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	Particle	
	Given Name	<b>Oleg N.</b>
	Prefix	
	Suffix	
	Role	
	Division	
	Organization	Northumbria University
	Address	NE1 8ST, Newcastle upon Tyne, United Kingdom
	Email	oleg.kirillov@northumbria.ac.uk
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# Classical Results and Modern Approaches to Nonconservative Stability



Oleg N. Kirillov

**Abstract** Stability of nonconservative systems is nontrivial already on the linear level, especially, if the system depends on multiple parameters. We present an overview of results and methods of stability theory that are specific for nonconservative applications. Special attention is given to the topics of flutter and divergence, reversible- and Hamiltonian-Hopf bifurcation, Krein signature, modes and waves of positive and negative energy, dissipation-induced instabilities, destabilization paradox, influence of structure of forces on stability and stability optimization.

## 1 Introduction

### 1.1 “It was Greenhill who Started the Trouble...”

...though he never knew it,” remarked Gladwell (1990) in his historical account of the genesis of the field of nonconservative stability. As many of his scientific contemporaries, Greenhill successfully combined his interest to pure mathematical subjects, such as elliptic functions, with contributions to applied problems of ballistics (Greenhill 1879), hydrodynamics (Greenhill 1880), and elasticity (Greenhill 1881) coming from the flourishing industries of the British Empire. In particular, motivated by the problem of buckling of propeller-shafts of steamers he analyzed in Greenhill (1883) stability of an elastic shaft of a *circular* cross-section, length  $L$ , and mass per unit length  $m$  under the action of a compressive force,  $P$ , and an axial torque,  $M$ . Figure 1 taken from Gladwell (1990) illustrates five possible in this system boundary conditions:

- I. Symmetric clamped-clamped shaft
- II. Asymmetric clamped-clamped shaft
- III. Clamped-free shaft

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O. N. Kirillov (✉)  
Northumbria University, NE1 8ST, Newcastle upon Tyne, United Kingdom  
e-mail: oleg.kirillov@northumbria.ac.uk

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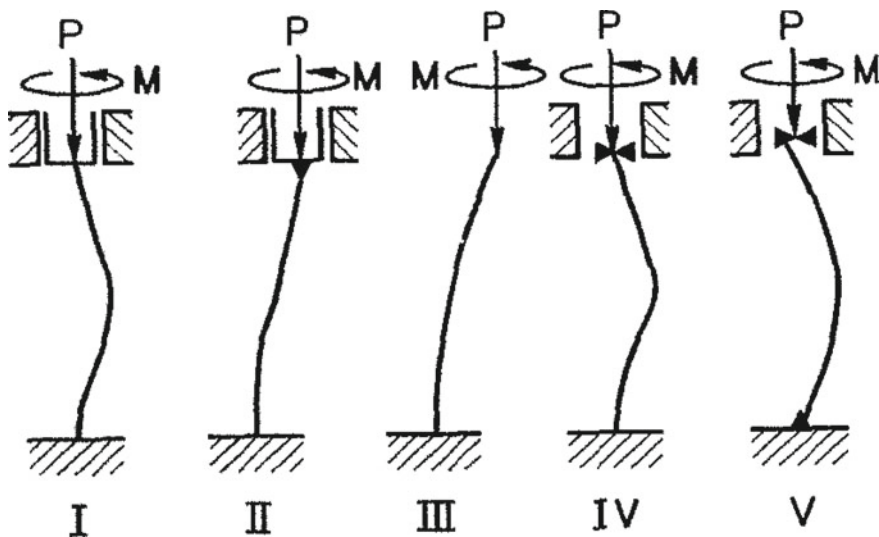


Fig. 1 Five realizations of Greenhill's elastic shaft loaded by a compressing force,  $P$ , and an axial torque,  $M$ , corresponding to five different boundary conditions (from Gladwell 1990)

IV. Clamped-hinged shaft

V. Hinged-hinged shaft

In the absence of the axial torque ( $M = 0$ ), the Greenhill problem reduces to the famous Euler's buckling under compression of 1757. The critical load at the onset of the static instability can be found by the equilibrium method, which seeks values of the axial force, for which there are nontrivial equilibrium configurations. This yields the Euler formula for the critical buckling force

$$P_{cr} = k \frac{\pi^2 EI}{L^2}, \quad \text{where} \quad \frac{\text{BC}}{k} \begin{array}{c|c|c|c|c} \text{I} & \text{II} & \text{III} & \text{IV} & \text{V} \\ \hline 4 & 1 & 1/4 & 2.046 & 1 \end{array}, \quad (1)$$

$E$  is the Young modulus and  $I$  is the moment of inertia of a (circular) cross-section of the shaft.

In contrast to the Euler buckling case, Greenhill set  $P = 0$  and tried to find the critical torque that causes buckling of the shaft. Using the equilibrium method, he managed to find the critical torque for the boundary conditions I, II, and V (Greenhill 1883; Ziegler 1953a, b; Gladwell 1990)

$$M_{cr} = k \frac{\pi EI}{L}, \quad \text{where} \quad \frac{\text{BC}}{k} \begin{array}{c|c|c|c|c} \text{I} & \text{II} & \text{III} & \text{IV} & \text{V} \\ \hline 2.861 & 2 & ? & ? & 2 \end{array}. \quad (2)$$

The cases III and IV have not been analyzed by Greenhill and remained untreated until Nicolai (1928) reconsidered a variant of the case IV, in which the axial torque

41 is replaced with the *follower* torque,  $M$ , such that the vector of the torque is directed  
 42 along the tangent to the deformed axis of the shaft at the end point (Gladwell 1990).

43 Nicolai (1928) had established that no nontrivial equilibrium configuration of the  
 44 shaft exists different from the rectilinear one, meaning stability for all magnitudes,  
 45  $M$ , of the follower torque and thus  $k = \infty$  in (2). Being unsatisfied with this overop-  
 46 timistic result, Nicolai realized that the equilibrium method does not work properly  
 47 in the case of the follower torque. He decided to study small oscillations of the shaft  
 48 about its rectilinear configuration using what is now known as the Lyapunov stability  
 49 theory (Lyapunov 1992) that, in particular, can predict instability via eigenvalues of  
 50 the linearized problem.

51 Surprisingly, it turned out that there exist eigenvalues with positive real parts  
 52 (instability) for all magnitudes of the torque, meaning that the critical value of the  
 53 follower torque for an elastic shaft of a circular cross-section is actually  $M_{cr} = 0$ , i.e.  
 54  $k = 0$  in (2). Because of its unusual behavior, this instability phenomenon received  
 55 a name “Nicolai’s paradox” (Nicolai 1928; Gladwell 1990).

56 In 1951-56 Ziegler re-considered the five original Greenhill problems with the  
 57 Lyapunov approach and found that at  $P = 0$  the shaft is unstable in cases III and IV  
 58 for all values of the axial torque  $M$ , just as in Nicolai’s problem with the follower  
 59 torque (Ziegler 1951a, b, 1953a, b, 1956).

60 
$$M_{cr} = k \frac{\pi EI}{L}, \quad \text{where} \quad \frac{BC}{k} \begin{array}{c|c|c|c|c} I & II & III & IV & V \\ \hline 2.861 & 2 & 0 & 0 & 2 \end{array} \quad (3)$$

61 Moreover, Ziegler realized that “Stability problem for a shaft loaded by an axial  
 62 torque  $M$ , is generally *non-conservative*, as in the cases III, IV, and V, where the  
 63 end slope is unconstrained. Only in exceptional cases the work of such torques in a  
 64 virtual deformation can be represented as a variation of an integral” and the problem  
 65 is conservative, as in cases I and II, where the equilibrium method gives the correct  
 66 critical torque. “In any case”, concluded Ziegler, “the results show that even very  
 67 simple models are *not conservative* and, if they occur as stability problems, they  
 68 should be treated dynamically”, i.e. with the use of the Lyapunov approach (Ziegler  
 69 1951a, b).

70 Note that already Nicolai (1929) realized that the cases III and IV do not represent  
 71 generic situations because it is possible to modify the end conditions, or consider a  
 72 shaft with *unequal* stiffness (non-circular cross-section) yielding a nonzero critical  
 73 torque (Bolotin 1963; Gladwell 1990). These conclusions were later confirmed by  
 74 Ziegler (1956) and developed further in the recent works on the Nicolai paradox by  
 75 Seyranian and Mailybaev (2011) and Luongo et al. (2016).

76 **1.2 Greenhill’s Shaft as a Non-self-adjoint Problem**

77 Small vibrations of the Greenhill’s shaft near its non-deformed rectilinear configu-  
 78 ration are described by the following partial differential equation (Bolotin 1963)

Editor Proof

$$79 \quad \mathbf{l}_0 \partial_z^4 \mathbf{w} + \mathbf{l}_1 \partial_z^3 \mathbf{w} + \mathbf{l}_2 \partial_z^2 \mathbf{w} + m \partial_z \mathbf{w} = 0, \quad z \in [0, L], \quad \mathbf{w} = \begin{pmatrix} w_1 \\ w_2 \end{pmatrix} \quad (4)$$

80 where the matrices  $\mathbf{l}_0$ ,  $\mathbf{l}_1$ , and  $\mathbf{l}_2$  are

$$81 \quad \mathbf{l}_0 = \begin{pmatrix} EI & 0 \\ 0 & EI \end{pmatrix}, \quad \mathbf{l}_1 = \begin{pmatrix} 0 & M \\ -M & 0 \end{pmatrix}, \quad \mathbf{l}_2 = \begin{pmatrix} P & 0 \\ 0 & P \end{pmatrix} \quad (5)$$

82 The nonconservative clamped-free case (III) is characterized by the following  
83 boundary conditions

$$84 \quad \mathbf{w}(0) = \mathbf{w}'(0) = 0, \\ 85 \quad \mathbf{l}_0 \mathbf{w}''(L) + \mathbf{l}_1 \mathbf{w}'(L) = \mathbf{0}, \\ 86 \quad \mathbf{l}_0 \mathbf{w}'''(L) + \mathbf{l}_1 \mathbf{w}''(L) + \mathbf{l}_2 \mathbf{w}'(L) = \mathbf{0}, \quad (6)$$

87 corresponding to the constrained deflection and slope at the clamped end ( $z = 0$ ) and  
88 vanishing axial force and axial torque at the free end ( $z = L$ ).

Separating time with  $\mathbf{w} = \mathbf{u}e^{\lambda t}$ , and introducing the matrix

$$\mathbf{l}_4(\lambda) = \lambda^2 \begin{pmatrix} m & 0 \\ 0 & m \end{pmatrix},$$

89 we come to the boundary eigenvalue problem

$$90 \quad \mathbf{L}(\lambda)\mathbf{u} = \mathbf{l}_0 \partial_z^4 \mathbf{u} + \mathbf{l}_1 \partial_z^3 \mathbf{u} + \mathbf{l}_2 \partial_z^2 \mathbf{u} + \mathbf{l}_4(\lambda)\mathbf{u} = \mathbf{0} \quad (7)$$

91 with the boundary conditions

$$92 \quad \mathbf{u}(0) = \mathbf{u}'(0) = \mathbf{0}, \\ 93 \quad \mathbf{l}_0 \mathbf{u}''(L) + \mathbf{l}_1 \mathbf{u}'(L) = \mathbf{0}, \\ 94 \quad \mathbf{l}_0 \mathbf{u}'''(L) + \mathbf{l}_1 \mathbf{u}''(L) + \mathbf{l}_2 \mathbf{u}'(L) = \mathbf{0}, \quad (8)$$

95 where prime denotes partial differentiation with respect to  $z$ . The equilibrium state  
96 is unstable if there is a value of  $\lambda$  with positive real part.

Integrating by parts the inner product

$$(\mathbf{L}\mathbf{u}, \mathbf{v}) = \bar{\mathbf{v}}^T \mathbf{L}(\lambda)\mathbf{u},$$

97 where the bar indicates complex conjugation, we obtain (Kirillov 2010)

$$98 \quad \int_0^L \bar{\mathbf{v}}^T \mathbf{L}(\lambda)\mathbf{u} dx = \int_0^L (\bar{\mathbf{L}}^T(\bar{\lambda})\mathbf{v})^T \mathbf{u} dx + \mathbf{v}^T \mathcal{L}\mathbf{u}. \quad (9)$$

99 Here  $\bar{\mathbf{L}}^T(\bar{\lambda})\mathbf{v} =: \mathbf{L}^\dagger(\bar{\lambda})\mathbf{v}$  is the *adjoint* differential expression (Kirillov 2010)

$$\mathbf{L}^\dagger(\bar{\lambda})\mathbf{v} = \sum_{q=0}^4 (-1)^{4-q} \partial_z^{4-q} (\bar{\mathbf{I}}_q^T \mathbf{v}) = \mathbf{l}_0 \partial_z^4 \mathbf{v} + \mathbf{l}_1 \partial_z^3 \mathbf{v} + \mathbf{l}_2 \partial_z^2 \mathbf{v} + \mathbf{l}_4(\bar{\lambda})\mathbf{v}, \quad (10)$$

the vectors  $\mathbf{u}$  and  $\mathbf{v}$  are

$$\begin{aligned} \mathbf{u}^T &= (\mathbf{u}^T(0), \mathbf{u}'_z{}^T(0), \mathbf{u}''_z{}^T(0), \mathbf{u}'''_z{}^T(0), \mathbf{u}^T(L), \mathbf{u}'_z{}^T(L), \mathbf{u}''_z{}^T(L), \mathbf{u}'''_z{}^T(L)) \\ \mathbf{v}^T &= (\mathbf{v}^T(0), \mathbf{v}'_z{}^T(0), \mathbf{v}''_z{}^T(0), \mathbf{v}'''_z{}^T(0), \mathbf{v}^T(L), \mathbf{v}'_z{}^T(L), \mathbf{v}''_z{}^T(L), \mathbf{v}'''_z{}^T(L)) \end{aligned}$$

and the block matrix  $\mathcal{L} := (\mathbf{l}_{ij})$

$$\mathcal{L} = \begin{pmatrix} -\mathfrak{L}(0) & 0 \\ 0 & \mathfrak{L}(L) \end{pmatrix}, \quad \mathfrak{L}(z) = \begin{pmatrix} \mathbf{l}_{00} & \mathbf{l}_{01} & \mathbf{l}_{02} & \mathbf{l}_{03} \\ \mathbf{l}_{10} & \mathbf{l}_{11} & \mathbf{l}_{12} & 0 \\ \mathbf{l}_{20} & \mathbf{l}_{21} & 0 & 0 \\ \mathbf{l}_{30} & 0 & 0 & 0 \end{pmatrix}.$$

The matrices  $\mathbf{l}_{ij}$  are expressed through the matrices of the differential expression as (Kirillov 2010, 2013a)

$$\mathbf{l}_{ij} = \sum_{k=i}^{3-j} (-1)^k M_{ij}^k \partial_z^{k-i} \mathbf{l}_{3-j-k}, \quad M_{ij}^k := \begin{cases} \frac{k!}{(k-i)!i!}, & i+j \leq 3 \cap k \geq i \geq 0 \\ 0, & i+j > 3 \cap k < i \end{cases}$$

which yields

$$\mathfrak{L}(z) = \begin{pmatrix} 0 & \mathbf{l}_2 & \mathbf{l}_1 & \mathbf{l}_0 \\ -\mathbf{l}_2 & -\mathbf{l}_1 & -\mathbf{l}_0 & 0 \\ \mathbf{l}_1 & \mathbf{l}_0 & 0 & 0 \\ -\mathbf{l}_0 & 0 & 0 & 0 \end{pmatrix},$$

where 0 denotes the  $2 \times 2$  zero matrix.

