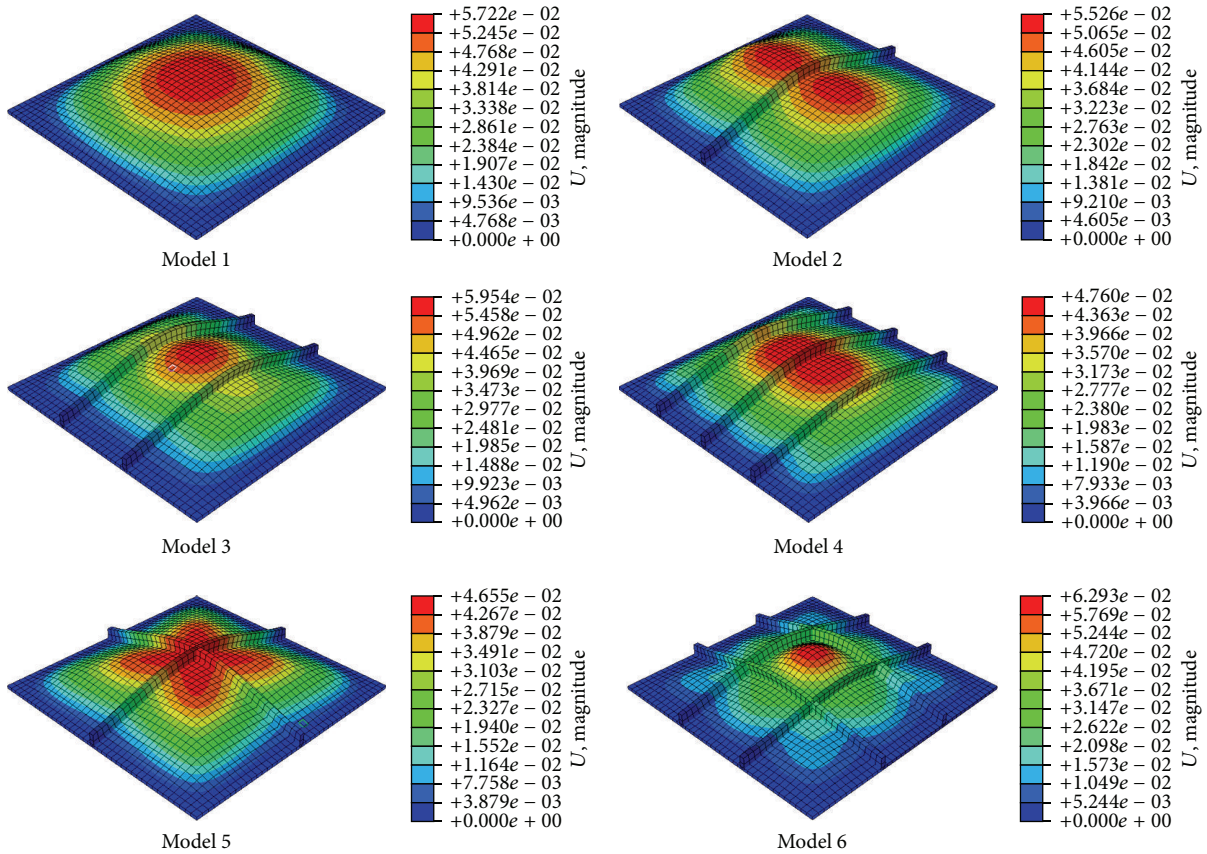


FIGURE 9: Displacement distribution for all models at 1E-2 sec and SF equal 0.2.

the effect of different shock loads as shown in Figure 6. It can be further noted that the time it takes for each model to reach maximum displacement at the same position is roughly the same.

Also, the transverse center-line deformation of all models is recorded as shown in Figure 7. It can be noted that the maximum deformation occurs at the center of the plates and reduced gradually towards the fixed boundaries. From Figure 7(a), model 2 has the maximum deformation until the distances are equal to 1.1m, after that it decreases towards the center of the plates. Model 6 has the minimum displacement from distance equaling 0.9 m to 1.1m after that the displacement increases rapidly towards the center of the plate. Also, model 5 has the minimum displacement.

Figures 7(b), 7(c), 7(d), and 7(f) show the transverse center-line deformation. Model 1 has the maximum deformation and this deformation increases linearly as the distance from the edge of the plate increases and reaches the maximum value at the center of the plate. Also model 5 has the best performance at all SFs. Figure 8 shows the transverse center-line deformation for all models at different SFs. The displacement distribution contours with time are presented in Figure 9.

