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Variational Multiscale Stabilization of Finite and Spectral Elements for Dry and Moist Atmospheric Problems

by
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ad inexplorata

Motto, Air Force Test Pilots -Edwards Air Force Base

Abstract

In this thesis the finite and spectral element methods (FEM and SEM, respectively) applied to problems in atmospheric simulations are explored through the common thread of Variational Multiscale Stabilization (VMS). This effort is justified by three main reasons. (i) the recognized need for new solvers that can efficiently execute on massively parallel architectures –a spreading framework in most fields of computational physics in which numerical weather prediction (NWP) occupies a prominent position. Element-based methods (e.g. FEM, SEM, discontinuous Galerkin) have important advantages in parallel code development; (ii) the inherent flexibility of these methods with respect to the geometry of the grid makes them a great candidate for dynamically adaptive atmospheric codes; and (iii) the localized diffusion provided by VMS represents an improvement in the accurate solution of multi-physics problems where artificial diffusion may fail. Its application to atmospheric simulations is a novel approach within a field of research that is still open. First, FEM and VMS are described and derived for the solution of stratified low Mach number flows in the context of dry atmospheric dynamics. The validity of the method to simulate stratified flows is assessed using standard two- and three-dimensional benchmarks accepted by NWP practitioners. The problems include thermal and gravity driven simulations. It will be shown that stability is retained in the regimes of interest and a numerical comparison against results from the literature will be discussed. Second, the ability of VMS to stabilize the FEM solution of advection-dominated problems (i.e. Euler and transport equations) is taken further by the implementation of VMS as a stabilizing tool for high-order spectral elements with advection-diffusion problems. To the author’s knowledge, this is an original contribution to the literature of high order spectral elements involved with transport in the atmosphere. The problem of monotonicity-preserving high order methods is addressed by combining VMS-stabilized SEM with a discontinuity capturing technique. This is an alternative to classical filters to treat the Gibbs oscillations that characterize high-order schemes. To conclude, a microphysics scheme is implemented within the finite element Euler solver, as a first step toward realistic atmospheric simulations. Kessler microphysics is used to simulate the formation of warm, precipitating clouds. This last part combines the solution of the Euler equations for stratified flows with the solution of a system of transport equations for three classes of water: water vapor, cloud water, and rain. The method is verified using idealized two- and three-dimensional storm simulations.

Resumen

En esta tesis los métodos de elementos finitos y espectrales (FEM - *finite element method* y SEM- *spectral element method*, respectivamente), aplicados a los problemas de simulaciones atmosféricas, se exploran a través del método de estabilización conocido como *Variational Multiscale Stabilization* (VMS). Tres razones fundamentales justifican este esfuerzo: (i) la necesidad de tener nuevos métodos de solución de las ecuaciones diferenciales a las derivadas parciales usando máquinas paralelas de gran escala –un entorno en expansión en muchos campos de la mecánica computacional, dentro de la cual la predicción numérica de la dinámica atmosférica (NWP-*numerical weather prediction*) representa una aplicación importante. Métodos del tipo basado en elementos (por ejemplo, FEM, SEM, Galerkin discontinuo) presentan grandes ventajas en el desarrollo de códigos paralelos; (ii) la flexibilidad intrínseca de tales métodos respecto a la geometría de la malla computacional hace que esos métodos sean los candidatos ideales para códigos atmosféricos con mallas adaptativas; y (iii) la difusión localizada que VMS introduce representa una mejora en las soluciones de problemas con física compleja en los cuales la difusión artificial clásica no funcionaría. La aplicación de FEM o SEM con VMS a problemas de simulaciones atmosféricas es una estrategia innovadora en un campo de investigación abierto. En primera instancia, FEM y VMS vienen descritos y derivados para la solución de flujos estratificados a bajo número de Mach en el contexto de la dinámica atmosférica. La validez del método para simular flujos estratificados es verificada por medio de test estándar aceptado por la comunidad dentro del campo de NWP. Los test incluyen simulaciones de flujos térmicos con efectos de gravedad. Se demostrará que la estabilidad del método numérico se preserva dentro de los regímenes de interés y se discutirá una comparación numérica de los resultados frente a aquellos hallados en la literatura. En segunda instancia, la capacidad de VMS para estabilizar métodos FEM en problemas de advección dominante (i.e. ecuaciones de Euler y ecuaciones de transporte) se implementa además en la solución a elementos espectrales de alto orden en problemas de advección-difusión. Hasta donde el autor sabe, esta es una contribución original a la literatura de métodos basados en elementos espectrales en problemas de transporte atmosférico. El problema de monotonicidad con métodos de alto orden es tratado mediante la combinación de SEM+VMS con una técnica de *shock capturing* para un mejor tratamiento de las discontinuidades. Esta es una alternativa a los filtros que normalmente se aplican a SEM para eliminar las oscilaciones de Gibbs que caracterizan las soluciones de alto orden. Como último punto, se implementa un esquema de humedad acoplado con el núcleo en elementos finitos; este es un primer paso hacia simulaciones atmosféricas más realistas. La microfísica de Kessler se emplea para simular la formación de nubes y tormentas cálidas (*warm clouds*: no permite la formación de hielo). Esta última parte combina la solución de las ecuaciones de Euler para atmósferas estratificadas con la solución de un sistema de ecuaciones de transporte de tres estados de agua: vapor, nubes y lluvia. La calidad del método es verificada utilizando simulaciones de tormenta en dos y tres dimensiones.

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Chapter 1

Introduction

Throughout the decades, there has been a non-unique approach to weather prediction. The approach to forecasting based on the numerical solution of partial differential equations sets the basis of that branch of mathematical-physics known as *Numerical Weather Prediction* (NWP). NWP is the main topic of discussion of this thesis, with a special dedication to the numerical methods of solution of the governing equations.

In spite of the ninety years that have passed since the work of Richardson during World War I and published in the 20s (Richardson, 1922), and the ever increasing computational power available today, NWP still represents one of the most challenging problems in computational sciences. The difficulties are due to the amount of physical information whose foundations are yet to be fully understood (e.g., turbulence, radiation, microphysical processes, cloud formation, precipitation) and that, eventually, must be understood as a whole and implemented with efficiency and accuracy on a computer. Certainly, great advances have been made since 1922, and the advent of massively parallel computers and the great progress in measurement techniques gave an important impulse to this evolution. In the atmospheric community there are still different views starting from the definition of a “most proper” set of equations, to the numerical method to solve them. In this thesis, we partially select the governing equations based on the comparison found in Giraldo and Restelli (2008), and combine their analysis with the experience that we gained in Computational Fluid Dynamics (CFD) for compressible flows in the low Mach number regime.

1.1 Trends in numerical methods for atmospheric simulations

Parallel scientific computing has seen a great deal of advancement in the past decade. Nowadays, petascale systems¹ are the driving force in high performance computing, with core counts approximating $\mathcal{O}(10^5)$ and $\mathcal{O}(10^6)$ (Nair et al., 2011). To take full

¹Petascale: 10^{15} floating point operations per second.

advantage of the performance of these architectures, the need for specific characteristics in new models/codes drove scientists from different fields to go back to the design board and start from scratch in the construction of their numerical algorithm. This is required by the need for very specific features that the numerical method must have to reach very high levels of scalability on the new machines. Most NWP codes in use today (at least in operational mode) are based on the finite-difference (FD) and spectral transform (ST) space discretization of the governing equations (see Tables 1.1-1.3). The ease of coding and their good performance has made FD popular with limited area models (LAM), whereas global circulation models (GCM) are mostly based on ST.

In the era of high-performance computing, the search for efficient parallel codes on hundreds of thousands of processors is suggesting the implementation of alternative numerical methods that are based on local operation properties (element-based techniques). Of these, Galerkin methods (see Giraldo and Restelli (2008), Kelly and Giraldo (2012), or Dennis et al. (2012), among others) are the most common. The spectral element method (SEM) and discontinuous Galerkin (DG) are two examples. These alternatives are justified by the proven high parallel efficiency of local methods (Nair et al., 2011); their efficiency on large to very large machines is given by the minimal parallel communication footprint that is of vital importance as resolution increases. For a better understanding, let us make reference to Fig. 1.1, where two different computational grids are represented and compared against each other. In (a), the grid consists of nine finite elements Ω_h^{el} . Using element-based methods such as finite and spectral elements, the solution is sought on an element-by-element basis, making each element dependent on the others only through its shared boundary nodes. When the finite element grid is partitioned into smaller portions of the global domain, the only information that needs to be exchanged among the subdomains of the partition is that on the boundary nodes (or edges with DG) that each subdomain shares with its neighbors. In contrast, in (b) the grid is a classical structured, rectangular finite difference grid that here is plotted to be a direct analogue (in terms of node count) of the finite element mesh. Because a finite difference stencil is such that differentiation on each node in the domain requires information from a set of adjacent nodes (in the Cartesian directions only) that varies with the order of differentiation, when the domain is partitioned, certain specific nodes will belong to two overlapping subdomains. Because of this, a lot more communication is necessary. The belief that there is no possibility of efficient scaling with FD is false, if taken in an absolute sense. What may cause FD to lose efficiency in parallel is a combination of factors such as the order of differentiation or the number of nodes within each overlapping region. In the case of element-based schemes communication is naturally low by construction.

The use of Galerkin methods in atmospheric simulations began five decades ago with the work on finite elements by Holmstrom (1963) and Simons (1968) in the 60s. This continued in the 70s (e.g. Cullen (1974), Francis (1972)) and was followed by an extensive production of articles in the 80s and 90s with, e.g., Staniforth (1984), Beland et al. (1983), or Burridge et al. (1986), who set the foundations of the operational *Global Environmental Multiscale* (GEM) model (Cote et al., 1998a; Yeh et al., 2002) of the

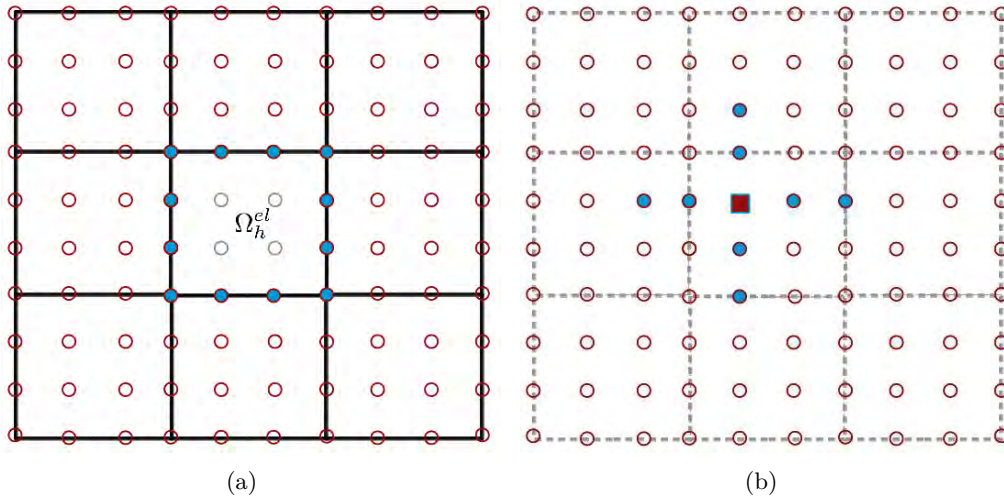


Figure 1.1: Examples of the adjacency pattern for a finite element Ω_h^{el} (a), and for a node that belongs to a finite difference grid (b). In (a) and (b) information is exchange, respectively, element- and node-wise. In (b), the only nodes that allow information to be shared between elements are the shared nodes on the boundary of neighboring elements (blue dots on the boundary of Ω_h^{el}). In (b), the cross made of blue circular nodes and a central red node is the stencil of a 2nd-order differentiation performed on the central node. How this plots relate to parallelization is described in the text.

Canadian Meteorological Center & Meteorological Research Branch (CMC-MRB). In the UK, [Untch and Hortal \(2004\)](#) used finite elements for the vertical discretization of a semi-Lagrangian transport scheme and introduced it in the operational version of the European Centre for Medium-Range Weather Forecasts (ECMWF) global spectral model (IFS), with great improvement with respect to the FD version of the code. At the same time, since [Davies et al. \(2006\)](#), the British Met Office committed to a FE approach for the treatment of the vertical atmosphere within their global and regional models. In the domain of Geophysical Fluid Dynamics (GFD), more Galerkin-type models appear since the beginning of the new millennium: in [Brdar et al. \(2012\)](#), [Lauter et al. \(2008\)](#), [Nair et al. \(2005\)](#), [Giraldo and Rosmond \(2004\)](#), [Giraldo et al. \(2002\)](#), and [Giraldo \(2000\)](#), different variational formulations mostly based on SEM and DG are employed to solve the shallow water equation, hyperbolic systems on the sphere, or the Navier-Stokes and Euler equations for non-hydrostatic problems on unstructured grids. Similarly, and with dynamic mesh adaptivity in mind, finite volume (FV) based softwares such as the German *ICOsahedral Non-hydrostatic General Circulation Model* (ICON) ([Gassmann and Herzog, 2008](#); [Bonaventura and Ringler, 2005](#)), the Japanese *Nonhydrostatic Icosahedral Atmospheric Model* (NICAM) ([Tomita and Satoh, 2004](#); [Satoh et al., 2008](#)), and the American *Operational Multiscale Environmental model with Grid Adaptivity* (OMEGA) by [Bacon et al. \(2000\)](#), are further examples of new trends in NWP. More recently, examples of element-based models are the SE/DG *Nonhydrostatic Unified Model for the Atmosphere* (NUMA), whose linear scalability up to 32,000 cores for DG using MPI was shown in [Kelly and Giraldo \(2012\)](#), the SE *Community Earth System Model* (CESM) with scala-

bility shown up to 160,000 CPUs in Dennis et al. (2012), and DUNE (Brdar et al., 2012), a DG implementation of a finite element research code. Müller et al. (2011) are working on adaptive grid refinement and built an adaptive solver for atmospheric problems. In 2010, Aubry et al. (2010) presented their results from an edge-based finite element solver built on a fully unstructured grid. A descriptive list of existing research and operational models is reported in Section 1.2.

1.2 Existing models in NWP

Tables (1.1-1.3) report a non-exhaustive list of numerical models existing today in the atmospheric literature. The tables are organized by the type of spatial discretization that is used in each code. Ten out of the twenty-nine codes listed are based on the finite difference method. Except for the multiscale models *Unified Model* (UM), *Nonhydrostatic Multiscale Model* (NMMB), and *EULerian LAGrangian* (EULAG), all FD-based codes are limited area models (LAM). Spectral transform and finite volumes represent the second major trend (six codes each). Codes based on the spectral transform are common for GCM only. High-order element-based methods (spectral elements and DG) follow, while finite elements, only used in IFS (partially) and GEM, are the least common of all. For reasons that will become clearer in the chapters that follow, the temporal integration schemes that are mostly used are the split-explicit and the semi-implicit methods.

1.3 Aim of this thesis

Based on the consideration reported above on element-based methods, this thesis delves into the use of FEM and SEM to solve problems of typical interest in atmospheric simulations: (i) nonhydrostatic, stratified, dry and moist flows, and (ii) transport in the atmosphere. Because of the instability that occurs in the numerical solution of (i) and (ii) (i.e., Euler and transport-diffusion equations), we stress the analysis on stabilization of FEM and SEM by the variational multiscale (VMS) scheme first introduced by Hughes (1995) and used extensively for incompressible flows and transport problems ever since.

In this framework, this thesis has three main objectives.

- Solution of the Euler equations using low-order finite elements stabilized by a compressible extension of VMS. Testing is performed using standard two- and three-dimensional benchmarks to verify the ability of this technique to solve compressible flows at low and very low Mach numbers. The algorithm is implemented within the multiphysics, multiscale, massively parallel code *Alya*².
- Development of a variational multiscale stabilization scheme for high-order spectral elements to solve the advection-diffusion equation in atmospheric transport problems. Because monotonicity-preserving methods are particularly important in atmospheric transport, SEM + VMS is enhanced by a discontinuity capturing

²Alya is developed at Barcelona Supercomputing Center, Spain. See Appendix B.

Model	Origin	Type	Equations and Numerical Scheme
- Unified Model (UM) (Malcolm, 1996)	UK, Met. Office	NH LAM/GCM	Fully compr. <i>FD</i> semi-impl., semi-Lagrangian
- COSMO (Doms and Schattler, 2002)	Germany, DWD	NH LAM	Fully compr. <i>FD</i> split-expl. + semi-implicit
- WRF-ARW (Skamarock et al., 2007)	USA, NCAR and collaborations	NH LAM	Fully compr. <i>FD</i> semi-Impl., split Expl.
- WRF-NMM (Janjic et al., 2001)	USA, NCEP and collaborations	HS/NH LAM	Primitive eqns. <i>FD</i> semi-Impl., split Expl.
- MM5 (Dudhia, 1993)	USA, NCAR and collaborations	NH LAM	Primitive eqns. <i>FD</i> leap frog
- Hirlam (Room, 2001)	France, Meteo France + coll.	NH LAM	Primitive eqns. <i>FD</i> semi-Impl., semi-Lagrangian (vert.)
- MC2 (Benoit et al., 1997)	Canada, Res. Centr. for NWP	NH LAM	Euler eqns. <i>FD</i> semi-Lagrangian, semi-impl.
- RAMS (Pielke et al., 1992)	USA, Colorado State U.	HS/NH LAM	Primitive eqns. <i>FD</i> leap frog
- ETA (Janjic, 1994)	USA, NCEP and collaborations	HS/NH LAM	Primitive eqns. <i>FD</i> semi Impl., split-expl
- NMM-B (Janjic, 2003)	USA, NCEP and collaborations	HS/NH LAM/GCM	Primitive eqns. <i>FD</i> semi Impl., split-expl
- COAMPS (Hodur, 1997)	USA, Naval Research Lab	NH LAM	Fully compr. <i>FD</i> semi-impl., expl.
- EULAG (Prusa et al., 2008)	USA, NCAR	NH LAM/GCM	Comprx./Incomprx. <i>FD</i> on generalized coords. NFT (Non-oscil Forward in Time).

Table 1.1: Existing atmospheric models in NWP

- IFS (Untch and Hortal, 2004)	UK, ECMWF	NH	Fully compr.+ T, <i>ST + FE</i> in z
- CAM EUL (Neale et al., 2010)	USA, NCAR	LAM	semi-Impl., semi-Lagrangian.
- NOGAPS (until 1998) (Hogan et al., 1991)	USA, Naval Research Lab	HS	shallow atmo., <i>ST + FD</i> in z
- ALADIN-NH (Laprise, 1992)	France, Meteo France	GCM	semi-Impl.
- ARPEGE (Courtier et al., 1991)	France, Meteo France	NH	Primitive eqns. <i>ST</i>
- AROME <i>www.cnrm.meteo.fr/arome/</i>	European consortium	GCM	cetranl time-diff.+ semi-impl. corr.
- GME (Majewski et al., 2002)	Germany, DWD	HS/NH	Fully compr. <i>ST + FD</i> in z
- OMEGA (Bacon et al., 2000)	USA, Centr. Atmo. Phys.	LAM	semi-impl., semi-Lagrangian
- ICON (Gassmann and Herzog, 2008)	Germany, MPIfM + DWD.	HS/NH	Primitive eqns. <i>ST</i>
- NICAM (Satoh et al., 2008)	Japan	LAM/GCM	(split-expl) semi Impl.
- CAM FV (Neale et al., 2010)	USA, NCAR	NH	As ALADIN-NH+soph. physics <i>ST</i>
- MCore (Ullrich et al., 2010)	USA, U. Mich.	LAM	semi-impl.
		NH	Primitive eqns. <i>FV/SE</i>
		GCM	semi-implicit
		NH	Fully compr. <i>FV</i>
		GCM	split-expl, semi-impl.
		NH	Fully compr. <i>FV</i>
		GCM	semi-implicit
		NH, GCM	Euler Eqns. <i>FV</i>
		LAM	split-expl, semi-impl.
		HS	primitive eqns. <i>FV</i>
		GCM	Explicit
		NH	Fully compr. <i>FV</i>
		GCM	

Table 1.2: Existing atmospheric models in NWP. (Continuation of Tab.1.1).

- NUMA (Kelly and Giraldo, 2012)	USA, Naval Postgraduate School	NH unified: LAM/GCM	Fully compr. <i>SE/DG</i> Implicit/Explicit (IMEX)
- DUNE (Brdar et al., 2012)	Germany, U. Freiburg	NH unified: LAM	Fully compr. <i>DG</i> Expl. (other options)
- NSEAM (from NOGAPS) (Giraldo and Rosmond, 2004)	USA, Naval Research Lab	NH GCM	Primitive eqns. <i>SE</i> semi-impl., semi-Lagrangian
- HOMME (Thomas and Loft, 2005)	USA, NCAR	HS GCM	Primitive eqns. <i>SE</i> Explicit + other options
- GEM (Cote et al., 1998b)	Canada, CMC & MRB	HS+NH LAM/GCM	Primitive eqns., <i>FE</i> semi-Lagrangian, impl.

Table 1.3: Existing atmospheric models in NWP (Continuation of Tab.1.1)

scheme to approach a quasi-monotonic solution. The algorithm is developed within the massively parallel, multiscale, atmospheric code *NUMA*³.

- To close the circle of applications, the Euler and transport equations are coupled to model the flow of moist atmospheres where warm, precipitating clouds form. This represents a proof-of-concept of the ability of Galerkin methods stabilized by VMS to solve problems of atmospheric relevance. To achieve this goal, the low-order finite elements that define the first objective of this study are extended to coupled problems with complex physics.

To our knowledge, the FEM and SEM algorithms with VMS proposed in this thesis represent the first continuous Galerkin methods with VMS stabilization applied to problems that are important for atmospheric modelers.

1.3.1 Publications derived from this work

Based on this work, the following publications were derived: three peer-reviewed research papers, two short conference articles, one technical report, and a set of talks given in international conferences and workshops.

Peer-reviewed journal articles

- Marras, S., Kelly, J. F., Giraldo, F. X., and Vázquez, M. 2012. “Variational Multiscale Stabilization of High-Order Spectral Elements for the Advection-Diffusion Equation” *Journal of Computational Physics* **231**, 7187-7213.
- Marras, S. Moragues, M. Vázquez, M. Jorba, O., and Houzeaux, G. “A Variational Multiscale Stabilized Finite Element Method for the Solution of the Euler Equations of Nonhydrostatic Stratified Flows”, *Journal of Computational Physics* (In revised form).
- Marras, S. Moragues, M. Vázquez, M. Jorba, O., and Houzeaux, G., “A Variational Multiscale Stabilized Finite Element Method for the Solution of the Euler Equations of Moist Atmospheric Flows”, *Journal of Computational Physics* (Submitted, July 2012).

In proceedings

- Marras, S., Vázquez, M. Jorba, O. Aubry, R., and Houzeaux, G. 2010 “Application of a Galerkin Finite Element Scheme to Atmospheric Buoyant and Gravity Driven Flows” AIAA paper 0690, p. 7860-7869. In *Proceedings of the 48th AIAA*

³NUMA is developed at the department of Applied Mathematics at the Naval Postgraduate School, U.S.A. See Appendix B.

Aerospace Sciences Meeting Vol. 9, ISBN: 978-1-61738-422-6, Curran Associates, Inc.

- Aubry, R., Vázquez, M., Houzeaux, G., Cela, J. M., and Marras, S. 2010 “An Unstructured CFD Approach for Numerical Weather Prediction” AIAA paper 0691, p. 7870-7895. In *Proceedings of the 48th AIAA Aerospace Sciences Meeting* Vol. 9, ISBN: 978-1-61738-422-6, Curran Associates, Inc.

Talks at conferences and workshops

- Marras, S., Kelly, J. F., and Giraldo, F. X. 2011 “Towards Positive High-Order Spectral Elements for the Advection Equation”, Institute for Pure and Applied Mathematics (IPAM), UCLA, U.S.A., Reunion meeting. Lake Arrowhead, December.
- Marras, S., Vázquez, M., Jorba, O., Houzeaux, G., and Folch, A. 2010, “Solving Nonhydrostatic Dynamics with a Variational Multiscale Galerkin Solver: Tests and Parallel Performance”, Institute for Pure and Applied Mathematics (IPAM), UCLA, U.S.A. (2010) Long program *Model and Data Hierarchies for Simulating and Understanding Climate*, Los Angeles, March 2010 - June 2010.
- Vázquez, M., Marras, S., Moragues, M., Jorba, O., Houzeaux, G., and Aubry, R. 2010, “A Massively Parallel Variational Multiscale FEM Scheme Applied to Nonhydrostatic Atmospheric Dynamics”, EGU Annual Meeting, EGU-2010-9060, Vienna, Austria

1.4 Organization of the manuscript

The remainder of this thesis is organized as follows. Chapter 2 presents a brief introduction to the physical problem of atmospheric dynamics and its mathematical modeling. It is followed by a general description of the finite and spectral element methods with a brief overview to the issues related to their stabilization. Chapter 4 reports on the application of finite elements and variational multiscale stabilization for the solution of the fully compressible Euler equations. Chapter 5 shows the solution of the advection-diffusion equation by means of high-order spectral elements. The main issue of monotonicity-preserving high-order methods is covered as well. The solution of moist atmospheres with phase change is described in Chapter 6. Conclusions are presented in Chapter 7. Appendix A presents some CFD grid generation techniques that can be of practical use in atmospheric modeling. The two computational environments are presented in Appendix B. The relationship between the equation of θ and T is reported in Appendix C.

Chapter 2

The physical problem

The best world has the greatest variety of phenomena regulated by the simplest laws

–Gottfried W. Leibniz, c. 1700

In this chapter we present the set of equations that govern atmospheric dynamics and transport phenomena in the atmosphere. We discuss the different formulations and justify the selection of the set that is used throughout this thesis.

2.1 Dynamics of dry atmospheres

The motion of the earth atmosphere can be described by the laws of fluid mechanics under the assumption that the air is treated as a continuum. As such, the state of the gas can be described by density, ρ , pressure, p , absolute temperature, T , and a velocity field, \mathbf{u} (Ockendon and Ockendon, 2004). At given T , \mathbf{u} , and height h , the total energy e of the fluid flow is given by the contribution of internal energy $e_i = c_v T$, kinetic energy $e_k = (\mathbf{u} \cdot \mathbf{u})/2$, and potential energy $\Phi = g h$, where c_v is the gas heat coefficient at constant volume and $g = 9.81 \text{ m s}^{-2}$ is the modulus of the acceleration of gravity. From the principles of conservation of mass, momentum, and energy, the Navier-Stokes equations of fluid dynamics are a proper set to describe atmospheric motion. Subgrid viscous effects in atmospheric simulations are typically introduced through the subgrid-scale eddy viscosity of turbulence. Since eddy viscosity is much larger than molecular viscosity, the effects of molecular viscosity in mesoscale models are typically neglected. In this thesis, however, turbulence effects will not be considered and the Euler equations will be used as a suitable model for the problems described throughout.

Nor will we consider the forces due to Coriolis¹ because of the relatively small scales of interest considered throughout. Viscous effects in modeling the atmosphere are indeed

¹For the time being, the hypothesis of a non-rotating domain is considered. In fact, the main goal of this work is not that of modeling a real weather problem, in which case, it is likely that Coriolis effects cannot be neglected, but rather testing a numerical algorithm to verify its suitability for non-hydrostatic stratified simulations in idealized conditions.

introduced when needed, but this is achieved through the use of proper turbulence closures. The problems treated in this thesis are sufficiently idealized that inclusion of turbulence is not necessary. The governing equations of interest are described in what follows.

Let $\mathbf{x} = (x, z)$ and $\mathbf{x} = (x, y, z)$ be a Cartesian, fixed frame of reference of dimensions $d = 2, 3$ and let $t \in \mathbb{R}^+$ be the time variable. Assuming that ρ is a non-negative function of \mathbf{x} and t , mass conservation can be expressed by the conservation law

$$\frac{\partial \rho}{\partial t} + \nabla \cdot \mathbf{U} = 0, \quad (2.1)$$

where $\nabla \cdot$ is the divergence operator acting upon momentum $\mathbf{U} = \rho \mathbf{u}$. \mathbf{U} has components $(U, W)^T$ and $(U, V, W)^T$ in two and three dimensions, respectively. Conservation of \mathbf{U} for a non-viscous fluid reads

$$\frac{\partial \mathbf{U}}{\partial t} + \nabla \cdot \left[\frac{\mathbf{U} \otimes \mathbf{U}}{\rho} + \mathbf{I}p \right] = -g\mathbf{e}_z\rho, \quad (2.2)$$

where \mathbf{I} and \otimes are the identity tensor and tensor product, respectively, and \mathbf{e}_z is the unit vector directed along the vertical direction (z).

In the case of an incompressible flow, the conservation of mass reduces to $\nabla \cdot \mathbf{u} = 0$ and the system of Eqs. (2.1)-(2.2) is self-contained. This is true when the flow Mach number $M = \|\mathbf{u}\|_2/c$ is smaller than 0.3 approximately ($c = \sqrt{\partial p / \partial \rho}$ is the speed of sound). Otherwise, compressibility effects become important in that a variation of pressure implies a variation of density and temperature. A constitutive equation is needed to express the relationship among the three thermodynamics variables. The variation in temperature is modeled by the additional equation of conservation of total energy:

$$\frac{\partial \rho e}{\partial t} + \nabla \cdot [(\rho e + p)\mathbf{u}] = 0. \quad (2.3)$$

At present, only a few atmospheric models are based on Eqs. 2.1-2.3. This set is at the base of the global, icosahedral, nonhydrostatic model for global cloud resolving simulations (NICAM) described by Tomita and Satoh (2004) and Satoh et al. (2008), and was compared against other sets by Giraldo and Restelli (2008).

2.2 Approximations and sets of common use in atmospheric simulations

By algebraic manipulation and/or suitable approximations, the Euler equations are often re-expressed by alternative formulations as a way, for example, to filter motions and solutions that are of no interest for the problem being considered (e.g., sound waves) (Thuburn, 2011a), or to inherently find a direct link between the physics and the set of

variables that describes it. In this respect, each set has its advantages and drawbacks. Unquestionably, no set is optimal per se; it is optimal within a very specific context that, in the case of atmospheric simulations, is not unique. For example, let us introduce the Exner function $\pi = (p/p_0)^{R/c_p}$, a normalized pressure given a reference base-state pressure p_0 , and the relation between π and potential temperature as $\theta = T/\pi$. A change of variables from ρ , p , and e to π and θ yields the self-contained system

$$\frac{\partial \pi}{\partial t} + \mathbf{u} \cdot \nabla \pi - \frac{R}{c_v} \pi \nabla \cdot \mathbf{u} = 0, \quad (2.4a)$$

$$\frac{\partial \mathbf{u}}{\partial t} + \mathbf{u} \cdot \nabla \mathbf{u} + c_p \theta \nabla \pi = -g \mathbf{e}_z, \quad (2.4b)$$

$$\frac{\partial \theta}{\partial t} + \mathbf{u} \cdot \nabla \theta = 0, \quad (2.4c)$$

where $(\pi, \mathbf{u}, \theta)^T$ is the vector of the solution variables (Durrán, 1998; Cullen, 1990). The advantage is clear: the system of (2+d) equations only has (2+d) unknowns so that there is no need for an extra equation to close the system (e.g., equation of state). Not only that, but the use of θ rather than energy is advantageous, in terms of the physics, because information on the stability of the atmosphere is given by $\partial \theta / \partial z$ (Restelli, 2007; Smith, 1979). Of the most common models that use this formulation we list MM5 developed at Penn State and NCAR (Dudhia, 1993), NMM based on the work by Janjic (2003) at NCEP, COAMPS (Hodur, 1997) developed at the U.S. Naval Res. Lab., and HIRLAM (Room, 2001, 2002) by a consortium of European Numerical Weather Services.

Another set of common use is given by the conservation laws of $(\rho, \mathbf{U}, \Theta)^T$ (\mathcal{T} indicates the vector's transpose):

$$\frac{\partial \rho}{\partial t} + \nabla \cdot \mathbf{U} = 0 \quad (2.5a)$$

$$\frac{\partial \mathbf{U}}{\partial t} + \nabla \cdot \left[\frac{\mathbf{U} \otimes \mathbf{U}}{\rho} + p \mathbf{I} \right] = -\rho g \mathbf{e}_z \quad (2.5b)$$

$$\frac{\partial \Theta}{\partial t} + \nabla \cdot \left[\frac{\Theta \mathbf{U}}{\rho} \right] = 0 \quad (2.5c)$$

where $\Theta = \rho \theta$ is density potential temperature. The state law for pressure

$$p = p_0 \left(\frac{R \Theta}{p_0} \right)^\gamma \quad (2.6)$$

completes the system. $\gamma = c_p/c_v$, where c_p and c_v are the heat coefficients at constant pressure and volume, respectively.

The ARW-WRF model (Skamarock et al., 2007) is based on this set, and so are the finite volume simulations by Ahmad and Lindeman (2007), the UK Meteorological Office Unified Model (UM) (Jackson et al., 2001; Malcolm, 1996), and the German LM model (COSMO, 1998).

Finally, computational efficiency suggested the use of the following set in research codes such as the *Nonhydrostatic Unified Model for the Atmosphere* (NUMA) developed at the Naval Postgraduate School (Kelly and Giraldo, 2012):

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{u}) = 0, \quad (2.7a)$$

$$\frac{\partial \mathbf{u}}{\partial t} + \mathbf{u} \cdot \nabla \mathbf{u} + \frac{1}{\rho} \nabla p \mathbf{I} = -g \mathbf{e}_z, \quad (2.7b)$$

$$\frac{\partial \theta}{\partial t} + \mathbf{u} \cdot \nabla \theta = 0, \quad (2.7c)$$

plus equation (2.6).

2.2.1 Remarks on the equation of total energy

The conservation equation of total energy can be replaced by the equation of transport of total temperature. The two equations are mathematically equivalent. However, in the presence of strong pressure waves (i.e. shocks), the numerical solution of the T equation places the discontinuity in the wrong position. Clearly, in the case of atmospheric flows there are no shocks to worry about, so that equation

$$\frac{\partial T}{\partial t} = -\mathbf{u} \cdot \nabla T + \frac{1}{c_v \rho} p \nabla \cdot \mathbf{u} \quad (2.8)$$

is equivalent to Eq. (2.3). Because atmospheric stability is directly linked to the variation of θ along z , in atmospheric simulations the use of θ is advantageous over the use of T or E . For this reason, all but one set described above express energy in terms of potential temperature. One way to see the equivalence is reported in Appendix C. Due to practical reasons related to the stabilization scheme (See Chapter 4 below), the set of equations used in this thesis is composed by the continuity and momentum equations of set (2.5), and by Eq. (2.7c). ρ , θ , and p are related by (2.6). In summary, we are interested in the system

$$\frac{\partial \rho}{\partial t} + \nabla \cdot \mathbf{U} = 0, \quad (2.9a)$$

$$\frac{\partial \mathbf{U}}{\partial t} + \nabla \cdot \left[\frac{\mathbf{U} \otimes \mathbf{U}}{\rho} + p \mathbf{I} \right] = -\rho g \mathbf{e}_z, \quad (2.9b)$$

$$\frac{\partial \theta}{\partial t} + \frac{\mathbf{U}}{\rho} \cdot \nabla \theta = 0. \quad (2.9c)$$

For how linearization is constructed (see page 37), we express ∇p with respect to ρ and θ . From the state equation $p = c_0(\rho\theta)^\gamma$, where $c_0 = R^\gamma/p_0^{\gamma-1}$, we have that

$$\nabla p = c_0\gamma(\rho\theta)^{\gamma-1} [\rho\nabla\theta + \theta\nabla\rho].$$

What has been stated so far applies to dry environments only. The necessary corrections, definitions, and derivations for a moist atmosphere will be given in Chapter 6.

2.2.2 Nearly-hydrostatic flows

Dynamics in the atmosphere is characterized by small variations of the thermodynamic quantities with respect to some background reference state (Marchuk, 1974; Klein, 2000). This is expressed by the splitting $\rho(\mathbf{x}, t) = \rho'(\mathbf{x}, t) + \bar{\rho}(z)$, $p(\mathbf{x}, t) = p'(\mathbf{x}, t) + \bar{p}(z)$, and $\theta(\mathbf{x}, t) = \theta'(\mathbf{x}, t) + \bar{\theta}(z)$, where the primed and barred quantities represent, respectively, the perturbation and the background state of ρ , p , and θ . They are such that $\rho' \ll \bar{\rho}$, $p' \ll \bar{p}$ and $\theta' \ll \bar{\theta}$. When vertical acceleration is zero, the vertical component of Eq. (2.9b) simplifies to the equation of hydrostatic balance

$$\partial_z \bar{p} = -g\bar{\rho}. \quad (2.10)$$

Given these considerations and the analysis of nearly-hydrostatic flows for well-balanced methods (Botta et al., 2004), set (2.9) changes to

$$\frac{\partial \rho}{\partial t} + \nabla \cdot \mathbf{U} = 0 \quad (2.11a)$$

$$\frac{\partial \mathbf{U}}{\partial t} + \nabla \cdot \left[\frac{\mathbf{U} \otimes \mathbf{U}}{\rho} + p' \mathbf{I} \right] = -\rho' g \mathbf{e}_z \quad (2.11b)$$

$$\frac{\partial \theta}{\partial t} + \frac{\mathbf{U}}{\rho} \cdot \nabla \theta = 0 \quad (2.11c)$$

System (2.11) is used in this thesis.

2.3 Hydrostatic vs nonhydrostatic models

Atmospheric models can be distinguished as hydrostatic and non-hydrostatic. If we assume the vertical acceleration to be negligible, the vertical momentum equation of the hydrostatic system reduces to the diagnostic equilibrium equation (2.10). At every time-step, this time-independent equation is solved instead of the full equation for vertical

momentum. Sound waves in the vertical direction are eliminated (Durrán, 1998). They are not, however, eliminated in the horizontal direction. Because the size of the domain in the horizontal direction is typically much larger than the vertical depth of the atmosphere, and the grid size along x and y may be orders of magnitude larger than the grid spacing along z , the stiffness of the problem in the sense of grid size and corresponding time-step size is relaxed.

The hydrostatic approximation has been the core of NWP for the past four decades. This approximation is valid for horizontal grid spacing larger than 10 km (Janjic, 1994; Thuburn, 2011a). The hydrostatic approximation is still appropriate to simulate synoptic scale phenomena where the vertical acceleration can be neglected, but is no longer considered in any mesoscale simulation. With the availability of more powerful computers, nonhydrostatic formulations have eventually been investigated (see, Janjic et al. (2001); Benoit et al. (1997); Bonaventura (2000); Gassmann (2005); Grell et al. (1995); Hodur (1997); Janjic (2003); Skamarock et al. (2007); Xue et al. (2000); Giraldo and Restelli (2008)) and are today the rule in the numerical approximation of mesoscale dynamics.

2.4 Transport in the atmosphere

In this thesis, the governing equations of a dry atmosphere are coupled to a set of transport equations for tracers (see Chapter 6). Generally speaking, a tracer is any quantity that is transported by the flow. This includes water species, chemicals, aerosols, and others. Their transport is described by the same equations. As for the case of Eq. (2.1), conservation of mass for a tracer i of density ρ_i and mixing ratio $q_i = \rho_i/\rho$ is expressed by

$$\frac{\partial \rho q_i}{\partial t} = -\nabla \cdot (\rho q_i \mathbf{u}) + S_i, \quad (2.12)$$

where S_i represents sources or sinks and where, for simplicity, diffusion is neglected. For example, in the case of water vapor, S_i is driven by evaporation and condensation. In spite of the mathematical simplicity of Eq. (2.12), its correct numerical approximation is still an open field of research. The importance of its correct numerical representation is vital in NWP. To understand this we introduce the concept of *tracer-air mass consistency* (see, Lauritzen et al. (2011) and references therein). In the case of the continuous equations, if $q_i = 1$ Eq. (2.12) reduces to the equation of conservation of air. In discrete space, this is certainly not achieved if the equation of conservation of air mass and of the tracer are solved by two different numerical methods. This is a classical issue in the case of either online applications that use different numerics for different equations (e.g., transport of reacting chemicals vs transport of dry air), and even more so in the case of offline simulations where the air mass properties come from a different model or observations. Although this topic is not treated in this thesis, we believe that it is worthwhile to keep this in mind throughout the development of a new atmospheric model, such as the case of the one described below.

To conclude this introduction to the physical problem, we write the advective form of

(2.12) that will be used later on. We are aware of the inconvenient properties (e.g., conservation) that this form carries (Lauritzen et al., 2011). Nevertheless, it is a classical formulation of the equation of transport of water concentration (water vapor, cloud water, rain, ice, etc.) that is used in weather forecasting. By using the equation of mass of dry air to eliminate ρ from (2.12), we obtain:

$$\frac{\partial q_i}{\partial t} = -\mathbf{u} \cdot \nabla q_i + S_i. \quad (2.13)$$

This is not the equation of conservation of q_i . Moreover, no numerical method of solution will be conservative on (2.13). We will touch more on this in Chapter 5.

As moisture is an extremely noisy variable that could cause serious stability and convergence problems, the capacity of a numerical scheme to produce a monotonic solution to (2.13) is another relevant point to analyze. If, for example, our system produced negative moisture, the physical parametrization would have to resolve this issue in some way (e.g. filtering the negative values); the wrong feedback that this condition will send to many other variables would, in turn, affect the moisture and hence cause *artificial rain* to be produced. The words of John P. Boyd are an amusing conclusion to this paragraph: "[...] Clever adaptive algorithms that work for smooth, straight shocks disintegrate into computational anarchy when flayed by gravity waves, assaulted by moist convective instability, battered by highly temperature-sensitive photochemistry, and coupled to the vastly different time and space scales of the ocean[...]" (SIAM News, Multiscale Numerical Algorithms for Weather Forecasting and Climate Modeling: Challenges and Controversies. Nov 2008, Vol.41 issue 9). Monotonic solutions are certainly more difficult to achieve with high order numerical methods. The problem is particularly challenging when Eq. (2.13) is solved by the spectral element method. In Chapter 5 we will specifically address this problem.

2.5 Characteristic scales in dynamic meteorology

Since the work by Ligda (1951) on radar observations of storms, atmospheric motions have been categorized into three spatial scales as follows: microscale: $l < 20 \text{ km}$, mesoscale: $20 \text{ km} < L < 1000 \text{ km}$, and synoptic scale: $L > 1000 \text{ km}$. A similar subdivision comes from Pielke (2002), but according to whom mesoscales only extend from 20 to 200 km, leaving a larger synoptic range ($> 200, \text{ km}$). Earlier on, Stull (1988) called macro what is synoptic to Pielke, defined mesoscale the range between 200 m and 200 km, and assigned micro and micro δ the ranges 2 m–10 km and 0–2 m respectively. Further scales were defined as macro- α ($L > 10000 \text{ km}$), macro- β ($10000 \text{ km} > L > 2000 \text{ km}$), meso- α ($2000 \text{ km} > L > 200 \text{ km}$), meso- β ($200 \text{ km} > L > 20 \text{ km}$), meso- γ ($20 \text{ km} > L > 2 \text{ km}$), micro- α ($2 \text{ km} > L > 200 \text{ m}$), micro- β ($200 \text{ m} > L > 20 \text{ m}$) and micro- γ ($L < 20 \text{ m}$) by Orlandi (1975), and based on other observations, atmospheric phenomena have also been categorized into masoscale, mesoscale, misoscale, mososcale and musoscale by Fujita (1981).

Table (2.1) lists the main characteristic scales of interest.

The difference between *global*, *synoptic*, *meso*, and *urban* ($L < 200\text{ m}$) is relevant when looking at the terms to be considered in the equations, and consequently when deciding on the numerical solution.

In the case of the work that we present, the equations and the numerical method apply to problems in mesoscale meteorology. The finite element method in itself is not scale-selective; this means that a finite element-based dynamical core, after proper modification of the equations (i.e., additional Coriolis effects), could, in theory, be transferred onto the global/synoptic scale in a unified manner as reported in the recent work by Kelly and Giraldo (2012). This will not be treated, but it is a great advantage to consider for it lies at the basis of the global extension of the algorithm to global scales.

2.6 Summary

In this chapter, we described the different sets of equations that are commonly adopted in atmospheric modeling. Of these, we focused on different formulations of the Euler equations in stratified environments and defined the equations that will be used in this thesis (set 2.11). Leaving the details for a dedicated part of the document (Chapter 6), we also introduced the transport equations of passive tracers that describe advection of water tracers in the atmosphere (Eq. (2.13)). No further details were given on the coupling of the two systems because it would require explanations that, at this point along the manuscript, may result unclear to the reader who is not familiar with wet atmospheric dynamics.

The material that was presented is sufficient to introduce part of the notation and the framework within which the thesis is developed. Certain topics of equal weight in NWP, such as the construction of the anelastic and pseudo-incompressible equations to filter acoustic waves (see, e.g., Durran (2008) and references therein), or the shallow-water equations fall beyond the scope of this thesis.

Table 2.1: Scales of atmospheric motions. Adapted from page 5 of [Holton \(2004\)](#), and from Fig. 1.1 of [Thuburn \(2011a\)](#)

Type of motion	Horizontal scale (m)	Time scale (s)
Molecular diffusion	10^{-2} - 10^2	10 - 10^8
Minute turbulent eddies	$10^{-2} - 10^{-1}$	10^1
Small eddies	$10^{-1} - 1$	10^1
Dust devils	1 - 10	10^1
Gusts	10 - 10^2	10^1 - 10^2
Tornadoes	10^2	10^3
Cumulonimbus clouds	10^3	10^3 - 10^4
Fronts, squall lines	$10^4 - 10^5$	10^4
Hurricanes	10^5	10^5
Synoptic cyclones	10^6	10^5 - 10^6
Planetary waves	10^7	10^6

Chapter 3

Numerical methods: Finite and Spectral Elements

It is necessary to solve differential equations

–Isaac Newton, c. 1700

In this chapter, the fundamentals of the approximation of partial differential equations (PDEs) by Galerkin methods are introduced. The finite element and the spectral element methods, FEM and SEM respectively, are a special type of Galerkin approximation techniques. We introduce the ideas behind Galerkin schemes in general and then distinguish between FEM and SEM in particular. We define their properties and underline the most salient differences. The FEM solution of a 1D steady-state, scalar, linear, advection-diffusion (AD) equation is then used to present the idea of unbounded solutions and the need for numerical stabilization. Finally, we introduce the fundamentals of a particular category of stabilization techniques in the context of Galerkin methods to familiarize with one of the numerical contents used in the chapters that follow.

3.1 Galerkin methods

Introduced in the early 40s in the study of vibration and equilibrium (Courant, 1943), but extensively developed only in the late 1950s by structural dynamicists in the aircraft industry, finite element methods¹ are among the most common numerical methods in use today in a wide range of applications (e.g. structural analysis and design (Yang, 1985), fluid dynamics (Zienkiewicz et al., 2005), electromagnetism (Bastos and Sadowski, 2003)). Accepted by scientists and engineers in theoretical studies and applications, the ease in modeling complex geometries, the flexible and general purpose programming format that it implies, and the intrinsic treatment of differential-type boundary conditions made it a robust tool for the solution of any differential problem (Donea and Huerta, 2003).

¹*Spectral elements* are finite elements as well. The difference, that will result clear by reading this chapter, lies in the type of element that is used in the sense of approximation.

In the following, we will deal with the idea behind the method of weighted residuals, of which the Galerkin finite and spectral element methods represent a special case. For a simple but quasi-rigorous analysis of the method we use a problem of real engineering interest and that, as we presented in Chapter 2, is a fundamental problem in numerical weather prediction: the advection-diffusion equation. The reader is referred to the books by Fletcher (1987), Quarteroni and Valli (1994), and Karniadakis and Sherwin (1999) as a reference for the more mathematical aspects of Galerkin methods.

Let us take a general differential problem

$$\mathcal{L}(q) = f, \quad (3.1)$$

where \mathcal{L} is the combination of differential operators in space \mathbf{x} and time t , and f is a forcing function. Let d indicate the space dimension and let $\Omega \subset \mathbb{R}^d$ be the domain bounded by the boundary $\partial\Omega$ where (3.1) is defined within the time interval $(0, t_f)$, and $t_f \in \mathbb{R}^+$.

For the problem to be well-posed, suitable boundary and initial conditions must be added to (3.1). Unless otherwise stated, given a known function g , Dirichlet boundary conditions $q(\mathbf{x}) = g$ for $\mathbf{x} \in \partial\Omega$ will be applied to the problems described throughout this work.

As previously said, Galerkin methods are a particular case of the method of weighted residuals. The idea behind this method is the numerical representation of the solution variable q by a finite dimensional approximation q^h obtained by the expansion

$$q^h(\mathbf{x}) = \sum_{k=0}^N \psi_k(\mathbf{x}) \hat{q}_k, \quad (3.2)$$

where N is the number of k nodes p_k of a possible partition of the continuous physical domain Ω . On its discrete counterpart, Ω^h , a set of $k = 0, \dots, N$ known analytic test functions ψ_k are defined (The two terms *test* and *basis* will be used interchangeably. Basis comes from the properties of ψ in the context of function spaces). The unknown coefficients \hat{q}_k correspond to the physical values of q at node p_k . The finite difference method is conceptually different in that what is approximated in the differential problem are the differential operators and not the solution variable. Substitution of (3.2) into (3.1) is such that $\mathcal{L}(q^h) - f \neq 0$. The method is called *method of weighted residuals* because a linear system of algebraic equations in the unknowns \hat{q} is built by imposing that

$$\int_{\Omega} w R d\Omega = 0, \quad (3.3)$$

where $R = \mathcal{L} - f$ is the residual of (3.1) and w is the weight function that has certain properties. Different methods arise from the selection of different w . The Galerkin

method is found when $w = \psi_k$. We can then write the following:

$$\int_{\Omega} \psi [\mathcal{L} - f] d\Omega = 0, \quad (3.4)$$

This is the weak form of the original equation to be solved.

Remark 3.1. So far, no distinction between the finite and spectral element methods has been made. The difference stems from the definition of ψ_k and will be reported shortly.

3.1.1 Suitable function spaces

Not every basis function ψ is accepted for the Galerkin formulation of a differential problem to make sense. In the specific cases of the advection-diffusion equation and the Euler equations of compressible flows, the highest order of derivation is 2, and the choice of the basis functions and the space to which they belong must depend on this regularity condition. The first step to take in the construction of the Galerkin method after defining the weak form of the original differential equation (e.g. Eq. (3.4)) is integration by parts to eliminate the second derivatives and hence impose a lower regularity on the solution variable (Quarteroni, 2009). In the simple case where $\mathcal{L}(q) \doteq \nabla \cdot (\nu \nabla q) = 0$ is the Laplace equation for diffusion of q in a medium with diffusivity ν , integration by parts is such that

$$\int_{\Omega} \psi \nabla \cdot (\nu \nabla q) d\Omega = - \int_{\Omega} \nu \nabla \psi \cdot \nabla q d\Omega, \quad (3.5)$$

where the assumption $\psi(\partial\Omega) = 0$ was used given Dirichlet boundary conditions on q . To understand under what conditions the integral on the right-hand side of Eq. (3.5) is defined, we first need to define the space of functions v that are Lebesgue integrable up to power $p = 1, \dots, \infty$ as:

$$L^p(\Omega) \doteq \{v : \Omega \rightarrow \mathbb{R} \quad \text{s.t.} \quad \|v\|_{L^p} = \left(\int_{\Omega} |v|^p d\Omega \right)^{\frac{1}{p}} < \infty\}. \quad (3.6)$$

We also need to refer to the Cauchy-Schwarz inequality according to which, given two functions $u, v \in L^2$, we have that

$$\int_{\Omega} |uv| d\Omega \leq \|u\|_{L^2} \|v\|_{L^2}. \quad (3.7)$$

It was stated within the definition of L^p that all the norms $\|\cdot\|_{L^p}$ are bounded. (3.6) and (3.7) simply imply the boundedness of $\int_{\Omega} |uv| d\Omega$, which is equivalent to saying that $uv \in L^1(\Omega)$. These steps are a very simple analysis of the least regularity requirement for the product $\nabla\psi \cdot \nabla q$. This requirement is fulfilled if $u = \nabla q$ and $v = \nabla\psi$ belong

to $L^2(\Omega)$. This means that we need to define a space where not only the functions, but their first derivatives are square integrable as well. We define:

$$H^1(\Omega) \doteq \{v \in L^2(\Omega) \quad \text{s.t.} \quad \frac{\partial v}{\partial x_j} \in L^2, j = 1, \dots, d\}. \quad (3.8)$$

The space W of test functions ψ and trial solutions q of problem (3.4), is a subset of H^1 such that

$$W \doteq \{\psi, q \in H^1(\Omega) \quad \text{s.t.} \quad \psi = 0 \text{ and } q = g \text{ on } \partial\Omega\}. \quad (3.9)$$

The previous analysis is far from being exhaustive but more of it would fall beyond the scope of this work. Nevertheless, the basic definitions reported so far are necessary to set the foundations of the Galerkin methods that will be used throughout.

3.1.2 Finite and Spectral Elements: discretization and basis functions

To discretize the problem in a finite and spectral element sense, the domain Ω is first decomposed into a finite element partition $\mathcal{P}^h = \{K^i\}_{i \in \ell=1, \dots, n_{el}}$ of n_{el} conforming² elements K^i such that

$$\Omega = \bigcup_{i \in \ell=1}^{n_{el}} K^i, \quad \text{and} \quad \bigcap_{i \in \ell=1}^{n_{el}} K^i = \emptyset, \quad (3.10)$$

where every element K^i is the image of the reference element $\mathbf{I} = [-1, 1]^d$ by a non-singular bijective mapping $\mathbf{x} = \mathcal{H}^i(\boldsymbol{\xi})$ from physical space \mathbf{x} to computational space $\boldsymbol{\xi}$. $\mathbf{J} = d\mathbf{x}/d\boldsymbol{\xi}$ is the transformation Jacobian matrix. A two-dimensional example of mapping is represented in Fig. 3.1.

The need for mapping is purely practical and forms the foundations of finite element computation. For details see Hughes (2000).

Basis functions: Finite Elements. Due to the properties of Lagrange polynomials, Lagrange basis functions are a common choice in finite elements. These functions, defined by h_k from now on, have the property of being piecewise continuous and are such that

$$h_k(x_l) = \delta_{kl} \quad k, l = 0, \dots, N,$$

where δ_{kl} is the Kronecker delta.

In the case of linear Lagrange polynomials, this translates into piecewise linear functions. This applies for any space dimension. The basis functions, in practice, are constructed only once on the reference element. At the moment of computing the integrals by a suitable quadrature rule, the mapping to physical space is computed.

²Being conforming is not a requirement as far as the construction of Galerkin methods goes. It is, however, a constraint if the code cannot treat hanging nodes.

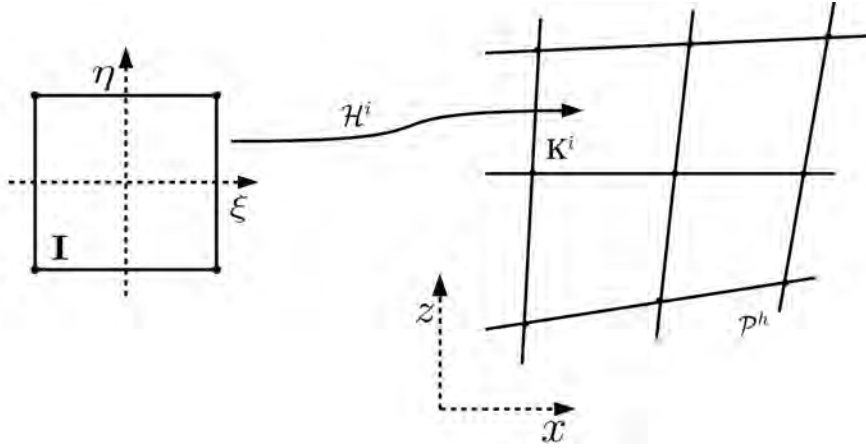


Figure 3.1: Mapping from reference, (ξ, η) , to physical space, (x, z) . $\forall K^i \in \mathcal{P}^h : K^i = \mathcal{H}^i(I)$.

For linear, quadratic, and cubic finite elements, the roots of the basis function along the reference element \mathbf{I} are the $N+1$ equi-spaced nodes within the element. Using the definition of the Lagrange polynomials

$$h_k(\xi) = \prod_{l=0, l \neq k}^N \frac{\xi - \xi_l}{\xi_k - \xi_l}, \quad (3.11)$$

in Fig. 3.2 we plot h_k along a reference element up to 2nd-order. A 4th-order finite element and corresponding basis function are plotted in Fig. 3.3 (left).

Basis functions: Spectral Elements. Unlike the case of high-order finite elements, the polynomials used with spectral elements are associated with zeros that are not equi-spaced. A classical and convenient set is represented by the Legendre-Gauss-Lobatto (LGL) points. LGL nodes ξ_i are the roots of

$$(1 - \xi^2)P'_N(\xi) = 0, \quad (3.12)$$

being $P_n(\xi)$ the N^{th} -order Legendre polynomial whose construction by recursive formulas can be found in [Karniadakis and Sherwin \(1999\)](#). The polynomials that are used have the same δ -property of the Lagrange polynomials defined above. Their analytic expression is given by

$$h_k(\xi) = \frac{(\xi^2 - 1)P'_N(\xi)}{N(N+1)(\xi - \xi_k)P_N(\xi)}, \quad k = 0, \dots, N, \quad (3.13)$$

where P' indicates derivation with respect to \mathbf{x} . The 4th-order k -polynomials along $\mathbf{I} = [-1, 1]$ are plotted on the right panel of Fig. 3.3. The comparative plot (finite

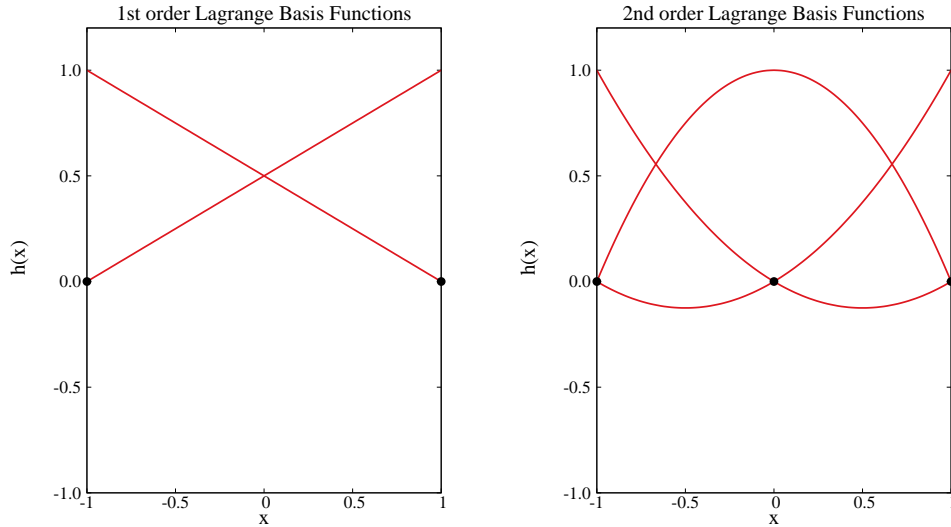


Figure 3.2: Lagrange polynomials of order 1 (left) and 2 (right) along the 1D reference element $\mathbf{I} = [-1, 1]$. Clearly, they are equivalent for FE and SE.

element on the left and spectral element on the right) is used to show that, if high-order is required, equi-spaced nodes produce unsatisfactory types of basis functions in the proximity of the edge points of the element. In other words, we lose control on the maximum and minimum values of h_k at the extrema of the element. When this happens, interpolation of any function is likely to suffer from such condition. To show how this feature translates into the interpolation of a known analytic function, we use the following example from [Giraldo \(2011\)](#). We define the *Witch of Agnesi* of unitary height as

$$z(x) = \frac{1}{1.0 + 50x^2},$$

where $z(x)$ is smooth and continuous, and interpolate it using the basis functions $\psi(x) = h_k(x)$ defined above. The test is performed by 4th-order interpolation. Equi-spaced and non equi-spaced points are used along the unitary domain. Fig. 3.4 shows how, the more the polynomial order is increased, and the better the result is when LGL nodes are employed. Roughly speaking, this analysis serves as a practical way of showing one reason for the use of LGL points in high-order simulations rather than high-order elements with evenly distributed nodes.

Figure 3.5 is a schematic representation of two 4th-order elements in two dimensions.

3.2 Galerkin method and unbounded solutions

The straight numerical approximation of problems with dominating advection may result in unphysical oscillations in the solution. Galerkin methods represent no exception

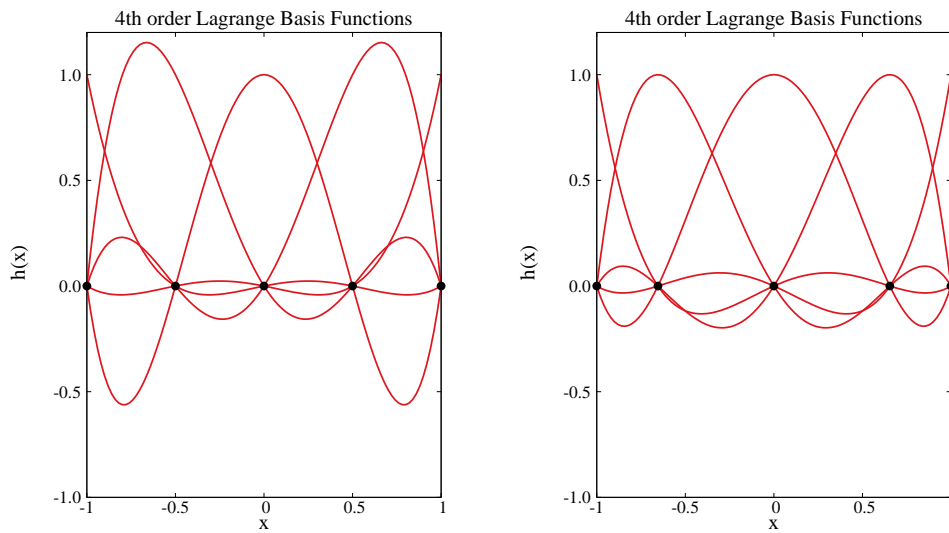


Figure 3.3: Basis of order 4 along the 1D reference element $\mathbf{I} = [-1, 1]$. Left: the nodes within the element are equi-spaced as for classical high-order FE. Right: Lagrange-Legendre polynomials of order 4 whose roots are the non-equi-spaced Legendre-Gauss-Lobatto (LGL) quadrature points. Nodal SE and DG may employ LGL or LG quadrature. However, to obtain a diagonal mass matrix then LGL is the only choice for SE, while LG can still be used for DG.

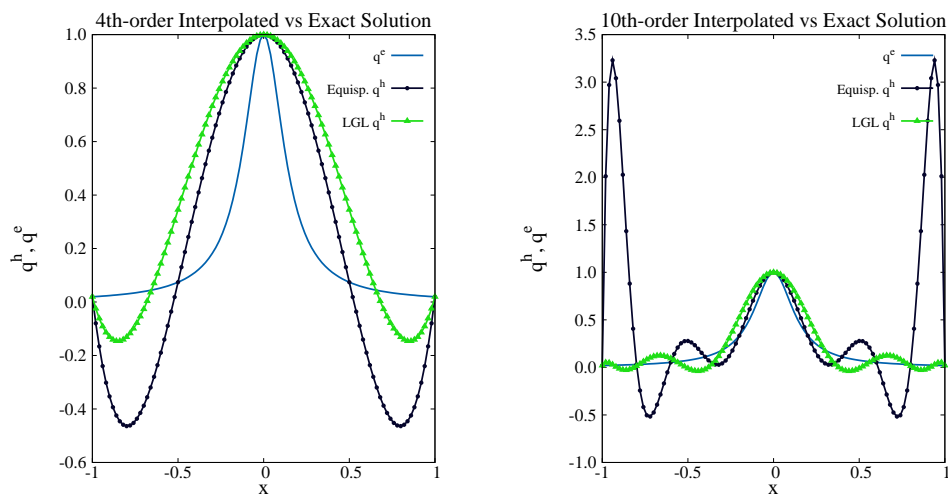


Figure 3.4: Interpolation of a known function (Witch of Agnesi) using high-order interpolating functions with equi-spaced and LGL points. Left: 4^{th} -order interpolation. Right: 10^{th} -order interpolation

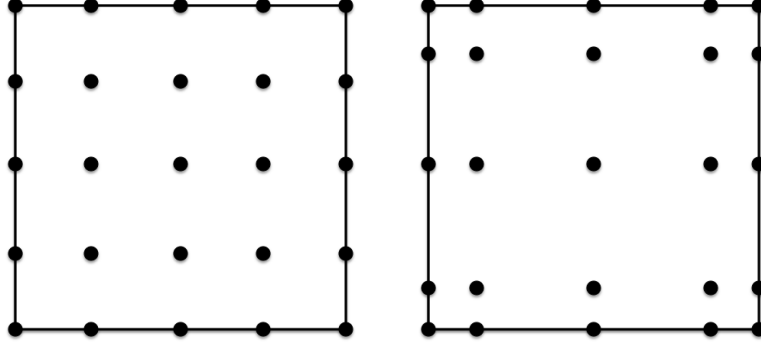


Figure 3.5: Nodes disposition for a two-dimensional 4th-order finite element (left), and spectral element (right).

