

Experimental study of dislocation damping using a rock analogue

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Key Points:

- The dominant deformation mechanism of borneol changes from diffusion to dislocation creep at about 2MPa
- The attenuation spectra of a deformed sample display a high frequency peak superposed on a high-temperature background
- The strain-induced microstructure and anelastic response demonstrate the potential importance of dislocations

Abstract

In order to explore the effects of dislocations on seismic velocity and attenuation, we conducted a series of forced oscillation and ultrasonic tests on rock analogue samples (polycrystalline borneol) that were pre-deformed under various differential stress $\Delta\sigma$. Additionally, creep experiments were conducted to determine the steady-state flow law for borneol. The dominant deformation mechanism of polycrystalline borneol changes from diffusion to dislocation creep at about $\Delta\sigma = 2$ MPa. At high stresses, power law creep with a stress exponent of ~ 4 was measured. Microstructure of the deformed samples showed wavy grain boundaries due to dislocation-induced migration, and the occasional existence of microcracks. A borneol sample deformed in the dislocation creep regime showed a significant reduction in Young's modulus E and a slight increase in attenuation Q^{-1} at frequencies lower than 100 Hz, whereas E at ultrasonic frequency (10^6 Hz) did not reduce. Therefore, a major part of the dislocation creep-induced anelastic relaxation is a peak with a characteristic frequency between 100 and 10^6 Hz, which is much higher than the range of grain boundary-induced anelasticity of this material. Further experiments under higher confining pressure are needed to assess the relative contribution from dislocations and microcracks to this peak.

1 Introduction

Dislocations are linear defects, whose mobility (glide and climb) through the crystalline lattice facilitates shear deformation. Although dislocation density is considered to be controlled by the magnitude of long-term background stress (Kohlstedt & Goetze, 1974), the stress magnitude in the Earth's interior is subject to large uncertainty. Dislocations and dislocation related microstructures, such as subgrain boundaries, are commonly observed in mantle nodules (e.g., Toriumi & Karato, 1978). Hence, dislocations as well as grain boundaries are considered to be pervasive in the mantle. Seismic waves propagating through these rocks drive dislocation motion and grain boundary sliding, causing dispersion and attenuation of the waves (rock anelasticity) (e.g., Karato & Spetzler, 1990). Understanding of rock anelasticity is necessary to interpret the seismological structures in the Earth. Although significant progress has been made in the understanding of anelasticity associated with grain boundary sliding (Cooper, 2002; Gribb & Cooper, 1998; Jackson & Faul, 2010; Jackson et al., 2004, 2014, 2002; McCarthy et al., 2011; Morris & Jackson, 2009; Raj, 1975; Takei et al., 2014; Yamauchi & Takei, 2016),

46 the effects of dislocations on rock anelasticity are poorly understood. Two preliminary
47 studies explored the influence of dislocations on attenuation with olivine samples that
48 were pre-deformed in dislocation creep (Farla et al., 2012; Guéguen et al., 1989). Although
49 these studies demonstrated a clear dependence of attenuation on dislocation density, a
50 comprehensive understanding of the mechanisms, needed for a seismological application,
51 is lacking.

52 Anelasticity measurements using a forsterite single crystal (Guéguen et al., 1989)
53 or olivine aggregates (Farla et al., 2012) are difficult because of the high temperature
54 and/or high confining pressure needed to measure the anelastic relaxations active in the
55 mantle. In studying anelasticity caused by grain boundary mechanisms, analogue exper-
56 iments using organic polycrystals (high-purity borneol and borneol+diphenylamine bi-
57 nary eutectic system), that have a low melting temperature, have been playing an im-
58 portant role by providing well constrained data over a broad frequency range (McCarthy
59 & Takei, 2011; McCarthy et al., 2011; Takei et al., 2014; Yamauchi & Takei, 2016). In
60 this study, by using the same analogue material (high-purity borneol) as these previous
61 studies, we performed a series of experiments to investigate the effects of dislocations on
62 anelasticity. Because the effects of grain boundary mechanisms on the anelastic prop-
63 erties of this analogue material are well known, the effects of dislocations can be detected
64 as a deviation from the grain boundary effects.

65 Borneol is designated a “plastic organic crystal”, which is considered to undergo
66 ductile deformation by the same kinds of dislocation and diffusion processes as miner-
67 als, metals, and ceramics (Sherwood, 1979). However, a deformation mechanism map
68 of borneol was not available, and hence had to be determined in this study. In the fol-
69 lowing sections, we first report an experiment newly conducted to investigate the flow
70 law of borneol, which clarifies the stress condition required for dislocation creep, and which
71 enabled us to investigate the effects of dislocations on the anelastic properties of a bor-
72 neol polycrystal.

73 **2 Experimental Details**

74 **2.1 Sample fabrication**

75 Samples and testing conditions used in this study are summarized in Table 1. Poly-
76 crystalline aggregates of organic borneol ($C_{10}H_{18}O$, melting temperature = 204.5°C, D’Ans

Table 1. High-purity borneol samples used in this study.

Sample No. (Geometry) ^a	Density (10 ³ kg/m ³)	Porosity ^b (%)	Grain size ^c d (μm)	Mechanical tests ^d	Final strain ^e ε_{fin}	Barrel distortion ^f
#120 (B)	1.006	0.5	15.3	creep (40°C)	0.10	0.004
#124 (B)	1.006	0.5	14.9 ^g	creep (40°C)	0.45	0.07
#121 (B)	1.007	0.4	-	creep (50°C)	0.36	0.01
#121A (A)	1.002	0.9	24.1 ^g	creep (50°C) without P	0.09	-
#122 (C)	1.005	0.6	16.2	creep & forced oscillation	0.16	0.01
#123 (B)	1.007	0.4	16.6	creep & ultrasonic	0.21	0.008
#127 (B)	1.008	0.3	17.7	creep & ultrasonic	0.13	0.004

^aDiameter $2R$ (mm) and length L (mm) are, respectively, 15 and 13 for A, 30 and 30 for B, and 30 and 65 for C.

^bCalculated from the sample density relative to the density of borneol, $1.011 \times 10^3 \text{ kg/m}^3$.

^cGrain size was not measured when deformed microstructure was far from equilibrium.

^dEach type of test (creep, forced oscillation, and ultrasonic) was performed using a different apparatus.

^eCalculated as $(L_i - L_f)/L_i$ from the initial (L_i) and final (L_f) sample lengths.

^fCalculated as $(2D_m - D_t - D_b)/(D_t + D_b)$ from the top (D_t), middle (D_m), and bottom (D_b) diameter.

^gBefore deformation.

et al., 1964; Takei, 2000) were prepared following the procedure in Takei et al. (2014).

Fine powder ($\sim 3 \mu\text{m}$) of high-purity borneol, prepared from a cold-ball milling (-40°C ,

6 days), was placed in a cylindrical die and pressed to 13.9 MPa at room temperature

for 4 days to produce fully dense, polycrystalline right circular cylinders (diameter $2R$,

length L). Then, isothermal annealing was performed at about 3 MPa and 60°C for 8

days, so that grain growth during the mechanical experiments in the following sections

was negligible. (Sample #120 was annealed at about 3 MPa and 50°C for 5 days.) Af-

ter annealing, the sample was removed from the die. The initial sample geometry and

mass were measured accurately and were used to calculate the initial sample density and

porosity. The samples were almost transparent, consistent with small porosity (support-

ing information Fig. S1). Further details of the sample preparation are presented in Takei

et al. (2014).

90 The equilibrium microstructure of the as-fabricated polycrystalline aggregates of
91 high-purity borneol is known from previous studies (McCarthy et al., 2011; Takei et
92 al., 2014). Therefore, in this study, microstructure was observed mostly after the me-
93 chanical tests to investigate the effects of dislocations. After all mechanical tests, the cylin-
94 drical samples were cut in half lengthwise parallel to the axis, and the cross section was
95 polished to a mirror finish by a microtome and observed using a confocal laser micro-
96 scope (Optelics C130, Lasertec). Mean grain size was measured by the line intercept method
97 with a correction factor of 1.5 (Underwood, 1970). Each specimen was analyzed in three
98 positions and the values attained were averaged (about 400 intercepts per sample). For
99 samples #124 and #121A, microstructural observation was also performed before the
100 mechanical tests, by polishing the top surface.

101 2.2 Experimental apparatuses

102 Three different experimental apparatuses were used in this study: triaxial defor-
103 mation apparatus, uniaxial forced-oscillation apparatus, and ultrasonic testing appara-
104 tus. The triaxial deformation apparatus was used for the creep tests to determine the
105 flow law and for deformation of samples before anelasticity tests ('creep' in Table 1). A
106 confining pressure of 0.8 MPa was applied to suppress microcracking during dislocation
107 creep of borneol samples. The forced-oscillation apparatus was used for accurate mea-
108 surement of anelastic properties under a small cyclic stress over a broad frequency range
109 (10^2 – 10^{-4} Hz). The ultrasonic testing apparatus was used to non-destructively measure
110 modulus and attenuation at 1 MHz.

111 In the triaxial deformation apparatus, a custom triaxial cell was placed in a com-
112 mercial loading frame (Instron Co., Ltd., 5567, maximum load 3000 kgf). Fig. 1 shows
113 a schematic illustration of the triaxial cell consisting of a pressure vessel (maximum con-
114 fining pressure 1 MPa) and a uniaxial loading piston supported by a linear ball bushing
115 (Tatsuoka, 1988). The pressure vessel is made of a thick transparent acrylic pipe that
116 is reinforced with steel rods. The vessel was filled with water, leaving a small volume of
117 air at the top. The pressure in the vessel was kept constant by connecting this air space
118 to an air compressor through a pneumatic pressure regulator (Fairchild Co., Ltd., 10262U).
119 The pressure in the vessel was monitored by a pressure gauge (Kyowa Co., Ltd., PGMC-
120 A). Temperature inside the vessel (ambient $\leq T \leq 60^\circ\text{C}$) was monitored and controlled
121 using a type K thermocouple and heater, which were connected to a PID controller (Om-

ron Co., Ltd., E5ER). The water in the vessel was stirred by circulating fans. Load F^* and displacement u of the loading piston of the vessel were measured by the load cell and encoder, respectively, of the loading frame (sampling frequency was 1 Hz). We confirmed that friction on the piston was negligible by equivalent load values measured by inner and outer load cells. Differential load on the sample, F (N), was calculated as $F = F^* + Mg - F_P$, where F^* is the load applied by the loading frame, M (=1.5 kg) is the mass of the uniaxial loading piston, g is gravitational acceleration, and $F_P = \pi r^2 P$ is the upward force applied to the piston with radius r (=10 mm) by the confining pressure P in the pressure vessel (Fig. 1). Based on the ability of the air compressor and the yield strength of the acrylic pressure cell, a maximum confining pressure of 0.8 MPa and differential stress $\Delta\sigma$ of 3 MPa can be applied to the samples.

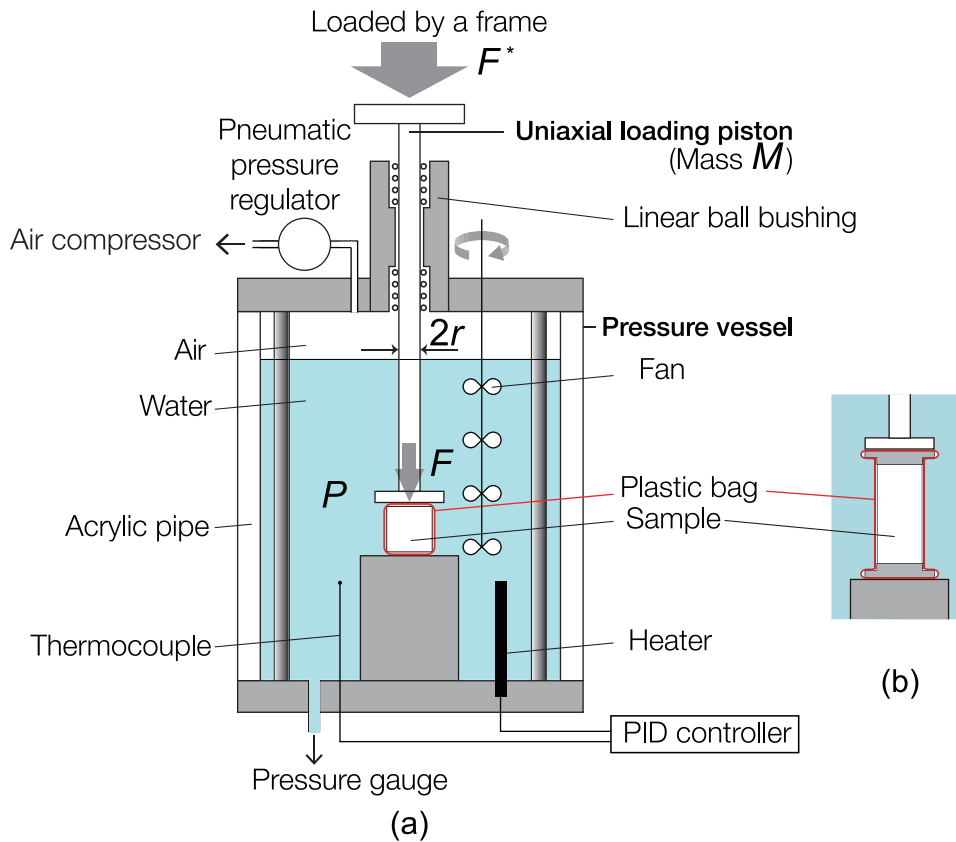
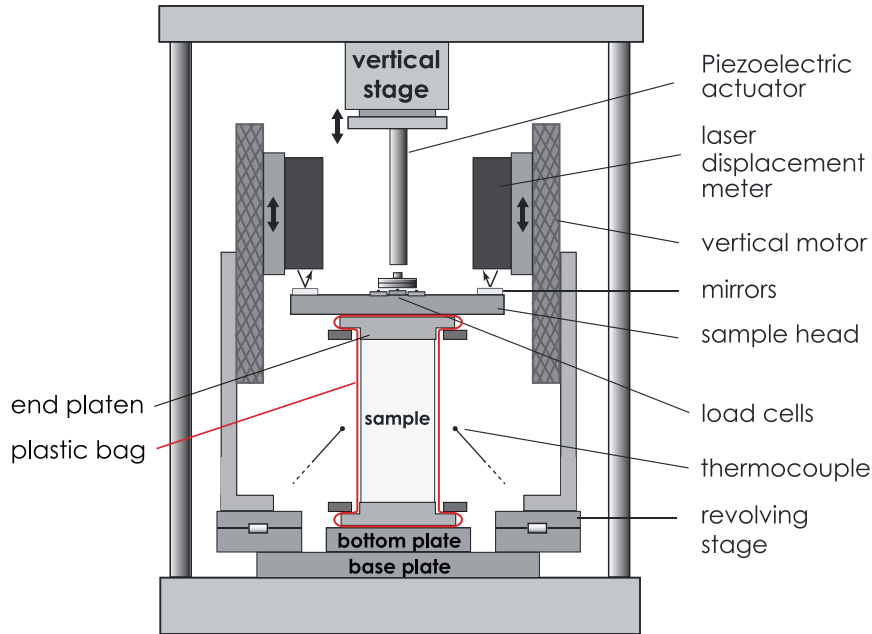


Figure 1. (a) Triaxial cell consisting of a pressure vessel and a uniaxial loading piston. The volume of air is compressed to provide up to 0.8 MPa of confining pressure in the vessel. This sample configuration was used for experiments described in sections 2.3 and 2.5. (b) Sample configuration for experiments described in section 2.4.

137 The uniaxial forced-oscillation apparatus used in this study is shown in Fig. 2 and
 138 described fully elsewhere (Takei et al., 2014). Cyclic compressive loading tests from 10^2 Hz
 139 to 10^{-4} Hz and creep tests can be performed under various temperatures ($0^\circ\text{C} \leq T \leq$
 140 50°C) and ambient pressure. The maximum differential stress $\Delta\sigma$ that can be applied
 141 to the samples by this apparatus is 0.28 MPa, which is much smaller than that needed
 142 for dislocation creep of borneol to be dominant.

143 The ultrasonic testing apparatus is shown in Fig. 2 in Takei (2000). Details of the
 144 apparatus used in this study are described in section 2.4 of Takei et al. (2014). In this
 145 study, this apparatus sits at room temperature and ambient pressure.



146 **Figure 2.** Forced oscillation apparatus. From Yamauchi and Takei (2016).