Boundary conditions (8) can be written in the matrix form as

$$\mathbf{U}_k \mathbf{u} = \sum_{j=0}^3 \mathbf{A}_{kj} \mathbf{u}_z^{(j)}(z=0) + \sum_{j=0}^3 \mathbf{B}_{kj} \mathbf{u}_z^{(j)}(z=L) = 0, \quad k = 1, \dots, 4$$

where

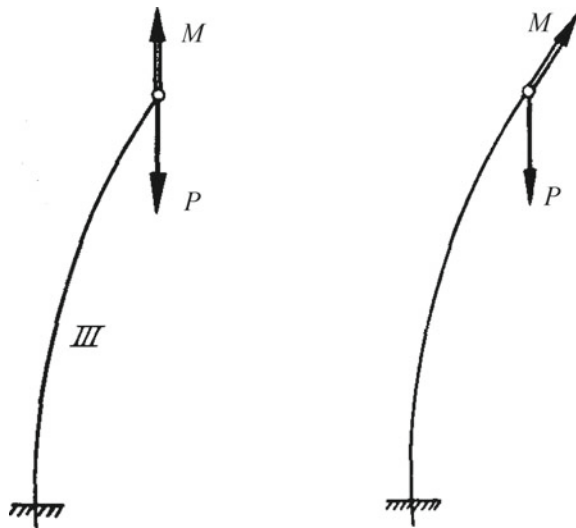
$$\mathbf{A}_{10} = \mathbf{A}_{21} = \mathbf{I}, \quad \mathbf{B}_{32} = \mathbf{l}_1, \quad \mathbf{B}_{33} = \mathbf{l}_0, \quad \mathbf{B}_{42} = \mathbf{l}_2, \quad \mathbf{B}_{43} = \mathbf{l}_1, \quad \mathbf{B}_{44} = \mathbf{l}_0$$

and all of other matrices  $\mathbf{A}_{kj}$  and  $\mathbf{B}_{kj}$  are zero. Introducing the matrices  $\mathfrak{A} = (\mathbf{A}_{kj})|_{z=0}$  and  $\mathfrak{B} = (\mathbf{B}_{kj})|_{z=L}$  and composing the block matrix  $\mathfrak{L} = [\mathfrak{A}, \mathfrak{B}]$  we can finally write the boundary conditions (8) in the compact matrix form (Kirillov 2010, 2013a)

$$\mathfrak{L} \mathbf{u} = [\mathfrak{A}, \mathfrak{B}] \mathbf{u} = 0. \quad (11)$$



**Fig. 2** (Left) Greenhill-III problem with the axial torque described by the problem (14). (Right) Nicolai's variant of the Greenhill-III problem with the follower torque described by the problem (13) which is adjoint to (14) (from Ziegler 1951a)



Extend the original matrix  $\mathcal{L}$  to a square non-degenerate matrix  $\mathcal{U}$  by an appropriate choice of the auxiliary matrices  $\tilde{\mathfrak{A}}$  and  $\tilde{\mathfrak{B}}$

$$\mathcal{L} = [\mathfrak{A}, \mathfrak{B}] \hookrightarrow \mathcal{U} = \begin{pmatrix} \mathfrak{A} & \mathfrak{B} \\ \tilde{\mathfrak{A}} & \tilde{\mathfrak{B}} \end{pmatrix}, \quad \det \mathcal{U} \neq 0.$$

Then, we can obtain the formula for calculation of the matrix  $\mathfrak{B}$  of the boundary conditions for the adjoint differential expression (10)

$$\mathfrak{B}v = 0$$

109 and the auxiliary matrix  $\tilde{\mathfrak{B}}$

$$110 \quad \begin{bmatrix} -\tilde{\mathfrak{B}} \\ \mathfrak{B} \end{bmatrix}^T = \mathcal{L}\mathcal{U}^{-1} = \begin{pmatrix} -\mathcal{L}(0) & 0 \\ 0 & \mathcal{L}(L) \end{pmatrix} \begin{pmatrix} \mathfrak{A} & \mathfrak{B} \\ \tilde{\mathfrak{A}} & \tilde{\mathfrak{B}} \end{pmatrix}^{-1} \quad (12)$$

Choosing

$$\tilde{\mathfrak{A}} = \begin{pmatrix} 0 & 0 & \mathbf{I} & 0 \\ 0 & 0 & 0 & \mathbf{I} \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{pmatrix}, \quad \tilde{\mathfrak{B}} = \begin{pmatrix} 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & \mathbf{I} & 0 & 0 \\ \mathbf{I} & 0 & 0 & 0 \end{pmatrix},$$

111 where  $\mathbf{I}$  is the  $2 \times 2$  identity matrix and  $0$  denotes the  $2 \times 2$  zero matrix, we find that  
 112  $\det \mathcal{U} = (EI)^4 \neq 0$ .

113 Then, the differential expression (10) and the relation (12) yield the adjoint bound-  
 114 ary eigenvalue problem:

$$\begin{aligned}
 115 \quad & \mathbf{I}_0 \partial_z^4 \mathbf{v} + \mathbf{I}_1 \partial_z^3 \mathbf{v} + \mathbf{I}_2 \partial_z^2 \mathbf{v} + \mathbf{I}_4(\bar{\lambda}) \mathbf{v} = 0, \\
 116 \quad & \mathbf{v}(0) = \mathbf{v}'(0) = 0, \\
 117 \quad & \mathbf{v}''(L) = 0, \\
 118 \quad & \mathbf{I}_0 \mathbf{v}'''(L) + \mathbf{I}_2 \mathbf{v}'(L) = 0, \tag{13}
 \end{aligned}$$

119 which is instructive to compare with the original boundary eigenvalue  
 120 problem (7), (8):

$$\begin{aligned}
 121 \quad & \mathbf{I}_0 \partial_z^4 \mathbf{u} + \mathbf{I}_1 \partial_z^3 \mathbf{u} + \mathbf{I}_2 \partial_z^2 \mathbf{u} + \mathbf{I}_4(\lambda) \mathbf{u} = 0, \\
 122 \quad & \mathbf{u}(0) = \mathbf{u}'(0) = 0, \\
 123 \quad & \mathbf{I}_0 \mathbf{u}''(L) + \mathbf{I}_1 \mathbf{u}'(L) = 0, \\
 124 \quad & \mathbf{I}_0 \mathbf{u}'''(L) + \mathbf{I}_1 \mathbf{u}''(L) + \mathbf{I}_2 \mathbf{u}'(L) = 0. \tag{14}
 \end{aligned}$$

125 It is easy to see that the differential expressions of the problems (14) and (13)  
 126 are identical and the difference comes from the terms in the boundary conditions  
 127 that contain the matrix  $\mathbf{I}_1 = \begin{pmatrix} 0 & M \\ -M & 0 \end{pmatrix}$  that is non-zero at nonzero torque  $M$ . Only  
 128 if  $M = 0$  the matrix  $\mathbf{I}_1 = 0$  and the boundary conditions of the original boundary  
 129 eigenvalue problem and the adjoint boundary eigenvalue problem coincide.

130 Therefore, only in the absence of the torque ( $M = 0$ ), the problem (14) as well  
 131 as its adjoint (13), is self-adjoint and represents a conservative system, which is not  
 132 surprising in view that it is the Euler buckling problem for an elastic shaft.

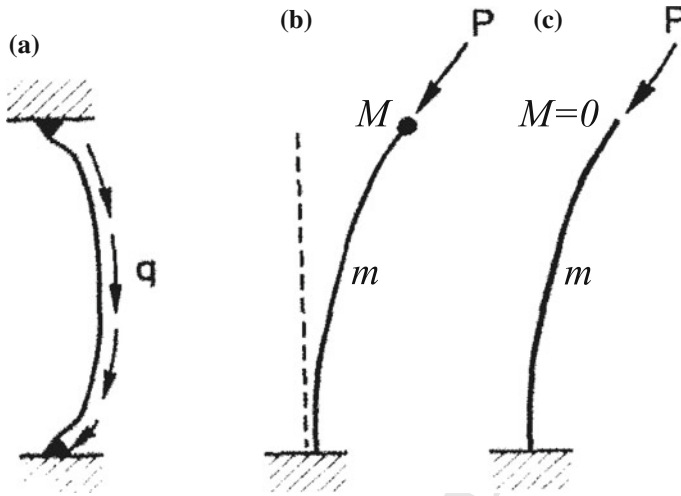
133 In case when  $M \neq 0$  the boundary conditions of the adjoint problem (13) do  
 134 not coincide with the boundary conditions of the original problem (14), manifesting  
 135 the non-self-adjoint nature of the non-conservative Greenhill-III problem (Ziegler  
 136 1951a, b, 1956).

137 It is well-known that adjoint problems have the same characteristic equation that  
 138 determines eigenvalues. Hence, stability properties of (14) and (13) are identical  
 139 despite they have different mechanical meaning.

140 The boundary value problem (14) corresponds to the original Greenhill-III  
 141 clamped-free shaft loaded by the axial force and the axial torque, Fig. 2(left). It  
 142 turns out that its adjoint given by (13) corresponds to the Nicolai's variant of the  
 143 Greenhill-III problem with the axial force and the follower torque, Fig. 2(right),  
 144 Bolotin (1963).

145 Both mechanical systems shown in Fig. 2 are nonconservative but have the same  
 146 spectrum and, therefore, the same stability properties. In the both problems the critical  
 147 value of the torque at  $P = 0$  is  $M_{cr} = 0$  (Nicolai's paradox) no matter whether the  
 148 torque is axial or follower.

Editor Proof



**Fig. 3** **a** Pflüger's hinged-hinged column loaded by the distributed follower force (static instability, or divergence), **b** Pflüger's clamped-free column of the mass per unit length,  $m$ , carrying the end mass,  $M$ , and loaded by the concentrated follower force at the tip, **c** Beck's column loaded by the concentrated follower force is a particular case of **b** with the end mass  $M = 0$  (dynamic instability, or flutter), from Gladwell (1990)

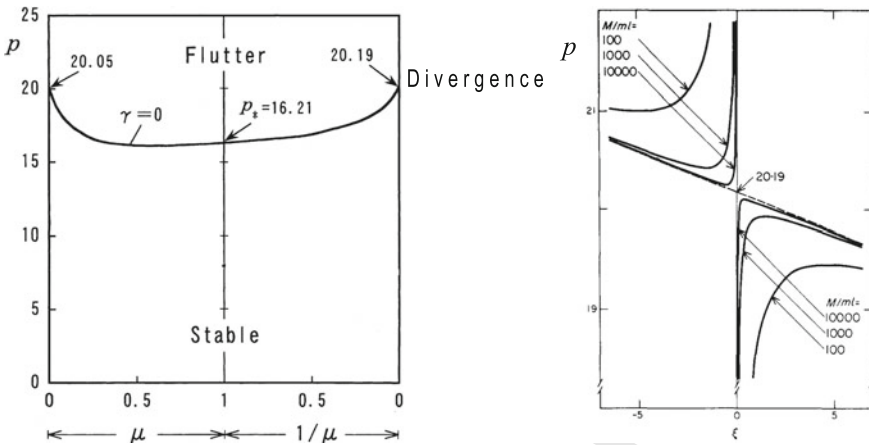
### 1.3 From Follower Torques to Follower Forces

A remarkable property of the Greenhill's five problems established by Nicolai and Ziegler is that, depending on boundary conditions, they could be both conservative and nonconservative. In conservative cases I and II, the Greenhill's shaft loses stability of the rectilinear equilibrium statically, i.e. without vibrations (*divergence* instability). In the nonconservative cases III and IV (and their Nicolai's variants with the *follower torque*), however, the mechanism of instability involves growing oscillations about the rectilinear equilibrium and is called *flutter*. Whereas divergence is the only possible type of instability in conservative systems, the nonconservative systems possess both flutter and divergence.

For instance, the nonconservative Greenhill-V shaft loses its stability by divergence (Greenhill 1883; Ziegler 1951a; Gladwell 1990). In 1950 Pflüger established divergence instability of a nonconservative hinged-hinged elastic column loaded by a distributed *follower force*, Fig. 3a.

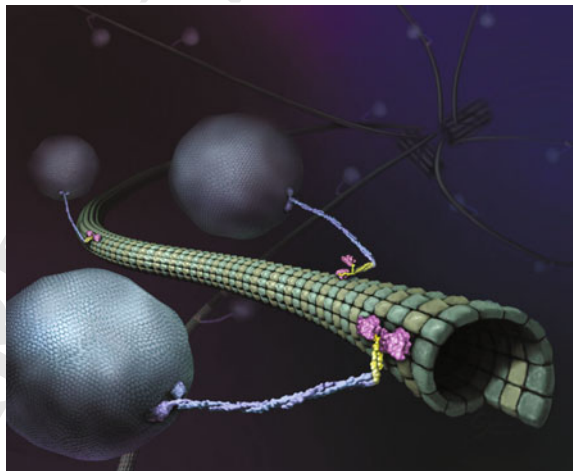
Note that columns loaded by distributed follower forces provide a basis for mathematical modeling of some biomechanical objects. We mention, for instance, recent works on the human spine (Rohlmann et al. 2009), centipede locomotion (Aoi et al. 2013), and flutter of flagella under the action of distributed tangential follower forces caused by cytoskeletal motor proteins (Bayly and Dutcher 2016).

Immediately after the Pflüger's work, Beck (1952) has found flutter of a clamped-free elastic column of length,  $L$ , and mass per unit length,  $m$ , loaded by the



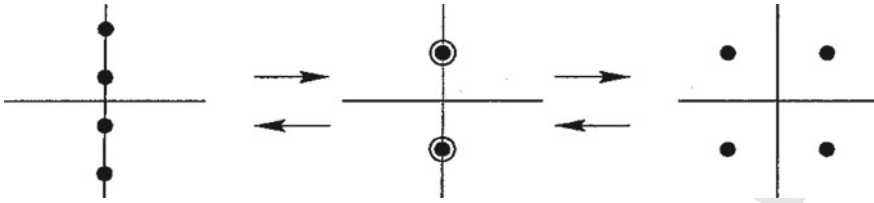
**Fig. 4** (Left) Stability map for the elastic Pflüger column in the “load” - “mass ratio” plane (from Ryu and Sugiyama 2003). (Right) Load parameter  $p = \frac{PL^2}{EI}$  versus dimensionless squared vibration frequency  $\xi = \frac{m\omega^2 L^4}{EI}$  for the Pflüger column at different mass ratios  $\mu$  when  $1/\mu$  is close to zero (from Sugiyama et al. 1976)

**Fig. 5** Molecular motors (kinesin) transporting membranes along microtubules (cytoskeletal filaments) inside a cell cause tangential follower forces acting on the microtubules (from Vale Lab web site <https://valelab4.ucsf.edu/external/moviepages/moviesMolecMotors.html>)



170 concentrated follower force at its tip, Fig. 3c. In 1955 Pflüger re-considered the  
 171 Beck’s column with an end mass,  $M$ , (see Fig. 3b) and found it flutter-unstable  
 172 for almost all mass ratios  $\mu = \frac{M}{mL}$ , except for the case when  $m = 0$  or  $\mu \rightarrow \infty$ ,  
 173 Fig. 4(left).

174 Figure 4(right) shows the load parameter of the Pflüger column as a function of the  
 175 squared dimensionless eigenfrequency at small values of  $\mu^{-1}$ . The lower hyperbolic  
 176 branch has its maximum at the critical flutter value of the load. The interval of loads



**Fig. 6** Linear reversible-Hopf bifurcation: (Left) eigenvalues of a stable reversible system are all imaginary and semi-simple; (centre) a pair of two simple imaginary eigenvalues (as well as the complex conjugate pair) merges into a pair of double imaginary eigenvalues with the Jordan block at the flutter threshold; (right) the pair of the double non-semi-simple eigenvalues unfolds into a complex quadruplet inside the flutter domain (from Lamb and Roberts 1998)

177 corresponding to flutter is between the minimum of the upper hyperbolic branch and  
 178 the maximum of the lower hyperbolic branch in Fig. 4(right). As  $\mu$  increases, the size  
 179 of the flutter interval tends to zero so that in the limit  $\mu \rightarrow \infty$  the two hyperbolic  
 180 branches merge and form a crossing at the load  $p \approx 20.19$  (Sugiyama et al. 1976).  
 181 Exactly at the crossing the eigenfrequency is double zero with the Jordan block,  
 182 which corresponds to the onset of the divergence instability. In the  $\mu \rightarrow \infty$  limit  
 183 the Pflüger column is *weightless* and is known as the *Dzhanelidze column* (Bolotin  
 184 1963). The opposite limit,  $\mu \rightarrow 0$ , of the Pflüger column is known as the *Beck column*  
 185 with the critical flutter load  $p \approx 20.05$ . It is instructive to note that the critical load  
 186 reaches its local maxima exactly in these two limiting cases, Fig. 4(left).

187 Connection of a maximum of the critical load and a crossing in the load-frequency  
 188 plane (Fig. 4(right)) is not a coincidence. Already Mahrenholtz and Bogacz (1981)  
 189 emphasized that “In the case of complicated structures there may appear different  
 190 shapes of characteristic curves, and only an analysis in the [load-frequency] plane  
 191 may assure the correct results for the design of structures subjected to nonconser-  
 192 vative loads”. A general perturbation approach to local extrema associated with  
 193 the crossings of characteristic curves has been developed in Kirillov and Seyranian  
 194 (2002a, b).