5.2. Strain Rate Sensitivity. In order to investigate the strain rate sensitivity on the plate’s performance under the effect of different shock loading, the analysis has been first carried out ignoring this effect, after that the strain-rate effect is included by adjusting the material dynamic yield stress at each Gauss point according to the Cowper-Symonds strain rate mode as mentioned in (11). In this investigation the material parameters defined as $D = 40 \text{ s}^{-1}$ and $q = 5$ being the commonly accepted values for steel have been used as recommended by Jones [23].

Figure 10 shows the displacement history of the central node for both with and without rate dependant for all models at different SFs. It shows the effect of inclusion of the strain rate on the performance of the different plates. The material with rate dependant analysis leads to decreasing the maximum displacement, which assures that the results are very sensitive to the material data. Figure 11 shows the maximum displacement values of the central node for both with/without rate dependant analysis and the improving ratio I_R at SF equal to 0.2, 0.238, and 0.27, respectively.

It can be concluded that, for model 1, the maximum displacement decrease from 99.69 mm, 117.2 mm, and 132.7 mm to 84.88 mm, 101 mm, and 114.7 mm at SF equal to 0.2, 0.238, and 0.27, respectively. For model 2, the maximum mid-point

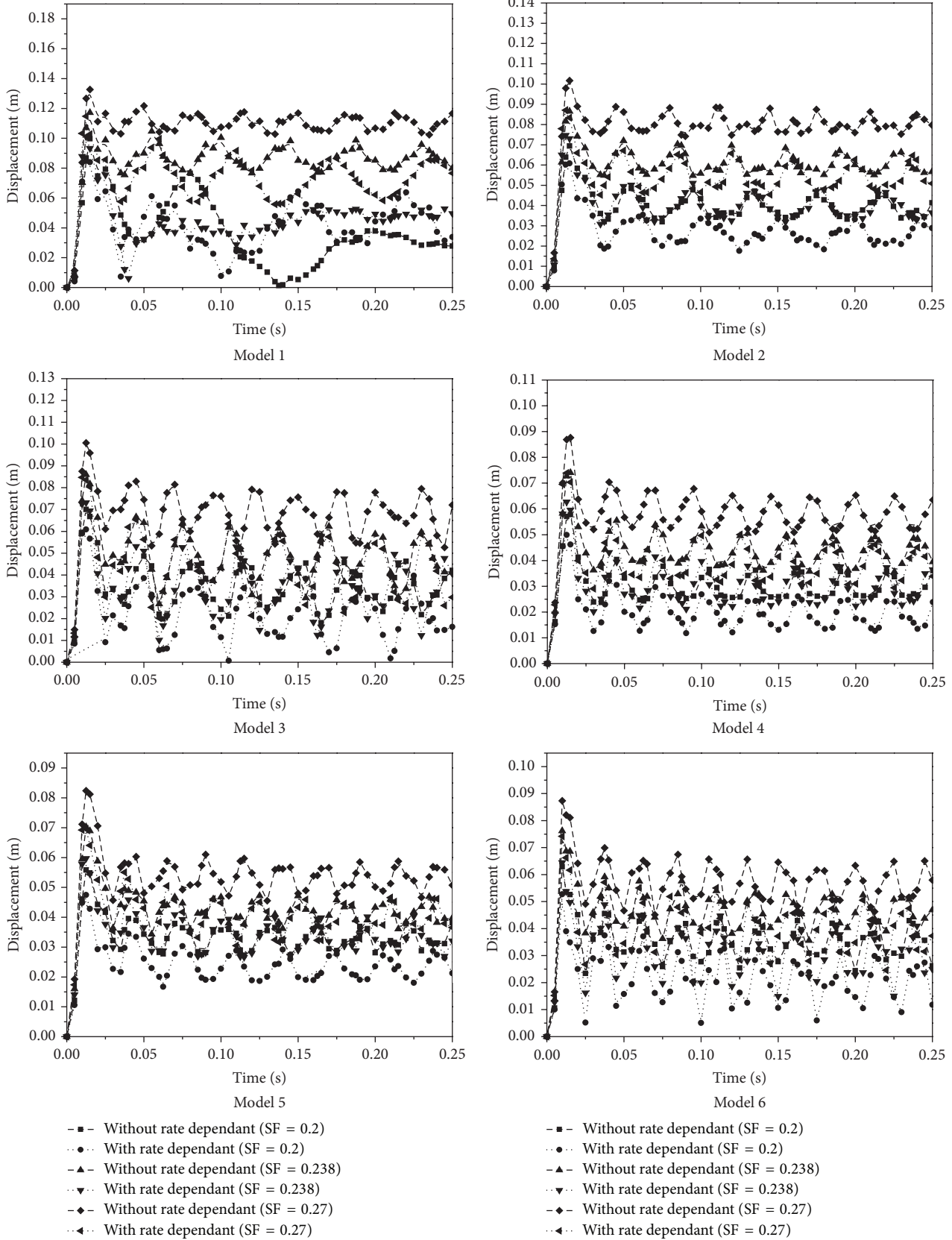


FIGURE 10: Rate dependant effect on the displacement in the central node.