147 2.3 Flow law measurement

148 The steady-state deformation behavior of polycrystalline material is described by
 149 the semi-empirical flow law relating strain rate $\dot{\epsilon}$ to stress $\Delta\sigma$, grain size d , and temper-
 150 ature T :

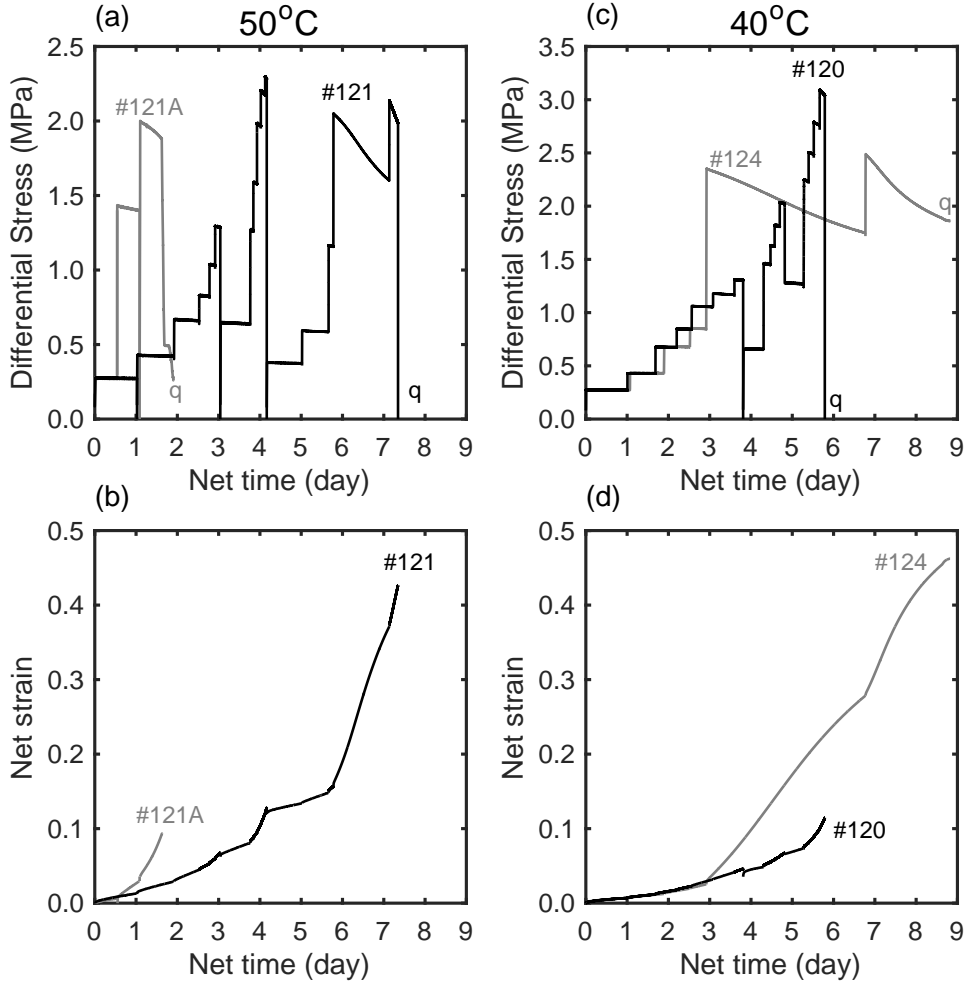
$$\dot{\epsilon} = A(\Delta\sigma)^n d^{-p} \times e^{-H/R_g T}, \quad (1)$$

151 where stress exponent n , grain size exponent p , and activation enthalpy H are specific
 152 to the deformation mechanism dominating at a given set of conditions. A and R_g rep-

153 represent a material constant and the gas constant, respectively. Since no previous defor-
 154 mation mechanism map has been developed for borneol, our first task was to explore the
 155 steady-state deformation behavior so as to determine the conditions at which disloca-
 156 tions are introduced into the sample by dislocation creep. Using the triaxial deforma-
 157 tion apparatus, creep tests were performed under various differential stresses $\Delta\sigma$ from
 158 0.28 to 3 MPa and a constant confining pressure of 0.8 MPa. Samples #120 and #124
 159 were tested at 40°C, and #121 was tested at 50°C. An additional sample (#121A) was
 160 tested at 50°C without confining pressure to check the sensitivity of the result to con-
 161 fining pressure.

162 Each sample was sealed in a nonreactive plastic bag and placed in the triaxial cell
 163 (Fig. 1a). The plastic bag (thickness = 77 μm) prevents water invasion, and also reduces
 164 the friction at the top and bottom surfaces of the sample. After filling the cell with wa-
 165 ter, temperature was increased to a run temperature. After a thermal equilibration (2
 166 hours), confining pressure was increased to 0.8 MPa (with the exception of #121A) and
 167 kept constant. Then, mechanical tests were performed by changing the uniaxial load step-
 168 wise (within ten minutes) to a relative maximum load, after which we removed the load.
 169 We refer to each stepwise segment as a creep test, such that a series of creep tests was
 170 performed on samples #120, #121, #121A, and #124 (Fig. 3). For each k -th creep test
 171 ($k = 1, 2, \dots$) performed under a constant differential load F (N), time dependent sam-
 172 ple length $L(t)$ was calculated from the initial length, L_i , and the displacement at t , $u(t)$,
 173 as $L(t) = L_i - u(t)$, where $t = 0$ represents the starting time of the creep test. Strain
 174 rate $\dot{\epsilon}(t)$ was calculated as $\dot{u}(t)/L(t)$. Differential stress $\Delta\sigma(t)$ was calculated as $\Delta\sigma(t) =$
 175 $\frac{F}{\pi(R(t))^2}$, where sample radius $R(t)$ was calculated from the initial radius R_i and $L(t)$
 176 by assuming a constant volume: $\pi R_i^2 L_i = \pi R^2(t) L(t)$. Under a constant F , $\Delta\sigma$ grad-
 177 ually decreases due to thickening of the sample (Figs. 3a and 3c). The net strain shown
 178 in Figs. 3b and 3d was calculated as $u(t)/L_i$.

183 After the last creep test on samples #120, #121, and #121A, we removed the ax-
 184 ial load and confining pressure, and quenched the sample to room temperature. We quenched
 185 #121 and #120 quickly by exposing them to cool tap water (#121) or room air (#120),
 186 and quenched #121A slowly by allowing the water in the cell to cool naturally using only
 187 the circulating fans inside the vessel (about 8 hours). In contrast, after the last creep
 188 test on #124, the last (highest) axial load and confining pressure were maintained, while
 189 the water was allowed to slowly cool to room temperature using the circulating fans in-



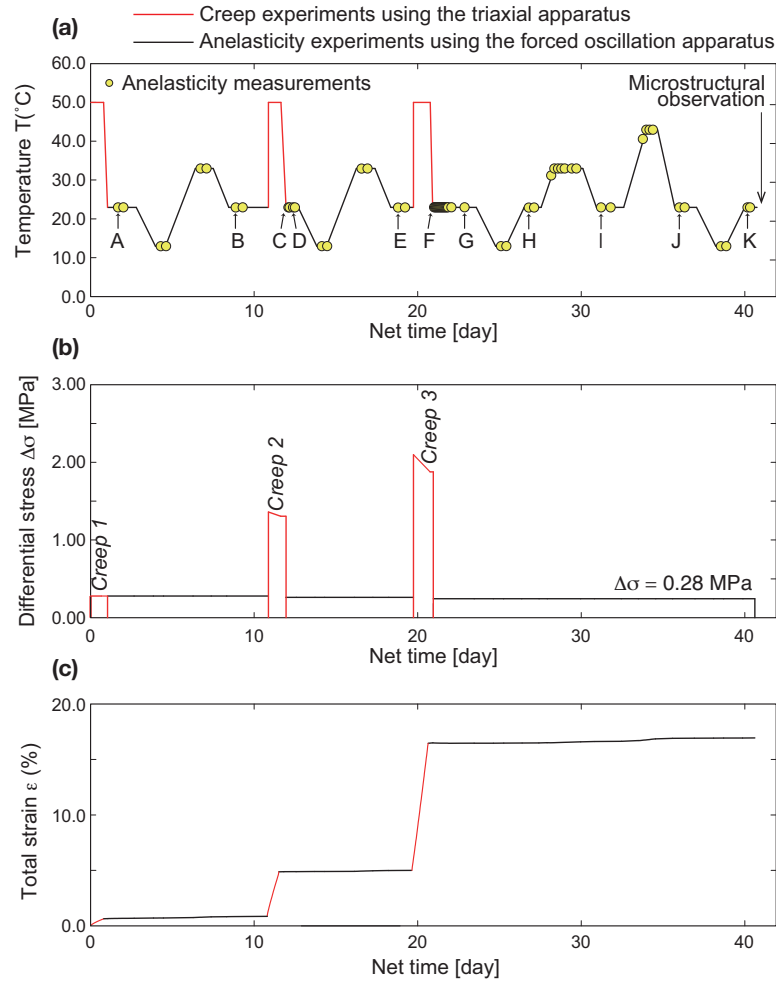
179 **Figure 3.** A series of creep tests performed on samples #120, #121, #121A, and #124: (a,
 180 c) differential stress and (b,d) net strain. Each creep test was performed under a constant differ-
 181 ential load. Stepwise change in stress took about 10 mins. Letter ‘q’ in panels (a) and (c) means
 182 ‘quench’.

190 side the vessel and an electric fan standing outside the vessel (about 6 hours). We then
 191 removed the axial load first and confining pressure next. This slow, and hence nonde-
 192structive, quench under the deviatoric stress was also used in the anelasticity experiments
 193 in sections 2.4 and 2.5. The final sample geometry (length and diameter at the top, mid-
 194dle, and bottom part) was measured accurately to calculate final axial strain ε_{fin} and bar-
 195rel distortion (Table 1).

2.4 Anelasticity measurement of a deformed sample

In order to investigate the effect of dislocations on anelasticity, we performed forced oscillation experiments on a sample that was deformed in both the diffusion and dislocation creep regimes (as determined from the flow law study) and measured the resulting change in anelastic behavior. Fig. 4 shows the variations of temperature, differential stress, and strain of sample #122 imposed during the testing protocol. The red parts in Fig. 4 were conducted in the triaxial deformation apparatus (Fig. 1), and the black parts were conducted in the uniaxial forced-oscillation apparatus (Fig. 2).

In order to conduct the multi-step testing shown in Fig. 4 on the single specimen, sample #122 (diameter = 30 mm, length = 65 mm) was assembled as follows. After attaching two brass end platens, the sample and platens were sealed in a nonreactive plastic bag, while air was removed by a vacuum pump. The end platens are needed to attach the sample tightly to the forced oscillation apparatus (Fig. 2). The vacuum sealed “sample + end platens” is hereafter referred to as the sample unit. We first placed the sample unit in the triaxial deformation apparatus (Fig. 1b), and deformed the sample under differential stress $\Delta\sigma = 0.28$ MPa, confining pressure 0.8 MPa, and temperature $T = 50^\circ\text{C}$ for about 19 hours (referred to as creep 1). Then, we quenched the sample under deviatoric stress (slow quench described in §2.3, which takes about 6 hours), removed the sample unit from the triaxial cell, and attached it to the forced oscillation apparatus. As shown in Fig. 4a, we then measured the anelastic property of this sample first at 20°C , then at 10°C and 30°C , and finally at 20°C again. One yellow circle in Fig. 4a represents a series of forced oscillation tests, in which Young’s modulus E and attenuation Q^{-1} were measured for 8–20 frequencies between $68.1\text{--}10^{-4}$ Hz, where the highest 8 frequencies between $68.1\text{--}1$ Hz were always included. We repeated this measurement at each temperature, so that reproducibility of the data could be checked and a temporal variation of anelasticity, if any, could be detected. The zero-to-peak amplitude of the cyclic stress and strain was less than 0.05 MPa and 2×10^{-5} , respectively. As shown in Fig. 4b, in the forced oscillation apparatus, constant differential stress $\Delta\sigma = 0.28$ MPa was always applied to the sample. Therefore, we can also measure the diffusion creep viscosity of the sample (e.g., Takei et al., 2014). After all anelasticity measurements, we removed the sample unit from the forced oscillation apparatus and placed it again in the triaxial deformation apparatus to conduct the next increment of creep deformation (‘creep 2’ or ‘creep 3’ in Fig. 4b).



204 **Figure 4.** Temporal variation of (a) temperature, (b) differential stress and (c) uniaxial strain
 205 of sample #122 during the experiment to investigate the effect of deformation (creep 1,2,3) on
 206 anelastic properties. Creep 1, 2, and 3 (red parts) were performed in the triaxial deformation
 207 apparatus (Fig. 1b) and the other mechanical testing (black parts) were performed in the forced
 208 oscillation apparatus (Fig. 2). Each yellow circle in panel (a) indicates an anelasticity measure-
 209 ment. The measurements marked by letters are referred to in section 3.2.1.

235 Creep 2 was conducted under $\Delta\sigma = 1.3$ MPa for 17 hours, and creep 3 was con-
 236 ducted under $\Delta\sigma \simeq 2.0$ MPa for 23 hours, where the other conditions (temperature,
 237 confining pressure) were equal to those in creep 1. After each increment of creep, we quenched
 238 the sample and measured the anelastic property by following the same procedures as those
 239 after creep 1, but with a larger number of repetitions and for a wider temperature range
 240 (Fig. 4a).

241 The data obtained by the forced oscillation tests were analyzed in the same man-
 242 ner as Takei et al. (2014). In the previous study, however, deformation of the sample
 243 was small ($\varepsilon_{\text{fin}} \leq 0.016$), and hence in correcting the modulus data for the effect of no-
 244 slip boundary condition at the two end platens, the length/diameter ratio, $L/2R$, was
 245 fixed to 2.2. In this study, because $L/2R$ changes from 2.14 (after creep 1) to 1.67 (af-
 246 ter creep 3), we derived the correction factor as functions of both Poisson's ratio and length/diameter
 247 ratio by conducting a FEM calculation (Appendix A). Also, the stiffness of the appa-
 248 ratus used in the data correction was measured by a blank test with a stainless steel sam-
 249 ple that was superglued to the two end platens and vacuum sealed in a plastic bag.

250 Takei et al. (2014) estimated the random and systematic errors of Young's mod-
 251 ulus measurements as $< 2\%$ and $\leq \sim 3\%$, respectively. One of the important factors
 252 affecting this accuracy is contact between the sample and the two end platens. In this
 253 study, the sample unit was vacuum packed, so that perfect contact between the sample
 254 and the end platens, which is expected to be attained during creep 1, can be maintained
 255 even after removing the sample unit from the triaxial cell. In previous measurements of
 256 anelasticity with a plastic bag, air in the bag was not removed (McCarthy & Takei, 2011;
 257 McCarthy et al., 2011; Yamauchi & Takei, 2016), so that the plastic bag did not touch
 258 the sample. In that case, the bag did not affect the dissipation, as was confirmed by a
 259 comparison of the data to those obtained without a bag (e.g., Takei et al., 2014). In or-
 260 der to examine the effect of the vacuum-sealed bag on the anelasticity data, we cut the
 261 bag and let air in just before the first anelasticity measurement after creep 3 (just be-
 262 fore run F in Fig. 4a).

263 **2.5 Ultrasonic measurement of samples before and after deformation**

264 In order to constrain the characteristic frequency of the dislocation creep-induced
 265 anelasticity, we additionally measured the effects of deformation by dislocation creep on

266 the elastic properties of samples #123 and #127 using ultrasonic testing. Each sample
267 (diameter = 30 mm, length = 30 mm) was sealed in a nonreactive plastic bag; #123 was
268 sealed by leaving some air in the bag, similarly to the samples in the flow law measure-
269 ment in section 2.3, whereas #127 was sealed by removing air with a vacuum pump, sim-
270 ilarly to #122 in section 2.4. In the same manner as the flow law measurement, each sam-
271 ple was placed in the triaxial cell (Fig. 1a) and was deformed under $\Delta\sigma = 0.28$ MPa
272 for 26 hours (creep 1), $\Delta\sigma \simeq 1.7$ MPa for 25 hours (creep 2) and $\Delta\sigma \simeq 2.0$ MPa for
273 16 hours (creep 3) in succession. Run temperature was 50°C and confining pressure 0.8
274 MPa. We then quenched the sample under deviatoric stress (slow quench described in
275 section 2.3) and removed it from the triaxial cell.

276 Before and after the deformation, longitudinal and shear wave velocities, V_P and
277 V_S , of the sample were measured at room temperature ($\sim 22^\circ\text{C}$) and ambient pressure
278 by using the ultrasonic method. The velocities were calculated from P- and S-wave travel
279 times and sample length. The sample length was measured directly (without a plastic
280 bag) with a micrometer just before the ultrasonic test. Travel times were measured by
281 the pulse transmission method by attaching transducers (PS-dual transducers D7054,
282 Panametrics Inc., center frequency = 1 MHz) to the top and bottom surfaces of the sam-
283 ple, which was sealed in a plastic bag. The data were corrected for the time lags due to
284 the transducers, plastic bag, and coupler, as described in Takei et al. (2014). From the
285 measured V_P , V_S , and sample density (Table 1), Young’s modulus at the ultrasonic fre-
286 quency (1 MHz) was calculated. After ultrasonic testing was completed, we measured
287 the final sample geometry and cut the sample for microstructural observation.

288 **3 Results**

289 First, in section 3.1, we summarize the mechanical and microstructural results of
290 the creep tests performed on samples #120, #121, #121A, and #124 to determine the
291 flow law of borneol. Then, in section 3.2, we summarize the results of forced oscillation
292 and ultrasonic tests performed on the pre-deformed samples (#122, #123, and #127),
293 where creep data during the prior deformation stage and microstructures of these sam-
294 ples are summarized in section 3.1.