195 The follower force problems of 1950-s are increasingly popular nowadays in  
 196 the mathematical modeling of mechanics underlying complex cellular phenomena  
 197 caused by molecular motors that translocate along cytoskeletal filaments, carrying  
 198 cargo, Fig. 5. It turns out that molecular motors produce piconewton tangential fol-  
 199 lower forces acting on filaments and resulting in their flutter, which is well described  
 200 by the classical continuous models of Beck and Pflüger and their discrete analogue  
 201 — the *Ziegler pendulum* (Ziegler 1952; Saw and Wood 1975) — as is shown in  
 202 the recent work by De Canio et al. (2017). Note that the Ziegler pendulum has been  
 203 realized experimentally by Bigoni and Noselli (2011) and the Pflüger column by  
 204 Bigoni et al. (2018).

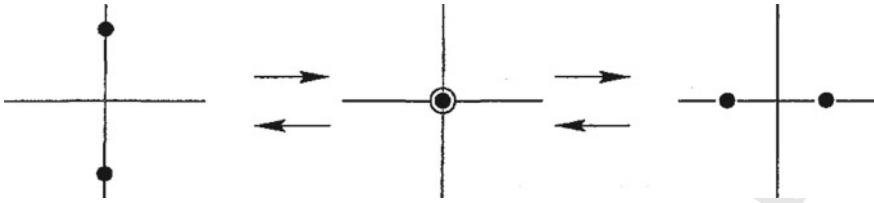


Fig. 7 Steady-state bifurcation in a reversible system: (left) eigenvalues of a stable reversible system are all imaginary and semi-simple; (centre) a conjugate pair of simple imaginary eigenvalues merges into a double zero eigenvalue with the Jordan block at the divergence threshold; (right) the double zero non-semi-simple eigenvalue splits into two real eigenvalues of opposite signs inside the divergence domain

## 2 Reversible and Circulatory Systems

O'Reilly, Malhotra and Namachchivaya (1995, 1996) observed that the governing equations of the classical structures with nonconservative follower loads possess a special type of symmetry, which largely determines their stability properties.

This symmetry, known as the *reversible symmetry*, can be defined with reference to the differential equation (Lamb and Roberts 1998)

$$\frac{d\mathbf{x}}{dt} = \mathbf{g}(\mathbf{x}), \quad \mathbf{x} \in \mathbb{R}^n$$

which is said to be **R**-reversible ( $\mathbf{R}^{-1} = \mathbf{R}$ ) if it is invariant with respect to the transformation  $(\mathbf{x}, t) \mapsto (\mathbf{R}\mathbf{x}, -t)$ , implying that the right hand side should satisfy  $\mathbf{R}\mathbf{g}(\mathbf{x}) = -\mathbf{g}(\mathbf{R}\mathbf{x})$ .

If  $\mathbf{x} = \mathbf{x}_0$  is a *reversible equilibrium* such that  $\mathbf{R}\mathbf{x}_0 = \mathbf{x}_0$ , and  $\mathbf{A} = \nabla\mathbf{g}$  is the linearization matrix about  $\mathbf{x}_0$ , then  $\mathbf{A} = -\mathbf{R}\mathbf{A}\mathbf{R}$ , and the characteristic polynomial

$$\det(\mathbf{A} - \lambda\mathbf{I}) = \det(-\mathbf{R}\mathbf{A}\mathbf{R} - \mathbf{R}\lambda\mathbf{R}) = (-1)^n \det(\mathbf{A} + \lambda\mathbf{I}),$$

implies that  $\pm\lambda, \pm\bar{\lambda}$  are eigenvalues of  $\mathbf{A}$  (Lamb and Roberts 1998). Due to the spectrum's symmetry with respect to both the real and imaginary axes of the complex plane, stability requires that *all* the eigenvalues of  $\mathbf{A}$  stay on the imaginary axis, Fig. 6(left).

Transition from stability to flutter instability occurs through the *reversible-Hopf bifurcation* (Lamb and Roberts 1998) that requires the generation of a non-semi-simple double pair of imaginary eigenvalues and its subsequent separation into a complex quadruplet, Fig. 6.

Transition from stability to divergence instability is accompanied by the *steady-state bifurcation* in which two simple imaginary eigenvalues merge at zero and then split into a real couple with the opposite signs, Fig. 7.

223 An important for applications fact is that reversible are all equations of second  
224 order (Lamb and Roberts 1998):

$$225 \quad \frac{d^2 \mathbf{x}}{dt^2} = \mathbf{f}(\mathbf{x}). \quad (15)$$

Indeed, denoting  $\mathbf{x}_1 = \mathbf{x}$  and  $\mathbf{x}_2 = \frac{d\mathbf{x}}{dt}$  we can write the first-order system

$$\dot{\mathbf{x}}_1 = \mathbf{x}_2, \quad \dot{\mathbf{x}}_2 = \mathbf{f}(\mathbf{x}_1),$$

which is invariant under the transformation

$$\mathbf{x}_1 \rightarrow \mathbf{x}_1, \quad \mathbf{x}_2 \rightarrow -\mathbf{x}_2, \quad t \rightarrow -t.$$

The system (15) is reversible also in the case when the positional force  $\mathbf{f}(\mathbf{x})$  has a non-trivial *curl*

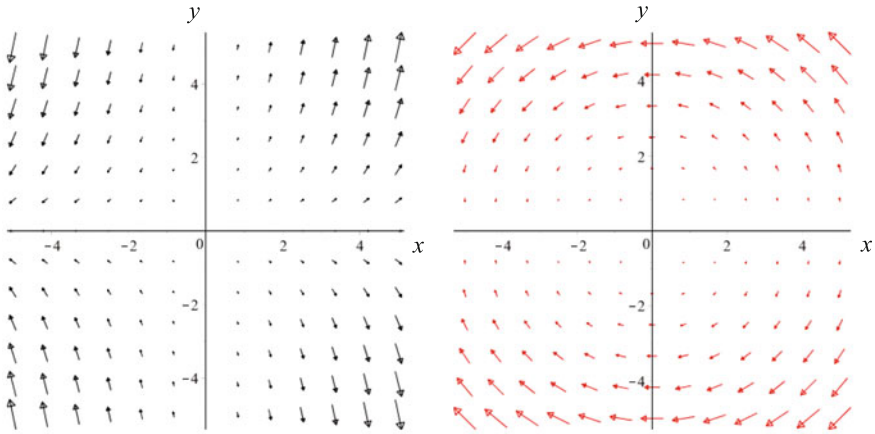
$$\nabla \times \mathbf{f}(\mathbf{x}) \neq 0,$$

226 which makes the reversible system *nonconservative*. Such nonconservative *curl*  
227 *forces* (Berry and Shukla 2016) that cannot be derived from any potential appear  
228 in modern opto-mechanical applications, including optical tweezers (Wu et al. 2009;  
229 Simpson and Hanna 2010; Sukhov and Dogariu 2017) and *light robotics* (Phillips  
230 et al. 2017). In mechanics these nonconservative positional forces are known as *circu-*  
231 *latory forces* for producing non-zero work along a closed circuit (Ziegler 1953a, b).  
232 A circulatory force acting on an elastic structure and remaining directed along the  
233 tangent line to the structure at the point of its application during deformation is the  
234 already familiar to us *follower force* (Ziegler 1952; Bolotin 1963).

235 We notice that in aeroelasticity the term ‘circulatory’ is frequently associated with  
236 the lift force in the Theodorsen lift model (Theodorsen 1935) that was developed to  
237 explain flutter instability occurring in aircrafts at high speeds. The Kutta–Joukowski  
238 theorem relates the lift on an airfoil to a circulatory component (circulation) of  
239 the flow around the airfoil. The circulation is the contour integral of the tangential  
240 velocity of the air on a closed loop (circuit) around the boundary of an airfoil. Hence  
241 the name *circulatory lift force*, see Pigolotti et al. (2017). Remarkably, the Theodorsen  
242 model is nonconservative and the non-potential positional forces arising in it due to  
243 the circulatory lift are simultaneously the circulatory forces in the sense of Ziegler  
244 (Pigolotti et al. 2017).

## 245 **2.1 Zubov-Zhuravlev Decomposition of Non-potential Force** 246 **Fields**

247 Zubov (1970) established the following instructive result:



**Fig. 8** (Left) The non-potential force field  $\mathbf{f} = (x, xy)^T = \mathbf{f}' + \mathbf{f}''$ ; (right) its circulatory part  $\mathbf{f}'' = \left(-\frac{y^2}{3}, \frac{xy}{3}\right)^T$

248 **Theorem 2.1** (Zubov 1970) Let  $f(t, \mathbf{x}) : \mathbb{R}_+ \times \mathbb{R}^n \rightarrow \mathbb{R}^n$  be a real-valued con-  
 249 tinuous vector-function and let  $w(t, \mathbf{x}) = \mathbf{f}^T \mathbf{x} = \sum_{i=1}^n x_i f_i(t, \mathbf{x})$  be a continuously  
 250 differentiable function with respect to components of  $\mathbf{x}$ . Then,

- 251 (a) there exists a real-valued function  $V(t, \mathbf{x}) : \mathbb{R}_+ \times \mathbb{R}^n \rightarrow \mathbb{R}$ , which is continuous  
 252 and continuously differentiable with respect to components of  $\mathbf{x}$ ;  
 253 (b)  $f(t, \mathbf{x})$  possesses the following representation

$$\mathbf{f}(t, \mathbf{x}) = -\nabla_{\mathbf{x}} V(t, \mathbf{x}) + \mathbf{P}\mathbf{x}, \tag{16}$$

254 where  $\mathbf{P}(t, \mathbf{x})$  is an  $n \times n$  skew-symmetric matrix ( $\mathbf{P}^T = -\mathbf{P}$ ) with the elements  
 255 that are continuous functions of  $t$  and components of  $\mathbf{x}$ .  
 256

257 **Example:** Let

$$\mathbf{f}(t, \mathbf{x}) = \begin{pmatrix} x \\ xy \end{pmatrix}, \quad \mathbf{x} = \begin{pmatrix} x \\ y \end{pmatrix} \tag{17}$$

258 According to Theorem 2.1, there exists the following decomposition  
 259

$$\begin{aligned} \mathbf{f}(t, \mathbf{x}) &= -\begin{pmatrix} \frac{\partial V}{\partial x} \\ \frac{\partial V}{\partial y} \end{pmatrix} + \frac{y}{3} \begin{pmatrix} 0 & -1 \\ 1 & 0 \end{pmatrix} \begin{pmatrix} x \\ y \end{pmatrix} \\ &= \begin{pmatrix} x + \frac{y^2}{3} \\ \frac{2xy}{3} \end{pmatrix} + \begin{pmatrix} -\frac{y^2}{3} \\ \frac{xy}{3} \end{pmatrix} = \begin{pmatrix} x \\ xy \end{pmatrix}, \end{aligned} \tag{18}$$



where  $V(t, \mathbf{x}) = -\frac{x^2}{2} - \frac{xy^2}{3}$ , see Fig. 8. Notice that many examples of nonconservative force fields and their curls can be found in the modern literature on optical tweezers, see e.g. Wu et al. (2009), Simpson and Hanna (2010), Sukhov and Dogariu (2017), and light robotics (Phillips et al. 2017).

Zhuravlev (2007, 2008) proposed an algorithm for constructing the Zubov decomposition, in particular, of nonlinear generalized forces in the Lagrange equations. Here we are interested in positional forces only.

Let  $T$  denote kinetic energy of a mechanical system. Consider the Lagrange equations

$$\frac{d}{dt} \left( \frac{\partial T}{\partial \dot{q}_i} \right) - \frac{\partial T}{\partial q_i} = f_i(t, q_1, \dots, q_n), \quad i = 1, \dots, n.$$

We assume that the generalized forces  $f_i$  have positional character, being functions of time and generalized coordinates only.

Let us first assume that the generalized forces  $\mathbf{f}$  are *linear*

$$\mathbf{f} = -\mathbf{A}\mathbf{q}, \quad \mathbf{A} \neq \mathbf{A}^T.$$

Recall that the  $n \times n$  matrix  $\mathbf{A}$  can be uniquely represented as the sum

$$\mathbf{A} = \frac{\mathbf{A} + \mathbf{A}^T}{2} + \frac{\mathbf{A} - \mathbf{A}^T}{2} = \mathbf{K} + \mathbf{N},$$

where  $\mathbf{K} = \mathbf{K}^T$  is a real symmetric matrix and  $\mathbf{N} = -\mathbf{N}^T$  is a real skew-symmetric matrix. Then, we can write the generalized positional force as

$$\mathbf{f} = -\mathbf{K}\mathbf{q} - \mathbf{N}\mathbf{q},$$

where the force  $\mathbf{f}' = -\mathbf{K}\mathbf{q}$  is derived from the potential  $V(\mathbf{q}) = \frac{1}{2}\mathbf{q}^T \mathbf{K}\mathbf{q}$ :

$$\mathbf{f}' = -\nabla V(\mathbf{q})$$

and the circulatory force  $\mathbf{f}'' = -\mathbf{N}\mathbf{q}$  is orthogonal to the vector of generalizes coordinates

$$\mathbf{q}^T \mathbf{f}'' = 0.$$

Indeed,

$$\mathbf{q}^T \mathbf{f}'' = -\mathbf{q}^T \mathbf{N}\mathbf{q} = (\mathbf{q}^T \mathbf{N}^T \mathbf{q})^T = \mathbf{q}^T \mathbf{N}\mathbf{q} \Rightarrow \mathbf{q}^T \mathbf{N}\mathbf{q} = 0.$$

A *linear circulatory system* is thus defined as (Ziegler 1953a,b, 1956)

$$\ddot{\mathbf{q}} + \mathbf{K}\mathbf{q} + \mathbf{N}\mathbf{q} = 0.$$

This is a reversible system (O'Reilly, Malhotra and Namachchivaya 1996).

Let us calculate the work of the linear positional force  $\mathbf{f}$  on the displacement  $\mathbf{q}$  with the frozen time

$$W = \int_0^1 \mathbf{q}^T \mathbf{f}(s\mathbf{q}) ds = - \int_0^1 \mathbf{q}^T \mathbf{K} \mathbf{q} s ds - \int_0^1 \mathbf{q}^T \mathbf{N} \mathbf{q} s ds = -\frac{1}{2} \mathbf{q}^T \mathbf{K} \mathbf{q}$$

Therefore, the potential component of the linear positional force  $\mathbf{f}$  is  $\mathbf{f}' = \nabla W = -\nabla V$  and the circulatory component is just  $\mathbf{f}'' = \mathbf{f} - \mathbf{f}'$ . Zhuravlev (2007, 2008) employs this idea for the decomposition of *nonlinear* non-potential force fields into a potential and circulatory parts.

Following Zhuravlev (2007, 2008), we define the potential part of  $\mathbf{f}$  as  $\mathbf{f}' = -\nabla V$ , where

$$V = - \int_0^1 \mathbf{q}^T \mathbf{f}(s\mathbf{q}) ds.$$

Then, the circulatory part of the nonlinear force  $\mathbf{f}$  is  $\mathbf{f}'' = \mathbf{f} - \mathbf{f}'$ ,  $\mathbf{f}'' \cdot \mathbf{q} = 0$ .