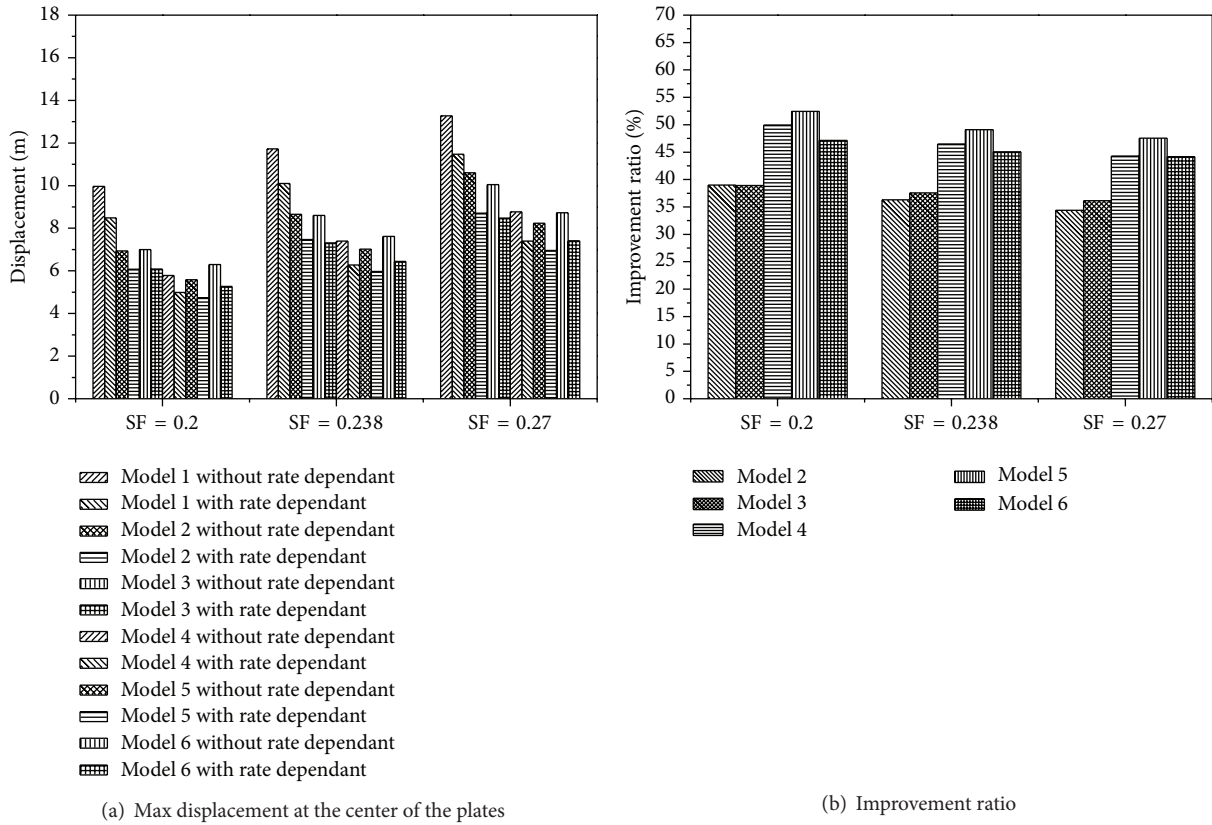


FIGURE 11: Max displacement at the central node and the improvement ratio for plats with/without rate dependant.

displacement without strain rate is 69.37 mm, 86.5 mm, and 106 mm and with strain rate are 60.8 mm, 74.67 mm, and 87.06 mm with I_R equal to 39%, 36.28%, and 34.39% at SF equal to 0.2, 0.238, and 0.27, respectively.

For model 3, the maximum mid-point displacement with strain rate is 60.88 mm, 73.16 mm, and 84.73 mm with I_R equal to 38.9%, 37.57%, and 36.15% at SF equal to 0.2, 0.238, and 0.27, respectively.

For model 4, the maximum mid-point displacement with strain rate is 49.89 mm, 62.71 mm, and 73.93 mm with I_R equal to 49.95%, 46.49%, and 44.28% at SF equal to 0.2, 0.238, and 0.27, respectively.