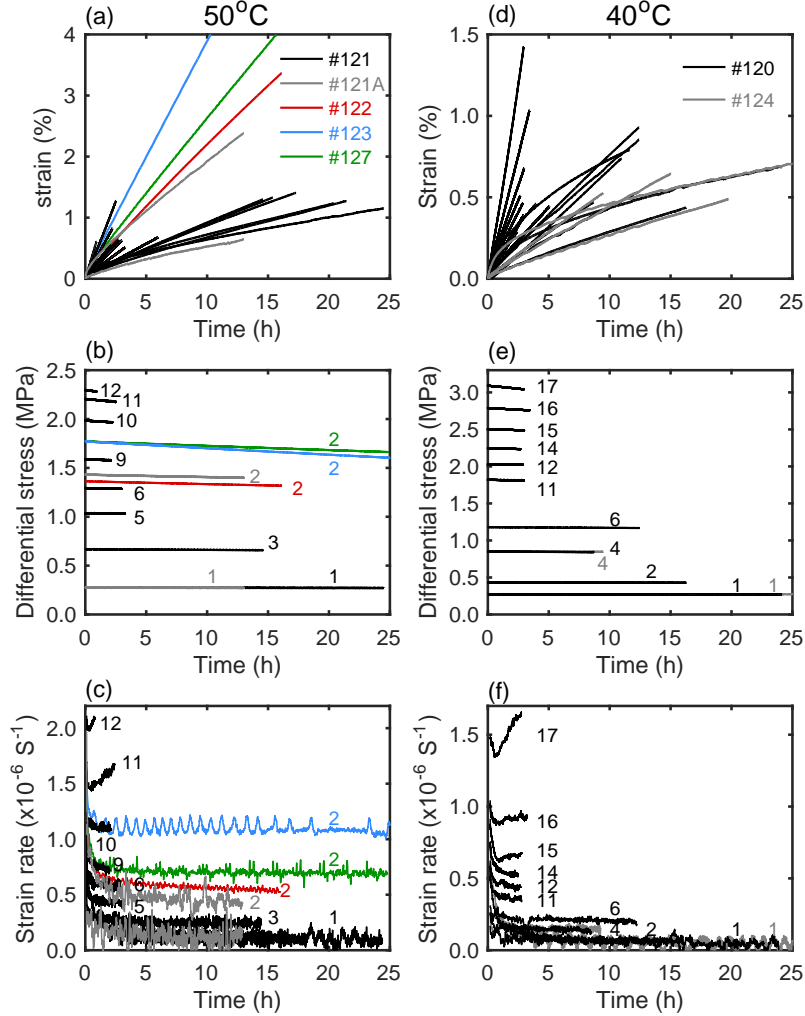
3.1 Flow law of borneol

3.1.1 Creep data

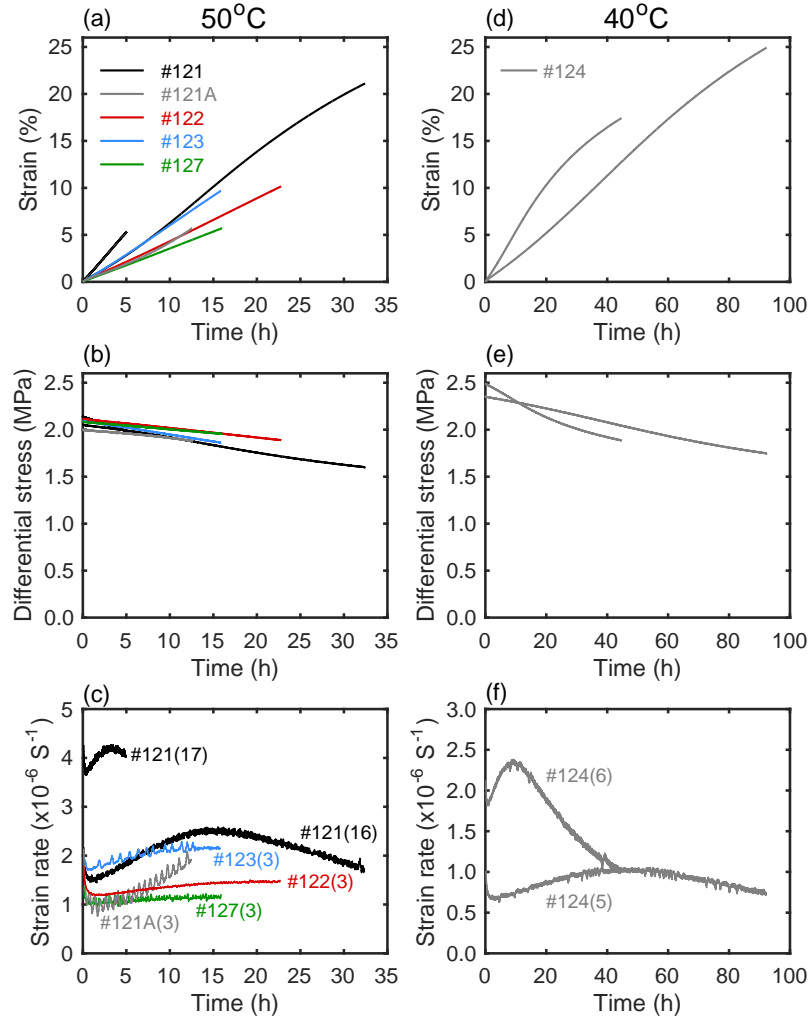
All creep curves obtained in the flow law measurement (Fig. 3) are shown in Figs. 5 and 6 by black and gray lines, where long creep curves obtained under high stress are shown in the latter. Several creep curves obtained during the creep tests performed before forced oscillation and ultrasonic tests are also shown in these figures by red (#122), blue (#123) and green (#127) lines to make up for the lack of long creep curves under medium stress around $\Delta\sigma \simeq 1.5$ MPa (Fig. 5) and high stress around $\Delta\sigma \simeq 2$ MPa (Fig. 6).

Most of the creep curves in Fig. 5 show a transient creep and a steady-state creep, in which stress and strain rate were nearly constant (middle and bottom rows of Fig. 5). These data can be used to determine the flow law and are plotted in Fig. 7 by the symbols. Fig. 7 shows that the flow law at $\Delta\sigma < 1$ MPa follows a linear stress-strain rate relationship (i.e. $n = 1$ in equation (1)), with viscosity $\eta = \Delta\sigma/\dot{\epsilon} = 2.8 \times 10^{12}$ Pas at 50°C (#121) and 6.2×10^{12} Pas at 40°C (#120, #124). Consistency of this result with the previous result for high-purity borneol (Takei et al., 2014) is shown in supporting information Fig. S2. Fig. 7 also shows that the flow law at $\Delta\sigma > 1$ MPa deviates from a linear relationship. The emergence of a nonlinear flow law (i.e., $n > 1$ in equation (1)) at higher stress can be also confirmed by wider vertical spacing in the strain rate plots (Figs. 5c and 5f) than that in the stress plots (Figs. 5b and 5e).

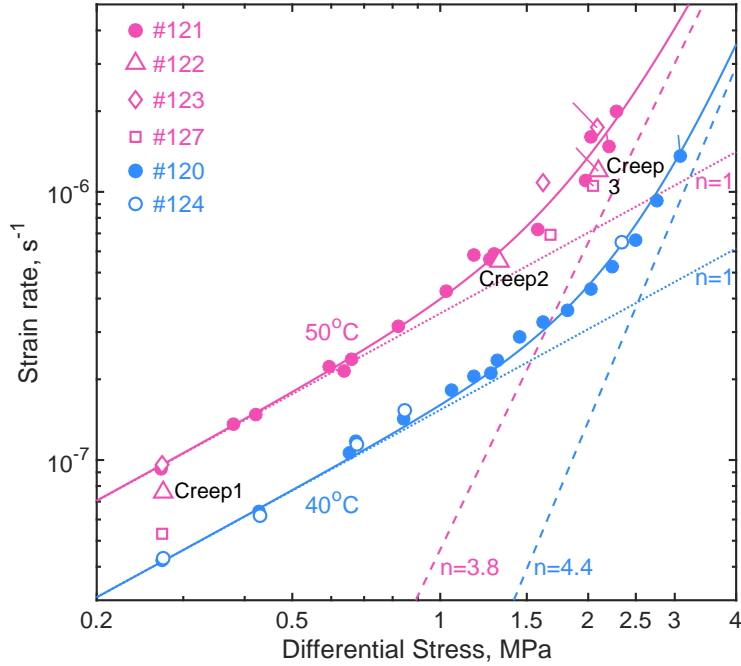
The three creep curves in Fig. 5 obtained under the highest two stresses at 50°C (creep 11 and 12 on sample #121) and the highest stress at 40°C (creep 17 on #120) showed a different behavior from the others: the initial strain-hardening stage was followed by a strain-weakening stage (Figs. 5c and 5f), and steady-state creep was not observed. For these three tests, the stress-strain rate data at the end of the initial strain hardening stage (that is, at the point of minimum strain rate) were plotted in Fig. 7 by circular symbols with an attached bar, such that the free end point of the bar indicates the data at the end of each creep test. Although these bars are short, if these creep tests had been performed for a longer time interval, the bars would have been longer. For a detailed investigation of the strain weakening stage, we performed four very long creep tests under high stress (creep 16 and 17 on #121, creep 5 and 6 on #124). As typically shown by creep 16 of #121 (Fig. 6c) and creep 5 of #124 (Fig. 6f), the obtained creep



304 **Figure 5.** Results of the creep tests on #121 and #121A at 50°C (left column) and on #120
 305 and #124 at 40°C (right column). (a,d) Strain, (b,e) differential stress (for selected creep tests),
 306 and (c,f) strain rate (for selected creep tests), versus time. Color shows sample and number k
 307 near the curve shows the k -th creep test on each sample (Fig. 3). Some results from #122 (red),
 308 #123 (blue) and #127 (green) are also shown to make up for the lack of long creep curves under
 309 medium stress around $\Delta\sigma \simeq 1.5$ MPa. Sample #121A was deformed without confining pressure.



310 **Figure 6.** Results of the creep tests performed for a long time under high stress ($\Delta\sigma \simeq 2$
 311 MPa). (a,d) Strain, (b,e) differential stress, and (c,f) strain rate, versus time. Color shows sam-
 312 ple, and number k in the parentheses in panels (c) and (f) shows the k -th creep test on each
 313 sample. Sample #121A was deformed without confining pressure.



325 **Figure 7.** Flow law of polycrystalline borneol. Color shows temperature (pink=50°C,
 326 blue=40°C), and symbol type corresponds to a sample. The pink and blue solid curves show
 327 the best fit models obtained from the data of #121 (50°C, pink) and #120 (40°C, blue), which
 328 are composed of linear creep (dotted line) and power-law creep (dashed line). Symbols with a bar
 329 indicate tests at high stress where samples demonstrated a strain weakening stage, such that the
 330 stress and strain rate conditions just before the strain weakening stage are shown by the symbol
 331 and the conditions at the end of the creep tests are shown by the free end point of the bar.

344 curves consisted of three stages: initial deceleration stage, second acceleration stage, and
 345 final deceleration stage. Because steady-state creep was not obtained at these high stresses,
 346 in order to estimate the steady-state flow law, it is important to clarify the physical pro-
 347 cesses responsible for the three stages.

348 When a creep test is conducted under a constant load, stress gradually decreases
 349 due to thickening of the sample. When stress is high, this thickening and hence the stress
 350 reduction are fast (Figs. 6b and 6e). Because strain rate is sensitive to stress at high stress,
 351 the final deceleration stage can be explained well by the response of the sample to the
 352 stress reduction. In other words, the third deceleration stage is not considered to rep-
 353 resent an intrinsic property of the sample. In contrast, it is not evident whether the sec-
 354 ond acceleration stage represents an intrinsic or extrinsic process of the sample. Strain

355 weakening prior to steady-state dislocation creep was reported in the deformation ex-
356 periments of olivine single crystals (Cooper et al., 2016; Hanson & Spetzler, 1994), in-
357 dicated the time-dependent evolution of dislocation density or dislocation microstruc-
358 ture in the deforming crystals. A similar process, if active in borneol crystal, can explain
359 the strain weakening stage, while the first strain hardening stage preceding it may be
360 explained by a similar mechanism to the transient creep in the diffusion creep regime (i.e.,
361 diffusively accommodated grain boundary sliding, McCarthy et al., 2011; Takei et
362 al., 2014). With this intrinsic mechanism, the weakening stage is expected to reach a steady
363 state, if the test is performed for a long time under a constant $\Delta\sigma$. However, there is an-
364 other possibility. If confining pressure is not high enough, microcracking and/or pore growth
365 are expected to occur at high stress, providing an alternative explanation for the strain
366 weakening stage. With this extrinsic mechanism, the weakening stage may not reach a
367 steady state even under a constant $\Delta\sigma$. Therefore, a creep test under constant $\Delta\sigma$ will
368 be helpful to clarify the mechanism, but it is difficult in the present apparatus.

369 In order to clarify the physical process responsible for the strain weakening stage,
370 we investigated the effect of confining pressure on this stage. Microcracking and/or pore
371 growth are expected to be strongly sensitive to confining pressure, whereas dislocation
372 generation and motion are not. The gray lines in the left column of Fig. 6 show the re-
373 sult of the creep test with $\Delta\sigma \simeq 2$ MPa without applying confining pressure (creep 3
374 on #121A). In comparing this result with the other results in Fig. 6 obtained under con-
375 fining pressure, it is important to introduce the concept of “effective confining pressure”.
376 We noted that strain weakening was small in samples #122 and #127 (red and green
377 lines in Fig. 6), which were sealed airtight, but was greater in #121, #123, and #124,
378 which were sealed with some air left in the plastic bag. This difference under the same
379 confining pressure can be explained by the effective confining pressure (confining pres-
380 sure minus pressure of the air remaining in the bag), which was 0.8 MPa in #122 and
381 #127, 0 in #121A, and unknown, but between these two values, in the other samples.
382 The amplitude of strain weakening showed a good correlation with the effective confin-
383 ing pressure (Figs. 6c and 6f), suggesting the extrinsic origin (due to microcracks or pores)
384 of the strain weakening stage. This result was also supported by the microstructural ob-
385 servation in section 3.1.2. Therefore, we do not use the data obtained during and after
386 the weakening stage in determining the steady-state flow law in Fig. 7.

397

Table 2. Flow law parameters in eq.(2)

	η	n	B
50°C(#121)	2.8×10^{12} Pas	3.8	4.226×10^{-8} (MPa) $^{-3.8}$ s $^{-1}$
40°C(#120)	6.2×10^{12} Pas	4.4	6.587×10^{-9} (MPa) $^{-4.4}$ s $^{-1}$

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The intrinsic creep strength around $\Delta\sigma \simeq 2$ MPa may be best estimated by the stress and strain rate data just before the weakening stage. For creep 16 on #121 and creep 5 on #124, such data were plotted in Fig. 7 by circular symbols, whereas creep 17 on #121 and creep 6 on #124 after experiencing a significant strain weakening (more than 20% strain) were not used in the flow law analyses. The data from #121A were not used, because #121A had larger grain size than the others (Table 1). We estimated the steady-state flow law at 50°C and 40°C by fitting the formula

$$\dot{\epsilon} = \Delta\sigma/\eta + B\Delta\sigma^n \quad (2)$$

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to the data sets obtained from #121 and #120, respectively, where the values of η estimated in the linear range were used. The best fit results are presented in Table 2 and plotted in Fig. 7 (pink and blue lines).

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3.1.2 Microstructure of the deformed samples

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We first report the results of microstructural observation supporting the occurrence of dislocation creep indicated by the emergence of power law creep. We then report the observation of microcracks and/or pores indicated by the emergence of the strain weakening stage.