**Example:** Decompose the non-potential vector field  $\mathbf{f}$  into the potential and circulatory parts

$$\mathbf{f} = \begin{pmatrix} x \\ xy \end{pmatrix} = \mathbf{f}' + \mathbf{f}''.$$

First, construct the potential function  $V$  of the potential part of the field

$$\begin{aligned} V &= - \int_0^1 [xf_x(sx, sy) + yf_y(sx, sy)] ds = - \int_0^1 (x^2s + xy^2s^2) ds \\ &= -\frac{x^2}{2} - \frac{xy^2}{3}. \end{aligned}$$

Then, find the potential part of  $\mathbf{f}$

$$\mathbf{f}' = - \begin{pmatrix} \frac{\partial V}{\partial x} \\ \frac{\partial V}{\partial y} \end{pmatrix} = \begin{pmatrix} x + \frac{y^2}{3} \\ \frac{2xy}{3} \end{pmatrix}.$$

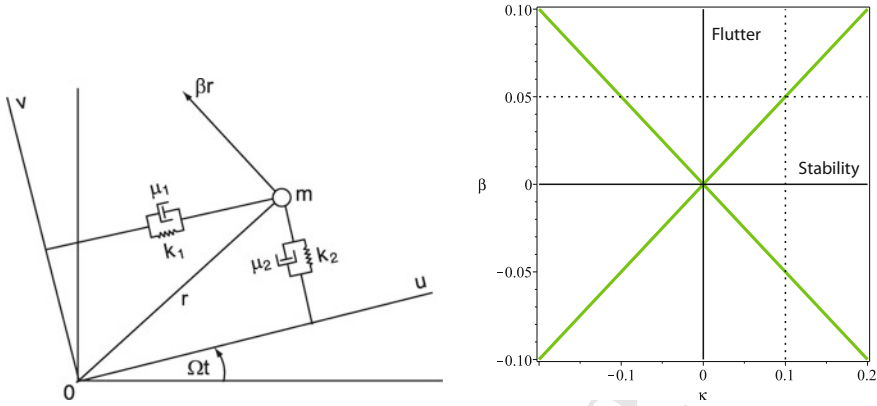
Finally, determine the circulatory part of  $\mathbf{f}$

$$\mathbf{f}'' = \mathbf{f} - \mathbf{f}' = \begin{pmatrix} -\frac{y^2}{3} \\ \frac{xy}{3} \end{pmatrix} = \frac{y}{3} \begin{pmatrix} 0 & -1 \\ 1 & 0 \end{pmatrix} \begin{pmatrix} x \\ y \end{pmatrix}, \quad \mathbf{f}'' \cdot \mathbf{q} = 0,$$

in agreement with Theorem 2.1. Note that  $\nabla \times \mathbf{f}'' = y\mathbf{e}_z \neq 0$ .

The decomposition is unique up to the class of potential forces that are simultaneously orthogonal to the vector of coordinates:  $\mathbf{q}^T \nabla V = 0$ . For instance, the force derived from the potential  $V(x, y) = x/(x + y)$  belongs to this class (Zhuravlev 2007, 2008)

Editor Proof



**Fig. 9** (Left) Rotating shaft by Shieh and Masur (1968). (Right) Stability map of the model (21) with  $k_1 = 1$  and  $m = 1$

$$\mathbf{f} = - \begin{pmatrix} \frac{\partial V}{\partial x} \\ \frac{\partial V}{\partial y} \end{pmatrix} = \frac{1}{(x + y)^2} \begin{pmatrix} -y \\ x \end{pmatrix}, \quad \mathbf{f} \cdot \mathbf{q} = 0.$$

279 In this case, obviously,  $\nabla \times \mathbf{f} = -\nabla \times (\nabla V) = 0$ .

## 2.2 Circulatory Forces in Rotor Dynamics

281 Non-potential circulatory forces historically originated in equations of rotor dynamics  
 282 when dissipation both in rotor and stator was taken into account. The two types  
 283 of damping were introduced by Kimball (1925) in order to explain a new type of  
 284 instability observed in built-up rotors at high speeds in the early 1920s. Smith (1933)  
 285 implemented this idea in a model of a rotor carried by a flexible shaft in flexible  
 286 bearings with the linearization given by the equation

$$\ddot{\mathbf{z}} + \mathbf{D}\dot{\mathbf{z}} + 2\Omega\mathbf{G}\dot{\mathbf{z}} + (\mathbf{K} + (\Omega\mathbf{G})^2)\mathbf{z} + \beta\mathbf{N}\mathbf{z} = 0 \tag{19}$$

288 where  $\mathbf{z}^T = (x, y)$  is the position vector in the frame rotating with the shaft's angular  
 289 velocity  $\Omega$ ,  $\mathbf{D} = \text{diag}(\delta + \beta, \delta + \beta)$ ,  $\mathbf{G} = \mathbf{J}$ ,  $\mathbf{J} = \begin{pmatrix} 0 & -1 \\ 1 & 0 \end{pmatrix}$ ,  $\mathbf{K} = \text{diag}(k_1, k_2)$ , and  
 290  $\mathbf{N} = \Omega\mathbf{J}$ . In Smith's model (19) the stationary (in the laboratory frame, and thus  
 291 external with respect to the shaft) damping coefficient  $\beta > 0$  represents the effect of  
 292 viscous damping in bearing supports while the rotating damping coefficient  $\delta > 0$   
 293 represents the effect of viscous damping in the shaft itself (internal damping). The  
 294 term  $\beta\mathbf{N}\mathbf{z}$  in Eq. (19) corresponds to circulatory forces.

295 In a more general model of the rotating shaft by Shieh and Masur (1968), the  
 296 diagonal elements of the damping matrix in Eq. (19) are allowed to be different. In  
 297 fact, Shieh and Masur (1968) model the shaft as the point mass  $m$  which is attached  
 298 by two springs with the stiffness coefficients  $k_1$  and  $k_2 = k_1 + \kappa$  and two dampers  
 299 with the coefficients  $\mu_1$  and  $\mu_2$  to a Cartesian coordinate system  $Ouv$  rotating at  
 300 constant angular velocity  $\Omega$ , Fig. 9 (left).

301 A non-conservative positional force which is proportional to the radial distance  
 302 of the mass from the origin and perpendicular to the radius vector  $\mathbf{f}'' = \begin{pmatrix} -\beta v \\ \beta u \end{pmatrix}$  acts  
 303 on the mass. Such a force on the shaft in the bearings may arise in a rotating fluid or  
 304 in an electromagnetic field. The linearized equations of motion of the shaft have the  
 305 form (Shieh and Masur 1968; Kirillov 2013a, 2011a, b)

$$306 \quad m\ddot{u} + \mu_1\dot{u} - 2m\Omega\dot{v} + (k_1 - m\Omega^2)u + \beta v = 0,$$

$$307 \quad m\ddot{v} + \mu_2\dot{v} + 2m\Omega\dot{u} + (k_2 - m\Omega^2)v - \beta u = 0. \quad (20)$$

308 Assuming that damping is absent ( $\mu_1 = 0, \mu_2 = 0$ ) and that the shaft is not rotating  
 309  $\Omega = 0$  we reduce the model (20) to the motion of the planar oscillator under the action  
 310 of a nonconservative circulatory force

$$311 \quad m\ddot{u} + k_1u + \beta v = 0,$$

$$312 \quad m\ddot{v} - \beta u + k_2v = 0. \quad (21)$$

313 Separating time in (21) with  $u = \tilde{u}e^{\lambda t}$  and  $v = \tilde{v}e^{\lambda t}$ , introducing the stiffness  
 314 anisotropy  $\kappa = k_2 - k_1$ , and writing the solvability condition for the resulting system  
 of two algebraic equations we end up with the quadratic equation in  $\lambda^2$ . Its solutions

$$\lambda = \pm i \frac{\sqrt{2m(2k_1 + \kappa \pm \sqrt{-4\beta^2 + \kappa^2})}}{2m}$$

313 are imaginary (stability) if  $\kappa^2 > 4\beta^2$  and form a complex quadruplet with negative  
 314 and positive real parts (flutter) if

$$315 \quad \beta^2 > \frac{\kappa^2}{4}. \quad (22)$$

316 This conical flutter domain is shown in Fig. 9(right) in the  $(\kappa, \beta)$ -plane of the stiffness  
 317 anisotropy,  $\kappa$ , and magnitude of the circulatory force,  $\beta$ . Note that flutter instability  
 318 occurs already at  $\beta > 0$  if the stiffness is symmetric ( $\kappa = 0$ ), similarly to the Nico-  
 319 lai paradox for the cantilever rod of circular cross-section under a follower or axial  
 320 torque. However, stiffness anisotropy ( $\kappa \neq 0$ ), no matter how small, increases the  
 321 flutter threshold as  $|\beta_f| = |\kappa|/2$ . Again, similar to the disappearance of the Nico-  
 322 lai's paradox in rods of non-circular cross-section (Nicolai 1929). This is not just  
 323 a coincidence. Indeed, the linearization of a two-degrees-of-freedom model of the

Editor Proof

324 Greenhill-Nicolai problem considered recently by Luongo and Ferretti (2016) is  
 325 described exactly by Eq. (21).

### 326 2.3 Stability Criteria for Circulatory Systems

327 Let us consider a circulatory system

$$328 \quad \ddot{\mathbf{x}} + (\mathbf{K} + \mathbf{N})\mathbf{x} = 0 \tag{23}$$

329 where  $\mathbf{K} = \mathbf{K}^T$  and  $\mathbf{N} = -\mathbf{N}^T$  are real  $m \times m$  matrices.

Separating time in (23) with the standard substitution  $\mathbf{x} = \mathbf{u}e^{\lambda t}$ , write the characteristic polynomial  $p(\lambda) = \det(\lambda^2 + \mathbf{K} + \mathbf{N})$

$$p(\lambda) = a_0\lambda^{2m} + a_1\lambda^{2m-2} + a_2\lambda^{2m-4} + \dots + \lambda^2 a_{m-1} + a_m.$$

330 Write the  $2m \times 2m$  discriminant matrix for  $p(\lambda)$

$$331 \quad \Delta = \begin{pmatrix} a_0 & a_1 & a_2 & a_3 & \dots & a_n & 0 & 0 & 0 \\ 0 & ma_0 & (m-1)a_1 & (m-2)a_2 & \dots & a_{m-1} & 0 & 0 & 0 \\ 0 & a_0 & a_1 & a_2 & \dots & a_{m-1} & a_m & 0 & 0 \\ 0 & 0 & ma_0 & (m-1)a_1 & \dots & 2a_{m-2} & a_{m-1} & 0 & 0 \\ \dots & \dots & \dots & \dots & \dots & \dots & \dots & \dots & \dots \\ \dots & \dots & \dots & \dots & \dots & \dots & \dots & \dots & \dots \\ 0 & 0 & 0 & \dots & 0 & a_0 & a_1 & \dots & a_m \\ 0 & 0 & 0 & \dots & 0 & 0 & ma_0 & \dots & a_{m-1} \end{pmatrix} \tag{24}$$

332 Consider a sequence of determinants of all even-order submatrices along the main  
 333 diagonal of  $\Delta$  starting from the upper left corner

$$334 \quad \det \Delta_1 = \det \begin{pmatrix} a_0 & a_1 \\ 0 & ma_0 \end{pmatrix}, \quad \det \Delta_2, \quad \dots, \quad \det \Delta_m = \det \Delta \tag{25}$$

**Theorem 2.2** (Gallina criterion Gallina 2003) *A necessary and sufficient condition for all the eigenvalues  $\lambda$  of the eigenvalue problem for the undamped circulatory system (23) to be imaginary is that the elements of the discriminant sequence corresponding to the discriminant matrix  $\Delta$  are all nonnegative and that the coefficients of the polynomial  $p(\lambda)$  are either all non-positive or all non-negative:*

$$\det \Delta_1 \geq 0, \quad \det \Delta_2 \geq 0, \quad \dots, \quad \det \Delta_m = \det \Delta \geq 0,$$

$$a_0 \geq 0, \quad a_1 \geq 0, \quad a_2 \geq 0, \quad \dots, \quad a_m \geq 0.$$

Editor Proof

335 With the use of the Leverrier-Barnett algorithm, see e.g. Kirillov (2013a), one can  
 336 write the characteristic polynomial of the system (23) as

$$337 \quad p(\lambda) = \lambda^{2m} + \text{tr}\mathbf{K}\lambda^{2m-2} + \frac{1}{2}((\text{tr}\mathbf{K})^2 - \text{tr}\mathbf{K}^2 - \text{tr}\mathbf{N}^2)\lambda^{2m-4} + \dots \quad (26)$$

338 Since for the polynomial (26) we have  $\det \Delta_1 = m > 0$ , then, Gallina criterion gives  
 339 a sufficient condition for instability if

$$340 \quad a_0^2((a_1^2 - a_2a_0)m - a_1^2) < 0. \quad (27)$$

341 With the explicit expressions for the coefficients of the polynomial from (26), we  
 342 re-write (27) as

$$343 \quad \text{tr}\mathbf{K}^2 + \text{tr}\mathbf{N}^2 < \frac{1}{m}(\text{tr}\mathbf{K})^2. \quad (28)$$

Taking into account that

$$\text{tr}\mathbf{K}^2 = \text{tr}(\mathbf{K}^T \mathbf{K}) = \|\mathbf{K}\|^2,$$

and

$$\text{tr}\mathbf{N}^2 = \text{tr}(-\mathbf{N}^T \mathbf{N}) = -\|\mathbf{N}\|^2,$$

344 where the norm is understood as the Frobenius norm, we represent (28) in the form

$$345 \quad \|\mathbf{N}\|^2 > \|\mathbf{K}\|^2 - \frac{1}{m}(\text{tr}\mathbf{K})^2. \quad (29)$$

346 The inequality (29) is known as the *Bulatovic flutter condition*.

**Theorem 2.3** (Bulatovic flutter condition Bulatovic 2011, 2017) *If*

$$\|\mathbf{N}\|^2 > \|\mathbf{K}\|^2 - \frac{1}{m}(\text{tr}\mathbf{K})^2$$

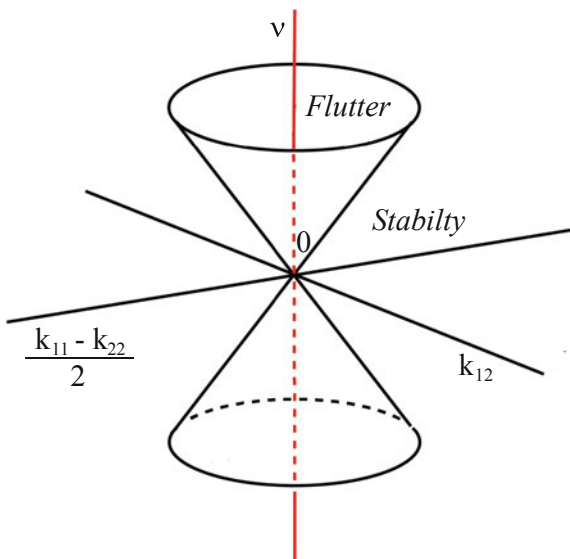
*the equilibrium of the circulatory system*

$$\ddot{\mathbf{x}} + (\mathbf{K} + \mathbf{N})\mathbf{x} = \mathbf{F}(\mathbf{x}, \dot{\mathbf{x}}),$$

347 where  $\mathbf{K} = \mathbf{K}^T$ ,  $\mathbf{N} = -\mathbf{N}^T$ , and  $\mathbf{F}$  is a collection of terms of no lower than second  
 348 order, is unstable.

349 In a particular case when the stiffness matrix is proportional to the identity matrix,  
 350  $\mathbf{K} = \kappa\mathbf{I}$ , we have  $\text{tr}\mathbf{K} = \kappa m$  and  $\text{tr}\mathbf{K}^2 = \|\mathbf{K}\|^2 = \kappa^2 m$ . With this, the flutter condi-  
 351 tion (29) reduces to the inequality  $\|\mathbf{N}\|^2 > 0$ , which is always fulfilled if  $\|\mathbf{N}\| \neq 0$ .  
 352 Instability in this degenerate case occurs at arbitrary small circulatory forces. This  
 353 statement is the famous Merkin theorem, see e.g. Krechetnikov and Marsden (2007);  
 354 Udwardia (2017).

**Fig. 10** Geometrical interpretation of the Bulatovic flutter condition and Merkin theorem for  $m = 2$  degrees of freedom



355 **Theorem 2.4** (Merkin Theorem (Merkin 1956)) *Perturbation by arbitrary linear*  
 356 *circulatory forces of a stable pure potential system with eigenfrequencies coinciding*  
 357 *into one with the algebraic multiplicity equal to the dimension of the system destroys*  
 358 *the stability of the equilibrium regardless of the form of the nonlinear terms.*

359 **2.4 Geometrical Interpretation for  $m = 2$  Degrees of**  
 360 **Freedom**

361 Let us now assume that  $m = 2$  in Eq. (23). Notice that the  $2 \times 2$  matrix  $\mathbf{A} = \mathbf{K} + \mathbf{N}$   
 362 has the following decomposition

363 
$$\mathbf{A} = \frac{k_{11} + k_{22}}{2} \begin{pmatrix} 1 & 0 \\ 0 & 1 \end{pmatrix} + \frac{1}{2} \begin{pmatrix} k_{11} - k_{22} & 2k_{12} \\ 2k_{12} & k_{22} - k_{11} \end{pmatrix}$$
  
 364 
$$+ \begin{pmatrix} 0 & -\nu \\ \nu & 0 \end{pmatrix} = \mathbf{C} + \mathbf{H} + \mathbf{N}, \tag{30}$$

365 where the matrix  $\mathbf{C}$  corresponds to potential forces of spherical type,  $\mathbf{H}$  to potential  
 366 forces of hyperbolic type, and  $\mathbf{N}$  to circulatory forces (Zhuravlev 2007, 2008). When  
 367  $\mathbf{H} = 0$  we are in the conditions of the Merkin theorem.