(Johnson et al., 1984). An error estimate of the standard Galerkin approximation of the problem proves it (see, e.g., Quarteroni and Valli (1994)). Here, we show it by working out the 1D finite element solution of the advection-diffusion problem with Dirichlet boundary conditions. The problem is that of solving

$$\mathcal{L}(q) \doteq \mathbf{u} \cdot \nabla q - \nabla \cdot (\nu \nabla q) = f \quad (3.14)$$

with linear finite elements. In (3.14), $\nu > 0$ is a constant, uniform, diffusivity coefficient, $\mathbf{u} = (u, 0, 0)$ is the velocity vector, and f is a forcing function that, for simplicity, we set to zero for now. The domain of interest is the unitary interval $\Omega = [0, 1]$. A uniform partition \mathcal{P}^h of Ω in n_{el} elements of $k = 0, \dots, N$ nodes with coordinates \mathbf{p}_k and length $h = \|\mathbf{p}_k - \mathbf{p}_{k-1}\|_2$ is assumed. For uniqueness of solution, $q(0) = 0$ and $q(1) = 1$ are the assigned boundary conditions. Let $W^h \subseteq H^1$ be the space of piece-wise linear Lagrange polynomials of class \mathcal{C}^0 (Fig. 3.2, left.)

Projection of Eq. (3.14) onto W^h by the L^2 scalar product yields the discrete weak problem

$$\int_{\Omega^h} \psi^h \mathbf{u} \cdot \nabla q^h d\Omega^h + \int_{\Omega^h} \nu \nabla \psi^h \cdot \nabla q^h d\Omega^h = 0 \quad \forall \psi^h \in W^h, \quad (3.15)$$

where \mathbf{q}^h and ψ^h are the projection of \mathbf{q} and ψ onto W^h . q^h is expanded by (3.2). Skipping the algebra to build the linear system explicitly, the 1D finite element discretization of (3.15) yields the discrete equation

$$\left(\frac{u}{2} - \frac{\nu}{h}\right) \hat{q}_{k+1} + \frac{2\nu}{h} \hat{q}_k - \left(\frac{u}{2} + \frac{\nu}{h}\right) \hat{q}_{k-1} = 0, \quad k = 1 \dots, N-1 \quad (3.16)$$

(3.16) is equivalent to the 1D finite difference discretization of the same problem by centered differentiation. After algebraic manipulation and the definition of the local

Péclet number³

$$Pe = \frac{u h}{2\nu}, \quad (3.17)$$

(3.16) is a function of Pe :

$$(Pe - 1) \hat{q}_{k+1} + 2\hat{q}_k - (Pe + 1) \hat{q}_{k-1} = 0, \quad k = 1, \dots, N - 1. \quad (3.18)$$

It represents a tridiagonal linear system in the unknowns q_k , $k = 1, N - 1$, whose solution is the exponential function (see [Quarteroni et al. \(2000\)](#)):

$$\hat{q}_k = \frac{\left(\frac{1+Pe}{1-Pe}\right)^k - 1}{\left(\frac{1+Pe}{1-Pe}\right)^n - 1}, \quad k = 1, N - 1. \quad (3.19)$$

The power of $(1+Pe)/(1-Pe)$ at the numerator produces an oscillatory behavior of the solution whenever $Pe > 1$, as it is plotted in Fig. 3.6. Pe is a linear function of h so that the grid, in principle, could be always constructed in such a way that, for a given value of u and ν , $Pe \leq 1$. However, this is not viable for most real problems because of the extremely high number of grid points that may be necessary to achieve that condition. The only way to solve the problem of boundedness in the Galerkin solution of transport problems with dominant advection remains that of stabilization by proper means. Issues and their solution will be described in the framework of low Mach number flows in atmospheric dynamics first and transport then, in the following sections and chapters.

3.2.1 First steps towards stabilization

Artificial Viscosity (AV) ([Johnson, 1987](#)), *Streamline Upwind Petrov-Galerkin* (SUPG) ([Brooks and Hughes, 1982](#)), *Galerkin/Least-Squares* (GLS) ([Hughes et al., 1989](#)), Galerkin methods with bubble functions ([Brezzi et al., 1992](#); [Baiocchi et al., 1993](#); [Brezzi et al., 1997](#)), or sub-grid projection methods ([Guermont et al., 2006](#)) are some of the most used stabilization techniques for finite elements. The Taylor-Galerkin method ([Donea, 1984](#)), the Characteristic-Galerkin formulation ([Pironneau et al., 1992](#)), and the Characteristic-Based Split (CBS) method ([Zienkiewicz and Codina, 1995](#); [Zienkiewicz et al., 1999](#)) are more ways for FE stabilization that, however, rely on a reasoning that has no relationship with the methods used in this thesis. We mention them here but we will not delve into their description.

Artificial Viscosity/Diffusion and streamline-upwind (SU). AV is the most dated of stabilization methods. It is, however, still very common today for different reasons (see [Jablonski and Williamson \(2011\)](#)). It is based on the addition of

³ Pe gives a measure of the local advective against the viscous effects –Just like the Reynolds number does for the Navier-Stokes equations.

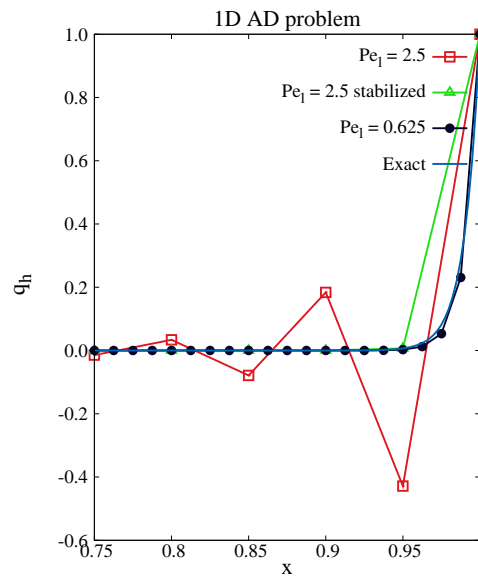


Figure 3.6: Finite element solution of the advection diffusion problem (3.14) using uniform, linear elements. $u = 10$, $\nu = 0.1$, in a domain of unitary total length. With these values, the global Péclet is $Pe_g = 50$. The plot shows the approximate solutions obtained for different grid spacing ($Pe = 2.5$ and $Pe = 0.625$) with and without stabilization. It is shown how the computed solution can approach the exact solution by either increasing the number of grid points ($Pe = 0.625$), or by maintaining the grid sufficiently coarse but with the addition of a stabilizing term (How this term is built has not been shown yet, but the result gives a hint on what to expect from it).

a viscosity-type term to the left-hand side of the discrete weak form of Eq. (3.14). The term, without rewriting the full equation, is simply

$$b_{AV} = \int_{\Omega_h} \tau \nabla \psi^h \cdot \nabla q^h d\Omega^h, \quad (3.20)$$

where, for the time being, we limit ourselves to say that τ is a diffusivity coefficient whose properties may be either uniform and constant or not. The addition of an artificial term such as (3.20) is a perturbation to the original equation. If the perturbation does not go to zero as the grid is refined, the method is not consistent. This simply means that the exact solutions of the original and of the perturbed problems are not equivalent when $h \rightarrow 0$. As we will show below (see Fig. 5.22 in Chapter 5), these methods add an uncontrolled and not-localized diffusion that yields a certain deterioration of the solution. The problem of isotropic smearing of the solution was partially solved by Hughes and Brooks (1982) with the construction of the Streamline-Upwind (SU) method. With SU, stabilization is projected in the direction of the flow only, as visible from

$$b_{SU} = \int_{\Omega_h} \tau \mathbf{u} \cdot \nabla \psi^h \mathbf{u} \cdot \nabla q^h d\Omega^h. \quad (3.21)$$

However, the method is not consistent either. The Streamline-Upwind/Petrov-Galerkin (SUPG) method described below is the consistent evolution of SU and will be among the most common methods of stabilization of finite elements used since its introduction.

Streamline-upwind/Petrov-Galerkin (SUPG). The SUPG method was designed by Brooks and Hughes (1982) and was later generalized for multidimensional problems by Hughes and Mallet (1986). It is a consistent alternative to the AV approach or to the overly diffusive SU. Its use has been ubiquitous in the solution of transport problems by the finite element method (e.g., Hughes and Tezduyar (1984); Franca et al. (1992); Brezzi et al. (1992); Tezduyar and Senga (2007)). The application of this strategy to higher-order schemes was first tested for spectral methods by Canuto and colleagues in Canuto (1994), Canuto and Puppò (1994), Canuto and Van Kemenade (1996), Canuto et al. (1998), and later by Hughes and coworkers in Hughes et al. (2005) using *non-uniform rational B-splines* (NURBS). Recently, it was applied to spectral elements in the context of atmospheric flows by Marras et al. (2012a) (see also Chapter 5).

SUPG is a Petrov-Galerkin method in that it does not assume that the basis and test functions live in the same space. We introduce the additional space Ψ^h of test functions w^h defined by

$$\Psi^h \doteq \left\{ w^h : w^h = \psi^h + \tau \mathbf{u} \cdot \nabla \psi^h : \psi^h \in W^h \right\}.$$

We have the problem of finding the function $q^h \in W^h$ such that

$$\int_{\Omega^h} w^h \mathbf{u} \cdot \nabla q^h d\Omega^h - \int_{\Omega^h} w^h \nabla \cdot (\nu \nabla q^h) d\Omega^h = \int_{\Omega^h} w^h f^h d\Omega^h \quad \forall w^h \in \Psi^h. \quad (3.22)$$

Some algebra and rearrangement of (3.22) yields the problem of finding $q^h \in W^h$ such that

$$\underbrace{\int_{\Omega^h} \psi^h \mathbf{u} \cdot \nabla q^h d\Omega^h + \int_{\Omega^h} \nu \nabla \psi^h \cdot \nabla q^h d\Omega^h}_{\text{Galerkin}} + b_{SUPG} = \underbrace{\int_{\Omega^h} \psi^h f^h d\Omega^h}_{\text{Galerkin}} \quad \forall \psi^h \in W^h, \quad (3.23)$$

where

$$b_{SUPG} = \int_{\Omega} (\mathbf{u} \cdot \nabla \psi^h) \tau \underbrace{[\mathbf{u} \cdot \nabla q^h - \nabla \cdot (\nu \nabla q^h) - f]}_{\mathcal{L}(q^h)-f} d\Omega^h \quad (3.24)$$

is the consistent SUPG stabilizing term. In (3.24), $\mathbf{u} \cdot \nabla q^h - \nabla \cdot (\nu \nabla q^h) - f$ is the residual of (3.14) and τ is the stabilization parameter to be defined later.

Galerking/Least-square (GLS). A generalization of SUPG was obtained by Hughes et al. (1989) as

$$b_{GLS} = \int_{\Omega} \underbrace{[\mathbf{u} \cdot \nabla \psi^h - \nabla \cdot (\nu \nabla \psi^h)]}_{\mathcal{L}(\psi^h)} \tau \underbrace{[\mathbf{u} \cdot \nabla q^h - \nabla \cdot (\nu \nabla q^h) - f]}_{\mathcal{L}(q^h)-f} d\Omega^h. \quad (3.25)$$

In analogy with the findings of Douglas and Wang (1989) to stabilize the Stokes equation, a sign change in the Laplace term of the stabilizing term in the perturbed equation proved to yield better stabilization characteristics (more accurate results) than the original generalized SUPG (or GLS) method (Franca et al., 1992). In (3.25), for better properties, instead of using the differential operator \mathcal{L} , the method should use the negative part of the adjoint \mathcal{L}^* of the original operator \mathcal{L} . We have that the ultimate perturbation term of the original AD equation should be

$$b = - \int_{\Omega} \mathcal{L}^*(\psi^h) \tau \underbrace{[\mathbf{u} \cdot \nabla q^h - \nabla \cdot (\nu \nabla q^h) - f]}_{\mathcal{L}(q^h)-f} d\Omega^h, \quad (3.26)$$

where

$$\mathcal{L}^* = -\mathbf{u} \cdot \nabla - \nabla \cdot (\nu \nabla).$$

Based on what was learned on stabilization of the scalar advection-diffusion equation, researchers in fluid dynamics applied these methods and their evolution widely for the stabilization of fluid problems. Due to its high efficiency, robustness, and validity at all

Mach regimes, in this thesis we focus on the Variational Multiscale approach pioneered by Hughes (1995) and Hughes et al. (1998) and that we introduce in the next paragraph first, and enter into in Chapter 4 for the solution of the Euler equations.

Variational Multiscale Stabilization (VMS). In 1995 and 1996, groups of researchers lead by Hughes (Hughes, 1995) and Brezzi (Brezzi et al., 1996) proposed a theory to explain the reasons of instabilities and a new way to attack the problem. They concluded that the unresolved scales (the scales that cannot be captured by the computational grid) are responsible for the numerical instabilities of the Galerkin solution of the differential problem. The analysis, that continues with Hughes and Stewart (1996) and Hughes et al. (1998), forms the unifying theory of all stabilized finite element methods. According to this theory, stabilized methods are subgrid scale models where the unresolved scales are intimately related to the instabilities at the level of the resolved scales, and thus should be used in the construction of the stabilization term. More specifically, in the formulation of the discrete problem, the effects of the unresolved scales must be introduced by modeling them on the grid. Avoiding the details that will be described later, the multiscale description of the stabilization scheme relies on the splitting of the solution into a resolved, q^h , and a sub-grid, unresolved component, \tilde{q} , to give $q = q^h + \tilde{q}$. Details will be given in Chapters 4 and 5. These schemes are known as *Variational Multiscale Stabilization* (VMS) methods. Generally speaking, the stabilization term of VMS corresponds to b defined in (3.26). VMS has been extensively applied to the solution of the advection-diffusion-reaction equation (or systems of them), and to the solution of the Navier-Stokes equations for incompressible flows (e.g., Codina (2000) and on). To our knowledge, VMS for compressible flows today only appears in Rispoli and Saavedra (2006), Koobus and Farhat (2004), van der Bos et al. (2007), and Moragues et al. (2010) for Computational Fluid Dynamics (CFD) simulations where compressibility effects are large (i.e. $M \geq 0.3$). In Chapters 4 and 6 of this thesis we will use the method that was recently used in Marras et al. (2012c) to stabilize the Euler equations of stratified atmospheric flows at low Mach number, whereas, in Chapter 5 VMS will be applied to high-order spectral elements.

3.3 Summary and discussion

This chapter was an introductory analysis of the numerical framework used in this thesis. We introduced the concept of Galerkin methods to build the structure of finite and spectral elements to approximate partial differential equations. The main similarities and differences between the two schemes were underlined. Finally, we used a pragmatic approach to show why stabilization of the above mentioned methods is needed and what possible solutions could be used with the Galerkin approximation of the problems at hand. In particular, we want to underline a few facts on stabilization:

- The finite element solution of the Euler equations needs stabilization. The Variational Multiscale Stabilization briefly introduced above is at the basis of the next

chapter where the Euler equations of stratified atmospheric flows are solved by the finite element method.

- In Chapter 4, low-order Q1 finite elements will be used. Because high-order methods suffer from unwanted Gibbs oscillations, in Chapter 5 we test the same stabilization technique to treat this problem in the spectral element solution of the advection-diffusion equation. In the framework of high-order methods in general, and spectral elements in particular, we are only aware of the use of VMS in the work on large eddy simulations by Wasberg et al. (2009). Its application to the spectral element solution of the transport equation is the central topic of Chapter 5.

Once they are stabilized, the main advantages of element-based Galerkin methods (of which FEM and SEM represent two special cases) can be summarized as follows.

- Element-based methods is halo-free, which is one of the biggest advantages for efficient parallelization regardless of the order of accuracy of the scheme. The only information that is needed by each element is that of the shared nodes between neighboring elements. This makes the method fully local and highly parallelizable (Giraldo and Rosmond, 2004).
- Although quadrilateral elements are used in this work, finite elements are fully unstructured by construction and do not rely on specific characteristics of the grid. No dimensional splitting is used in that both the horizontal and vertical directions are not distinguished in the discretization process. In other words, the equations can be solved by the same numerical method in all space dimensions. The great advantage of this is that the code becomes totally free from the geometry of the grid. This approach is classical in CFD and was first applied in NWP by Bacon et al. (2000), Bonaventura (2000), and Giraldo and Restelli (2008) using finite volumes, finite differences, and high order spectral elements, respectively.
- The proposed algorithm is designed to cover compressible flow regimes in a wide range of Mach numbers. In Chapter 4, we specifically focus on the treatment of low to very-low Mach number flows typical of atmospheric dynamics using finite elements.

Chapter 4

Solution of the Euler equations of dry nonhydrostatic flows

The main topic of analysis of this chapter is the finite element approximation of the governing equations of dry dynamics. Because of the numerical instability of the finite element solution of hyperbolic problems (the case of the Euler equations), we stress the analysis on stabilization by means of variational multiscale methods. We also consider and briefly analyze the independence of finite elements with respect to the type of computational grid and geometry of the domain.

Finally, with a suite of 2D and 3D standard benchmarks, we assess the solution method to solve nonhydrostatic flows at low Mach numbers in stratified atmospheres.

4.1 Mathematical model of compressible flows

In this section, the Euler equations (2.11) are solved by the finite element method. Stabilization of the finite element approximation is achieved by VMS. The current numerical treatment is independent of the number of space dimensions. This is proved in Sections 4.6 and 4.7 by solving two- and three-dimensional problems alike.

To render the development of the numerical method simpler, we re-express set (2.11) in compact form as

$$\frac{\partial \mathbf{q}}{\partial t} + \mathcal{L}(\mathbf{q}) = \mathbf{f}(\mathbf{q}), \quad (4.1)$$

where \mathbf{q} is the vector of the unknowns, \mathcal{L} is the differential operator, and \mathbf{f} is the source vector. We have:

$$\mathbf{q} = \begin{bmatrix} \rho \\ \mathbf{U} \\ \theta \end{bmatrix}, \quad \mathcal{L}(\mathbf{q}) = \begin{bmatrix} \nabla \cdot \mathbf{U} \\ \nabla \cdot \left(\frac{\mathbf{U} \otimes \mathbf{U}}{\rho} + p' \mathbf{I} \right) \\ \frac{\mathbf{U}}{\rho} \cdot \nabla \theta \end{bmatrix}, \quad \mathbf{f}(\mathbf{q}) = \begin{bmatrix} 0 \\ -\rho' g \mathbf{e}_z \\ 0 \end{bmatrix},$$

The equation of state

$$p = p_0 \left(\frac{R \rho \theta}{p_0} \right)^\gamma$$

completes the system of $(d+2)$ equations in $(d+3)$ unknowns. The problem consists in finding $\mathbf{q}(\mathbf{x}, t)$ such that Eq. (4.1) with proper boundary and initial conditions is verified $\forall (\mathbf{x}, t) \in \Omega \times (0, t_f)$.

4.2 Numerical formulation

Set (4.1) is discretized in space using finite elements and in time using an explicit finite difference scheme. It is understood that the finite element approximation described in the previous chapter equally applies to the discretization of (4.1). After a brief refresher on notation, in this section we take it from there and stress on stabilization via VMS.

4.2.1 Finite element approximation

A partition $\mathcal{P}^h = \{K^i\}_{i=1, \dots, n_{el}}$ of Ω_h in n_{el} quadrilateral finite elements is taken. We consider the finite trial and basis functions space W^h associated with \mathcal{P}^h as the space of 1^{st} -order Lagrange polynomials. Let $\{p_k\}_{k=1, \dots, n_{nodes}}$ be the nodes of the grid and ψ_k^h the Lagrange polynomial associated with node p_k , then $\{\psi_k^h\}_{k=1, \dots, n_{nodes}}$ is a basis for W^h . As before, we project Eq. (4.1) onto W^h by the L^2 scalar product. We hence obtain the weak (or variational) equation

$$\int_{\Omega^h} \psi^h \frac{\partial \mathbf{q}}{\partial t} d\Omega^h + \int_{\Omega^h} \psi^h \mathcal{L}(\mathbf{q}) d\Omega^h = \int_{\Omega^h} \psi^h \mathbf{f}(\mathbf{q}) d\Omega^h \quad \forall \psi^h \in W^h. \quad (4.2)$$

We define \mathbf{q}^h as the projection of \mathbf{q} onto W^h and expands it as

$$\mathbf{q}^h(\mathbf{x}, t) = \sum_{k=1}^{n_{nodes}} \psi_k^h(\mathbf{x}) \hat{\mathbf{q}}_k^h(t), \quad (4.3)$$

where $\hat{\mathbf{q}}_k^h(t)$ is the unknown value of \mathbf{q}^h at node p_k and time t . (To keep notation simple, the hat symbol will be omitted from now on unless otherwise needed for clarity).

4.2.2 Stabilization by the Variational Multiscale Method

In the framework of VMS, the solution vector is split into a grid scale and a sub-grid scale component as:

$$\mathbf{q} = \mathbf{q}^h + \tilde{\mathbf{q}}, \quad (4.4)$$

where $\tilde{\mathbf{q}} = (\tilde{\rho}, \tilde{U}, \tilde{W}, \tilde{\theta})^T$ represents the sub-grid scales (i.e.: the component of the solution that, unlike \mathbf{q}^h , cannot be resolved by the computational grid). See *Subgrid-scale approximation* below for details. The split (4.4) yields the reformulation of Eq. (4.2) as

$$\begin{aligned} & \int_{\Omega^h} \psi^h \frac{\partial \mathbf{q}^h}{\partial t} d\Omega^h + \int_{\Omega^h} \psi^h \mathcal{L}(\mathbf{q}, \mathbf{q}^h) d\Omega^h + \int_{\Omega^h} \psi^h \frac{\partial \tilde{\mathbf{q}}}{\partial t} d\Omega^h + \int_{\Omega^h} \psi^h \mathcal{L}(\mathbf{q}, \tilde{\mathbf{q}}) d\Omega^h \\ &= \int_{\Omega^h} \psi^h \mathbf{f}(\mathbf{q}^h) d\Omega^h + \int_{\Omega^h} \psi^h \mathbf{f}(\tilde{\mathbf{q}}) d\Omega^h \quad \forall \psi^h \in W^h. \end{aligned} \quad (4.5)$$

where, due to the non-linearity of the operator \mathcal{L} , the notation $\mathcal{L}(\cdot, \cdot)$ in (4.5) indicates that $\mathcal{L}(\mathbf{q}) \neq \mathcal{L}(\mathbf{q}^h) + \mathcal{L}(\tilde{\mathbf{q}})$.

Different possible linearization strategies are possible. We choose the following:

$$\mathcal{L}(\mathbf{q}, \mathbf{q}^*) = \begin{bmatrix} \nabla \cdot \mathbf{U}^* \\ \nabla \cdot \left(\frac{\mathbf{U} \otimes \mathbf{U}^*}{\rho} + \frac{c_\sigma}{2} (\rho\theta)^{\gamma-1} (\theta\rho^* + \rho\theta^*) \mathbf{I} \right) \\ \frac{\mathbf{U}}{\rho} \cdot \nabla \theta^* \end{bmatrix}, \quad (4.6)$$

where $\mathbf{q}^* = (\rho^*, U^*, W^*, \theta^*)^T$ represents \mathbf{q}^h or $\tilde{\mathbf{q}}$ in (4.6), and \mathbf{q} is the part of the non-linear term that is given as a known quantity at every time-step. Note that $\mathcal{L}(\mathbf{q}, \mathbf{q}^h + \tilde{\mathbf{q}}) = \mathcal{L}(\mathbf{q}, \mathbf{q}^h) + \mathcal{L}(\mathbf{q}, \tilde{\mathbf{q}})$ and $\mathbf{f}(\mathbf{q}^h + \tilde{\mathbf{q}}) = \mathbf{f}(\mathbf{q}^h) + \mathbf{f}(\tilde{\mathbf{q}})$. The hypothesis of quasi-static subscales is used so that $\int_{\Omega^h} \psi^h \frac{\partial \tilde{\mathbf{q}}}{\partial t} d\Omega^h = 0$ in (4.5). For the sake of algorithmic simplicity, we make the approximations $\mathcal{L}(\mathbf{q}, \tilde{\mathbf{q}}) \approx \mathcal{L}(\mathbf{q}^h, \tilde{\mathbf{q}})$ and $\mathcal{L}(\mathbf{q}, \mathbf{q}^h) \approx \mathcal{L}(\mathbf{q}^h, \mathbf{q}^h) \rightarrow \mathcal{L}(\mathbf{q}^h)$. What this means is that we do not carry around the total variable although the rigorous approximation tells us to do so. Rather, we only consider \mathbf{q}^h when \mathbf{q} appears.

In (4.5), we group the terms in \mathbf{q}^h and $\tilde{\mathbf{q}}$ to obtain

$$\begin{aligned} & \int_{\Omega^h} \psi^h \frac{\partial \mathbf{q}^h}{\partial t} d\Omega^h + \int_{\Omega^h} \psi^h \mathcal{L}(\mathbf{q}^h) d\Omega^h - \int_{\Omega^h} \psi^h \mathbf{f}(\mathbf{q}^h) d\Omega^h \\ &+ \int_{\Omega^h} \psi^h \mathcal{L}(\mathbf{q}^h, \tilde{\mathbf{q}}) d\Omega^h - \int_{\Omega^h} \psi^h \mathbf{f}(\tilde{\mathbf{q}}) d\Omega^h = 0 \quad \forall \psi^h \in W^h. \end{aligned} \quad (4.7)$$

To eliminate the $\tilde{\mathbf{q}}$ derivatives in $\mathcal{L}(\mathbf{q}^h, \tilde{\mathbf{q}})$ (the convective term of the subgrid scales), we integrate it by parts and assume the resulting boundary term equal to zero, obtaining:

$$\boxed{\begin{aligned} & \int_{\Omega^h} \psi^h \frac{\partial \mathbf{q}^h}{\partial t} d\Omega^h + \underbrace{\int_{\Omega^h} \psi^h \mathcal{L}(\mathbf{q}^h) d\Omega^h - \int_{\Omega^h} \psi^h \mathbf{f}(\mathbf{q}^h) d\Omega^h}_{\text{Galerkin term}} \\ & - \underbrace{\int_{\Omega^h} \mathbf{b}(\psi^h, \mathbf{q}^h, \tilde{\mathbf{q}}) d\Omega^h - \int_{\Omega^h} \psi^h \mathbf{f}(\tilde{\mathbf{q}}) d\Omega^h}_{\text{Stabilization term}} = 0 \quad \forall \psi^h \in W^h \end{aligned}} \quad (4.8)$$

where

$$\mathbf{b}(\psi^h, \mathbf{q}^h, \tilde{\mathbf{q}}) = \underbrace{\nabla \psi^h \cdot \begin{bmatrix} \tilde{\mathbf{U}} \\ \frac{\mathbf{U}_h \otimes \tilde{\mathbf{U}}}{\rho_h} + \frac{c_o}{2} (\rho_h \theta_h)^{\gamma-1} (\theta_h \tilde{\rho} + \rho_h \tilde{\theta}) \mathbf{I} \\ \frac{\mathbf{U}_h \tilde{\theta}}{\rho_h} \end{bmatrix}}_{\text{From integration by parts of } \mathcal{L}(\mathbf{q}^h, \tilde{\mathbf{q}})} + \psi^h \begin{bmatrix} 0 \\ 0 \\ \nabla \cdot \left(\frac{\mathbf{U}_h}{\rho_h} \right) \tilde{\theta} \end{bmatrix}. \quad (4.9)$$

Equation (4.8) is the weak stabilized form of Eq. (4.1).

Subgrid-scale approximation. The subscales are approximated in every element as a function of the resolved scales as originally proposed by Hughes (1995) using:

$$\tilde{\mathbf{q}} = \tau \mathbf{R}(\mathbf{q}^h), \quad (4.10)$$

where $\mathbf{R}(\mathbf{q}^h) = -\mathcal{L}(\mathbf{q}^h) + \mathbf{f}(\mathbf{q}^h)$ is the residual of the governing equations of the system and $\tau = \text{diag}(\tau_\rho, \tau_U, \tau_W, \tau_\theta)$ is the stabilization diagonal matrix (τ is often referred to as *intrinsic time*). In Chapter 3 τ was simply mentioned but never defined. It is because its definition has a direct relation with the physical problem to be solved and could not be generalized beforehand. In the case of the coupled Euler equations, the physics is considered through the characteristic advective and sound velocities. The two scales are linked to the maximum allowed time-step through the Courant-Friedrichs-Lewy (CFL) condition (Courant et al., 1928). The acoustic is more restrictive than the advective CFL. We will not enter this topic anywhere in this thesis, but it is worthwhile mentioning that the approximation of the Euler equations by the anelastic (e.g. Durran (1989)) or pseudo-incompressible approximations (Durran, 2008; Nance and Durran, 1994) are possible options to bypass the problem of the explicit solution of acoustic waves and hence keep Δt sufficiently large. Many research and operational atmospheric codes, however, are based on the solution of the fully compressible Euler equations (see Tables

Table 4.1: Stabilization methods

Method	Stabilization term	$\mathbf{R}(\mathbf{q}^h)$
SUPG/GLS	$\int_{\Omega^h} \nabla \psi^h \mathbf{A}(\mathbf{q}^h) \cdot \tau \mathbf{R}(\mathbf{q}^h) d\Omega^h$	$\frac{\partial \mathbf{q}^h}{\partial t} + \mathbf{A} \nabla \mathbf{q}^h - \mathbf{f}(\mathbf{q}^h)$
VMS	$-\int_{\Omega^h} \mathbf{b}(\psi^h, \mathbf{q}^h, \tau \mathbf{R}(\mathbf{q}^h)) d\Omega^h - \int_{\Omega^h} \psi^h \mathbf{f}(\tau \cdot \mathbf{R}(\mathbf{q}^h)) d\Omega^h$	$-\mathcal{L}(\mathbf{q}^h) + \mathbf{f}(\mathbf{q}^h)$

1.1-1.3). Other ways to filter acoustics are the use of properly designed time-integration methods. Of these, we mention the split-explicit scheme (e.g. [Klemp et al. \(2007\)](#)) where the slow-modes (i.e. advection) are integrated with sufficiently large Δt . Within one advective Δt , smaller time-steps are used to integrate the acoustic modes. We also mention the semi-implicit ([Cullen, 1990](#); [Giraldo et al., 2009](#)) and semi-Lagrangian methods (see [Robert \(1982\)](#), [Benoit et al. \(1997\)](#), [Bonaventura \(2000\)](#), and [Restelli \(2007\)](#)) as the most commonly used in atmospheric models. Refer to [Durrant \(2011\)](#) for a recent review of this topic. We will discuss time integration below, but using a fully explicit time integrator, Δt and the smallest of τ 's are directly connected in the current formulation of the code. Finally, since the work of [Turkel \(1987\)](#), the use of preconditioners is another option to attack the problem from the point of view of the linear algebra of the system at hand. A preconditioner can be seen as a way of artificially slowing sound waves down by manipulating the eigenvalues of the matrix problem. Preconditioners will not be treated in this work.

We define τ element-wise as

$$\tau_\rho = \tau_U = \tau_W = \frac{1}{4} \left[\frac{\|\mathbf{u}^i\| + c^i}{h^i} \right]^{-1} \quad \text{and} \quad \tau_\theta = \frac{1}{4} \left[\frac{\|\mathbf{u}^i\|}{h^i} \right]^{-1}, \quad (4.11)$$

where c is the speed of sound and the superscript i indicates that the value is computed on element K^i . h^i is taken as the size of the longest edge of the physical element. τ_ρ , τ_U , and τ_W depend on the speed of sound (see, for instance, [Hughes and Tezduyar \(1984\)](#); [Le Beau et al. \(1993\)](#); [Rispoli and Saavedra \(2006\)](#)). In contrast, we observe better stabilization if τ_θ does not depend on c . There is clearly an algorithmic issue with τ_θ when $\mathbf{u} \rightarrow 0$. In this event, τ_θ is set to zero in the code.

It is important to see the local nature of the sub-scales that are meant to exist only where residuals are important. This, with non-constant values, marks the major difference with respect to artificial diffusion. Figs. 4.5, 4.7, and 4.11 show the spatial distribution and the values of $\tilde{\mathbf{q}}$ normalized with respect to the corresponding maximum value of \mathbf{q}^h computed in three of the benchmarks of Section 4.6. In Table 4.1, we summarize the form of the stabilization term. For comparison, SUPG for compressible flows is reported as well. With no viscosity, SUPG and GLS for the Euler equations are the same. In the definition of SUPG/GLS, \mathbf{A}_i , $i = 1, \dots, d$, are the Jacobian matrices (i.e. the derivative of the convective fluxes with respect to \mathbf{q}^h).

Stabilization b for every equation. The algebraic development, equation by equation, of the stabilization terms $b_{\rho,U,\theta}$ defined in (4.9) yields the expressions:

ρ equation: $[kg m^{-3} s^{-1}]$

$$b_\rho = \psi_x \tau_U R(U) - \tau_W \psi_z R(W) \quad (4.12)$$

U equation: $[kg m^{-2} s^{-2}]$

$$\begin{aligned} b_U &= u \psi_x^h \tau_U R(U) \\ &+ \frac{1}{2} \psi_x^h c_0 \rho (\rho\theta)^{\gamma-1} \tau_\theta R(\theta) \\ &+ \frac{1}{2} \psi_x^h c_0 \theta (\rho\theta)^{\gamma-1} \tau_\rho R(\rho) \end{aligned} \quad (4.13)$$

W equation: $[kg m^{-2} s^{-2}]$

$$\begin{aligned} b_W &= w \psi_z^h \tau_W R(W) \\ &+ \frac{1}{2} \psi_z^h c_0 \rho (\rho\theta)^{\gamma-1} \tau_\theta R(\theta) \\ &+ \frac{1}{2} \psi_z^h c_0 \theta (\rho\theta)^{\gamma-1} \tau_\rho R(\rho) \\ &\quad - \psi^h \rho \tau_\rho R(\rho) \end{aligned} \quad (4.14)$$

θ equation: $[kg m^{-3} K s^{-1}]$

$$b_\theta = \left(u \psi_x^h + w \psi_z^h \right) \tau_\theta R(\theta) \quad (4.15)$$

The residuals in (4.12)-(4.15) are the following:

$$R(\rho) = U_x + W_z \quad (4.16)$$

$$R(U) = \left(\frac{UU}{\rho} + p \right)_x + \left(\frac{UW}{\rho} \right)_z \quad (4.17)$$

$$R(W) = \left(\frac{WU}{\rho} \right)_x + \left(\frac{WW}{\rho} + p \right)_z + \rho g \quad (4.18)$$

$$R(\theta) = \frac{U}{\rho}\theta_x + \frac{W}{\rho}\theta_z \quad (4.19)$$

Note how ρ and p that are used within the stabilization term are the total variables, unlike the case of the physical system.

4.2.3 Time integration

Although we are fully aware of its limits when applied to atmospheric simulations, in this work we apply a linear explicit Euler time integration scheme. Furthermore, even by increasing the accuracy of the time discretization, because the method is first-order accurate in space, the leading order of accuracy will not be larger than one (see, e.g., [Gaberšek et al. \(2012\)](#)). Time discretization of (4.8) gives

$$\begin{aligned} \int_{\Omega^h} \psi^h \frac{\mathbf{q}^{h,n+1} - \mathbf{q}^{h,n}}{\Delta t} d\Omega^h &= - \int_{\Omega^h} \psi^h \mathcal{L}(\mathbf{q}^{h,n}) d\Omega^h + \int_{\Omega^h} \psi^h \mathbf{f}(\mathbf{q}^{h,n}) d\Omega^h \\ &+ \sum_{i=1}^{n_{el}} \int_{K^i} \mathbf{b}(\psi^h, \mathbf{q}^{h,n}, \tilde{\mathbf{q}}^n) dK^i + \int_{\Omega^h} \psi^h \mathbf{f}(\tilde{\mathbf{q}}^n) d\Omega^h \quad \forall \psi^h \in W^h, \end{aligned} \quad (4.20)$$

where $\tilde{\mathbf{q}}^n = \tau \mathbf{R}(\mathbf{q}^{h,n})$ and the time step Δt is computed from the critical time step Δt_{cr} defined as

$$\Delta t_{cr} = \min_{i=1, \dots, n_{el}} \left[\frac{\|\mathbf{u}^i\| + c^i}{h^i} \right]^{-1}.$$

From the definition $CFL \doteq C\Delta t/h$, where $C = \|\mathbf{u}^i\| + c$ is the flow characteristic speed, the value of CFL is given explicitly such that

$$\Delta t = CFL \Delta t_{cr}. \quad (4.21)$$

Unless otherwise stated, $CFL = 0.5$ for every simulation.

Remark 4.1. Δt is computed dynamically, at every time-step, as the smallest among all those computed element by element across the computational domain. $CFL = 0.5$ does not necessarily correspond to the maximum allowed for a stable solution. However, because the main topic of this thesis is related to the space discretization and its stability properties in terms of bounded finite element solution, we did not explore the limits of CFL in any way. From what we mentioned previously, either a split-explicit or implicit methods and preconditioning should be explored in the future.

If we consider that $\{\psi_k^h\}_{k=1,\dots,N=n_{nodes}}$ is a basis for W^h , equation (4.20) only needs to hold for all $\psi_k^h \in W^h$, $k \in \{1, \dots, N\}$, thus (4.20) is a system of $N(d+2)$ equations. Using (4.3) in Eq. (4.20), we obtain the discrete matrix problem

$$\mathbf{M} \frac{\mathbf{q}^{h,n+1} - \mathbf{q}^{h,n}}{\Delta t} = \mathbf{G}(\mathbf{q}^{h,n}) + \mathbf{b}(\mathbf{q}^{h,n}, \tilde{\mathbf{q}}^n), \quad (4.22)$$

where

1. \mathbf{q}^h has dimension $N(d+2)$.
2. \mathbf{M} is the global Mass matrix of dimension $N(d+2) \times N(d+2)$ obtained by assembly of the local element matrices

$$\mathbf{M}^{kl} = \int_{\Omega_{el}^h} \psi_k^h \psi_l^h d\Omega^h, \quad k, l \in \{1, \dots, n_{points}^{el}\}.$$

3. \mathbf{G} and \mathbf{b} are the Galerkin and stabilization vector terms, respectively. They are constructed by assembly of the local differentiation and stabilization element terms:

$$\mathbf{G}^{kl}(\mathbf{q}^{h,n}) = - \int_{\Omega^h} \psi_k^h \mathcal{L}_l(\mathbf{q}^{h,n}) d\Omega^h + \int_{\Omega^h} \psi_k^h \mathbf{f}(\mathbf{q}^{h,n}) d\Omega^h$$

$$\mathbf{b}^k(\mathbf{q}^{h,n}, \tilde{\mathbf{q}}^n) = \int_{\Omega^h} \mathbf{b}(\psi_k^h, \mathbf{q}^{h,n}, \tilde{\mathbf{q}}^n) d\Omega^h + \int_{\Omega^h} \psi_k^h \mathbf{f}(\tilde{\mathbf{q}}^n) d\Omega^h, \quad k \in \{1, \dots, n_{points}^{el}\}.$$

All integrals are approximated by Gaussian quadrature that, for a function f , writes:

$$\int_{K^i} f(\mathbf{x}) dK^i = \int_I f(\mathcal{H}^i(\xi)) |J^i(\xi)| d\xi \approx \sum_{k=1}^{n_{Gauss}} f(\mathcal{H}^i(\xi^k)) |J^i(\xi^k)| \omega^k, \quad (4.23)$$

where $I = [-1, 1]^d$ is the reference element and $|J^i|$ is the determinant of the Jacobian matrix $\mathbf{J} = \frac{d\mathbf{x}}{d\xi}$ of the space transformation. n_{Gauss} is the number of Gauss integration points ξ^k inside K^i and ω^k their weights.

Remark 4.2. To preserve hydrostatic equilibrium, computation of the functions at the Gauss points is described in Section 4.4.

Remark 4.3. \mathbf{M} is computed once before the time-loop because it only depends on functions that are time-independent. It is a sparse matrix with dominant diagonal that results from the assembly of the mass matrices of each finite element. Because it is not diagonal, the solution of (4.22) would require the solution of a matrix inversion problem at every time-step. To avoid it, the mass lumping technique is adopted. Briefly, mass lumping consists of diagonalizing \mathbf{M} in some way. For bi-linear elements we use $\mathbf{M} \leftarrow \mathbf{M}_L = \text{diag}(\check{m}_{ii})$ where

$$\check{m}_{kk} = \sum_{l=k-1}^{k+1} m_{kl},$$

with k, l indicating the matrix rows and columns. This diagonalization may produce a singular matrix for non-quadrilateral high-order elements. In that case, other methods are usually considered. On the other hand, we will see later that lumping is not needed with quadrilateral spectral elements because \mathbf{M} is diagonal by construction using the property of tensor-product basis functions. We show this in Chapter 5.

From Eq. (4.22), the value of $\mathbf{q}^{h,n+1}$ at each node of the computational grid is obtained as

$$\mathbf{q}^{h,n+1} = \mathbf{q}^{h,n} + \Delta t \mathbf{M}_L^{-1} \cdot \left(\mathbf{G}(\mathbf{q}^{h,n}) + \mathbf{b}(\mathbf{q}^{h,n}, \tilde{\mathbf{q}}^n) \right). \quad (4.24)$$

4.3 Boundary conditions

For the well-posedness of the numerical solution of the mathematical problem, the compressible Euler equations at subsonic regimes require the imposition of one boundary value at the outflow and two at the inflow of the domain. At solid walls, such as the terrain of the domain, no-penetration is modeled as usual.

The problem of boundary conditions in NWP is still an open field of research (Nickovic et al., 2011). At least as far as open boundaries are concerned. Details on this are reported below.

Open boundaries A perturbation of a stratified atmosphere is responsible for the formation of fast-moving gravity waves. The perturbation may be caused by a multiplicity of reasons. The presence of an obstacle in the flow (mountain) or the formation of convective clouds are two examples (see Fig. 6.4 at page 138). At short running times gravity waves do not affect the solution in the inner region and the solution remains unperturbed during a characteristic time that depends on the extension of the domain. However, if the simulation is left free to evolve in time and the outflow boundary is not properly modeled, the local perturbation at the subsonic exit boundary will propagate inward, hence affecting the stability and accuracy of the solution (LeVeque, 2002). The effect of gravity waves needs consistent treatment with respect to the imposition of the boundary conditions. In the finite domain of limited vertical and horizontal extensions, damping of gravity waves by a sponge-like layer defined inside the flow domain is a simple solution that has been used by most modelers for many years. However, this technique needs quite a good portion of the computational grid for it to be effective, with clear disadvantage in terms of computational cost. The part of the domain occupied by the absorbing layer has no meaning for the physical problem, while it still contributes to the number of algorithmic operations in the code execution. Although more sophisticated methods have been studied throughout the years (see Dea et al. (2009) and references

therein), the non-linearity of the Euler equations still sets a limitation to the correct treatment of the outgoing waves using non-reflecting boundary conditions (typically referred to as NRBC). One example is the characteristic-based method implemented by [Aubry et al. \(2010\)](#) that, however, was not designed for problems with important gravity wave effects (e.g. mountain wave problems).

Although the literature reports an heterogeneous set of absorbing boundary conditions and corresponding implementations, in this work we did not dig into this topic and, instead, selected a classical Rayleigh-type scheme based on [Durrant and Klemp \(1983\)](#). Efficiency is left for the sake of simplicity. The use of a Rayleigh-type absorbing layer has the sole purpose of allowing the solution to reach steady-state whenever it is necessary, without feeling the effects of reflecting gravity waves. In the area occupied by the sponge, the solution variables $(\rho_d, \mathbf{U}, \theta)$ are corrected as $\rho = \mathcal{D}(\rho - \rho_b)$, $\mathbf{U} = \mathcal{D}(\mathbf{U} - \mathbf{U}_b)$, $\theta = \mathcal{D}(\theta - \theta_b)$, where b indicates the value of the unknown at the nodes on the physical boundaries and \mathcal{D} is the damping coefficient built from two contributions: \mathcal{D}^g for the internal gravity waves and \mathcal{D}^a for the high-frequency acoustic waves; $\mathcal{D} = \mathcal{D}^g + \mathcal{D}^a$. We have:

$$\mathcal{D}^g(z) = \begin{cases} 0 & \text{for } z \leq z_d \\ \frac{\alpha}{2} \left[1 - \cos \left(\frac{z - z_d}{z_{top} - z_d} \pi \right) \right] & \text{for } 0 \leq \frac{z - z_d}{z_{top} - z_d} \leq \frac{1}{2} \\ \frac{\alpha}{2} \left[1 + \cos \left(\left(\frac{z - z_d}{z_{top} - z_d} - \frac{1}{2} \right) \pi \right) \right] & \text{for } \frac{1}{2} < \frac{z - z_d}{z_{top} - z_d} \leq 1, \end{cases} \quad (4.25)$$

and

$$\mathcal{D}^a(z) = \frac{1 - \tanh \left(\frac{z_{top} - z}{d_z} \right)}{\tanh \left(\frac{z_{top} - z}{d_z} \right)}, \quad (4.26)$$

where z_d and z_{top} are, respectively, the coordinates of the base of the absorbing layer along the vertical direction, and the coordinate of the top of the physical domain. α is a coefficient whose value may be different from simulation to simulation as a function of the size of the domain, of the obstacle in the event that there is one (e.g. for a mountain of half-width a , subject to a horizontal flow moving with velocity $\mathbf{u} = (\bar{u}, 0, 0)$, we could have $\alpha = 6\pi\bar{u}/(2a)$, where $2a$ is taken as the dominant horizontal wavelength), and of the flow velocity. d_z is a characteristic length that is much smaller of $z_{top} - z_d$ (see [Klemp and Lilly \(1978\)](#) or [Restelli \(2007\)](#) for details). The upper boundary is more critical than the lateral boundaries ([Restelli, 2007](#)). Following Restelli's experience on

this, the lateral boundary sponge parameter is taken as

$$\mathcal{D}(x) = \frac{1 - \tanh\left(\frac{x_{bdy}-x}{d_x}\right)}{\tanh\left(\frac{x_{bdy}-x}{d_x}\right)}, \quad (4.27)$$

where x_{bdy} is the coordinate of the physical boundary. d_x is the horizontal analogue of d_z .

4.4 Interpolation error and well-balanced discretization

Numerical integration by Gauss quadrature demands a linear interpolation of the integrand from the grid nodes to the Gauss points displaced inside the mapped element. In a problem where small perturbations from a reference state are the quantities of interest, if interpolation of the reference values is not done properly, the effort described above (page 15) to define a balanced state may result insufficient. This is because the error of interpolation of the reference state would be orders of magnitude larger than the approximated values of the perturbed variables. To maintain a balanced state up to machine precision, we opted for a technique that consists of computing the reference quantities explicitly at the Gauss points, while interpolating the deviation variables only.

Before proceeding any further, we test this methodology by the equilibrium test presented in [Botta et al. \(2004\)](#) and [Gassmann \(2010\)](#). A resting atmosphere over a steep mountain must remain still at all times. The homentropic atmosphere initially at rest evolves until $t_f = 56000 s$ (approx. 16 hrs). Although the numerical truncation error in the discretization of $\partial_z \bar{p} + g\bar{\rho}$ is maintained sufficiently small, on a machine of finite precision acceleration of the order of machine error are still expected. That of equilibrium above a steep topography is of relevant importance in NWP; most models based on terrain-following coordinates must rely on the filtering of the orography to maintain stability and preserve equilibrium. In this test, the steep topography is given by a *Witch of Agnesi*

$$h(x, z) = \frac{h_c}{1 + \left(\frac{x-x_c}{a_c}\right)^2}, \quad (4.28)$$

where $h_c = 2000 m$, $a_c = 800 m$, and $x_c = 0 m$ within the domain $[-8000, 8000] \times [0, 8000] m^2$. The grid consists of 64×32 quadrilateral elements for an average resolution equal to $250 m$ both horizontally and vertically. See Fig. 4.1.

The background environment is characterized by the *Brunt – Väisälä* frequency $N = 0.0195$ and defined as

$$N^2 = g \frac{\partial}{\partial z} \ln \bar{\theta}. \quad (4.29)$$

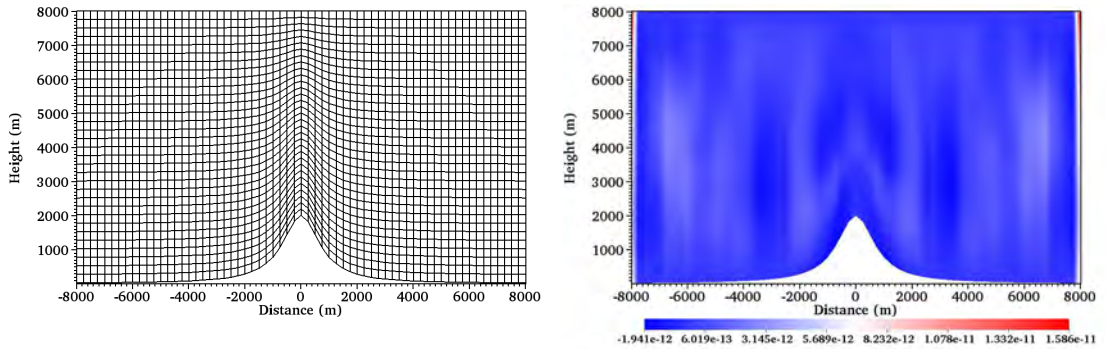


Figure 4.1: Hydrostatic equilibrium of an atmosphere at rest above steep topography. Left: computational grid. Right: filled contours of vertical velocity w at $t_f = 56000$ s. Vertical velocity: $-1e-12 \leq w \leq 1e-11$ ms^{-1} .

Integration of (4.29) along z yields

$$\bar{\theta} = \theta_0 e^{\frac{N^2}{g}z}, \quad (4.30)$$

where ground potential temperature $\theta_0 = 250$ K. From equations (2.6) and (2.10) and under the hypothesis of isothermal atmosphere, the background pressure is found as:

$$\bar{p} = p_0 e^{-\frac{g}{R\theta_0}z}. \quad (4.31)$$

$\bar{\rho}$ is derived from Eq. (2.6).

Fig. 4.1 shows the grid over a topography with maximum slope $\sigma \sim 60^\circ$ and the filled contours of vertical velocity after 16 hrs approximately. No-flux boundary conditions are imposed at the bottom and top boundaries, while periodicity is assigned to the lateral boundaries. The time-step is computed dynamically giving an average value of $\Delta t = 0.1$ s in this run. The distribution of vertical velocity is bounded in the range $-1e-12 \leq w \leq 1e-11$ ms^{-1} after $t_f = 56000$ s, with a symmetry pattern that is visible from the filled contours of the figure. This result is important because it shows that equilibrium is preserved with this simple technique; the same does not apply if the same exercise is run without the explicit computation of the reference state at the Gauss nodes. It also shows the ease of treatment of topography and the full independence of the method on the structure and shape of the computational grid.

4.5 Vertical discretization

What mentioned in Section 4.4 leads the discussion to the issue of vertical discretization in atmospheric models. Because of the classical use of finite differences with Cartesian rectangular grids, the accurate approximation of topography has always been a major concern both in atmospheric and ocean models. The vertical coordinate systems can be distinguished into two main branches: σ terrain-following (Phillips, 1957)

and height-coordinates. Terrain-following coordinates have the advantage of the accurate representation of topography and ease of application of boundary conditions as the grid cells follow the shape of the varying bottom of the domain. However, the large truncation errors that increase at increasing topography slope (Sundqvist, 1976; Janjic, 1989) suggested vertical coordinates that could be more suitable for steep topographies. The height-coordinate system was first proposed as the η -system by Mesinger et al. (1988). It consists of the use of a rectangular grid that intersects the topography and defines the orographic height at the cells edges. Modification of both approaches have been later defined. Examples are the hybrid terrain-following (Simmons and Burridge, 1981) as an improvement of σ , or the shaved-cell method in z-coordinates introduced by Adcroft et al. (1997) for ocean models. Fig. 4.2 shows a schematic of these grids.

Finite elements and Galerkin methods in general (finite volumes included) are free of all the drawbacks of methods that are not flexible about the grid. Finite elements depend on computational grids of quadrilateral and triangular elements (in 2D) or hexahedra and tetrahedra (in 3D) that adjust to the physical geometry to be discretized without affecting the formulation of the governing equations. The grid shape is inherently defined in the numerical formulation of the method. Generally speaking, they are z-coordinate based methods with full control of the shape of the topography. The grid itself looks like a σ -grid, but the fundamental difference is that finite difference methods with σ grids require re-expressing the equations using a coordinate transformation.

In the case of finite elements the boundary conditions are applied straightforwardly on the discretized geometry as it comes. The slope of the topography does not affect the solution (see test result in Fig. 4.1). Furthermore, in a time when high resolution is the rule, complex orographies can be modeled with ease with better grids. High resolution terrain-following coordinates induce grids to lose the property of orthogonality at the boundaries. The internal elements as well would suffer great stretching up to a point that the grid is no longer good for the numerical method to perform correctly. For example, in a finite difference discretization, if the Jacobian matrix of the transformation from physical to computational space is singular, large numerical errors and instability in the solution would occur (see, e.g., Thuburn (2011b)). In appendix A we report on the application of CFD grid generation techniques for the use of finite elements in atmospheric problems. We will explore the idea of quasi-orthogonal elliptic grids and their advantages in the study of turbulent boundary layers on small-scale simulations. The idea of multi-block grids is also described as a possible way to split the computational domain into a part that is fully rectangular and that can hence be optimal for finite difference models in the regions away from the topography.

4.6 2D Numerical tests

In the following sections, the FE-VMS algorithm presented in Section 4.2 is tested against a suite of seven standard tests used in dynamical core development. We divide the runs in two subsets according to the physics of the problems. In Section 4.6.1, *Numerical tests I*, we perturb the background atmosphere with thermal anomalies that vary in definition

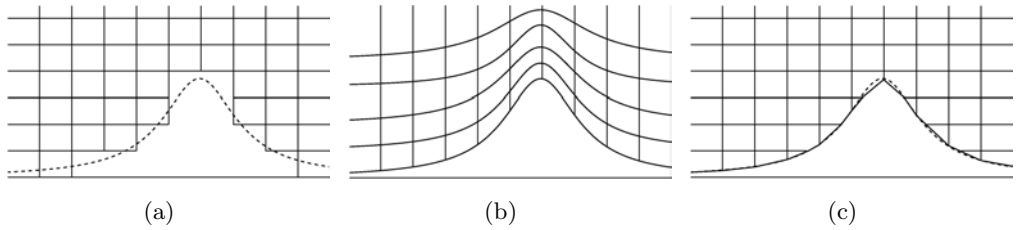


Figure 4.2: Representation of a smooth mountain using: (a) height coordinate system with step topography, (b) σ -terrain following coordinates, and (c) height coordinate system with shaved cells.

and size. These tests do not have analytic solution and the metrics that we use are based on comparison with the literature using symmetry considerations, front velocity of the moving thermal perturbation, and the amount of extrema. This set includes the rising thermal bubble in a large domain (Ahmad and Lindeman, 2007), the rising thermal bubble in a small domain (Robert, 1993), a modified density current of Straka et al. (1993), the density current of Haertel et al. (2001), and the inertia-gravity wave advected by a uniform velocity field in a uniformly stratified environment (Skamarock and Klemp, 1994). In Section 4.6.2, *Numerical tests II*, we solve two mountain problems that have semi-analytic solution based on the linear theory of small perturbations (Smith, 1979).

4.6.1 Numerical tests I: Thermally-induced flows

Given that the analytical solution does not exist, it must be understood that these tests can only give a qualitative (and relative) information on the accuracy that one model can achieve in the simulation of dynamic events in a low Mach environment.

Background state The background state is characterized by a neutral atmosphere with uniform potential temperature $\bar{\theta}$ and background pressure \bar{p} in hydrostatic equilibrium satisfying Eq. (2.10) such that

$$\bar{p} = p_0 \left(1 - \frac{g}{c_p \theta_0} z \right)^{c_p/R}, \quad (4.32)$$

where θ_0 is the potential temperature at the surface. The equation of state (2.6) is used to derive $\bar{\rho}$:

$$\bar{\rho} = \frac{p_0^{R/c_p}}{R\theta_0} \bar{p}^{c_v/c_p}. \quad (4.33)$$

Case 1: Warm bubble in a large domain

The convection of a warm bubble in a uniform environment has been widely used by different authors (e.g. Robert (1993); Janjic et al. (2001); Giraldo and Restelli (2008); Ahmad and Lindeman (2007)) to test their codes. Like Ahmad and Lindeman (2007)

after [Wicker and Skamarock \(1998\)](#), in *Case 1* a domain that extends within $[0, 20000] \times [0, 10000] m^2$ is defined. A large bubble of radius $r_0 = 2000 m$ and centered in $(x_c, z_c) = (10000, 2000) m$ is initially at rest and used to perturb the atmosphere at uniform $\theta = \theta_0 = 300 K$. The perturbation is given as a linear function of $R = \sqrt{(x - x_c)^2 + (z - z_c)^2}$ by

$$\theta' = \begin{cases} 0 & \text{if } R > r_o \\ A [1.0 - R/r_o] & \text{if } R \leq r_o, \end{cases} \quad (4.34)$$

where the oscillation constant is $A=2K$. The initial velocity field is zero everywhere. No-flux boundary conditions are set for all the boundaries.

Results Case 1. To compare directly against [Ahmad and Lindeman \(2007\)](#), the final time is set to $t_f = 1020 s$. We perform three runs on three different resolutions: 1) $\Delta x = \Delta z = 50 m$, 2) $\Delta x = \Delta z = 125 m$, and 3) $\Delta x = \Delta z = 250 m$. Fig. 4.3 shows the values of θ' and p' for the two finest grids. For θ' , the results qualitatively agree with those of the referenced article, where, however, pressure is not shown. However, quantitatively our results show a higher degree of diffusivity that can be quantified by the values in Table 4.2. A definitive construction of τ in VMS does not exist yet and a different definition could improve this results.

Values $\theta' < 0 K$ in Table 4.2 indicate the presence of local undershoots. In our case, a possible solution is the addition of a discontinuity capturing scheme to preserve monotonicity. This is not tested in this thesis. At equal resolutions velocity $\mathbf{u} = (u, w)$ matches well with the FV solution of [Ahmad and Lindeman \(2007\)](#). A relatively slower vertical propagation is observed in Table 4.2 for the solution on the coarsest grid.

Fig. 4.4 shows a horizontal slice of θ' along $z = 6000 m$ for three different resolutions. Symmetry with respect to the central axis is preserved, but we observe a loss of accuracy and sharpness at the coarsest grid.

Remark 4.4. In Table 4.2, the results relative to the *Advanced Research Weather Research and Forecasting* modeling system (WRF-ARW) ([Skamarock et al., 2007](#)) were obtained by Ahmad and Lindeman using fifth- and third-order FD in the horizontal and vertical directions, respectively, and adding artificial viscosity with diffusivity coefficient $K = 15 m^2 s^{-1}$ to preserve stability. The run of the *f-wave* solver in the same reference employed a second-order FV scheme.

Fig. 4.5 shows the normalized values and distribution of the subscales $\tilde{\mathbf{q}}$ on the domain. Normalization is done with respect to the maximum value of the solution variable at hand. The subgrid scales closely follow the structure of the solution (e.g. the warm bubble in this case). Because we would like for stabilization to only affect the regions of important gradients, this behavior is what we seek when modeling $\tilde{\mathbf{q}}$.

As expected from Eq. (4.10), their values are more important where the residual is large. Table 4.3 shows the extrema of $\tilde{\mathbf{q}}$ and \mathbf{q}^h .

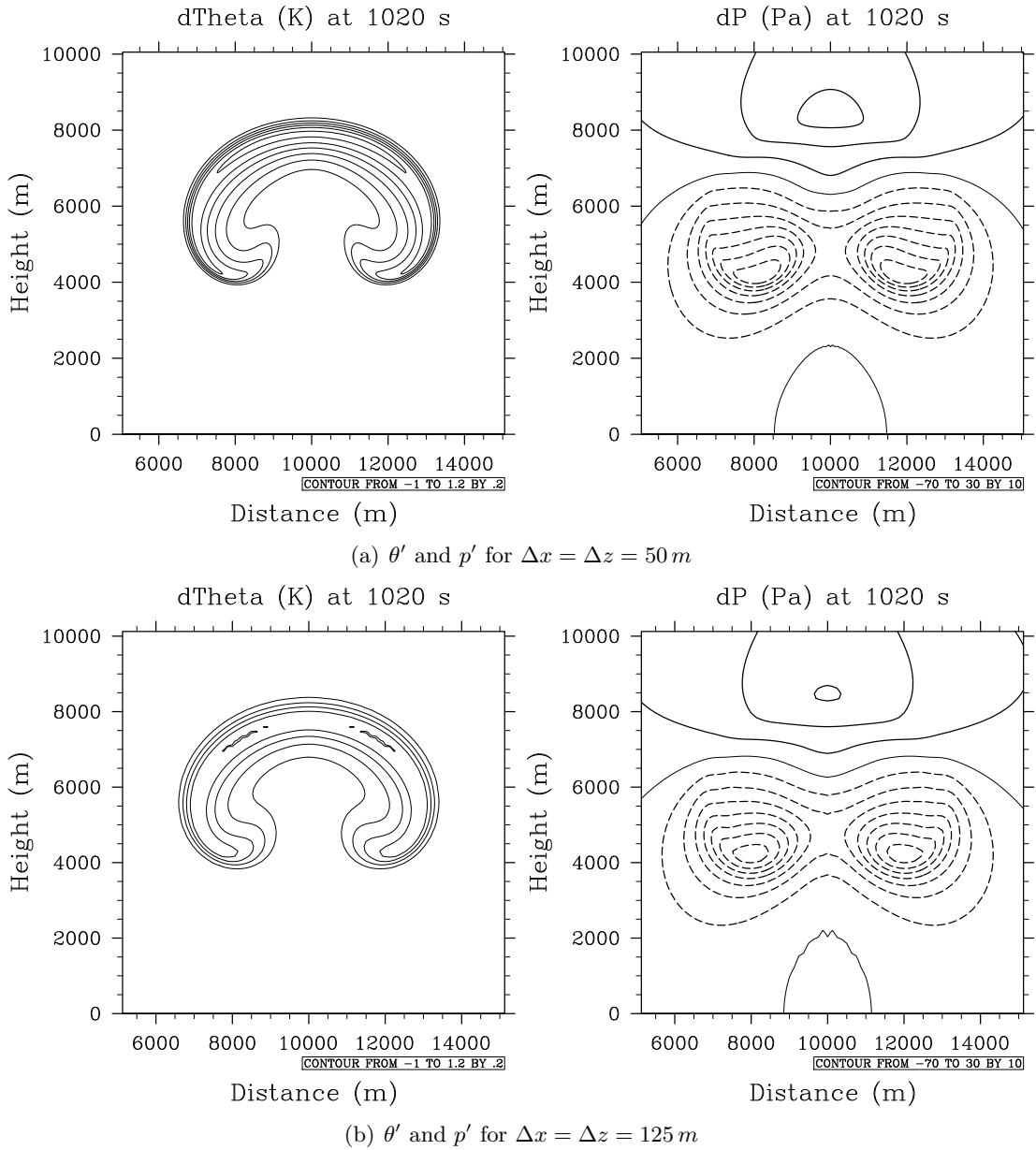


Figure 4.3: Case 1. θ' (left column) and p' (right column) contours at 1020 s on the finest grid with $\Delta x = \Delta z = 50\text{ m}$ (top row), and medium resolution grid with $\Delta x = \Delta z = 125\text{ m}$ (bottom row). Negative values are dashed.