Model 5 has the least deformation and the material with rate dependant analysis leads to decreasing the maximum displacement to 47.4 mm, 59.67 mm, and 69.61 mm with I_R equal to 52.45%, 49.08%, and 47.54% at SF equal to 0.2, 0.238, and 0.27, respectively.

For model 6, the material with rate dependant analysis leads to decreasing the maximum displacement to 52.7 mm, 64.4 mm, and 74.1 mm with I_R equal to 47.13%, 45.05%, and 44.16% at SF equal to 0.2, 0.238, and 0.27, respectively.

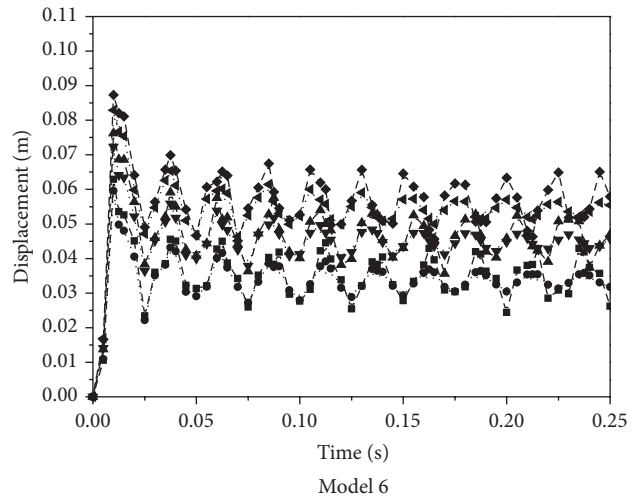
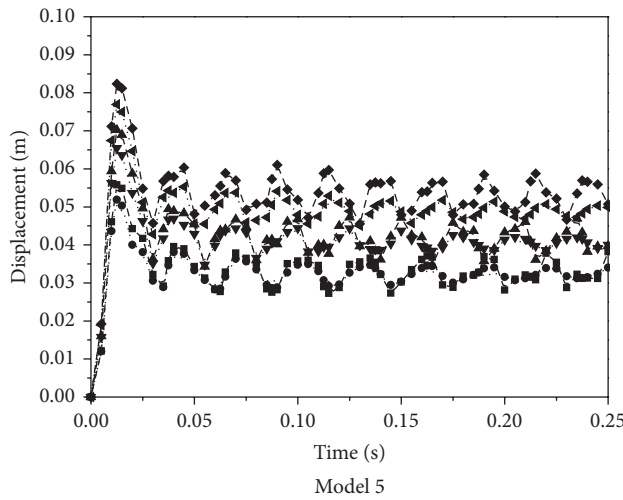
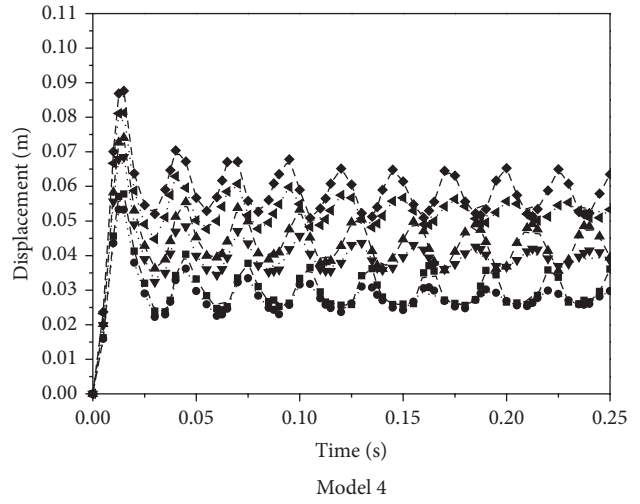
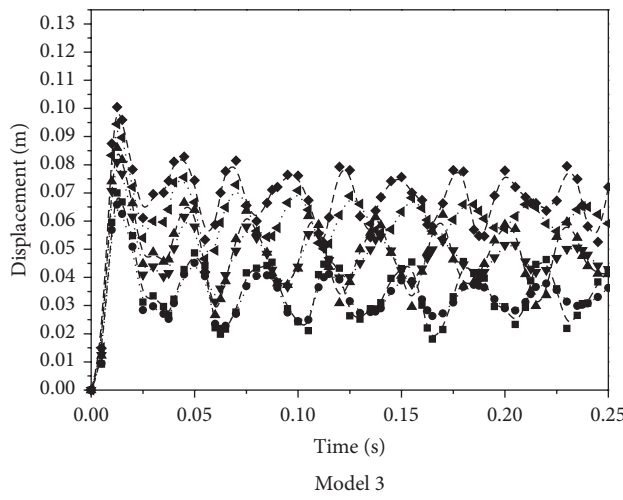
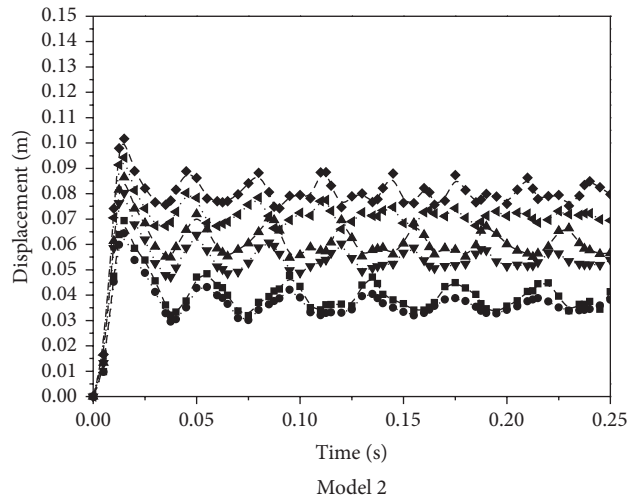
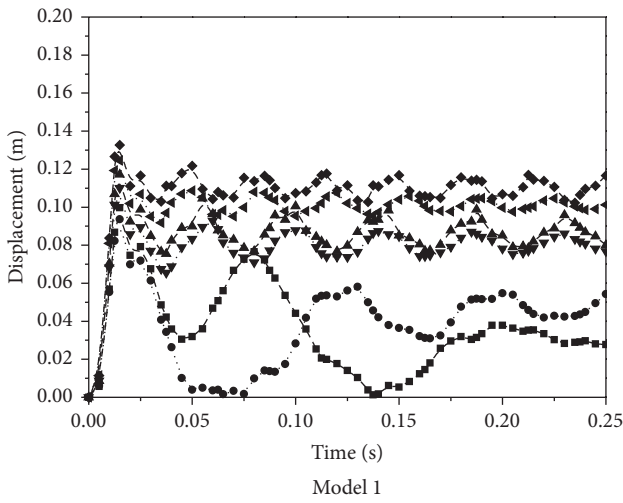
Finally it can be concluded that the best I_R is 52.45% for model 5 and the lowest I_R is 38.93% for model 3 at SF equal to 0.2. By increasing the SF the I_R decreases, the best I_R is 49.08% also for model 5 and the lowest I_R is 36.28% for model 2 at SF equal to 0.238. At SF equal to 0.27 the best I_R is 47.54% for model 5 and the lowest I_R is 34.39% for model 2. The results