368 Calculating the eigenvalues of the corresponding eigenvalue problem, which are  
 369 the roots of the polynomial  $\det(\lambda^2 \mathbf{I} + \mathbf{A})$ , we find

$$\lambda^2 = -\frac{k_{11} + k_{22}}{2} \pm \frac{1}{2}\sqrt{(k_{11} - k_{22})^2 + 4k_{12}^2 - 4\nu^2}. \tag{31}$$

which is complex (flutter) if

$$\nu^2 > \frac{(k_{11} - k_{22})^2}{4} + k_{12}^2. \tag{32}$$

This condition determines an interior of a double cone in the space of parameters  $\frac{k_{11}-k_{22}}{2}$ ,  $k_{12}$ , and  $\nu$ , see Fig. 10.

Let us establish a connection between the stability diagram of Fig. 10 and already known to us Bulatovic’s flutter condition and Merkin’s theorem.

Observing that

$$\|\mathbf{K}\|^2 = k_{11}^2 + k_{22}^2 + 2k_{12}^2, \quad (\text{tr}\mathbf{K})^2 = (k_{11} + k_{22})^2, \quad \|\mathbf{N}\|^2 = 2\nu^2$$

we find

$$\|\mathbf{K}\|^2 - \frac{1}{2}(\text{tr}\mathbf{K})^2 = \frac{(k_{11} - k_{22})^2}{2} + 2k_{12}^2.$$

Hence,

$$\nu^2 > \frac{(k_{11} - k_{22})^2}{4} + k_{12}^2 \Leftrightarrow \|\mathbf{N}\|^2 > \|\mathbf{K}\|^2 - \frac{1}{m}(\text{tr}\mathbf{K})^2.$$

and we establish the equivalence of the Bulatovic flutter condition (29) and (32). Therefore, the Bulatovic flutter condition determines the conical flutter domain in Fig. 10. The axis of the cone passing through the origin at  $k_{11} - k_{22} = 0$ ,  $k_{12} = 0$  and  $\nu = 0$  lies in the flutter domain, corresponding to the condition  $\nu^2 > 0$  or  $\|\mathbf{N}\|^2 > 0$  given by the Merkin theorem.

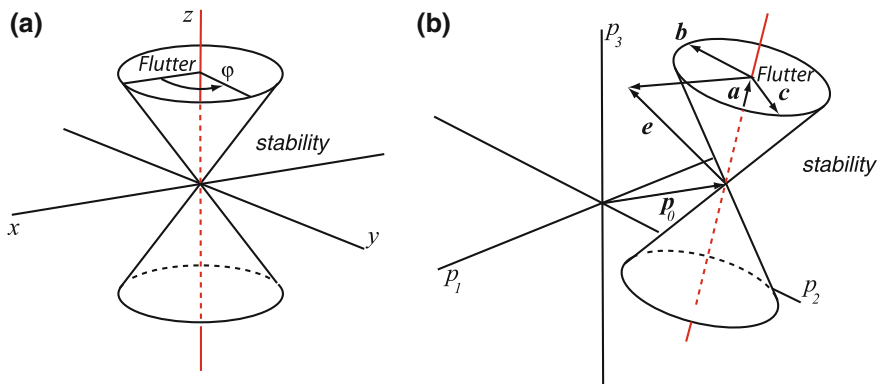
The apex of the cone at  $k_{11} - k_{22} = 0$ ,  $k_{12} = 0$  and  $\nu = 0$  corresponds to the potential system under the action of potential forces of spherical type, which is stable. Potential forces of spherical type and circulatory forces imply Merkin’s instability at all values of  $\nu \neq 0$ . Potential forces of hyperbolic type stabilize the Merkin-unstable system at  $\nu < \nu_{cr} = \frac{(k_{11}-k_{22})^2}{4} + k_{12}^2$ . This is equivalent to the finite threshold for a torque in the Nicolai shaft with a non-circular cross-section (Nicolai 1929; Ziegler 1951a, b; Bolotin 1963; Seyranian and Mailybaev 2011; Luongo and Ferretti 2016).

## 2.5 Approximating Flutter Cone by Perturbation of Eigenvalues

Consider the matrix  $\mathbf{A}$  defined by Eq.(30) as a function of three parameters  $\mathbf{A} = \mathbf{A}(k_{22}, k_{12}, \nu)$ , whereas the parameter  $k_{11}$  is fixed, and the eigenvalue problem for it

$$\mathbf{A}(k_{22}, k_{12}, \nu)\mathbf{u} = \sigma\mathbf{u}, \tag{33}$$





**Fig. 11** Conical flutter domain of a circulatory system in the vicinity of a point in the parameter space corresponding to a semi-simple eigenvalue of the matrix  $\mathbf{A} = \mathbf{K} + \mathbf{N}$ . **a** Given by Eq. (39). **b** Given by Eq. (40)

394 where  $\sigma = -\lambda^2$ .

395 Let  $\mathbf{A}_0 = \mathbf{A}(k_{22} = k_{11}, k_{12} = 0, \nu = 0)$ . Then

396 
$$\mathbf{A}_0 = \begin{pmatrix} k_{11} & 0 \\ 0 & k_{11} \end{pmatrix}. \tag{34}$$

397 This matrix has a *semi-simple* real eigenvalue  $\sigma_0 = k_{11}$  with the two linearly-  
 398 independent right eigenvectors  $\mathbf{u}_1$  and  $\mathbf{u}_2$  and two linearly-independent left eigen-  
 399 vectors,  $\mathbf{v}_1$  and  $\mathbf{v}_2$ . In general, left and right eigenvectors of a non-symmetric matrix  
 400 differ but in our example  $\mathbf{A}_0$  is real and symmetric and we can choose

401 
$$\mathbf{u}_1 = \mathbf{v}_1 = \begin{pmatrix} 0 \\ 1 \end{pmatrix}, \quad \mathbf{u}_2 = \mathbf{v}_2 = \begin{pmatrix} 1 \\ 0 \end{pmatrix}. \tag{35}$$

402 Let us introduce the vector of parameters  $\mathbf{p} = (k_{22}, k_{12}, \nu)$  and denote  $\mathbf{p}_0 =$   
 403  $(k_{11}, 0, 0)$ . Then,  $\mathbf{A}(k_{22}, k_{12}, \nu) = \mathbf{A}(\mathbf{p})$  and  $\mathbf{A}_0 = \mathbf{A}(\mathbf{p}_0)$ .

In the following, we briefly consider a perturbative approach to the study of stability of circulatory systems following (Kirillov 2010, 2013a). We introduce a scalar parameter  $\varepsilon$  and consider a smooth path in the parameter space  $\mathbf{p}(\varepsilon)$  and consider it in the vicinity of  $\mathbf{p}_0 = \mathbf{p}(\varepsilon = 0)$

$$\mathbf{p}(\varepsilon) = \mathbf{p}_0 + \varepsilon \frac{d\mathbf{p}}{d\varepsilon} + o(\varepsilon).$$

Then, the matrix family  $\mathbf{A}(\mathbf{p}(\varepsilon))$  takes an increment

$$\mathbf{A}(\mathbf{p}(\varepsilon)) = \mathbf{A}_0 + \varepsilon \mathbf{A}_1 + o(\varepsilon),$$

404 where  $\mathbf{A}_1 = \sum_{s=1}^n \frac{\partial \mathbf{A}}{\partial p_s} \frac{dp_s}{d\varepsilon}$ . In our example  $n = 3$ ,  $p_1 = k_{22}$ ,  $p_2 = k_{12}$ , and  $p_3 = \nu$ .

405 It can be shown by perturbation argument (Kirillov 2010, 2013a) that the double  
 406 semi-simple eigenvalue  $\sigma_0$  splits into two simple eigenvalues as follows

$$407 \quad \sigma(\varepsilon) = \sigma_0 + \varepsilon \frac{(\mathbf{A}_1 \mathbf{u}_1, \mathbf{v}_1) + (\mathbf{A}_1 \mathbf{u}_2, \mathbf{v}_2)}{2} \pm \frac{\varepsilon}{2} \sqrt{D} + o(\varepsilon), \quad (36)$$

408 where  $D = x^2 + y^2 - z^2$  and

$$409 \quad x = \langle \mathbf{f}_*, \mathbf{e} \rangle, \quad y = \langle \mathbf{f}_+, \mathbf{e} \rangle, \quad z = \langle \mathbf{f}_-, \mathbf{e} \rangle. \quad (37)$$

410 The vector  $\mathbf{e} = \left( \frac{dp_1}{d\varepsilon}, \dots, \frac{dp_n}{d\varepsilon} \right)^T$ . The components of the vectors  $\mathbf{f}_*$ ,  $\mathbf{f}_+$  and  $\mathbf{f}_-$   
 411 are given by the expressions

$$412 \quad f_{*,s} = (\partial_{p_s} \mathbf{A} \mathbf{u}_1, \mathbf{v}_1) - (\partial_{p_s} \mathbf{A} \mathbf{u}_2, \mathbf{v}_2),$$

$$413 \quad f_{\pm,s} = (\partial_{p_s} \mathbf{A} \mathbf{u}_1, \mathbf{v}_2) \pm (\partial_{p_s} \mathbf{A} \mathbf{u}_2, \mathbf{v}_1). \quad (38)$$

414 The brackets  $\langle \cdot, \cdot \rangle$  in (37) denote the inner product of vectors in  $n$ -dimensional space  
 415 and the brackets  $(\cdot, \cdot)$  in (38) denote the inner product of vectors in  $m$ -dimensional  
 416 space. Recall that in our example  $m = 2$  and  $n = 3$ .

417 The perturbed eigenvalues (36) are complex if

$$418 \quad z^2 > x^2 + y^2, \quad (39)$$

419 that is, inside the conical surface in the  $(x, y, z)$ -space, see Fig. 11 a.

In order to describe this conical flutter domain in the space of parameters  $\mathbf{p}$ , we  
 introduce the vectors

$$\mathbf{a} = \mathbf{f}_* \times \mathbf{f}_+, \quad \mathbf{b} = \mathbf{f}_* \times \mathbf{f}_-, \quad \mathbf{c} = \mathbf{f}_- \times \mathbf{f}_+$$

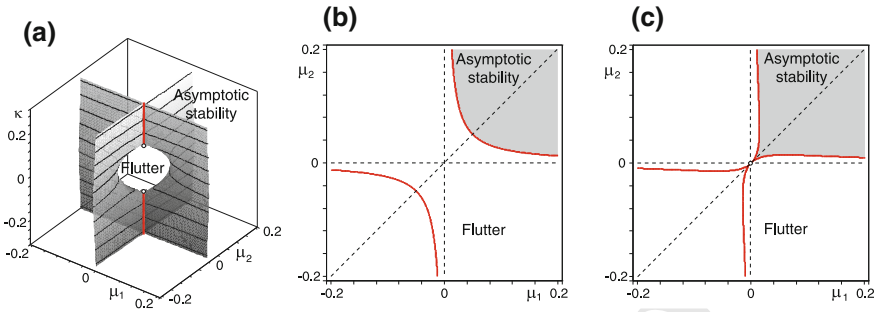
420 and the polar angle  $\varphi$  through the relations  $x = z \cos \varphi$  and  $y = z \sin \varphi$ . Then we  
 421 can describe the flutter cone at the point  $\mathbf{p}_0$  as the *tangent cone* to the flutter domain,  
 422 i.e. as a set of directions  $\mathbf{e}$  in which from the given point one can send a curve that  
 423 lies in the flutter domain:

$$424 \quad \{\mathbf{e} : \mathbf{e} = t(\mathbf{a} + d(\mathbf{b} \sin \phi + \mathbf{c} \cos \phi)), t \in \mathbb{R}, \varphi \in [0, 2\pi], d \in [0, 1)\}. \quad (40)$$

Taking into account the eigenvectors (35) of the matrix (34) and constructing the  
 gradient vector

$$\mathbf{e} = \begin{pmatrix} k_{22} - k_{11} \\ k_{12} \\ \nu \end{pmatrix}$$

we find the vectors



**Fig. 12** Flutter instability of the shaft (41) at weak damping and weak stiffness anisotropy for  $k_1 = 1$ ,  $m = 1$  and  $\beta = 0.05$ . **a** Stability domain (42) with two Whitney umbrella singular points in the  $(\mu_1, \mu_2, \kappa)$ -space. **b** Instability at weak damping and zero stiffness anisotropy ( $\kappa = 0$ ). **c** Stabilization by weak damping at large stiffness anisotropy ( $\kappa = 2\beta = 0.1$ )

$$\mathbf{f}_* = \begin{pmatrix} 1 \\ 0 \\ 0 \end{pmatrix}, \quad \mathbf{f}_+ = \begin{pmatrix} 0 \\ 2 \\ 0 \end{pmatrix}, \quad \mathbf{f}_- = \begin{pmatrix} 0 \\ 0 \\ -2 \end{pmatrix}.$$

Substituting these vectors into the flutter condition

$$\langle \mathbf{f}_*, \mathbf{e} \rangle^2 + \langle \mathbf{f}_+, \mathbf{e} \rangle^2 - \langle \mathbf{f}_-, \mathbf{e} \rangle^2 < 0,$$

we reproduce the flutter cone (32).

Note that the conical singularity is one of the eight generic singularities of codimension 3 that can occur on stability boundaries of circulatory systems with at least three parameters (Kirillov 2013a). In case of two parameters the number of generic singular points reduces to four, and in one-parameter families of circulatory systems we have only two singular points, corresponding to the reversible-Hopf bifurcation and to the steady-state bifurcation shown in Figs. 6 and 7, respectively.

### 3 Perturbing Circulatory Systems

#### 3.1 Shieh–Masur Shaft with Dissipative Forces

Let us return to the model (20) of a rotating shaft by Shieh and Masur (1968) in the case when the shaft is non-rotating ( $\Omega = 0$ ) and take into account damping

$$\begin{aligned} m\ddot{u} + \mu_1\dot{u} + k_1u + \beta v &= 0 \\ m\ddot{v} + \mu_2\dot{v} + k_2v - \beta u &= 0 \end{aligned} \tag{41}$$

438 Separating time with  $u = \tilde{u}e^{\lambda t}$  and  $v = \tilde{v}e^{\lambda t}$  and applying to the characteristic  
 439 polynomial of the resulting system of two algebraic equations the Hurwitz stability  
 440 criterion, we find that the trivial solution  $u = 0, v = 0$  is stable asymptotically, if  
 441 and only if

$$442 \quad (\mu_1 + \mu_2)^2(\mu_1\mu_2k_1 - m\beta^2) + \mu_1\mu_2\kappa(\kappa m + \mu_1(\mu_1 + \mu_2)) > 0,$$

$$443 \quad \mu_1 + \mu_2 > 0. \quad (42)$$

444 The stability conditions (42) ensure the exponential decay with time of all no-trivial  
 445 solutions  $u(t)$  and  $v(t)$  of the Eq. (41).

446 The conditions (42) have a complicated form in contrast to the undamped case  
 447 corresponding to  $\mu_1 = 0$  and  $\mu_2 = 0$  when the shaft is stable at  $\beta^2 < \kappa^2/4$ . How the  
 448 damped and undamped cases are connected? Does the undamped flutter condition  
 449 always follow from the damped one in the limit of vanishing damping coefficients?  
 450 Let us investigate.

451 Equate the left side of Eq.(42)<sub>1</sub> to zero and solve the resulting equation with  
 452 respect to  $\kappa$ . Then assume in the result  $\mu_1 = b\mu_2$  and consider its limit as  $\mu_2 \rightarrow 0$ .  
 453 This yields

$$454 \quad \kappa(b) = \pm\beta \left( \sqrt{b} + \frac{1}{\sqrt{b}} \right), \quad b = \frac{\mu_1}{\mu_2}. \quad (43)$$

455 The function  $\kappa(b)$  has a minimum equal to  $2\beta$  and a maximum equal to  $-2\beta$  at  $b = 1$ .  
 456 This means that the threshold of stability of the dissipative system coincides with the  
 457 threshold of the undamped system ( $\kappa^2 = 4\beta^2$ ) in the limit of vanishing dissipation  
 458 only if  $\mu_1 = \mu_2$ , or  $b = \mu_1/\mu_2 = 1$ .