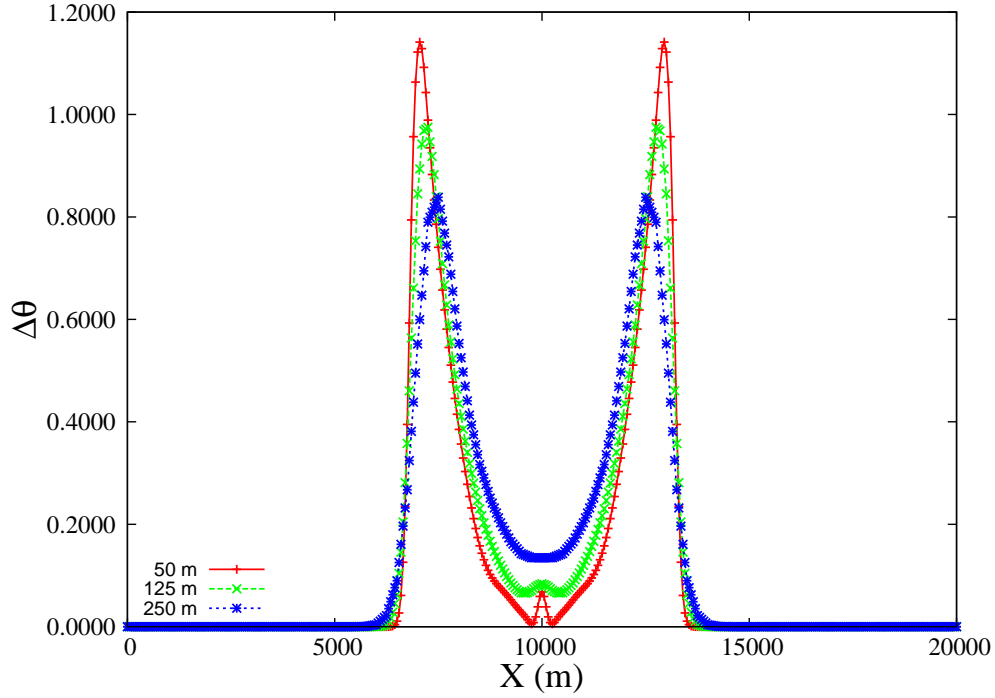


Figure 4.4: Case 1. Profiles of θ' along $z = 6000\text{ m}$ after 1020 s, at three different resolutions: $\Delta x = \Delta z = 50\text{ m}$, $\Delta x = \Delta z = 125\text{ m}$, and $\Delta x = \Delta z = 250\text{ m}$.

Table 4.2: Case 1. Comparative results of p' (Pa), θ' (K), and (u, w) (m s^{-1}) extrema at 1020 s. VMS (FE) at different resolutions, WRF-ARW (FD), and f-wave (FV) are compared. The values of WRF-ARW and f-wave decomposition in table are from [Ahmad and Lindeman \(2007\)](#). The dash symbol "-" indicates that data are not available.

Model	p'_{min}	p'_{max}	θ'_{min}	θ'_{max}	u_{min}	u_{max}	w_{min}	w_{max}
VMS (50 m)	-77.43	32.35	-1.38e-2	1.244	-10.21	10.21	-8.93	14.47
VMS (125 m)	-75.74	30.38	-1.08e-2	1.008	-8.903	8.903	-7.43	14.03
VMS (250 m)	-67.90	29.13	-1.53e-2	0.891	-7.18	7.18	-6.651	12.09
WRF-ARW (125m)	-	-	-6.0e-2	1.65	-5.05	5.05	-5.0	11.5
f-wave (125m)	-	-	-1.0e-2	1.39	-8.53	8.53	-7.75	13.95

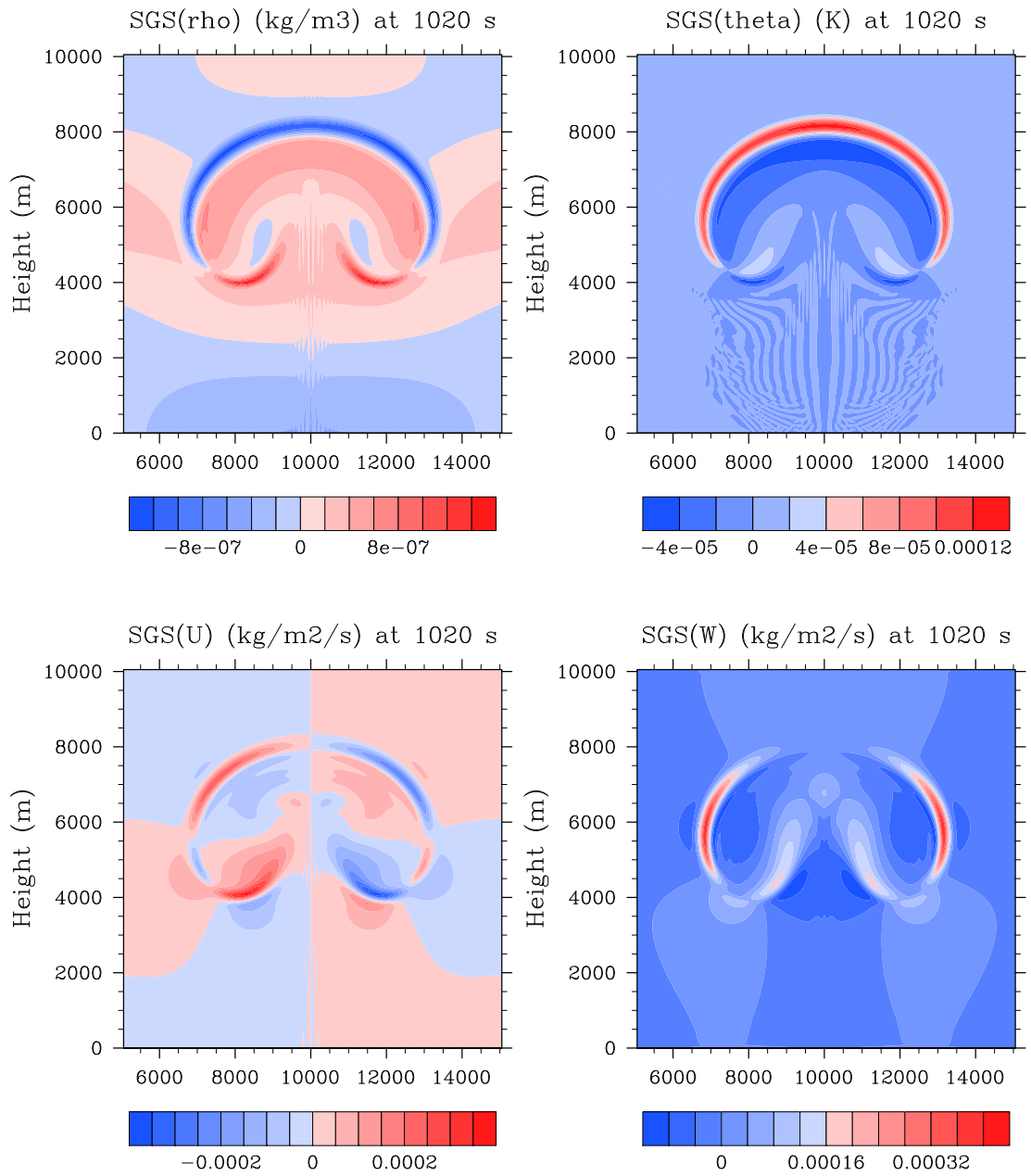


Figure 4.5: Case 1. Filled contours of the normalized subscales at $t_f = 1020$ s and $\Delta x = \Delta z = 50$ m resolution. Top row: $\tilde{\rho}/\rho_{max}$, $\tilde{\theta}/\theta_{max}$. Bottom row: \tilde{U}/U_{max} , \tilde{W}/W_{max} .

Table 4.3: Case 1. $\tilde{\mathbf{q}}$ and \mathbf{q}^h extrema at $t_f = 1020s$ and $\Delta x = \Delta z = 50 m$ resolution.

	ρ^h	$\tilde{\rho}$	U^h	\tilde{U}	W^h	\tilde{W}	θ^h	$\tilde{\theta}$
max.	1.161	1.736e-6	8.475	3.178e-3	11.05	4.429e-3	348.2	4.486e-2
min.	0.4343	-1.551e-6	-8.475	-3.178e-3	-6.095	-1.322e-3	130.3	1.883e-2

Case 2: Warm bubble in a small domain

Unlike Case 1, the thermal perturbation is defined by a Gaussian as in the original problem of Robert (1993), where the domain is smaller and extends within $[0, 1000] \times [0, 1500] m^2$. The thermal perturbation with amplitude $A = 0.5 K$, radius $r_o = 50 m$, and $\sigma = 100 m$ is initially centered in $(x_c, z_c) = (500, 260) m$, and is defined by the law:

$$\theta' = \begin{cases} A & \text{if } R \leq r_o \\ A e^{-(R-r_o)^2/\sigma^2} & \text{if } R > r_o \end{cases} \quad (4.35)$$

where the perturbation boundary is $R = \sqrt{(x - x_c)^2 + (z - z_c)^2}$. Initial velocity is zero everywhere. Zero normal velocity is imposed at the four boundaries to impose no-flux conditions.

Results Case 2. For comparison against Robert (1993), the final time is set to $t_f = 1080 s$. We perform four runs on four different resolutions: 1) $\Delta x = \Delta z = 2.5 m$, 2) $\Delta x = \Delta z = 5 m$, 3) $\Delta x = \Delta z = 10 m$, and 4) $\Delta x = \Delta z = 25 m$. The plots of θ' , p' , u , and w are shown in Fig. 4.6 for the finest resolution. The solution presents rotating structures qualitatively comparable to the result of the reference. However, the two wakes that originate from the upper thermal have a characteristic thickness that makes us think the solution to be more diffusive than the one presented in the original paper by Robert, where 2^{nd} order FD in space is used. In Case 1, the different initial distribution of θ' and the different ratio between the diameter of the bubble and the length of the domain do not allow these structures to form.

The only comparison possible for this test is qualitative because no numerical values are presented by any author. However, Table 4.4 shows the extrema obtained with different resolutions. As for all the tests to follow, we do not modify the set of equations with the addition of an artificial diffusivity term. The main purpose of this study is the analysis of the stabilization properties of VMS. This, and the non-linearity and non-steady nature of the thermal problems that we present are the cause of variation in the solution computed on different grids. This appears from Table 4.4, where p' , θ' , and (u, w) are reported. See *Density Current I* below for more on this.

Table 4.4: Case 2. Comparative results of p' , θ' , and (u, w) extrema at 1080 s for four different grid resolutions.

Model	p'_{min}	p'_{max}	θ'_{min}	θ'_{max}	u_{min}	u_{max}	w_{min}	w_{max}
VMS (2.5 m)	-5.31	1.57	-5.36e-2	3.64e-1	-2.61	2.61	-2.33	2.36
VMS (5 m)	-3.34	1.22	-7.01e-2	3.57e-1	-1.55	1.55	-2.38	1.72
VMS (10 m)	-3.05	1.06	-6.45e-2	3.11e-1	-1.58	1.58	-2.24	1.46
VMS (25 m)	-2.86	1.05	-3.07e-3	3.29e-1	-1.05	1.05	-2.12	1.26

Table 4.5: Case 2. $\tilde{\mathbf{q}}$ and \mathbf{q}^h extrema at $t_f = 1080s$ and $\Delta x = \Delta z = 2.5 m$ resolution.

	ρ^h	$\tilde{\rho}$	U^h	\tilde{U}	W^h	\tilde{W}	θ^h	$\tilde{\theta}$
max.	1.149	1.676e-7	2.906	2.656e-4	2.610	1.844e-4	303.5	1.464e-2
min.	1.015	-1.170e-7	-2.906	-2.656e-4	-2.576	-1.771e-4	303.1	-2.331e-2

Fig. 4.7 shows the normalized values and distribution of the subscales $\tilde{\mathbf{q}}$ on the domain. Their values are more important where residuals are bigger. The extrema of $\tilde{\mathbf{q}}$ and \mathbf{q}^h extrema are reported in table 4.5

Case 3: Density current I

After its introduction in different numerical (Carpenter et al., 1990) and experimental (Droegemeier and Wilhelmson, 1987) studies, Straka et al. (1993) documented in detail the density current problem as a standard benchmark in NWP. Like in Ahmad and Lindeman (2007), in this thesis Straka's test is modified and run without the artificial diffusion of diffusivity constant $K = 75 m^2 s^{-1}$ for two fundamental reasons: first, in this study we are interested in solving the Euler equations of inviscid flows, and second, we want to verify that VMS can indeed stabilize problems with dominant dynamics without the addition of any type of artificial viscosity. The initial base-state atmosphere is characterized by a uniform potential temperature $\bar{\theta} = \theta_0 = 300 K$ within the domain $[-25600, 25600] \times [0, 6400] m^2$. A perturbation of θ centered in $(x_c, z_c) = (0, 3000) m$ and

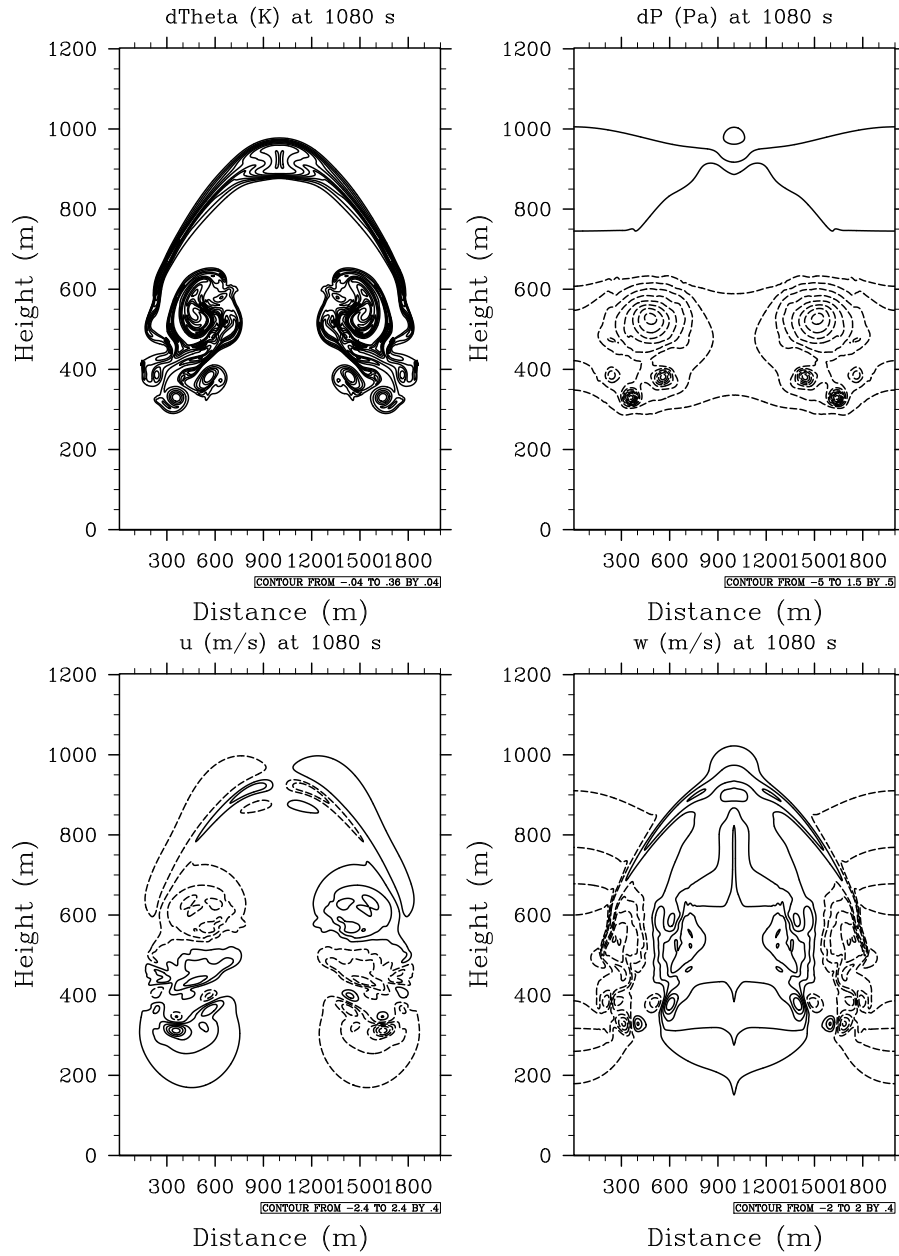


Figure 4.6: Case 2. Solution at $t_f = 1080$ s for grid resolution $\Delta x = \Delta z = 2.5$ m. Top row: θ' and p' contours. Bottom row: u and w contours.

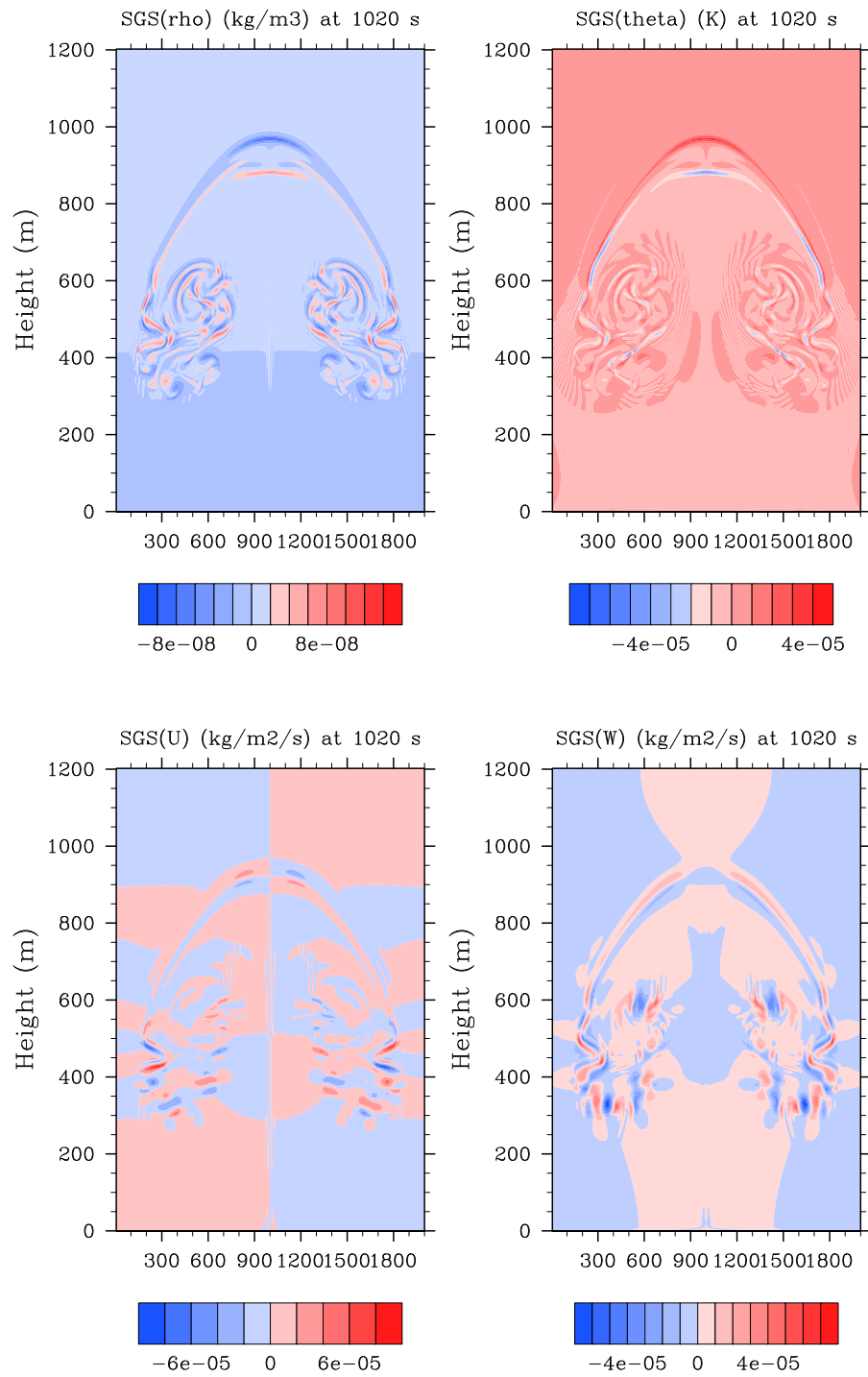


Figure 4.7: Case 2. Filled contours of the normalized subcales at $t_f = 1080$ s and $\Delta x = \Delta z = 2.5$ m resolution. Top row: $\tilde{\rho}/\rho_{max}$, $\tilde{\theta}/\theta_{max}$. Bottom row: \tilde{U}/U_{max} , \tilde{W}/W_{max}

with radii $(r_x, r_z) = (4000, 2000) m$ is given by the cosine distribution

$$\theta' = \begin{cases} 0 & \text{if } R > 1 \\ A \left[\frac{1 + \cos(\pi_c R)}{2} \right] & \text{if } R \leq 1 \end{cases} \quad (4.36)$$

where $A = -15 K$ is the perturbation amplitude, and R the analytical definition of the perturbation that goes as

$$R = \sqrt{\left(\frac{x - x_c}{r_x}\right)^2 + \left(\frac{z - z_c}{r_z}\right)^2}.$$

The four boundaries are defined as no-flux permitting walls. The initial velocity is zero.

Results Case 3. Fig. 4.8 shows the contours of θ' at $t_f = 900 s$ for four different resolutions: 1) $\Delta x = \Delta z = 25 m$, 2) $\Delta x = \Delta z = 50 m$, 3) $\Delta x = \Delta z = 75 m$, and 4) $\Delta x = \Delta z = 100 m$. The cold perturbation introduces a heavier pool of air whose downward motion is responsible for the development of the propagating front. Inertia causes the top layers of the front to pull back giving rise to Kelvin-Helmholtz structures.

Being inviscid, the result should be comparable to those of [Ahmad and Lindeman \(2007\)](#), where the solution is computed on a grid of size $\Delta x = \Delta z = 50 m$. At the same resolution, we observe the same number of rotating structures (see Fig. 4.8) and measure a difference of the front position normalized by the domain half-length equal to 0.0135 (see Table 4.6). We consider the front position defined as the last node on the ground where $\theta' = -1 K$. A comparison of the front positions is reported in Table 4.6. For the 50 m resolution, the front is at 14629 m at 900 s, giving a front speed of $16.25 m s^{-1}$. The front is slower on a coarser grid as it is also observed in Fig. 5 of [Straka et al. \(1993\)](#).

As the resolution is increased, the amount of structures that are resolved increases as well. The solution is inviscid, non-linear, and non-steady. Without viscosity, that could homogenize the solution by diffusive effects, we do not expect to reach space-convergence to a common solution. Rather, we expect more and more structures to be resolved until the smallest refinement of the order of the subgrid scales is reached. This goes way beyond the resolutions of interest in NWP and its analysis falls out of the scope of this work. In Fig. 4.8, oscillations in potential temperature appear in the wake of the density current. They are a consequence of the large inertia of the falling pool that, once it has reached the ground, abruptly changes direction of motion. The oscillations are partially dragged by the wake of the flow and stand still in the proximity of the lower layers of the domain. The small numerical dissipation that is added by VMS is not enough to suppress them. As previously mentioned, a localized discontinuity capturing method may be a viable solution.

Fig. 4.11 shows the normalized values and distribution of the subscales on the domain. Table 4.7 shows the extreme values of $\tilde{\mathbf{q}}$ and \mathbf{q}^h .

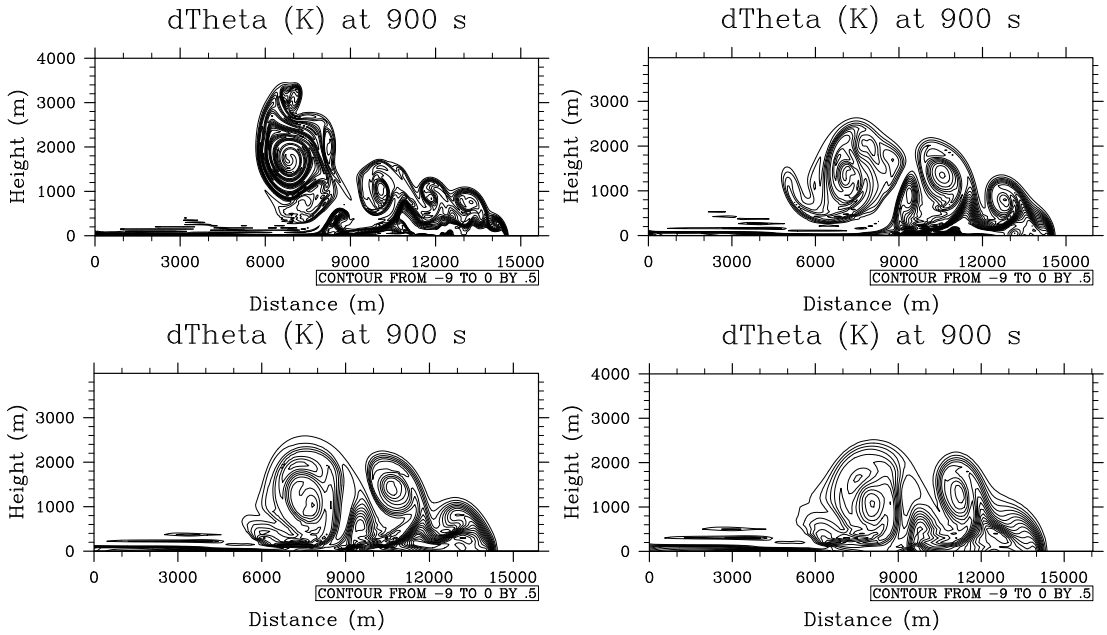


Figure 4.8: Case 3. Contours of θ' at $t_f = 900$ s for four different grid resolutions. Top row: $\Delta x = \Delta z = 25$ m and $\Delta x = \Delta z = 50$ m resolution. Bottom row: $\Delta x = \Delta z = 75$ m and $\Delta x = \Delta z = 100$ m resolution.

In Fig. 4.9 we reproduce the result by [Droegemeier and Wilhelmson \(1987\)](#) for a non-viscous density current generated by an initial step profile of θ' . The amount of structures increases as time evolves. A laboratory experiment of a density current with the intent of reproduce a thunderstorm outflow is shown in Fig. 4.10. It is a good visualization of what type of structures a real density current admits.

Case 4: Density current II

This is a test proposed by [Haertel et al. \(2001\)](#). They simulate an idealized thunderstorm outflow initialized as an instantaneous cooling of the lower layers of an isentropic atmosphere. The difference of this case with respect to *density current I* is in the geometrical definition of the cold region and in the quantitative analysis that can be extracted from the results. The cold pool is given by a uniform potential temperature $\theta = 297$ K within a semi-elliptic region centered in the middle of the domain $[-16000, 16000] \times [0, 3000]$ m² and extended for 10000 m in the x and 2000 m in the z directions. The outer atmosphere is isothermal in θ with a reference value $\bar{\theta} = 300$ K. The simpler shape of the outflow makes it easy to verify the kinematic relation that [Benjamin \(1968\)](#) gave between the velocity c of the leading edge and the depth d of the cold region a few meters upstream of the current. If *Density Current I* provides important information on the quality of the simulation, Benjamin's law sets a bridge towards the quantitative evaluation of the

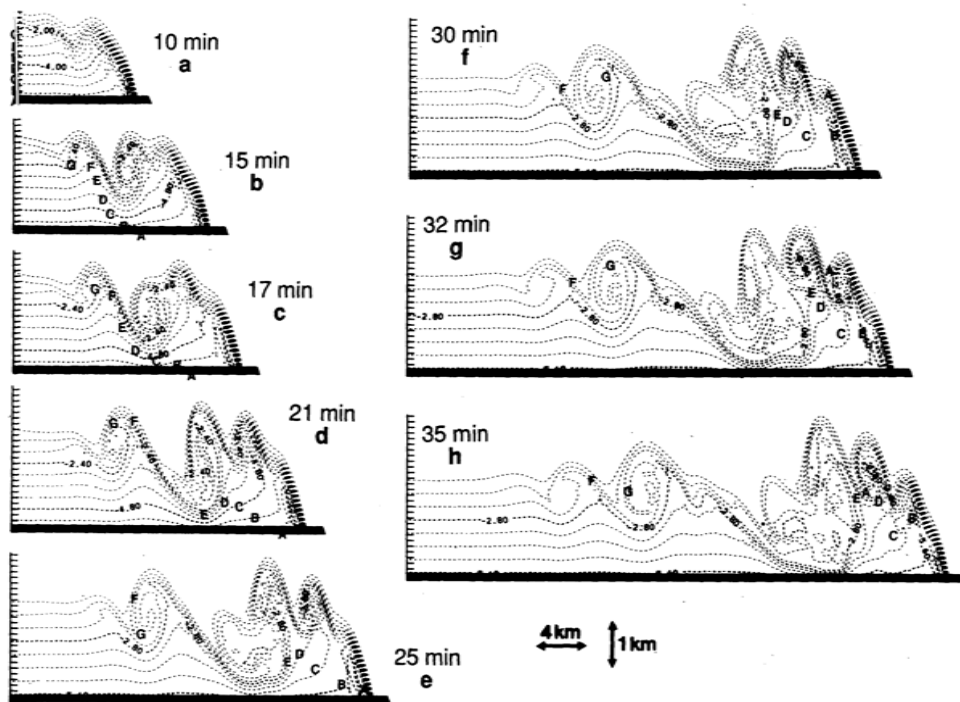


FIG. 13. A sequence describing the motion of seven air parcels (labeled A through G) in the interior of a simulated outflow. For reference, the perturbation potential temperature field is shown by the dashed contours (note the different vertical and horizontal scales). See the text for further details.

Figure 4.9: Case 3. Reference simulation of a non-viscous density current. From Fig. 13 of Droegemeier and Wilhelmson (1987), (c)American Meteorological Society. Reprinted with permission.

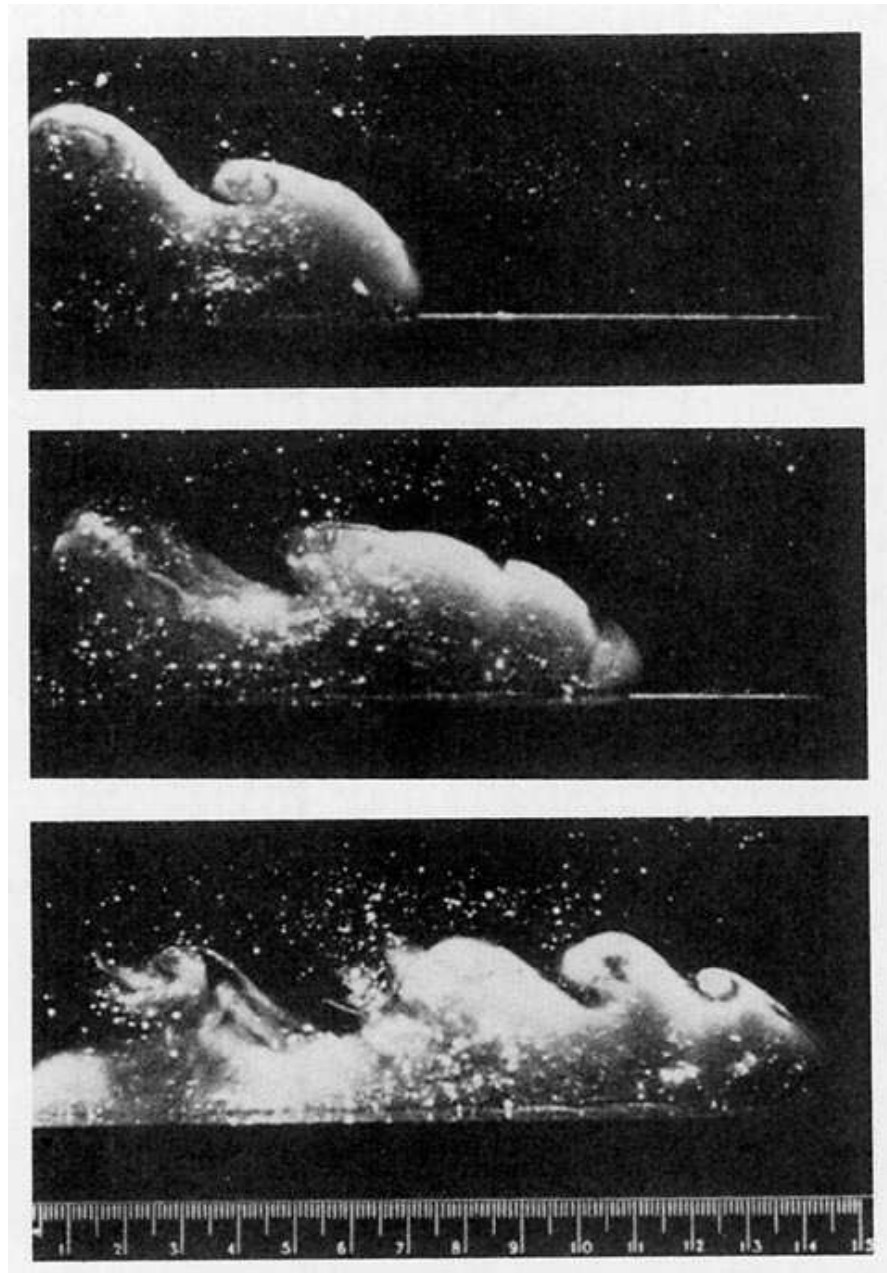


Figure 4.10: Case 3. Lab experiment of storm outflow. From Fig. 2 of [Droegemeier and Wilhelmson \(1987\)](#), (c)American Meteorological Society. Reprinted with permission. Original picture from [Simpson \(1969\)](#).

Table 4.6: Case 3. Comparative results of θ' extrema and front location at 900s. VMS (FE) at different resolutions, WRF-ARW V2.2 (FD), f-wave (FV), Spectral Elements (SE), Discontinuous Galerkin (DG), *REFC*, *REFQ* and *PPM* results are compared.

Model	Front Location [m]
VMS (25 m)	14890
VMS (50 m)	14629
VMS (75 m)	14487
VMS (100 m)	14355
VMS (200 m)	13811
VMS (400 m)	12733
WRF-ARW 50 m	14470
SE Giraldo and Restelli (2008) 50m	14767
DG Giraldo and Restelli (2008) 50m	14767
f-wave (FV) Ahmad and Lindeman (2007) 50 m	14975
REFC Straka et al. (1993) 50 m	14437
REFQ Straka et al. (1993) 50 m	14409
PPM Straka et al. (1993) 50 m	15027

Table 4.7: Case 3. $\tilde{\mathbf{q}}$ and \mathbf{q}^h extrema at $t_f = 900s$ and $\Delta x = \Delta z = 50 m$ resolution.

	ρ^h	$\tilde{\rho}$	U^h	\tilde{U}	W^h	\tilde{W}	θ^h	$\tilde{\theta}$
max.	1.206	1.038e-4	40.76	2.606e-2	18.60	2.448e-2	348.8	0.3932
min.	0.6472	-4.739e-5	-40.76	-2.606e-2	-25.97	-4.748e-2	194.2	-0.5834

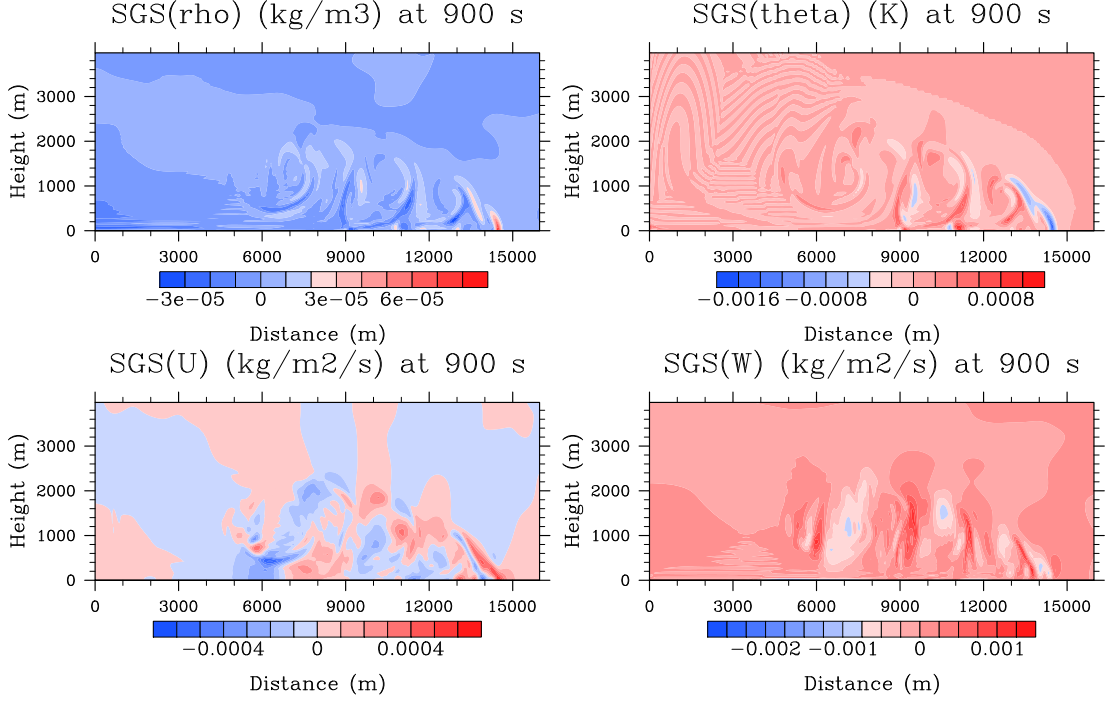


Figure 4.11: Case 3. Filled contours of the normalized subscales at $t_f = 900$ s and $\Delta x = \Delta z = 50$ m resolution. Top row: $\tilde{\rho}/\rho_{max}$, $\tilde{\theta}/\theta_{max}$. Bottom row: \tilde{U}/U_{max} , \tilde{W}/W_{max} .

test. Benjamin's equation is the following:

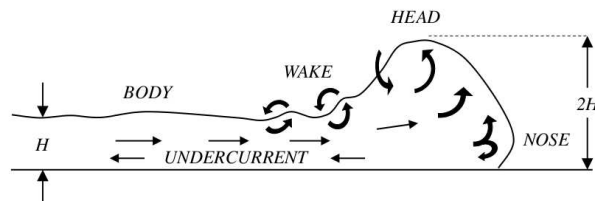
$$c = k\sqrt{g'd}, \quad (4.37)$$

where $g' = g\rho'/\rho$, $k = \sqrt{(\alpha^2 - 3\alpha + 2)/(1 + \alpha)}$, $\alpha = d/h_{domain}$, and h_{domain} is the depth of the domain. In physical thunderstorm outflows values of α between 0.71 and 1.25 are usually observed for k (e.g. [Mueller and Carbone \(1987\)](#); [CharbaJ. \(1974\)](#)). The largest k-value theoretically obtained is limited to $\sqrt{2}$ when $\alpha \rightarrow 0$, decreasing for increasing values of α . At equal geometry, [Klemp et al. \(1994\)](#) observed that theoretical studies and physical outflows can be characterized by different values of k partly because of surface friction and mixing effects that may differ in the two cases.

Results Case 4. Fig. 4.12 is a schematic representation of a physical current, while Fig. 4.13 shows the evolution of the simulated outflow on a grid $\Delta x = \Delta z = 50$ m at different final times. The vector field in Fig. 4.13 is shown for direct comparison with the plots of the reference. We consider the front to be identified by the last grid cell on the ground where $\theta < \bar{\theta} = 300, K$: in our problem this coincides with 9350 m at 600 s, giving a front speed of 15.66 m s^{-1} . If we take the largest absolute value of perturbation density ρ'_{max} inside the perturbed flow, by Eq. (4.37) we obtain the entries of Table 4.8, giving a theoretical leading edge speed of $c = 14.05 \text{ m s}^{-1}$. The difference between

Table 4.8: Case 4. Entries for Benjamin's law

Time	d	$\alpha = \frac{d}{h_{domain}}$	ρ	$ \rho'_{max} $	$k(\alpha)$	$g'(g, \rho', \rho)$
600 s	800 m	0.125	1.16	0.02	1.208	0.16

Figure 4.12: Case 4. Schematic of the structure of a physical density current as represented in [Mueller and Carbone \(1987\)](#) from the data of [Simpson \(1969\)](#); [Droegemeier \(1985\)](#).

the numerical speed measured in our simulation and that obtained by Benjamin's law is $\sim 1.5 m s^{-1}$.

Case 5: Inertia-gravity waves

Unlike the previous four, the inertia-gravity wave test consists of the motion of θ' along the horizontal direction x within a uniformly stratified environment. The domain is a straight channel defined within $[0, 300000] \times [0, 10000] m^2$ with periodic left and right boundaries. Like in [Skamarock and Klemp \(1994\)](#), the initial perturbation is defined as

$$\theta' = A \frac{\sin\left(\frac{z\pi}{h_c}\right)}{1 + \left(\frac{x-x_c}{a_c}\right)}, \quad (4.38)$$

where $A = 0.01 K$, $h_c = 10000 m$, and $a_c = 5000 m$. It is centered at $(x_c, z_c) = (100000, 5000) m$ and subject to a uniform horizontal flow of velocity $(\bar{u}, \bar{w}) = (20, 0) m s^{-1}$. The background environment is stratified with *Brunt-Väisälä* frequency $N = 0.01 s^{-1}$. At these conditions $\bar{\theta}$ is given by Eq. (4.30) with $\theta_0 = 300 K$. From Eqs. (2.10) and (4.29), the background pressure is

$$\bar{p} = p_0 \left[1.0 + \frac{g^2}{c_p \theta_0 N^2} \left(e^{\frac{-z N^2}{g}} - 1.0 \right) \right]^{c_p/R}. \quad (4.39)$$

Density $\bar{\rho}$ is computed from Eq. (2.6). Top and bottom boundary conditions are set to no-flux, while periodicity is imposed at the left and right boundaries.

Results Case 5. Fig. 4.14 shows θ' at the final time $t_f = 3000 s$ on a grid with $\Delta x = 1000 m$ and $\Delta z = 100 m$. For ease of comparison with the referenced cases, the

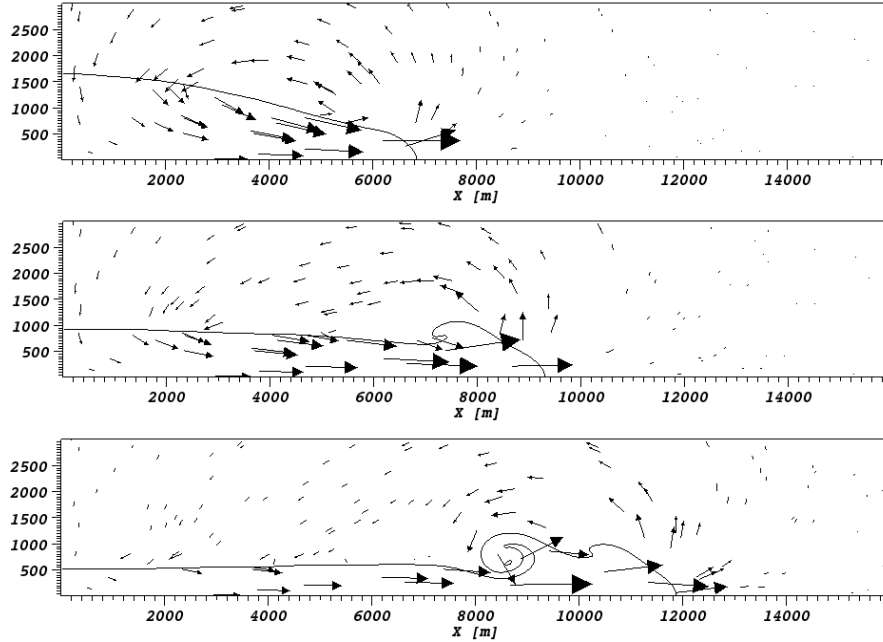


Figure 4.13: Case 4. From top: profiles of θ' with vector field at 300 s, 600 s, and 900 s.

same result is plotted first with the same contour levels of Fig. 13.6 of the *Advanced Regional Prediction System* (ARPS) V. 4.0 User's Guide (Arps, 2000), and then with those of the simulation of Ahmad and Lindeman (2007). As expected, at t_f the new center of the perturbation is positioned at 160000 m since it was advected by a flow with velocity $(\bar{u}, \bar{w}) = (20.0, 0.0) \text{ m s}^{-1}$ during 3000 s.

Reminded that we are employing first order discretization both in space and time, the best solution of reference that we should compare against is that obtained with ARPS, where symmetry is weakened as time approaches $t_f = 3000 \text{ s}$.

In the case reported in the referenced literature, symmetry is preserved if high order discretization schemes (in time and space) are employed. See, e.g., the 2nd-order accurate (in space and time) solution of Ahmad and Lindeman (2007), the 6th-order finite difference simulation of Skamarock and Klemp (1994) using WRF-ARW, and the high-order spectral elements and discontinuous Galerkin of Giraldo and Restelli (2008) with a high-order time integrator. On the other hand, in Arps (2000) it is concluded that first-order in space and time may cause the solution to fully lose symmetry. In the case presented here, however, the reason for symmetry loss cannot be blamed on the order only; by a test using bi-quadratic elements with an average dynamic time-step sufficiently small for time integration to be considered highly accurate, the same behavior is observed (plot not shown). Due to the high velocity field in the advection of θ' in the furthest right region of the domain, the diffusion due to VMS appears to be the cause for smearing the solution and hence causing symmetry to vanish.

The main conclusion that we obtain from this result is that the algorithm that we

are presenting is capable of reproducing the structure of the simulation in terms of correct advection of a very small perturbation of potential temperature in a uniformly stratified environment. However, the high speed that drive the stabilization in the region of maximum displacement (furthest right), makes the problem over-diffusive. Further investigation on VMS for long-lasting simulations of the Euler equations for thermally driven is necessary. Orthogonal subscales (OSS) are a possible option (Codina, 2000).

4.6.2 Numerical tests II: Mountain-induced waves

The correct simulation of vertically propagating linear hydrostatic (HS) and nonhydrostatic (NH) mountain waves is a classical test to verify a model's robustness and efficiency. Although the typical resolution used for many years in operational models was such that mountain-induced waves did not have quite a large importance, once the horizontal resolution approaches 10000 m or less, the phenomenon should be studied (Gallus and Klemp, 2000). Smith (1979) produced a useful summary of the problem, and variations of the tests therein described can be found in the literature. The Brunt – Väisälä frequency N , the height of the orography h_m , and its half-width amplitude a_c at ground are combined to create different atmospheric conditions. The deviation errors are quantified by the root-mean-square error (RMS) and normalized L_2 norm given by

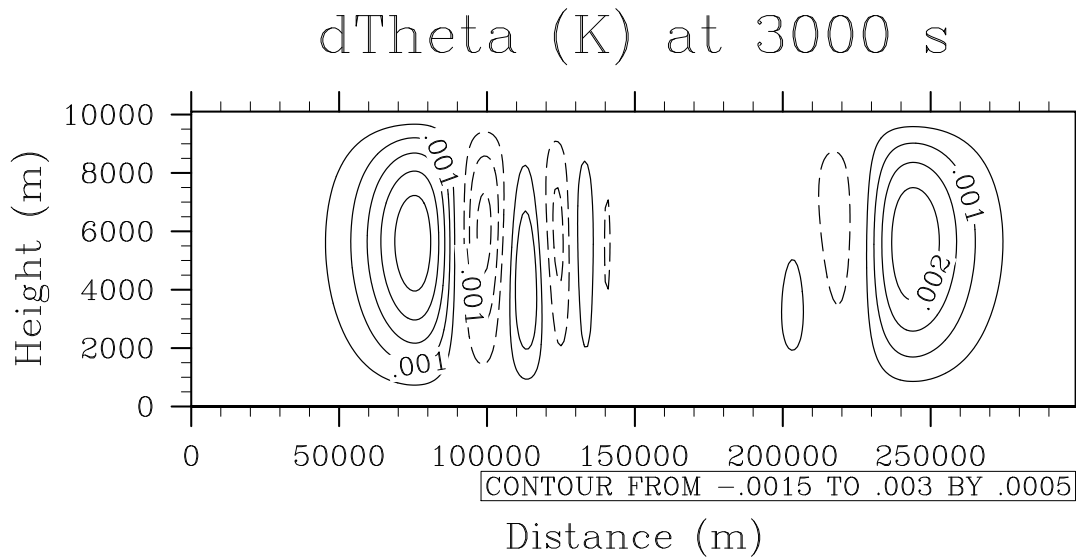
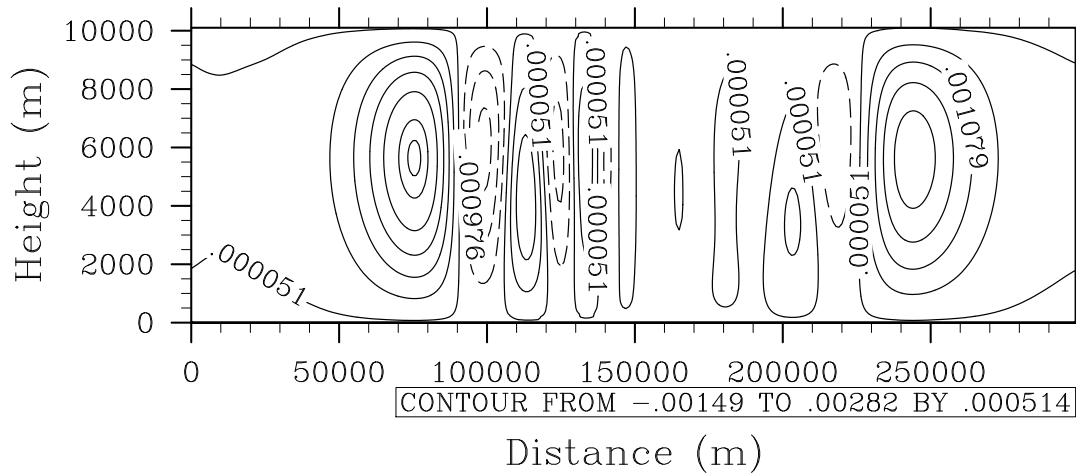
$$\text{RMS} = \sqrt{\sum_{k=1}^{n_{nodes}} (\mathbf{q}_k^h - \mathbf{q}_k^{ref})^2 / n_{nodes}},$$

$$\|\mathbf{q}^h - \mathbf{q}^{ref}\|_{L_2} = \sqrt{\frac{\sum_{k=1}^{n_{nodes}} (\mathbf{q}_k^h - \mathbf{q}_k^{ref})^2}{\sum_i^{n_{nodes}} (\mathbf{q}_k^{ref})^2}},$$

where \mathbf{q}^{ref} is a reference solution; i.e. either the analytic, $\mathbf{q}^{analytic}$, or the numerical solution computed on the finest grid $\mathbf{q}^{h,finest}$. Convergence to the stationary condition is measured by the time variation of vertical momentum flux. The momentum flux for hydrostatic and non-hydrostatic mountains is defined as

$$m(z) = \bar{\rho}(z) \int_{-\infty}^{\infty} u'(x, z) w'(x, z) dx. \quad (4.40)$$

For analysis, $m(z)$ is normalized by the analytic hydrostatic (HS) and non-hydrostatic (NH) momentum fluxes $m_{HS}(z) = -0.25\pi \rho_0 \bar{u} N h_c^2$ and $m_{NH}(z) = -0.457 m_{HS}(z)$, where 0.457 is a coefficient that makes the definition above of $m_{NH}(z)$ only valid when $N a_c / \bar{u} = 1$ (Klemp and Durran, 1983), and ρ_0 is the density on the ground. For these tests $\bar{\mathbf{u}}$ is the reference velocity, while \mathbf{u}' indicates its perturbation. Normalization is $m(z)/m_{HS}(z)$ and $m(z)/m_{NH}(z)$ for the hydrostatic and non-hydrostatic cases respectively. A vertical slice of the wave field is also plotted for further comparison between the numerical and analytic solution.

(a) Same contours of Fig. 13.6 in [Arps \(2000\)](#)(b) Same contours of Fig. 1 in [Ahmad and Lindeman \(2007\)](#)Figure 4.14: Case 5. θ' after 3000 seconds. Problem solved on a grid of $\Delta x = 1000\text{ m}$ $\Delta z = 100\text{ m}$.

Background state The background state of the atmosphere is a function of the *Brunt–Väisälä* frequency N and the potential temperature θ_0 measured at the surface. Hydrostatic balance is verified for \bar{p} . Given Eq. (4.29), $\bar{\theta}$ is defined by Eq. (4.30). From the linear analysis of Smith (1979), the linear problems below have a characteristic time $\hat{t} = a_c/\bar{u}$. The ratio between N and the characteristic frequency $\hat{f} = \hat{t}^{-1}$ defines the flow regime of the problem at hand.

Case 6: Linear Hydrostatic (HS)

In the case of linear hydrostatic flows, an isothermal atmosphere with $\theta_0 = 250\text{ K}$ flows above a single-peaked mountain. In this case, pressure is defined by Eq. (4.31). The mountain is the *Witch of Agnesi* of Eq. (3.1.2) with height $h_c = 1\text{ m}$, semi-width $a_c = 10000\text{ m}$ and center $x_c = 120000\text{ m}$ within the domain $[0, 240000] \times [0, 24000]\text{ m}^2$ (Durran and Klemp, 1983). The initial flow has uniform velocity $(\bar{u}, \bar{w}) = (20, 0)\text{ m s}^{-1}$ and stability $N = 0.0195\text{ s}^{-1}$. When $N/\hat{f} \gg 1$ we are considering a flow of strong stability over a wide mountain. This condition is such that a gravity wave propagates only vertically, with negligible vertical accelerations. The steady state solution is expected at $t_f > 10\text{ hrs} \sim 86\hat{t}$ approximately. No-flux boundary conditions are used on the bottom boundary while the top and lateral boundaries are treated with non-reflecting boundary conditions. The absorbing layer is built at $z \geq 12000\text{ m}$ and for $x \leq -80000\text{ m}$ and $x \geq 80000\text{ m}$.

Results Case 6. In Fig. 4.15 the numerical solution (black) of u' , w' , and θ' is plotted on top of the analytic solution (red) for a computational grid with 400×200 bilinear elements in x and z , giving a resolution of 600 m in x and 120 m in z .

RMS and the normalized L_2 -norm of $\mathbf{q}^h - \mathbf{q}^{analytic}$ are computed inside the portion of domain that is not affected by the Rayleigh-type boundary conditions and reported in Table 4.9. It emerges that the vertical perturbed velocity approximates the analytic solution best. This is expected from the hydrostatic regime of the problem, where vertical acceleration should be sufficiently small. The discrepancy of u' and θ' with respect to the analytic solution appears within the acceptable values from previous studies using element-based methods (e.g., Giraldo and Restelli (2008)).

From the normalized momentum flux represented in Fig. 4.17 we deduce that the simulation has reached steady-state after $t_f = 12\text{ hrs}$. In Fig. 4.16 the plot of a vertical slice of \mathbf{u}' , \mathbf{w}' , and θ' along $x = 120\text{ km}$ better shows the increase in the difference between the numerical and analytic solution as z approaches the level of the absorbing layer at $z = 12\text{ km}$. The non-reflecting boundary conditions used here are not optimal; this explains the larger error in the higher layers than at the surface. It also explains the oscillations in momentum flux that appear in Fig. 4.17 in the proximity of $z = 12\text{ km}$.

Case 7: Linear Nonhydrostatic (NH)

In the case of linear nonhydrostatic flows, the atmosphere above the Agnesi mountain is uniformly stratified with $\bar{\theta}$ given by Eq. (4.30), $\theta_0 = 280\text{ K}$, $N = 0.01\text{ s}^{-1}$, and

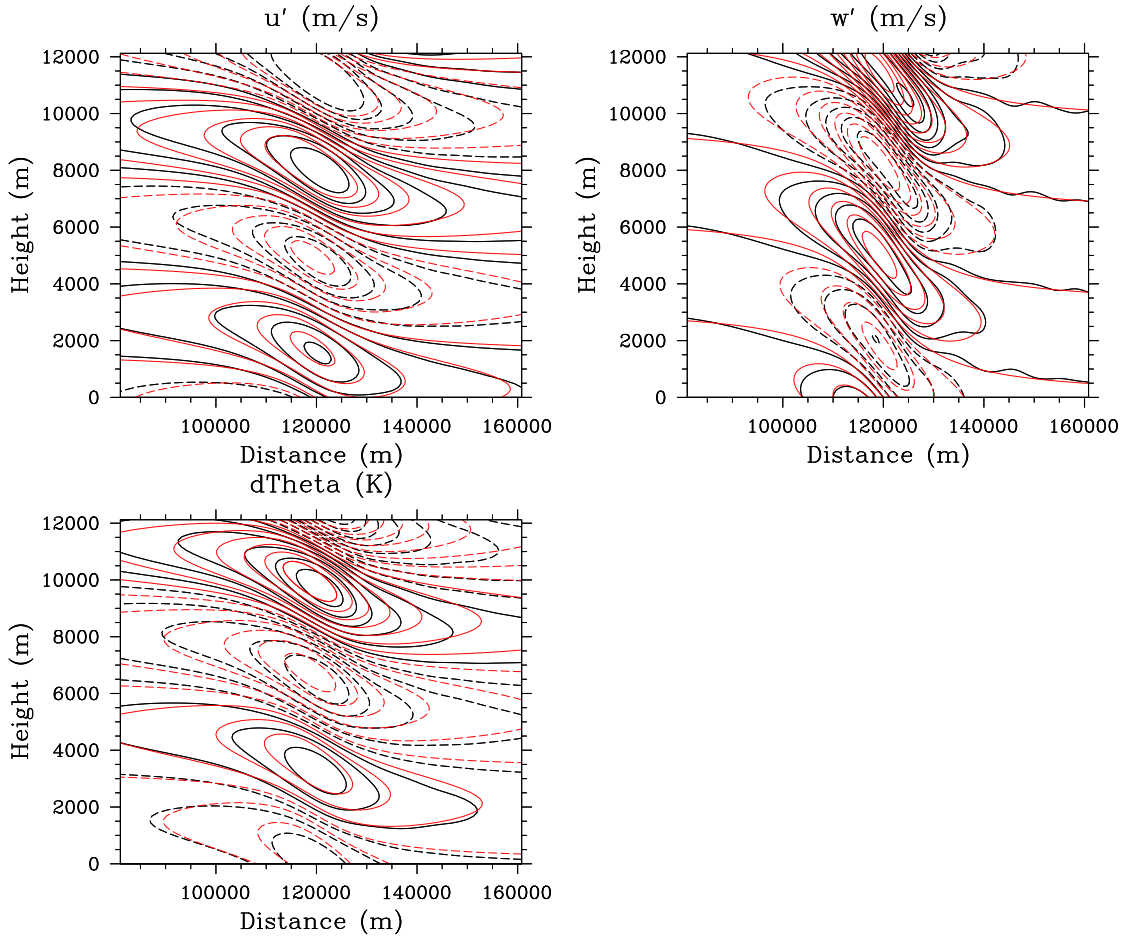


Figure 4.15: Case 6. From top left, in clockwise direction: u' , w' , and θ' contours at $t_f = 12 \text{ hrs}$ computed on a grid of 400×200 bilinear elements. The numerical and analytic solutions are black and red, respectively. The negative values are dashed. The contours are plotted as: $-0.025 \leq u' \leq 0.025$ with contour interval of 0.005; $-0.005 \leq w' \leq 0.005$ with contour interval of 0.0005; $-0.025 \leq \theta' \leq 0.025$ with contour interval of 0.00357.

Table 4.9: Case 6. *RMS* and normalized L_2 norm of $\mathbf{q}^h - \mathbf{q}^{analytic}$ for 400×200 elements.

	θ'	u	w
RMS	1.519e-3	2.294e-3	2.784e-4
L_2	1.972e-2	2.350e-2	1.127e-2

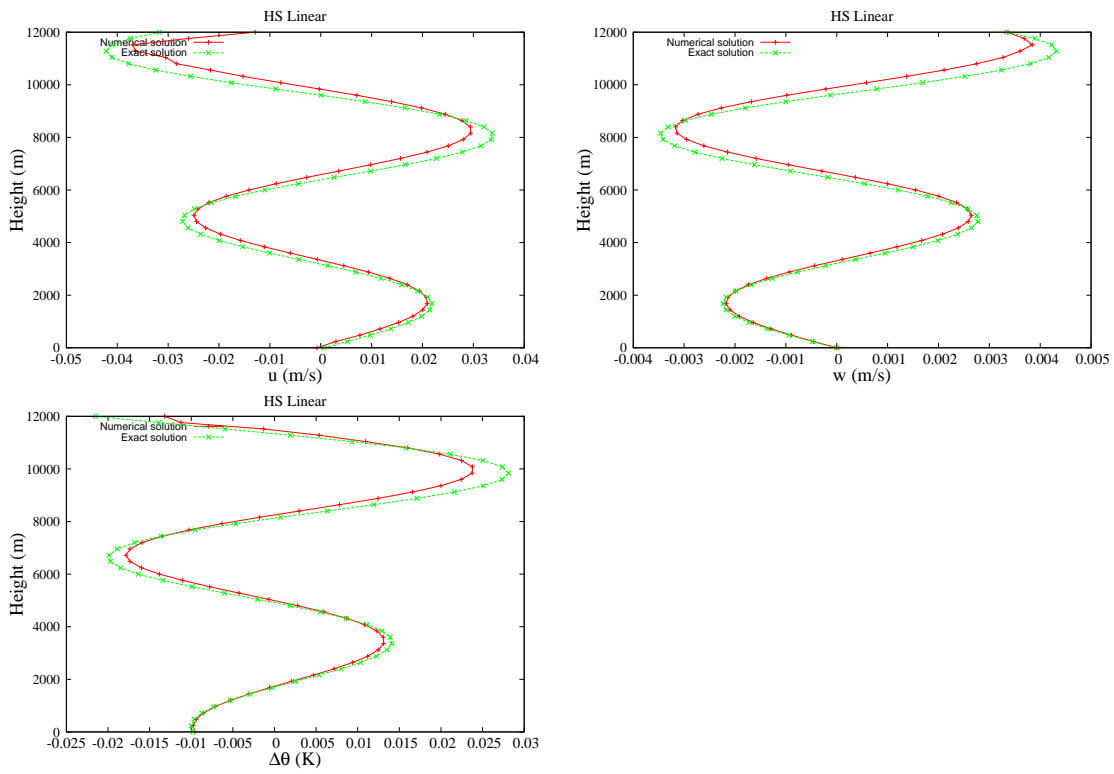


Figure 4.16: Case 6. Vertical cut in the middle of the domain. From top left, in clockwise direction: u' , w' , and θ' values at $t_f = 12$ hrs.

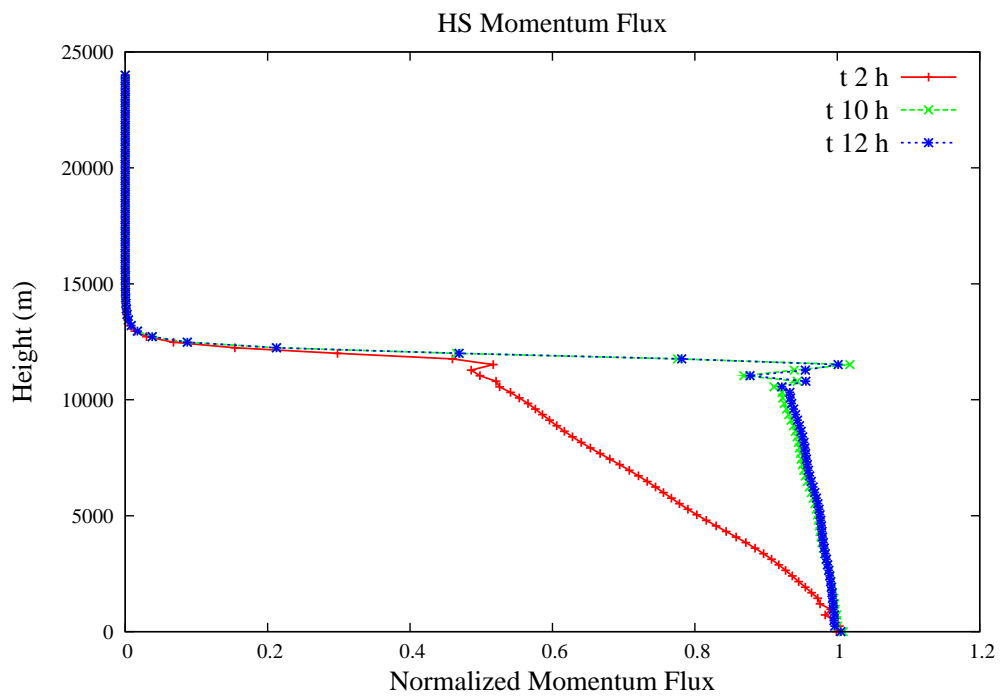


Figure 4.17: Case 6. Normalized momentum flux at $t_f = 2$ hrs, 10 hrs, 12 hrs.

Table 4.10: Case 7. RMS and normalized L_2 norm of $\mathbf{q}^h - \mathbf{q}^{analytic}$ for 400×200 elements.