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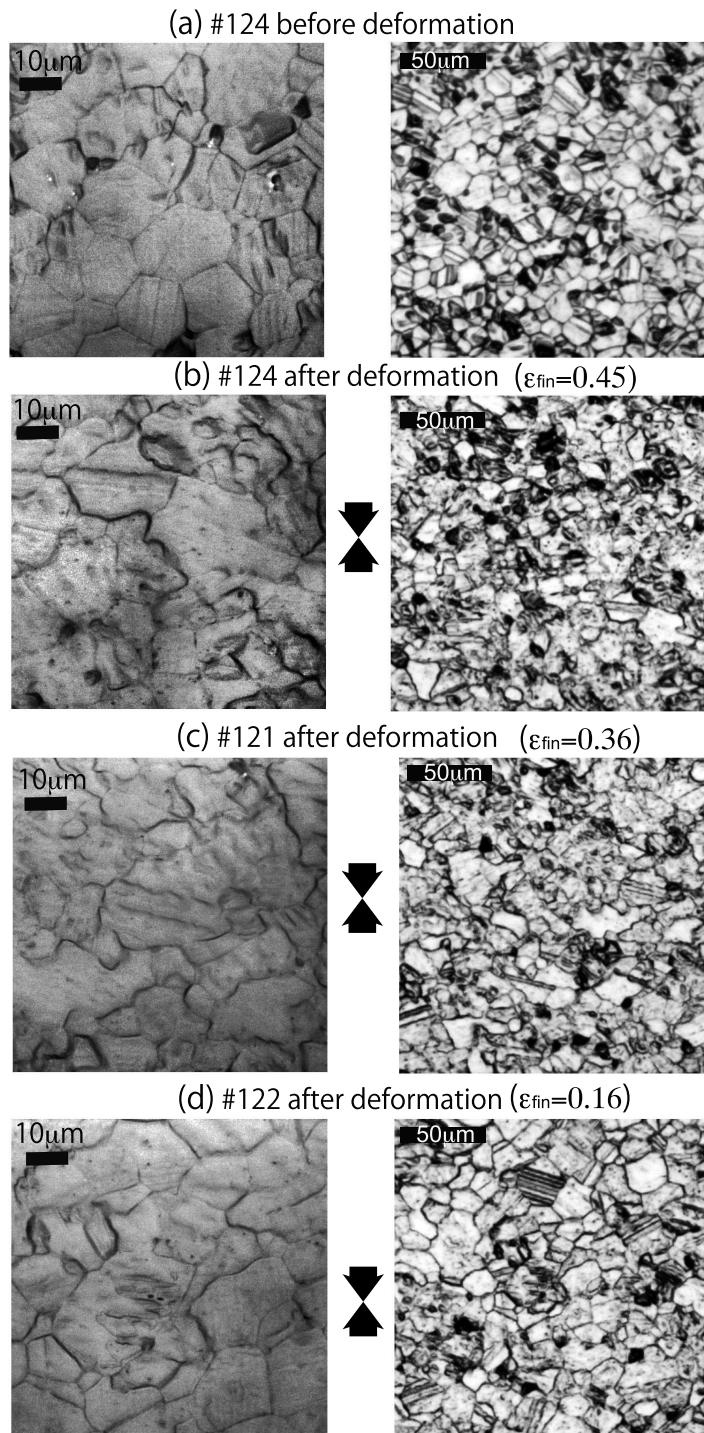
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Figures 8a–8b show the microstructure of sample #124 before and after deformation. During the high-temperature annealing (for grain growth) performed before the deformation, microstructure was equilibrated to minimize interfacial energy. As shown in Fig. 8a, grain boundaries in the undeformed sample are observed mostly as straight lines, which intersect at about 120° at the triple junctions. Grains are mostly polygonal and equi-axial. In contrast, after the creep tests in the power law regime, the boundaries are observed as wavy lines and the angles at the triple junctions frequently devi-

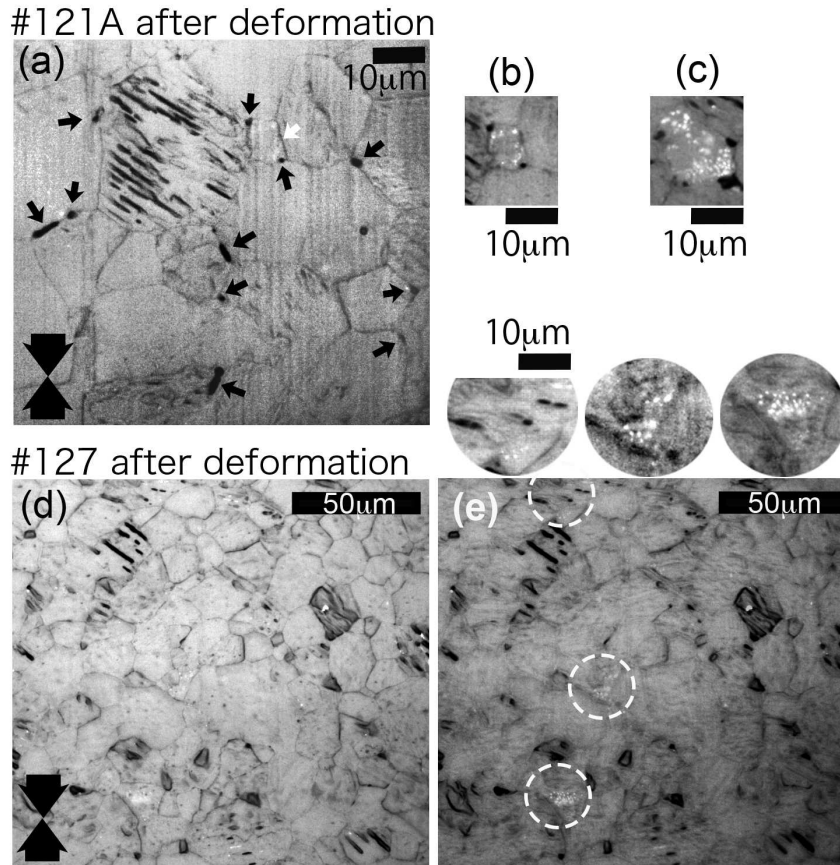


403 **Figure 8.** Light microscopic images of polycrystalline borneol at high and low magnification.
 404 (a,b) Sample #124 before and after deformation. (c,d) Samples #121 and #122 after deforma-
 405 tion. (b)–(d) show the images near the center part of the samples. Arrows show the uniaxial
 406 compression direction.

414 ate from 120° (Fig. 8b). Irregularly shaped grains, sometimes elongated in the horizon-
 415 tal direction, were observed frequently. The microstructure observed after diffusion creep
 416 was similar to the equilibrium microstructure in Fig. 8a (Fig. 4b–d in Takei et al., 2014).
 417 Hence, the disequilibrium microstructure in Fig. 8b indicates the occurrence of dislocation-
 418 induced grain boundary migration. We did not observe a remarkable change in grain size
 419 by dislocation creep.

420 The deviation from the equilibrium microstructure tends to increase with increas-
 421 ing final strain and/or temperature. As shown in Fig. 8c, similar disequilibrium texture
 422 in #124 ($\varepsilon_{\text{fin}} = 0.45$, 40°C) was observed in #121 ($\varepsilon_{\text{fin}} = 0.36$, 50°C) both of which
 423 were deformed to similarly high strains. Whereas in the microstructure of #121A ($\varepsilon_{\text{fin}} =$
 424 0.09), #122 ($\varepsilon_{\text{fin}} = 0.16$), #123 ($\varepsilon_{\text{fin}} = 0.21$), and #127 ($\varepsilon_{\text{fin}} = 0.13$) deformed at
 425 50°C , the deviation from the equilibrium texture was small compared to #121 and #124.
 426 A typical microstructure of these low strain samples is shown in Fig. 8d and from these
 427 samples we obtained grain size measurements. However, the linear intercept method does
 428 not work well with wavy grain boundaries, since each grain can cut a line more than twice.
 429 Hence, we measured grain size only for the near equilibrium microstructures (Table 1).

430 Figures 9a–9c show microstructures of sample #121A deformed without confining
 431 pressure. These figures were obtained near the outer cylindrical surface. We observed
 432 micropores (pores much smaller than grain size) at or near triple junctions and along
 433 grain boundaries (marked by small black arrows). The grain in Fig. 9a labelled by the
 434 white arrow is shown in Fig. 9b, where the faint spots can be interpreted as pores in the
 435 near subsurface. We occasionally observed multiple micropores on a single grain bound-
 436 ary (Fig. 9c), which may be called a crack. Based on these observations, we consider that
 437 the physical process responsible for the strain weakening stage was formation of micro-
 438 pores and cracks, which also explains well the observed sensitivity to the effective con-
 439 fining pressure. Because the strain weakening stage was slower at lower temperature (Fig. 6c
 440 and 6f), it may be a more ductile process than brittle fracturing. Micropores and cracks
 441 were mostly found within 1 mm of the outer cylindrical surface, were only occasionally
 442 found inside the cylinder, and rarely found near the top and bottom surfaces. Consis-
 443 tently, transparency of sample #121A was not lost even after the deformation (support-
 444 ing information Fig. S1).



445 **Figure 9.** Light microscope images of samples #121A and #127 observed near the outer
 446 cylindrical surface. (a, b, c) Microstructures of #121A deformed without confining pressure. (d,
 447 e) Microstructures of #127 deformed with confining pressure. Thick black arrows in (a) and (d)
 448 show the uniaxial compression direction. Small black arrows in (a) show deformation-induced
 449 micropores and white arrow shows the grain in (b). The faint spots in (b) and (c) can be inter-
 450 preted as pores in the near subsurface. Image (e) was obtained by focusing the microscope to
 451 a depth 1 μm below the surface shown in panel (d). Faint spots similar to (c) are found in the
 452 dashed circles, for which enlarged and highly contrasted views are provided above (e).

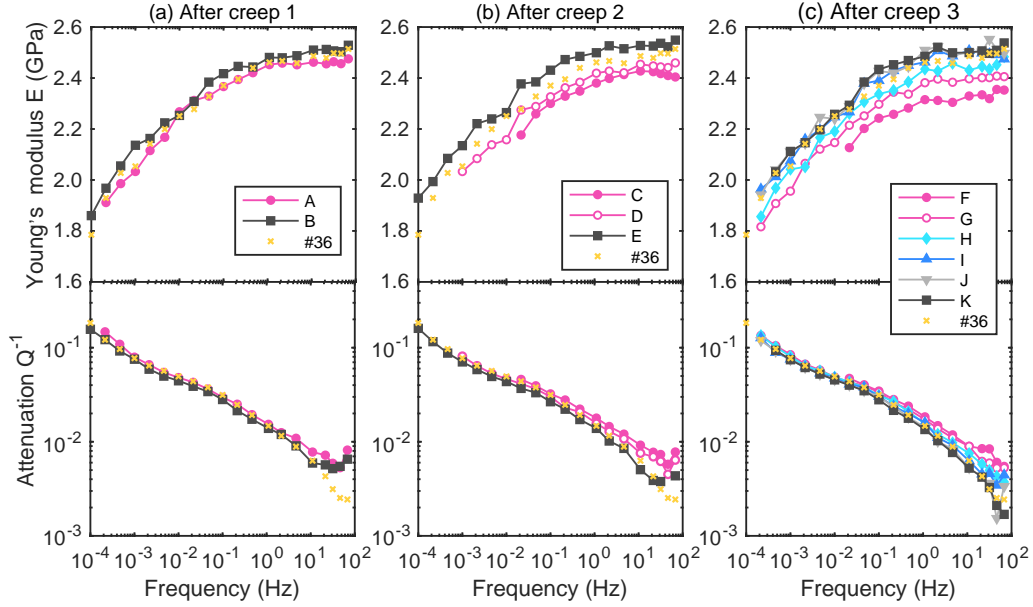
453 Figures 9d–9e show the microstructures of sample #127 deformed under the effec-
 454 tive confining pressure of 0.8 MPa. The strain weakening stage was much smaller in am-
 455 plitude than #121A, which had no confining pressure (Fig. 6c). Indeed, micropores sim-
 456 ilar to Fig. 9a were rarely recognized in this sample. Nevertheless, we did observe some
 457 cracks. Fig. 9e was obtained by focusing the microscope on the subsurface plane at a depth
 458 of $1 \mu\text{m}$ beneath that shown in Fig. 9d. We could find multiple micropores indicating
 459 cracks (dashed circles in Fig. 9e). Figs. 9d–9e were obtained near the outer cylindrical
 460 surface. Inside the cylinder, we found only one or two of such cracks per $S = 284 \times$
 461 $227 (\mu\text{m})^2$. By assuming that these cracks are 45° to the compression axis (*i.e.*, shear
 462 cracks), and mean radius a is given by $a = d/2 \simeq 10 \mu\text{m}$, the number density N is
 463 estimated as $N = (1 \sim 2)/(Sd/\sqrt{2}) (\mu\text{m})^{-3}$. Then, crack density parameter $c = Na^3$
 464 (e.g., O’Connell & Budiansky, 1974) can be semi-quantitatively estimated as $c = 0.002$ –
 465 0.004 . Even under the highest effective confining pressure, deformation in the disloca-
 466 tion creep regime is not free from microcracking. In #121A, micropores were more re-
 467 markable than microcracks, probably because #121A was slowly quenched without de-
 468 viatoric stress and hence thin cracks were able to heal.

469 3.2 Effects of dislocation creep on anelasticity

470 3.2.1 Results of forced oscillation tests

471 The stress and strain rate conditions of the deformation of sample #122 prior to
 472 the forced oscillation tests are shown in Fig. 7 (pink open triangles labelled creep 1,2,3).
 473 Creep 1 is in the diffusion creep regime, creep 2 is transitioning into dislocation creep,
 474 and creep 3 is also transitioning but has the largest contribution from dislocation creep.
 475 Young’s modulus E and attenuation Q^{-1} measured after creep 1, 2, and 3 are shown in
 476 Figs. 10a, 10b, and 10c, respectively, where only the data at 20°C are plotted. Pink cir-
 477 cular symbols show the data of the first 20°C testing after each creep (solid pink and open
 478 pink circles in 10b and 10c show, respectively, the first and last data of the repeated mea-
 479 surements), and black square symbols show the data of the final 20°C testing. Also shown
 480 in Figure 10c are the data of the 2nd (light blue diamond), 3rd (blue triangle), and 4th
 481 (gray reverse triangle) 20°C testings showing an evolution with time. Figure 10 clearly
 482 shows that the Young’s modulus E was significantly reduced and attenuation Q^{-1} was
 483 slightly increased after deformation by dislocation creep, and that the magnitude of these
 484 changes systematically depends on the deviatoric stress amplitude of the prior deforma-

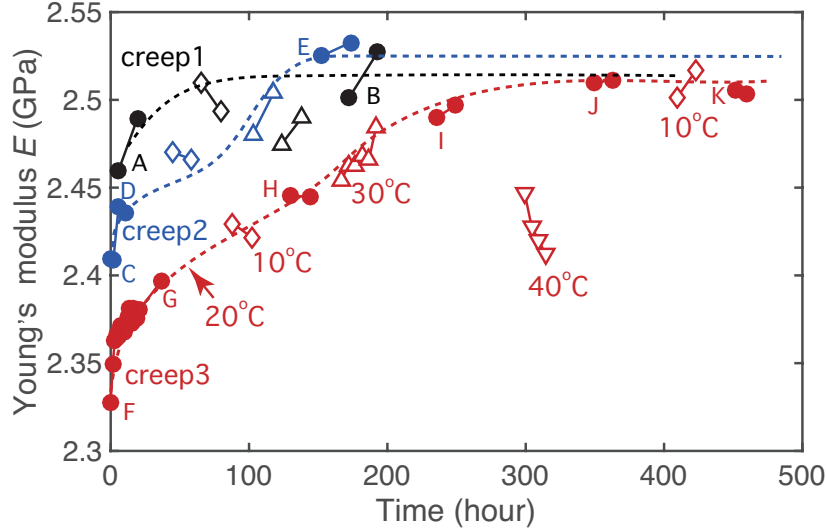
485 tion. Moreover, modulus E gradually increased and attenuation Q^{-1} gradually decreased
 486 during the repeated anelasticity measurements, and finally recovered to the values ob-
 487 tained after creep 1, that is, the values before dislocation creep. These final data (black
 488 square symbols in Figs 10a, b, and c) agree well with the previous data of a high-purity
 489 borneol sample with similar grain size (yellow cross symbols in Fig. 10).



490 **Figure 10.** Young's modulus and attenuation versus frequency of sample #122 measured
 491 after (a) creep 1 in the diffusion creep regime, (b) creep 2 with some contribution from dislo-
 492 cation creep, and (c) creep 3 with the largest contribution from dislocation creep (pink open
 493 triangles in Fig. 7). All data shown in these figures were obtained at 20°C. Alphabetic data set
 494 labels indicate the timing of the measurement described in Figure 4a. Pink circular and black
 495 square symbols show the data in the first and the last 20°C testing, respectively, after each creep.
 496 Yellow cross symbols show the previous data by Takei et al. (2014) (sample #36, $d = 15.9 \mu\text{m}$).

497 Figure 11 plots the time series data showing the recovery of Young's modulus E .
 498 All data points in this figure show the average E over the highest 8 frequencies (from
 499 68.1 to 1 Hz), so that the effect of data scattering is removed. In addition to the data
 500 at 20°C (solid circle), those at 10°C (open diamond), 30°C (open triangle), and 40°C
 501 (open reverse triangle) are shown. Because modulus data tend to decrease with increas-
 502 ing temperature (mostly due to the anelastic effect) (e.g., McCarthy et al., 2011; Takei
 503 et al., 2014), by using the 10°C symbols as an upper bound and 30–40°C symbols as a

504 lower bound, we estimated the recovery curves connecting the modulus at 20°C (dashed
 505 lines). The recovery curves after creep 2 and creep 3 clearly show that the recovery slows
 506 down at lower temperature (10°C) and is enhanced at higher temperature (30°C).



507 **Figure 11.** Recovery of Young's modulus during the repeated anelasticity measurements after
 508 creep 1 (black), creep 2 (blue), and creep 3 (red) performed on sample #122 is shown as time
 509 series data. Each symbol shows the average of the modulus over the same 8 frequencies from
 510 68.1 to 1 Hz. Solid circles show the data at 20°C, open diamonds 10°C, open triangles 30°C, and
 511 open reverse triangles 40°C. Letters are equal to those in Figure 4a. Dashed lines are schematic
 512 recovery curves showing E at 20°C versus time (see text for detail).

513 The present results show that the deformation under high differential stress ($\Delta\sigma >$
 514 1 MPa) reduced the Young's modulus and slightly increased the attenuation, but these
 515 effects mostly disappeared over the course of the anelasticity measurements performed
 516 under the small differential stress ($\Delta\sigma = 0.28$ MPa). Although Young's modulus gen-
 517 erally depends on both bulk and shear moduli, because of the large Poisson's ratio of bor-
 518 neol ($\nu = 0.371$, Takei, 2000), it is quite insensitive to the bulk modulus (McCarthy
 519 et al., 2011). Therefore, the observed change in the Young's modulus can be attributed
 520 to the change in the shear modulus.