459 Let us expand  $\kappa(b)$  in a Taylor series in the vicinity of  $b = 1$

$$460 \quad \kappa = \pm 2\beta \pm \beta \frac{(b-1)^2}{4} + o((b-1)^2). \quad (44)$$

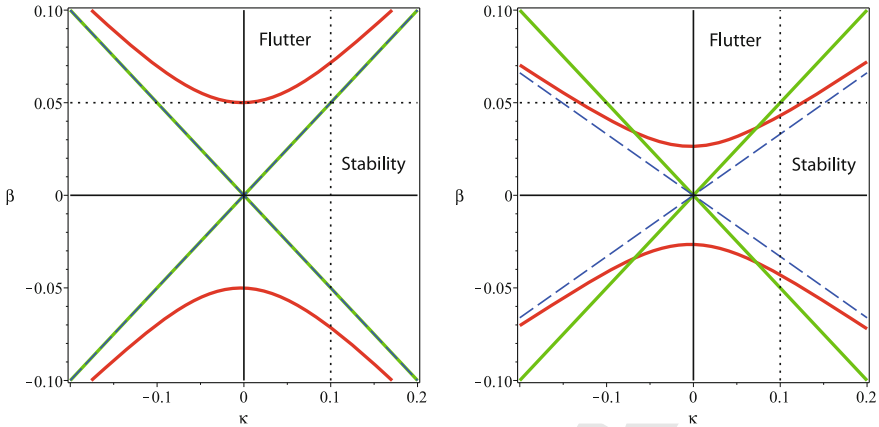
461 Truncating the series and taking into account that  $b = \mu_1/\mu_2$ , we write

$$462 \quad \kappa(\mu_1, \mu_2) = \pm 2\beta \pm \beta \frac{(\mu_1 - \mu_2)^2}{4\mu_2^2}. \quad (45)$$

463 Equation (45) is in the form  $zy^2 = x^2$ , which is the normal form of a surface in the  
 464  $Oxyz$ -space that has the Whitney umbrella singular point at the origin (Bottema 1956;  
 465 Arnold 1972; Langford 2003; Kirillov and Verhulst 2010). The function  $z(x, y) =$   
 466  $x^2/y^2 > 0$  at all  $x, y$  except for the specific line  $x = 0$ , where  $z(0, y) = 0$ .

467 In our case the line  $x = 0$  is the line  $\mu_1 = \mu_2$  in the  $(\mu_1, \mu_2)$ -plane, see Fig. 12  
 468 where the stability domain (42) is shown with the two Whitney umbrella singu-  
 469 lar points situated on the  $\kappa$ -axis at  $\kappa = \pm 2\beta$ . It is remarkable that a weak stiff-  
 470 ness anisotropy in the presence of weak damping does not prevent the system from  
 471 flutter when circulatory forces are acting, Fig. 12b. Indeed, at  $\kappa = 0$  criterion (42)

Editor Proof



**Fig. 13** Stability map of the rotating shaft with  $k_1 = 1$ ,  $m = 1$  (green lines) without dissipation and (red curves) with dissipation when (left) dissipation coefficients are equal,  $\mu_1 = \mu_2 = 0.05$ , (right) when  $\mu_1 = 0.07$  and  $\mu_2 = 0.01$ . The asymptotic dashed lines are given by Eqs. (48) and (49), respectively

472 yields stability beyond a hyperbolic branch in the first quadrant of the  $(\mu_1, \mu_2)$ -  
 473 plane

474 
$$\mu_1 \mu_2 k_1 - m \beta^2 > 0, \tag{46}$$

475 at some distance from the origin. Notice that stability condition (46) traces back to  
 476 Kapitsa (1939) who derived it in his study of transition to supercritical speeds in a  
 477 special high-frequency expansion turbine that he developed for liquefaction of air.

478 As soon as the absolute value of the stiffness anisotropy increases, the stability  
 479 domain comes closer to the origin and touches it in a cuspidal point exactly when  
 480  $\kappa = \pm 2\beta$ , Fig. 12c, i.e. at the Whitney umbrella singular points. We observe that  
 481 at  $\kappa = \pm 2\beta$  there exists only one direction pointing to the stability domain from  
 482 the origin, and this direction is along the line  $\mu_1 = \mu_2$ , in full agreement with (45).  
 483 Decreasing dissipation along this line yields tending the critical flutter load smoothly  
 484 to its values  $\kappa = \pm 2\beta$  for the undamped shaft. However, this is not true for all other  
 485 directions, i.e. damping ratios  $b = \mu_1/\mu_2$  different from 1.

In fact, near the Whitney umbrella points the stability boundary behaves much like a ruled surface, which has exactly two rulers  $\mu_1 = b_{\pm} \mu_2$ , where

$$b_{\pm} = 1 + \frac{\kappa^2 - 4\beta^2}{2\beta^2} \pm \kappa \frac{\sqrt{\kappa^2 - 4\beta^2}}{2\beta^2}$$

486 at every  $\kappa$  such that  $\kappa^2 > 4\beta^2$ . Consequently, tending damping to zero along either  
 487 of the two directions,  $\mu_1 = b_{\pm} \mu_2$ , will result in the value of  $\kappa$  that does not coincide  
 488 with the undamped values  $\pm 2\beta$ . The flutter load of the damped shaft has therefore  
 489 a singular zero-dissipation limit at the Whitney umbrella points. At every damping

490 ratio, except for 1, the flutter load in the limit of vanishing dissipation differs by  
 491 a finite value from the flutter load of the undamped system. This is the famous  
 492 *Ziegler–Bottema destabilization paradox* (Ziegler 1952; Bottema 1956).

493 Now we are prepared to answer how dissipation affects the conical flutter domain  
 494 of the undamped shaft given by the Bulatovich flutter condition that is shown in  
 495 Fig. (9)(right). From (43) an expression for the two lines in the  $(\kappa, \beta)$ -plane follows

$$496 \quad \beta = \pm \frac{\sqrt{\mu_1 \mu_2}}{\mu_1 + \mu_2} \kappa. \tag{47}$$

497 The slope of the lines depends on the damping ratio in the manner dictated by the  
 498 ruled surface geometry near the Whitney umbrella singularities. Indeed, for equal  
 499 damping coefficients,  $\mu_1 = \mu_2$ , the lines (47) are

$$500 \quad \beta = \pm \frac{\sqrt{\mu_1 \mu_2}}{\mu_1 + \mu_2} \kappa = \pm \frac{1}{2} \kappa. \tag{48}$$

501 They coincide with the flutter boundaries of the undamped system, Fig. 13(left). If  
 502 we plot the stability domain (42) in the  $(\kappa, \beta)$ -plane for different damping coeffi-  
 503 cients that satisfy the constraint  $\mu_1 = \mu_2$ , we will see that the stability boundary is  
 504 a hyperbolic curve with the asymptotes (48). In the limit of vanishing dissipation  
 505 such that  $\mu_1 = \mu_2$  the stability boundary of the dissipative system degenerates into  
 506 the cone  $\kappa^2 = 4\beta^2$ .

507 However, taking the limit of vanishing dissipation at any other constraint on the  
 508 damping coefficients, say,  $\mu_1 = 7\mu_2$ , results in the different conical domain with the  
 509 boundaries

$$510 \quad \beta = \pm \frac{\sqrt{\mu_1 \mu_2}}{\mu_1 + \mu_2} \kappa = \pm \frac{\sqrt{7}}{6} \kappa. \tag{49}$$

511 The flutter domain in the limit of vanishing dissipation given by the inequality  $36\beta^2 >$   
 512  $7\kappa^2$  is therefore larger than the flutter domain of the undamped shaft,  $\kappa^2 < 4\beta^2$ ,  
 513 Fig. 13(right), providing an instructive example of a dissipation-induced instability  
 514 (Bloch et al. 1994; Krechetnikov and Marsden 2007).

### 515 3.2 A Circulatory System Perturbed by Dissipative Forces

516 The Shieh and Masur (1968) shaft is a non-conservative system with two degrees of  
 517 freedom illustrating the properties summarized in the remark by Leipholz (1987):  
 518 “*Independent works of Bottema (1956) and Bolotin (1963) for second-order systems*  
 519 *has shown that in the non-conservative case and for different damping coefficients*  
 520 *the stability condition is discontinuous with respect to the undamped case.*”

Editor Proof

521 Let us build a general theory proving this effect in a finite-dimensional mechanical system of *arbitrary order* under the action of positional conservative forces  
 522 represented by a real symmetric matrix  $\mathbf{K} = \mathbf{K}^T$  and positional non-conservative (or  
 523 circulatory) forces with the real skew-symmetric matrix  $\mathbf{N} = -\mathbf{N}^T$ :  
 524

$$525 \quad \mathbf{M}\ddot{\mathbf{x}} + (\mathbf{K} + \mathbf{N}(q))\mathbf{x} = 0. \quad (50)$$

526 The matrix of circulatory forces smoothly depends on a parameter  $q$ .

527 Assuming solution to the problem (50) in the form  $\mathbf{x} = \mathbf{u} \exp \lambda t$ , we arrive at the  
 528 eigenvalue problem

$$529 \quad \mathbf{L}(\lambda, q)\mathbf{u} := (\mathbf{K} + \mathbf{N}(q))\mathbf{u} + \lambda^2\mathbf{M}\mathbf{u} = 0. \quad (51)$$

530 Let at the value of the parameter  $q = q_0$  there exist an algebraically double imaginary  
 531 eigenvalue  $\lambda_0 = i\omega_0$  with the Jordan block that satisfies the following equations

$$532 \quad (\mathbf{K} + \mathbf{N}(q_0))\mathbf{u}_0 - \omega_0^2\mathbf{M}\mathbf{u}_0 = 0$$

$$533 \quad (\mathbf{K} + \mathbf{N}(q_0))\mathbf{u}_1 - \omega_0^2\mathbf{M}\mathbf{u}_1 = -2i\omega_0\mathbf{M}\mathbf{u}_0, \quad (52)$$

534 where  $\mathbf{u}_0$  is an eigenfunction and  $\mathbf{u}_1$  is an associated function at  $\lambda_0$ .

535 Note that the eigenfunction  $\mathbf{v}_0$  and the associated function  $\mathbf{v}_1$  at the eigenvalue  
 536  $\bar{\lambda}_0 = -i\omega_0$  are governed by the adjoint equations

$$537 \quad (\mathbf{K} - \mathbf{N}(q_0))\mathbf{v}_0 - \omega_0^2\mathbf{M}\mathbf{v}_0 = 0$$

$$538 \quad (\mathbf{K} - \mathbf{N}(q_0))\mathbf{v}_1 - \omega_0^2\mathbf{M}\mathbf{v}_1 = 2i\omega_0\mathbf{M}\mathbf{v}_0. \quad (53)$$

539 Let us perturb the parameter  $q$  in the vicinity of  $q_0$  as

$$540 \quad q(\varepsilon) = q_0 + \varepsilon q_1 + o(\varepsilon^2). \quad (54)$$

541 Then,

$$542 \quad \mathbf{N}(q(\varepsilon)) = \mathbf{N}(q_0) + \varepsilon\mathbf{N}_1 + o(\varepsilon) \quad (55)$$

543 where  $\mathbf{N}_1 = \left. \frac{\partial \mathbf{N}}{\partial q} \frac{dq}{d\varepsilon} \right|_{\varepsilon=0}$  and

$$544 \quad \lambda(\varepsilon) = \lambda_0 + \lambda_1\varepsilon^{1/2} + \lambda_2\varepsilon + o(\varepsilon),$$

$$545 \quad \mathbf{u}(\varepsilon) = \mathbf{u}_0 + \mathbf{z}_1\varepsilon^{1/2} + \mathbf{z}_2\varepsilon + o(\varepsilon). \quad (56)$$

546 Substituting the expansions (55) and (56) into (51), we get

$$547 \quad (\mathbf{K} + \mathbf{N}(q_0) + \varepsilon\mathbf{N}_1 + o(\varepsilon))(\mathbf{u}_0 + \mathbf{z}_1\varepsilon^{1/2} + \mathbf{z}_2\varepsilon + o(\varepsilon))$$

$$548 \quad + (\lambda_0^2 + 2\varepsilon^{1/2}\lambda_0\lambda_1 + \varepsilon(2\lambda_0\lambda_2 + \lambda_1^2) + o(\varepsilon))\mathbf{M}(\mathbf{u}_0 + \mathbf{z}_1\varepsilon^{1/2} + \mathbf{z}_2\varepsilon + o(\varepsilon))$$

$$549 \quad = 0. \quad (57)$$

550 Collecting terms at  $\varepsilon^0$ ,  $\varepsilon^{1/2}$ , and  $\varepsilon^1$  we obtain the equations

$$\begin{aligned}
 551 \quad & (\mathbf{K} + \mathbf{N}(q_0))\mathbf{u}_0 + \lambda_0^2 \mathbf{M}\mathbf{u}_0 = 0 \\
 552 \quad & (\mathbf{K} + \mathbf{N}(q_0))\mathbf{z}_1 + \lambda_0^2 \mathbf{M}\mathbf{z}_1 = -2\lambda_0 \mathbf{M}\lambda_1 \mathbf{u}_0 \\
 553 \quad & (\mathbf{K} + \mathbf{N}(q_0))\mathbf{z}_2 + \lambda_0^2 \mathbf{M}\mathbf{z}_2 = -2\lambda_0 \lambda_1 \mathbf{M}\mathbf{z}_1 - \mathbf{N}_1 \mathbf{u}_0 - (2\lambda_0 \lambda_2 + \lambda_1^2) \mathbf{M}\mathbf{u}_0.
 \end{aligned} \tag{58}$$

554 Let  $(\mathbf{a}, \mathbf{b}) = \bar{\mathbf{b}}^T \mathbf{a}$  be an inner product of vectors  $\mathbf{a}$  and  $\mathbf{b}$ . Taking the inner product  
 555 of the last of the Eq. (58) with the vector  $\mathbf{v}_0$ , we find

$$\begin{aligned}
 556 \quad & ((\mathbf{K} + \mathbf{N}(q_0))\mathbf{z}_2, \mathbf{v}_0) + \lambda_0^2 (\mathbf{M}\mathbf{z}_2, \mathbf{v}_0) = -2\lambda_0 \lambda_1 (\mathbf{M}\mathbf{z}_1, \mathbf{v}_0) - (\mathbf{N}_1 \mathbf{u}_0, \mathbf{v}_0) \\
 557 \quad & \quad \quad \quad - (2\lambda_0 \lambda_2 + \lambda_1^2) (\mathbf{M}\mathbf{u}_0, \mathbf{v}_0).
 \end{aligned} \tag{59}$$

558 In view of the property  $(\mathbf{L}\mathbf{u}, \mathbf{v}) = (\mathbf{u}, \mathbf{L}^\dagger \mathbf{v})$ , where the adjoint matrix polynomial  
 559 is just  $\mathbf{L}^\dagger = \mathbf{K} - \mathbf{N} + \bar{\lambda}^2 \mathbf{M}$ , and taking into account that  $\mathbf{L}^\dagger \mathbf{v}_0 = 0$ , we find

$$560 \quad 2\lambda_0 \lambda_1 (\mathbf{M}\mathbf{z}_1, \mathbf{v}_0) + (\mathbf{N}_1 \mathbf{u}_0, \mathbf{v}_0) + (2\lambda_0 \lambda_2 + \lambda_1^2) (\mathbf{M}\mathbf{u}_0, \mathbf{v}_0) = 0. \tag{60}$$

561 Observing that  $\mathbf{z}_1 = \lambda_1 \mathbf{u}_1 + C_1 \mathbf{u}_0$  and  $(\mathbf{M}\mathbf{u}_0, \mathbf{v}_0) = 0$  we arrive at the equation

$$562 \quad 2\lambda_0 \lambda_1^2 (\mathbf{M}\mathbf{u}_1, \mathbf{v}_0) + (\mathbf{N}_1 \mathbf{u}_0, \mathbf{v}_0) = 0. \tag{61}$$

563 Hence,

$$564 \quad \lambda_1^2 = \frac{i (\mathbf{N}_1 \mathbf{u}_0, \mathbf{v}_0)}{2\omega_0 (\mathbf{M}\mathbf{u}_1, \mathbf{v}_0)}. \tag{62}$$

565 In these conditions with  $\varepsilon \mathbf{N}_1 = \left. \frac{\partial \mathbf{N}}{\partial q} \frac{dq}{d\varepsilon} \right|_{\varepsilon=0}$   $\varepsilon = \left. \frac{\partial \mathbf{N}}{\partial q} \right|_{q=q_0} \Delta q = \mathbf{N}'_q \Delta q$  we obtain

$$566 \quad \lambda(q) = i\omega_0 \pm \sqrt{\Delta q \frac{i (\mathbf{N}'_q \mathbf{u}_0, \mathbf{v}_0)}{2\omega_0 (\mathbf{M}\mathbf{u}_1, \mathbf{v}_0)}} + o(\sqrt{|\Delta q|}), \tag{63}$$

$$568 \quad \mathbf{u}(q) = \mathbf{u}_0 \pm \mathbf{u}_1 \sqrt{\Delta q \frac{i (\mathbf{N}'_q \mathbf{u}_0, \mathbf{v}_0)}{2\omega_0 (\mathbf{M}\mathbf{u}_1, \mathbf{v}_0)}} + o(\sqrt{|\Delta q|}), \tag{64}$$

$$570 \quad \mathbf{v}(q) = \mathbf{v}_0 \pm \mathbf{v}_1 \sqrt{\Delta q \frac{i (\mathbf{N}'_q \mathbf{u}_0, \mathbf{v}_0)}{2\omega_0 (\mathbf{M}\mathbf{u}_1, \mathbf{v}_0)}} + o(\sqrt{|\Delta q|}). \tag{65}$$

571 Therefore, we have approximations to the eigenvalues and eigenvectors of the  
 572 undamped circulatory system in the vicinity of  $q = q_0$ , i.e. in the vicinity of the  
 573 flutter boundary corresponding to the reversible-Hopf bifurcation.