	θ'	u	w
RMS	1.537e-4	4.738e-4	2.152e-4
L_2	2.468e-1	2.599e-1	2.529e-1

reference pressure from Eq. (4.31) The domain extends along $144000 m$ and $30000 m$ in the x and z directions, respectively. The initial environment has uniform velocity $(\bar{u}, \bar{w}) = (10, 0) m s^{-1}$ flowing over the same Agnesi profile of the previous case. now with semi-width $a_c = 1000 m$, height $h_m = 1 m$, and $x_c = 72000 m$. Under these conditions $\hat{f} = 0.01 s^{-1}$. This corresponds to a flow that is neither hydrostatic nor irrotational. Unlike the hydrostatic case, vertical accelerations in this regime are not negligible. No-flux conditions are used on the bottom boundary while non-reflecting conditions are applied along the top and lateral boundaries. The absorbing layer is now set at $z \geq 12000 m$, and on for $x \leq -60000 m$, and $x \geq 60000 m$.

Results Case 7. In Fig. 4.18 the numerical solution (black) is plotted on top of the analytic solution (red) at $t_f = 400 \hat{t} \sim 12 hrs$ on a computational grid with 400×200 bilinear elements, giving a resolution of $360 m$ in x and $150 m$ in z .

The degree of similarity between the numerical and analytic solution is measured by RMS and the normalized L_2 norm of $\mathbf{q}^h - \mathbf{q}^{analytic}$. Values are reported in Table 4.10. In this case, the error of u' and w' are of the same order of magnitude. Because the regime is non-hydrostatic, in contrast to *Case 6* we do not have a reason to expect the error of w' to be any lower than that of u' . The discrepancy of all variables with respect to the analytic solution appears within the acceptable values of previous studies that use element-based methods like Giraldo and Restelli (2008) where, however, spectral elements of order ten are used.

In Fig. 4.19, the plot of a vertical slice of u' , w' , and θ' along $x = 72000 m$ better shows the increase in the difference between the numerical and analytic solution as z approaches the level of the absorbing layer at $z = 12000 m$.

From the normalized momentum flux represented in Fig. 4.20 it appears that the simulation has reached a steady-state at $t = 10 hrs$ approximately. Unlike other studies, at $t = 5 hrs$ our numerical solution is still evolving. With a different definition and thickness of the absorbing layer, we believe that this may be one possible reason for the difference; however, a sensitivity study to different non-reflecting boundaries should be made to verify this hypothesis.

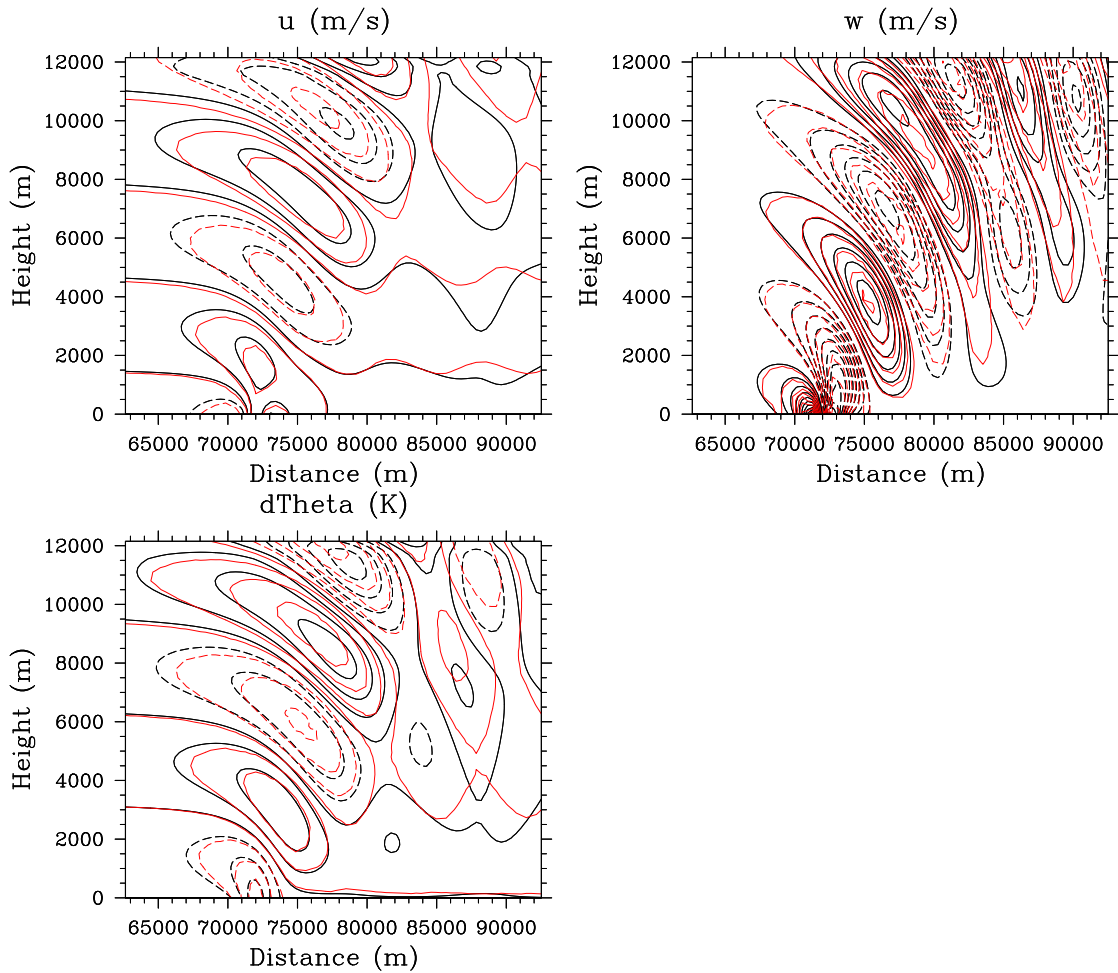


Figure 4.18: Case 7. From top left, in clockwise direction: u' , w' , and θ' contours at $t_f = 10$ hrs computed on a grid of 400×200 bilinear elements. The numerical and analytic solutions are black and red, respectively. The negative values are dashed. The contours are plotted as: $-0.025 \leq u' \leq 0.025$ with contour interval of 0.0025; $-0.005 \leq w' \leq 0.005$ with contour interval of 0.0005; $-0.0036 \leq \theta' \leq 0.0036$ with contour interval of 0.00072.

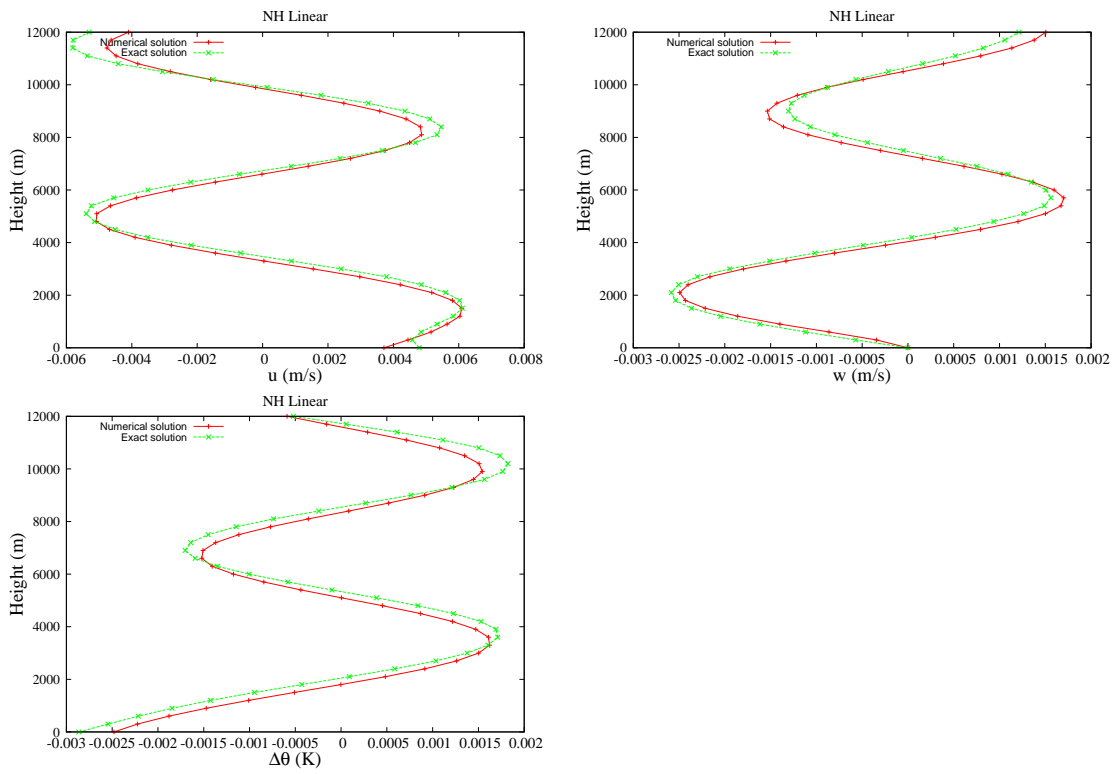


Figure 4.19: Case 7. Vertical cut in the middle of the domain. From top left, in clockwise direction: u' , w' , and θ' values at $t_f = 10$ hrs.

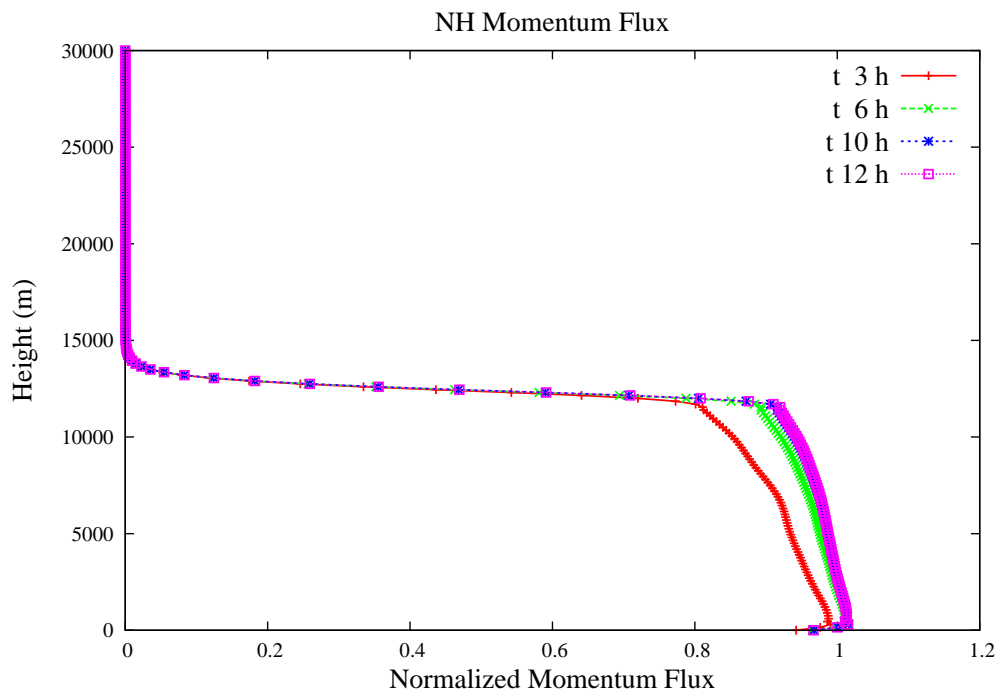


Figure 4.20: Case 7. Normalized momentum flux at $t_f = 3 \text{ hrs}, 6 \text{ hrs}, 10 \text{ hrs}, 12 \text{ hrs}$ for 360 m and 150 m resolution in x and z , respectively.

4.7 3D Numerical tests

Two-dimensional tests are well documented and standardized, whereas in three dimensions the same does not necessarily apply. From the recent works of [Ahmad \(2008\)](#) and [Kelly and Giraldo \(2012\)](#), we define two specific thermal problems. They are described in detail in *Case 9* and *Case 10*. The numerical verification for these problems proceeds in two steps. We start with the same approach of Kelly and Giraldo, who defined a pseudo-3D rising thermal bubble test and compared it against its analogue 2D result. For reasons that will be explained below, the two outputs should be identical. Were they not, it would mean that the algorithm will certainly not perform correctly in 3D. In step two, the code is run in real 3D mode and tested with the simulation of a buoyant rising bubble.

Case 9: Pseudo-3D warm bubble

We consider an initially neutrally stratified atmosphere in the 2D square domain of size $[0, 1000] \times [0, 1000] m^2$ first, and in the 3D box of dimension $[0, 1000] \times [0, 100] \times [0, 1000] m^3$ next. Note that, because only one element and periodic boundary conditions are used in the y -direction of the box, the actual extension along y is irrelevant. In both runs, a perturbation θ' with radius $r_0 = 250 m$ and centered in $(x_c, z_c) = (500, 350) m$ is initially at rest and used to perturb the atmosphere that is originally at uniform $\theta = \theta_0 = 300 K$. The perturbation is

$$\theta' = A \left[1 + \cos \left(\frac{\pi R}{r_0} \right) \right], \quad (4.41)$$

where $R = \sqrt{(x - x_c)^2 + (z - z_c)^2}$ and $A = 2 K$. The initial velocity field is zero everywhere. In the already familiar 2D problem the four boundaries are no-flux permitting. For the pseudo-3D problem, solid walls are defined everywhere except for the front and back xz -planes, where periodicity is applied. Under these conditions, the perturbation defined in the periodic problem is a cylinder of infinite extension along y .

Results Case 9. 40×40 bilinear elements are used for the reference 2D problem, giving a resolution of $\Delta x = \Delta z = 25 m$. The pseudo-3D problem is based on a grid of equivalent resolution using $40 \times 1 \times 40$ hexahedra. Figs. [4.21](#) and [4.22](#) display the xz cross-sections of θ' , p' , u and w computed with the 2D and pseudo-3D runs at $t_f = 600 s$. Not only the structure, but the values of every variable is expected to be identical because of the periodicity and the use of one element along y . As previously observed, none of these tests has analytic solution and we cannot state with blind certainty that the result is correct. However, the physics that is reproduced by both runs is plausible and certainly comparable with previous studies presented either in this work or in the literature. Given that the pseudo-3D output seems encouraging, we extended the problem to three dimensions and describe it in *Case 10*.

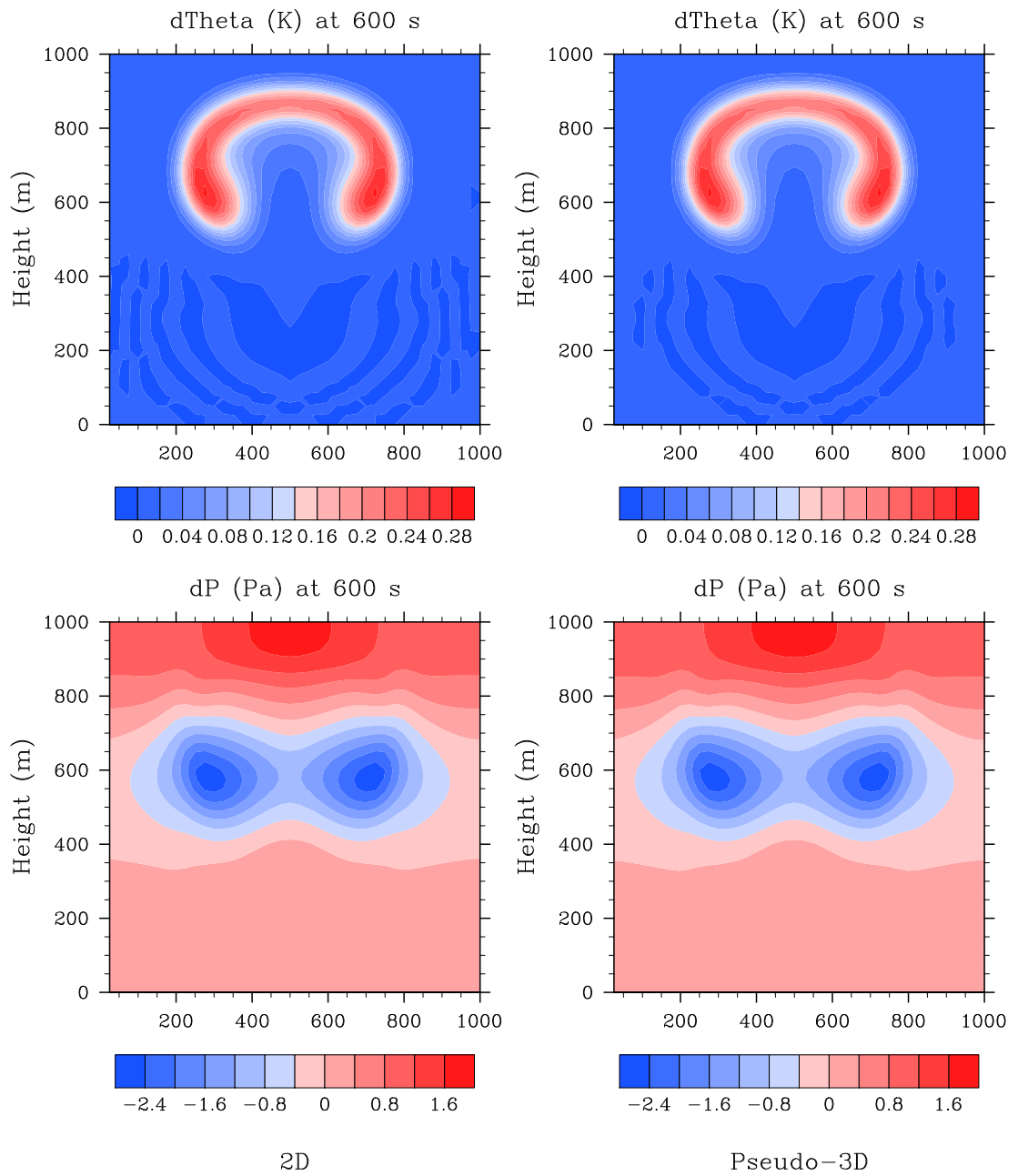


Figure 4.21: Case 9. Filled contours of θ' (top row) and p' (bottom row) for $\Delta x = \Delta z = 25$ m. The left and right columns represent, respectively, the simulation in 2D and pseudo-3D modes (i.e. semi-infinite box with periodic xz -plane). Values are shown at $t = 600$ s.

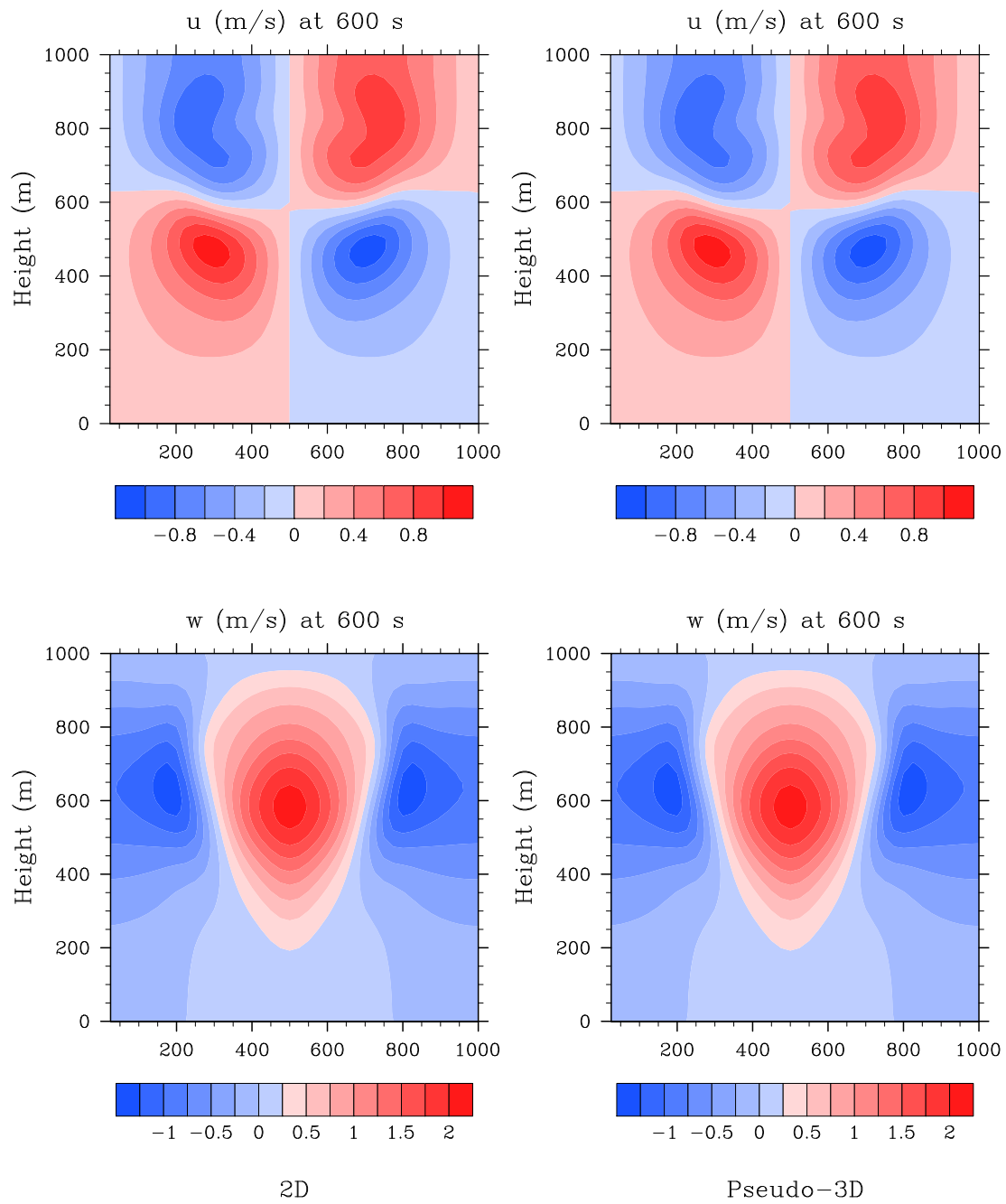


Figure 4.22: Case 9. As Fig. 4.21, but representing u (top row) and w (bottom row).

Case 10: 3D warm bubble

The problem of a 3D rising thermal bubble is taken from [Ahmad \(2008\)](#). The domain is bounded within $[0, 3200] \times [0, 3200] \times [0, 4000] m^3$. The initial atmosphere is neutrally stratified with $\bar{\theta} = \theta_0 = 300 K$. Its perturbation, θ' , is now spherical and is centered in $(x_c, y_c, z_c) = (1600, 1600, 500) m$. It is a linear function of space and defined as

$$\theta' = A \left(1 - \frac{R}{r_0} \right), \quad (4.42)$$

where $R = \sqrt{(x - x_c)^2 + (y - y_c)^2 + (z - z_c)^2}$ and $A = 2 K$. The initial velocity field is zero everywhere. Solid boundaries are prescribed.

Results Case 10. As for two-dimensional advective thermals, the warm perturbation generates acceleration in the inner region of the bubble (where temperature is higher), with subsequent downdrafts originating at the boundaries of the perturbation. A classical mushroom shape generates because of the faster, vertically moving, center of the thermal. The mesh resolution is uniform and set to $40 m$ using $80 \times 80 \times 100$ hexahedra. Symmetry considerations and direct comparison against Ahmad are used to evaluate the results. Figs. [4.23](#) and [4.24](#) display the xz and yz cross-sections of θ' , p' , u , and w at $t_f = 480 s$. Due to the symmetric nature of the problem and the initial spherical shape of θ' , the evolution of the perturbation of potential temperature is expected to preserve axisymmetry at all times. The same consideration applies to the perturbation of density (ρ' , not shown), and hence pressure. By observing the vertical sections on two orthogonal planes, this is indeed verified with very high accuracy. A slower front is observed by comparison with the referenced work. The different numerical dissipation introduced may be one cause, although a thorough sensitivity analysis on grid resolution and stabilization should be addressed in a future work.

4.8 Summary and discussion

The finite element method has two fundamental qualities. As we observed above, one is certainly the complete freedom with respect to the structure of the computational grid. The other one is represented by its impressive parallel efficiency¹. With this in mind, in this chapter we proposed the use of the finite element approximation of the compressible Euler equations for stratified flows at low Mach numbers typical of atmospheric dynamics. Because of the instability that occurs in the numerical solution of advective problems (i.e. Euler equations in this case), we stressed the analysis on stabilization by the Variational Multiscale Method. The algorithm is an extension of VMS to compressible flows and was adapted to the treatment of thermally perturbed stratified

¹Although in this thesis we are not describing parallelization issues in specific, the underlying code which this work is built on is a massively parallel code whose characteristics are described in [Houzeaux et al. \(2009\)](#). A brief description of the software is reported in Appendix B of this thesis.

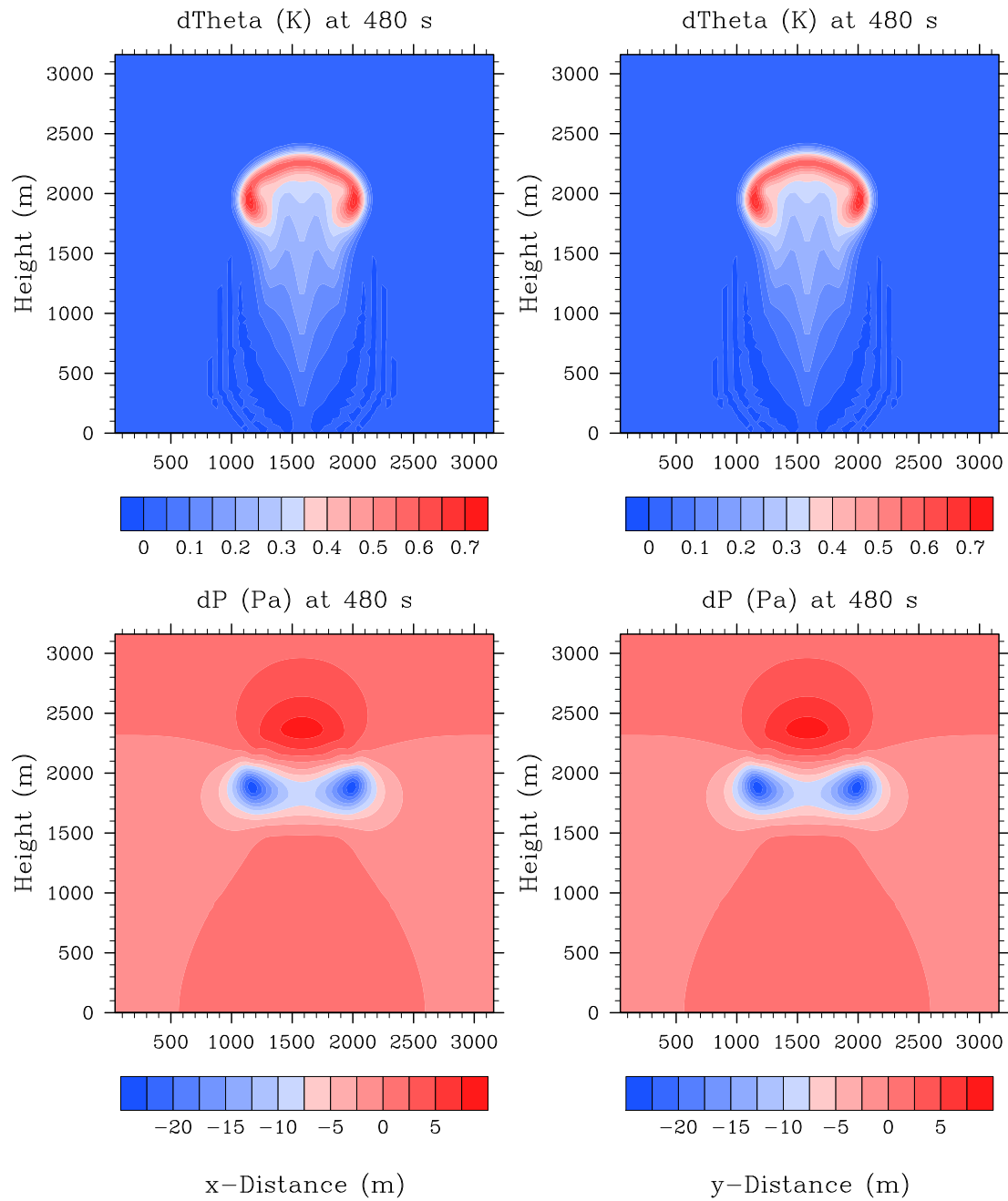


Figure 4.23: Case 10. Filled contours of θ' (top row) and p' (bottom row) are computed on a grid of $80 \times 80 \times 100$ elements. The left and right columns represent, respectively, the xz and yz planes. For plotting reasons, the plotted domain is shorter than the computational domain. Values are shown at $t = 480$ s.

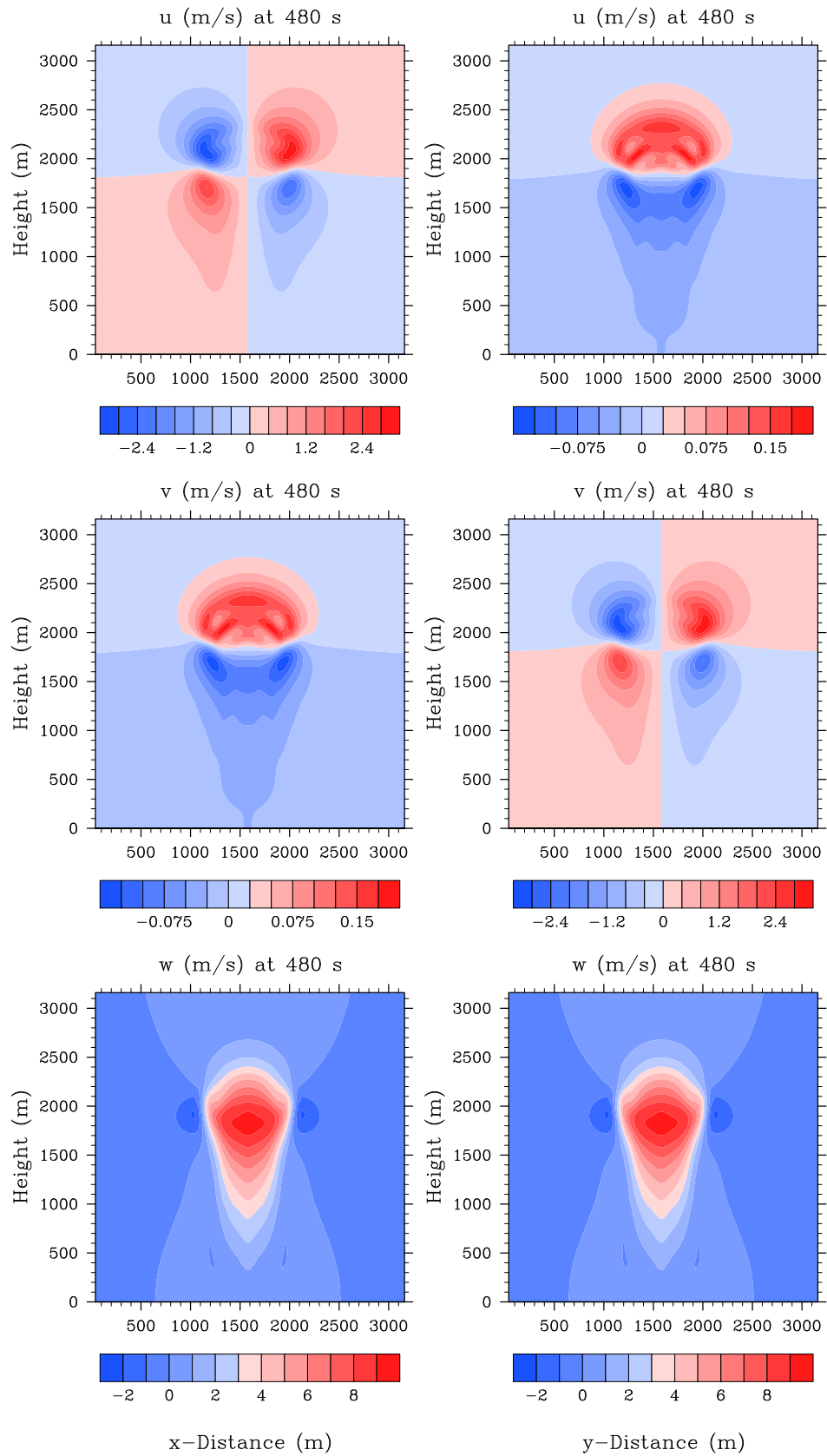


Figure 4.24: Case 10. As Fig. 4.23 but for u (top row), v (middle row), and w (bottom row).

atmospheres. We also introduced a simple methodology that limits the truncation errors committed in the solution of the hydrostatic states that, otherwise, would cause the generation of unphysical vertical accelerations. With this strategy, only the deviations from the background state in hydrostatic balance are considered in the evaluation of the element integrals, while the background state is explicitly computed at the quadrature points (i.e. Gauss points).

Performance and accuracy were measured by means of standard benchmarks adopted by atmospheric modelers. In terms of stability, the compressible VMS technique that was tested appears to stabilize the solution of low-Mach number compressible flows solved by the finite element method. Unlike artificial diffusion, the viscosity that is added by VMS is controlled and localized in the regions where the residual of the solution is important. Nevertheless, VMS is not a method that, alone, preserves monotonicity. This can be proved with the error estimate of [Codina \(2000\)](#). If monotonicity is a requirement, VMS should be coupled to a proper shock capturing method. This problem is even more relevant when, instead of linear finite elements, the solution is computed by high-order methods such as spectral elements and discontinuous Galerkin. If limiters are an effective solution for discontinuous Galerkin methods, this is not true for spectral elements. This will be treated in [Chapter 5](#), where the Spectral Element method is stabilized by VMS as an alternative to classical filters used with SEM approximations. In this chapter we have not investigated shock capturing methods to render the solution monotonic because, in NWP, this condition can be relaxed with no harm when solving the equations of the dynamical core.

The major drawback of the current implementation is the fully explicit time integration scheme that translates into a severe Δt restriction. Neither research nor operational nonhydrostatic NWP models can accept this constraint not even on the fastest machine designed for high-performance computing. We were aware of this constraint since the beginning of this work; however, throughout the development, we have never specifically focused on its solution and it is thus left as a starting point for future work.

The software environment that contains the algorithm described so far is described in [Appendix B](#).

Chapter 5

Toward high-order positivity preserving spectral elements for the advection-diffusion equations

Element-based high-order methods have shown great success in atmospheric simulations. Of these, the spectral element method (SEM) and discontinuous Galerkin (DG) are among the most common (e.g., [Fournier et al. \(2004\)](#), [Levy et al. \(2007\)](#), [Kelly and Giraldo \(2012\)](#)). Higher accuracy and impressive parallel performance on massively parallel computers are the main reasons for their success. Like most numerical methods, SEM for advection dominated problems needs stabilization. This is classically achieved by filtering high frequency modes to keep the solution bounded. This methodology is good for stabilization purposes, but the solution may lose smoothness especially in the proximity of large gradients. With the goal of improving their characteristics, this chapter is an extension of [Chapter 4](#) to stabilize high-order spectral elements in the solution of the advection-diffusion equation. Furthermore, discontinuity capturing methods are applied as well to enhance monotonicity given its importance in transport phenomena in geophysical fluid dynamics. This study is supportive of the wide spectrum of spectral-element codes used nowadays by the atmospheric community, where, though, the preservation of positivity is challenged by the Gibbs oscillations typical of high-order space discretization. No results will be presented for low order finite elements because, in this respect, the stabilization features of VMS in the FEM approximation of the advection-diffusion equation has been largely treated and assessed throughout the years.

5.1 Introduction

As mentioned in the previous chapters, a large number of physical applications relies on the accurate solution of [Eq. \(2.13\)](#) or, with the addition of diffusion, on the solution of

the advection-diffusion equation

$$\frac{\partial q}{\partial t} + \mathcal{L}(q) = f, \quad (5.1)$$

where $\mathcal{L}(q) = \mathbf{u} \cdot \nabla q - \nabla \cdot (\nu \nabla q)$. It is of great importance for the solution of (5.1) in atmospheric simulations to respect two significant properties: (i) positivity must be preserved, and (ii) smearing at internal and boundary layers should not be excessive. These properties are extremely important in the context of transport in the atmosphere. Both limited-area and global atmospheric models for weather prediction need monotonic advection of tracers and moisture variables, otherwise the wrong amount of precipitation would be forecasted. Simple microphysics schemes, such as Kessler’s parametrization (see Kessler (1969); Klemp and Wilhelmson (1978) and Chapter 6), require three variables (water vapor, cloud water, and rain), whereas more sophisticated parametrizations include additional variables such as ice and snow (Houze, 1993). Similarly, climate models require transport of hundreds of tracers, each representing a different chemical species. Regardless of the physical scales of the model, tracers must preserve monotonicity since the physical parametrizations that govern sub-grid scale processes such as auto-conversion and sedimentation, implicitly assume such a condition. These issues have been addressed for both transient and stationary problems (see, e.g., Restelli et al. (2006)) and, in the context of finite element methods, the *stabilized* methods introduced above have been an active topic of research since their introduction in the early 1980s. In this Chapter, we address the problem of solving (5.1) by spectral elements without losing the ability to approach a monotone solution to the problem. High-order accuracy comes at the price of aliasing phenomena in the solution (Lax, 1978), but the anti-aliasing filters typically used to give a stable spectral element solution do not respect conditions (i) and (ii).

Regardless of the type of numerical method that atmospheric models apply, dissipation of some sort is always added for reasons that do not only involve stabilization, but also subgrid-scale processes such as the modeling of turbulence effects (see the review of Jablonowski and Williamson (2011) and references therein). As far as stabilization is concerned, explicit dissipation is very much used to control the Gibbs oscillations in spectral codes, which is the fundamental problem that we will discuss below.

Traditionally, explicit dissipation –or Hyper-Viscosity (HV)– is added to Eq. (5.1) as a linear diffusion operator defined as

$$b_{HV} = (-1)^{\alpha+1} \nabla^\alpha \cdot (\bar{\nu}_\beta \nabla^\alpha q), \quad (5.2)$$

where $\alpha = \beta/2$, being β a positive even power of the hyper-viscosity operator, and $\bar{\nu}_\beta$ is the coefficient of thermal diffusivity corresponding to the operator of order β .

Although HV is scale-selective (i.e., it damps only higher frequencies), it is not consistent, nor is it physical. In fact, to maintain the correct physical dimensions of the hyper-viscous term, the value of $\bar{\nu}$ must be different when different α are used. Its selection not only is not trivial, but it has a great impact on the solution of the problem.

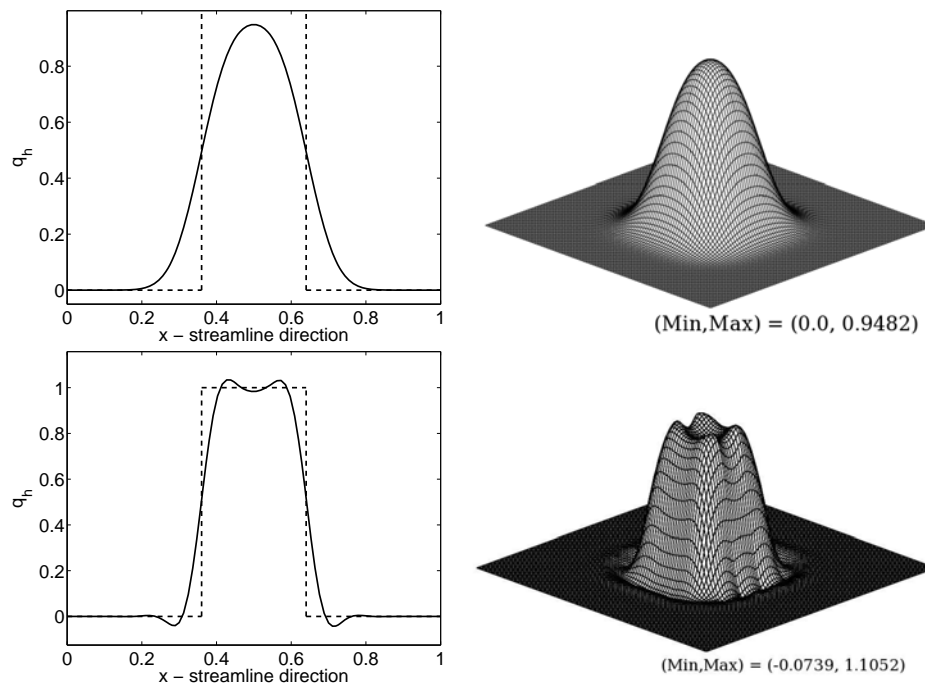


Figure 5.1: Advection of a 2D square wave in a doubly-periodic channel with velocity directed along the x-axis. Stabilization achieved by second order HV (top row) and by fourth order HV (bottom row). The exact solution is the dashed lined in the vertical section plots on the left column. $\bar{v}_2 = 0.001 \text{ m}^2 \text{ s}^{-1}$ and $\bar{v}_4 = 0.000001 \text{ m}^2 \text{ s}^{-1}$.

Let us quote [Jablonowski and Williamson \(2011\)](#) (p. 393):

The choice of the ∇^2, ∇^4 or even higher-order diffusion coefficient is most often motivated by empirical arguments and chosen in a somewhat arbitrary manner. It is sometimes even considered a model tuning parameter [...]

This is a measure of the possible limitation of these methods in stabilizing advection problems where accuracy is mandatory, especially if you think that we are looking at values of the order of $10e - 05$ (e.g. kg of cloud water over kg of dry air in cloud simulations). A small error in computation may translate into a very big error in the numerically estimated concentration q . Certainly, a method that could avoid the need for tuning and arbitrary selection would certainly improve the solution of the problem at hand. To gather a rough idea of what effects higher-order viscosity may have on the solution of a 2D advection problem of a passive tracer in a doubly periodic square domain $\Omega = [-1, 1] \times [-1, 1]$, we implemented a 4th-order version of (5.2)¹. Fig. 5.1 shows the structure of the solution after one full revolution along the x-direction.

¹Detailed examples using different stabilization techniques are described and analyzed later in this chapter. This first attempt is simply to gain some insight on how much control we have on the artificial diffusion that we add when we solve a transport problem with high-order spectral elements.

The implementation that conducted to the results of Fig. 5.1 is not optimal and was made in the most simple of ways. The selection of $\bar{\nu}$ itself is not optimal. Unlike the second order operator that has a physical meaning (molecular diffusivity in the medium), and hence $\bar{\nu}_2$ can be chosen on physical reasoning (for reasons that will be explained later, $\bar{\nu}_2$ in this test is not physical either), the same does not apply to higher-order operators. The over-diffusive 2^{nd} -order artificial diffusion (top row in the figure) and the less diffusive result obtained with the 4^{th} -order operator show that tuning is indeed necessary to optimize their performance, and the selection of the value of the diffusivity coefficient requires prior knowledge of the problem to be solved. As Jablonowski and Williamson (2011) stress out, there is no “right” or “wrong” subgrid-scale mechanism that can be effective at all scales and that is globally optimal for all models. By applying methods that helped improve stabilization of linear and quadratic finite elements, in this chapter we suggest a possible way out to the problem of oscillating solutions with high order spectral elements without relying on hyper-viscosity schemes.

To pursue our goal we apply VMS to stabilize SEM without spectral filters. VMS methods are residual-based schemes that improve the stability properties of the solution and preserve the accuracy of the underlying numerical scheme (Hughes et al., 2005). Neither VMS, SUPG, or GLS, however, preclude the formation of over- and under-shoots in the proximity of sharp gradients. For this reason, *shock/discontinuity capturing* (DC) techniques, also referred to as *Spurious Oscillations at Layers Diminishing methods* (SOLD), are used in combination with VMS to introduce an additional term to the stabilized form of the equation. A detailed review of most existing SOLD schemes can be found in a two-part paper by John and Knobloch (2007, 2008). We have previously seen that this category of stabilization methods strongly depend on a parameter identified by τ . τ for high order finite elements was defined for quadratic and cubic elements only, whose interpolation nodes are equi-spaced (see Codina et al. (1992), Hughes and Sangalli (2007), and Houzeaux et al. (2009)). τ is a function of the element characteristic length (e.g., h as the largest edge of a bilinear-element). The challenge stands in the definition of this characteristic length for high order elements; even more for high order elements whose nodes are unevenly distributed (i.e. spectral elements). For this reason, in this chapter we propose τ for higher-order spectral elements and use it to construct the VMS term. To further improve the solution, we combine VMS and DC with a method referred to as FOS (for First-Order Subcells) (Marras et al., 2012a). This technique subdivides a tensor product spectral element of order p and dimension d into p^d subcells and then uses 1^{st} -order basis functions and integration rules on every subcell of the element. In this chapter, all of these methods are described and tested.

5.2 Numerical formulation

Given the same definitions and notations of Section 3.1, the fundamental difference that distinguishes spectral from finite elements are the polynomial basis functions that are

used.

Given the spaces L^2 and H^1 (see Chapter 3), the general d -dimensional basis functions that are used for the problems of this chapter are built as the tensor product of d one-dimensional basis $h_i(\xi) \in W \subseteq H^1$ of order N (Eq. (3.13)). In the case of $d = 2$, we write:

$$\psi_k(\mathbf{x}) = h_i(\xi(\mathbf{x})) \otimes h_j(\eta(\mathbf{x})), \quad \forall i, j = 0, \dots, N, \quad (5.3)$$

where \otimes indicates the tensor product and the two-dimensional index k varies as $k = 1, \dots, (N + 1)^2$. The use of tensor product to take advantage of the 1D machinery for higher-dimension elements does not limit spectral elements to the use of quadrilaterals or hexahedra. SEM, just like FEM, has full flexibility on the shape of the grid. Simply, for triangular or tetrahedral elements the shape functions cannot be built as in (5.3) and have to be computed otherwise. In the following, we describe the spectral element method with variational multiscale stabilization and show a possible derivation of the stabilization parameter τ for higher-order elements. For better monotonicity properties, we also introduce a method referred to as FOS for *First Order Subcells* when the combination of shock capturing and high-order elements does not perform best. As usual, the technique is assessed using standard benchmarks at the end of the chapter.

5.2.1 Spectral element formulation

As we briefly did in Chapter 3 for a general differential problem, and later developed for the system of Euler equations in Chapter 4, we now proceed and construct the spectral element approximation of the non-steady advection-diffusion equations (5.1). As usual, let us project equation (5.1) onto $W \subseteq H^1$ by the L^2 scalar product to obtain the weak problem of finding a solution $q \in W$ such that

$$\int_{\Omega} \psi \left[\frac{\partial q}{\partial t} + \mathbf{u} \cdot \nabla q - \nabla \cdot (\nu \nabla q) - f \right] d\Omega = 0, \quad \forall \psi \in W. \quad (5.4)$$

For the solution to belong to H^1 , we integrate the diffusive term by parts as follows:

$$\int_{\Omega} \psi \nabla \cdot (\nu \nabla q) d\Omega = - \int_{\Omega} \nu \nabla \psi \cdot \nabla q d\Omega + \int_{\partial\Omega} \psi \mathbf{n} \cdot \nabla q d\partial\Omega. \quad (5.5)$$

We defined W in (3.9) as the space of functions that support homogeneous Dirichlet boundary conditions and hence such that $\psi(\partial\Omega) = 0$. Under this condition, substitution of (5.5) into (5.4) yields the problem of finding $q \in W$ such that

$$\int_{\Omega} \psi \left[\frac{\partial q}{\partial t} + \mathbf{u} \cdot \nabla q + \nu \nabla \psi \cdot \nabla q - f \right] d\Omega = 0, \quad \forall \psi \in W. \quad (5.6)$$

Stabilization is described in what follows.

5.2.2 Stabilization techniques

The way stabilization is achieved in this chapter is somewhat different from the way it was done in Chapter 4 for the non-linear, coupled, Euler equations. The common denominator is the splitting of the solution into a resolvable and non-resolvable components. The fundamental ideas are the same, however, the algorithmic implementation is different for practical more than theoretical reasons. We see how shortly.

Additional definitions. Although they have not been needed so far, for practical reasons in the next derivations it is useful to introduce the inner product (\cdot, \cdot) which $L^2(\Omega)$ is endowed with and the bilinear form² $a(\cdot, \cdot)$ that satisfies

$$a(\psi, q) = (\psi, \mathcal{L}(q)).$$

We can hence express the weak problem (5.6) compactly as that of finding $q \in W$ such that

$$\left(\psi, \frac{\partial q}{\partial t} \right) + a(\psi, q) = (\psi, f) \quad \forall \psi \in W, \quad (5.7)$$

where

$$\begin{aligned} \left(\psi, \frac{\partial q}{\partial t} \right) &\doteq \int_{\Omega} \psi \frac{\partial q}{\partial t} d\Omega, \\ a(\psi, q) &\doteq \int_{\Omega} \psi \mathbf{u} \cdot \nabla q d\Omega + \int_{\Omega} \nu \nabla \psi \cdot \nabla q d\Omega, \\ (\psi, f) &\doteq \int_{\Omega} \psi f d\Omega. \end{aligned}$$

The variational multiscale formulation (VMS). Let $W = W^h \oplus \tilde{W}$ be the space decomposition such that \tilde{W} completes W^h in W . This translates into the decomposition $q = q^h + \tilde{q}$ of the solution variables. The same applies to the basis as $\psi = \psi^h + \tilde{\psi}$. Substitution of the decomposition into Eq. (5.7) and anticipating that we will consider \tilde{q} to be quasi-static (Codina, 2002), we obtain:

$$\left(\psi^h + \tilde{\psi}, \frac{\partial q^h}{\partial t} \right) + a(\psi^h + \tilde{\psi}, q^h + \tilde{q}) = (\psi^h + \tilde{\psi}, f) \quad \forall \psi^h \in W^h, \tilde{\psi} \in \tilde{W}. \quad (5.9)$$

²Definition: bilinear form XXX

By virtue of the linear independence of ψ^h and $\tilde{\psi}$ we can first take $\tilde{\psi} = 0$ and then $\psi^h = 0$ and find the split problem:

$$\left(\psi^h, \frac{\partial q^h}{\partial t} \right) + a(\psi^h, q^h) + a(\psi^h, \tilde{q}) = (\psi^h, f) \quad \forall \psi^h \in W^h \quad (5.10a)$$

$$\left(\tilde{\psi}, \frac{\partial q^h}{\partial t} \right) + a(\tilde{\psi}, q^h) + a(\tilde{\psi}, \tilde{q}) = (\tilde{\psi}, f) \quad \forall \tilde{\psi} \in \tilde{W}. \quad (5.10b)$$

We make the assumptions that $\tilde{\psi}(\partial\Omega) = 0$ and $\tilde{q}(\partial\Omega) = 0$, and that $\tilde{\psi}(\partial\Omega_{el}) = 0$.

Following [Hughes \(1995\)](#), we integrate by parts the bilinear forms that depend on the subgrid-scales in (5.10):

$$\begin{aligned} a(\psi^h, \tilde{q}) &= (\mathcal{L}^* \psi^h, \tilde{q}) \quad \forall \psi^h \in W^h \text{ and } \tilde{q} \in \tilde{W} \\ a(\tilde{\psi}, q^h) &= (\tilde{\psi}, \mathcal{L}q^h) \quad \forall \tilde{\psi} \in \tilde{W} \text{ and } q^h \in W^h \\ a(\tilde{\psi}, \tilde{q}) &= (\tilde{\psi}, \mathcal{L}\tilde{q}) \quad \forall \tilde{\psi} \in \tilde{W} \text{ and } \tilde{q} \in \tilde{W}, \end{aligned}$$

where \mathcal{L}^* is the dual (or adjoint) of \mathcal{L} . The following system is then found:

$$\left(\psi^h, \frac{\partial q^h}{\partial t} \right) + a(\psi^h, q^h) + (\mathcal{L}^* \psi^h, \tilde{q}) = (\psi^h, f) \quad \forall \psi^h \in W^h \quad (5.11a)$$

$$\left(\tilde{\psi}, \frac{\partial q^h}{\partial t} \right) + (\tilde{\psi}, \mathcal{L}q^h) + (\tilde{\psi}, \mathcal{L}\tilde{q}) = (\tilde{\psi}, f) \quad \forall \tilde{\psi} \in \tilde{W}, \quad (5.11b)$$

where

$$\begin{aligned} (\mathcal{L}^* \psi^h, \tilde{q}) &= \int_{\Omega'} \mathcal{L}^*(\psi^h) \tilde{q} \, d\Omega', \\ (\tilde{\psi}, \mathcal{L}q^h) &= \int_{\Omega'} \tilde{\psi} \mathcal{L}(q^h) \, d\Omega', \\ (\tilde{\psi}, \mathcal{L}\tilde{q}) &= \int_{\Omega'} \tilde{\psi} \mathcal{L}(\tilde{q}) \, d\Omega'. \end{aligned}$$

The integrals are defined by

$$\int_{\Omega'} (\cdot) \, d\Omega' \doteq \sum_{el} \int_{\Omega_{el}} (\cdot) \, d\Omega_{el}, \quad (5.13)$$

where Ω' is the union of the element interiors only, given that the subscales are assumed to be zero on the elements' boundaries.

Approximation of the sub-grid scales. The unresolved quantity \tilde{q} has not been defined yet. Like in Chapter 4, in the following the subscales are constructed by algebraic approximation following the technique described by Houzeaux et al. (2009) for linear, quadratic, and cubic elements. They take \tilde{W} as the space of vanishing functions on the boundaries of each element. These are called bubble functions (see Baiocchi et al. (1993); Brezzi et al. (1997)). By incorporating the time-dependent term of Eq. (5.11b) within the second inner product and by re-arranging the terms, the equation for the subgrid scales \tilde{q} ,

$$\int_{\Omega'} \tilde{\psi} \mathcal{L}(\tilde{q}) d\Omega' = \int_{\Omega'} \tilde{\psi} [f - \mathcal{L}(q)] d\Omega' \quad \forall \tilde{\psi} \in \tilde{W}, \quad (5.14)$$

is found. The equivalent strong form of (5.14) is

$$\mathcal{L}(\tilde{q}) = f - \mathcal{L}(q^h) = -R(q^h), \quad (5.15)$$

where $R(q^h)$ is the residual of the original equation. For the solution of (5.15) we first define \tilde{q} as a function of the bubbles $b(x)$; we have $\tilde{q} = -b(x)R(q^h)$ that is plugged into (5.15) to find the differential problem

$$\mathcal{L}(b(x)R(q^h)) = R(q^h). \quad (5.16)$$

Eq. (5.16) is solved for b with Dirichlet boundary conditions $b(x_1) = 0$ and $b(x_2) = 0$ on every element $\Omega = [x_1, x_2]$ of length $h = |x_1 - x_2|$. By thinking that $R(q^h)$ is always known from the previous time-step, we consider it constant and take out of the $\mathcal{L}(\cdot)$ operator so that $\mathcal{L}(b(x)) = 1$. For the one-dimensional steady-state advection-diffusion equation³ the problem is:

$$\mathcal{L}(b(x)) \doteq u b_x(x) - \nu b_{xx}(x) = 1, \quad (5.17)$$

with exact solution

$$b(x) = \frac{x}{u} + \frac{h}{u} \frac{1 - e^{xu/\nu}}{e^{hu/\nu}}. \quad (5.18)$$

Now that we computed the bubble functions along the element length, we construct the stabilization parameter τ as the mean value of $b(x)$ along each element. We have

³The bubbles are computed only once, out of the time loop; only if they depend on non-constant coefficients, they are computed once per time-step. In either case, the problem to be solved is not time-dependent.

that

$$\tau = \frac{1}{|x_1 - x_2|} \int_{x_1}^{x_2} b(x) dx. \quad (5.19)$$

Integration of $b(x)$ (5.18) in the interval $[x_1, x_2] = [0, h]$ yields the following definition of τ :

$$\tau = \frac{h}{2u} \left(\coth(Pe_k) - \frac{1}{Pe_k} \right), \quad (5.20)$$

where

$$Pe_k = \frac{uh}{2\nu}$$

is the element Péclet number.

We have derived all the ingredients to define the sub-grid scales \tilde{q} by the algebraic approximation

$$\tilde{q} = \tau R(q^h). \quad (5.21)$$

Eq. (5.11b) was used as the starting point to approximate \tilde{q} . Now, by plugging (5.21) into (5.11a), the VMS stabilized Galerkin method is found and expressed as follows:

Find $q^h \in W^h$ such that

$$\left(\psi^h, \frac{\partial q^h}{\partial t} \right) + a(\psi^h, q^h) + \int_{\Omega_h} \mathcal{L}^*(\psi^h) \tau R(q^h) d\Omega_{el} = (f, \psi^h) \quad \forall \psi^h \in W^h. \quad (5.22)$$

Eq. (5.22) differs from Eq. (5.7) by the additional term that models the subgrid scales. The extra term is the viscous-like contribution that stabilizes the equation.

5.2.3 Intrinsic time, τ , for spectral elements

For linear elements, $h = |x_1 - x_2|$ is simply the length of the element. At higher-order, h becomes a fraction of the total element size if the internal nodes are equi-spaced. In the case of spectral elements, where the *LGL* points are unevenly distributed, the integral is computed by using h as the local distance between two consecutive points. The stabilization parameter τ is built inside the element as a function of the bubbles on every segment delimited by two consecutive nodes. This means that equation $\mathcal{L}(b(x)) = 1$ is solved on every sub-element by applying homogeneous Dirichlet b.c. at the sub-element boundaries. For example, for a second-order element with one internal node we would

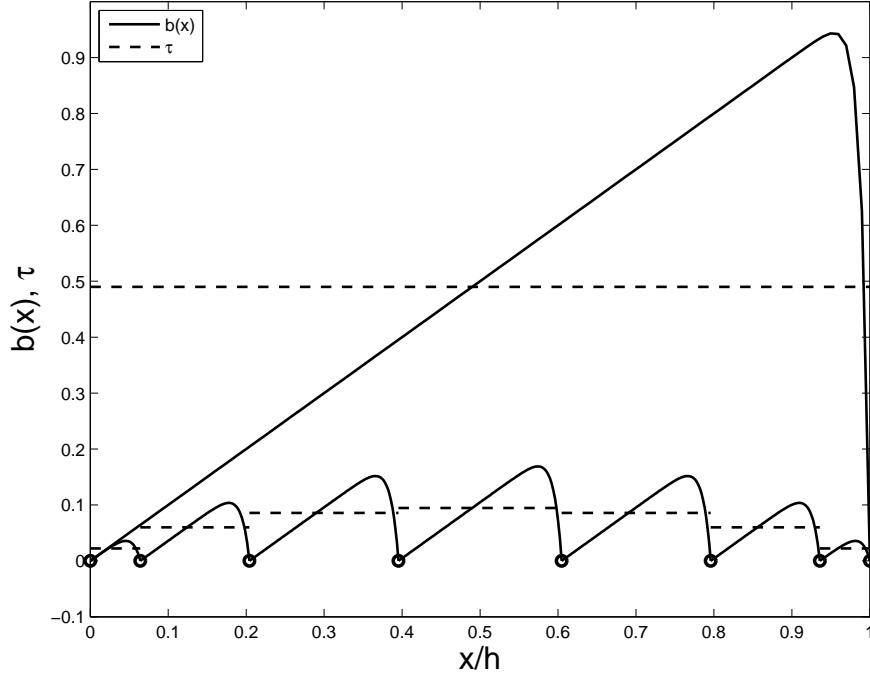


Figure 5.2: Bubbles $b(x)$ and τ for a 7th-order unitary spectral element. The biggest bubble in the plot is the bubble that a linear element would have.

solve $\mathcal{L}(b(x)) = 1$ on the two segments $[x_1, x_2]$ and $[x_2, x_3]$, respectively, by applying homogeneous b.c. as $b(x_1) = 0, b(x_2) = 0$ and $b(x_2) = 0, b(x_3) = 0$. With this, two τ 's would be computed as

$$\tau_i^{i+1} = \frac{1}{x_{lgl}(i+1) - x_{lgl}(i)} \int_{x_{lgl}(i)}^{x_{lgl}(i+1)} b(x) dx, \quad (5.23)$$

where $x_{lgl}(i+1)$ and $x_{lgl}(i)$ are the coordinates of two consecutive LGL points.

The most simple (but not unique) way of proceeding is that of taking the average value of all the sub- τ 's as the value of the full element τ . This way was used to produce the results shown below.

The uneven spacing of the element nodes is the major difference with respect to the definitions derived in previous studies. In this case, the intrinsic time is non-uniform along the element. In Fig. 5.2 the bubbles and corresponding τ 's are displayed for an element of order 7.

Explicit expression for τ on high-order elements. In this paragraph, we explicitly derive the expression for τ defined between two consecutive LGL points $[x_{lgl}(i), x_{lgl}(i+1)]$. The bubble obtained from the integration of (5.17) with boundary conditions $b(x_{lgl}(i)) = 0$ and $b(x_{lgl}(i+1)) = 0$ has expression:

$$b(x) = \frac{x}{u} - \frac{x(i+1) - x(i)}{u(e^{ux(i+1)/\nu} - e^{ux(i)/\nu})} e^{ux/\nu} - \frac{x(i)e^{ux(i+1)/\nu} - x(i+1)e^{ux(i)/\nu}}{u(e^{ux(i+1)/\nu} - e^{ux(i)/\nu})}.$$

The subscript *LGL* is omitted to keep the long expressions simple to read. The evaluation of integral (5.23) yields the expression:

$$\tau_{x(i)}^{x(i+1)} = \frac{1}{x(i+1) - x(i)} \left[\frac{x(i) - x(i+1)}{u} \left(\frac{\nu}{u} + \frac{x(i+1) - x(i)}{2} \right) - \frac{e^{ux(i+1)/\nu} (x(i) - x(i+1))^2}{e^{ux(i)/\nu} - e^{ux(i+1)/\nu}} \right].$$

When $x(i) = 0$ and $x(i+1) = h$, we have that

$$\tau_0^h = -\frac{\nu}{u^2} - \frac{h}{2u} + \frac{he^{uh/\nu}}{e^{uh/\nu} - 1},$$

from which, with little algebra, expression (5.20) is recovered.

Observations on time-dependent subgrid-scales: The time-dependent approximation (5.9) would include a contribution from the time evolution of the subscales given by $\partial_t \tilde{q}$ if the hypothesis of *quasi-static subscales* (i.e. $\partial_t \tilde{q} \approx 0$) had not been considered. Under this hypothesis, the contribution from the subgrid scales only appears in the steady part of the Galerkin approximation. If a sufficiently small time-step is used with an explicit time integrator, accuracy loss is limited under the quasi-static hypothesis. With the use of large time-steps with semi-implicit time integrators in atmospheric simulations, tracking of the subscales is hence needed. This issue is reserved for future work.

5.2.4 Spurious oscillations at layers diminishing (SOLD) methods

Methods in the form of (5.22) may produce overshoots and undershoots in the proximity of internal and boundary layers. These unwanted oscillations can be suppressed, without affecting the global solution, by adding an additional diffusive term of the form

$$(\nabla \psi^h, \tilde{\tau} \nabla q^h), \quad (5.24)$$

where consistency must be respected through a proper construction of $\tilde{\tau}$. We would like to have a method that does not modify the diffusion in the streamline direction since that is already accounted for by the stabilization term, but also avoids overdamping in the crosswind direction. The comprehensive set of tests performed by John and Knobloch

reveals that the method proposed by (Codina, 1993) is among the best ones that satisfy these conditions. In Codina (1993), $\tilde{\tau}$ is defined by:

$$\tilde{\tau} = \frac{1}{2} \max \left\{ 0, C - \frac{2\nu}{|\mathbf{u}_{||}|h_k} \right\} h_k \frac{|R(q^h)|}{\|\nabla q^h\|} \left(\mathbf{I} - \frac{\mathbf{u} \otimes \mathbf{u}}{|\mathbf{u}|^2} \right) \quad (5.25)$$

where C is a constant and $\mathbf{u}_{||}$ is the velocity component in the streamline direction. Codina suggests $C = 0.7$ for linear and bilinear elements, and $C = 0.35$ for quadratic and biquadratic elements. However, for higher order elements using LGL points, we found that the best results were obtained by setting $C = 1$, as long as h_k is selected properly in the construction of both $\tilde{\tau}$ and τ_k .

An alternative to (5.24) comes from Johnson et al. (1987) who defined the following:

$$(\tilde{\tau} \mathbf{u}^\perp \cdot \nabla \psi^h, \mathbf{u}^\perp \cdot \nabla q^h), \quad \mathbf{u}^\perp = \frac{(-w, u)}{|\mathbf{u}|}. \quad (5.26)$$

In the current work, (5.26) gives better results than (5.24), and was then used throughout. The results obtained with this technique are labeled with *DC* for *Discontinuity Capturing*.

5.2.5 First-Order Subcells (FOS)

FOS is one additional tool that can further help the suppression of Gibbs oscillations. The concept is simple and is easily coded on structured grids. If a solution has large gradients, the algorithm needs to identify the elements where the large gradients occur, and project the solution scheme to a 1st-order space. The gradient is sought with a proper error estimator. The simple physics of the advection-diffusion problems discussed below allows for the energy-norm of the gradient of the solution to be a sufficiently good estimator for the current study. As it is defined in this study, the error estimator depends on a parameter, ϵ , that may be a function of the numerical settings (e.g., grid resolution, time step). This point must be considered in the construction of FOS and in the selection of the error estimator. We did not explore this further in this study, although it is a very important issue for the best performance of FOS.