521 The previous data (yellow cross symbols in Fig. 10) follow the Maxwell frequency
 522 scaling and hence can be associated with diffusionally-accommodated grain boundary
 523 sliding (McCarthy et al., 2011; Takei et al., 2014). The deviation from this grain boundary-

524 induced anelasticity in the pre-deformed samples gives the dislocation creep-induced anelas-
 525 ticity, which appeared largely in the modulus and only slightly in the attenuation spec-
 526 tra. Due to the relationship between modulus dispersion and attenuation (e.g., equations
 527 11 and 12 in Takei et al., 2014), this result indicates that the characteristic frequency
 528 of the dislocation creep-induced anelasticity is higher than 68.1 Hz (the highest frequency
 529 of the forced oscillation data), and hence much higher than the major part of the dis-
 530 sipation band caused by the grain boundary sliding. Further constraint on the charac-
 531 teristic frequency is given by the ultrasonic data. As shown in Fig. 8d, sample #122 af-
 532 ter all mechanical tests had wavy grain boundaries. Because the anelasticity data ob-
 533 tained after the recovery agreed well with those after creep 1 and also those in the pre-
 534 vious studies, we can say that the small deviation from the equilibrium microstructure
 535 (wavy grain boundaries) does not significantly affect grain boundary sliding.

536 In Figs. 10a and 10b, we observe a slight increase in attenuation Q^{-1} with increas-
 537 ing frequency above 10 Hz, which is not observed in the Q^{-1} data in Fig 10c measured
 538 after releasing the vacuum of the plastic bag. Because the Q^{-1} spectra in Fig 10c agree
 539 well with the previous data without a plastic bag (yellow cross symbols), the slight in-
 540 crease in Q^{-1} above 10 Hz can be considered an effect of the vacuum sealed plastic bag.

541 In the forced oscillation apparatus, diffusion creep viscosity was measured just be-
 542 fore each change in the run temperature (Fig. 4a). (Viscosity was not measured at 10°C.)
 543 The effect of prior deformation by dislocation creep on the diffusion creep viscosity was
 544 not visible, and the obtained viscosity data were consistent with the previous result with-
 545 out dislocations (supporting information Fig. S2). This is consistent with the above men-
 546 tioned observation that the effect of deformation by dislocation creep on the low frequency
 547 attenuation is very small.

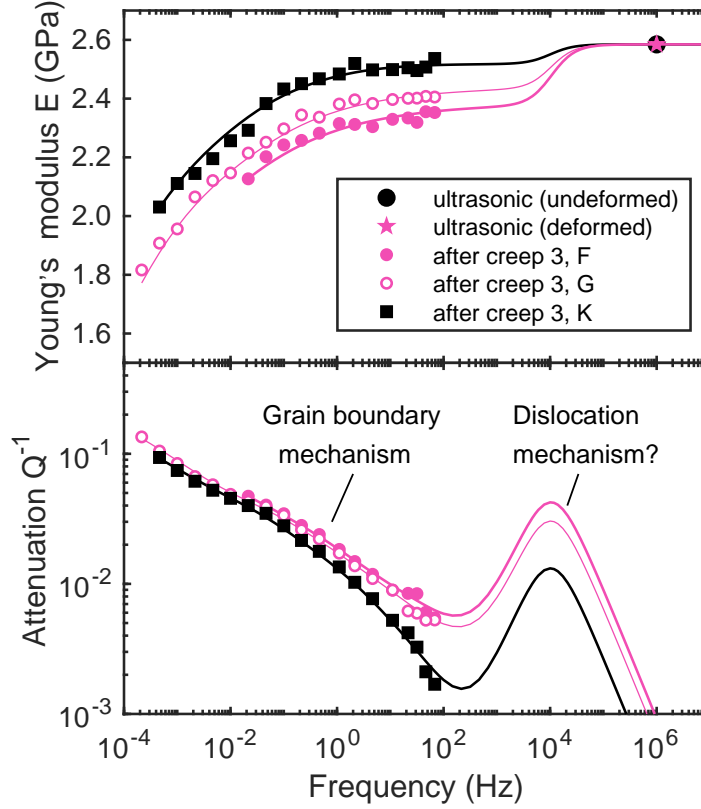
548 **3.2.2 Result of the ultrasonic test**

549 The stress and strain rate conditions during the deformation of samples #123 and
 550 #127 were plotted in Fig. 7 (pink open diamonds and squares). The conditions of the
 551 last, highest stress creep on these samples were similar to that in creep 3 of #122 and
 552 hence dislocation-creep effects similar to those on #122 by creep 3 can be expected. Typ-
 553 ical ultrasonic wave forms obtained before and after the deformation are shown in sup-
 554 porting information Fig. S3. For sample #123 (which was deformed to $\varepsilon_{\text{fin}} = 0.21$ at

555 $\Delta\sigma \leq 2.1$ MPa), we obtained $V_P = 2.222$ km/s, $V_S = 0.963$ km/s, and hence $E =$
 556 2.585 GPa before the deformation, and $V_P = 2.223$ km/s, $V_S = 0.963$ km/s, and hence
 557 $E = 2.585$ GPa after the deformation. For #127 (which was deformed to $\varepsilon_{\text{fin}} = 0.13$
 558 at $\Delta\sigma \leq 2.1$ MPa), we obtained $V_P = 2.217$ km/s, $V_S = 0.965$ km/s, and $E = 2.596$
 559 GPa before the deformation, and $V_P = 2.214$ km/s, $V_S = 0.963$ km/s, and $E = 2.588$
 560 GPa after the deformation. By considering the errors of $\pm 0.15\%$ in V_P and $\pm 0.2\%$ in V_S
 561 (Takei et al., 2014), for both samples, Young’s modulus measured at ultrasonic frequency
 562 (1 MHz) did not show any significant change after dislocation creep. The waveforms did
 563 not show any visible increase in attenuation at ~ 1 MHz (supporting information Fig. S3).
 564 These results show that the major part of the dislocation creep-induced anelastic relax-
 565 ation exists at frequencies lower than 1 MHz.

566 A possible distribution of the dislocation creep-induced anelasticity, explaining both
 567 ultrasonic and forced oscillation data is shown in Fig. 12 (solid curves). Each set of curves
 568 (E and Q^{-1}) was calculated from a relaxation spectrum $X(\tau)$ (Nowick & Berry, 1972)
 569 estimated as follows. First, from the modulus and attenuation data obtained by the forced
 570 oscillation test, we estimated the relaxation spectrum $X(\tau)$ by using equation (9) in Takei
 571 et al. (2014). Unrelaxed modulus was given by the ultrasonic result ($E_U = 2.585$ GPa),
 572 which agrees well with a previous value ($E = 2.600$ GPa, Takei et al., 2014). In this
 573 study, unlike Takei et al. (2014), a line spectrum was added to $X(\tau)$ at $\tau = (2\pi \times 10^4)^{-1}$ s,
 574 in order to make up the modulus deficit between $f = 10^6$ Hz and $f \leq 68.1$ Hz. There-
 575 fore, the calculated Q^{-1} spectrum includes a Debye peak with the center frequency at
 576 $f = 10^4$ Hz. The assumption of a line spectrum was used for mathematical simplicity
 577 and is modified below. Fig. 12 demonstrates that major part of the dislocation creep-
 578 induced anelastic relaxation forms a peak at much higher frequency than the major part
 579 of high-temperature background, which is caused by diffusionally-accommodated grain
 580 boundary sliding.

588 Figure 12 also shows that the Debye peak is gradually reduced by annealing, but
 589 some part of it ($\sim 30\%$) remains even after the recovery (black peak). This result might
 590 mean that once deformed in dislocation creep, the borneol samples always contain some
 591 dislocations that cannot be removed by high temperature annealing. However, because
 592 a small modulus deficit similar to that of data “K” ($\sim 3\%$), which is covered by the black
 593 peak in Fig. 12, exists even without dislocation creep (Fig. 17a in Takei et al., 2014),
 594 it might be alternatively explained by elastically accommodated grain boundary sliding



581 **Figure 12.** A possible distribution of the dislocation creep-induced anelasticity obtained from
 582 a combined analysis of both ultrasonic and forced oscillation data. Major part of the dislocation
 583 creep-induced dissipation is a peak at much higher frequency than the high-temperature back-
 584 ground that is dominant at $f \leq 100$ Hz, which is attributable to grain boundary sliding. Symbols
 585 show the experimental data ($T = 20^\circ\text{C}$, homologous temperature $T/T_m = 0.61$, and Maxwell
 586 frequency $f_M \simeq 1.3 \times 10^{-5}$ Hz). Each set of solid curves (E and Q^{-1}) was calculated from an
 587 appropriate relaxation spectrum.

595 and/or a systematic error of the modulus measured by the forced oscillation test (Takei
 596 et al., 2014).

597 The Debye peak in Fig. 12 resulted from the line spectrum assumption of $X(\tau)$ in-
 598 troduced for mathematical simplicity. However, as shown in supporting information Fig. S4,
 599 even if a broader distribution of relaxation time is assumed, the anelasticity induced by
 600 dislocation creep is a peak at much higher frequency than the major part of the high-
 601 temperature background. Therefore, the major conclusion of this section does not de-
 602 pend on the assumed form of $X(\tau)$.

603 4 Discussion

604 4.1 Dislocation creep

605 The newly obtained creep data show that the dominant deformation mechanism
 606 of high-purity borneol starts to transition from diffusion to dislocation creep at $\Delta\sigma \simeq$
 607 1 MPa for $T = 40\text{--}50^\circ\text{C}$. At high stress around $\Delta\sigma \simeq 2$ MPa, the initial hardening
 608 stage was followed by a strain weakening stage. Based on the sensitivity to the effective
 609 confining pressure, and also based on the results of microstructural observations, we at-
 610 tributed the strain weakening stage to a growth of microcracks and/or pores, particu-
 611 larly near the outer cylindrical surface of the sample. Therefore, the data during and af-
 612 ter the strain weakening stage were not used in determining the steady-state flow law
 613 provided in Fig. 7 and Table 2. The stress exponent of steady-state dislocation creep was
 614 estimated as $n \simeq 4$. The microstructure of the deformed samples showed wavy grain
 615 boundaries, indicating the occurrence of dislocation-induced grain boundary migration.
 616 Here, we compare the present results with transient and steady-state creep of olivine in
 617 the dislocation creep regime, experimentally studied in single crystals (Cooper et al., 2016;
 618 Hanson & Spetzler, 1994) and polycrystals (Chopra, 1975; Hansen et al., 2012).

619 Similar to the present result, grain boundary migration indicated by serrated grain
 620 boundaries is generally reported for olivine polycrystal (e.g., Mei & Kohlstedt, 2000).
 621 The observation of inverse transients, or strain weakening, in the dislocation creep of olivine
 622 is limited but to a few cases: deformation of single crystals with constant axial load in
 623 the direction of $[110]_c$ (Hanson & Spetzler, 1994) and with constant axial stress in the
 624 direction of $[011]_c$ (Cooper et al., 2016). Even in these cases, however, strain weaken-
 625 ing did not exceed a few percent strain. Olivine polycrystals generally show strain hard-
 626 ening (Chopra, 1975). Only at large strain do olivine polycrystals show strain weaken-
 627 ing mostly due to a grain-size evolution by recrystallization (Hansen et al., 2012). How-
 628 ever, grain-size evolution was not observed in the present study. Therefore, with the present
 629 conclusion that the strain weakening process observed in our study was not intrinsic (*i.e.*,
 630 was due to microcrack and/or pore formation), we can remove the extrinsic weakening
 631 and the resulting transient behavior of polycrystalline borneol is similar to that of poly-
 632 crystalline olivine. Furthermore, the stress exponent measured during steady-state dis-
 633 location creep of olivine single crystals and polycrystals is obtained as 3-4 (eg., Cooper

et al., 2016; Hansen et al., 2012; Mei & Kohlstedt, 2000), which is also similar to that of polycrystalline borneol obtained in this study.

4.2 Dislocation creep-induced anelasticity

As shown in Fig. 12, following deformation by dislocation creep, a large peak in the attenuation spectra is presumed to exist at high frequency, which gradually decreased during the anelasticity measurements under low differential stress ($\Delta\sigma \simeq 0.28$ MPa). We first compare the present anelasticity results with that obtained for the same material (borneol polycrystals) in which the grain-boundary mechanism was isolated. Anelastic relaxation by grain boundary sliding at various temperatures, grain sizes, and chemical compositions can be characterized by two nondimensional parameters; normalized frequency f/f_M and homologous temperature T/T_m , where $f_M = E_U/\eta$ represents the Maxwell frequency calculated by unrelaxed Young's modulus E_U and diffusion creep viscosity η , and T_m represents the melting temperature or solidus (Takei, 2017). The data in Fig. 12 (sample #122 at $T = 20^\circ\text{C}$) have $T/T_m = 0.61$ and $f_M \simeq 1.3 \times 10^{-5}$ Hz, where we used $\eta = 2.0 \times 10^{14}$ Pas measured in the forced oscillation apparatus at 20°C (supporting information Fig. S2). Therefore, using this same Maxwell frequency normalization on the data in Fig. 12, the peak at $f = 10^4$ Hz is $f/f_M \simeq 10^9$. A large and broad high-frequency peak caused by a grain boundary mechanism has been previously captured in borneol (+ diphenylamine) at near solidus temperatures $T/T_m > 0.9$, which was attributed to grain boundary premelting (Takei, 2017; Yamauchi & Takei, 2016). However, when $T/T_m = 0.61$, grain boundary effects mostly exist at $f/f_M < 10^5$ as high-temperature background (e.g., Fig. 9d in Takei et al., 2014), which corresponds to $f < 1$ Hz in Fig. 12. Therefore, dislocation creep-induced anelasticity obtained in this study has a much shorter relaxation time scale than that of grain boundary effects at this low homologous temperature.

As shown in section 3.1.2, dislocation creep in this study was not completely free from microcracking. For mantle rocks, the empirical condition for fully plastic deformation without any contribution from microcracks is known as $\sigma_3 \geq \sigma_1 - \sigma_3$, where σ_1 and σ_3 represent, respectively, the largest and smallest compressive principal stress (e.g., Evans et al., 1990). This condition, called the Goetze criterion, was not satisfied in this study ($\sigma_1 - \sigma_3 \simeq 2$ MPa, and $\sigma_3 = 0.8$ MPa). Higher confining pressure is needed to completely suppress microcracking. Although the emergence of the nonlinear creep and

666 the wavy grain boundaries indicate the presence of dislocations, the obtained change in
667 the anelastic properties cannot be solely attributed to the dislocations. If we estimate
668 the effect of microcracks on Young's modulus by using the observed crack density pa-
669 rameter ($c = 0.002 - 0.004$) and poroelastic theory (O'Connell & Budiansky, 1974),
670 the modulus reduction is estimated as 0.15–0.3% for closed cracks and 0.33–0.66% for
671 open cracks. (Calculations are given in the supporting information). Although the es-
672 timated reduction by microcracks is much smaller than the observed modulus change,
673 this estimation based on the visible microcracks can be an underestimation. We cannot
674 deny a possible existence of grain boundary cracks which were completely closed and in-
675 visible after the removal of stress, but still affect the grain boundary sliding. The observed
676 recovery of the anelastic properties by annealing can be also explained by both disloca-
677 tion recovery and crack healing. Indeed, multiple micropores may represent a healed crack,
678 and would be connected under high stress. Therefore, it is difficult to separate the con-
679 tributions from dislocations and microcracks. Further experiments under higher confin-
680 ing pressure are needed in order to assess the contribution from microcracks.

681 In addition, the removal of the deviatoric and confining stress and the slow quench
682 performed before the anelasticity measurements might have inflicted some damage to the
683 sample and/or sample-platen contact. In order to address this issue in future studies,
684 we developed a new experimental system in which dislocation creep under confining pres-
685 sure (0.8 MPa) and anelasticity measurement can be performed simultaneously. Prelim-
686 inary results obtained recently using this new system show that Young's modulus (mea-
687 sured at 2 Hz) decreases gradually during dislocation creep. Although details will be re-
688 ported elsewhere, this preliminary observation shows that the reduction of the Young's
689 modulus after dislocation creep is not due to damage during unloading nor quench. Such
690 in situ experiments will also enable us to investigate a possible effect of strain on anelas-
691 ticity, which was not examined in this present study.

692 **4.3 Comparison with dislocation damping**

693 Although the dislocation creep-induced anelasticity obtained in this study cannot
694 be attributed solely to a dislocation mechanism, but might be affected by microcracks,
695 it is, nonetheless, meaningful to compare this result to the dislocation damping observed
696 in previous studies and examine whether the present anelasticity result can be under-
697 stood as a dislocation mechanism.