574 Assume that at  $q < q_0$  the eigenvalues of the circulatory system are imaginary  
 575 and at  $q > q_0$  the eigenvalues are complex-conjugate (instability).



576 Let us study how simple imaginary eigenvalues of a circulatory system change  
 577 due to dissipative perturbation with the matrix  $\mathbf{D}(\mathbf{p})$  where  $\mathbf{p} = (p_1, p_2, \dots, p_n)^T$   
 578 and  $\mathbf{D}(\mathbf{p} = 0) = 0$ . Write the dissipatively perturbed eigenvalue problem (51)

$$579 \quad \mathbf{L}(\lambda, q, \mathbf{p})\mathbf{u} := (\mathbf{K} + \mathbf{N}(q))\mathbf{u}(q) + \lambda(q)\mathbf{D}(\mathbf{p})\mathbf{u}(q) + \lambda^2(q)\mathbf{M}\mathbf{u}(q) = 0. \quad (66)$$

580 as well as its adjoint

$$581 \quad \mathbf{L}^\dagger(\lambda, q, \mathbf{p})\mathbf{v} := (\mathbf{K} - \mathbf{N}(q))\mathbf{v}(q) + \bar{\lambda}(q)\mathbf{D}(\mathbf{p})\mathbf{v}(q) + \bar{\lambda}^2(q)\mathbf{M}\mathbf{v}(q) = 0. \quad (67)$$

582 We assume in the above equations that  $q$  is fixed such that  $q < q_0$ .

583 Let at  $\mathbf{p} = 0$  the eigenvalue problem (66) has a simple eigenvalue  $\lambda(q) = i\omega(q)$   
 584 with an eigenvector  $\mathbf{u}(q)$ . Assuming  $\mathbf{p} = \mathbf{p}(\varepsilon)$ , where  $\mathbf{p}(\varepsilon) = \varepsilon\mathbf{p}_1 + o(\varepsilon)$ , we obtain

$$585 \quad \mathbf{D}(\mathbf{p}(\varepsilon)) = \varepsilon\mathbf{D}_1 + o(\varepsilon) \quad (68)$$

586 with  $\mathbf{D}_1 = \sum_{s=1}^n \frac{\partial \mathbf{D}}{\partial p_s} \frac{dp_s}{d\varepsilon} \Big|_{\varepsilon=0}$ . Then, the eigenvalues of (66) are

$$587 \quad \lambda(\varepsilon) = \lambda(q) - \frac{(\mathbf{D}_1\mathbf{u}(q), \mathbf{v}(q))}{2(\mathbf{M}\mathbf{u}(q), \mathbf{v}(q))} \varepsilon + o(\varepsilon). \quad (69)$$

588 In other words

$$589 \quad \lambda(q, \mathbf{p}) = \lambda(q) - \frac{\sum_{s=1}^n (\mathbf{D}'_{p_s}\mathbf{u}(q), \mathbf{v}(q))\Delta p_s}{2(\mathbf{M}\mathbf{u}(q), \mathbf{v}(q))} + o(\|\Delta\mathbf{p}\|). \quad (70)$$

590 Following Andreichikov and Yudovich (1974) we require

$$591 \quad \sum_{s=1}^n (\mathbf{D}'_{p_s}\mathbf{u}(q), \mathbf{v}(q))\Delta p_s = 0 \quad (71)$$

592 as a condition for the imaginary eigenvalue remain imaginary after a dissipative  
 593 perturbation. This means that we approximately stay on the neutral stability surface  
 594 after the dissipative perturbation. Eq. (71) gives an exact linear approximation to the  
 595 neutral stability surface at every  $q < q_0$ , if we know exactly the dependencies  $\mathbf{u}(q)$ ,  
 596  $\mathbf{v}(q)$  and  $\lambda(q)$ . Usually, however, these functions are determined numerically, see  
 597 e.g. Andreichikov and Yudovich (1974); Luongo et al. (2016).

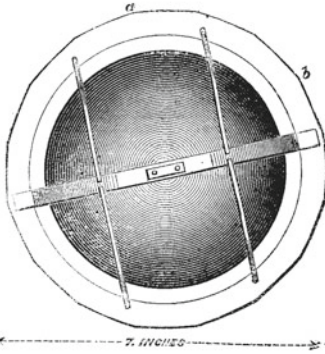
598 Kirillov (2007, 2013a) proposed to use in the method of Andreichikov and  
 599 Yudovich (1974) approximations to  $\mathbf{u}(q)$ ,  $\mathbf{v}(q)$  and  $\lambda(q)$  in the vicinity of  $q = q_0$   
 600 such as those given by Eqs. (63), (64) and (65). Substituting them into (71), we  
 601 express the approximate critical flutter load explicitly as

Nov. 18, 1880] NATURE 69

ON AN EXPERIMENTAL ILLUSTRATION OF MINIMUM ENERGY<sup>2</sup>

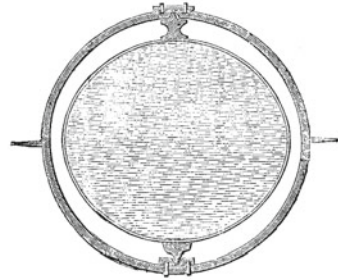
THIS illustration consists of a liquid gyrostat of exactly the same construction as that described and represented by the annexed drawing, repeated from NATURE, February 1, 1877, p. 297, 298, with the difference that the figure of the shell is prolate instead of oblate. The experiment was in fact conducted with the actual apparatus which was exhibited to the British Association at Glasgow in 1876, altered by the substitution of a

The oblate and prolate shells were each of them made from the two hemispheres of sheet copper which plumbers solder together to make their globular floaters. By a little hammering it is easy to alter the hemispheres to the proper shapes to make either the prolate or the oblate figure.



shell having its equatorial diameter about  $\frac{6}{5}$  of its axial diameter, for the shell with axial diameter  $\frac{1}{5}$  of equatorial diameter which was used when the apparatus was shown as a successful gyrostat.

<sup>1</sup> In illustration of this see an exhaustive mathematical paper on the values of iron ores, by Prof. A. Habets: *Cropper's Revue Universelle des Mines* (1877), t. 1, p. 594.  
<sup>2</sup> By Sir William Thomson, F.R.S. British Association, Swansea, Section A.



The result of the first trial was literally startling.

The spinning in the case of the oblate shell, as was known from previous experiments, would have given amply sufficient rotation to the contained water to cause the apparatus to act with great firmness like a solid gyrostat. In the first experiment with the oval shell the shell was seen to be rotating with great velocity during the last minute of the spinning; but the moment it was released from the cord, and when, holding the framework in my hands, I commenced carrying it towards the horizontal glass table to test its gyrostatic quality, the framework which I held in my hands gave a violent uncontrollable lurch, and in a few seconds the shell stopped turning. Its utter failure as a gyrostat is precisely what was expected from the theory, and presents a truly wonderful contrast from what is observed with the apparatus and operations in every respect similar, except having an oblate instead of a prolate shell to contain the liquid.

Fig. 14 The Kelvin gyrostat (Thomson 1880)

$$q = q_0 + \frac{1}{\lambda_1^2} \left( \frac{\sum_{s=1}^n (\mathbf{D}'_s \mathbf{u}_0, \mathbf{v}_0) \Delta p_s}{\sum_{s=1}^n [(\mathbf{D}'_s \mathbf{u}_0, \mathbf{v}_1) + (\mathbf{D}'_s \mathbf{u}_1, \mathbf{v}_0)] \Delta p_s} \right)^2. \quad (72)$$

In the particular case of  $n = 2$  parameters we assume that  $\Delta p_1 = \beta \Delta p_2$ . Introducing the quantity

$$\beta_0 = - \frac{(\mathbf{D}'_{p_2} \mathbf{u}_0, \mathbf{v}_0)}{(\mathbf{D}'_{p_1} \mathbf{u}_0, \mathbf{v}_0)}, \quad (73)$$

we can write  $q(\beta)$  retaining only the terms of order  $(\beta - \beta_0)^2$  and lower:

$$q = q_0 + \frac{\lambda_1^{-2} (\mathbf{D}'_{p_1} \mathbf{u}_0, \mathbf{v}_0)^2 (\beta - \beta_0)^2}{[(\mathbf{D}'_{p_1} \mathbf{u}_0, \mathbf{v}_1) \beta_0 + (\mathbf{D}'_{p_1} \mathbf{u}_1, \mathbf{v}_0) \beta_0 + (\mathbf{D}'_{p_2} \mathbf{u}_0, \mathbf{v}_1) + (\mathbf{D}'_{p_2} \mathbf{u}_1, \mathbf{v}_0)]^2}. \quad (74)$$

Therefore, we have derived a general analogue of the expression (44), which gives the quadratic approximation to the vanishing-dissipation limit of the critical flutter load,  $q(\beta)$ , in the vicinity of  $\beta = \beta_0$  in a rigorous sense. This approximation is sufficient to capture the Whitney umbrella singularity that is responsible for the Ziegler–Bottema destabilization paradox.

Editor Proof

## 613 4 Krein Signature and Stability of Hamiltonian Systems

614 An attempt to spin a hard-boiled egg always ends up successfully: when spun suffi-  
 615 ciently rapidly, its symmetry axis can even rise to the vertical position demonstrat-  
 616 ing a gyroscopic stabilization. The mathematical model of this effect is the rotating solid  
 617 prolate spheroid known as Jellett's egg, see e.g. Kirillov (2013a). In contrast, trying  
 618 to spin a raw egg containing a yolk inside, surrounded by a liquid, will generally lead  
 619 to its slow wobbling motion.

620 Thomson (1880) experimentally demonstrated that a thin-walled and slightly  
 621 oblate spheroid completely filled with liquid remains stable if rotated fast enough  
 622 about a fixed point, which does not happen if the spheroid is slightly prolate, Fig. 14.  
 623 In the same year this observation was confirmed theoretically by Greenhill (1880),  
 624 who found that rotation around the center of gravity of the top in the form of a  
 625 weightless ellipsoidal shell completely filled with an ideal and incompressible fluid,  
 626 is unstable when  $a < c < 3a$ , where  $c$  is the length of the semiaxis of the ellipsoid  
 627 along the axis of rotation and the lengths of the two other semiaxes are equal to  $a$   
 628 (Greenhill 1880).

629 Quite similarly, bullets and projectiles fired from the rifled weapons can relatively  
 630 easily be stabilized by rotation, if they are solid inside. In contrast, the shells, contain-  
 631 ing a liquid substance inside, have a tendency to turn over despite seemingly revolved  
 632 fast enough to be gyroscopically stabilized. Motivated by such artillery applications,  
 633 in 1942 Sobolev, then director of the Steklov Mathematical Institute in Moscow,  
 634 considered stability of a rotating heavy top with a cavity entirely filled with an ideal  
 635 incompressible fluid (Moiseyev and Rumyantsev 1968; Ramodanov and Sidorenko  
 636 2017)—a problem that is directly connected to the classical XIXth century models  
 637 of astronomical bodies with a crust surrounding a molten core (Stewartson 1959).

638 For simplicity, the solid shell of the top and the domain  $V$  occupied by the cavity  
 639 inside it, can be assumed to have a shape of a solid of revolution. They have a common  
 640 symmetry axis where the fixed point of the top is located. The velocity profile of the  
 641 stationary unperturbed motion of the fluid is that of a solid body rotating with the  
 642 same angular velocity  $\Omega$  as the shell around the symmetry axis.

643 Following Sobolev, we denote by  $M_1$  the mass of the shell,  $M_2$  the mass of the  
 644 fluid,  $\rho$  and  $p$  the density and the pressure of the fluid,  $g$  the gravity acceleration,  
 645 and  $l_1$  and  $l_2$  the distances from the fixed point to the centers of mass of the shell  
 646 and the fluid, respectively. The moments of inertia of the shell and the 'frozen' fluid  
 647 with respect to the symmetry axis are  $C_1$  and  $C_2$ , respectively;  $A_1$  ( $A_2$ ) stands for  
 648 the moment of inertia of the shell (fluid) with respect to any axis that is orthogonal  
 649 to the symmetry axis and passes through the fixed point. Let, additionally,

$$650 \quad L = C_1 + C_2 - A_1 - A_2 - \frac{K}{\Omega^2}, \quad K = g(l_1 M_1 + l_2 M_2). \quad (75)$$

651 The solenoidal ( $\operatorname{div} \mathbf{v} = 0$ ) velocity field  $\mathbf{v}$  of the fluid is assumed to satisfy the  
 652 no-flow condition on the boundary of the cavity:  $\mathbf{v}_n|_{\partial V} = 0$ .

653 Stability of the stationary rotation of the top around its vertically oriented sym-  
 654 metry axis is determined by the system of linear equations derived by Sobolev in  
 655 the frame  $(x, y, z)$  that has its origin at the fixed point of the top and rotates with  
 656 respect to an inertial frame around the vertical  $z$ -axis with the angular velocity of the  
 657 unperturbed top,  $\Omega$ . If the real and imaginary part of the complex number  $Z$  describe  
 658 the deviation of the unit vector of the symmetry axis of the top in the coordinates  $x$ ,  
 659  $y$ , and  $z$ , then these equations are, see e.g. Kopachevskii and Krein (2001); Kirillov  
 660 (2013a):

$$\begin{aligned}
 661 \quad & \frac{dZ}{dt} = i\Omega W, \\
 662 \quad & (A_1 + \rho\kappa^2) \frac{dW}{dt} = i\Omega LZ + i\Omega(C_1 - 2A_1 + \rho E)W \\
 663 \quad & \quad + i\rho \int_V \left( v_x \frac{\partial \chi}{\partial y} - v_y \frac{\partial \chi}{\partial x} \right) dV, \\
 664 \quad & \partial_t v_x = 2\Omega v_y - \rho^{-1} \partial_x p + 2i\Omega^2 W \partial_y \bar{\chi}, \\
 665 \quad & \partial_t v_y = -2\Omega v_x - \rho^{-1} \partial_y p - 2i\Omega^2 W \partial_x \bar{\chi}, \\
 666 \quad & \partial_t v_z = -\rho^{-1} \partial_z p, \tag{76}
 \end{aligned}$$

667 where  $2\kappa^2 = \int_V |\nabla \chi|^2 dV$ ,  $E = i \int_V (\partial_x \bar{\chi} \partial_y \chi - \partial_y \bar{\chi} \partial_x \chi) dV$ , and the function  $\chi$  is  
 668 determined by the conditions

$$669 \quad \nabla^2 \chi = 0, \quad \partial_n \chi|_{\partial V} = z(\cos nx + i \cos ny) - (x + iy) \cos nz, \tag{77}$$

670 with  $n$  the absolute value of a vector  $\mathbf{n}$ , normal to the boundary of the cavity.