Algorithm 1 is a simple implementation of this concept within our code. We present the pseudo-code below for the sake of clarity. The method was applied to a two-dimensional advection-diffusion problem with internal and boundary layers in a skew velocity field. Results are shown in Figures 5.9-5.12 and 5.14-5.18. A detail of Figures 5.18(a,b) is presented in Fig. 5.19.

In the tests that use Algorithm 1, ϵ was set to 0.5.

5.2.6 Time discretization

So far we have described the details of space discretization. The time-dependent problem must then be discretized and integrated forward in time. Substitution of expansion

Algorithm 1 Compute the 1st-order *rhs*

Initialize ϕ^0 and φ^0

for $iel = 1$ to $nelem$ **do**

 Check if the element contains a sufficiently large gradient

if iel is s.t. $\|\nabla q^h\|_2 > \epsilon$ **then**

 Treat element iel as a sub-domain with $(ngl - 1) \times (ngl - 1)$ sub-elem. (*isubel*):

for $isubel = 1$ to p^d **do**

 Create *mass* and *rhs* using 1st-order basis functions and integration rule

end for

else

 Create *rhs* for the high-order spectral element.

end if

end do

(3.2) into the Galerkin problem (5.22), we obtain the matrix problem of $(N + 1)$ scalar equations written as:

$$\mathbf{M} \frac{\partial \mathbf{q}}{\partial t} = \mathbf{R}(\mathbf{q}), \quad (5.27)$$

where $\mathbf{R}(\mathbf{q}) = \mathbf{G}(\mathbf{q}^h) + \mathbf{b}(\mathbf{q}^h, \tilde{\mathbf{q}})$ and

1. \mathbf{q}^h is the vector of $N + 1$ time-dependent unknowns.
2. \mathbf{M} is the global mass matrix of dimension $(N + 1) \times (N + 1)$ obtained by the assembly (also referred to as DSS for Direct Stiffness Summation) of the local element matrices

$$\mathbf{M}^{kl} = \int_{\Omega_{el}} \psi_k^h \psi_l^h d\Omega^h, \quad k, l \in 1, \dots, n_{points}^{el}$$

By construction and by the quadrature rule in use, \mathbf{M} in this case is diagonal without having to rely on lumping techniques (see Chapter 4).

3. \mathbf{G} is the differentiation matrix composed of the Galerkin terms constructed by assembly of the local matrices

$$\mathbf{G}^{kl} = - \int_{\Omega_{el}} \mathbf{u} \cdot \psi_k^h \nabla \psi_l^h d\Omega_{el} - \int_{\Omega_{el}} \nabla \psi_k^h \cdot \nabla \psi_l^h d\Omega_{el}, \quad k, l \in 1, \dots, n_{points}^{el}$$

and

4. \mathbf{b} results from assembly of the element stabilization. We have:

$$\mathbf{b}(\mathbf{q}^h, \tilde{\mathbf{q}}) = \sum_{k=1}^{n_{el}} \int_{\Omega_{el}} \mathcal{L}^*(\psi^h)_k \tau R(\mathbf{q})^h d\Omega_{el}, \quad k \in 1, \dots, n_{points}^{el}.$$

High-order time integration. Time integration is constructed by the strong-stability preserving 5-stage, 3rd-order Runge-Kutta (SSPRK) (RK35) method of Spiteri and Ruuth (2002). SSP methods avoid the production of additional oscillations or excessive damping of the solution. Given system (5.27) as

$$\frac{\partial \mathbf{q}}{\partial t} = \mathbf{M}^{-1} \mathbf{R}(\mathbf{q}), \quad (5.28)$$

the solution vector using SSPRK with $s = 1, \dots, N_s = 5$ stages is given as

$$\mathbf{q}^s = \alpha_0^s \mathbf{q}^n + \alpha_1^s \mathbf{q}^{s-1} + \beta^s \Delta t \mathbf{M}^{-1} \mathbf{R}(\mathbf{q}^{s-1}),$$

where $\mathbf{q}^0 = \mathbf{q}^n$ and $\mathbf{q}^{N_s} = \mathbf{q}^{n+1}$. α and β are tabulated constants.

5.3 Mass conservation

For problems in geophysical fluid dynamics and, more specifically, atmospheric simulations, mass conservation of tracers is an important factor. In this section we address this issue and illustrate how the algorithm reported in this chapter behaves in this respect. Under the suitable hypothesis of a divergence-free flow, Eq. (5.1) for the transport of the mixing ratio

$$q = \frac{\rho_{tracer}}{\rho},$$

where ρ_{tracer} and ρ are, respectively, the densities of the tracer and of the advecting fluid, is derived from the equation of conservation of mass of the tracer⁴

$$\frac{\partial \rho q}{\partial t} + \nabla \cdot (\mathbf{u} \rho q) = 0 \quad (5.29)$$

by elimination of ρ from equation (2.1) that we report here again:

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\mathbf{u} \rho) = 0. \quad (5.30)$$

In (5.29) and (5.30) the quantities that are conserved are $\rho_{tracer} = \rho q$ and ρ , but not q . In the case of $\rho(t) = \rho(t=0) = constant$ in a non-divergent flow, Eq. (5.1) is equivalent to

$$\frac{\partial q}{\partial t} + \nabla \cdot (\mathbf{u} q) = 0. \quad (5.31)$$

⁴For simplicity, the diffusion and source terms were dropped.

This allows the equal treatment of ρq and q in the modeling of the physical system [Nair and Lauritzen \(2010\)](#). Eq. (5.1) is of common use within atmospheric models (e.g., [Klemp and Wilhelmson \(1978\)](#); [Nair and Lauritzen \(2010\)](#); [Gaberšek et al. \(2012\)](#)). We track mass during the simulations to evaluate the amount of mass loss that we would run into if using SEM+VMS knowing that no method will conserve for this equation since it is not in conservation form or a conservation law. For this computation, we use the advection of a square wave in a periodic channel and report the results in *Case 5*.

5.4 Numerical tests

The algorithms discussed throughout this chapter are tested by using standard one- and two-dimensional problems. The problems are organized according to the nomenclature listed below:

- *Case 1*: 1D Steady-state homogeneous advection-diffusion
- *Case 2*: 1D Steady-state advection-diffusion with source
- *Case 3*: 2D Steady-state advection-diffusion with internal and boundary layers
- *Case 4*: 2D Time-dependent advection-diffusion with “L”-shaped discontinuity
- *Case 5*: 2D Time-dependent advection of a sharp tracer in a doubly periodic channel
- *Case 6*: 2D Smooth solid-body rotation - Convergence study
- *Case 7*: Pseudo-3D advection in a neutrally stratified atmospheric flow

Case 1. One-dimensional steady-state advection-diffusion. q^h is propagated with constant velocity $u = 1 \text{ m s}^{-1}$ and diffusivity $\nu = 1/512 \text{ m}^2 \text{ s}^{-1}$ first on two elements of order $p = 10$ (Fig. 5.3), and then on four elements of order $p = 12$ (Fig. 5.4). The domain is the line segment $\Omega = [-1, 1]$ with Dirichlet boundary conditions $q^h(-1) = 0$ and $q^h(1) = 1$. We compared the filtered against the stabilized solution and observe a decrease of oscillations and undershoots. Also, at higher order and finer resolution, the capabilities of the filter are clearly being challenged by the presence of the boundary layer at $x = 1$. At the same time, small oscillations near the nodes of the element by the boundary layer are not completely suppressed by the stabilized method either; hence, additional localized smoothing is necessary.

Case 2. Steady state advection-diffusion with source term $f = 1$. q^h is propagated with constant velocity $u = 1 \text{ m s}^{-1}$ and two different diffusivities: $\nu = 5 \times 10^{-3} \text{ m}^2 \text{ s}^{-1}$ and $\nu = 5 \times 10^{-2} \text{ m}^2 \text{ s}^{-1}$. The domain is the line segment $\Omega = [0, 1]$ and homogeneous Dirichlet boundary conditions are imposed. The domain is subdivided into two elements

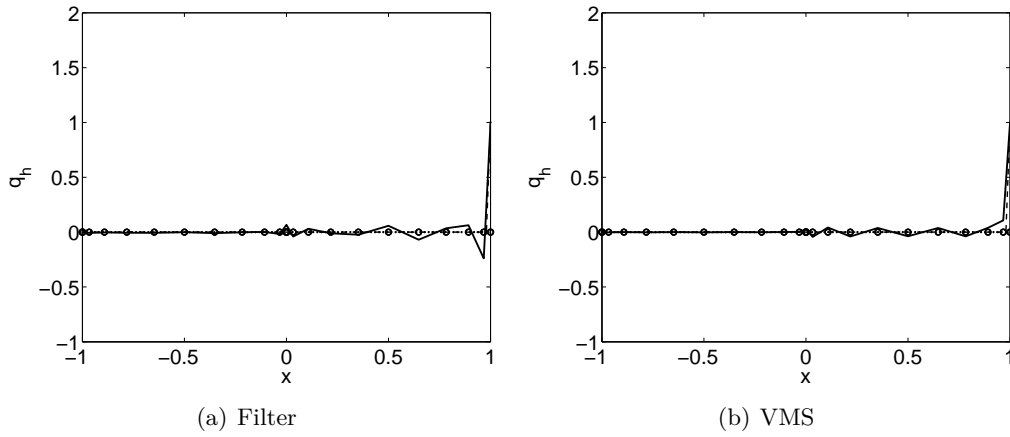


Figure 5.3: *Case 1*. 1D steady-state AD problem with $\nu = 1/512 m^2 s^{-1}$. Solution obtained with 2 elements of order 10. (a) filtering, and (b) using VMS. The exact solution is dashed. The circles indicate the grid points.

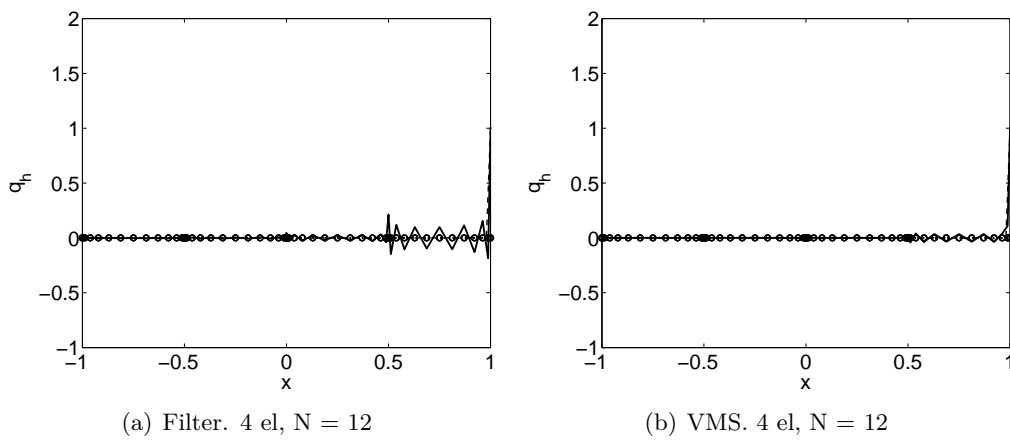


Figure 5.4: *Case 1*. 1D steady-state AD problem with $\nu = 1/512 m^2 s^{-1}$. Solution obtained with 4 elements of order 12. (a) filtering, and (b) using VMS. The exact solution is dashed. The circles indicate the grid points.

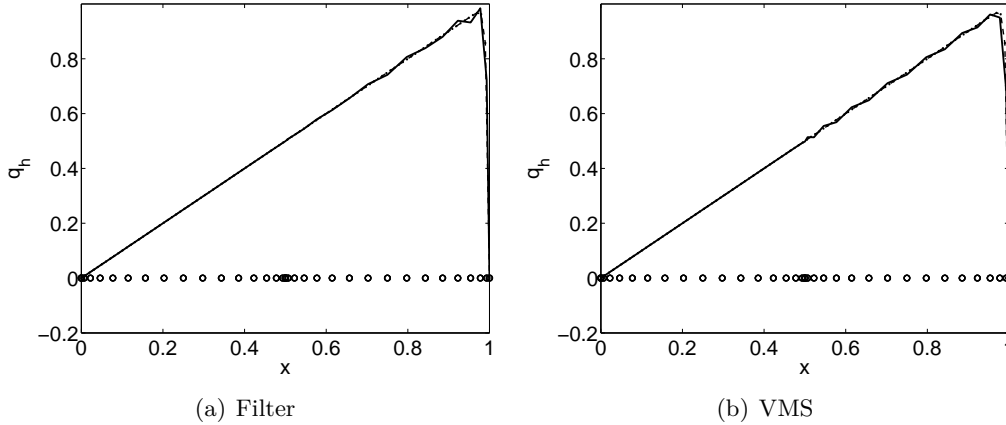


Figure 5.5: *Case 2.* 1D steady-state AD problem with Source. $\nu = 5 \times 10^{-3} m^2 s^{-1}$. Solution obtained with 2 elements of order 16. (a) filtering, and (b) using VMS. The exact solution is dashed. The circles indicate the grid points.

of order $p = 16$ and runs are compared using filtered SE (left plots in Figures 5.5 and 5.6), and VMS (right plots). For the smooth problem in Fig. 5.5, we observe a very similar behavior of the solution between the two different cases. The results are comparable to the ones obtained by Houzeaux et al. (2009) using τ for quadratic and cubic elements.

Case 3. Standard steady advection-diffusion skewed to the mesh (e.g., Codina (1993)): a discontinuity is propagated with constant velocity $\mathbf{u} = (1, -2)m s^{-1}$ and diffusivity $\nu = 10^{-8} m^2 s^{-1}$ in the unit square $\Omega = [0, 1] \times [0, 1]$. The initial configuration with prescribed Dirichlet boundary conditions is defined by

$$q^h = \begin{cases} 1 & \text{if } y = 1, \\ 1 & \text{if } x = 0 \text{ and } y \geq 0.7, \\ 0 & \text{otherwise.} \end{cases}$$

and is shown in Fig. 5.7.

In Figures 5.8-5.12 we illustrate the run of the same case with a different number of elements and order of the interpolating polynomials. For direct comparison of our solution with the ones in the existing literature of finite elements, we first run the test with linear elements ($p = 1$) and present the results in Fig. 5.8. The multiscale solution of this problem (see Fig. 5.8a) shows important boundary and internal layers that are damped by the discontinuity capturing techniques of Section 5.2.4. The application of the discontinuity capturing scheme greatly improves the solution and yields monotonicity (see Fig. 5.8b). In Fig. 5.9 we maintained the same number of nodes of the previous run, but increased to 4th the order of interpolation to assess the algorithm in the context of this chapter (i.e. 50 elements of order 4 were used instead of 200 elements of order 1). The similar behavior of the solution with respect to the 1st-order polynomial run

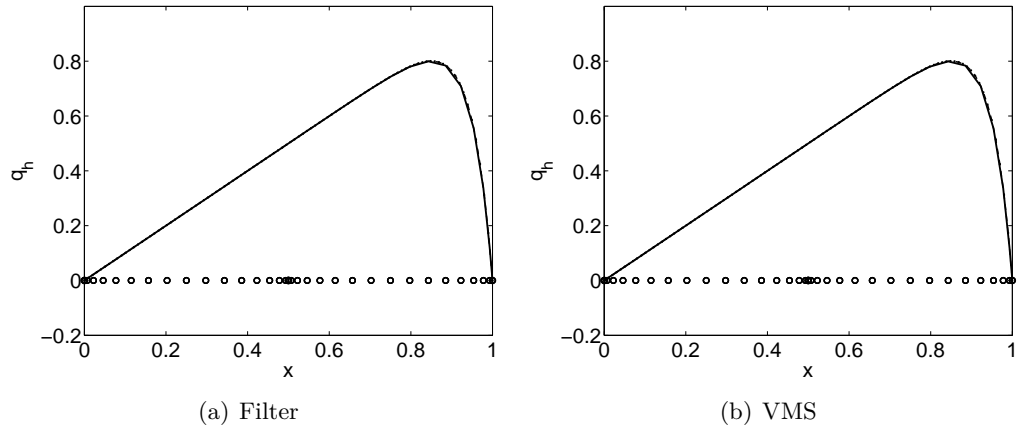


Figure 5.6: *Case 2*. 1D steady-state AD problem with Source. $\nu = 5 \times 10^{-2} m^2 s^{-1}$. Solution obtained with 2 elements of order 16. (a) filtering, and (b) using VMS. The exact solution is dashed. The circles indicate the grid points.

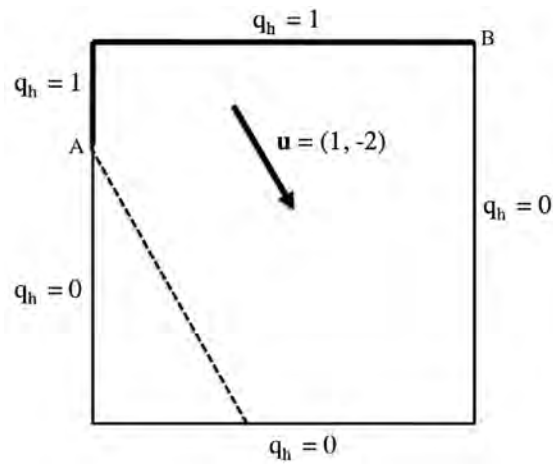


Figure 5.7: *Case 3*: initial configuration of the steady-state problem

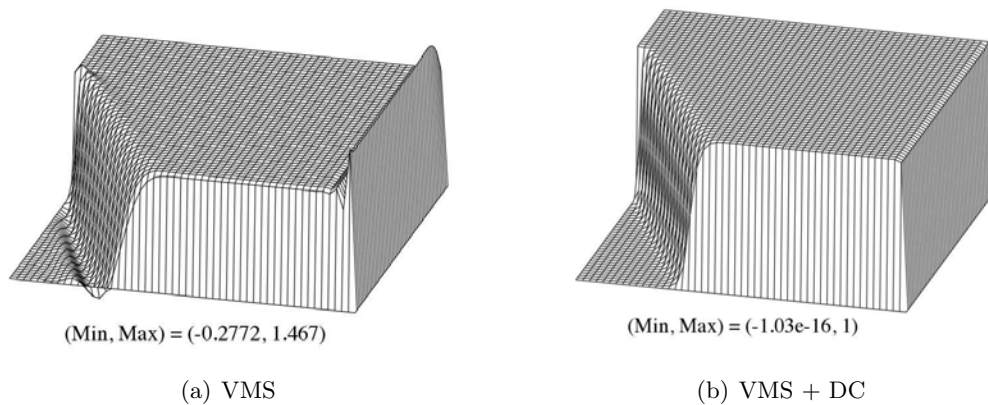


Figure 5.8: *Case 3*: steady-state solution on 200×200 1^{st} -order elements. (For plotting only, the data are interpolated to a 50×50 node grid using Octave (Eaton, 2002)).

suggests that the residual-based methods as implemented in this study may not be sensitive to the distribution of the interpolation nodes within the elements edges. As it appears in Figures 5.9c, the behavior is completely analogous to the previous run. However, monotonicity is lost in two singular nodes: with reference to Fig. 5.9c, the 4^{th} -order solution is smooth and monotone everywhere except for the nodes represented as points A and B in Fig. 5.7. This is not surprising: at A and B the tracer is leaving the boundary with a skew angle; an incorrect imposition of boundary conditions at these nodes may be causing the problem. The numerical singularity at this points should be addressed but it will not be done in the current work. These are fully suppressed by applying the FOS algorithm described above, as it is shown in Fig. 5.9d.

Decreasing the number of computational nodes by doubling the order from 4 to 8 and setting the number of elements to 10 in x and z , even with a discontinuity capturing term, the solution starts to lose monotonicity. This appears in Figures 5.11 and 5.12, where extrema get larger than in the previous cases. This problem shows that the construction of the stabilizing parameter τ should include information on the order of the interpolating polynomial.

For a better view of the problem, in Figures 5.10 and 5.12 we present a vertical slice of the solution. The boundary layers are evident. Their damping, however, is clear if *VMS*, *DC*, and *FOS* are applied.

Case 4. Transient advection-diffusion of an L-shaped discontinuity in a flow where $\nu = 10^{-6} m^2 s^{-1}$ and the velocity \mathbf{u} of magnitude $|\mathbf{u}| = 0.5\sqrt{2} m s^{-1}$ is at 45° with respect to the axis (x, z) . The initial configuration is shown in Fig. 5.13.

The convex shape of the sharp discontinuity makes this problem more challenging than the previous case (Bazilevs et al., 2007). It is chosen to analyze robustness and accuracy of the algorithm. Runs were performed at two different resolutions and two different orders of interpolating polynomials. In particular we have: approx. 100 points

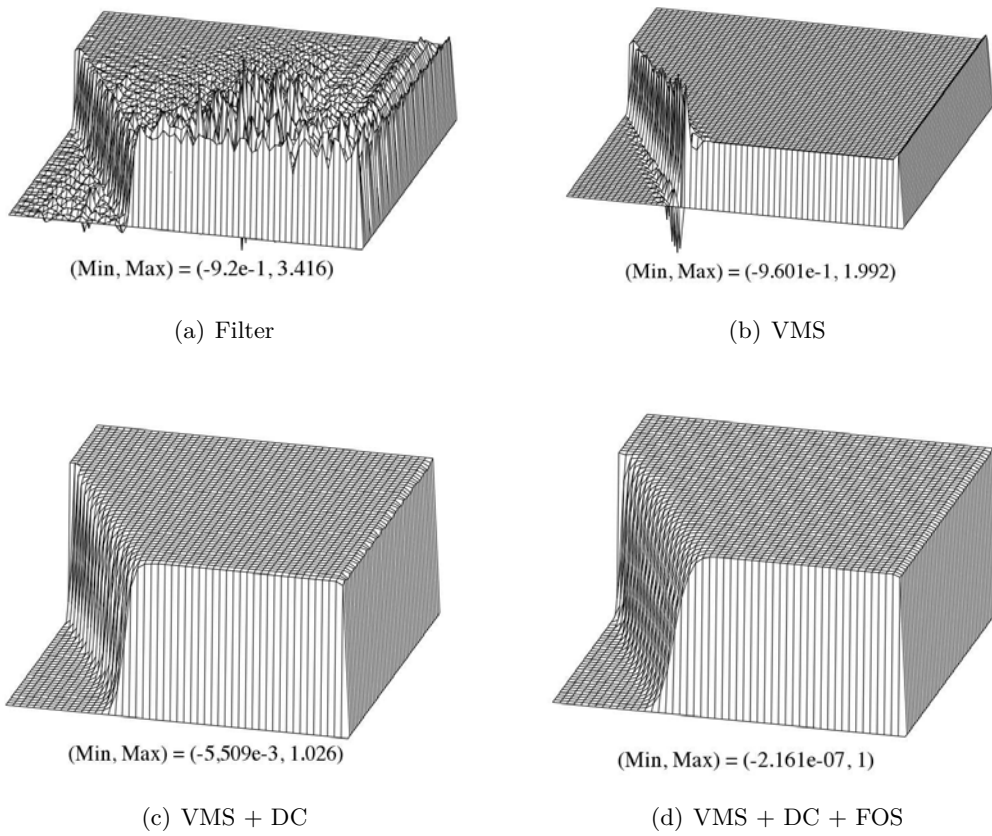


Figure 5.9: *Case 3*: steady-state solution on 50×50 4^{th} -order elements.

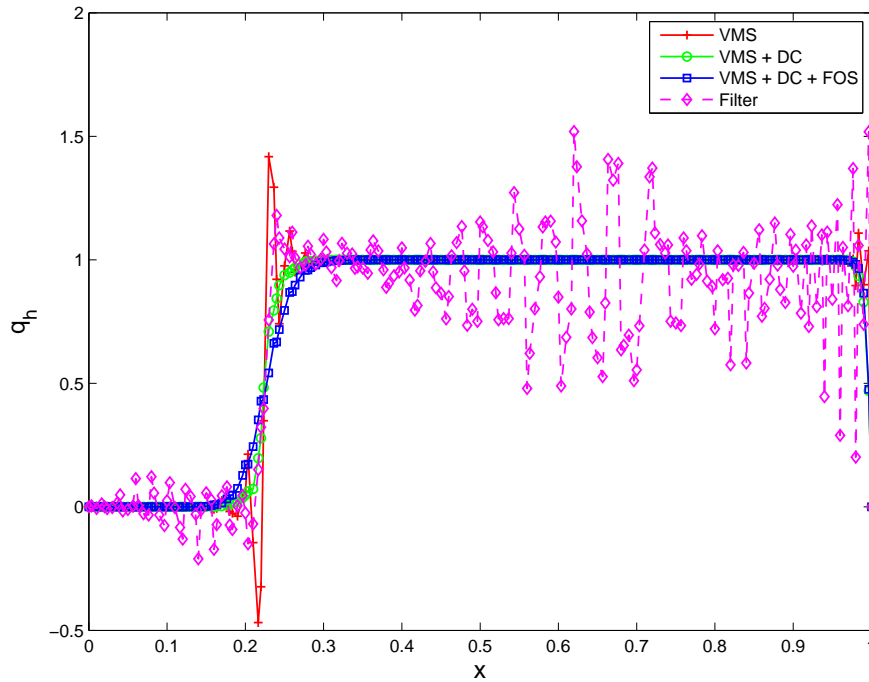


Figure 5.10: *Case 3*: steady-state solution on 50×50 4^{th} -order elements. Vertical slice at $z = 0.3$

per side using 25×25 4^{th} -order elements (Fig. 5.14), and 12×12 8^{th} -order elements (Fig. 5.15); and approx. 200 points per side using 50×50 4^{th} -order (Fig. 5.16), and 25×25 8^{th} -order (Fig. 5.17). In the figures, *Filter* means that the SEM solution was filtered at every time-step. *VMS* and/or *DC* indicate that the SEM solution is stabilized by VMS with or without a discontinuity capturing term (DC). *VMS + DC + FOS* indicates the contribution of FOS as well. Positivity is not preserved in the solution obtained with a filter. The sharp front, in fact, makes the filter inappropriate. However, similarly to the steady advection-diffusion test *Case 3*, the VMS-stabilized solution of this problem is characterized as well by the formation of internal layers that run along the edges of the tracer in the direction of the flow (see, e.g., Fig. 5.14b), and VMS is not sufficient to preserve monotonicity unless it is supplemented by the additional *DC* term defined in (5.26). This effect is displayed in Figures 5.14, 5.15, 5.16, and 5.17.

The consideration made for problem *Case 3* on the singular peaks that form at the nodes where the tracer leaves the boundary at an angle, applies here at nodes A and B of Fig. 5.13. This is visible in Fig. 5.18 obtained by slicing the tracer along $z = 0$ in Figures 5.16 and 5.17, respectively. The problem is solved by the application of FOS.

As the order of interpolation is increased from 4^{th} to 8^{th} , the smooth solution begins to lose positivity. As interpreted for *Case 3*, the solution is clearly being affected when the interpolation nodes are densely clustered towards the boundaries of the elements, as is the case for higher order.

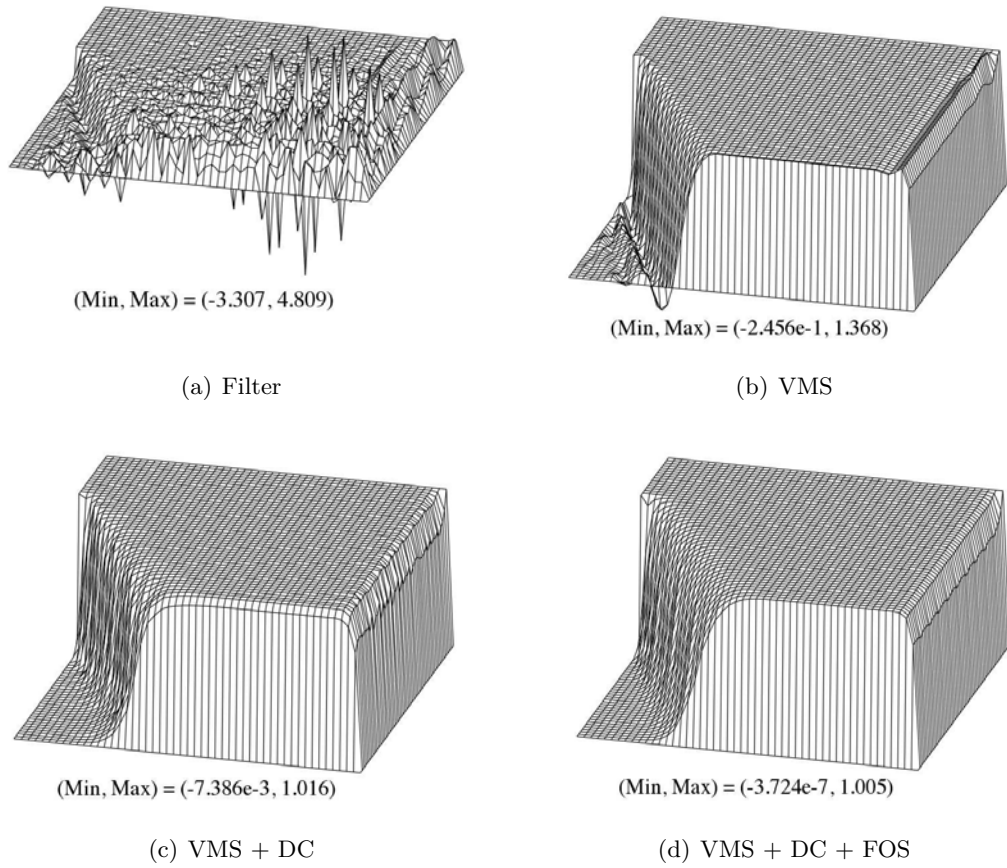


Figure 5.11: *Case 3*: steady-state solution on 10×10 8^{th} -order elements.

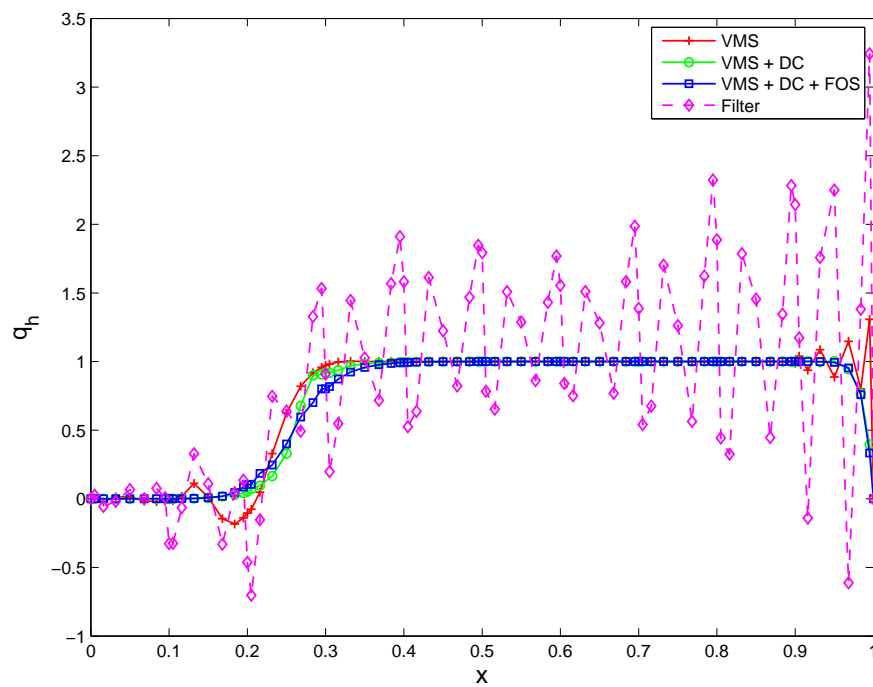


Figure 5.12: *Case 3*: steady-state solution on 10×10 8^{th} -order elements. Vertical slice at $z = 0.3$

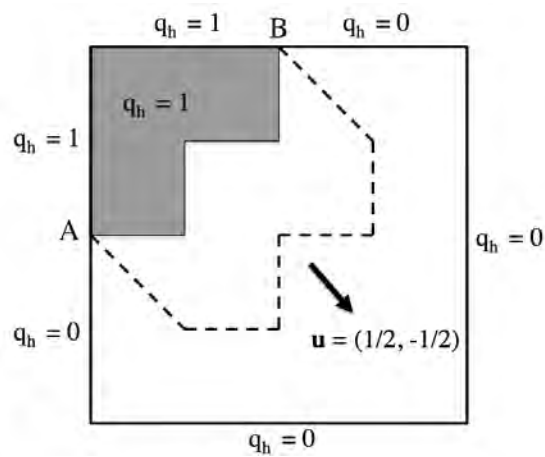


Figure 5.13: *Case 4*: initial configuration of the L-shaped problem

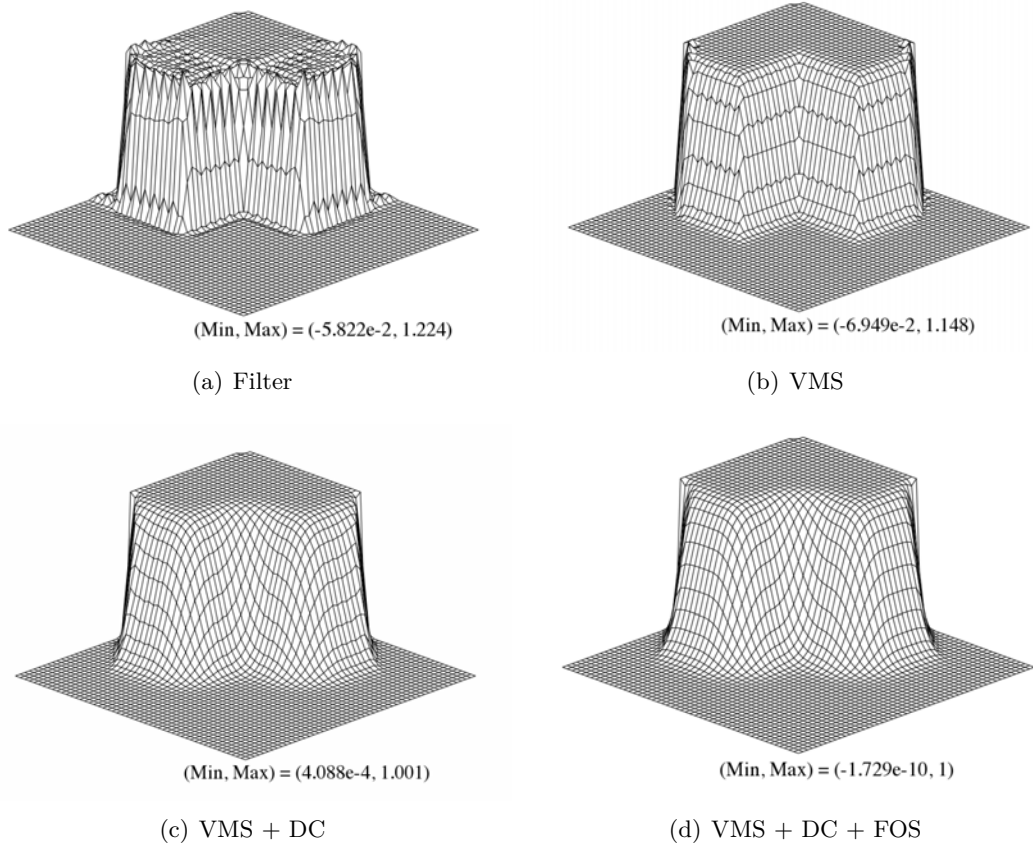


Figure 5.14: Case 4: 4th – order 25×25. $t = 0.25$ s.

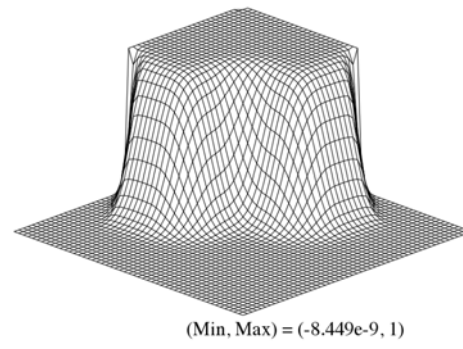
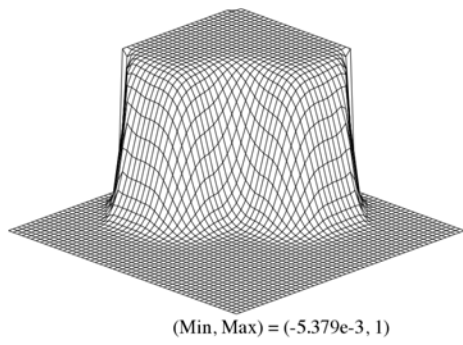
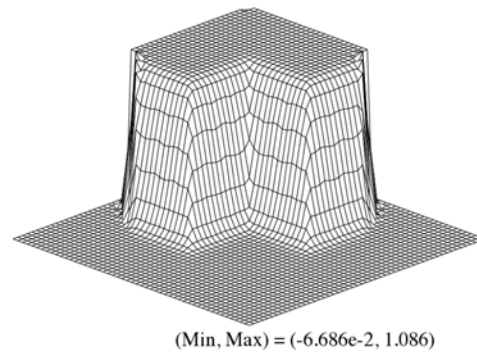
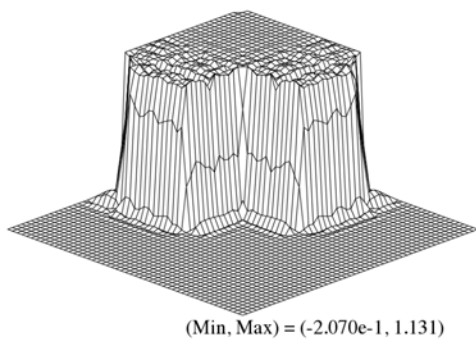


Figure 5.15: Case 4: 8th - order 12×12. $t = 0.25$ s.

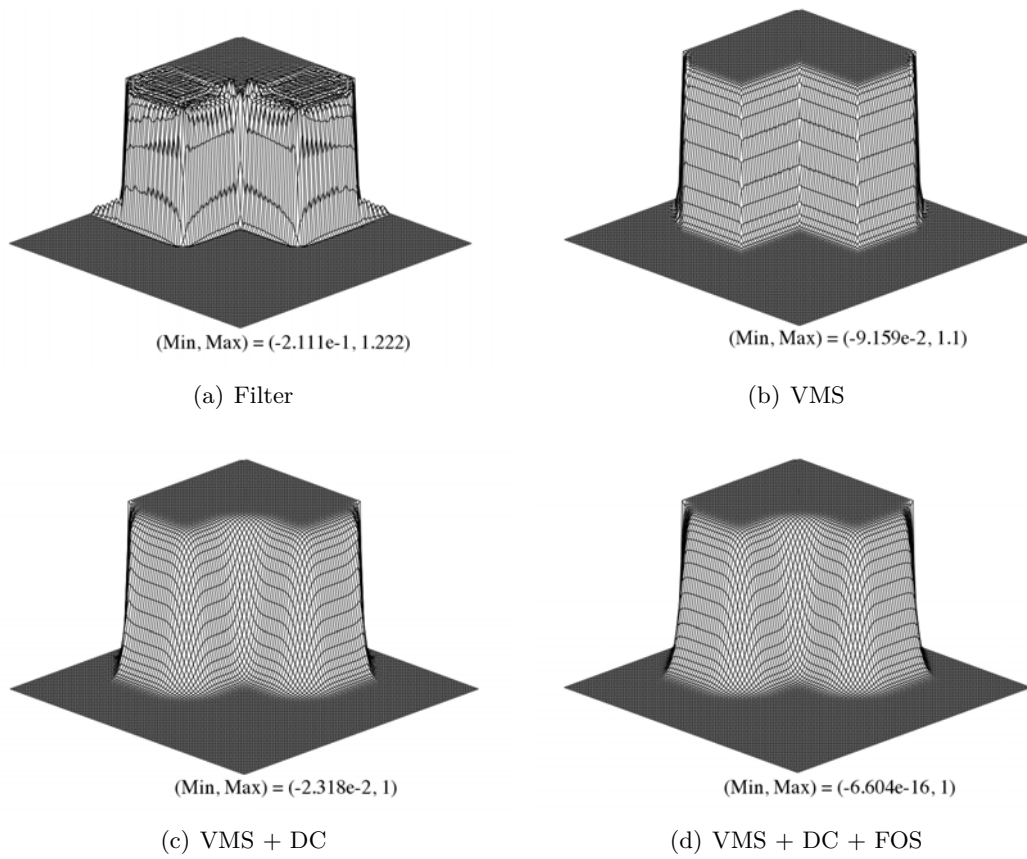
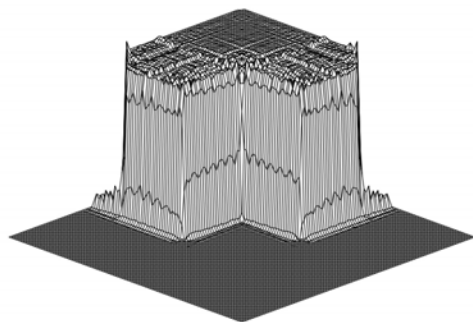
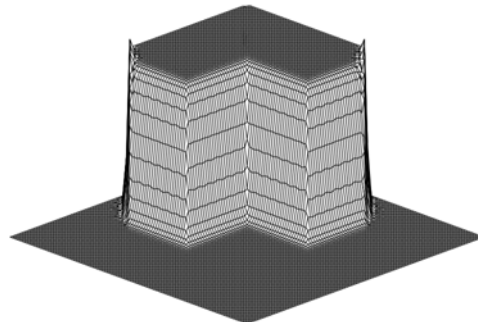


Figure 5.16: Case 4: 4th - order 50×50 . $t = 0.25$ s.



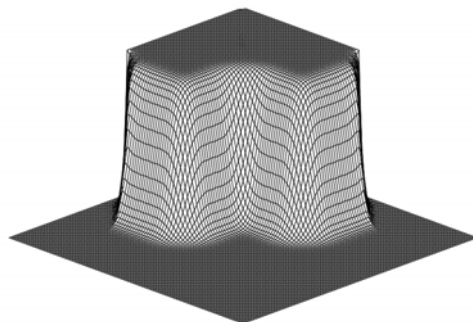
(Min, Max) = (-1.708e-1, 1.158)

(a) Filter



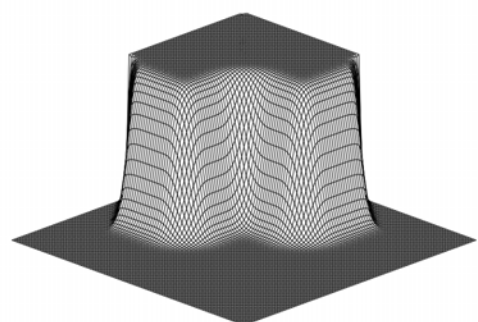
(Min, Max) = (-1.337e-1, 1.149)

(b) VMS



(Min, Max) = (-7.071e-2, 1.009)

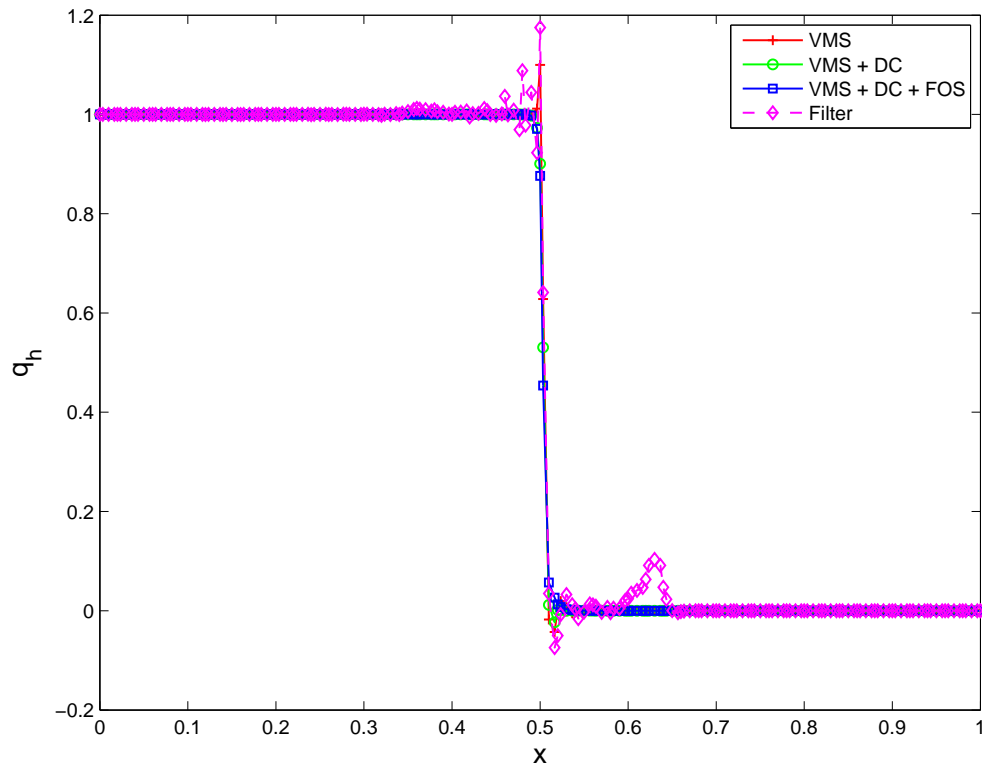
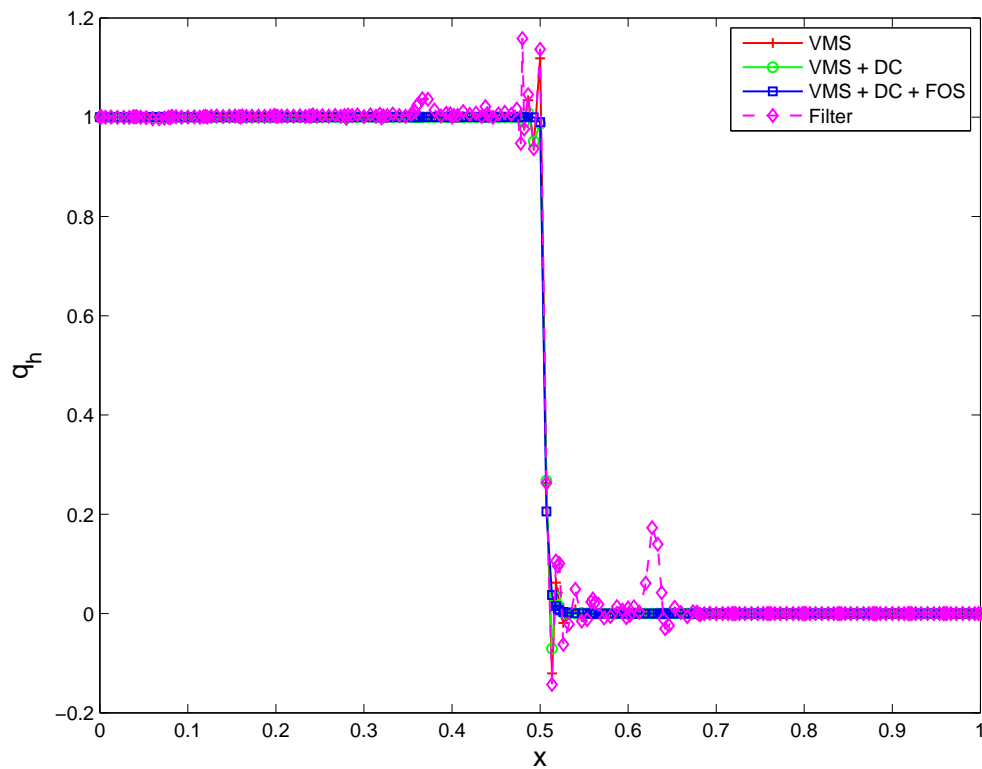
(c) VMS + DC

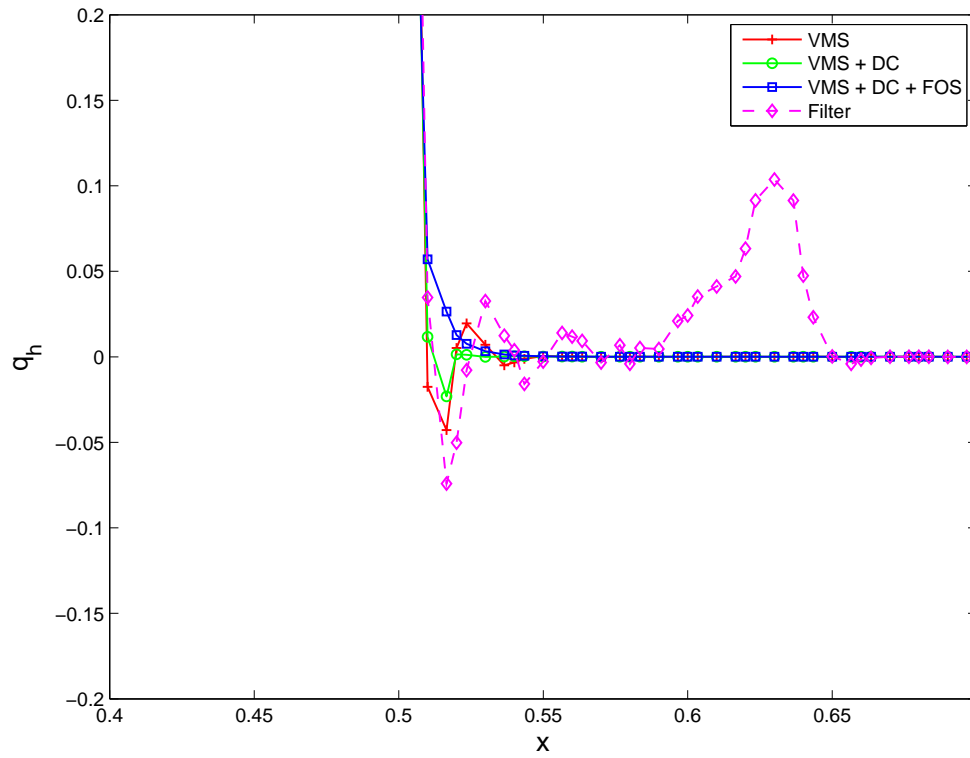
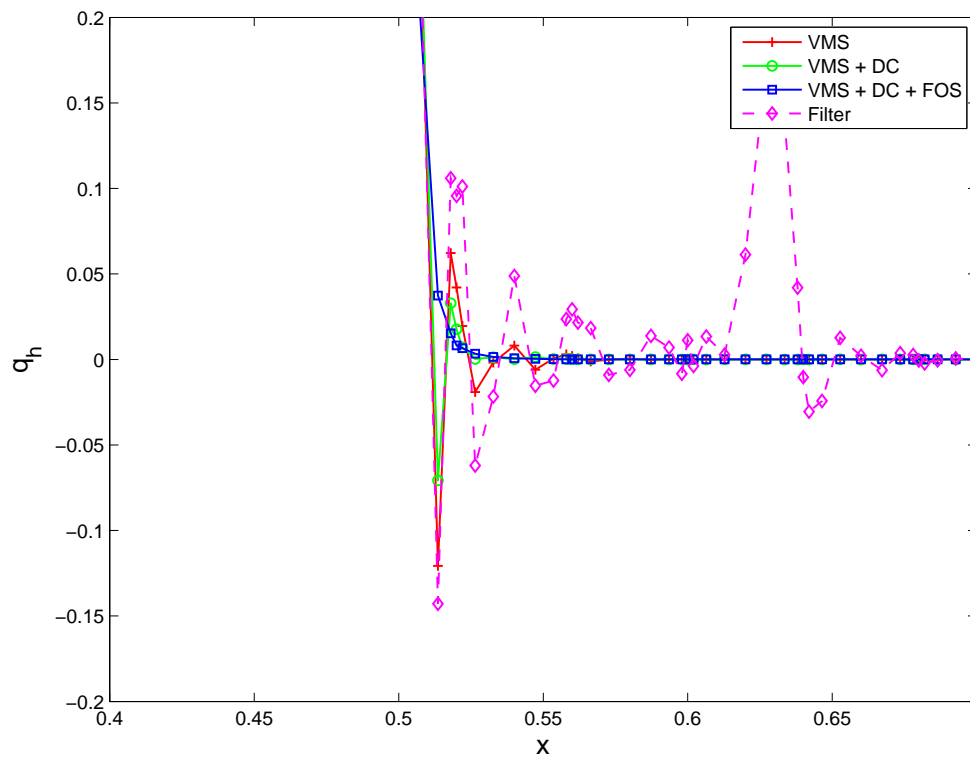


(Min, Max) = (-9.330e-9, 1)

(d) VMS + DC + FOS

Figure 5.17: Case 4: 8th - order 25×25. $t = 0.25$ s.

(a) 4th - order 50×50 (b) 8th - order 25×25 Figure 5.18: Case 4: Vertical slice at $z = 0.0$.

(a) 4th – order 50×50(b) 8th – order 25×25Figure 5.19: *Case 4*: detail of Fig. 5.18. Region with undershoots. Vertical slice at $z = 0.0$.

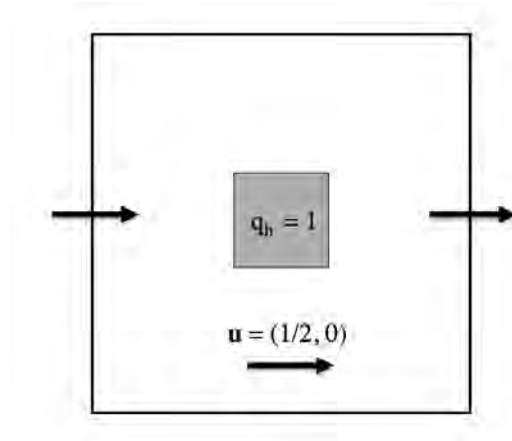


Figure 5.20: *Case 5*: initial configuration of the pure advection problem.

Case 5. Linear advection of a 2D square wave along x in the periodic domain $\Omega = [0, 1] \times [0, 1]$: the tracer is transported with velocity $\mathbf{u} = (1/2, 0) \text{ m s}^{-1}$ for one periodic revolution along x . The initial concentration $q^h = 1$ is centered at $(x_c, z_c) = (0.5, 0.5)$ (Fig. 5.20). The computational finite domain consists of 11×11 quadrilaterals of order 11.

As in the steady case, Figures 5.21-5.23 display improvement of the solution in terms of monotonicity when the VMS method is used instead of the filter. The combination of VMS and filtering is not recommended (result not shown); although VMS alone controls the over- and under-shootings along the streamlines, the addition of the filter at the end of every time step degrades positivity in the neighborhood of large gradients.

In Figures 5.22 and 5.23 we present the streamline and crosswind sections of the solution obtained by slicing the tracer along $z = 0.5$ and $x = 0.5$, respectively. Unlike the previous problems characterized by internal and boundary layers, for pure advection the VMS preserves the maximum and minimum concentrations $q_{max}^h = 1$, and $q_{min}^h = 0$ and is free of spurious oscillations. As a point of comparison, we present the result of classical artificial-viscosity in Figures 5.21-5.23d.

Testing mass conservation. Because of the periodic boundary conditions applied here, we compute mass conservation properties for this test. At every time-step, the total mass loss of ρq (for $\rho = 1$) is computed as

$$M_{loss}(t) = \frac{\int_{\Omega} (\rho q(t) - \rho q(t_0)) d\Omega}{\int_{\Omega} \rho q(t_0) d\Omega}, \quad (5.32)$$

where Ω is the domain volume and t_0 indicates the values at the initial time. Fig. 5.24 shows the evolution of the mass loss that occurs during 100 revolutions around the periodic channel of *Case 5*. 100 revolutions happen in 100 s.

Although in Fig. 5.24 there seems to be an asymptotic trend to an upper bound,

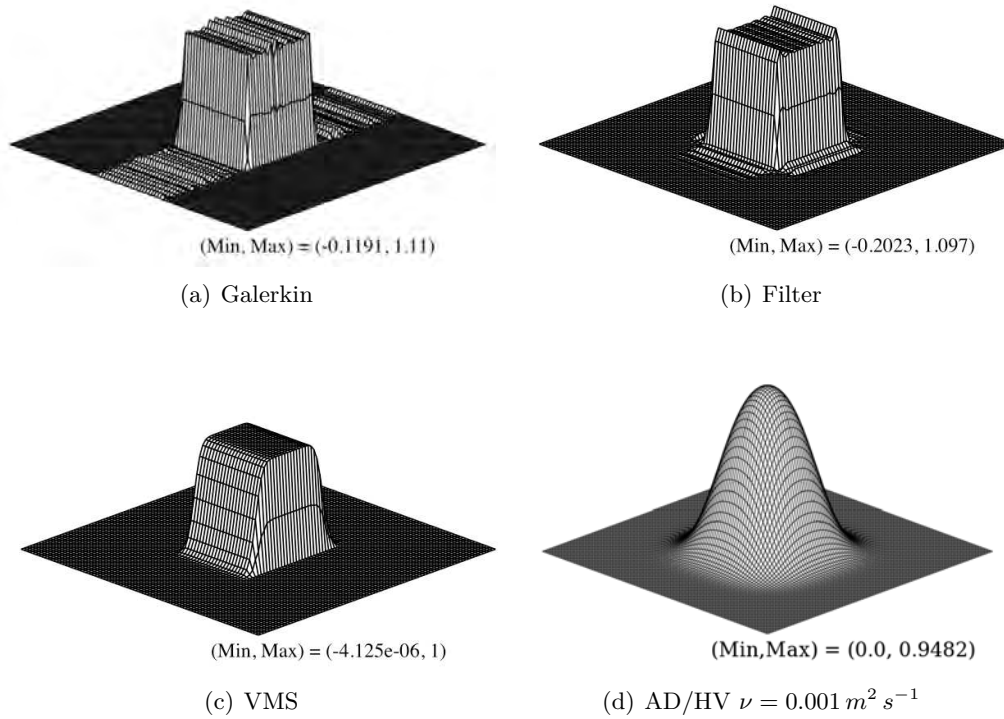


Figure 5.21: *Case 5*: Surface plot of the concentration field: $\Delta t = 0.001 \text{ s}$ (except for HV: $\Delta t = 0.0002 \text{ s}$), 11×11 elements with 11^{th} order polynomials. Results at $t = 2.0 \text{ s}$ (after 1 periodic revolution along x).

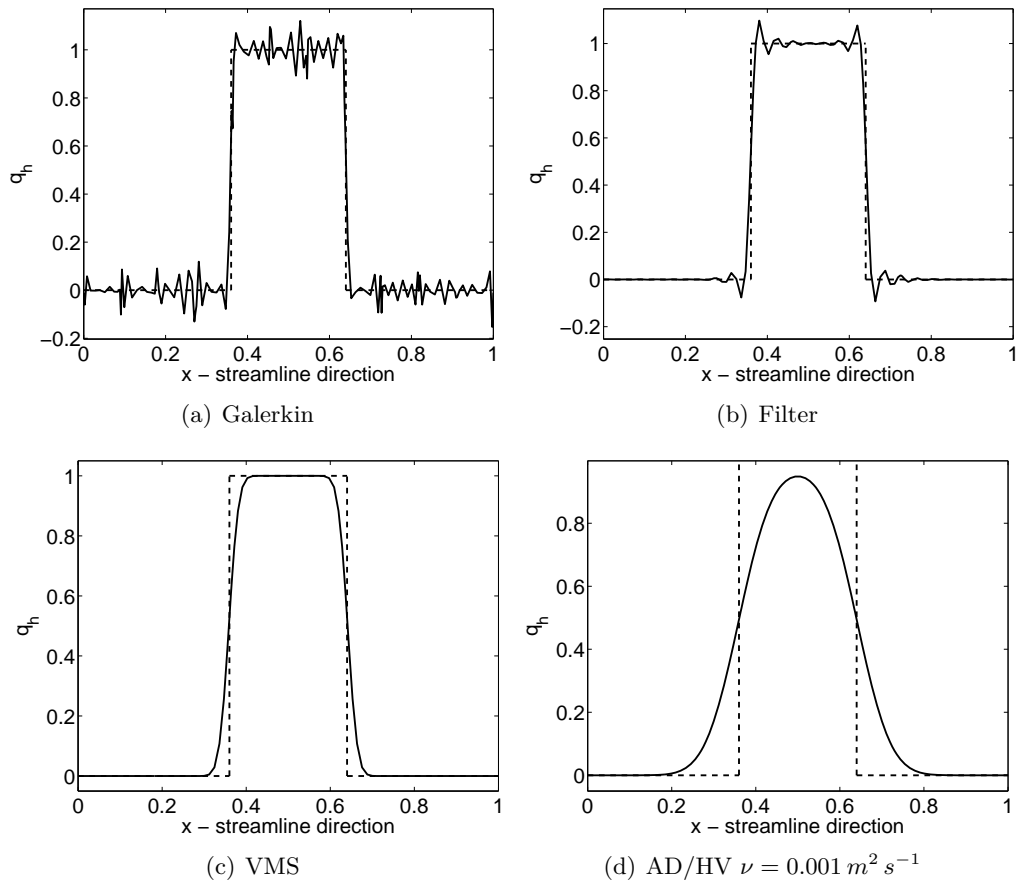


Figure 5.22: *Case 5: Streamline cut at 0.5 m in the y-direction. $\Delta t = 0.001 \text{ s}$, 11×11 elements with 11^{th} order polynomials. Results at $t = 2 \text{ s}$ (after 1 periodic revolution along x). Solid line indicates the computed solution. The dashed line is the analytic solution.*

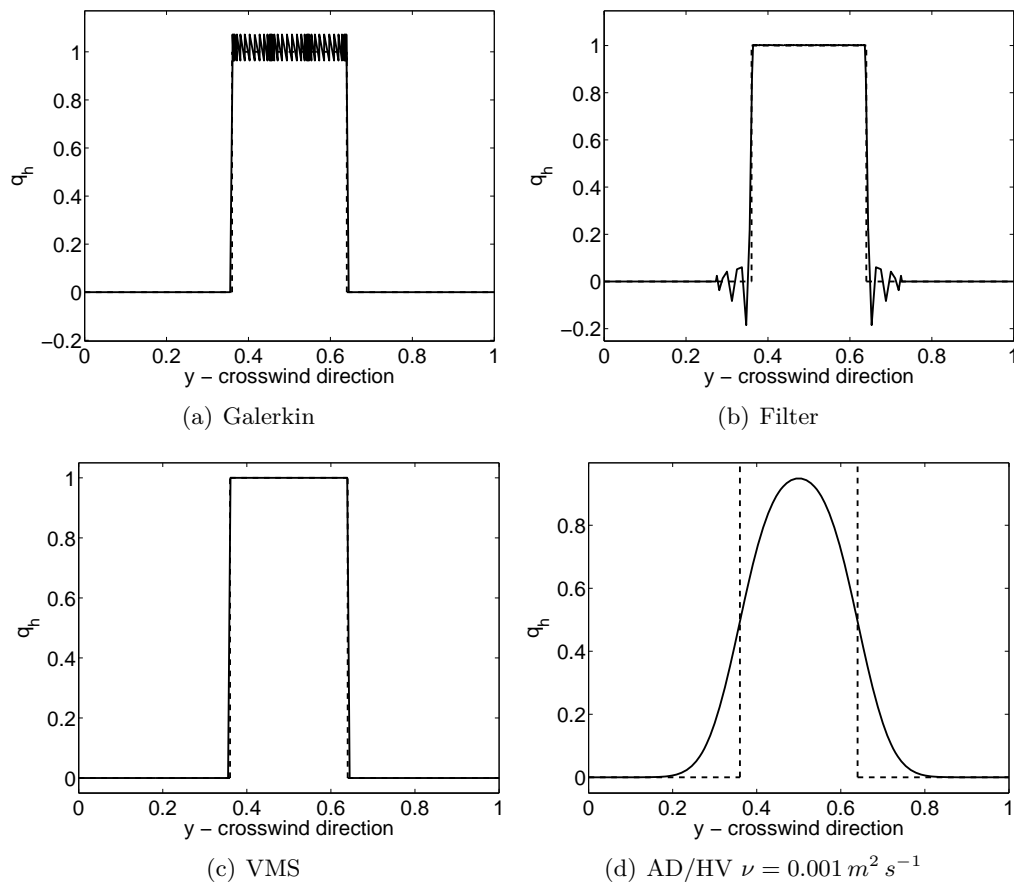


Figure 5.23: *Case 5: Crosswind cut at 0.5 m in the x-direction. $\Delta t = 0.001 \text{ s}$, 11×11 elements with 11^{th} order polynomials. Results at $t = 2 \text{ s}$ (after 1 periodic revolution along x). Solid line indicates the computed solution. The dashed line is the analytic solution.*

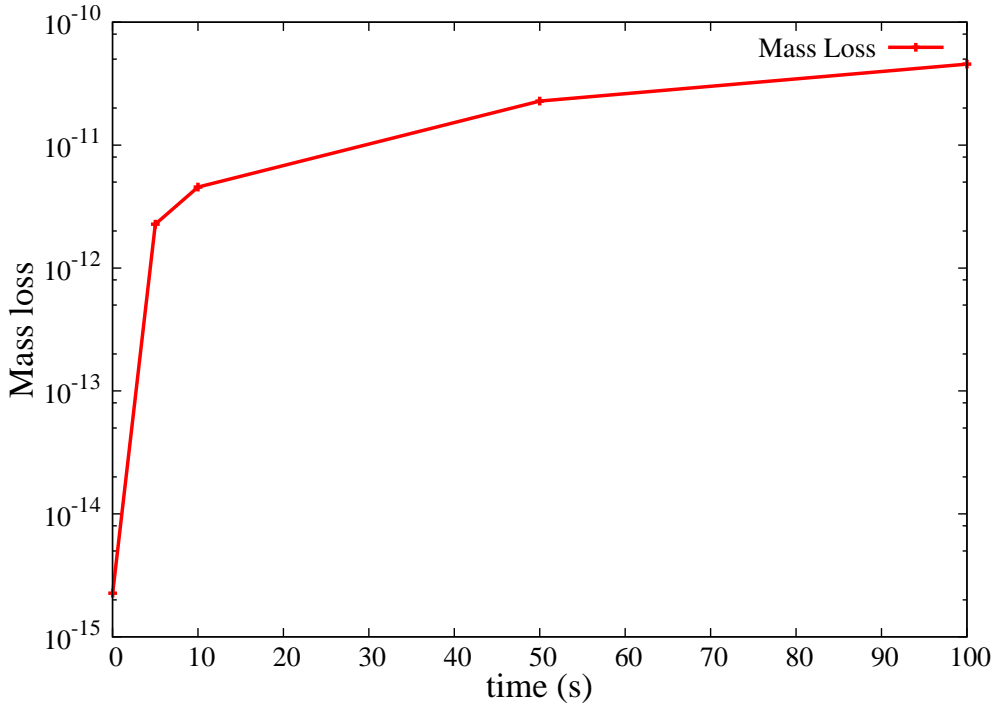


Figure 5.24: *Case 5*: evolution of the mass loss during 100 s, or 100 revolutions of the square wave around the periodic channel of Fig. 5.20.

this is obtained at the expenses of accuracy during long runs. Regardless of the type of equations (conservative or non-conservative), the method here proposed is certainly unable to retain all mass. Because of this, at this stage we can only think of applying this technique to short term weather forecast but not climate. This is the first application of VMS and DC to Spectral Elements to solve the advection equation; in the future we will work on a fix to this problem for better (or total) mass conservation. A first improvement of accuracy for long-time runs may be achieved using orthogonal sub-grid scales (OSS) as proposed by Codina (2000). Further analysis should be done.