show that at low SF the performance of model 2 is better than model 3 and by increasing the SF the performance of model 3 is better than model 2 which refers to the importance of different SFs in this analysis.

The results show that when the strain-rate effect is taken into account, the yield stress increases because the elastic modulus is higher than the plastic modulus. It is noted that the analysis with strain rate will be much stiffer, resulting in a decrease in the mid-point displacement as in [26–28].

5.3. Damping Effect. Under the effect of load applying, the undamped structures continue to vibrate with constant amplitude. A constant amplitude vibration is not the response that would be expected in practice since the vibrations in this type of structure would tend to die out over time and effectively disappear. The energy loss typically occurs by a variety of mechanisms, so it is needed to consider the presence of damping in the analysis to model this energy loss. The results of the damped analysis clearly show the effect of mass proportional damping.

Figure 12 shows the displacement history of the central node for both the damped and undamped material for all models at different SFs. It show that the material with damping analysis leads to decreasing the maximum displacement. For model 1, the maximum displacement decrease from 99.69 mm, 117.2 mm, and 132.7 mm to 93.7 mm, 110.1 mm, and



- Without damping (SF = 0.2)
- ...●... With damping (SF = 0.2)
- ▲- Without damping (SF = 0.238)
- ...▼... With damping (SF = 0.238)
- ◆- Without damping (SF = 0.27)
- ...◀... With damping (SF = 0.27)

- Without damping (SF = 0.2)
- ...●... With damping (SF = 0.2)
- ▲- Without damping (SF = 0.238)
- ...▼... With damping (SF = 0.238)
- ◆- Without damping (SF = 0.27)
- ...◀... With damping (SF = 0.27)

FIGURE 12: Damping effect on the displacement in the central node.