698 We compare the present result with the previous knowledge on the dislocation-induced
699 anelasticity in metallic crystals (e.g., Blanter et al., 2007; Nowick & Berry, 1972). Re-
700 laxation timescale of the dislocation effects in metals extends over a vast frequency range
701 depending on the mechanism controlling the mobility of dislocation, as schematically il-
702 lustrated in Fig.3.3.30 in Gremaud (2001). When pinned dislocation segments bow by
703 glide without any interaction with other defects, the relaxation timescale is estimated
704 as $B_v l^2 / 12\zeta$ with segment length l (m), dislocation energy ζ (J/m), and viscous drag co-
705 efficient B_v (Ns/m²), which is caused by the interaction of the moving dislocation with
706 phonons (e.g., Blanter et al., 2007; Nowick & Berry, 1972). In metals, a dissipation peak
707 caused by this mechanism exists at high frequencies (>10 MHz Gremaud, 2001). In con-
708 trast, when the dislocation glide is rate-controlled by drag of point defects by diffusion,
709 anelastic relaxation occurs at much lower frequencies (Gremaud, 2001). Because bow-
710 ing of the dislocation segments without interaction with other defects does not involve
711 diffusion (e.g., Kocks et al., 1975), a similar mechanism, if active in the borneol crystal,
712 explains well the presence of the large peak at much higher frequency than the high tem-
713 perature background caused by the diffusion-controlled grain boundary sliding. Unlike
714 metallic crystals, however, the peak did not reach the ultrasonic regime (~ 1 MHz). The
715 above mentioned model suggests that borneol has larger B_v , larger l , and/or smaller ζ
716 than metals, but so far it is difficult to determine the reason for this discrepancy. In metal-
717 lic crystals, the frequency and width of the peak(s) are known to depend on the crystal
718 structure and height of Peierls potential (Karato & Spetzler, 1990; Nowick & Berry,
719 1972). With low Peierls potential, a simple string model works (e.g., Granato & Lücke,
720 1956), but with high Peierls potential, kink motion has to be taken into account (e.g.,
721 Karato, 1998). Therefore, detailed knowledge of the borneol crystal will facilitate the un-
722 derstanding of the present result. To our knowledge, the crystal structure and Peierls
723 potential of borneol (molecular crystal) have not been reported, although the high tem-
724 perature phase of borneol ($\geq 72^\circ\text{C}$ at ambient pressure) is known as f.c.c. (Sherwood,
725 1979).

726 We also compare the present result with the previous results of dislocation-induced
727 anelasticity in olivine. The major results of Guéguen et al. (1989) and Farla et al. (2012)
728 are summarized in Table 3. Guéguen et al. (1989) first reported a significant enhance-
729 ment of attenuation in pre-deformed forsterite single crystals. They reported that the
730 enhancement is more significant at lower frequencies and/or higher temperatures. Sim-

Table 3. Summary of dislocation-induced anelasticity

	This study ^a	Farla et al. (2012)	Guéguen et al. (1989)	McCarthy & Cooper (2016)
sample ^b	borneol PC	olivine PC	forsterite SC	ice PC
f dependence	peak	broad absorption band	$Q^{-1} \propto f^{-0.21}$	broad absorption band
ρ dependence	yes	$Q^{-1} = 0.024 \times \rho^c$	yes	–
T dependence	–	yes	$Q^{-1} \propto e^{-H/R_g T^d}$	yes
nonlinearity	–	no compelling evidence	(probably)	yes
anisotropy	–	yes	–	–

^aContribution from microcracks might be included.

^bPC=polycrystals, SC=single crystals

^c ρ is dislocation density in $(\mu\text{m})^{-2}$. Result of a 101-s period for compressively pre-deformed sample.

^d $H = 440$ kJ/mol

731 ilar results were also reported by Farla et al. (2012). The dislocation-induced anelas-
732 ticity captured by Farla et al. (2012) shows a frequency dependence similar to the high-
733 temperature background caused by grain boundary sliding. Also, the deviation from the
734 grain boundary effect was significant at high temperatures ($\geq 1000^\circ\text{C}$). A similar re-
735 sult was found in an anelasticity study of polycrystalline ice that was actively deform-
736 ing under high stress (McCarthy & Cooper, 2016). In this study, we could constrain well
737 the frequency dependence of the dislocation creep-induced anelasticity. Unlike these pre-
738 vious studies, the obtained attenuation spectrum is a peak with a characteristic frequency
739 much higher than those of the high-temperature background (Fig. 12). Such a peak has
740 not been reported in these previous studies. However, the possible existence of a peak
741 at much higher frequency than the testing frequencies cannot be denied in these stud-
742 ies, because of a relatively large uncertainty in absolute value of the modulus measured
743 from the forced oscillation tests; modulus data were not shown in Guéguen et al. (1989),
744 were not analyzed quantitatively in McCarthy and Cooper (2016), and were discussed
745 only tentatively in Farla et al. (2012). Because all relaxations existing at higher frequen-
746 cies than seismic waves affect the wave velocity, the present approach using a rock ana-
747 logue, for which Young’s modulus can be measured accurately, can play a complemen-
748 tary role for the understanding of rock anelasticity.

5 Conclusions

Creep tests of polycrystalline borneol were conducted at $T = 40\text{--}50^\circ\text{C}$ with differential stress up to $\Delta\sigma \simeq 2$ MPa and confining pressure 0.8 MPa. The dominant deformation mechanism changes from diffusion to dislocation creep at $\Delta\sigma \simeq 2$ MPa. The stress exponent of steady-state dislocation creep was estimated as ~ 4 . Wavy grain boundaries were observed throughout the majority of the deformed samples, indicating the occurrence of dislocation-induced grain boundary migration, whereas microcracks were occasionally found, particularly near the outer cylindrical surface.

Effects of dislocation creep on anelasticity were captured by the forced oscillation test at $f < 100$ Hz as a significant reduction in Young's modulus and a slight increase in attenuation. In contrast, the Young's modulus measured at the ultrasonic frequency ($f = 1$ MHz) did not change. Therefore, a major part of the dislocation creep-induced anelastic relaxation is a peak with the characteristic frequency much higher than the major part of high-temperature background caused by grain-boundary sliding. These effects of prior high stress deformation mostly disappeared during the anelasticity measurements under low differential stress ($\Delta\sigma \simeq 0.28$ MPa). The recovery was thermally activated. Further experimental study under higher confining pressure is needed to assess the relative contribution from dislocations and microcracks to the observed changes in anelasticity.

A Correction factor for no-slip boundary conditions.

Let E_{app} be an apparent Young's modulus of a cylindrical sample (radius R and length L) measured by compressing the top and bottom surfaces with a rigid plane without slip. Here, sample geometry is represented by aspect ratio $\alpha = R/L$. Let E_{true} be the true Young's modulus. Takei et al. (2014) defined correction factor γ as $E_{\text{true}} = \gamma E_{\text{app}}$, where γ depends on the Poisson's ratio ν and aspect ratio α of the sample. The correction factor used in this present study is explicitly written as

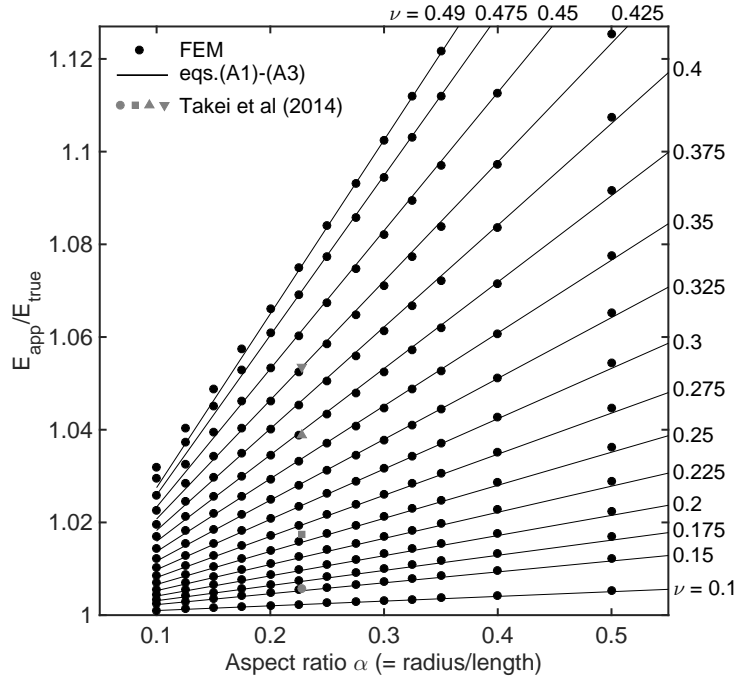
$$\frac{1}{\gamma} = a(\nu) \times \alpha + b(\nu), \quad (\text{A.1})$$

where $a(\nu)$ and $b(\nu)$ are given by

$$a(\nu) = 0.99314 \times \nu^2 + 2.3671 \times \nu^4 \quad (\text{A.2})$$

$$b(\nu) = 1 - 0.025144 \times \nu^3 - 0.50741 \times \nu^6. \quad (\text{A.3})$$

777 We obtained equations (A.1)-(A.3) as an approximation formula for the variation of $E_{\text{app}}/E_{\text{true}}$
 778 with α and ν numerically calculated by the finite element method described below (Fig. A.1).
 779 The difference between $E_{\text{app}}/E_{\text{true}}$ from FEM and from equations (A.1)-(A.3) is within
 780 ± 0.002 for a range of $0.175 \leq \alpha \leq 0.4$ and $0 \leq \nu \leq 0.49$, or for a range of $0.1 \leq$
 781 $\alpha \leq 0.5$ and $0 \leq \nu \leq 0.425$. In this study, α was between 0.23 and 0.3, and ν was be-
 782 tween 0.371 and 0.42 (the largest value of ν occurs at the highest temperature and low-
 783 est frequency). The γ^{-1} accurately determined in this study is slightly (< 0.008) larger
 784 than the previous one used by Takei et al. (2014) (gray symbols in Fig. A.1).



785 **Figure A.1.** Apparent Young's modulus E_{app} divided by the true Young's modulus E_{true}
 786 versus sample aspect ratio α (= radius R / length L) for various Poisson's ratio ν . Black symbols
 787 show the FEM results, black lines show the approximation formula (A.1)-(A.3), and gray symbols
 788 at $\alpha = 0.227$ ($L/2R = 2.2$) show the factors used by Takei et al. (2014) for $\nu = 0.2$ (circle), 0.3
 789 (square), 0.4 (triangle), and 0.45 (reverse triangle).

790 The finite element method was used to solve the axisymmetric equations of linear
 791 elasticity in a cylinder of radius R and height L . Owing to mirror symmetry, it is suf-
 792 ficient to solve the equations just for the top half of the cylinder. The boundary condi-

793 tions are

$$\sigma_{rr}, \sigma_{rz} = 0 \quad \text{on } r = R, \quad (\text{A.4})$$

$$u_r = 0 \quad \text{on } z = L/2, \quad (\text{A.5})$$

$$u_z = \epsilon_0 L/2 \quad \text{on } z = L/2, \quad (\text{A.6})$$

$$u_z, \sigma_{rz} = 0 \quad \text{on } z = 0, \quad (\text{A.7})$$

794 where ϵ_0 is the imposed strain, \mathbf{u} is the displacement, and σ is the stress tensor. Equa-
 795 tion A.4 imposes zero traction on the curved sides of the cylinder. Equations A.5 and
 796 A.6 prescribe the displacement at the ends of the cylinder, which is bonded such that
 797 there is zero radial displacement. Equation A.7 imposes mirror symmetry about the cylin-
 798 der midplane. The apparent Young's modulus can be determined from the volume av-
 799 erage of σ_{zz} ,

$$E_{\text{app}} = \frac{1}{\epsilon_0 V} \int_V \sigma_{zz} \, dV, \quad (\text{A.8})$$

800 where V is the volume. The governing equations were discretised using the FEniCS soft-
 801 ware (Logg & Wells, 2010; Logg et al., 2012). Cubic triangular Lagrange elements were
 802 used to represent the displacement \mathbf{u} . Mesh points were evenly spaced with N_r points
 803 in the radial direction and N_z in the vertical, where

$$N_r = \max(N, RN/L), \quad (\text{A.9})$$

$$N_z = \max(N/2, LN/2R), \quad (\text{A.10})$$

804 and $N = 128$, to ensure an equal spacing of grid points in both directions.

805 Acknowledgments

806 The authors would like to thank Ayako Suzuki for helpful discussions and experimen-
 807 tal supports, M. Uchida for technical assistance, and A. Yasuda for allowing us to ac-
 808 cess the microscope. L. Hansen and an anonymous referee provided thoughtful reviews,
 809 which greatly improved this paper. This work was supported by JSPS KAKENHI Grant
 810 Number JP15K13560. ERI International Office supported the visit of CM. Experimen-
 811 tal dataset of this study can be accessed at http://www.eri.u-tokyo.ac.jp/people/ytakei/Sasaki_etal_JGR2019/

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Figure 1.

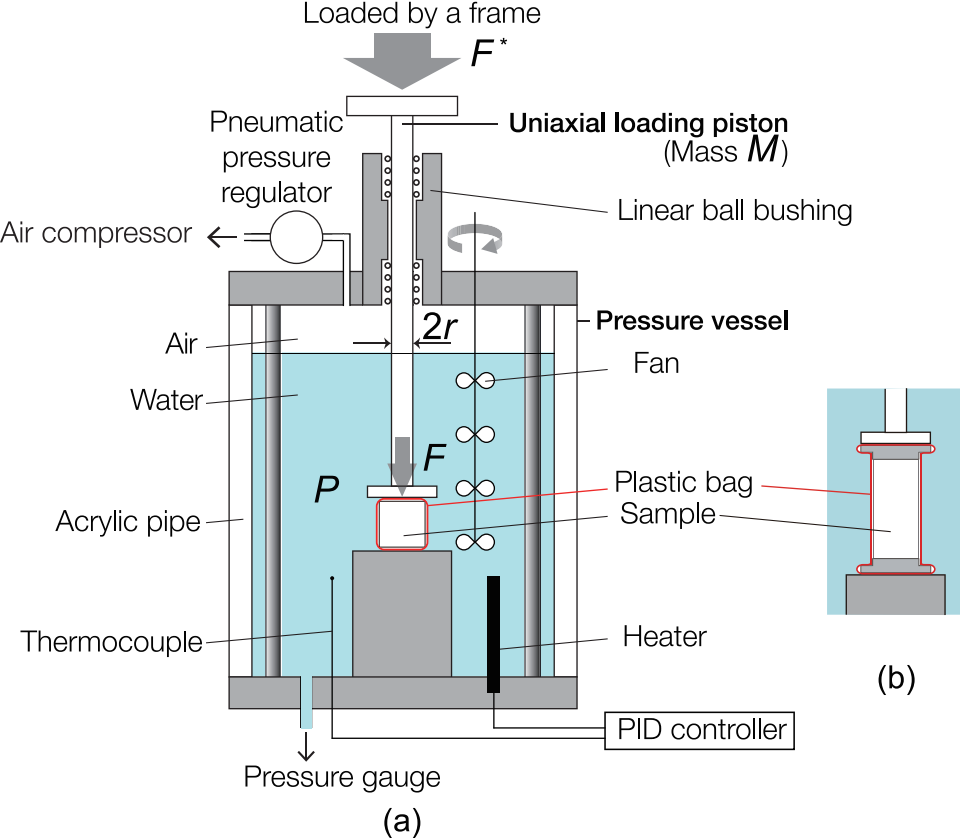


Figure 2.