Case 6. The smooth solid-body rotation test with a smooth function is used for a grid convergence study (Levy et al., 2007; Nair et al., 2011). A Gaussian hill $q^h = \exp[-5((x - x_c)^2 + (y - y_c)^2)]$ is originally centered in $(x_c, y_c) = (0, 0)$ in a periodic domain $\Omega = [-\pi, \pi] \times [-\pi, \pi]$ with prescribed velocity $(u, w) = (-\pi y, \pi x)$. Convergence is computed after one full revolution ($t = 2$ s). The normalized standard L_2 error is computed with respect to the exact solution $q_e = q^h(x, y, t = 0)$ using $N_{el} \in \{10^2, 20^2, 40^2\}$ and polynomials of degree 4 and 8. Fig. 5.25 shows the h -error. The original data are reported in Table 5.1.

The experiment indicates that VMS does not affect the rate of convergence of SEM. However, the time-discretization error is approximately 10^{-11} ; because of this, there is no gain in accuracy with further grid refinement from 1600 to 6400 elements unless a

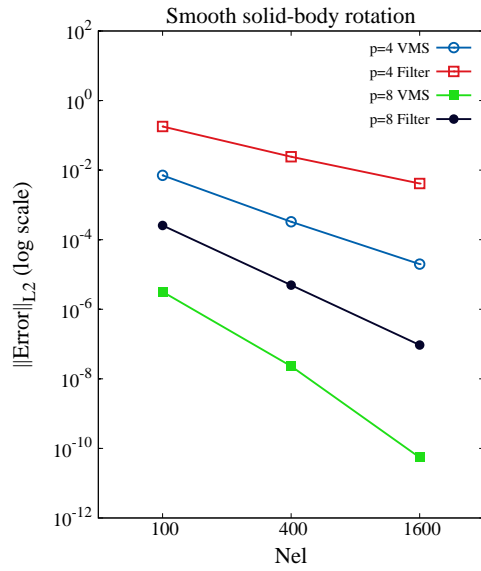


Figure 5.25: *Case 6*. Smooth Solid-Body rotation: Log-log plot of the normalized L_2 error vs. N_{el} using VMS or a Filter.

Table 5.1: *Case 6*: normalized L_2 error vs. N_{el} . Convergence rate of every setting is reported on the last row of the table.

N_{el}	VMS 4 th	FILTER 4 th	VMS 8 th	FILTER 8 th
100	0.7083E-02	0.1793E+00	0.3175E-05	0.2569E-03
400	0.3280E-03	0.2422E-01	0.2305E-07	0.4927E-05
1600	0.1982E-04	0.4127E-02	0.5473E-10	0.9388E-07
Convergence rate:	4.2406	2.7206	7.9168	5.7091

more accurate time discretization method is used.

Case 7. The transport of a passive tracer in a neutrally stratified atmosphere in a large domain represents an idealized application to a seemingly real atmospheric problem. This final test is a proof-of-concept to verify the behavior of the methodology over larger time and spatial scales that are of relevance for real applications.

The velocity field is no longer uniform and constant, but varies non-linearly in space and time during the evolution of a rising thermal perturbation originally centered at the central lower region of the domain. The difficulty of the test is expressed by the transient character of the velocity that, in the first instant of the motion, greatly affects the stability of the solution of the advection equation. The problem is defined as follows (Thomas et al., 2003). The domain extends within $\Omega = [0, 1000] \times [0, 1000] \times [0, 1000] m^3$. It is divided first into 10, and then into 20 spectral elements of order 4 along x and z , with 1 element along y . The simulation final time is $t_f = 600 s$. A neutral background state at uniform potential temperature $\theta_0 = 300 K$ is perturbed by a cylindrical thermal bubble of radius $r_c = 250 m$, centered in $(x_c, y_c, z_c) = (500, 500, 350) m$, and defined by

$$\theta' = A \left[1 + \cos \left(\frac{r\pi}{r_c} \right) \right], \quad (5.33)$$

where $r = \sqrt{(x - x_c)^2 + (z - z_c)^2}$ and $A = 0.5 K$. The top, bottom, left, and right boundaries are modeled as non-penetrating solid walls, while periodicity is imposed on the front and back boundaries (y -direction).

The thermal problem is modeled by the Euler equations of inviscid compressible flows and solved by the method described in Kelly and Giraldo (2012). The use of VMS to solve the Euler equations by SEM falls beyond the scope of this thesis. Currently, VMS for the finite element solution of the compressible Euler equations of dry nonhydrostatic stratified flows can be found in Marras et al. (2012c).

At time $t = 0 s$, the tracer q^h is centered in the same position of θ' , but is described by a cylindrical step function of maximum intensity $q'_{h_{max}} = 0.5$ within a radius $r \leq 250 m$. The initial state of q^h and θ' is shown in Fig. 5.26. After 600 s the rising bubble has developed into the structure plotted in Fig. 5.27. If properly resolved, the tracer is expected to have similar features given that the velocity field derives from the motion of the bubble.

The velocity field \mathbf{u} is still until the warm bubble begins to move due to buoyancy. As soon as $\mathbf{u} \neq 0$, the tracer begins to move as well. The sudden change of state from rest to moving generates oscillations at the boundaries of the tracer that are more difficult to treat with respect to its analogue steady-state case. Fig. 5.28 shows the tracer after 600 s on a grid of 10×10 elements of order 4. The filtered solution and the solution obtained with artificial diffusion (Figures 5.28a, 5.28b) have important under and overshoots that propagate in the whole domain. The constant diffusivity coefficient used with AV is taken as the average value of τ of VMS for the same simulation. As expected from the previous tests, VMS alone is not able to fully eliminate the oscillations

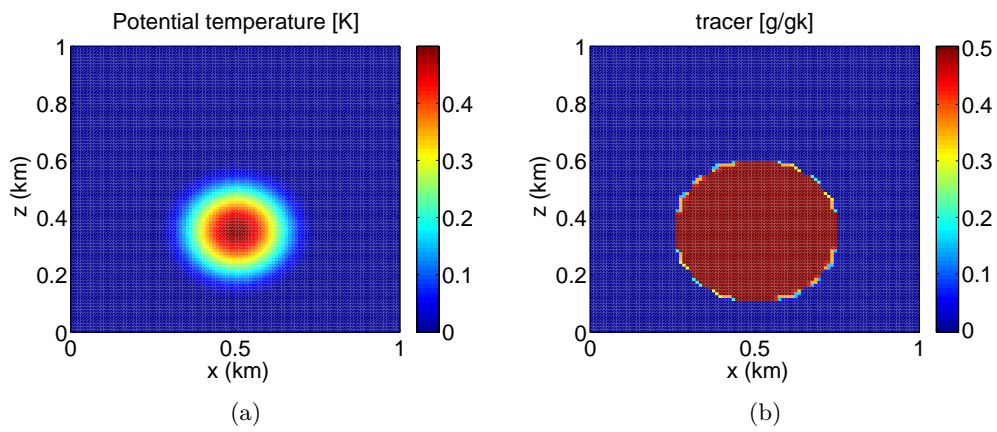


Figure 5.26: *Case 7*: $x - z$ -slice plot at $y = 500\text{m}$ of the initial conditions of θ' (left), and q^h (right).

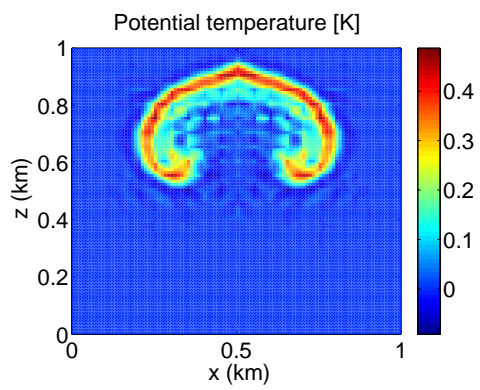


Figure 5.27: *Case 7*: θ' after 600 s. Grid: 20×20 elements of order 4.

Table 5.2: *Case 7*: Relative error ϵ on the maximum (0.5) and minimum (0.0) theoretical values. Results for 10×10 elements of order 4.

Method	ϵ_{min}	ϵ_{max}
AD/HV ($\nu = 0.001 m^2 s^{-1}$)	76.89 %	105.37 %
AD/HV ($\nu = 0.1 m^2 s^{-1}$)	61.79 %	45.00 %
Filter	51.05 %	53.53 %
VMS	33.67 %	36.21 %
VMS + DC	3.53 %	5.20 %

in the proximity of the discontinuity, however, improvement is evident (Figures 5.28c). The best performance is obtained with the combination VMS+DC. Using the theoretical extreme values for the tracer ($0 \leq q^{exact} \leq 0.5$), the relative error

$$\epsilon = \frac{q^h - q^{exact}}{q^{exact}}$$

is reported in Table 5.2. The same considerations apply for the finer-grid solution (20×20 elements of order 4). The results are plotted in Fig. 5.29 and the errors reported in Table 5.3.

Remark 5.1. Use of filters in the previous results In the current work, filtering was applied in the usual way that has been used previously in SE models (see, e.g., Fischer and Mullen (2001); Taylor et al. (1997); Giraldo and Rosmond (2004); Levin et al. (1997)). That is, the filtering coefficients were defined at the beginning of the simulation and applied after every time-step using the same filter matrix for all elements. It may be possible to obtain better results with filters if they are constructed in a specific way (e.g., each element uses a different filter matrix that is constructed dynamically) but a clear approach on how to do this remains an open topic since this can be viewed as a classical limiter but for Spectral Elements (see, e.g., Fournier et al. (2004)).

5.5 Summary and discussion

In this chapter, we proposed the use of the VMS to stabilize advection-dominated problems solved with spectral elements. In the regions characterized by strong gradients, we also combined VMS, discontinuity capturing method, and a First-Order Subcells

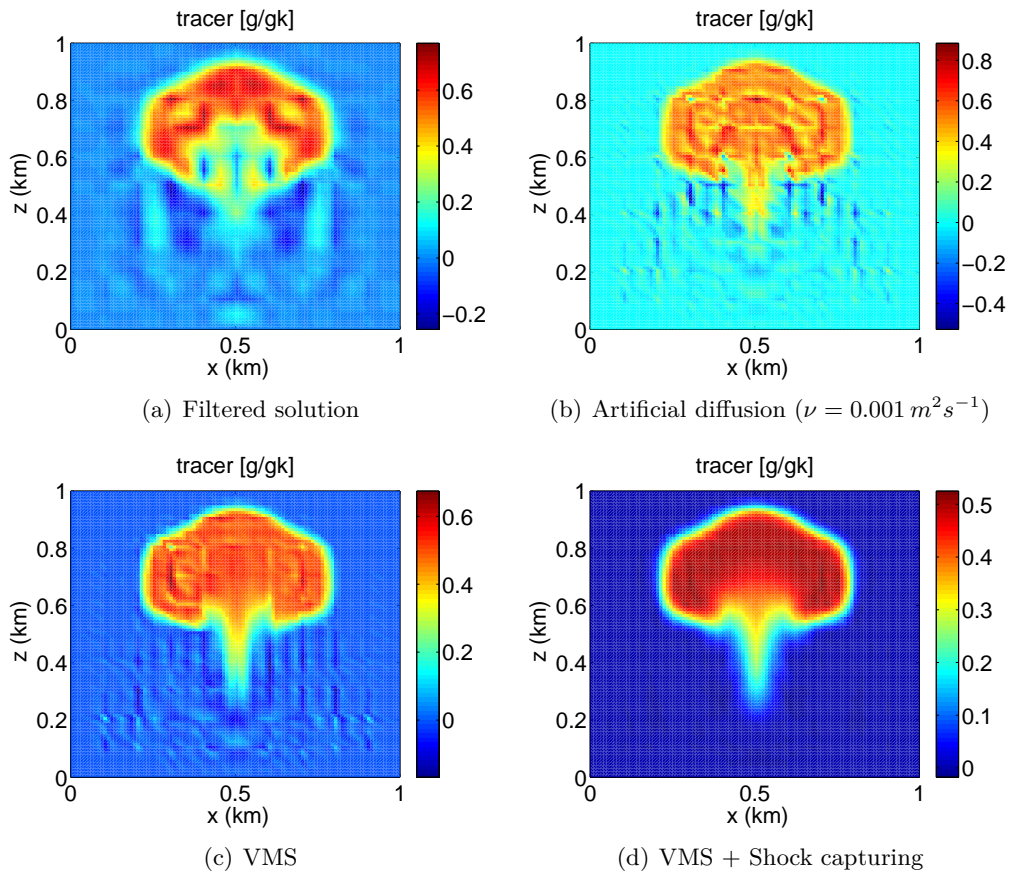


Figure 5.28: *Case 7*: Tracer after 600 sec. Grid: 10×10 elements of order 4.

Table 5.3: *Case 7*: As for Table 5.2, but for 20×20 elements of order 4.

Method	ϵ_{min}	ϵ_{max}
AD/HV ($\nu = 0.001 \text{ m}^2 \text{ s}^{-1}$)	65.26 %	62.59 %
AD/HV ($\nu = 0.1 \text{ m}^2 \text{ s}^{-1}$)	20.33 %	21.06 %
Filter	36.38 %	53.10 %
VMS	14.34 %	11.53 %
VMS + DC	1.24 %	4.63 %

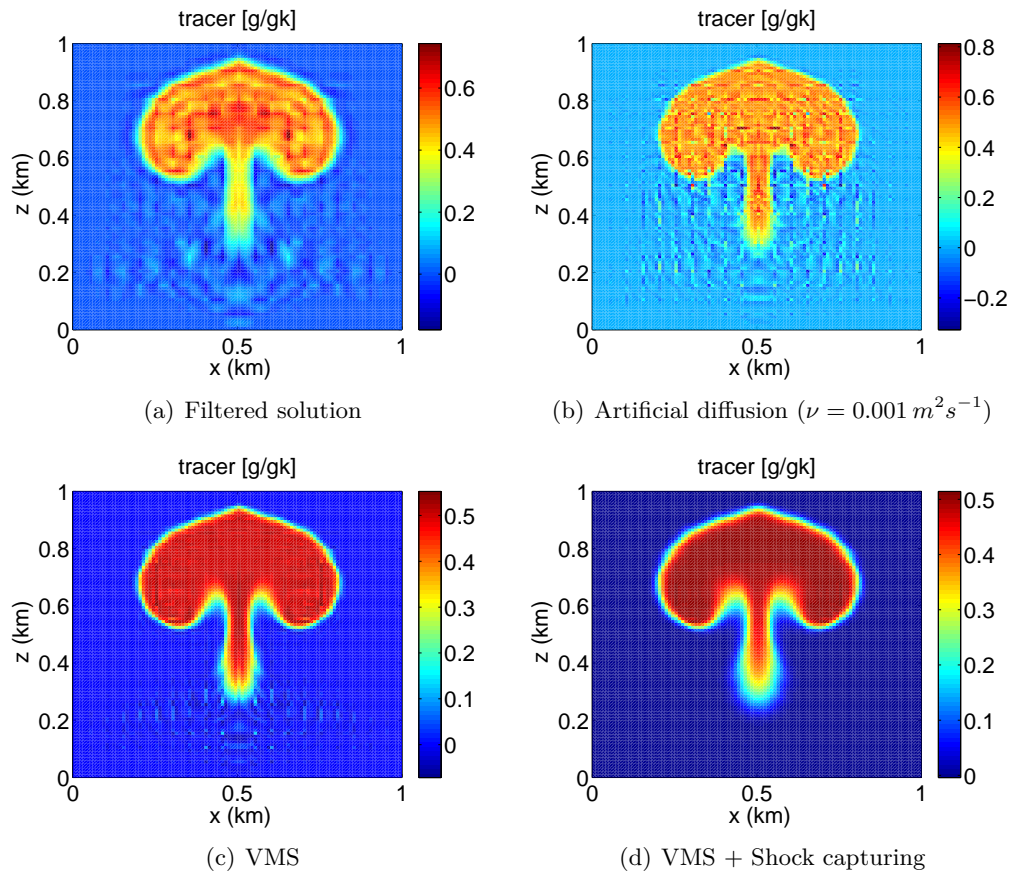


Figure 5.29: *Case 7*: Tracer after 600 sec. Grid: 20×20 elements of order 4.

method (FOS) for a better treatment of Gibbs phenomena in the proximity of boundary and internal layers. The stabilization parameter τ that appears in the VMS scheme was computed to include the characteristics of high-order spectral elements with LGL nodes. Numerically, we demonstrated that this approach is a possible alternative to the standard filters used in the stabilization of the spectral element solvers if the suppression of unwanted under- and over-shoots is the main concern. Stabilization by these methods is obtained by introducing a diffusion-like term that is controlled and localized where the residual is important (i.e. large gradients). Where needed, the combined action of VMS and FOS yields encouraging results. The algorithms were evaluated on a set of standard tests of increasing difficulty. A significant improvement was observed in the performance of the spectral element solver as far as the control of extrema is concerned, both in the purely advective and in the advective-diffusive regimes. The most important features of this new approach are the following:

- Unlike hyper-viscosity, the subgrid-scale diffusion is localized and controlled.
- Under- and over-shoots are greatly suppressed relative to traditional filters.
- VMS does not depend on a free-parameter assigned by the user. On the other hand, in Algorithm 1 ε , is a free-parameter related to the simple error-estimator that was used. A more sophisticated estimator should not depend on any user-defined constant.
- Currently, the method is not fully mass-conservative. This can be an issue for long term simulations such as those in climate applications.
- Pseudo-3D simulations on large space and time scales show that the method is suitable for extension to fully three-dimensional problems.

5.5.1 Application to atmospheric modeling in climate and weather prediction

In [Gaberšek et al. \(2012\)](#), a Kessler microphysics scheme within a spectral element framework requires the advection of three moisture variables (vapor, cloud, and rain mixing ratios). This microphysics scheme will be implemented within the *Nonhydrostatic Unified Model for the Atmosphere* (NUMA) ([Kelly and Giraldo, 2012](#)) in order to simulate both mesoscale and synoptic-scale atmospheric phenomena. As we have previously stressed, Galerkin-based methods yield (i) higher-order accuracy and (ii) excellent dispersion properties, which are both desirable for advection schemes; however, the resulting Gibbs oscillations produce strong gradients that must be remedied in some fashion. In the present high-order implementation by [Gaberšek et al. \(2012\)](#), a simple-minded “fixer” is applied whereby negative values of the moisture variables are set equal to zero. This fixer acts as an effective mass source, thus violating the conservation properties of the model. In addition, this fixer violates the function space that the spectral element solution inhabits. For these reasons, monotonic advection of tracer variables

is essential for any atmospheric model. The proposed VMS+DC+FOS technique is a candidate since it (i) preserves monotonicity better than the standard filter approach, (ii) does not significantly increase the cost of the spatial discretization scheme, and (iii) is completely local in nature (i.e., no additional communications are required in a parallel environment), which is necessary for scaling on modern distributed and hierarchical memory environments. However, the method should be improved for better mass conservation, especially when the time scales at hand are large.

In Chapter 6, the methods defined throughout this thesis will be applied to an atmospheric problem with moist dynamics. Specifically, the Euler system and the advection equations for water quantities will be coupled in the definition of a simple cloud model based on warm rain principles. The system of coupled equations will be solved by the finite element method reported mostly in Chapter 4.

Chapter 6

Idealized moist simulations: the case of convective storms

And then in the late A.M. the isolate clouds overhead start moving toward one another, and in the early P.M. they being very slowly interlocking like jigsaw pieces, and by evening the puzzle will be solved and the sky will be the color of old dimes.

– David Foster Wallace¹

So far the numerical methods for the solution of the Euler equations modeling dry stratified flows have been shown. Numerical Weather Prediction goes beyond dry atmospheres. Hence, the algorithms described above must be able to behave stably and accurately when the atmosphere is humid. This chapter describes a simple cloud model based on the solution of the Euler equations coupled to a set of three advection equations that model transport of water in the atmosphere. Phase change in the formation of clouds and precipitation is modeled by the warm rain microphysics scheme of [Kessler \(1969\)](#). All this is done within the finite element framework that was described in the previous chapters. Idealized moist simulations are defined and examined to verify the behavior of the numerical method in wet conditions. As it is for non-linear benchmarks in dry atmospheres, we cannot rely on the existence of any analytic solution to evaluate the results. Nor can we base the analysis on grid refinement until solution convergence is reached because, as it was proved in, e.g., [Weisman et al. \(1997\)](#), [Bryan and Fritsch \(2003\)](#), and [Bryan and Morrison \(2011\)](#) the solution is highly resolution dependent due to the type of physical processes involved. The metrics for analysis are macro-characteristics such as cloud formation, precipitation initiation, up- and downdrafts, spatial extension of the storm, and rain accumulation. The squall-line simulations of [Weisman et al. \(1988\)](#) and [Gaberšek et al. \(2012\)](#) are used as a reference for evaluation. A mountain initiated storm in two and three dimensions is also described at the end of the chapter.

¹*A Supposedly Fun Thing I'll Never Do Again*, 1997

6.1 Introduction

In this chapter, we assess the ability of FEM and VMS to simulate idealized moist dynamics. The Euler equations are coupled to a set of prognostic equations for water species that, by means of the Kessler microphysics are allowed to change phase and trigger a long-lasting convective storm similar to the one represented in Fig. 6.1. Convection in general is the process by which heat is transported. In meteorology, this definition restricts to the transport of heat in the vertical direction (by buoyancy), whereas the term advection classically refers to transport that is not necessarily caused by buoyant/vertical motion. Given this definition of convection, *convective storm* refers to that phenomenon by which vertical motion of moist atmospheres triggers condensation of water vapor into clouds that start developing and self-sustaining within an initially unstable atmosphere. The instability and vertical convection are caused by heat transfer in the lower troposphere (in the proximity of the surface). Air warms up, vertical motion begins and, if water vapor is transported to the colder regions by the vertically moving thermal, expansion of warmer, moist air may trigger condensation of vapor to cloud water. Condensation is a source of heat release to the environment. This induces a temperature increase and consequent expansion of the air parcel, causing the warmer, moist, region to rise until either the stability conditions of the atmosphere or the thermodynamic conditions permit more condensation. The lighter region within the condensed (cloudy) region is shown in Fig. 6.2, where the density and potential temperature inside a developing cloud are shown.

It must be borne in mind that the type of cloud/storm is a function of the initial state of the atmosphere (e.g. atmospheric stability), content of water vapor, and time evolution of surface heat flux. Based on the literature and on the data available to initialize the model, in what follows we mostly focus on the simulation of a convective storm; however, other phenomena could be simulated as well by solving the same system of equations. An example are the orographic clouds as we report in below.

Water in the atmosphere is present in different forms and phases that coexist simultaneously. We are interested in the simplest of water interactions; in particular, the microphysical processes that drive condensation of water vapor into clouds and formation of rain. In the absence of ice, snow, graupel, or hail, we talk about *warm* microphysics, where a simple bulk parametrization can predict cloud water and rain/drizzle only. Bulk models assume that the number of water categories in the atmosphere can be grouped such that as few species as possible need to be solved explicitly (Houze, 1993). Other methods are possible and have been investigated extensively.

Most of the analysis in this chapter is performed on two-dimensional tests. Due to the well known limitations of two-dimensional cloud modeling (see Orville and Kopp (1977); Wilhelmson (1974); Klemp and Wilhelmson (1978), and Weisman et al. (1988)), we do not expect a faithful representation of the most proper characteristics of convective storms. Nevertheless, two-dimensional simulations can still be used to verify the properties of the finite element algorithm for moist atmospheric simulations. The literature is sufficiently extensive in the presentation of thorough two-dimensional storm anal-

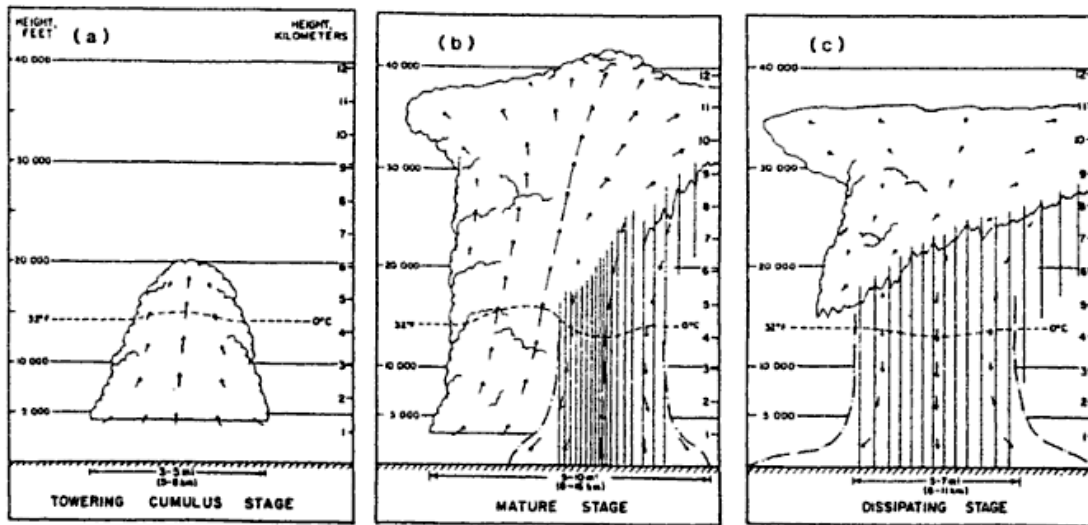


Figure 6.1: Graphical representation of the evolution of convective showers. (a) cumulus, (b) mature, and (c) dissipating stage. The arrows are the air velocity field. The region of rain-cooled air is delimited by the dash-dot lines. Image from Emanuel (1994), ©Oxford University Press. Reprinted with permission.

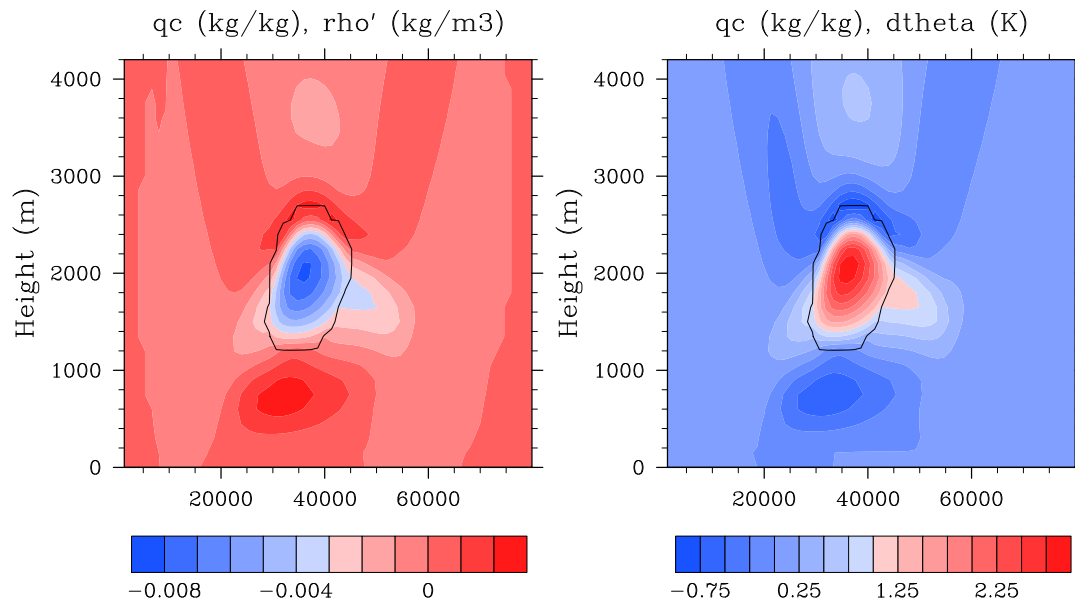


Figure 6.2: xz -slice of a fully 3D convective cloud in its formation stage at $t_f = 650$ s. The filled contours represent ρ' and θ' in the left and right panels, respectively. The contour line corresponds to $q_c = 10^{-5} \text{ kg kg}^{-1}$. The negative variation of ρ (left) and the positive perturbation of potential temperature indicate that condensation is taking place with heat release to the environment. Like in z , the distance along x is expressed in meters.

ysis for us to compare against (Gaberšek et al., 2012; Morrison and Grabowski, 2008; Grabowski and Clark, 1991; Grabowski, 2007) and draw some quantitative conclusions. An approach similar to that of Gaberšek et al. (2012) is used for the assessment of the model. Six tests are performed, two of which are fully three-dimensional: four simulate a convective storm triggered by a rising thermal perturbation, whereas two reproduce a class of orographic storms that are generated by pure mechanical motion of air aloft a smooth mountain.

The remainder of the chapter is organized as follows. Some notation and basic thermodynamics of moist air are given in section 6.2. The equations and the method of solution are reported in sections 6.3 and 6.4, respectively, while numerical tests are presented in section 6.6. Conclusions are given in section 6.7.

6.2 Definitions and thermodynamics of moist atmospheres

A great deal of difference between a moist and a dry gas stands in their thermodynamics. For a better understanding of the quantities and considerations presented throughout this chapter, it is instructive to introduce the necessary fundamentals of the thermodynamic quantities needed to represent moist dynamics in atmospheric simulations. The treatment is meant to be a simple and straightforward introduction to the new quantities that have not been relevant so far; for more details on moist convection the reader should make reference to Iribarne and Godson (1981), Emanuel (1994), Bannon (2002), and citations therein.

In the absence of solid water, water substances in the atmosphere are treated in terms of density of water vapor, ρ_v , cloud water, ρ_c , and precipitating water (or rain), ρ_r . From these, other quantities are derived and will be described as needed. Given a mass per unit volume of dry air, ρ_d , the mixing ratios of vapor, cloud, and rain are indicated by

$$q_i = \frac{\rho_i}{\rho_d} \quad i = v, c, r. \quad (6.1)$$

Pressure of moist air, p , is given by Dalton's law as the sum of the partial pressure of dry air, p_d , and the partial pressure of vapor given by $e = \rho_v R_v T$, where $R_v = 461 \text{ J kg}^{-1} \text{ K}^{-1}$ is the gas constant of vapor. Pressure is

$$p = p_d + e = \rho_d R T + \rho_v R_v T = \rho_d R T \left(1 + \frac{q_v}{\varepsilon} \right), \quad (6.2)$$

where $R = 287 \text{ J kg}^{-1} \text{ K}^{-1}$ is the usual gas constant of dry air, $\varepsilon = R/R_v$, and T is temperature.

In the presence of water vapor (i.e., unsaturated moist air), air density in a given volume V is simply

$$\rho = \frac{M_d + M_v}{V} = \rho_d(1 + q_v) = \frac{p_d}{RT}(1 + q_v).$$

Multiplication and division by $p_d + e$ yields:

$$\rho = \frac{p}{RT} \frac{p_d}{p_d + e} (1 + q_v) = \frac{p}{RT} \frac{1 + q_v}{1 + q_v/\varepsilon} = \frac{p}{T} \frac{1}{R'}, \quad (6.3)$$

where R' represents the gas constant of moist air as

$$R' = R \frac{1 + q_v/\varepsilon}{1 + q_v}. \quad (6.4)$$

R' is usually referred to as *effective gas constant*. Because $\varepsilon < 1$, from (6.4) and (6.3) we notice how a parcel of moist air is lighter than its dry counterpart at equal T and p . To consider the dependence of ρ on vapor, instead of using temperature and potential temperature as working variables, a modified temperature has been introduced and is of common use for the description of the Euler equations that govern moist dynamics. This modified temperature is called *virtual temperature*. It is derived by the considerations and definitions above and is given by

$$T_v = T \frac{1 + q_v/\varepsilon}{1 + q_v} \approx T(1 + q_v\varepsilon). \quad (6.5)$$

From T_v , virtual potential temperature is

$$\theta_v = T_v \left(\frac{p_0}{p} \right)^{R/c_p} \approx \theta_d(1 + q_v\varepsilon), \quad (6.6)$$

where $p_0 = 10^5 \text{ Pa}$ and $c_p = 1004 \text{ J kg}^{-1} \text{ K}^{-1}$ is the specific heat of dry air at constant pressure. In terms of θ_v , the equation of state becomes

$$p = p_0 \left(\frac{R \rho_d \theta_v}{p_0} \right)^{c_p/c_v}. \quad (6.7)$$

The use of virtual quantities is sometimes considered unnecessary. In fact, the relationship between T and θ for dry air given by

$$\theta \doteq T \left(\frac{p_0}{p} \right)^{R/c_p} \quad (6.8)$$

would still hold as long as the exponent of the pressure ratio is corrected by the presence of vapor to give:

$$\theta \doteq T \left(\frac{p_0}{p} \right)^{\frac{R}{c_p} \frac{1 + q_v/\varepsilon}{1 + q_v c_{pv}/c_p}}. \quad (6.9)$$

Then, why introduce the concept of θ_v if the physical meaning of θ is still valid? In the case of unsaturated air the reason is practical: the stability of the atmosphere is very much related to the vertical gradient of θ_v rather than that of θ (Emanuel, 1994). The *Brunt – Väisälä* frequency introduced for dry atmospheres in Chapter 4 becomes

$$N^2 = \frac{g}{\theta_v} \frac{\partial \theta_v}{\partial z}.$$

What has been said so far is valid only as long as the unsaturated air does not contain condensed particles (e.g. water droplets, ice) whose density is large enough to affect the total density of moist air. In that case, the total air density changes to $\rho = \rho_d + \rho_v + \rho_w$, where ρ_w includes the density of any possible extra water quantity contained in the atmosphere. Of these, rain drops and ice are two examples. For our needs the only water content in the atmosphere comes from water vapor, cloud water, and rain. The total mixing ratio is then $q_t = q_v + q_c + q_r$. In this case, a generalization of virtual temperature and potential temperature are the *density temperature* and *density potential temperature* given, respectively, by

$$T_\rho = T \frac{1 + q_v/\varepsilon}{1 + q_t}, \quad (6.10)$$

and

$$\theta_\rho = T_\rho \left(\frac{p_0}{p} \right)^{R/c_p}. \quad (6.11)$$

In the case of cloudy air (saturated, possible phase change), static stability (indicated by N_ρ to distinguish it from N used for dry environment) is not as straightforward to define as it is for the case of dry air or unsaturated atmospheres (i.e. $q_t = q_v$). The vertical displacement of saturated air is linked to strictly non-linear phenomena such as phase change. Because of this, N_ρ is not simply a function of the vertical gradient of θ_ρ , but of moist entropy and total water content. The derivation of N_ρ in this case is not difficult but it is not of interest for the remainder of this chapter. The interested reader can find it in (Emanuel, 1994, Ch. 6). Because this chapter partly serves to wrap up the concepts presented throughout the whole thesis, in Table 6.1 we reproduce Table 2.2 of Cotton et al. (2011) where the criteria to determine static stability of different atmospheres are summarized.

Definition: Convective Available Potential Energy (CAPE). The conditions that, most likely, allow convective clouds to form, can be identified by the buoyant stability of the environment (Houze, 1993). The amount of potential energy of a rising parcel within the environment is proportional to the integral of its buoyancy. This

Table 6.1: Static stability for different environments. Adapted from Table 2.2 of Cotton et al. (2011).

	Dry	Moist unsaturated	Moist saturated
Absolutely unstable	$\partial\theta/\partial z < 0$	$\partial\theta_v/\partial z < 0$	$N_\rho^2 < 0$
Neutral	$\partial\theta/\partial z = 0$	$\partial\theta_v/\partial z = 0$	$N_\rho^2 = 0$
Absolutely stable	$\partial\theta/\partial z > 0$	$\partial\theta_v/\partial z > 0$	$N_\rho^2 > 0$

quantity is identified by the *Convective Available Potential Energy*, or CAPE, that is defined as:

$$CAPE = \int_{LFC}^{Z_{top}} g (\theta_\rho - \bar{\theta}_\rho) dz, \quad (6.12)$$

where the term of integration is the buoyancy of moist air. As in Chapter 4, the barred quantities indicate the background state of the atmosphere. In the case of dry air, buoyancy is simply equivalent to $B = g(\theta - \bar{\theta})$ as usual. LFC (for Level of Free Convection) is the height at which buoyancy is positive and the parcel begins to rise dry adiabatically. With reference to Fig. 6.3, LFC is at the inflection point on the dash-dotted purple curve. If, while rising, saturation occurs, the parcel keeps rising moist adiabatically and stops at the level where $\theta_\rho = \bar{\theta}_\rho$ (i.e. zero buoyancy). On a skew $T - \log(p)$ diagram, the value of CAPE is proportional to the area between the θ_ρ curve of the rising parcel and that of the environment. In Fig. 6.3, the area of interest is the region between the red dashed curve and the thick continuous temperature line. If a convective storm with rain has formed and is self-sustaining, CAPE is likely to be larger than $2 kJ kg^{-1}$ (Emanuel, 1994).

The definitions reported so far provide the basis for us to proceed and describe the dynamics of moist atmospheres to simulate a convective storm. We start by the description of the equations of motion in the following section.

6.3 Basic equations of moist dynamics

In this section, we present how the set of the Euler equations of dry motion are modified to become suitable for cloud modeling. Like before, Coriolis effects are neglected and so is turbulence. Cloud systems can be modeled properly only if either the compressible or anelastic equations are used. The hydrostatic approximation is clearly not a possibility due to the important vertical motion of convective clouds.

We begin by taking the system (2.11) of compressible flows with stratification and

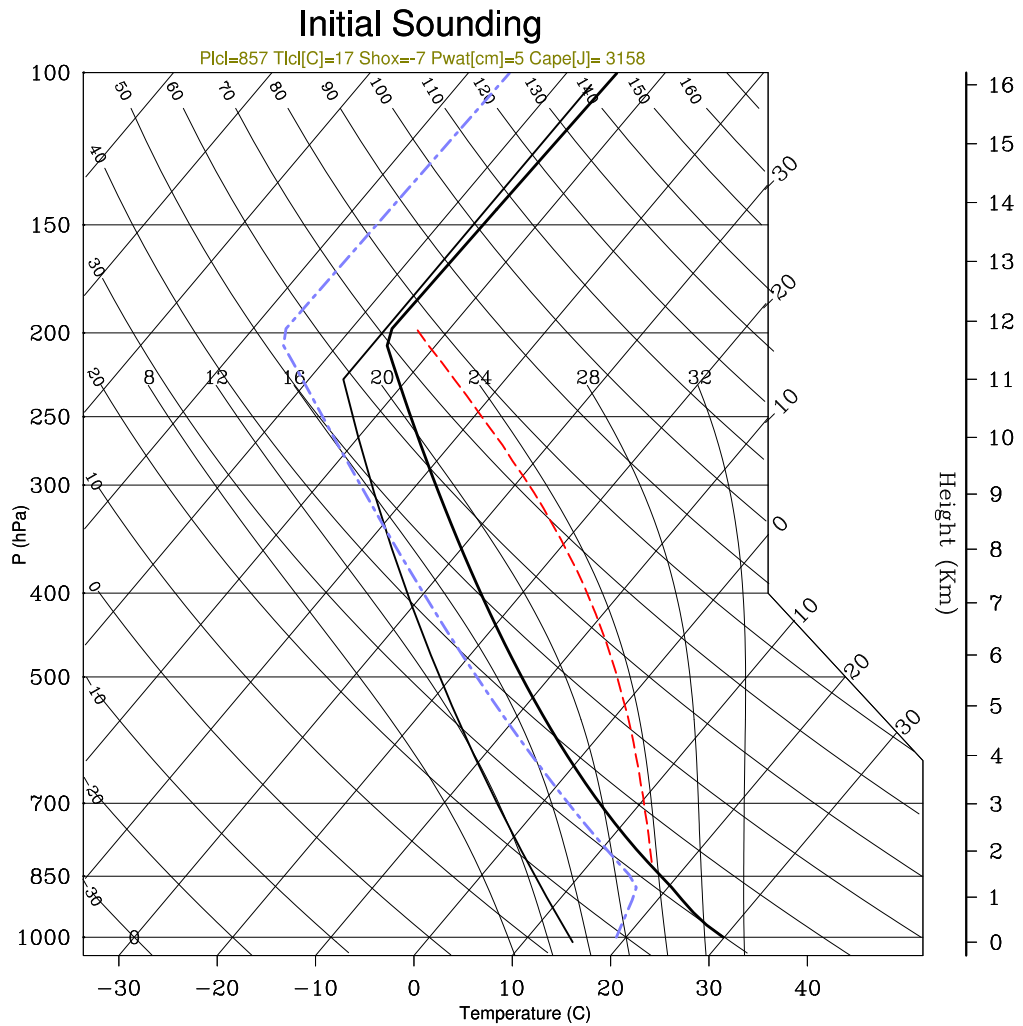


Figure 6.3: Skew-T plot of the initial sounding. Temperature and the dew point temperature are, respectively, the thick black continuous and the dash-dotted purple lines. The dashed red curve is the wet adiabatic curve that originates at LCL. The area between the wet adiabatic and the T curve is equivalent to the value of CAPE.

include the extra necessary terms. As usual, we write the system as

$$\frac{\partial \phi_1}{\partial t} + \mathcal{L}_1(\phi_1) = \mathbf{f}_1(\phi_1), \quad (6.13)$$

where ϕ_1 is the vector of the unknowns, \mathcal{L}_1 is the differential operator, and \mathbf{f}_1 is the source vector. We have:

$$\phi_1 = \begin{bmatrix} \rho_d \\ U \\ W \\ \theta_\rho \end{bmatrix}, \quad \mathcal{L}_1(\phi_1) = \begin{bmatrix} \nabla \cdot \mathbf{U} \\ \nabla \cdot \left(\frac{\mathbf{U}\mathbf{U}}{\rho_d} + p'\mathbf{e}_x \right) \\ \nabla \cdot \left(\frac{\mathbf{U}W}{\rho_d} + p'\mathbf{e}_z \right) \\ \frac{\mathbf{U}}{\rho_d} \cdot \nabla \theta_\rho \end{bmatrix}, \quad \mathbf{f}_1(\phi_1) = \begin{bmatrix} 0 \\ 0 \\ B \\ S_\theta \end{bmatrix}, \quad (6.14)$$

where $B = -\rho'_d g (1 + \varepsilon q'_v + q_c + q_r)$ represents buoyancy affected by moisture. The first difference with respect to the dry equations is the definition of B on the right-hand side of the momentum equation. As explained in Section 6.2, moist air contributes with positive buoyancy to the flow. This effect is modeled by adding the contribution of water concentration to the gravity term of the equation. The second major difference is in the temperature equation. The thermodynamic equation presented here is one of a handful of possible options presented in the literature. Different models differ from one another according to the assumptions that are made in terms of microphysical quantities to use, the definition of the variable of interest (entropy represented by equivalent potential temperature), or the way dry dynamics is defined. In this chapter, the equation for density potential temperature, θ_ρ , is used and is expressed using the same right-hand side reported in the recent paper by [Gaberšek et al. \(2012\)](#). The source or sink term S_{θ_ρ} includes the latent heat release or gain that occurs during the phase changes of the moisture variables. More details on S_θ are reported below along with the rest of the microphysical processes that are included in the set of equations for water quantities. In the total absence of moisture, the Euler equations for dry air (2.11) are clearly re-established because, in that case, $S_\theta = 0$, $\theta_\rho = \theta$, and $B = -\rho'_d g$.

In (6.14), the mixing ratios q'_v , q_c , and q_r that appear in B are computed from the solution of three additional equations that we write compactly as

$$\frac{\partial \phi_2}{\partial t} + \mathcal{L}_2(\phi_2) = \mathbf{f}_2(\phi_2), \quad (6.15)$$

with

$$\phi_2 = \begin{bmatrix} q'_v \\ q_c \\ q_r \end{bmatrix}, \quad \mathcal{L}_2(\phi_2) = \mathbf{u} \cdot \nabla \begin{bmatrix} q'_v \\ q_c \\ q_r \end{bmatrix}, \quad \mathbf{f}_2(\phi_2) = \begin{bmatrix} S_{q_v} \\ S_{q_c} \\ S_{q_r} \end{bmatrix}. \quad (6.16)$$

The quantities $S_{q_v}, S_{q_c}, S_{q_r}$ above, and S_θ in the energy equation, include the microphysical processes of condensation, coalescence, and evaporation that are described in what follows.

6.3.1 Microphysics

Cloud microphysics includes all the thermo-physical processes at the scales of the particles that form the cloud. Examples are the phase change of water quantities or the agglomeration of particles into larger ones. Most physical processes typical of storm dynamics (e.g., precipitation, freezing, deposition, or sublimation) have a physics and characteristic scales that make their explicit modeling too computationally challenging or even unviable (see Emanuel (1994), Ch. 10). For this reason, parametrization of some sort is commonly used within numerical models. Microphysical parametrization relies on the physical knowledge of certain processes without the need for a full understanding of all the microscale processes that are involved. The clear limitation is that certain phenomena cannot be represented with high accuracy if they lay outside of the conditions required by the parametrization. The most common representation of cloud microphysics was designed by Kessler (1969).

Kessler's is a *bulk* model, meaning that water species are categorized only with respect to the particles' type. In other words, if we speak about cloud water, we would model it through one equation that represents the transport of cloud water concentration with water droplets of one single size. Bulk models are contrasted by *explicit* models, where, within each category (e.g., cloud, rain) the size of the water particles is considered as well. Explicit models are, arguably, more physically accurate, but they certainly cost more due to the larger amount of quantities that must be accounted for. Here, we limit ourselves to note this. For more information on the topic, please, refer to Houze (1993) and more recent literature.

Kessler's is a simple scheme based on the main assumption that ice is not contemplated (warm rain). The main limitation of the warm condition is that only moist convection at the tropics or at mid-latitudes in the warm season can be represented. The three forms of water that are considered are: (i) water vapor; (ii) cloud water (liquid water whose size is so small that its terminal fall speed is negligible); and (iii) precipitating water that only includes rain (namely, drops whose diameter is $> 0.5 \text{ mm}$). Drizzle is excluded (rain of drop diameter between 0.2 and 0.5 mm).

The main processes resolved by a warm cloud microphysics scheme are touched on briefly. These processes dictate how the source terms of the previous equations are defined and how they affect the dynamics of the simulation. The illustration reported in this section is all but exhaustive. The reader is referred to the literature for a more thorough analysis (e.g. see [Houze \(1993\)](#) and references therein). Given the approximated Teten's formula ([Bolton, 1980](#)) for saturation vapor pressure as

$$e^* = 611.2 \exp\left(\frac{17.67T}{T + 243.5}\right),$$

the saturation mixing ratio is given by

$$q_{vs} = \frac{\varepsilon e^*}{p - e^*}. \quad (6.17)$$

From [Klemp and Wilhelmson \(1978\)](#), the S -terms in (6.14) and (6.16) are

$$\begin{aligned} S_\theta &= -\frac{L_v}{c_p T} (\dot{q}_{vs} + E_r), \\ S_{q_v} &= \dot{q}_{vs} + E_r, \\ S_{q_c} &= -\dot{q}_{vs} - A_r - C_r, \\ S_{q_r} &= \frac{1}{\rho} \frac{\partial}{\partial z} (\rho V_r q_r) - E_r + A_r + C_r, \end{aligned} \quad (6.18)$$

where, being c_{pl} and c_{pv} the heat coefficients at constant pressure of liquid water and water vapor, respectively, $L_v = L_{v0} - (c_{pl} - c_{pv})(T - T_0)$ is the latent heat of vaporization with reference value $L_{v0} = 2.5e + 6 \text{ J kg}^{-1}$, T_0 is a reference temperature, V_r is the terminal fall speed of raindrops (taken positive in the downward direction), and \dot{q}_{vs} is the rate of condensation or evaporation (the dot symbol indicates derivation with respect to time). A_r , C_r , and E_r are the rates of autoconversion, collection, and evaporation of rain. They are computed using the formulas:

$$A_r = \text{MAX}(0, k_1(q_c - a_T)), \quad (6.19a)$$

$$C_r = k_2 \rho^{0.375} q_c q_r^{0.875}, \quad (6.19b)$$

$$E_r = -\frac{1}{\rho} \frac{(q_v/q_{vs} - 1)k(\rho q_r)^{0.525}}{5.4 \times 10^5 + 2.55 \times 10^6(p q_{vs})}, \quad (6.19c)$$

where $k_1 = 0.001 \text{ s}^{-1}$, $k_2 = 2.2 \text{ s}^{-1}$, $a_T = 0.001 \text{ kg kg}^{-1}$ are Kessler's parameters and

k is the ventilation factor that is a function of the terminal fall speed. Eq. (6.19a) was derived by Kessler considering that cloud is converted into rainwater whenever q_c exceeds a threshold a_T . *autoconversion* is the rate at which the rain water content increases at the expenses of cloud water due to the coalescence of smaller particles. Yet, this process is not fully understood. Nor it is fully understood how *collection* occurs. As the name suggests, collection can be explained as cloud water particles being collected by the falling larger rainwater droplets that go through the cloud layers during their fall. *Evaporation* occurs when the sensible heat flux from the environment into the water droplet is balanced by the latent heat of evaporation of the water particle. As in Soong and Ogura (1973), the cloud droplets move at the same speed of the flow because they are considered having negligible terminal velocity.

The values of the constants in (6.19) are, to a certain extent, arbitrary (Houze, 1993); however, by observations, it is of common agreement that k_1, k_2 and a_T are non-linear terms with respect to q_c itself. They are also a function of temperature and of the distribution of the condensation nuclei. As Emanuel (1994) points out, the lack of understanding of the underlying physics is such that different results are being obtained by different and more sophisticated schemes. For the topic of this thesis, there is no need to go further into this. Nevertheless, it is important to stress out that microphysical parametrization has a major role in forecasting clouds and precipitation, but is still an open field of research (see the recent paper by Morrison and Grabowski (2008)).

Remark 6.1: subgrid-scale mixing. Turbulent mixing is an important ingredient in the simulation of moist atmospheres. The small scale turbulence ($< 1 km$) in the boundary layer regulates the intensity of a developing storm. Because in this thesis turbulence is not accounted for, the limitations of the model that we present are obvious; yet, we will see that the solution is sufficiently good to show that the algorithm performs properly in the presence of microphysical processes. In the absence of a proper turbulence closure scheme, many research codes represent subgrid-scale mixing with an artificial viscous term. The amount of diffusion that the current implementation of VMS introduces, localized and limited to the regions of important gradients, cannot serve as a type of turbulence closure technique. As its formulation suggests, however, it could be constructed in that way. This is an open topic of research that can be found in Bazilevs et al. (2007), among others.

6.4 Method of solution

Numerically speaking, (6.13) and (6.15) are solved by the finite element method described in Chapter 4. Regardless of the type of space approximation, phase changes are accounted for by using the *saturation adjustment technique* explained in detail in the appendix of Soong and Ogura (1973). In this section we center on this issue since all the numerical framework was covered in the previous chapters.

The *saturation adjustment technique* consists of solving the problem in two steps.

First, the prognostic equations are solved by neglecting all the terms that involve phase changes (all the S -terms are set to zero). This means that the dynamics and transport equations are advanced forward to an intermediate time-step n^* so that the intermediate values of the prognostic variables, $(\rho, p, \theta_\rho, q_{vs}, q_v, q_c, q_r)^*$, are obtained. These values are plugged into the **Kessler** module² to compute the S -quantities defined above. Once the computation of S is completed, the thermodynamic variables are updated and returned to the *Euler/transport* solver as the initial values for the next time step $n + 1$. A pseudo-code that describes this stream of operations is shown in Algorithm 2.

Algorithm 2 Steps to add microphysical processes using saturation adjustment

Initialize ϕ_1^0 and ϕ_2^0

for all time-steps n **do**

1. **Dry atmosphere and transport solution:**

Advance Eq. (6.13) forward to the intermediate time-step $n^* \rightarrow \phi_1^{n^*}$.

Advance Eq. (6.15) forward to the intermediate time-step $n^* \rightarrow \phi_2^{n^*}$.

2. **Phase change:**

$S_i \leftarrow \text{Kessler}(\phi_1^{n^*}, \phi_2^{n^*})$, $i = \theta, q_{vs}, q_v, q_c, q_r$.

3. **Moist solution:**

Update ϕ_1^{n+1} and ϕ_2^{n+1} using S_i .

end do

6.5 Boundary conditions

The top and bottom boundaries are rigid walls that allow free-slip for velocity, but allow water to exit. Given a domain sufficiently large compared to the extension of the cloud, periodic boundary conditions in the x direction are a suitable choice. The choice of lateral boundary conditions in cloud simulations is strictly linked to the type of simulation one is interested in. Whether it be a field of convective clouds or an isolated storm, the domain must extend enough in the horizontal direction(s) for the lateral boundaries not to affect the dynamics and thermodynamics of the convective system.

Non-reflecting boundaries. The formation of a lifting convective cloud is the cause of upward-propagating gravity-wave energy that reflects back to the computational domain, unless a radiation condition is implemented. We use the same absorbing layer of **Durran and Klemp (1983)** already described in Section 4.3. Fig. 6.4 shows the damping of the internal gravity waves in a 2D simulated squall line with and without wind shear. The Rayleigh layer used in both simulations is inserted at $z = 11.7 \text{ km}$. The effect of

²In this thesis, the term *module* is not to be intended in the sense of Fortran programming. It may be synonym of either a specific code function, set of functions, or any algorithmic entity that depends on the developer implementation.

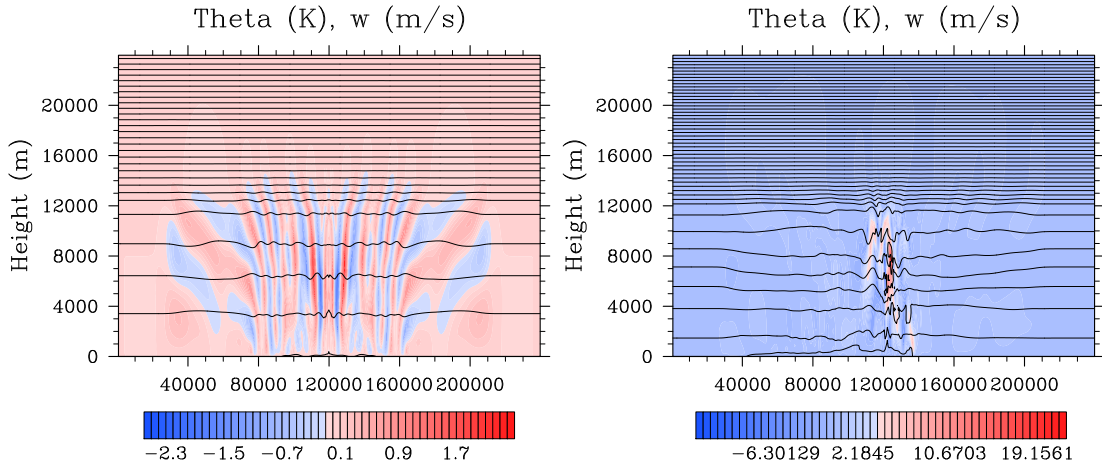


Figure 6.4: *Gravity waves*. Idealized squall line at $t = 7200$ s. Visualization of gravity waves through the θ_ρ isopleths plotted on top of the filled contours of vertical velocity. The left plot represents the simulation without wind shear, while the one on the right includes wind shear. Both simulations are solved on a uniform grid with $\Delta x = 580$ m and $\Delta z = 290$ m (414×83 quads). A 11.7 km deep upper absorbing layer is used.

the absorbing layer is evident in the upper region of the domain where the contour lines of θ_ρ are oscillation-free. Reflecting gravity waves could affect the thermodynamic fields by unwanted oscillations that eventually could either quench or boost unphysically the development of the storm.

6.6 Numerical tests

The algorithm is evaluated against a set of idealized two- and three-dimensional tests designed for wet dynamics that permits the formation of a storm and precipitation of warm rain. The initial background temperature, water vapor, and horizontal velocity for all the tests come from the sounding of Rotunno et al. (1988) depicted in Figs. 6.3 and 6.5. It consists of an initially saturated boundary layer in a typical environment of midlatitude squall lines. The air saturation decreases at the higher levels of the atmosphere. The weak stability of the atmosphere changes from a *Brunt–Väisälä* frequency $N = 0.01$ s⁻¹ below the tropopause (at approximately 12 km) to a more stable condition with $N = 0.02$ s⁻¹ above 12 km. Condensation starts in a volume that contains water vapor and is raised to regions of lower temperature and pressure. These conditions allow the expansion of the wet volume and hence vapor condensation. As in Rotunno et al. (1988), in the first three tests buoyancy is excited by an initial perturbation of potential temperature with an elliptical warm bubble whose characteristics are defined test-by-test. Once the rising air parcels reach the level of free convection, lifting continues as long as the parcels are less dense than the surroundings. At saturation (i.e. according to Kessler, when condition (6.19a) is fulfilled), rain begins to form and precipitate.

Before trying to simulate a pseudo-real, long-lasting storm with wind shear, we take a simpler step. Following the tests of [Klemp and Wilhelmson \(1978\)](#), we consider the case in which there is no initial shear and focus on the first few minutes during the storm development. The initial temperature perturbation is symmetric with respect to the central axis of the domain; for this reason, the cloud that forms from vapor lifting must remain symmetric for as long as the simulation advances forward. The simulation is run long enough for the cloud to form and lift, but for a time that is sufficiently short for the main updraft to be resolved before the cloud reaches the tropopause. Wind shear is particularly important. The duration, extension, and intensity of a storm are regulated, among other factors, by the type of vertical shear ([Weisman and Klemp, 1982](#)). As it will be observed with the first test (*Case 1: Simple*), a short-lived single cell is expected with none-to-weak wind shear. As the shear intensifies, the duration of the storm increases, with subsequent formation of long-lived multicell structures that may result into supercells and split storms caused by high shears. The second test is a two-dimensional version of the problem defined in [Rotunno et al. \(1988\)](#) and [Weisman et al. \(1988\)](#). The same grid resolution of Rotunno and colleagues is used for qualitative comparison. Specifically, it is a uniform grid of quadrilateral elements. The grid equivalence is advantageous when comparing results because of the resolution dependence that moist convection problems present. This test is identified by *Case 2: Storm-WKR88*. The third test, *Case 3: Storm-GGD12*, is the two-dimensional problem of [Gaberšek et al. \(2012\)](#) who used high-order spectral elements to evaluate the grid dependence of the solution. The structure of two-dimensional storms was analyzed by [Thorpe et al. \(1982\)](#), where different semi-analytic results are derived for different types of wind shears. We will show how the properties of the two-dimensional moist convection are reproduced according to the theory within the limits of similar initial conditions³. A description of different types of squall lines is given by [Bluestein and Jane \(1985\)](#).

The simulation of orographic clouds in two and three dimensions is presented as well. In *Case 4: O-Clouds* and *Case 5: 3D O-Clouds*, a mountain storm forms aloft an idealized *Witch of Agnesi*. We did not find similar simulations in the literature so that comparison was not possible. To evaluate this test, we verify that the structure of the storm has physically meaningful characteristics. Finally, a fully three-dimensional convective cell with vertical wind shear is simulated in *Case 6: 3D convective cloud*. It is the closure of a set of idealized tests with the idea of verifying the capabilities of the code in simulating realistic three-dimensional convective phenomena.

6.6.1 Case 1: Simple

This is the simplest among the set of problems which will be described in more detail in the sections that follow. The domain extends along 240 km in the horizontal direction

³The type of clouds that form depend on the sounding and thermodynamic conditions of the environment. A model is not designed to solve one type of cloud or another; rather, it is designed to solve moist environments that, according to the atmospheric conditions, are more likely to develop a certain type of clouds and not others.

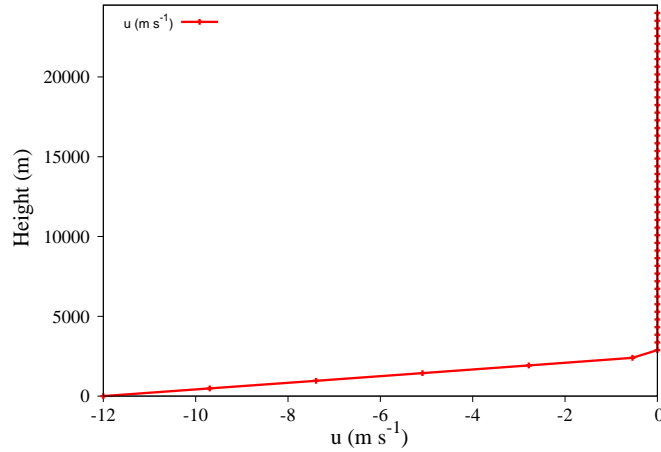


Figure 6.5: Representation of the vertical wind shear.

and is 24 km deep. The initial field given by the sounding represented in the skew-T plot of Fig. 6.3 is perturbed by an elliptical thermal θ' centered at $(x_c, z_c) = (120, 2)$ km and defined by

$$\theta' = \theta_c \cos^2 \frac{\pi r}{2} \quad \text{if } r \leq 1, \quad (6.20)$$

with amplitude $\theta_c = 3.0$ K and $r = \sqrt{(x - x_c)^2/x_r^2 + (z - z_c)^2/z_r^2}$. The radii of the ellipsis are $(x_r, z_r) = (10, 1.5)$ km. The size of the perturbation is certainly large. So is the value of its amplitude. This definition is appropriate because it triggers strong convection in a reasonable period of time that is advantageous for testing purposes.

To show the wind's effect on the longevity of the storm, the same problem is solved with and without wind shear. See the sounding of the wind shear in Fig. 6.5.

Results Case 1. The solution is reported at $t_f = 1550$ s (≈ 25 min) in Fig. 6.6. At this time, the updraft caused by the lifting cloud has reached its maximum intensity, with peak vertical velocity equal to $w = 16$ m s⁻¹, and rainwater has started to fall. The rainy region is very localized right below the main core of the cloud. After 3600 s (Fig. 6.7), when the cloud has reached the tropopause and started to spread leftward and rightward symmetrically, precipitation begins to decrease. The maximum concentration of approximately $q_r = 8$ g kg⁻¹ at 1550 s drops to approximately $q_r = 3$ g kg⁻¹ after 3600 s, as it is visible by comparing Figs. 6.6 and 6.7 (bottom-right panels).