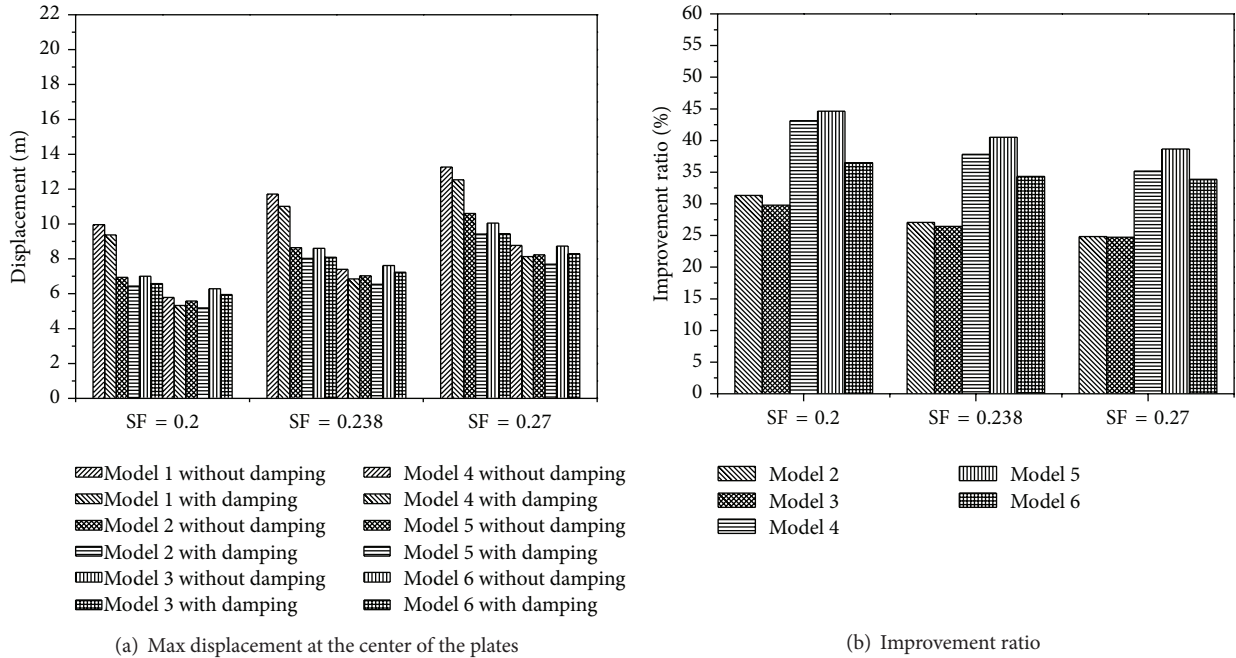


FIGURE 13: Damping effect on the plates performance.