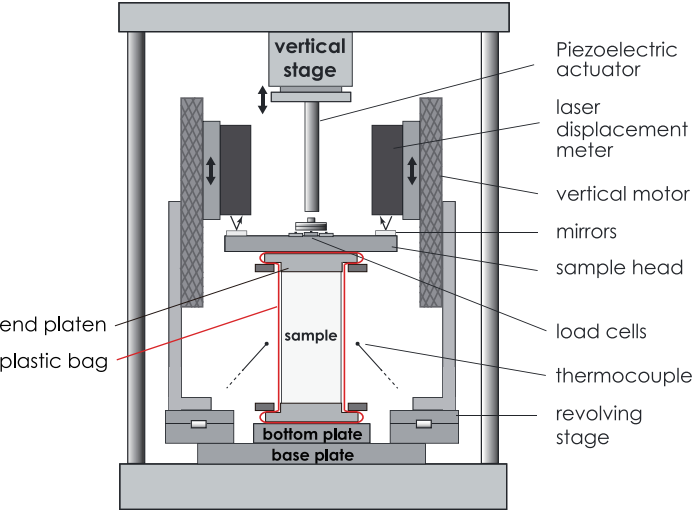


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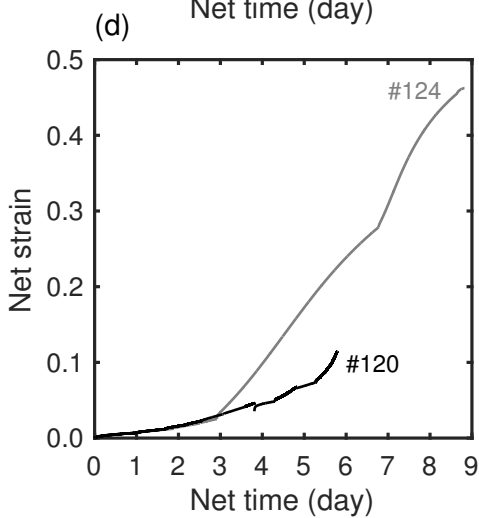
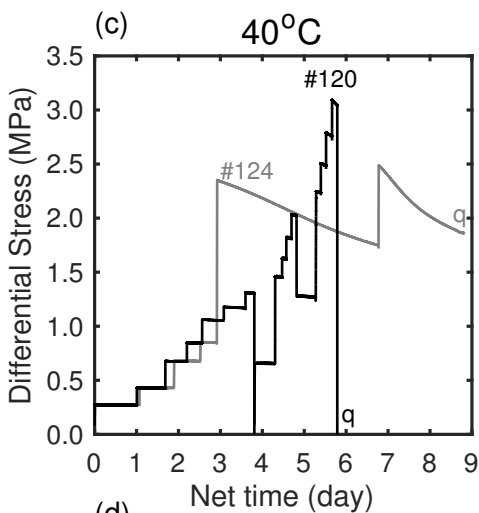
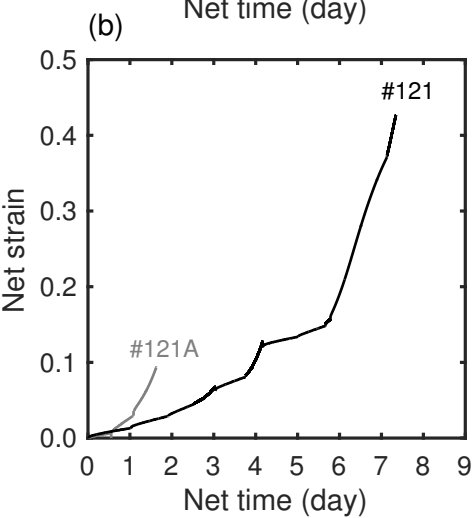
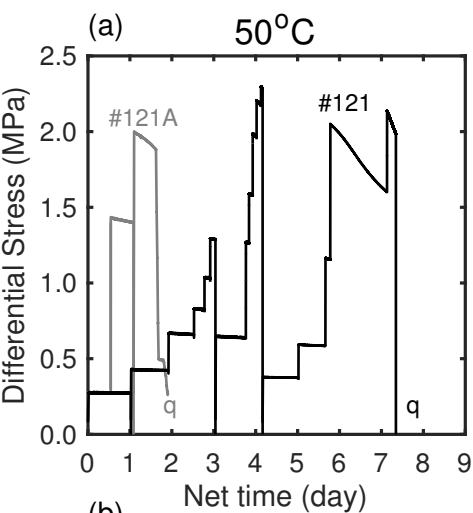


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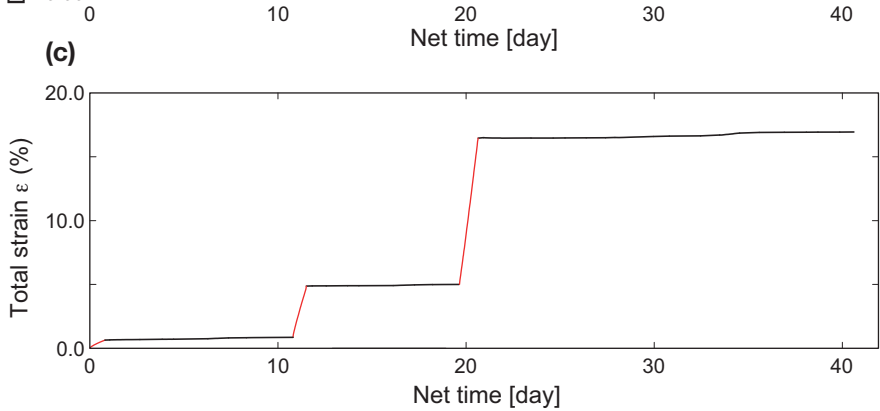
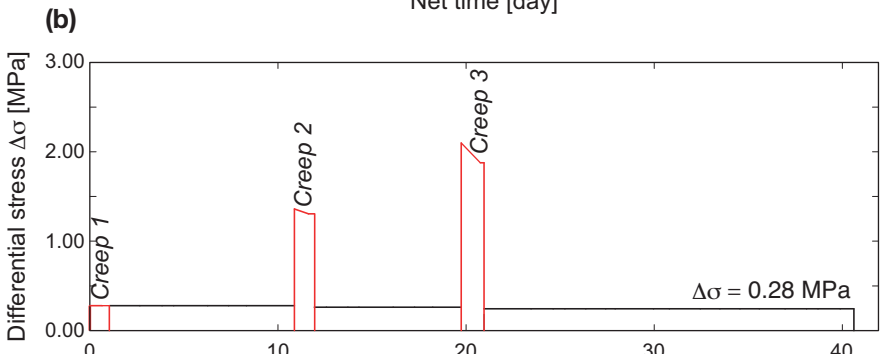
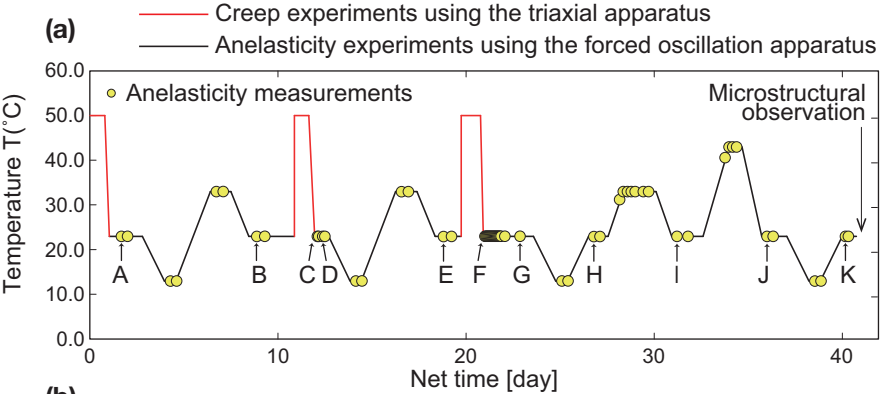


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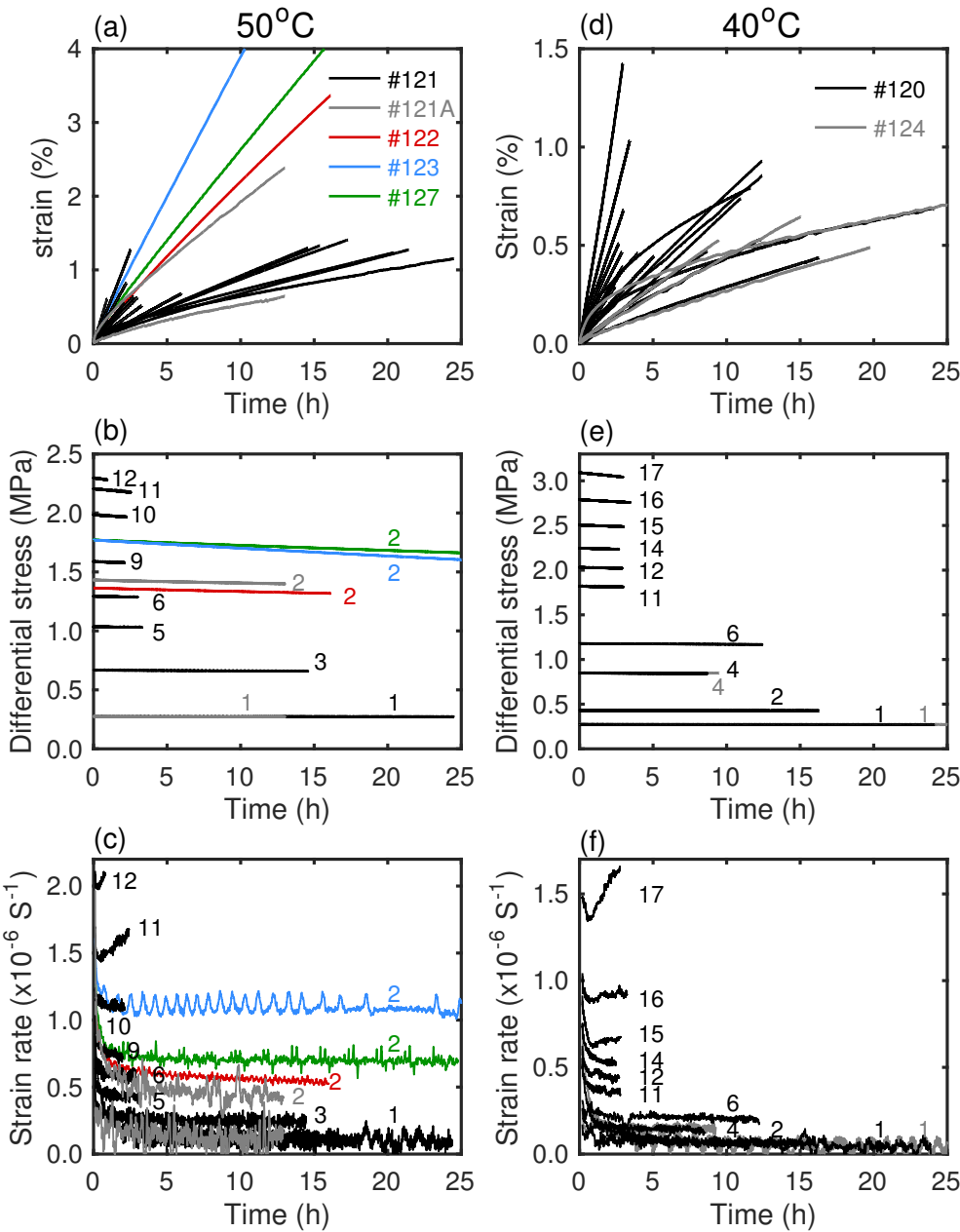


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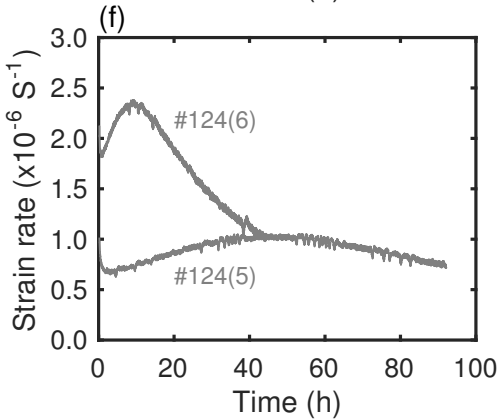
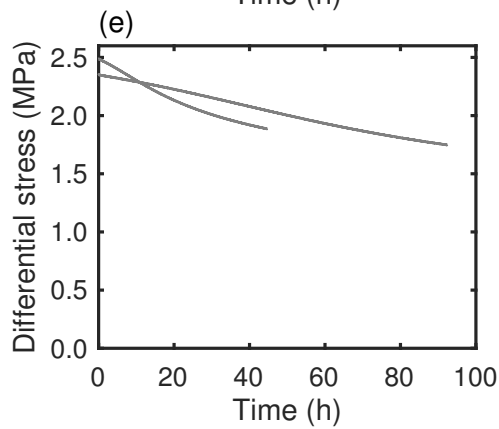
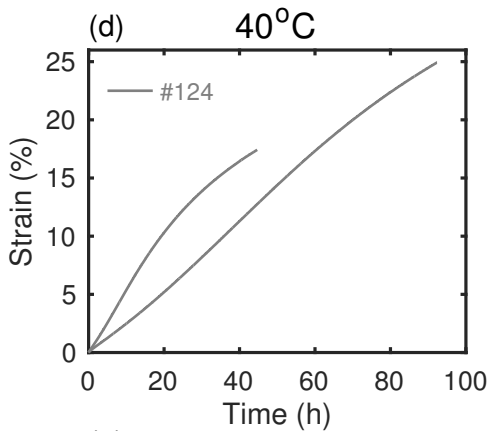
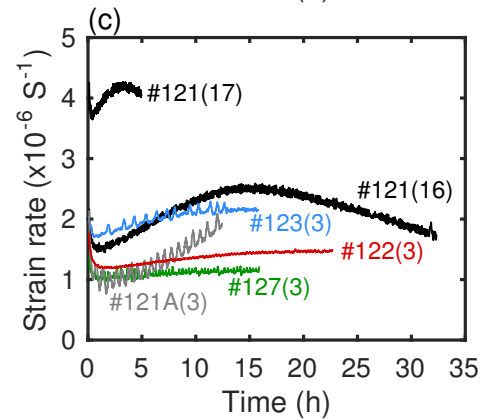
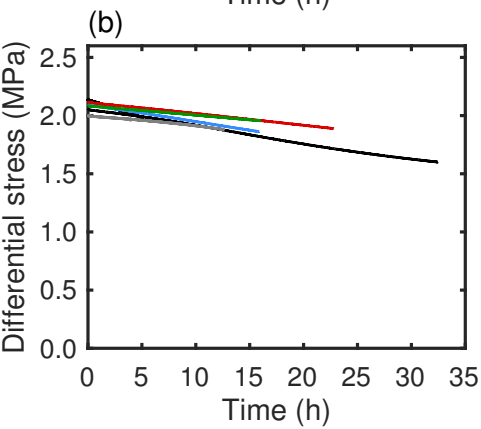
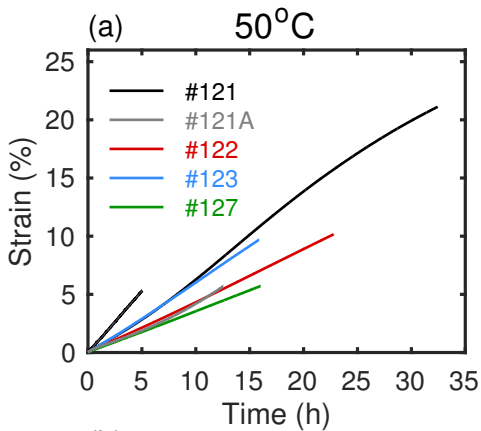


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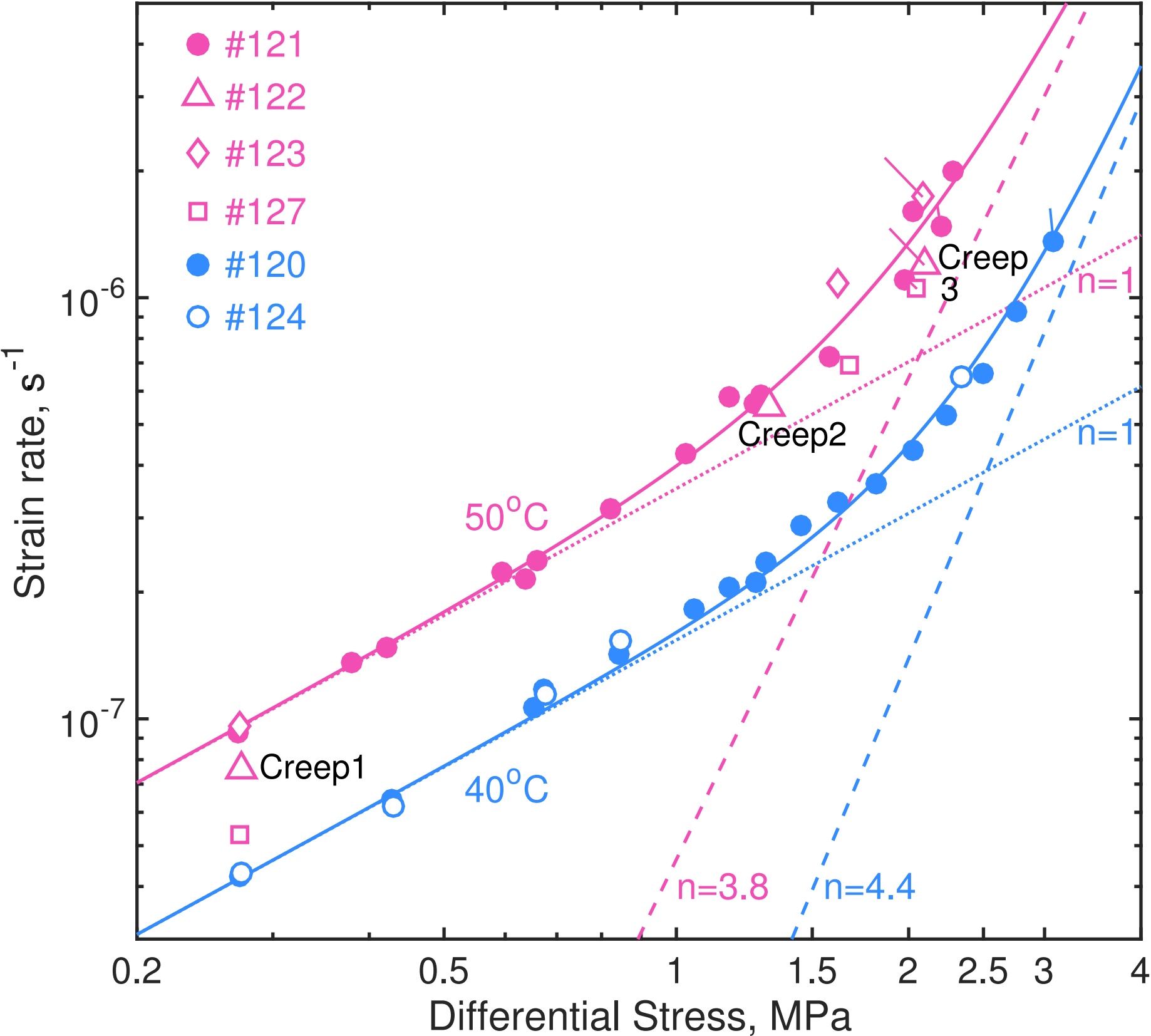
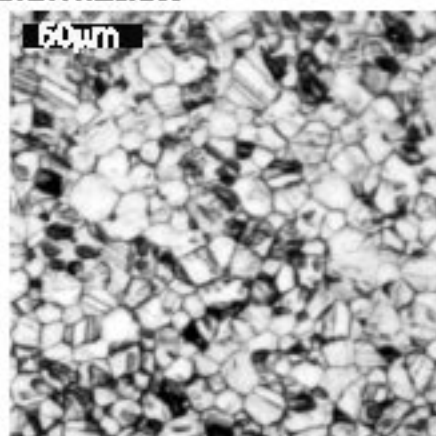
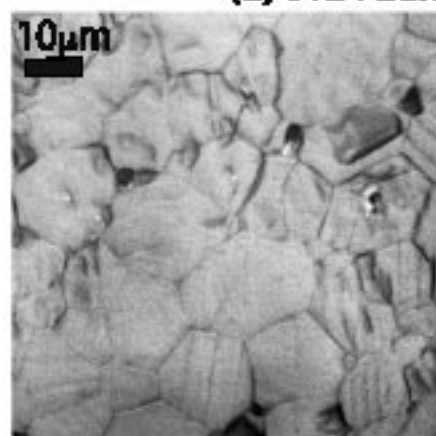
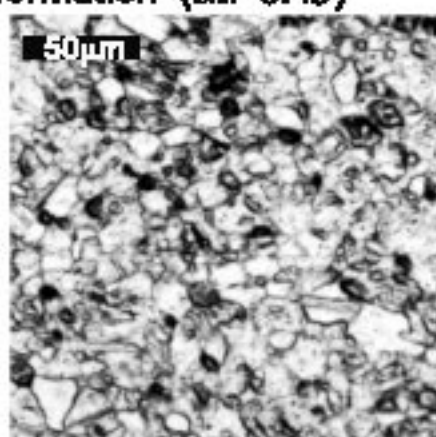
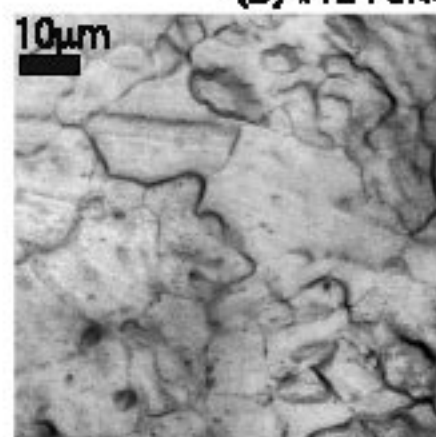