The longevity of the storm can be incremented if the updraft of the convective cell is not counteracted by the downdraft coming from precipitation. This can be achieved by adding horizontal wind shear in the lower layers of the atmosphere (Klemp and Wilhelmson, 1978). The presence of a vertical wind shear forces rain to fall away from the updraft so that the low-level vapor supply is not impeded from feeding the forming cloud. By comparison of Figs. 6.7 and 6.8, we observe that the extension and maximum con-

centrations of rainwater in the symmetric simulation are deficient with respect to the non-symmetric case. The same figures depict q_c and velocity (u, w) as well. By 7500 s (not shown), q_r of the symmetric case has reached a value of 0.017 g kg^{-1} versus a maximum $q_r = 7.23 \text{ g kg}^{-1}$ with wind shear.

6.6.2 Case 2: Storm-WKR88

From Weisman et al. (1988) (from now on, WKR88), the domain is 180 km wide and 18 km deep. A layer of unsaturated water vapor characterizes the lower troposphere. The amount of moisture decreases with height and the flow is forced by the vertical wind shear of Fig. 6.5. The bottom and top boundaries are modeled as non-viscous solid walls, with periodicity along the x -direction. The perturbation of θ_ρ is centered at $(x_c, z_c) = (90, 2) \text{ km}$.

Results Case 2. To compare directly with the reference, grid increments $\Delta x = 2 \text{ km}$ and $\Delta z = 0.72 \text{ km}$ (90×25 elements) are used. The top of the computational domain is at $z = 18 \text{ km}$, but the physical domain is limited by the absorbing layer which is set at 11.7 km . The evolution of the storm is represented in Fig. 6.10 at three consecutive simulation times. The cloud starts to form at approximately $t = 900 \text{ s}$ (not shown) with precipitating water reaching the ground at $t = 1800 \text{ s}$ (Fig. 6.10-top). Rain is accompanied by evaporation in the proximity of the ground. This mechanism causes the air to cool down and generate a density current that spreads at the surface. If the density current collides with the incoming wind from east, new deep lifting is triggered and consequently new cells are formed. This is in agreement with the theory that is graphically illustrated in Fig. 6.9. Fig. 6.10-bottom shows a fully formed new precipitating cell in the rear region of the main convective structure.

The strong updraft that starts with the formation of the first cell continues until the cloud has reached the tropopause at 12000 m . At that point, because of the change in atmospheric stability, the cloud begins to spread with the characteristic anvil shape of deep storms. Since there are no quantitative results available on the water quantities simulated by Weisman and colleagues, the main structure of the storm is not sufficient to get a sense of accuracy of our simulation with respect to a similar real situation. Nevertheless, the developing storm is characterized by important vertical velocities that are indeed reported in WLR88. We use their values as a metric for comparison and, in Fig. 6.11 (top-left), we report the time evolution of w_{max} . The maximum vertical velocity has peaks that are of the same order of magnitude of w_{max} plotted in reference. However, unlike the reference, the vertical velocity presents large peaks also when $t > 8000 \text{ s}$. We believe that this is due to the total lack of either artificial or subgrid-scale eddy viscosity that would, otherwise, have effects on the energy and momentum balance of the problem. From the curves of q_c , we observe that the maximum cloud concentration remains almost constant around a value of 2 g kg^{-1} starting from its full development at approximately 2000 s . Together with the representation of the cloud contour in Fig. 6.10, the simulated clouds change distribution without great variation of their mixing

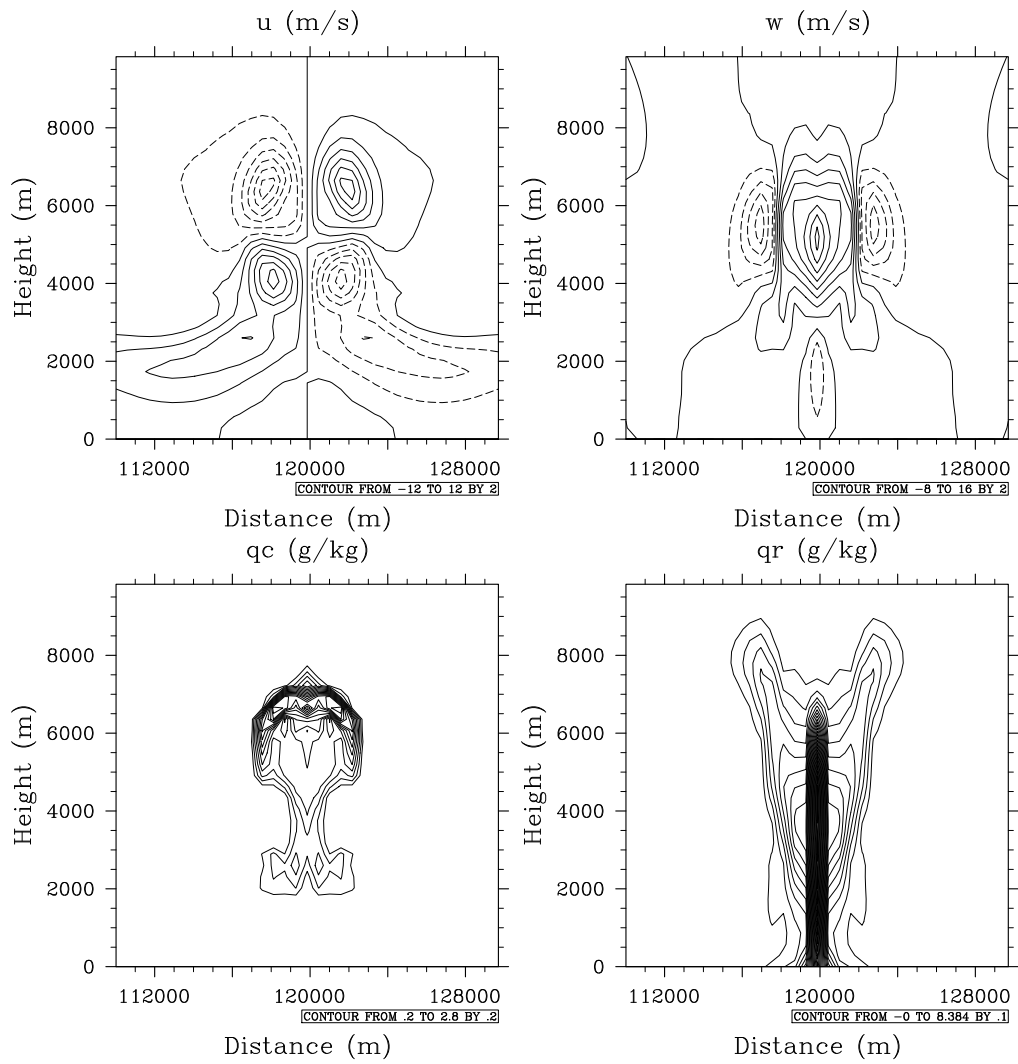


Figure 6.6: *Case 1*. Simulated fields after 1550 s without wind shear. u (top left), w (top right), q_c (bottom left), and q_r (bottom right). Given a shearless velocity field, the original symmetry is preserved. This plot is linked to Fig. 6.4-left where the gravity waves generated by the symmetric storm are plotted.

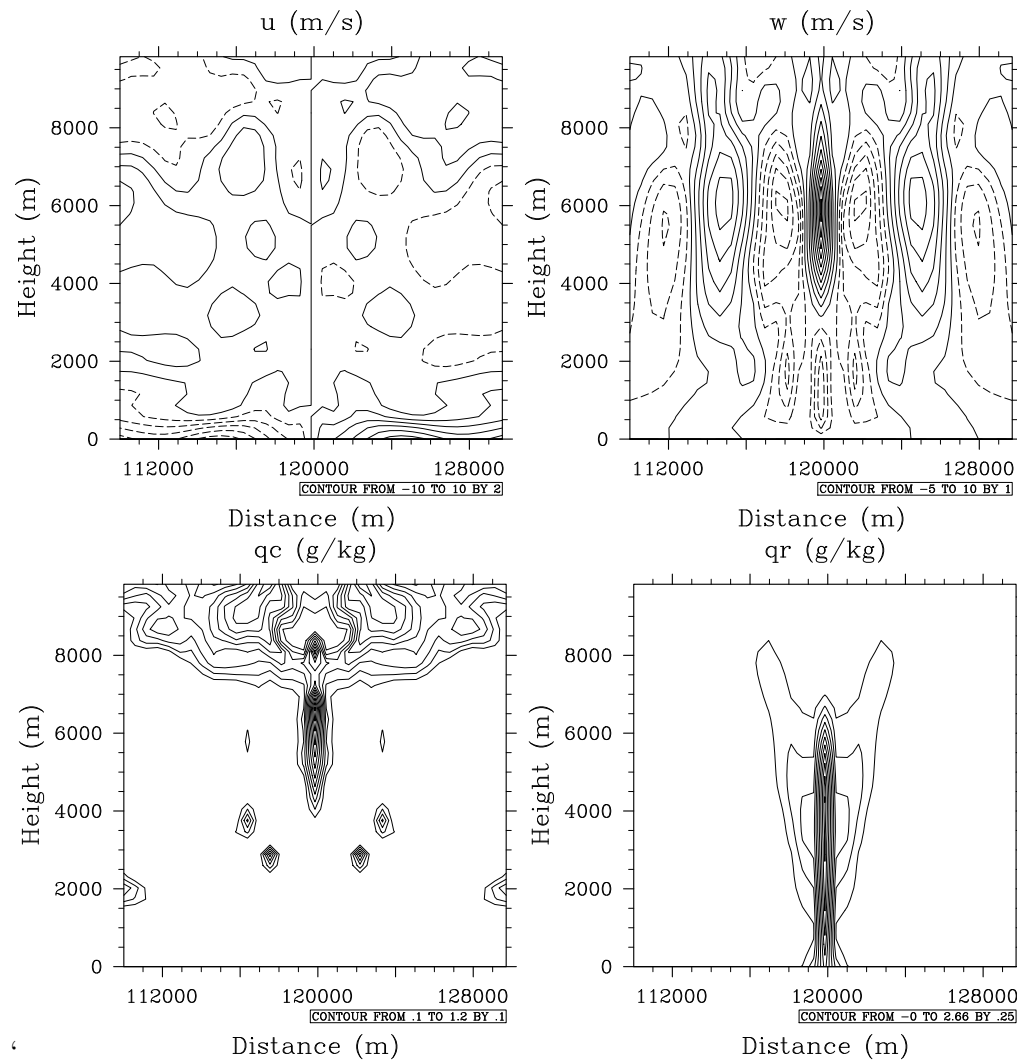


Figure 6.7: *Case 1*. Same as Fig. 6.6, but after 3600 s. The cloud has reached the tropopause where it tends to spread left- and rightward. The storm intensity is decreasing in terms of rain water concentration.

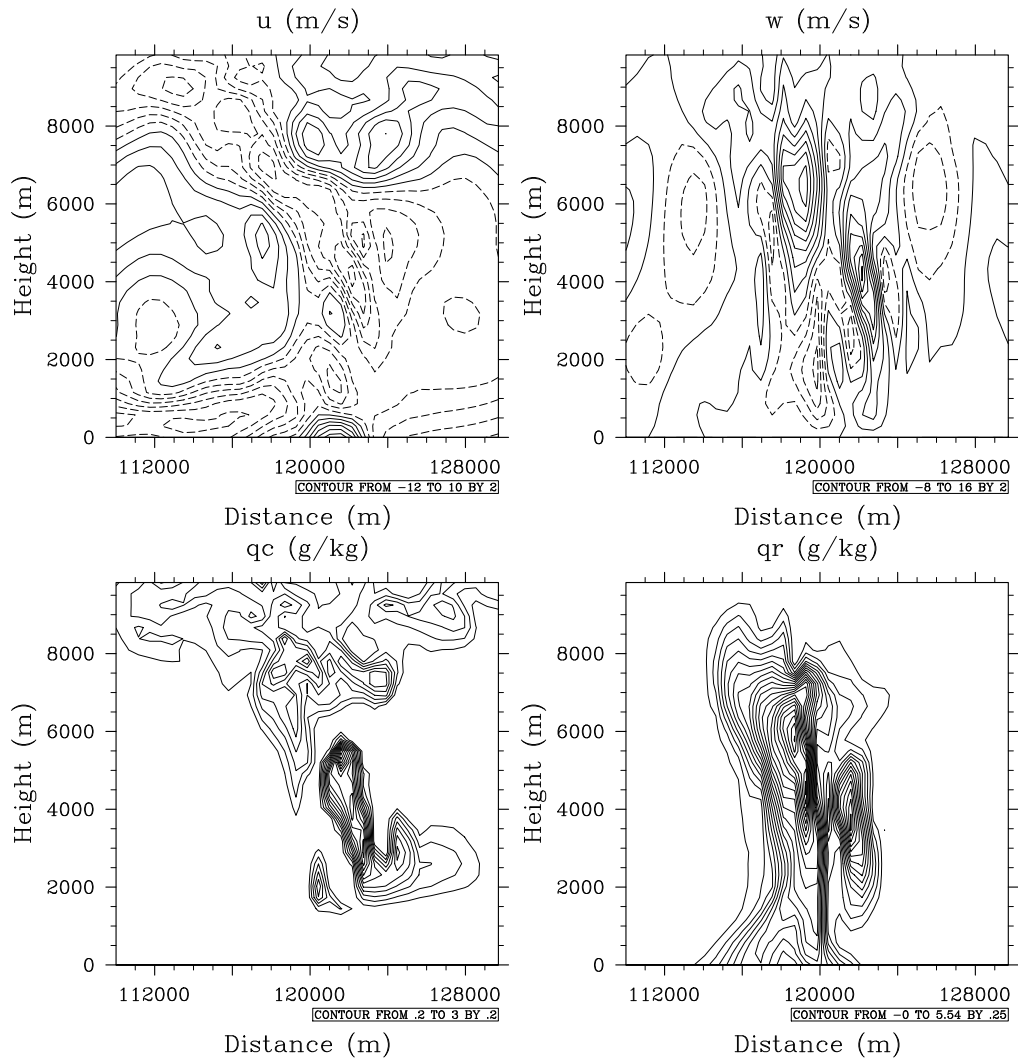


Figure 6.8: *Case 1*. Simulated fields after 3600 s with wind shear. u (top left), w (top right), q_c (bottom left), and q_r (bottom right). The wind shear has affected the evolution of the storm with respect to its shearless counterpart of Fig. 6.7. Concentration of rainwater is twice as large as that obtained by the symmetric simulation. The gravity waves for this problem are represented in Fig. 6.4-right.

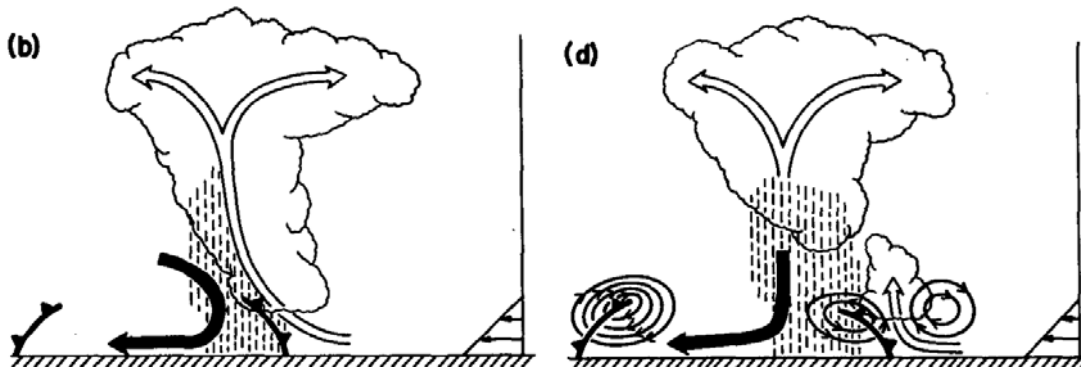


Figure 6.9: Reproduced from Fig. 5 of Rotunno et al. (1988). Rotunno et al. justify the presence of new cells triggered by the interaction of the low-level shear and the circulation of the rain-produced cold pools. With the plot in (d), they contradict the previous theory of Thorpe et al. (1982), represented in (b), who argue that a long-lived cell is produced because the cold pools remain beneath the cloud. In contrast, the interaction is such that new deep lifting is produced and triggers new cells. In the model calculations that follow, we observe the expectations of Rotunno and colleagues. Image reprinted with permission. ©American Meteorological Society.

ratio. On the other hand, once the storm has reached its mature stage, precipitation is expected to decrease. The trend in the time evolution of q_r (Fig. 6.11-bottom-right) is consistent with this consideration.

6.6.3 Case 3: Storm-GGD12

This test is the same as *Case 2* but it extends along 240 km in the horizontal and 24 km in the vertical directions as in Gaberšek et al. (2012) (from now on, GGD12). The vertical extension of the physical domain is limited by the absorbing layer which is set at 11.7 km . To analyze the dependence of the solution with respect to grid spacing, the simulation is performed on four computational grids with horizontal resolutions $\Delta x = \{4660, 2330, 1160, 580\}\text{ m}$ and a common vertical spacing $\Delta z = 290\text{ m}$. The grid spacing is chosen to reproduce, as faithfully as possible, the nominal grid used by GGD12 with high-order spectral elements of unevenly spaced nodes.

Results Case 3. The storm evolution resolved on the finest grid with $\Delta x = 580\text{ m}$ and $\Delta z = 290\text{ m}$ (414×83 quadrilateral elements) is presented in Figs. 6.12 and 6.13. Given the same initialization of GGD12, we obtain a similar behavior of the solution in terms of the structure (i.e. shape and extension), and accumulated water. The cloud starts to form at approximately $t = 900\text{ s}$ (not shown) with precipitating water reaching the ground at $t = 1800\text{ s}$ (Fig. 6.12, top). The vertical motion of the cloud top stops at the tropopause, where the stability of the background state increases. The cold pool that forms on the surface is visible in Figs. 6.13 and 6.17. The colder and denser air is the source of a density current that interacts upwind with the wind coming from east. This causes further lifting of water vapor to form new cells. By 6000 s one extra cell has formed and is visible in the middle plot of Fig. 6.12. Fig. 6.14 shows the velocity vector

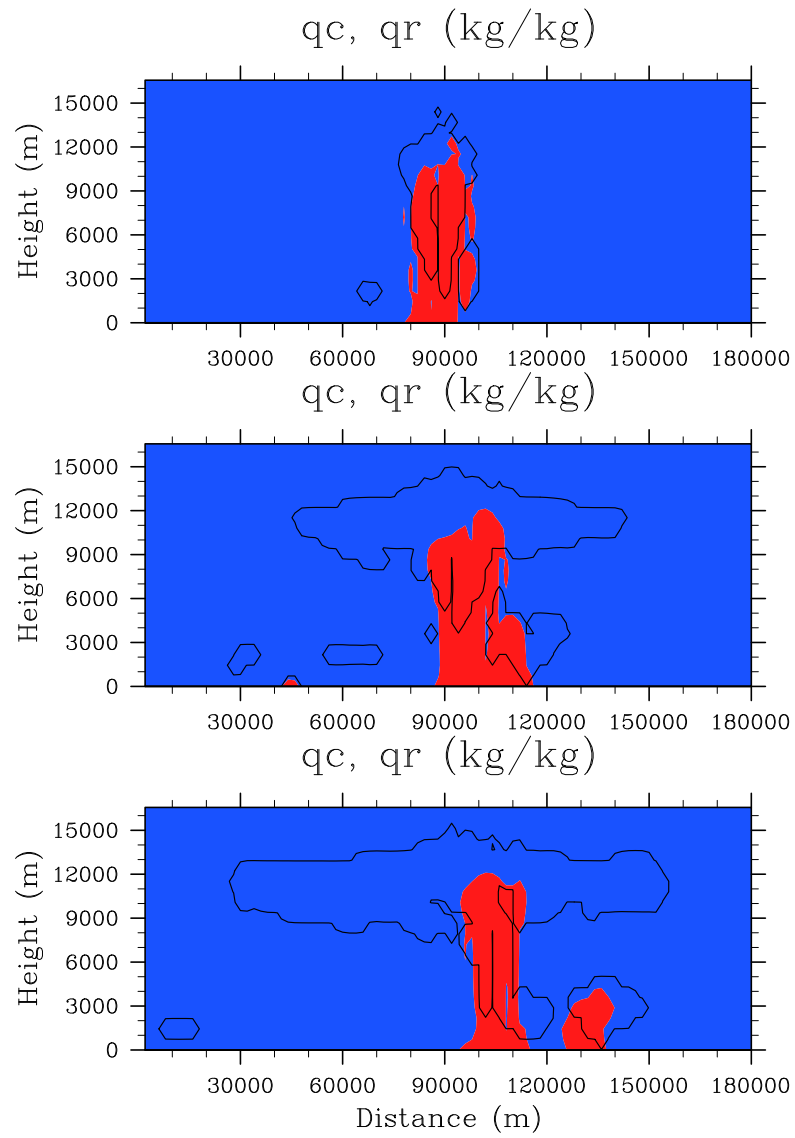


Figure 6.10: *Case 2*. Storm at 3600 s (top), 7200 s (middle), and 12000 s (bottom). The thick black contour represents the outline of the cloud where $q_c = 10^{-5} \text{ kg kg}^{-1}$. Precipitating water $q_r = 10^{-4} \text{ kg kg}^{-1}$ is plotted in solid red color. Solved on uniform grid with $\Delta x = 2 \text{ km}$ and $\Delta z = 0.72 \text{ km}$ (90×25 quads).

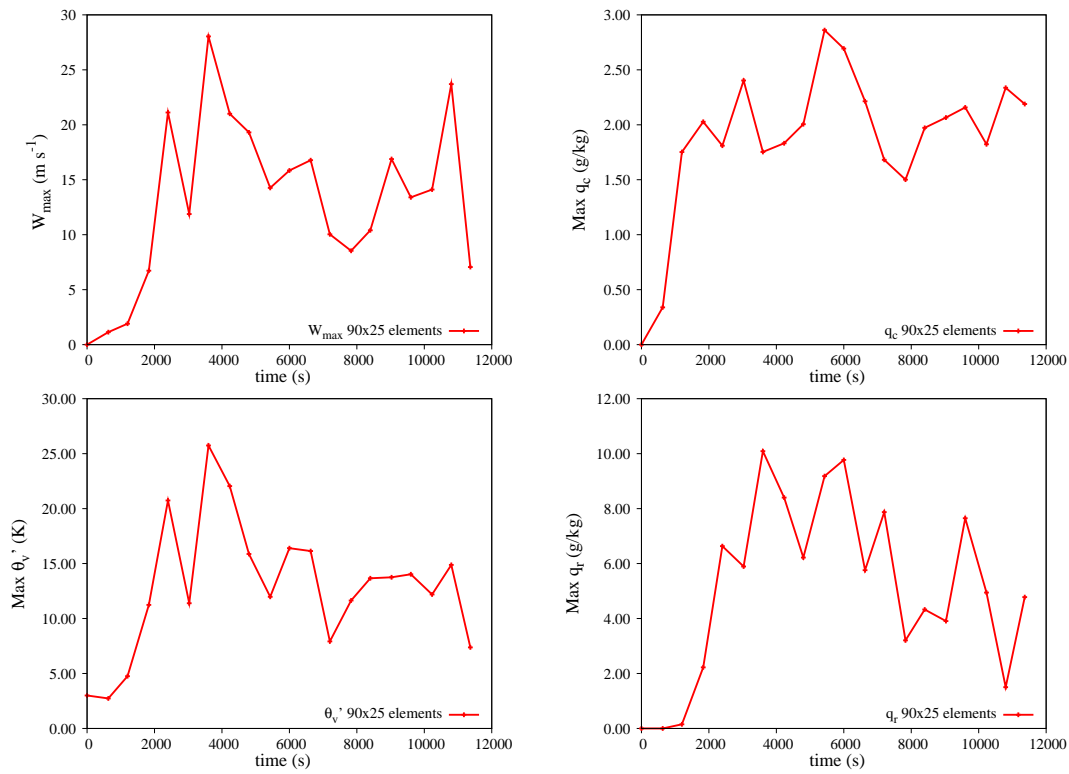


Figure 6.11: *Case 2*. Time series of: w_{max} (top-left), q_c (top-right), θ'_p (bottom-left), and q_r (bottom-right) at different resolutions. Wind shear from $u_s = -12, m\ s^{-1}$ on the ground to $u_s = 0\ m\ s^{-1}$ at $z = 2.6\ km$.

field of the developing storm. The downdraft associated with precipitation is particularly visible in the middle plot of the figure. The important vertical motion through the core of the developing cloud splits left and right as the cloud top approaches the tropopause.

The storm features important changes in vertical velocity during the first simulated hour, with peak vertical speeds reaching $w_{max} \sim 25 \text{ m s}^{-1}$ in the interval $3000 \text{ s} \leq t \leq 7000 \text{ s}$. From that point on the storm reaches a quasi steady-state with smaller updrafts and is characterized by the classic anvil shape of the cumulus along several hundred kilometers. The time evolution of w_{max} , q_c , q_r and θ'_ρ are plotted in Fig. 6.15, where the values are compared for different grid resolutions. The major differences are in the maxima of q_r and θ'_ρ .

The extension of the cloud (Figs. 6.16 and 6.17) and the amount of precipitation (Fig. 6.18) are very sensitive to Δx . Although the characteristic times of the storm are comparable at different resolutions, q_c and q_r are more localized as the grid is refined. The structure of the storm for $\Delta x = 580 \text{ m}$ and $\Delta x = 1160 \text{ m}$ is similar to that described by the theory of Rotunno et al. (1988). This specific structure is almost completely lost on the coarsest grid with $\Delta x = 4660 \text{ m}$. On the one hand, this variability is due to the non-linearity of the microphysical processes (Gaberšek et al., 2012). On the other, because a typical spatial extension of a convective cell is about 8 km , grid spacing should not exceed 2 km to represent the cloud evolution explicitly. Because the coarsest resolution is more than twice the size of the 2 km threshold, most of the sub-grid scale physics that affects the formation of the cloud is not resolved explicitly. For an extended analysis of grid sensitivity, refer to Weisman et al. (1997) and Bryan and Morrison (2011).

The accumulated precipitation represented in Fig. 6.18 is quantitatively comparable with the results of (Gaberšek et al., 2012).

6.6.4 Case 4: O-Clouds

As briefly mentioned in the introduction to this chapter, other types of clouds and storms can be simulated with the same model. The type of physics is totally dependent on the initial state of the atmosphere, on the wind shear, and on the topographical characteristics of the domain. In contrast to the convective storms presented in the previous cases, the orographic storm presented below features a cloud that, once it has formed, remains in a quasi-stationary state on the lee side of the mountain and does never propel upward to higher altitudes. The case is built as follows.

A single-peaked Agnesi mountain of height $h_c = 2.5 \text{ km}$ and semi-width $a_c = 10 \text{ km}$ is centered at $(x_c, z_c) = (40, 0) \text{ km}$ in a domain that is 80 km wide and 24 km deep. The mountain is defined by

$$h(x) = \frac{h_c}{1 + \left(\frac{x-x_c}{a_c}\right)^2}. \quad (6.21)$$

The initial field is equivalent to the sounding used so far in the previous tests. The initial values have been interpolated onto a terrain-following grid. The lifting of water vapor now occurs due to purely mechanical reasons: a sheared flow moving at velocity

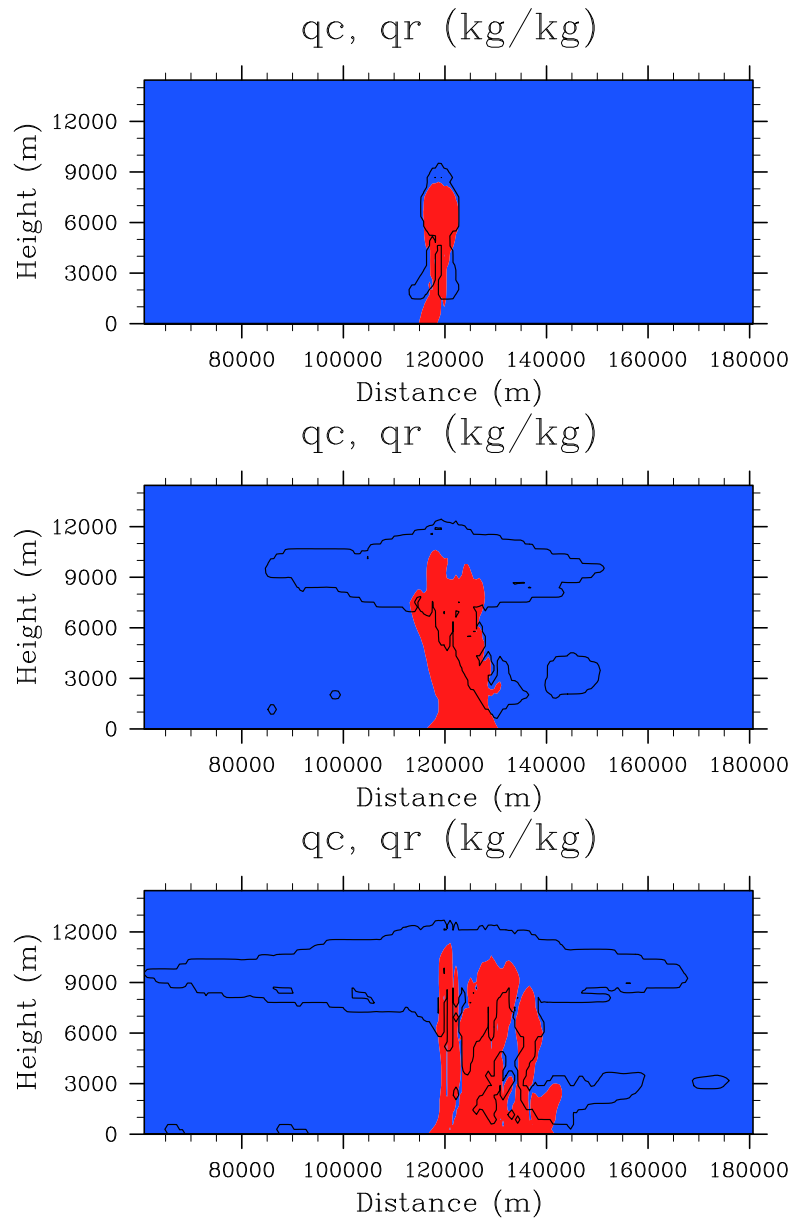


Figure 6.12: *Case 3*. Storm at 1800 s (top), 6000 s (middle), and 9000 s (bottom). The thick black contour represents the outline of the cloud where $q_c = 10^{-5} \text{ kg kg}^{-1}$. Precipitating water $q_r = 10^{-4} \text{ kg kg}^{-1}$ is plotted in solid red color. Solved on uniform grid with $\Delta x = 580 \text{ m}$ and $\Delta z = 290 \text{ m}$ (414×83 quads).

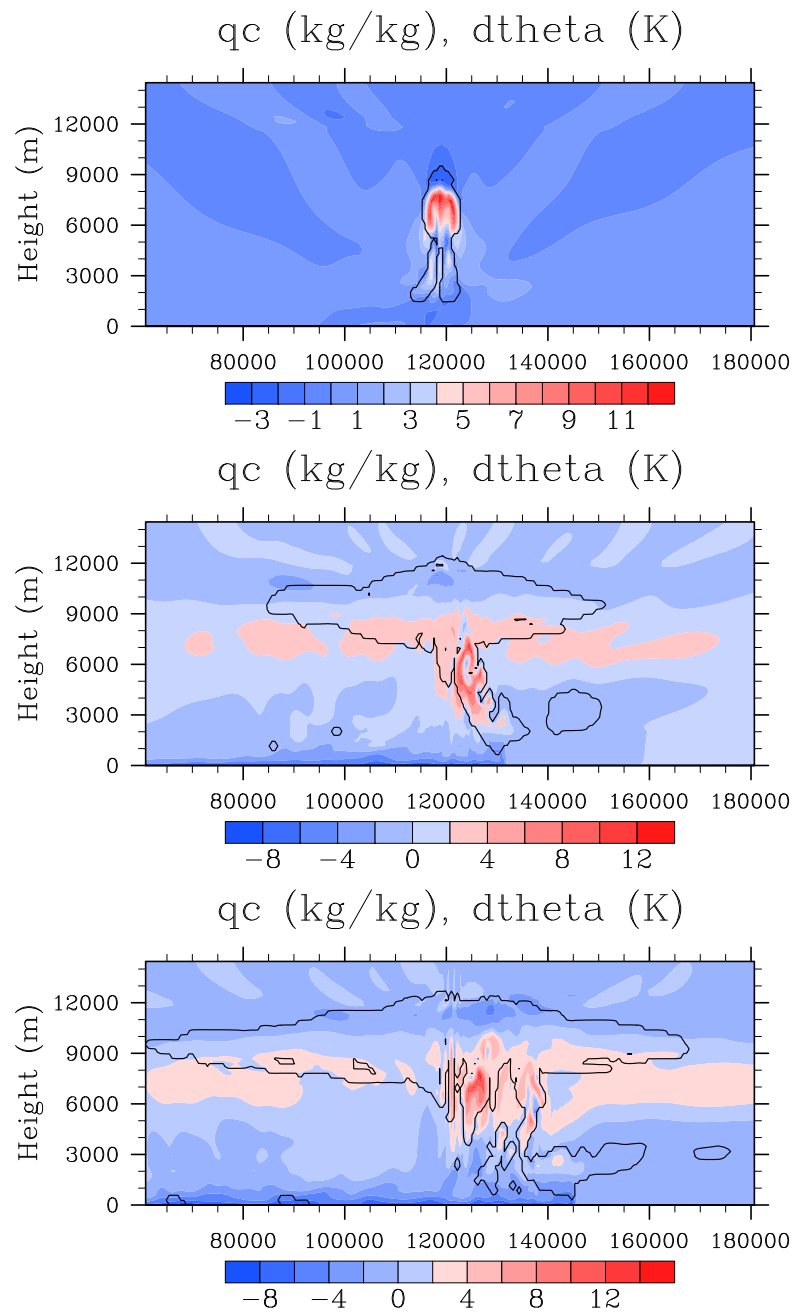


Figure 6.13: *Case 3*. As in Fig. 6.12, but with filled contours of θ'_p . At 1800 s (top), 6000 s (middle), and 9000 s (bottom). The color scales are left free to show the absolute extrema of θ'_p .

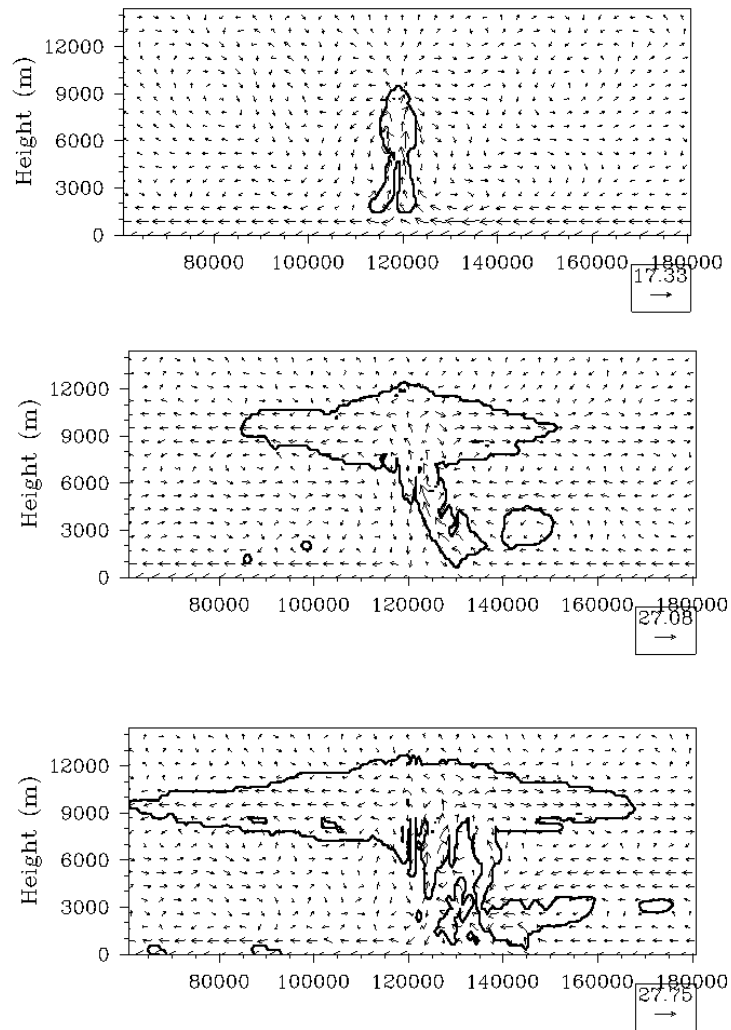


Figure 6.14: *Case 3*. Velocity vectors and cloud water contour ($q_c = 10^{-5} \text{ kg kg}^{-1}$) for the storm simulation on the grid of $\Delta x = 580 \text{ m}$ and $\Delta z = 290 \text{ m}$ (414×83 quads). At 1800 s (top), 6000 s (middle), and 9000 s (bottom).

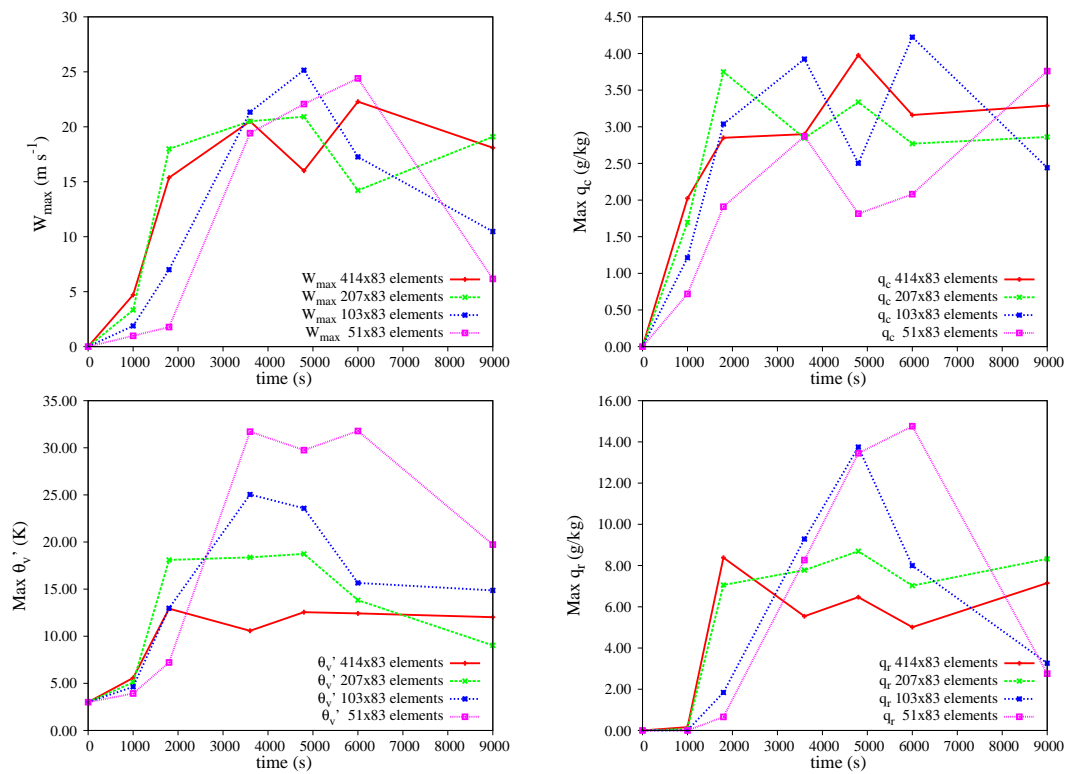


Figure 6.15: *Case 3*. Time series of: w_{max} (top-left), q_c (top-right), θ'_p (bottom-left), and q_r (bottom-right) at different resolutions. Wind shear from $u_s = -12, m s^{-1}$ on the ground to $u_s = 0 m s^{-1}$ at $z = 2.6 km$.

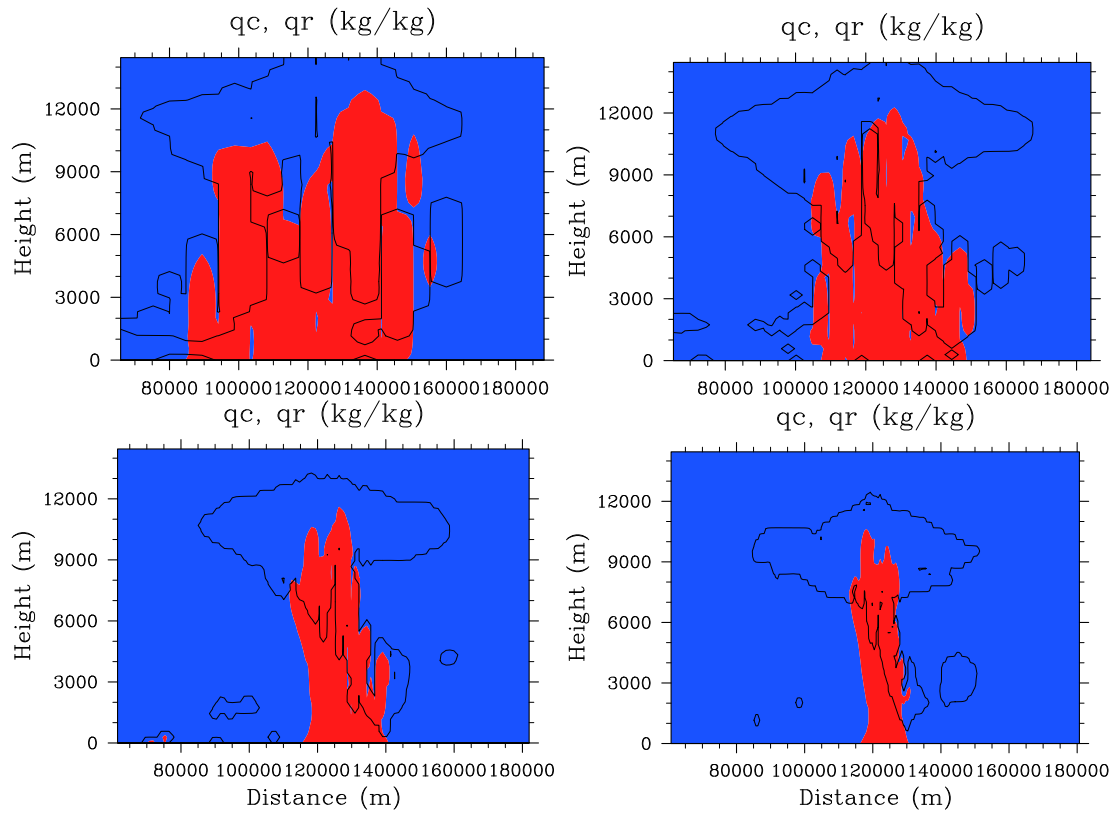


Figure 6.16: *Case 3*. Storm at 6000 s for different grid resolutions. Top-left: $\Delta x = 4660$ m and $\Delta z = 290$ m (51×83 quads). Top-right: $\Delta x = 2330$ m and $\Delta z = 290$ m (103×83 quads). Bottom-left: $\Delta x = 1160$ m and $\Delta z = 290$ m (207×83 quads). Bottom-right: $\Delta x = 580$ m and $\Delta z = 290$ m (414×83 quads). The thick black contour represents the outline of the cloud where $q_c = 10^{-5} \text{ kg kg}^{-1}$. Precipitating water $q_r = 10^{-4} \text{ kg kg}^{-1}$ is plotted in solid red color.

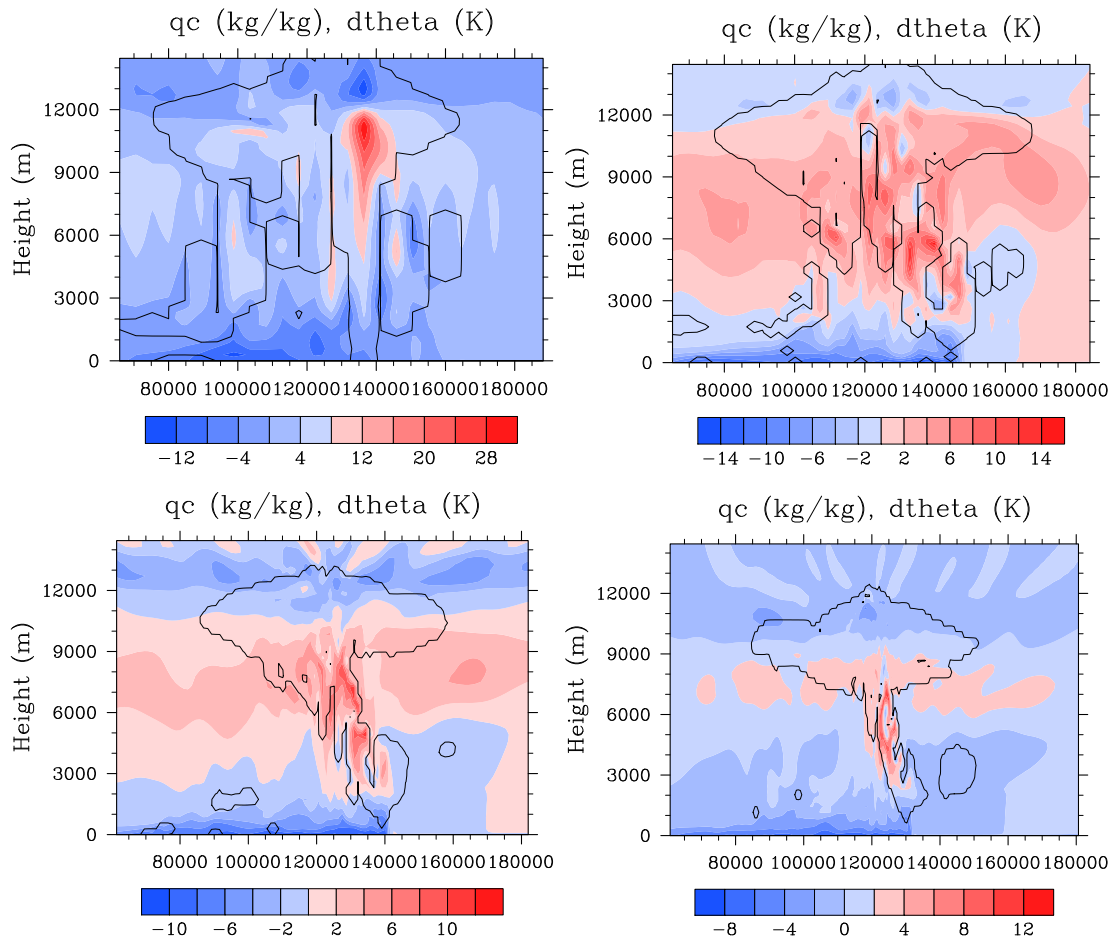


Figure 6.17: *Case 3*. As Fig. 6.16 but plotting the filled contours of θ'_ρ . The color scales are left free to show the absolute extrema of θ'_ρ for every grid resolution.

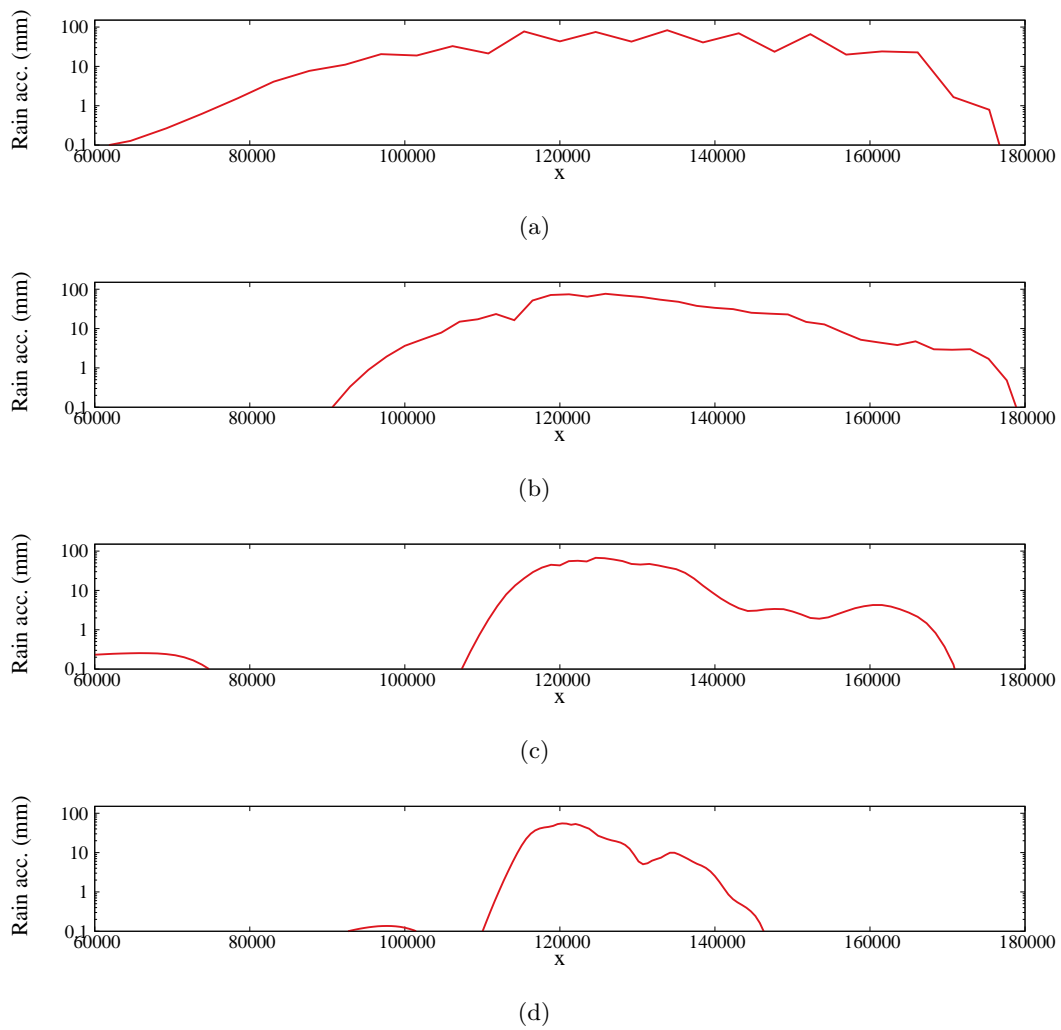


Figure 6.18: *Case 3*. Accumulation of rain water on the ground after 10000 s (≈ 3 hours). (a) $\Delta x = 4660\text{ m}$ and $\Delta z = 290\text{ m}$ (51×83 quads). (b) $\Delta x = 2330\text{ m}$ and $\Delta z = 290\text{ m}$ (103×83 quads). (c) $\Delta x = 1160\text{ m}$ and $\Delta z = 290\text{ m}$ (207×83 quads). (d) $\Delta x = 580\text{ m}$ and $\Delta z = 290\text{ m}$ (414×83 quads).

$(u, w) = (-12, 0) \text{ m s}^{-1}$ on the ground is forced to move along the topography. Because of the specific stability conditions of the atmosphere, the cloud forms and develops upstream of the mountain. This condition is known as *upstream cloud triggering* (see Houze (1993) for details).

Results Case 4. A grid of 50×30 bi-linear quadrilateral elements is used to discretize the domain. During the first minutes after the initial state, the wind advects the water vapor along the mountain surface and lifts it to the colder region at higher altitudes. Condensation starts to occur at approximately $t = 700 \text{ s}$. Fig. 6.19 shows the evolution of the cloud represented with contours of $q_c = 10^{-5} \text{ kg kg}^{-1}$ and $q_r = 10^{-4} \text{ kg kg}^{-1}$. Once the cloud begins to form, a deviation of velocity towards the region of lifting moisture is visible in the area upstream of the mountain. In the frontal region of the cloud, velocity increases and deviates upwards as the cloud develops. In the same region there is an important increase of θ'_ρ due to the heat release during condensation of cloud water into rain. The region of maximum rain and storm activity keeps steady and is approximately centered at $(x, z) = (45, 2) \text{ km}$.

As previously observed, precipitation is accompanied by evaporation at ground with direct formation of cold pools. The colder air generates important downslope winds on the windward side of the mountain. The westerly directed velocity vectors are counter-balanced by the downslope wind that is blowing east, as it is shown in Fig. 6.19 (bottom panel). Fig. 6.20 shows the accumulated precipitation after three hours approximately.

6.6.5 Case 5: O-Clouds 3D

A three-dimensional extension of *Case 4* is obtained by defining an Agnesi mountain of height $h_c = 2.5 \text{ km}$, semi-widths $(a_c, b_c) = (5, 3) \text{ km}$, and centered at $(x_c, y_c, z_c) = (30, 20, 0) \text{ km}$ in the domain $[0, 60] \times [0, 40] \times [0, 17.5] \text{ km}^3$. The mountain is defined by

$$h(x, y) = \frac{h_c}{1 + \left(\frac{x-x_c}{a_c}\right)^2 + \left(\frac{y-y_c}{b_c}\right)^2}. \quad (6.22)$$

The domain is crossed by a horizontal wind along the x-direction with vertical shear equivalent to the one used previously. A no-slip condition is applied on the surface boundary while periodic boundaries are applied along x and y . An absorbing layer is included at $z = 11.5 \text{ km}$.

Results Case 5. The domain is discretized using $40 \times 40 \times 30$ linear hexahedra. The thermodynamic conditions and the mountain characteristics induce the formation of a lenticular-type cloud similar to the one over Mount Hood (Oregon) shown in Fig. 6.21. q_v is advected along the mountain with maximum velocity of approximately 12 m s^{-1} on the surface. Once water vapor reaches saturation in the levels of lower temperature, condensation occurs on the peak of the mountain. Like its 2D analogue, as time goes by the cloud increases in volume but stands still around the mountain. This is possible as

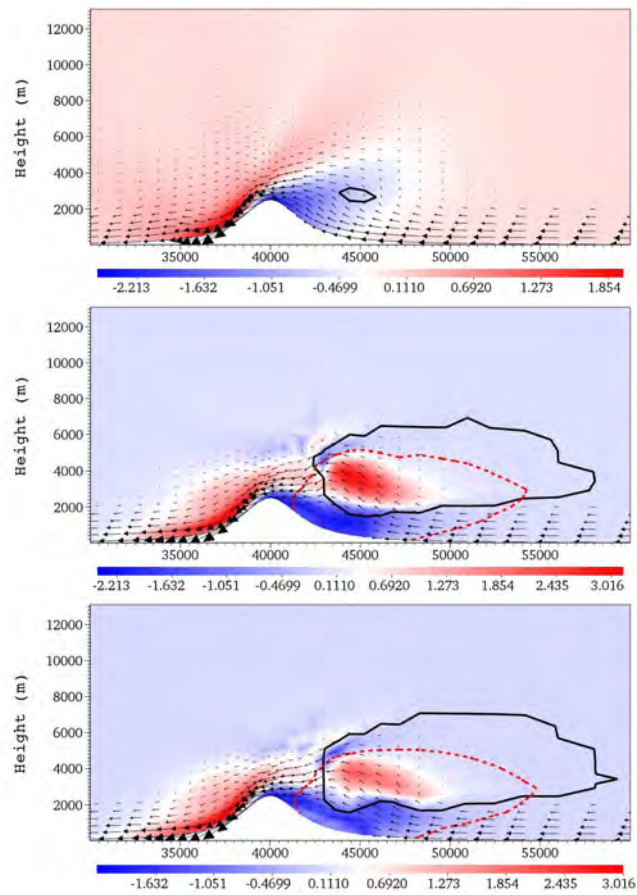


Figure 6.19: *Case 4*. (a): filled contours of θ'_ρ and velocity vectors at 725 s (top), 6000 s (middle), and 10000 s (bottom). Resolved on a grid of 50×30 elements, with an average horizontal grid size of $\Delta x = 1600 m$, and an average vertical grid size of $\Delta z = 750 m$. The thick black and the red dashed contours represent the outline of the cloud where $q_c = 10^{-5} kg kg^{-1}$ and $q_r = 10^{-4} kg kg^{-1}$, respectively.

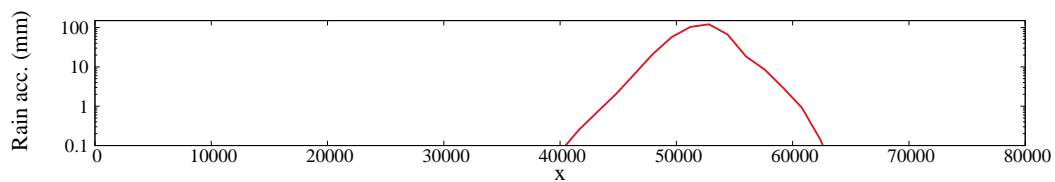


Figure 6.20: *Case 4*. Accumulated precipitation after 10000 s ($\approx 3 hrs$).



Figure 6.21: Lenticular cloud over Mount Hood, Oregon, U.S.A. (Photo by Lourdes Coronado).

long as there is incoming vapor that feeds the cloud from below. The simulated cloud with precipitation at $t_f = 3500 s$ is shown in Fig. 6.22. Like in the two-dimensional case, precipitation is localized in the upwind side of the mountain. In Fig. 6.23 we show three views of the volume rendering of the cloud after 8500 s. The size of the cloud has been increasing but the storm has not moved further downwind than where it originally formed. The real sounding of the event at Mount Hood is not available, so we are unable to say with certainty that the simulated cloud is the numerical representation of the lenticular cloud of the photograph. Nonetheless, it is encouraging to see that the method can simulate different types of storms given geographical and thermodynamic conditions that allow them. *Case 5* and *Case 6* close this work with the realistic simulation of two different three-dimensional events to prove that the code is three-dimensional by construction. As we have stated before, however, the lack of turbulence modeling is still a limitation that must be overcome to eventually simulate a storm event properly either in two or three dimensions. Only a qualitative analysis will be reported below, whereas a complete analysis of three-dimensional moist simulations will be the topic of future work.

6.6.6 Case 6: Convective cell 3D

The three-dimensional simulation of a convective cell is defined in the domain $[0, 80] \times [0, 80] \times [0, 20] km^3$. The initial field, given by the sounding of Fig. 6.3, is perturbed by an ellipsoidal thermal θ' centered at $(x_c, y_c, z_c) = (40, 40, 2) km$ and defined by

$$\theta' = \theta_c \cos^2 \frac{\pi r}{2} \quad \text{if } r \leq 1, \quad (6.23)$$

with amplitude $\theta_c = 3.5 K$ and $r = \sqrt{(x - x_c)^2/x_r^2 + (y - y_c)^2/y_r^2 + (z - z_c)^2/z_r^2}$. The ellipsoid has radii $(x_r, y_r, z_r) = (10, 10, 1.5) km$.

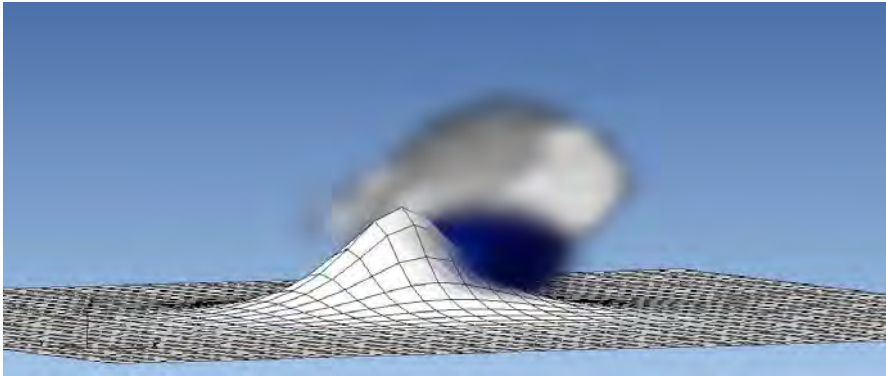


Figure 6.22: *Case 6*. Simulation at $t_f = 3500$ s. In grey, volume rendering of $q_c = 10^{-5} \text{ kgkg}^{-1}$. In blue, volume rendering of $q_r = 10^{-4} \text{ kgkg}^{-1}$. (Rendering done with [VisIt \(2000\)](#)).



(a)



(b)



(c)

Figure 6.23: *Case 6*. Volume rendering of q_c at $t_f = 8500$ s. (a) xz plane. (b) yz plane. (c) $-yz$ plane.



Figure 6.24: *Case 6.* A tornadic supercell thunderstorm. La Plata, Maryland, USA. 28 April 2002 (Photo by Steven Maciejewski).

An example of a developing thunderstorm is shown in Fig. 6.24. What we want to show with this photograph is the complex structure and the different scales that characterize the core of a convective storm. Although the cloud has reached the tropopause, as indicated by the anvil-shaped upper boundary in the photograph, the overshooting top visible in the central region of the cloud indicates that the storm is still developing.

Results Case 6. The volume is subdivided into $60 \times 60 \times 60$ elements first, and later into $80 \times 80 \times 60$ elements. The elements are clustered towards the center of the domain in x and y , given a small clustering on the lower layers along z as well (see Fig. 6.25). The horizontal resolution on the coarser grid ranges from $\Delta x = \Delta y = 500 m$ in the central region of the domain to $\Delta x = \Delta y = 2750 m$ at the external boundaries. The finer resolution ranges from $\Delta x = \Delta y = 300 m$ in the central region of the domain to $\Delta x = \Delta y = 1665 m$ at the external boundaries. Δz varies from $200 m$ in the lowest layer to $525 m$ in the upper atmosphere in both cases. The node clustering is used to keep the global grid size small but still maintain a sufficiently fine grid in the region where the cell is expected to form. The domain is crossed by a horizontal wind along the x -direction with vertical shear equivalent to the one used previously. A no-slip condition is applied on the surface boundary while periodic boundaries are applied along x and y . A vertical absorbing layer is included at $z = 15 km$. Along x and y a layer of lower absorption coefficient is defined in the regions $0 < x, y < 20 km$ and $60 < x, y < 80 km$ as well. A qualitative view of the storm after $6000 s$ is given in Fig. 6.26.

A qualitative comparison between the simulation on two grids is plotted in Figs.

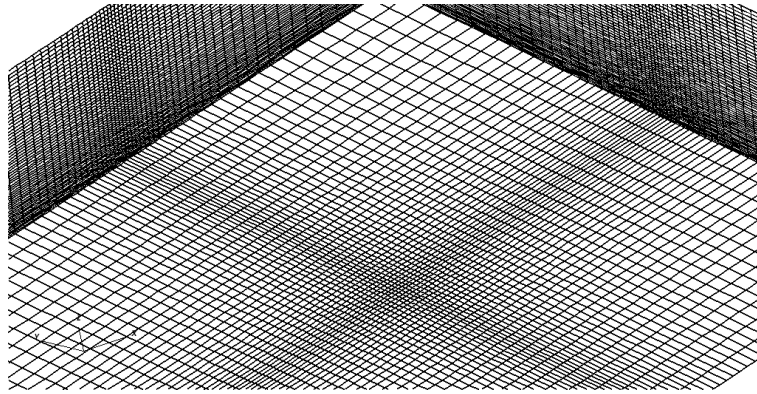


Figure 6.25: *Case 6*. Three slices of the 3D Computational grid.

6.27 and 6.28, where the evolution of the cloud is represented between $t = 1750 s$ and $t = 6000 s$. The same situation already observed in 2D is observed in 3D as well: the extension of the simulated cloud on the finer grid remains more compact as time evolves. Both clouds initiate in similar ways, with comparable sizes during the first minutes; however, as the storm evolves, the simulation on the coarse grid produces a convective cell that extends more in the horizontal and vertical directions. In Fig. 6.29, the complex structures inside the storm can be visualized by looking at the velocity field. The region where the cell flattens corresponds to approximately $z = 12 km$. The overshooting top is the evidence that the storm is in the middle of its activity.

In the future, a deeper investigation of the sensitivity to different wind shears and grid resolutions should be carried on for a full assessment of the code. Once the method is fully evaluated, the addition of Coriolis effects and turbulence should set the basis to simulate tornadic structures on high resolution grids in a way similar to that reported by Proctor et al. (2012).