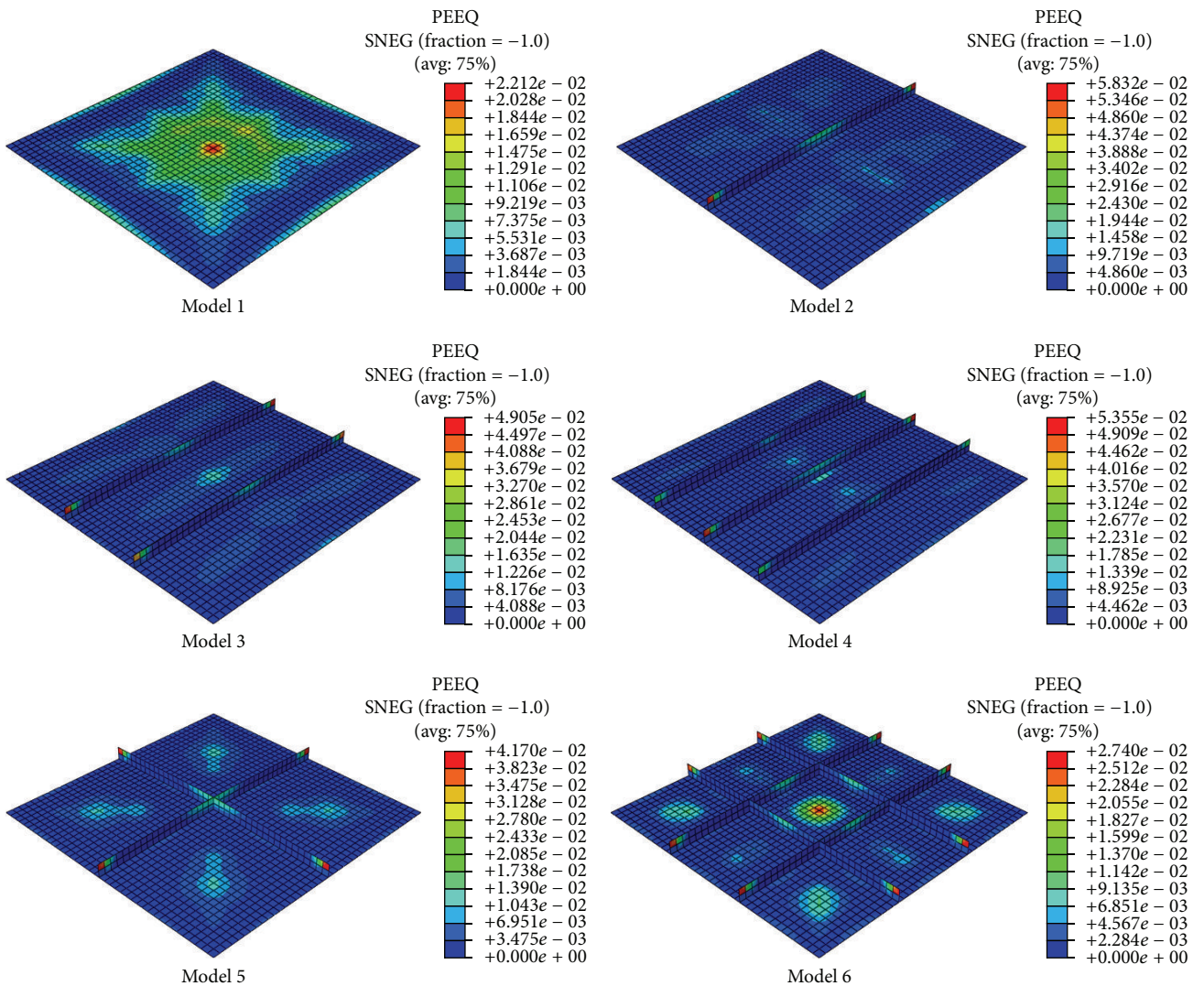


FIGURE 14: Contour plot of accumulated equivalent plastic strain (PEEQ).