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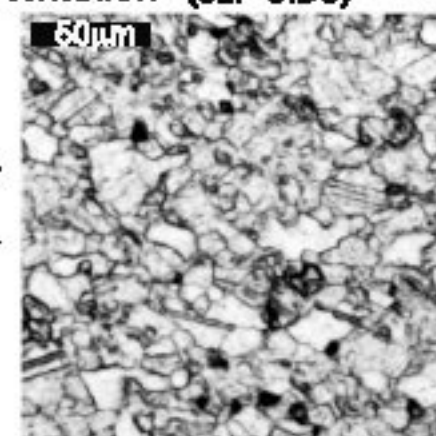
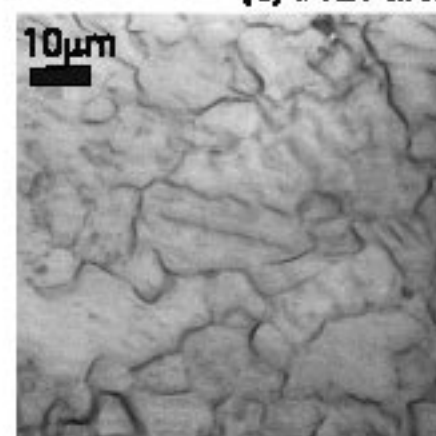
(a) #124 before deformation



(b) #124 after deformation ($\epsilon_{nr}=0.45$)



(c) #121 after deformation ($\epsilon_{nr}=0.36$)



(d) #122 after deformation ($\epsilon_{nr}=0.16$)

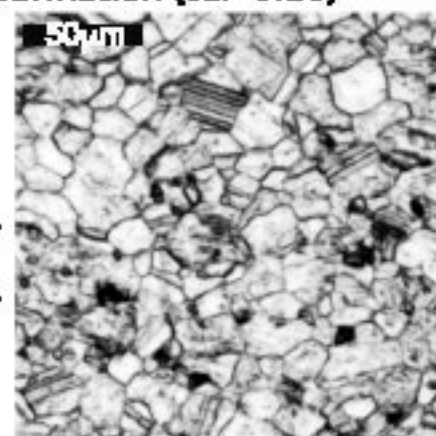
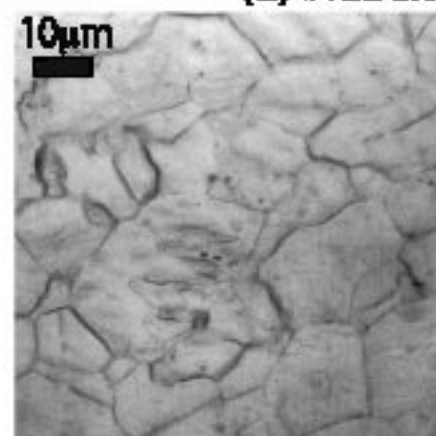
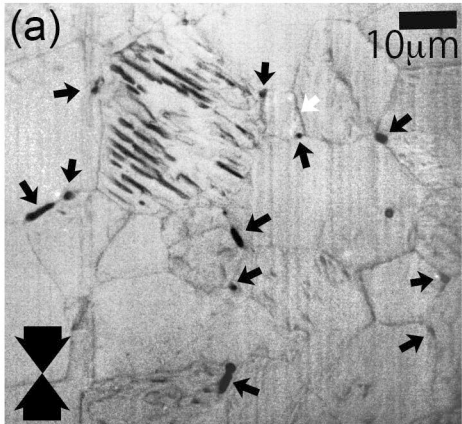


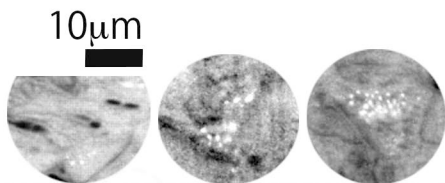
Figure 9.

#121A after deformation



(b) Micrograph (b) shows a small area of deformed material. A scale bar below indicates 10 μm.

(c) Micrograph (c) shows a small area of deformed material. A scale bar below indicates 10 μm.



#127 after deformation

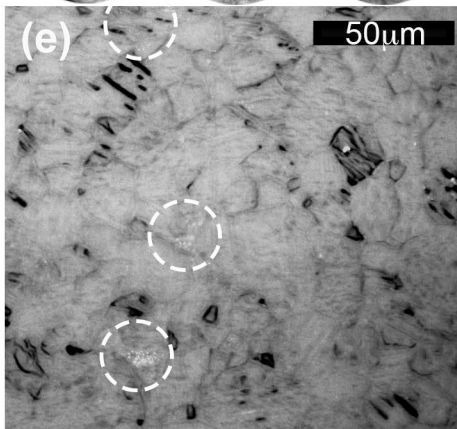
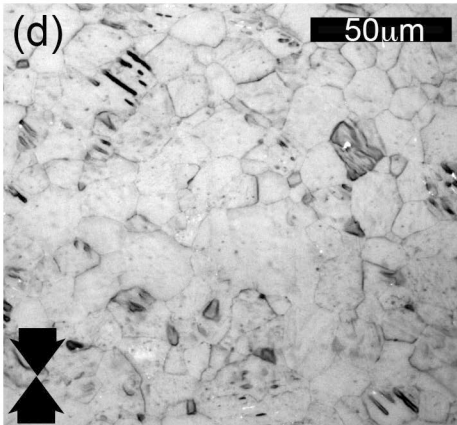


Figure 10.

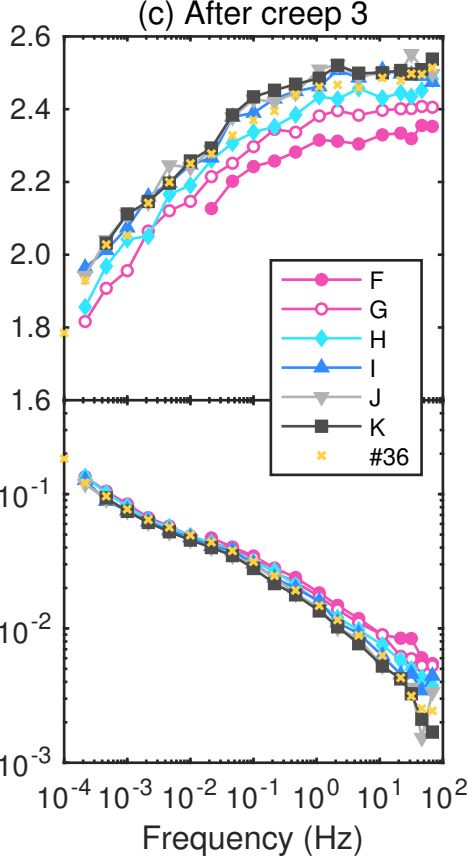
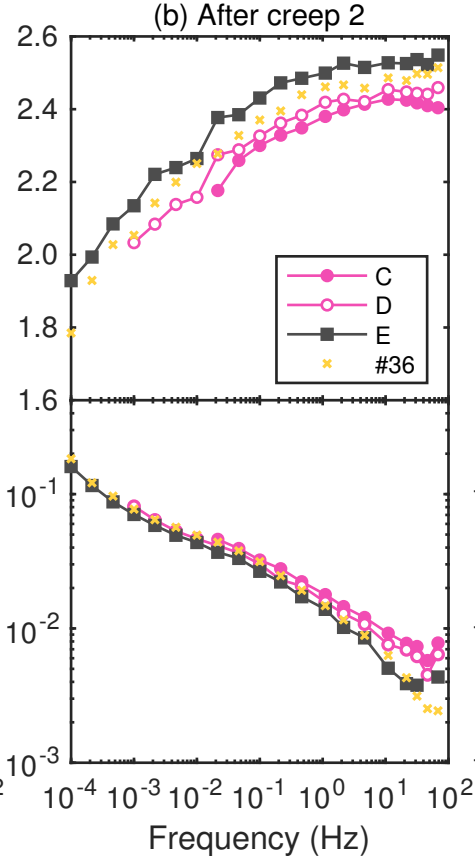
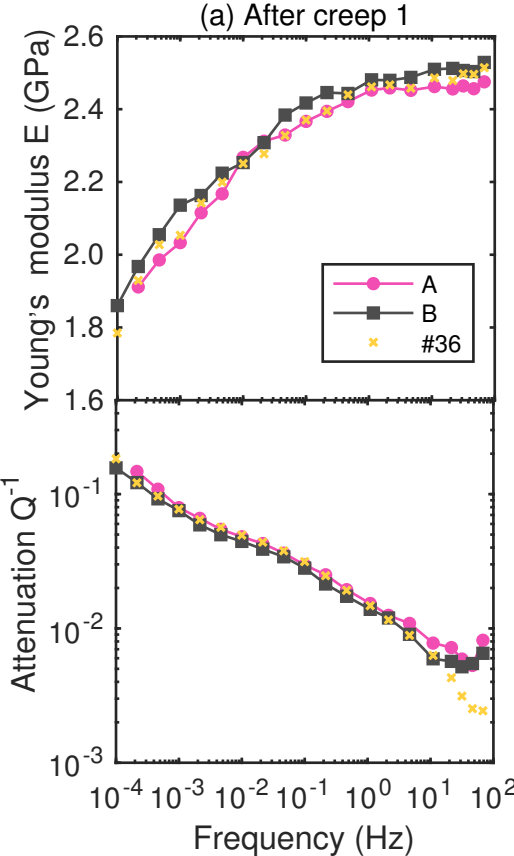


Figure 11.

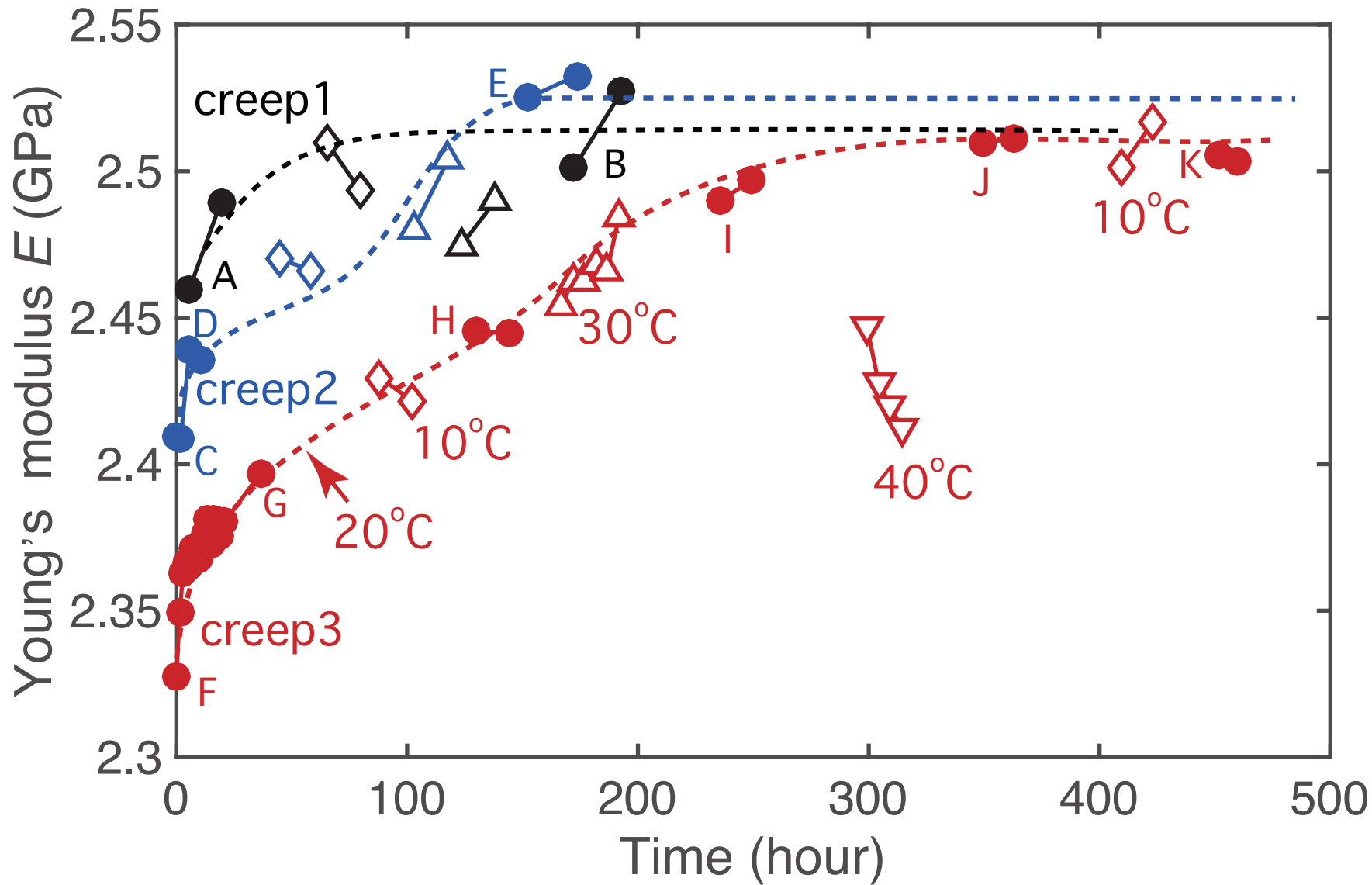


Figure 12.

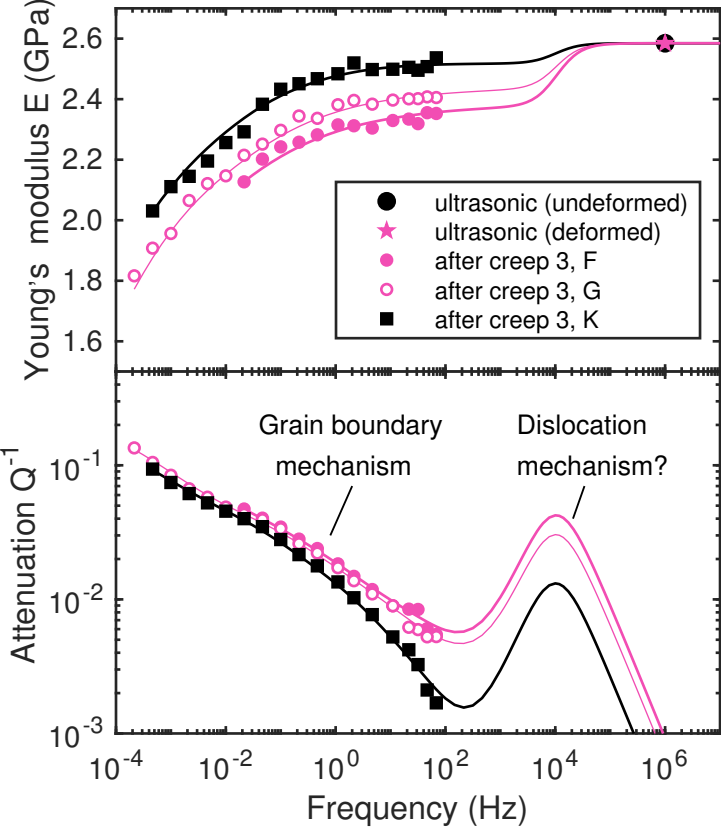


Figure A1.