6.7 Summary and discussion

In this chapter, the finite element algorithm stabilized by the variational multiscale method was assessed for its ability to simulate stratified moist atmospheric flows. The Euler equations were coupled to a set of three transport equations that model advection of water vapor, cloud water, and rain in the atmosphere. A Kessler microphysics parametrization was used to model phase changes during the formation and development of a convective storm. The algorithm was tested by simulating two- and three-dimensional idealized mesoscale warm clouds with and without wind shear given a synthetic initial state typical of mid-latitude squall-line formation. The main features of the convective cloud such as spatial distribution, storm propagation, maximum variation of the vertical velocities, and rain accumulation were used to compare against the literature. Although the majority of the tests that we presented are two-dimensional, it is understood that the simulation of two-dimensional storms is not a faithful representation

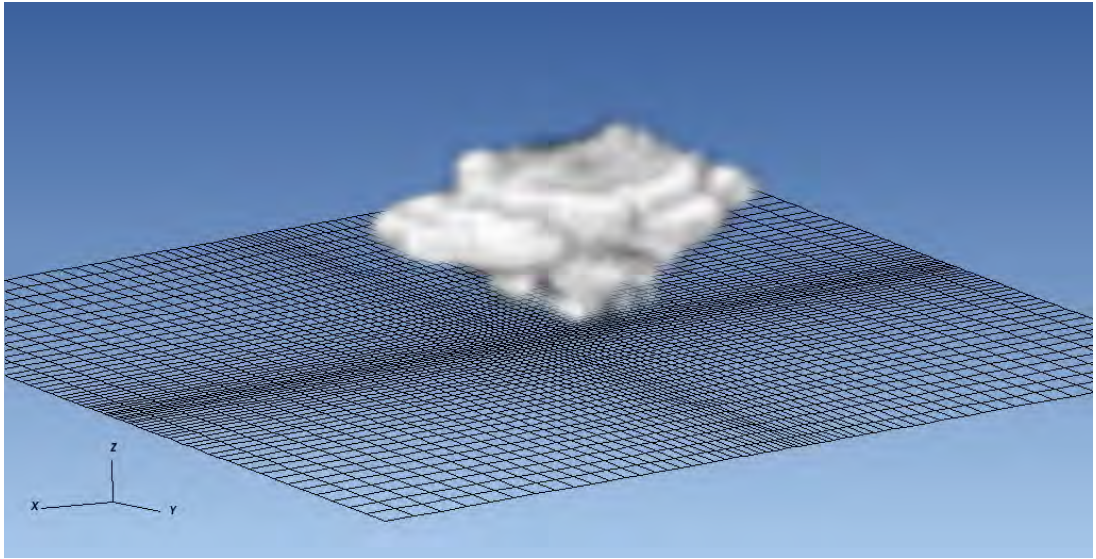


Figure 6.26: *Case 6*. Three-dimensional perspective of the simulated cloud (q_c) at $t_f = 6000$ s. Over-shooting tops are visible, indicating the full activity of the storm at this stage. Relative to this image, the x-directed wind comes from the left-hand side.

of reality due to the lack of important features that would exclusively be present in the case of three dimensions. However, the main task of this study was the assessment of the algorithm in terms of stability properties in the context of wet dynamics as a direct extension of the method reported in Chapter 4 for dry, stratified, low-Mach number flows. In this respect, the algorithm proved to be successful. The finite element solution of the two coupled systems of equations with phase change preserves stability during the long-lasting simulations. Furthermore, the physical results concur with those of other authors. Although the amount of cloud water, its distribution, and duration is indeed comparable with known results, the lack of a turbulence scheme in the current model suggests that, to a certain extent, the simulated phenomena may vary if a turbulence scheme were included. We cannot prove this statement until turbulence is accounted for in some way. This issue must be further investigated and is left open as a possible topic for future work.

We performed simulations for different grid resolutions for which we observed an important variability of the solution. Although the average shape of the cloud does not change substantially from grid to grid, the amount of precipitation and extension of the storm does change, with much more localized precipitation for the finest simulations. As known, the non-linearity of the microphysical processes has a lot to do with it. At the same time, by increasing the grid resolution, more dynamic effects are captured; this fact has important implications on the whole problem because a different dynamics is certain to affect the thermodynamics of the system and hence phase change, and storm triggering.

In conclusion, we proved FEM with VMS stabilization to be a suitable method for

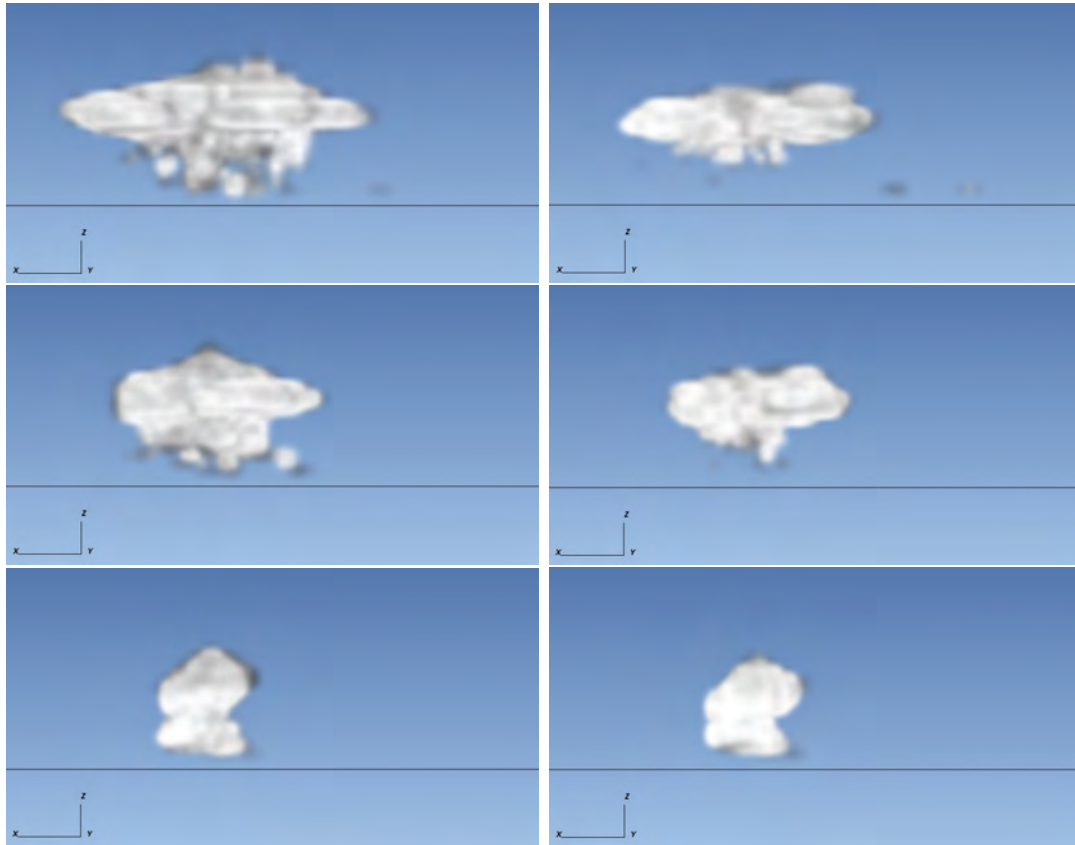


Figure 6.27: *Case 6*. Comparative side-view (xz -plane). $60 \times 60 \times 60$ elements (left column) vs. $80 \times 80 \times 60$ elements (right column). Volume rendering of the evolution of the storm. From bottom to top, q_c is represented at $t = 1750$ s, 3500 s, 6000 s.

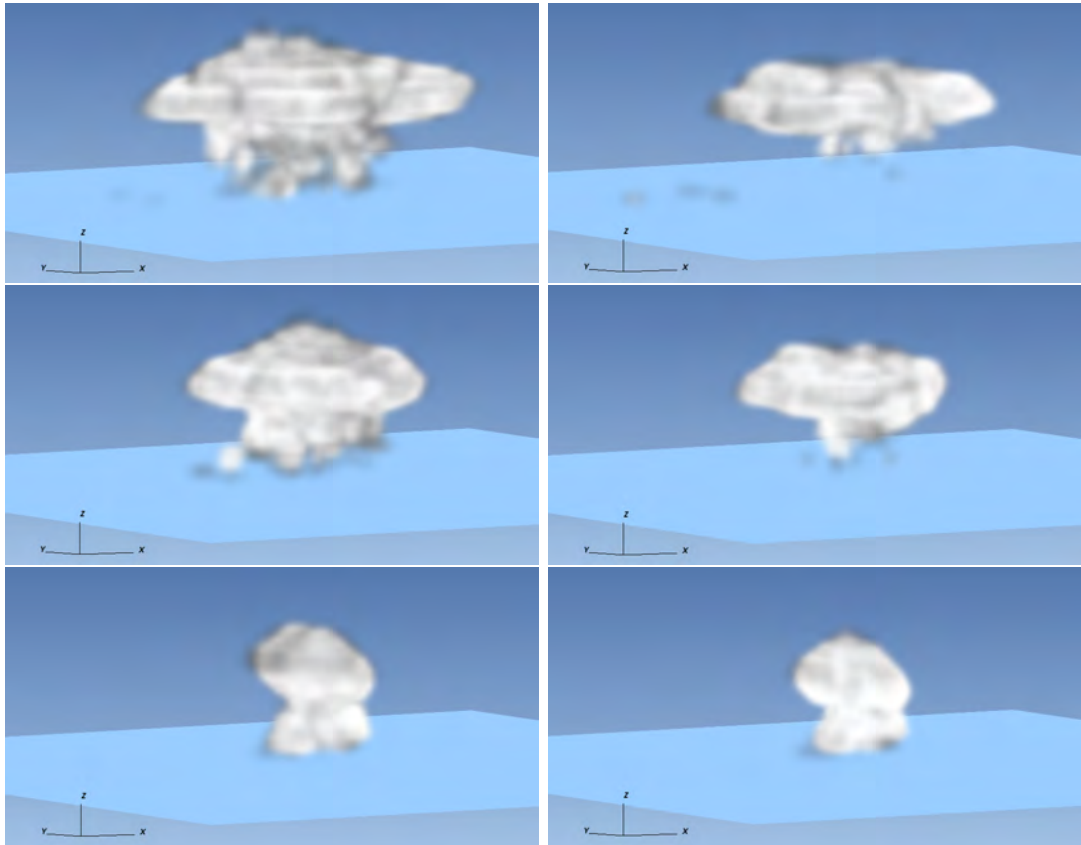
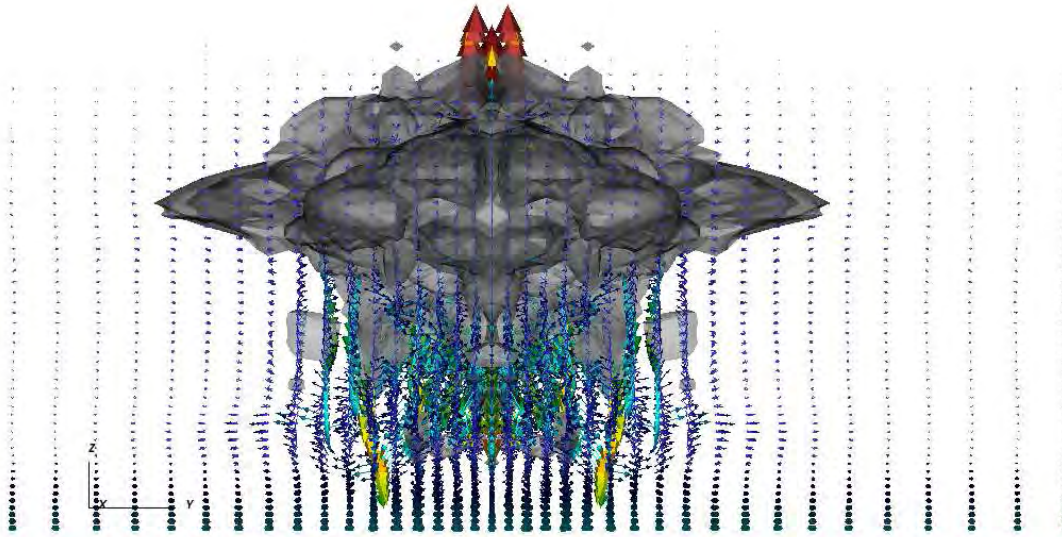
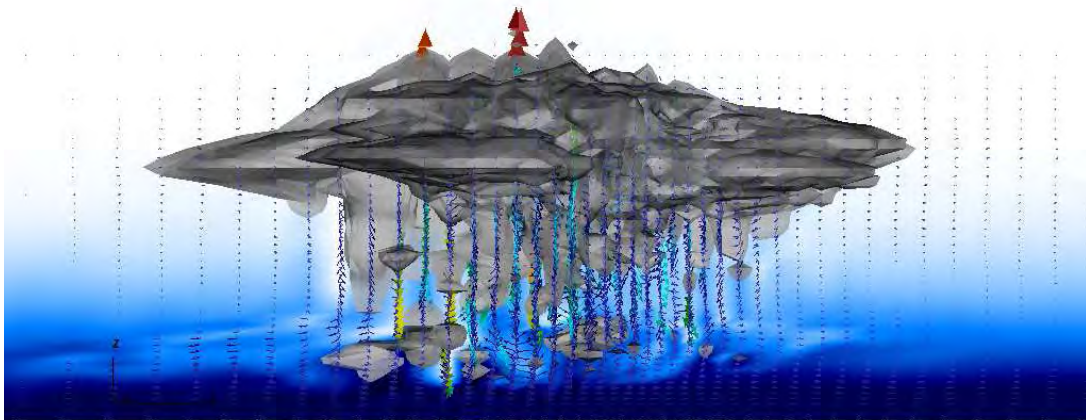


Figure 6.28: *Case 6*. Comparative 3D view. Volume rendering of the evolution of the storm. $60 \times 60 \times 60$ elements (left column) vs. $80 \times 80 \times 60$ elements (right column). From bottom to top, q_c is represented at $t = 1750\text{ s}$, 3500 s , 6000 s .



(a)



(b)

Figure 6.29: *Case 6*. (a) yz -view of q_c (grey scale) with a slice of the vector field on the plane $x = 0$. (b) xz -view of q_c with a slice of q_v (blue-scale filled contours) on the plane $y = 0$. $t_f = 4000$ s. $\|\mathbf{u}\|_{max} \approx 70$ m s $^{-1}$.

complex problems that involve transport phenomena of multi-phase flows in meteorology. The results are encouraging to suggest further analysis. The underlying finite element code can handle both structured and unstructured grids by construction. Because of this, problems with topography can be solved naturally without the need to change or transform the set of equations.

Chapter 7

Conclusions and future work

7.1 Summary

Finite and spectral elements (FEM and SEM) have two fundamental qualities. The complete freedom with respect to the structure of the computational grid is one. The other is represented by their impressive parallel efficiency (i.e., petascale capability). With this in mind, this thesis delves into the use of FEM and SEM to solve problems of typical interest in atmospheric simulations: (i) nonhydrostatic, stratified, dry and moist flows, and (ii) quasi-monotonic solution of the transport equations using high-order schemes.

Because of the instability that occurs in the numerical solution of transport phenomena (i.e. Euler and transport-diffusion equations), we stressed the analysis on stabilization by the variational multiscale method (VMS). The algorithm is an extension of VMS to compressible flows and was adapted to the treatment of thermally perturbed stratified atmospheres first, and then to improve the characteristics of high-order methods in terms of oscillation-free solutions.

After a brief introduction of the physical problems to be solved and of the fundamentals of finite element approximation, the main topics were discussed in three central chapters that are self-contained, yet connected by a common thread represented by the stabilization method on the one hand, and by the physics to be solved on the other. The main achievements of this work are reported as follows:

- In Chapter 4, the finite element method for the solution of the compressible Euler equations that model stratified, nonhydrostatic, dry atmospheric flows was analyzed and assessed. The need for stabilization was fulfilled by VMS for coupled equations that model stratified flows.

The goal of simulating nonhydrostatic, stratified dynamics by the finite element method with VMS stabilization was fulfilled. The main achievement lies in having a stable algorithm with full control on the stabilization parameter that, unlike artificial diffusion or hyperviscosity, is localized and consistent. Within the frame-

work of atmospheric simulations, this idea has important consequences in terms of solution accuracy for the following reasons. Atmospheric dynamics is characterized by very small variations of the thermo-physical quantities to be solved for. Pressure, to mention one, has a dominant, non-uniform distribution in the order of $10^5 Pa$. Nonetheless, it is its variations, orders of magnitude smaller, that drive the evolving thermodynamics of the flow. The same applies to density, velocity, and temperature. For example, $0.01 K$ variations of potential temperature over a reference state at $300 K$ (see *Case 5* in Chapter 4) require the numerical method to resolve, with sufficient accuracy, the small variations without incurring into unwanted cancellation due to possible overdiffusivity. A localized, proportional, consistent stabilization technique is hence necessary. The way variational multi-scale stabilization behaves makes it a candidate for this type of applications.

In this chapter we also designed a methodology to limit the truncation errors encountered in the solution of the hydrostatic states. If not properly computed, the dominant reference state in hydrostatic balance would induce unphysical, vertical accelerations that are not acceptable. The strategy that was implemented consists of explicitly computing the background state at the quadrature points, while interpolating only the deviations of the physical quantities. If the numerical method is not well-balanced with respect to the approximation of the vertical hydrostatics, the entire solution may be negatively affected. This problem has been a focus of atmospheric modelers especially in the presence of steep topographies. Thanks to the logically unstructured definition of finite elements, complex geometries can be treated without additional effort. Finite elements are unstructured in nature, and the simple balancing method described above applies regardless of the grid being used.

- In Chapter 5, the variational multiscale method was extended to stabilize high-order spectral elements to solve the advection-diffusion equation. This technique was introduced as an alternative to spectral filtering to treat the Gibbs oscillations typical of high-order approximations. The main challenge that we faced lies in the construction of the VMS stabilization parameter (τ) for high-order elements whose internal nodes are not evenly spaced. In the specific case of spectral elements that adopt Legendre-Gauss-Lobatto (LGL) nodes, we extended to LGL elements a technique that was recently introduced for quadratic and cubic finite elements. Still today the construction of τ is an open field of study. The idea that we proposed in this chapter consists of using a known definition of τ defined for low-order elements, and generalize it for elements of arbitrary order.

In the regions distinguished by strong gradients, VMS is not sufficient to treat the localized under- and overshoots that characterize the solution. For better performance, we combined VMS with a discontinuity capturing method (DC) first, and later further enhanced their properties by the First-Order Subcells method (FOS) for particularly difficult problems (e.g. boundary and internal layers with very high-orders of approximation). FOS can be viewed as a selecting limiter

for spectral elements. The idea is simple and its coding on structured grids is straightforward. However, the drawback of the current implementation comes from the need of an externally tuned parameter. For robustness of the code, free parameters are not recommended. A proper construction of the error estimator used to locate the discontinuity should be considered for future work.

Numerically, we demonstrated that VMS, VMS+DC, and VMS+DC+FOS are a possible alternative to the standard filters used to stabilize spectral elements and suppress unwanted under- and over-shoots of the computed advected tracer. We evaluated the algorithms on a set of standard tests of increasing difficulty. A significant improvement was observed in the performance of the spectral element solver as far as the control of extrema is concerned, both in the purely advective and in the advective-diffusive regimes.

The most important features of variational multiscale stabilization for both the solution of the Euler equations of compressible flows (Chapter 4) or the scalar advection-diffusion problem (Chapter 5), can then be summarized as follows.

- Unlike hyper-viscosity, the subgrid-scale diffusion is localized and controlled.
 - Gibbs oscillations are greatly suppressed relative to traditional filters for high-order methods.
 - VMS does not depend on a free-parameter assigned by the user.
 - For best performance, FOS needs a proper error estimator (e.g., those typically used for dynamic grid refinement).
 - Currently, VMS with SEM for transport of tracers is not fully mass-conservative. This is an issue for long term simulations such as those in climate applications.
- Finally, in Chapter 6 the scheme was further extended to solve coupled problems for convective storm simulations. The Euler equations for stratified flows were modified to include the effects of water species on the dynamics of the flow and were coupled to a set of three advection equations that model transport of water vapor, cloud water, and rain. We used a Kessler-type microphysical scheme to predict phase change of water species and solve the multiphase problem of cloud simulations.

We tested the algorithm by simulating idealized mesoscale convective clouds with and without wind shear given a synthetic initial state for mid-latitude squall-line formation. The main features of the convective cloud such as spatial distribution, storm propagation, maximum variation of the vertical velocities, and rain accumulation were used as the metrics for comparison with the literature. We must point out the total lack of turbulence parametrization whose effects would have major impact on the solution.

Because the main goal of this chapter was the assessment of the algorithm in terms of stability properties in the context of wet dynamics (i.e., extension to wet environments of the method reported in Chapter 4), we proved that VMS stabilization

of finite elements is suitable for complex problems that involve transport phenomena of multi-phase flows in the framework of mesoscale meteorology. At the same time, in terms of accuracy in the solution of problems of atmospheric relevance (either idealized, or seemingly real simulations), the method as a whole performs sufficiently well to take this work further.

7.2 Future work

From this thesis a few directions for future research can be extracted.

- In relation to Chapters 4 and 6, this work represents the very first step to the development of an atmospheric research code for short-range atmospheric simulations (Numerical Weather Prediction). A great deal of work is necessary in fields that were not covered in this thesis. Turbulence, radiation, and planetary boundary layer are three examples. Turbulence is vital for the correct modeling of atmospheric dynamics and must be added to the code as one of the first future steps. Along with turbulence, the forces of Coriolis must be added as well to obtain meaningful results for long-lasting simulations in extended domains that could, eventually, cover the entire sphere on a global scale.

VMS is a suitable stabilization technique whose local nature and low diffusivity properties should be further explored for atmospheric applications. Orthogonal subgrid-scales (OSS) and subscale tracking are also recommended to improve the accuracy of long-lasting simulations and simulations using larger time-steps. VMS alone cannot preserve monotonicity of the solution. We have seen how this is particularly important to treat transport of tracers in the atmosphere. In the future, FEM+VMS should be analyzed with respect to monotonic solutions to verify mass conservation properties and evaluate possible improvements.

The use of fully explicit time integration represents the major drawback that is afflicting the code at the present stage. Acoustic modes are physically irrelevant in atmospheric simulations; however, they are a constraint on the size of the time-step. This limits the usability of the current implementation of the code to the simulation of relatively short-lasting phenomena on (relatively) small grids. The availability of very powerful machines is not sufficient to account for the fact that large problems can still be executed in a reasonable time (as it was indeed done in the testing phase of this work); the development of an efficient time integration algorithm is mandatory for the evolution of this work. This is an on-going work within the department where this thesis was partially developed. Based on the extensive work presented by other researchers on semi-implicit methods for atmospheric simulations, this direction is being considered.

- In Chapter 5 a possible way to achieve monotonic high-order spectral elements was shown. For conservation issues, however, the method thereby described is limited to short-term transport problems. To be applicable to climate problems, where

mass-conserving transport of hundreds of tracers for very long times is expected, work must be done, starting from the definition of proper fixers to the method for mass conservation. Finally, the algorithm should be extended for transport problems on the sphere. This is a problem of growing interest in the atmospheric community which has been developing future generation, multiscale, global circulation models for the past few years.

FOS should be improved by the addition of a proper error estimator to better locate the large gradients in the solution.

- Although the idea was reported only briefly in Appendix [A](#), grid generation is a field of increasing interest in high-resolution atmospheric simulation. We covered the main concepts of quasi-orthogonal, smooth grids to suggest new possibilities in grid generation for atmospheric modelers. The use of unstructured grids implies additional work in the construction of non-column-based parametrization schemes because vertically irregular grids are the limiting factor for many microphysical schemes of widespread use.

Appendix A

3D elliptic grid generation for domains with orography and bathymetry

In this appendix, we describe the implementation of an elliptic, quasi-orthogonal, three-dimensional mesh generator designed for simply-connected volumes with difficult topographies. We also discuss the idea of multi-block grid generation and how it can be applied to problems in geophysical fluid dynamics paying particular attention to NWP. The elliptic grid generator is built around the solution of the Thompson-Thames-Mastin (TTM) equations. Both techniques are dated and are well known by CFD practitioners, however, to our knowledge they are not yet of common use in NWP. There is a fundamental reason behind this fact: microphysical processes, radiation, and other parametrizations are designed and coded column-wise; this requires that the vertical nodes be aligned along the z-coordinate in some way. The use of elliptic (or any unstructured) grids along the vertical direction would need recoding of most existing parametrization if the horizontal grid resolution is much larger than the vertical. The constraint on the vertical positioning of consecutive nodes becomes less stringent as horizontal resolution is increased. As high horizontal resolution becomes more common, meshing techniques that can treat steep slopes in mountainous regions should be further explored for future applications. Such techniques are indeed starting to draw attention by atmospheric practitioners who are interested in the development of high-resolution models. One is described in what follows.

A.1 Introduction

Volume grid generation in atmospheric models is classically performed by a one directional simplification of *Transfinite interpolation* (TFI) (Gordon and Hall, 1973; Eriksson, 1982). TFI is robust, simple, and arguably the fastest grid generation technique in use in many fields of computational mechanics, of which geophysical fluid dynamics represents

a particular case. Nevertheless, generally the quality of TFI grids degenerates when the geometric features of the domain boundaries present sharp corners, quasi-vertical boundary walls, or similar characteristics. This has direct effect onto the quality of the numerical solution of the problem (Mastin and Thompson, 1978). The problem exists regardless of the underlying numerical method of solution. In NWP, sharp mountain ridges and canyons are an example. With the ever increasing trend towards high spatial resolution that we are experiencing in numerical weather prediction today, sharp topographies are certainly an issue. In the following sections, we describe the current way of generating structured grids in topographical domains and present a few examples to underline the possible limitations. At that point, we introduce the idea behind elliptic grid generation and how grids may be improved in terms of smoothness and orthogonality properties by this simple technique. Most of the ideas presented in this appendix are found in the books by Knupp and Steinberg (1993) and Thompson et al. (1985), and in the recent paper by Kaul (2010).

A.2 Algebraic grid generation

As we have mentioned above, transfinite interpolation has a major drawback that comes from the constraint on the regularity of the boundaries. If the boundaries of the simply-connected domain are not sufficiently smooth, TFI fails to generate good grids. The sharpness of internal corners given by a possible discontinuity in the space derivative of the boundary functions, reflects into folding grids with unacceptable node overlapping (see Fig. A.1). The problem of folding grids with difficult geometries is usually solved by subdividing the domain into smaller subdomains with more regular boundaries. This technique is robust but difficult to automatize. In Fig. A.2, although the edges do not cross, the vertical wall on the left-hand side of the hill is a challenge for the grid generator, as it can be noted by the extremely stretched elements in the region of the hill's front.

Nonetheless, because topography is usually smooth in current operational models (at horizontal resolution of 1 km or coarser, all mountain peaks are likely to be smoothed out), TFI is yet the perfect and quick solution that can be properly modified for different types of vertical node distributions.

A.2.1 Classical vertical discretization

The domain of a limited area model consists of a regular, rectangular volume whose vertical boundary walls are flat, regular planes. A coordinate transformation is then usually performed in the vertical direction only, while a uniform subdivision is used along the horizontal plane. The most simple discretization is nothing more than the σ -grid (Gal-Chen and Somerville, 1975) mentioned in Section 4.5, and given by the following transformation along z :

$$\sigma(x, z) = H \frac{z - h(x)}{H - h(x)} \quad (\text{A.1})$$

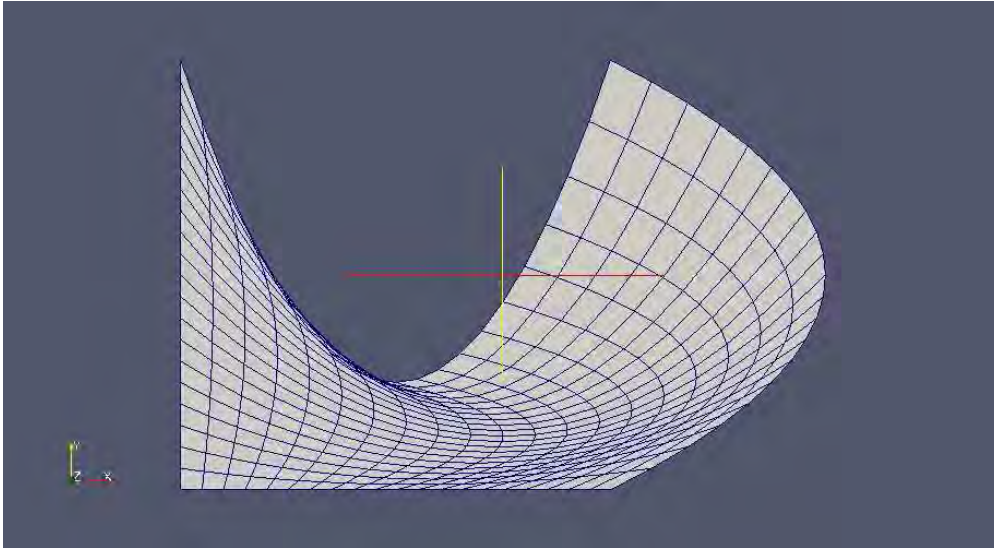


Figure A.1: Detail of a TFI grid with overlapping nodes near the lowest peak of the top boundary.

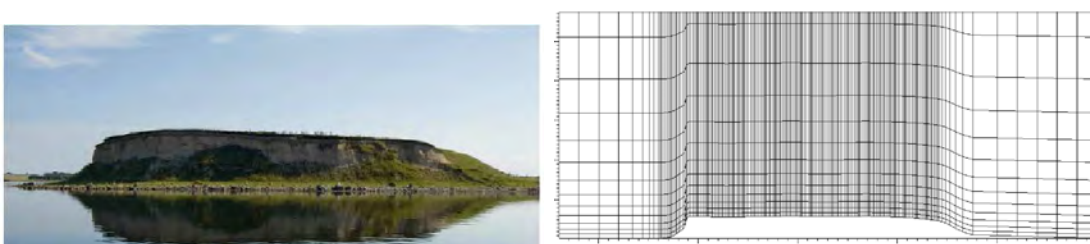


Figure A.2: Vertical slice of a TFI volume grid of the Bolund hill in Roskilde Fjord, Denmark. Source of the photograph: <http://windenergyresearch.org/2010/09/the-bolund-project/>.

that inverted gives yields the backward mapping from the sigma vertical coordinates into the original x-y space as:

$$z(x, \sigma) = h(x) + \sigma(x, z) \frac{H - h(x)}{H}. \quad (\text{A.2})$$

It is simple, but on steep topography the regularity of the grid is compromised. Because these grids are based on a map from logical, rectangular space to physical space, mathematically speaking, the irregularity comes from the value of the Jacobian of the transformation that approaches zero. To overcome the drawbacks of the σ transformation over steep slopes, Schar et al. (2002) introduced the *smooth level vertical* (SLEVE) mapping that helps maintain a sufficient degree of regularity of the node distribution in the inner domain. Given a mountain ridge, a SLEVE grid is obtained from the decomposition of a large and small scale variation of topography (e.g. a Gaussian terrain perturbed by a wave-like function). Through this solution the grid distortion is controlled from bottom to top by means of two parameters (s_i in equation A.3). The mapping is done through two successive steps: first, the topography $h(x)$ in a domain of height H is split into its large and small scale functions $h_{small}(x)$ and $h_{large}(x)$, and second, h_{small} and h_{large} are combined with the decay factors

$$b_i(\zeta) = \frac{\sinh[H - \zeta/s_i]}{\sinh[H/s_i]}, \quad (\text{A.3})$$

to give the inverse mapping

$$z(x, \zeta) = \zeta + h_{small}(x, z)b_1(\zeta) + h_{large}(x, z)b_2(\zeta). \quad (\text{A.4})$$

Here ζ defines the vertical coordinate in the same way of σ in equation (A.1) –we called it differently to avoid confusion about the two transformations. The combination of s_1 and s_2 drives the invertibility condition of the Jacobian (see Schar et al. (2002) for details).

Somewhere between the two methods described above, stands the grid of Simmons and Burridge (1981). Their map uses the same vertical coordinate σ and combines the topography $h(x)$ and the height of the domain through two functions $a(\sigma)$ and $b(\sigma)$ whose values are properly tabled and such that the the mapping is given by:

$$z(\sigma) = Ha(\sigma) + h(x)b(\sigma). \quad (\text{A.5})$$

The modified version of A.5 is the analogue of SLEVE with one control parameter s only. Its inverse mapping is:

$$z(x, \sigma) = \sigma + h(x) \frac{\sinh[(H - \sigma)/s]}{\sinh[H/s]}. \quad (\text{A.6})$$

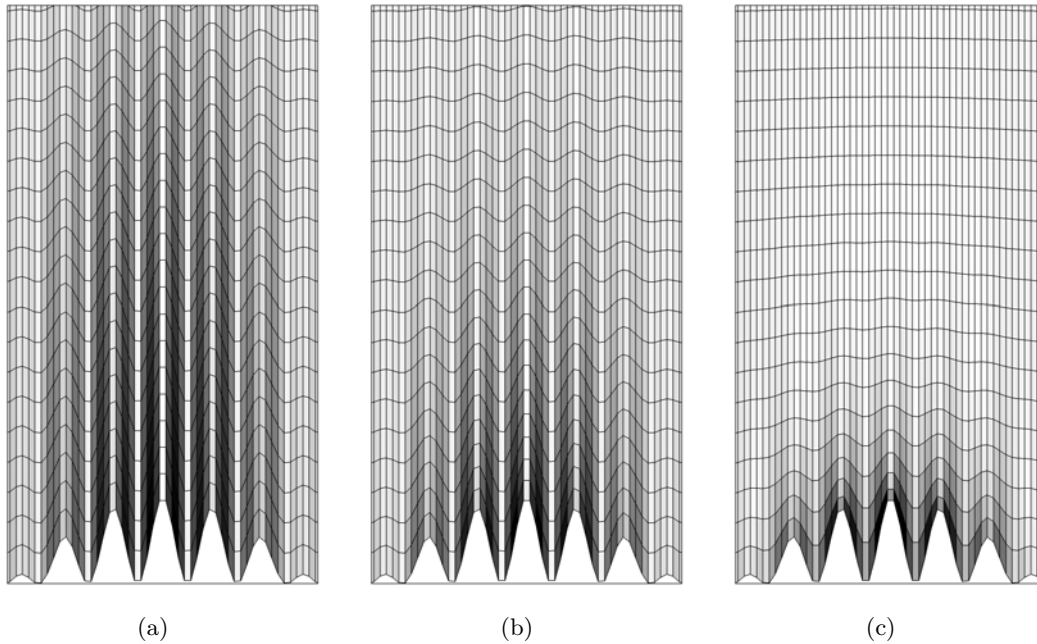


Figure A.3: (a) σ , (b) hybrid, (c) SLEVE. Shading indicates the degree of grid deformation.

The resulting vertical grids are plotted in Fig. A.3. The degree of deformation of the grid is represented in grey scale, where darker shading indicates that skewness is larger.

These improved methods are sufficiently good as long as the boundaries are never vertical. This is because the transformations are performed along z only. For full control of the nodes' distribution in all directions, these schemes should be incorporated into a full TFI interpolation.

A.3 Elliptic grid generation

One simple, yet efficient solution to the generation of smooth grids with sufficiently good properties is given by the solution of the Thompson-Thames-Mastin (TTM) problem (Thompson et al., 1974), an elliptic system of partial differential equations. The penalty of elliptic methods is the higher cost with respect to algebraic methods. The method that we apply is the inhomogeneous TTM to construct elliptic grids with the control of the node positioning in the grid's interior (e.g. boundary layers, regions of large gradients in the middle of the domain).

Three-dimensional grid generation by elliptic methods is described by the system of PDE's :

$$\xi_{xx} + \xi_{yy} + \xi_{zz} = P(\xi, \eta, \zeta) = -a_i \cdot \text{sgn}(\xi - \xi_i) \exp\{-b_i |\xi - \xi_i|\}, \quad (\text{A.7a})$$

$$\eta_{xx} + \eta_{yy} + \eta_{zz} = Q(\xi, \eta, \zeta) = -c_i \cdot \text{sgn}(\eta - \eta_i) \exp\{-d_i |\eta - \eta_i|\}, \quad (\text{A.7b})$$

$$\zeta_{xx} + \zeta_{yy} + \zeta_{zz} = R(\xi, \eta, \zeta) = -e_i \cdot \text{sgn}(\zeta - \zeta_i) \exp\{-f_i |\zeta - \zeta_i|\}, \quad (\text{A.7c})$$

where the triads (ξ, η, ζ) and (x, y, z) are generalized curvilinear and Cartesian coordinates, respectively. P , Q , and R are functions that control the grid points within the volume. They are controlled by the user-defined constants a_i, b_i, c_i, d_i, e_i and f_i . i indicates one particular boundary component associated with the specific problem. I.e., to have node clustering close to the bottom surface, $i = 3$ and a_i, b_i, c_i, d_i, e_i are given specific definitions. Since [Kaul \(2010\)](#), the constants a_i, b_i, c_i, d_i, e_i can be no longer user defined but rather, be part of the solution. For the sake of simplicity, in this appendix the original TTM is described, but with ease of implementation and subsequent improvement in the software usage, Kaul's automatic technique should be considered in the future.

Given the Jacobian $J = J(\mathbf{x}/\boldsymbol{\xi})$ of the transformation, the grid in physical space is obtained by solving the inverse problem of [\(A.7\)](#) given by:

$$\alpha_{11}x_{\xi\xi} + \alpha_{22}x_{\eta\eta} + \alpha_{33}x_{\zeta\zeta} + 2(\alpha_{12}x_{\xi\eta} + \alpha_{13}x_{\xi\zeta} + \alpha_{23}x_{\eta\zeta}) = -J^2(p_ix_{\xi} + q_ix_{\eta} + r_ix_{\zeta}) \quad (\text{A.8a})$$

$$\alpha_{11}y_{\xi\xi} + \alpha_{22}y_{\eta\eta} + \alpha_{33}y_{\zeta\zeta} + 2(\alpha_{12}y_{\xi\eta} + \alpha_{13}y_{\xi\zeta} + \alpha_{23}y_{\eta\zeta}) = -J^2(p_iy_{\xi} + q_iy_{\eta} + r_iy_{\zeta}) \quad (\text{A.8b})$$

$$\alpha_{11}z_{\xi\xi} + \alpha_{22}z_{\eta\eta} + \alpha_{33}z_{\zeta\zeta} + 2(\alpha_{12}z_{\xi\eta} + \alpha_{13}z_{\xi\zeta} + \alpha_{23}z_{\eta\zeta}) = -J^2(p_iz_{\xi} + q_iz_{\eta} + r_iz_{\zeta}) \quad (\text{A.8c})$$

where the subscript i varies according to the surface where the clustering is needed and the coefficients α are given by:

$$\begin{aligned} \alpha_{11} &= J^2(\xi_x^2 + \xi_y^2 + \xi_z^2), \\ \alpha_{22} &= J^2(\eta_x^2 + \eta_y^2 + \eta_z^2), \\ \alpha_{33} &= J^2(\zeta_x^2 + \zeta_y^2 + \zeta_z^2), \\ \alpha_{12} &= J^2(\xi_x\eta_x + \xi_y\eta_y + \xi_z\eta_z), \\ \alpha_{13} &= J^2(\xi_x\zeta_x + \xi_y\zeta_y + \xi_z\zeta_z), \\ \alpha_{23} &= J^2(\eta_x\zeta_x + \eta_y\zeta_y + \eta_z\zeta_z). \end{aligned}$$

In the case of flows over topography, the only grid clustering that is necessary to maintain is in the proximity of the ground. This is obtained when P and Q in equations [\(A.7\)](#) are set to zero, while R is given the expression:

$$R(\xi, \eta, \zeta) = -Q_0 \text{sgn}(\zeta - \zeta_0) \exp\{-\lambda |\zeta - \zeta_0|\}. \quad (\text{A.9})$$

Q_0 and λ are user-defined coefficients passed in input, and $\zeta_0 \equiv 0$ indicates that clustering is built around the ground surface only.

System (A.8) is solved by Gauss-Seidel iterations starting from a background TFI grid. For demanding problems a maximum of 100 iterative loops was necessary, although an average of 25 is usually sufficient in simple domains. The boundary conditions are simply given by the (x,y,z) coordinates of the domain boundaries. To add degrees of freedom to the boundary nodes and relax their distribution (if needed), Neumann boundary conditions should be considered.

Remark A.1. Quasi-orthogonality: Orthogonality in three-dimensional structured grid generation systems is still an open field of work (see [Thompson et al. \(1985\)](#); [Kaul \(2010\)](#)). The elliptic grid generation system herein described and implemented in [MMesh3D \(2010\)](#) is able to reach reasonable orthogonality properties at the lower boundary by either using Neumann boundary conditions to move the nodes, or by proper definition of the control functions as done in [Kaul \(2010\)](#). Currently, a quasi-orthogonal system is the best that we can achieve with the available algorithms from the literature. Fig. A.5 shows how a non-orthogonal grid (top panel of the figure) is transformed to a quasi-orthogonal mesh in the proximity of the boundary. This grid was deliberately relaxed to the point where the boundary layer is completely lost. It was meant this way to clearly show orthogonality at the boundary. However, maintaining a proper stretching ratio in the proximity of the boundary without affecting orthogonality is a problem that is not fully solved yet. A compromise is needed in building the grid, and experimentation on different topographies may be necessary. An example of a smooth grid with control on the boundary layer is shown in Fig. A.7.

A.4 Multiblock grids

Multiblock grids are a standard approach to structured grid generation of complex geometries in CFD. It consists in subdividing the global domain into smaller, simply connected regions, and mesh them independently as long as their boundary nodes match. In geophysical fluid dynamics, multi-blocks can be mostly found in ocean models, especially so when structured grids are used to mesh domains of complex horizontal definition (e.g. the Mediterranean basin as in [Ly and Luong \(1999\)](#)). The flexibility in the geometric features of the grid generated within every block is what makes this idea attractive with respect to previous existing systems. Recently, a similar approach using quadtrees was introduced by [Yamazaki and Satomura \(2012\)](#) for NWP. The flexibility in geometric features of the grid generated within every block is what makes the idea attractive. It is also a way of keeping most of the grid rectangular and Cartesian wherever possible, and even use fully unstructured grids in regions of complex geometry only (better resolution of topography). An example is reported in Fig. A.4.

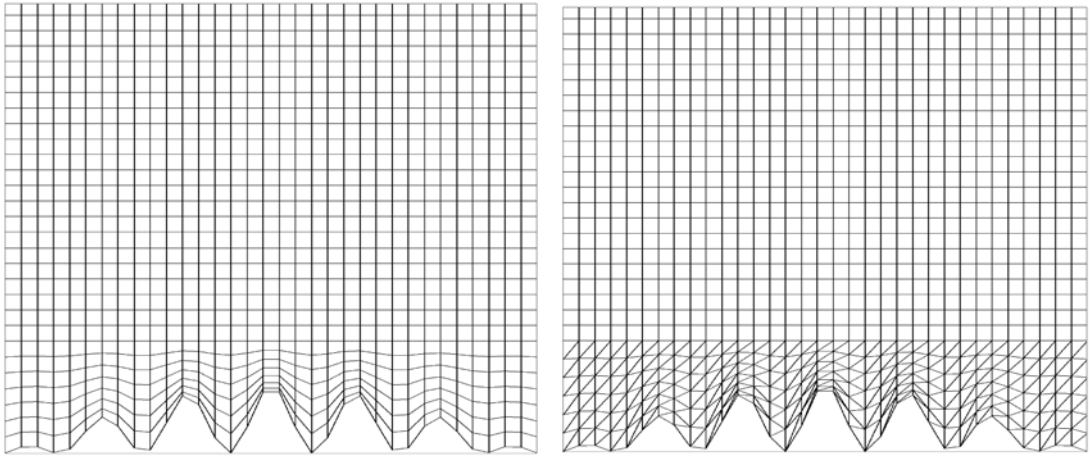


Figure A.4: Two-block grid: example of regular rectangular grid on the top part of the domain, coupled to a SLEVE grid in the mountain region.

A.5 Examples

We report a few two- and three-dimensional examples of grids generated using both algebraic and elliptic methods. Fig. A.5 shows the computational grid around a cosine functions obtained by TFI interpolation, TFI with orthogonal multi-surface method, and by the elliptic grid generator. The geometry is pretty straightforward to mesh. The three methods give similar results; however, using the elliptic method together with a multi-surface technique to achieve orthogonality seems to give an interesting grid for problems that involve the solution of boundary layers. The problem is taken a little further with the fully three-dimensional mesh of the Bolund region. The improvement in terms of regularity of the grid in the internal volume and in terms of quasi-orthogonality, is evident from panel (b) in Fig. A.6, where the elliptic solver was applied with 50 iterations to improve the algebraic grid of panel (a). The control of the grid points along the vertical direction to include a boundary layer grid is visible in Fig. A.7. Fig. A.8 shows the solution to the problem of overlapping nodes. Finally, the surface grids of two real domains are represented in Fig. A.9.

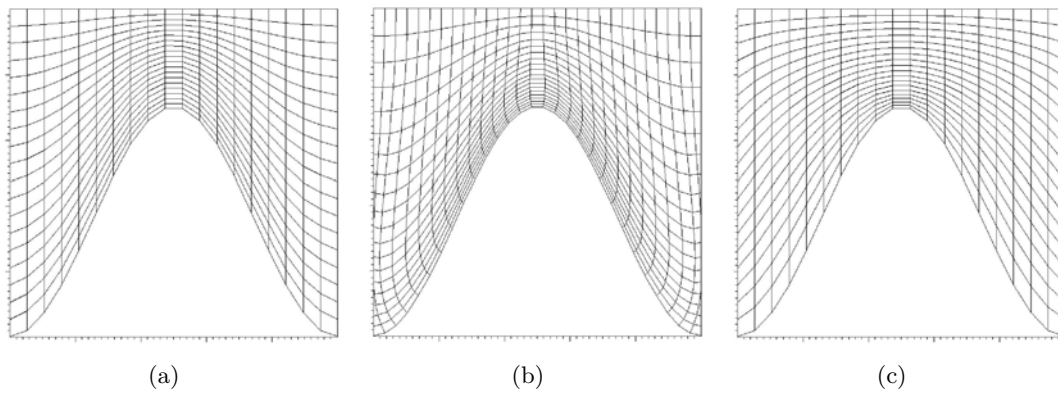
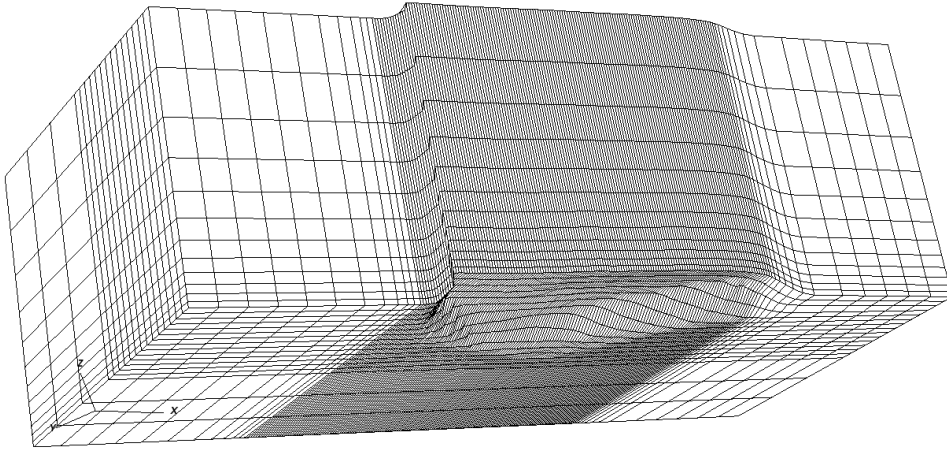
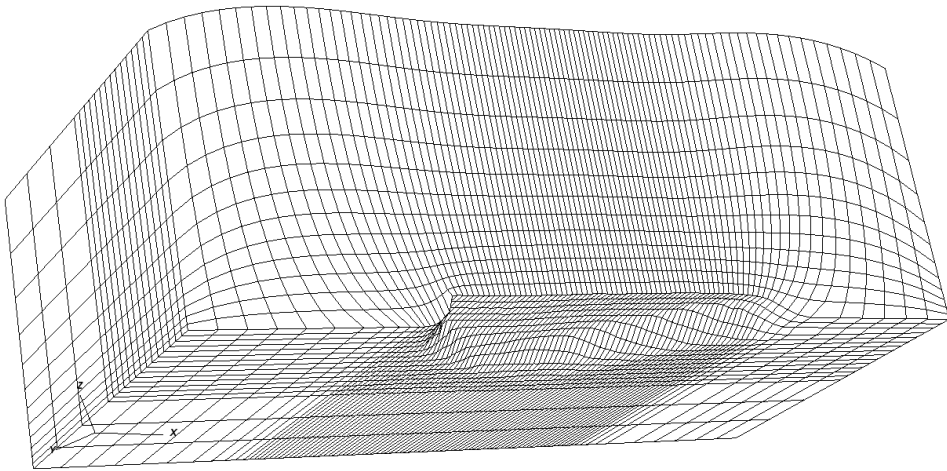


Figure A.5: (a) TFI, (b) orthogonal, (c) elliptic.



(a)



(b)

Figure A.6: (a) TFI and (b) elliptic volume grids. In this plot there is no grid control in the proximity of the boundary surface. Elliptic computed with 50 iterations.

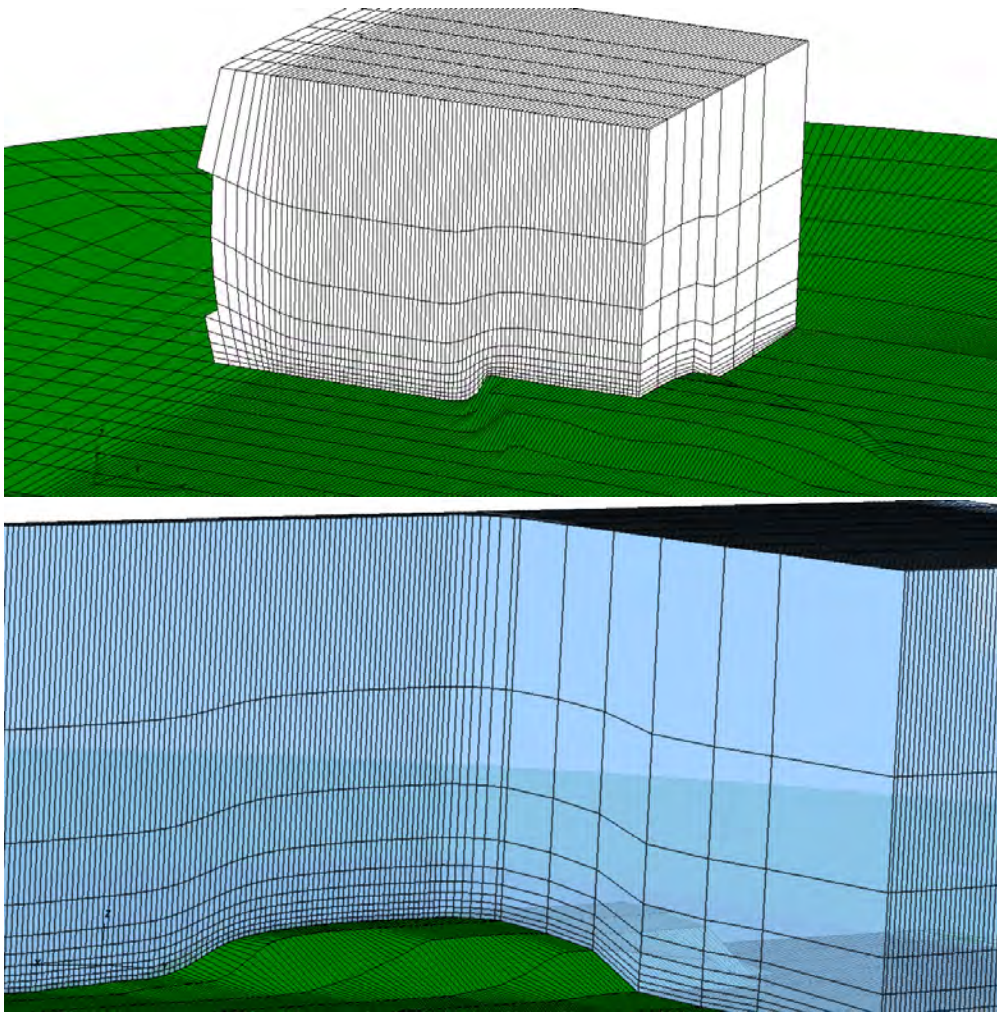
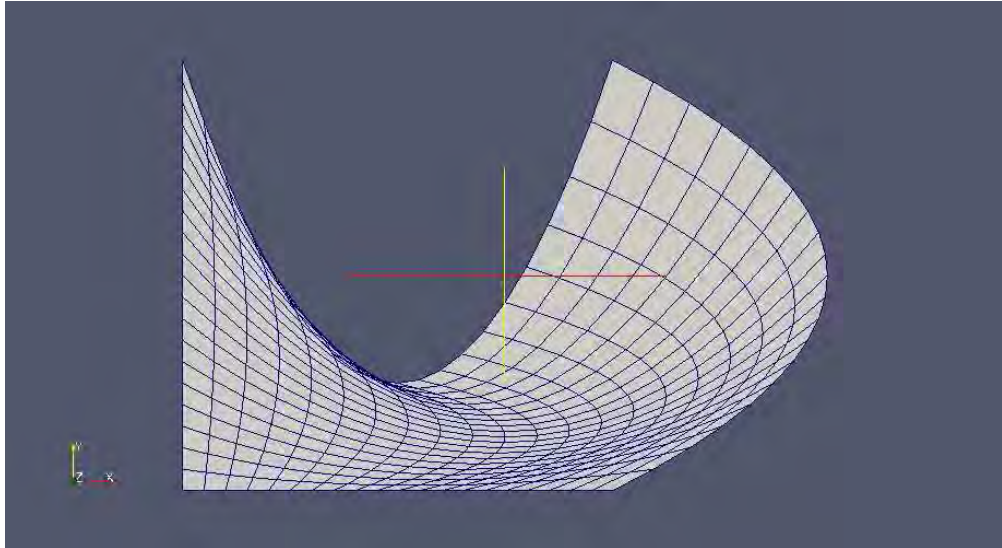
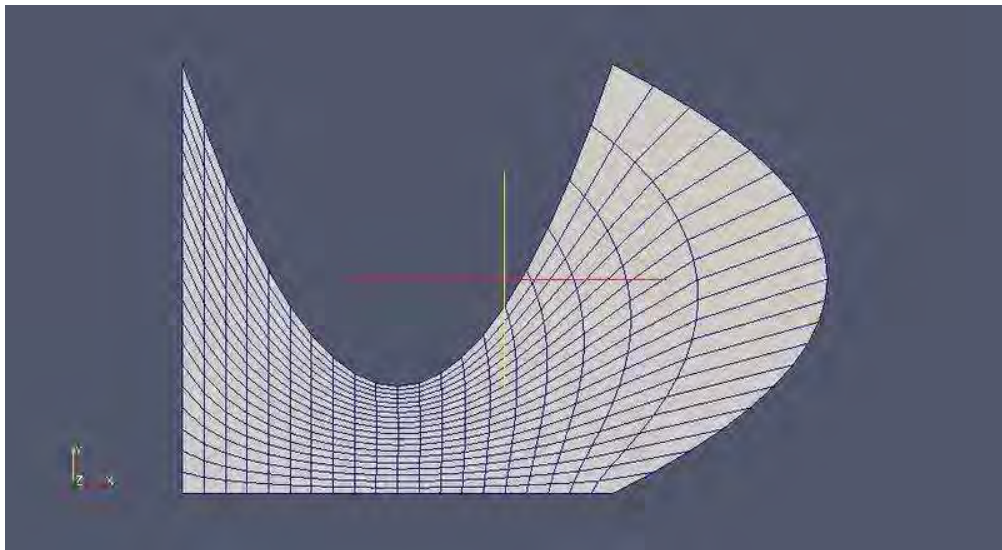


Figure A.7: Two views of the same elliptic volume grid. After 10 iterations.

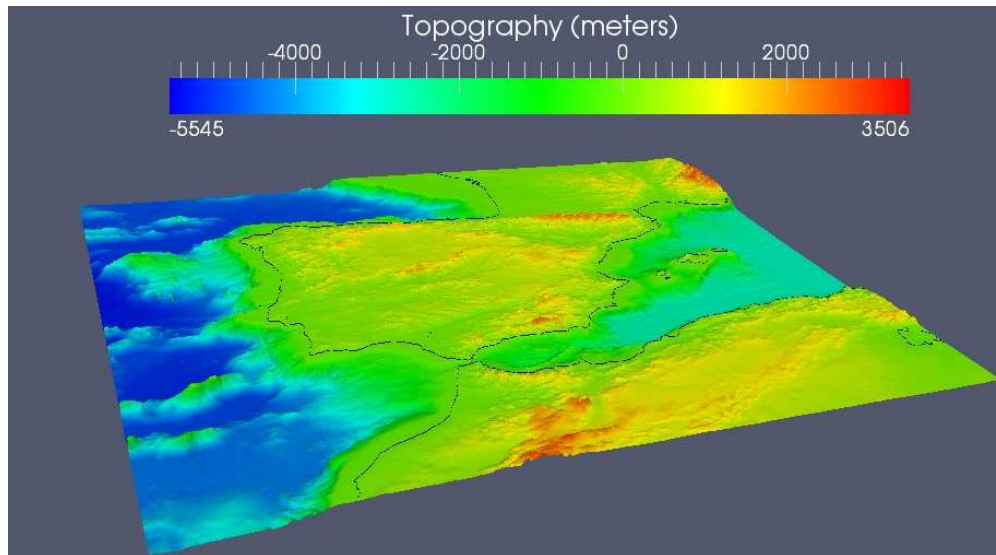


(a)

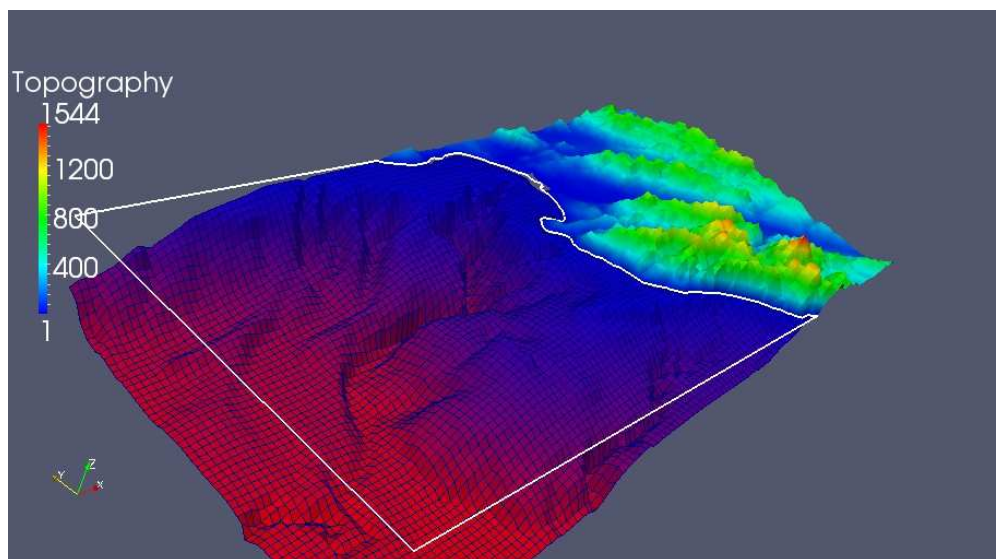


(b)

Figure A.8: As in Fig. A.1, but compared with the result after applying the elliptic solver (b).



(a)



(b)

Figure A.9: Surface grids of the topography of (a) the Iberian Peninsula and (b) the region of the Monterey Bay, California (b).

Appendix B

Computational environment

B.1 Alya

The algorithm described in Chapter 4 is implemented within a larger environment designed for heterogeneous problems in computational mechanics. The framework is called *Alya*. It is fully developed at Barcelona Supercomputing Center within the department of Computer Applications for Science and Engineering (CASE). This section describes its main features.

Alya is a multi-physics, three-dimensional modular code for high-performance computational mechanics. It was conceived as a numerical solver of linear and nonlinear PDE's of mathematical physics by means of variational methods (FEM) for discretization in space, and finite differencing for time integration. The efficient solution of large problems on massively parallel computers was the driving justification behind its development. As such, it was coded from scratch with parallelization and high performance computing (HPC) in mind to attain high scalability standards. It was benchmarked on different architectures such as Intel Nehalem, Sandy Bridge, and IBM PPC. The largest benchmark executed so far has been an implicit incompressible flow run using 16000 cores on JUGENE (Juelicher BlueGene/P) reaching 90% of linear scalability.

The software is organized into three main blocks: the **Kernel**, **Modules**, and **Services**, each of which fulfills very specific tasks. To describe the three branches, we shuffle the order a bit. We start by describing the function and structure of the modules because it will later help understand how the kernel and services work. Every physical problem is coded within its own specific module that is contained in **Modules**. For instance, the solver of the Euler equations is coded in a module (**nastal**) that has no direct link with the solver for incompressible viscous flows coded in **nastin**. The same applies to any module that corresponds to some specific physical problem. However, as much as each module is independent, the **Kernel** controls their possible coupling by exchanging a set of global variables common to everyone. The Kernel then, is responsible for the control of the workflow of the code by management of the interconnections between the modules and the services. Algorithmically speaking, the most important tasks taken care of by

the Kernel can be summarized as follows:

- Reading of computational grid and organize the information for domain splitting done by the `Parall` service.
- Construction of the finite element tools to be used by each module: e.g., basis functions.
- All (or most) linear algebra operations: e.g. solver of the linear system that results from the FE approximation

Finally, the services control very specific tools that are common to all but that are not necessary for the solution of the physical problem. Parallelization is an example. The importance of coupling in Alya is clear from the moist simulations described in Chapter 6, where `nastal` is coupled to `chemic`, the module that solves the transport equation of moisture in the atmosphere. The following recent publications are based on Alya: Houzeaux et al. (2009); Houzeaux et al. (2011); Marras et al. (2012c,b).

B.2 NUMA

NUMA, the *Nonhydrostatic Unified Model of the Atmosphere*, is developed by the group of Francis X. Giraldo at the department of Applied Mathematics of the Naval Postgraduate School in Monterey (U.S.A.). It is a three-dimensional massively parallel model based on the solution of the compressible Navier-Stokes equations for simulating non-hydrostatic and hydrostatic atmospheric processes. It is a unified model in the sense of (1) its numerics by allowing the user to choose either continuous or discontinuous high-order Galerkin methods; and (2), the model can be used as either a mesoscale (limited-area) or global model. The numerics of NUMA is flexible and the code is modular: the user/developer has the freedom to add new basis functions, grids, time-integrators, and data structures for his own needs. Parallelization is achieved by automatic domain partitioning using METIS (G. Karypis and Kuman, 2008) with Message Passing Interface (MPI) for parallel communication.

The Navy's hydrostatic global Spectral Element Atmospheric Model (NSEAM) is the precursor of NUMA. NSEAM is a complete atmospheric model with all the necessary sub-grid scale parametrizations and real orography.

The numerical methods and capabilities of the model are described in a set of research articles of which the most recent are Kelly and Giraldo (2012); Giraldo et al. (2012); Gaberšek et al. (2012); Müller et al. (2011); Marras et al. (2012a). See also http://faculty.nps.edu/fxgirald/projects/NUMA/Introduction_to_NUMA.html for more.

Appendix C

θ and T equations

Let us take Eq. (2.5c), express it in terms of $(\rho, \mathbf{u}, \theta)$, expand the derivatives, and write:

$$\theta \left(\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{u}) \right) + \rho \frac{\partial \theta}{\partial t} + \rho \mathbf{u} \cdot \nabla \theta = 0$$

Use of the continuity equation and the definition of total derivative D/Dt yield the equation

$$\rho \frac{D\theta}{Dt} = 0. \quad (\text{C.1})$$

By definition, $\theta = T (p_0/p)^{\frac{\gamma-1}{\gamma}}$. If we assign the identities $A = p_0^{\frac{\gamma-1}{\gamma}}$ and $\alpha = \frac{1-\gamma}{\gamma}$, then $\theta = ATp^\alpha$. We obtain:

$$A\rho T \frac{\partial p^\alpha}{\partial t} + A\rho p^\alpha \frac{\partial T}{\partial t} + A\rho T \mathbf{u} \cdot \nabla p^\alpha + A\rho p^\alpha \mathbf{u} \cdot \nabla T = 0. \quad (\text{C.2})$$

After reordering and simplifying by $A\rho$, equation (C.2) takes the form

$$\alpha T p^{\alpha-1} \frac{Dp}{Dt} + p^\alpha \frac{DT}{Dt} = 0.$$

Some final algebra and the temperature equation can be given the (T, p) formulation as follows:

$$\begin{aligned} \frac{1-\gamma}{\gamma} \frac{1}{p} \frac{Dp}{Dt} + \frac{1}{T} \frac{DT}{Dt} &= 0, \\ -\frac{1}{\rho c_p} \frac{Dp}{Dt} + \frac{DT}{Dt} &= 0. \end{aligned} \quad (\text{C.3})$$

Substitution of the state equation $p = \rho RT$ and developing the material derivative yields

$$\frac{DT}{Dt} - \frac{1}{\rho c_p} R \rho \frac{DT}{Dt} - \frac{1}{\rho c_p} RT \frac{D\rho}{Dt} = 0; \quad (\text{C.4})$$

Using now the equivalence $1 - R/c_p = c_v/c_p$ and the continuity equation to remove the density time-derivative, we get back to the first law of thermodynamics expressed by (2.8), which is thus equivalent to (C.1).

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