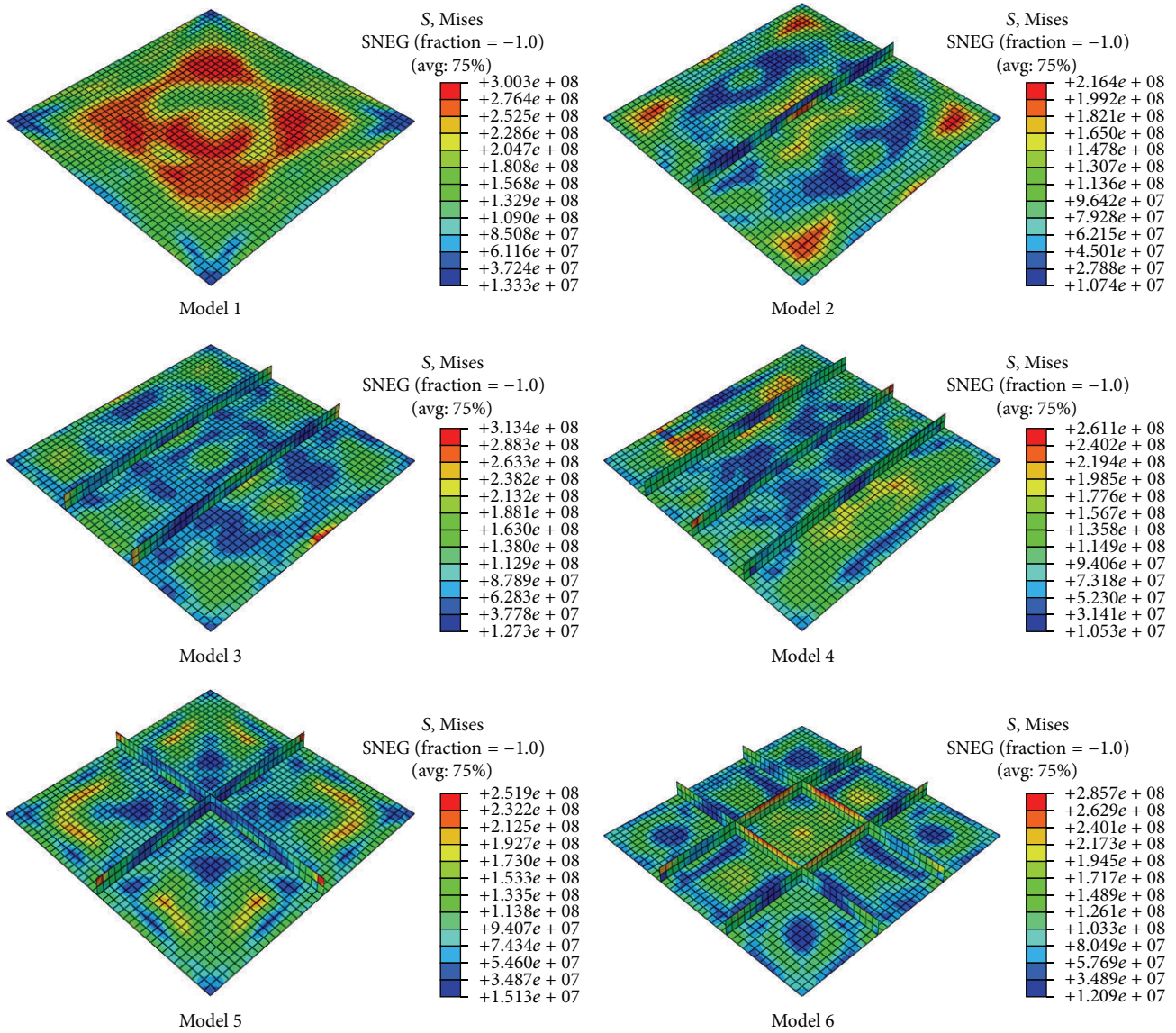


FIGURE 15: Contour plot of von Mises stress at the end of the analysis.

125.4 mm at SF equal to 0.2, 0.238, and 0.27, respectively. Also Figure 12 shows that the time it takes for all the models to reach maximum displacement at the same position is roughly the same.

Figure 13(a) shows the values of maximum central displacement in case of damped and undamped material and, at SF equal to 0.2, the best performance is model 5 which has the least deformation. The material with damping analysis leads to decreasing the maximum displacement up to 47.97% (model 5), 46.35% (model 4), 40.31% (model 6), 35.45% (model 2), and 34% in model 3. These values are based on the comparison to the value obtained from model 1 without damping. Figure 13(b) shows the improving ratio I_R in case of damped material analysis based on the comparison to the value obtained from model 1 with damping. It can be noted that as the SF increases the I_R decreases and the best performance is model 5 with I_R of 44.6%, 40.53%, and 38.7%

at SF equal to 0.2, 0.238, and 0.27, respectively. This confirms that the proposal with stiffeners can help the structure to sustain shock loads resulting from an underwater explosion and this result is matching the one in [27]. So, it can be concluded that the damping properties play a very important role in the response prediction due to underwater shock load.

5.4. *Structural Integrity.* Structural integrity is one of the major concerns of such analysis. The equivalent plastic strain in a material (PEEQ) is a scalar variable that is used to represent the material's inelastic deformation. If this variable is greater than zero, the material has yielded [17]. Figure 14 shows a contour plot of accumulated PEEQ for all models and permanent deformations are monitored which emphasize that the largest permanent deformation is obtained on model 1 and model 6. The FE analysis predicts the central peak

observed and permanent deformations which first occurred at the middle of the sides of the plates and then progressed towards the corners according to experimental results reported in [24–26]. The lowest permanent deformation is obtained on model 2, model 4, and model 5. Figure 15 shows a contour plot of von Mises stress at the end of the analysis. The regions of high stress concentration are analyzed which confirm that model 5 offered greater resistance and its I_R is 16.13%. It is found that I_R for models 4 and 6 is 13.05% and 4.86%, respectively. The lowest I_R is for model 3, which matches the experimental results reported in [24, 25, 29].

6. Conclusions

From the numerical simulation of stiffened plates subjected to noncontact underwater explosion the following can be concluded.

The displacement-time histories under different shock loadings are presented which will be used in designing stiffened panels to enhance resistance to underwater shock damage. The effect of stiffener configurations is very important, since it can drastically affect the overall behavior of the plates as indicated. The proposal with special damping system can help the structure to sustain shock loads resulting from an underwater explosion. Model 5 offered greater resistance to UNDEX load than all models which assure that stiffener location influenced the plate response. The inclusion of strain-rate effect and the proposal with special damping system results in lower mid-point displacement. Thus, the damping and the strain-rate effect should be taken into account when analyzing structures subjected to underwater shock loading. The inclusion of strain-rate effect is more effective than the proposal with special damping system to reduce the displacement of the plate.

Conflict of Interests

The authors declare that there is no conflict of interest is regarding the publication of this paper.

Authors' Contribution

Professor Lili Tong and Dr. Mahmoud Helal have contributed equally to this work.

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