671 Sobolev realized that some qualitative conclusions on the stability of the top can  
 672 be drawn with the use of the bilinear form

$$673 \quad Q(R_1, R_2) = L\Omega Z_1 \bar{Z}_2 + (A_1 + \rho\kappa^2) W_1 \bar{W}_2 + \frac{\rho}{2\Omega^2} \int_V \bar{\mathbf{v}}_2^T \mathbf{v}_1 dV \tag{78}$$

674 on the elements  $R_1$  and  $R_2$  of the space  $\{R\} = \{Z, W, \mathbf{v}\}$ . The linear operator  $B$   
 675 defined by Eq. (76) that can be written as  $\frac{dR}{dt} = iBR$  has all its eigenvalues real  
 676 when  $L > 0$ , which yields Lyapunov stability of the top. The number of pairs of  
 677 complex-conjugate eigenvalues of  $B$  (counting multiplicities) does not exceed the  
 678 number of negative squares of the quadratic form  $Q(R, R)$ , which can be equal only  
 679 to one when  $L < 0$ . Hence, for  $L < 0$  an unstable solution  $R = e^{i\lambda_0 t} R_0$  can exist  
 680 with  $\text{Im}\lambda_0 < 0$ ; all real eigenvalues are simple except for maybe one (Kopachevskii  
 681 and Krein 2001).

682 In the particular case when the cavity is an ellipsoid of rotation with the semi-axes  
 683  $a$ ,  $a$ , and  $c$ , the space of the velocity fields of the fluid can be decomposed into a  
 684 direct sum of subspaces, one of which is finite-dimensional. Only the movements  
 685 from this subspace interact with the movements of the rigid shell, which yields a  
 686 finite-dimensional system of ordinary differential equations that describes coupling  
 687 between the shell and the fluid.

Calculating the moments of inertia of the fluid in the ellipsoidal container

$$C_2 = \frac{8\pi\rho}{15}a^4c, \quad A_2 = l_2^2M_2 + \frac{4\pi\rho}{15}a^2c(a^2 + c^2),$$

denoting  $m = \frac{c^2 - a^2}{c^2 + a^2}$ , and assuming the field  $\mathbf{v} = (v_x, v_y, v_z)^T$  in the form

$$v_x = (z - l_2)a^2m\xi, \quad v_y = -i(z - l_2)a^2m\xi, \quad v_z = -(x - iy)c^2m\xi,$$

one can eliminate the pressure in Eq. (76) and obtain the reduced model

$$\frac{d\mathbf{x}}{dt} = i\Omega\mathbf{A}^{-1}\mathbf{C}\mathbf{x} = i\Omega\mathbf{B}\mathbf{x}, \quad (79)$$

where  $\mathbf{x} = (Z, W, \xi)^T \in \mathbb{C}^3$  and

$$\mathbf{A} = \begin{pmatrix} 1 & 0 & 0 \\ 0 & A_1 + l_2^2M_2 + \frac{4\pi\rho}{15}a^2c\frac{(c^2 - a^2)^2}{c^2 + a^2} & 0 \\ 0 & 0 & c^2 + a^2 \end{pmatrix},$$

$$\mathbf{C} = \begin{pmatrix} 0 & 1 & 0 \\ L & C_1 - 2A_1 - 2l_2^2M_2 - \frac{8\pi\rho}{15}a^2c^3m^2 & -\frac{8\pi\rho}{15}a^4c^3m^2 \\ 0 & -2 & -2a^2 \end{pmatrix}. \quad (80)$$

The matrix  $\mathbf{B} \neq \mathbf{B}^T$  in Eq. (79) after multiplication by a symmetric matrix

$$\mathbf{G} = \begin{pmatrix} L & 0 & 0 \\ 0 & A_1 + l_2^2M_2 + \frac{4\pi\rho}{15}a^2c\frac{(c^2 - a^2)^2}{c^2 + a^2} & 0 \\ 0 & 0 & \frac{4\pi\rho}{15}a^4c^3\frac{(c^2 - a^2)^2}{c^2 + a^2} \end{pmatrix} \quad (81)$$

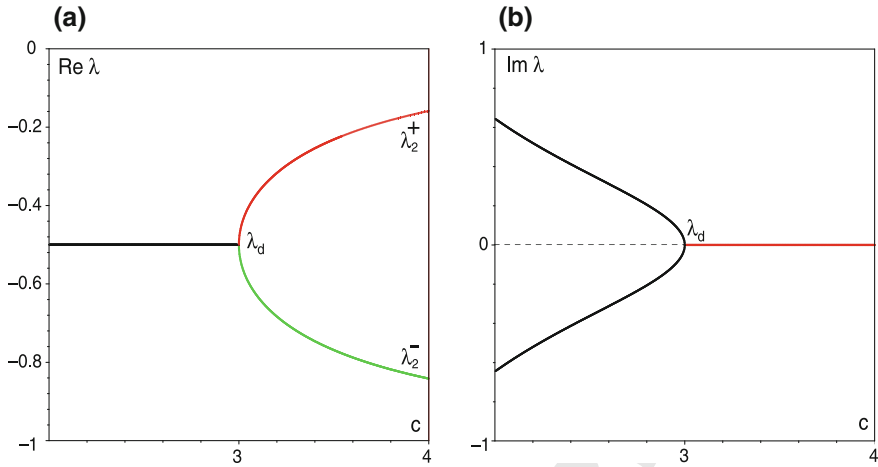
yields a Hermitian matrix  $\mathbf{GB} = \overline{(\mathbf{GB})}^T$ , i.e.  $\mathbf{B}$  is a self-adjoint operator in the space  $\mathbb{C}^3$  endowed with the metric

$$[\mathbf{u}, \mathbf{u}] := (\mathbf{Gu}, \mathbf{u}) = \bar{\mathbf{u}}^T \mathbf{Gu}, \quad \mathbf{u} \in \mathbb{C}^3, \quad (82)$$

which is *definite* when  $L > 0$  and *indefinite* with one negative square when  $L < 0$ . If  $\lambda$  is an eigenvalue of the matrix  $\mathbf{B}$ , i.e.  $\mathbf{Bu} = \lambda\mathbf{u}$ , then  $\bar{\mathbf{u}}^T \mathbf{GBu} = \lambda\bar{\mathbf{u}}^T \mathbf{Gu}$ . On the other hand,  $\bar{\mathbf{u}}^T \overline{(\mathbf{GB})}^T \mathbf{u} = \bar{\lambda}\bar{\mathbf{u}}^T \mathbf{Gu} = \bar{\lambda}\bar{\mathbf{u}}^T \mathbf{Gu}$ . Hence,

$$(\lambda - \bar{\lambda})\bar{\mathbf{u}}^T \mathbf{Gu} = 0,$$

implying  $\bar{\mathbf{u}}^T \mathbf{Gu} = 0$  on the eigenvector  $\mathbf{u}$  of the complex  $\lambda \neq \bar{\lambda}$ . For real eigenvalues  $\lambda = \bar{\lambda}$  and  $\bar{\mathbf{u}}^T \mathbf{Gu} \neq 0$ . The sign of the quantity  $\bar{\mathbf{u}}^T \mathbf{Gu}$  can be different for different real eigenvalues.



**Fig. 15** **a** Simple real eigenvalues (83) of the Sobolev’s top in the Greenhill’s case for  $a = 1$  with (red)  $\bar{\mathbf{u}}^T \mathbf{G} \mathbf{u} > 0$  and (green)  $\bar{\mathbf{u}}^T \mathbf{G} \mathbf{u} < 0$ . **b** At simple complex-conjugate eigenvalues (black) and at the double real eigenvalue  $\lambda_d$  we have  $\bar{\mathbf{u}}^T \mathbf{G} \mathbf{u} = 0$

701 For example, when the ellipsoidal shell is massless and the supporting point is at  
 702 the center of mass of the system, then  $A_1 = 0, C_1 = 0, M_1 = 0, l_2 = 0$ . The matrix  $\mathbf{B}$   
 703 has thus one real eigenvalue ( $\lambda_1^+ = -1, \bar{\mathbf{u}}_1^{+T} \mathbf{G} \mathbf{u}_1^+ > 0$ ) and the pair of eigenvalues

$$704 \quad \lambda_2^\pm = -\frac{1}{2} \pm \frac{1}{2} \sqrt{1 + \frac{32\pi\rho ca^4}{15L}}, \quad L = \frac{4\pi\rho}{15} a^2 c (a^2 - c^2), \quad (83)$$

705 which are real if  $L > 0$  and can be complex if  $L < 0$ . The latter condition together  
 706 with the requirement that the radicand in Eq.(83) is negative, reproduces the  
 707 Greenhill’s instability zone:  $a < c < 3a$  (Greenhill 1880). With the change in  $c$ ,  
 708 the real eigenvalue  $\lambda_2^+$  with  $\bar{\mathbf{u}}_2^{+T} \mathbf{G} \mathbf{u}_2^+ > 0$  collides at  $c = 3a$  with the real eigenvalue  
 709  $\lambda_2^-$  with  $\bar{\mathbf{u}}_2^{-T} \mathbf{G} \mathbf{u}_2^- < 0$  into a real double defective eigenvalue  $\lambda_d$  with the algebraic  
 710 multiplicity two and geometric multiplicity one, see Fig. 15. Note that  $\bar{\mathbf{u}}_d^T \mathbf{G} \mathbf{u}_d = 0$ ,  
 711 where  $\mathbf{u}_d$  is the eigenvector at  $\lambda_d$ .

712 Therefore, in the case of the ellipsoidal shapes of the shell and the cavity, the  
 713 Hilbert space  $\{R\} = \{Z, W, \mathbf{v}\}$  of the Sobolev’s problem endowed with the indefi-  
 714 nite metric ( $L < 0$ ) decomposes into the three-dimensional space of the reduced  
 715 model (79), where the self-adjoint operator  $B$  can have complex eigenvalues and  
 716 real defective eigenvalues, and a complementary infinite-dimensional space, which  
 717 is free of these complications. The very idea that the signature of the indefinite met-  
 718 ric can serve for counting unstable eigenvalues of an operator that is self-adjoint  
 719 in a functional space equipped with such a metric, turned out to be a concept of  
 720 a rather universal character possessing powerful generalizations that were initiated  
 721 by Pontryagin in 1944 (Yakubovich and Starzhinskii 1975; Kopachevskii and Krein  
 722 2001).

Editor Proof

## 723 4.1 Canonical and Hamiltonian Equations

724 Following Yakubovich and Starzhinskii (1975), we consider a complex vector space  
725  $\mathbb{C}^n$  with the inner product  $(\mathbf{x}, \mathbf{y}) = \bar{\mathbf{y}}^T \mathbf{x}$ . Define an indefinite inner product in  $\mathbb{C}^n$  as

$$726 \quad [\mathbf{x}, \mathbf{y}] = (\mathbf{G}\mathbf{x}, \mathbf{y}) = \bar{\mathbf{y}}^T \mathbf{G}\mathbf{x}, \quad (84)$$

727 where  $\mathbf{G} = \bar{\mathbf{G}}^T$  ( $\det \mathbf{G} \neq 0$ ) is an arbitrary (neither positive nor negative definite)  
728 Hermitian  $n \times n$  matrix. Hence,  $[\mathbf{x}, \mathbf{x}]$  is real but in contrast to  $(\mathbf{x}, \mathbf{x})$  it can be positive,  
729 negative, or zero for  $\mathbf{x} \neq 0$ .

730 The matrix  $\mathbf{A}^+$  with the property

$$731 \quad [\mathbf{A}\mathbf{x}, \mathbf{y}] = [\mathbf{x}, \mathbf{A}^+\mathbf{y}] \quad (85)$$

732 is said to be  $\mathbf{G}$ -adjoint to  $\mathbf{A}$ . From Eq. (85) it follows that

$$733 \quad \mathbf{A}^+ = \mathbf{G}^{-1} \bar{\mathbf{A}}^T \mathbf{G}. \quad (86)$$

734 A differential equation

$$735 \quad i^{-1} \mathbf{G} \frac{d\mathbf{z}}{dt} = \mathbf{H}\mathbf{z}, \quad (87)$$

736 where  $\mathbf{H}$  is Hermitian, is called *Hamiltonian equation*. The matrix  $\mathbf{A} = i\mathbf{G}^{-1}\mathbf{H}$   
737 yields

$$738 \quad [\mathbf{A}\mathbf{x}, \mathbf{y}] = -[\mathbf{x}, \mathbf{A}\mathbf{y}], \quad (88)$$

739 i.e.  $\mathbf{A}^+ = -\mathbf{A}$ , and is called the  $\mathbf{G}$ -Hamiltonian matrix (Yakubovich and Starzhinskii  
740 1975; Zhang et al. 2016). In terms of the  $\mathbf{G}$ -Hamiltonian matrix  $\mathbf{A}$ , the Hamiltonian  
741 system (87) takes the form

$$742 \quad \frac{d\mathbf{z}}{dt} = \mathbf{A}\mathbf{z}. \quad (89)$$

743 Since  $\mathbf{A} = -\mathbf{G}^{-1} \bar{\mathbf{A}}^T \mathbf{G}$ , the matrices  $-\bar{\mathbf{A}}^T$  and  $\mathbf{A}$  have the same spectrum. Con-  
744 sequently, if  $\lambda$  is an eigenvalue of  $\mathbf{A}$ , then so is  $-\bar{\lambda}$ . Hence, the spectrum of a  
745  $\mathbf{G}$ -Hamiltonian matrix is symmetric about the imaginary axis. The eigenvalue  $\lambda$  lies  
746 on the imaginary axis if and only if  $\lambda = -\bar{\lambda}$  (Yakubovich and Starzhinskii 1975).

747 Let  $\mathbf{I}$  be the unit  $k \times k$ -matrix and

$$748 \quad \mathbf{J} = \begin{pmatrix} 0 & -\mathbf{I} \\ \mathbf{I} & 0 \end{pmatrix} = -\mathbf{J}^{-1}, \quad (90)$$

749 the canonical *symplectic matrix*. The  $n \times n$  matrix  $\mathbf{G} = i\mathbf{J}$ , where  $n = 2k$ , is  
750 Hermitian:  $\bar{\mathbf{G}}^T = i\bar{\mathbf{J}}^T = -i(-\mathbf{J}) = i\mathbf{J} = \mathbf{G}$ . With  $\mathbf{G} = i\mathbf{J}$  and  $\mathbf{H} = \mathbf{H}^T$  real, the  
751 Hamiltonian equation (87) reduces to

$$\mathbf{J} \frac{d\mathbf{x}}{dt} = \mathbf{H}\mathbf{x} \tag{91}$$

that is referred to as the *canonical equation*, whereas the indefinite inner product takes the form (Yakubovich and Starzhinskii 1975).

$$[\mathbf{x}, \mathbf{y}] = \bar{\mathbf{y}}^T (i\mathbf{J})\mathbf{x} = i\bar{\mathbf{y}}^T \mathbf{J}\mathbf{x}. \tag{92}$$

The canonical Hamiltonian linear equation (91) describe motion of a system with  $k$  degrees of freedom

$$\frac{dx_s}{dt} = \frac{\partial H}{\partial x_{k+s}}, \quad \frac{dx_{k+s}}{dt} = -\frac{\partial H}{\partial x_s}, \quad s = 1, \dots, k, \tag{93}$$

where  $x_s$  are *generalized coordinates* and  $x_{k+s}$  are *generalized momenta*. The quadratic form  $H = \frac{1}{2}(\mathbf{H}\mathbf{x}, \mathbf{x})$ , where  $\mathbf{x}^T = (x_1, \dots, x_{2k})$ , is referred to as a *Hamiltonian function*. The real symmetric  $2k \times 2k$ -matrix  $\mathbf{H}$  of the quadratic form  $H$  is called the *Hamiltonian* (Yakubovich and Starzhinskii 1975).

Seeking for the solution to Eq. (91) in the form  $\mathbf{x} = \mathbf{u} \exp(\lambda t)$ , we find

$$\mathbf{H}\mathbf{u} = \lambda \mathbf{J}\mathbf{u}. \tag{94}$$

From Eqs. (92) and (94) it follows that if  $\lambda$  is a pure imaginary eigenvalue with the eigenvector  $\mathbf{u}$  of the  $(i\mathbf{J})$ -Hamiltonian matrix  $\mathbf{J}^{-1}\mathbf{H}$ , then

$$(\mathbf{H}\mathbf{u}, \mathbf{u}) = \text{Im} \lambda [\mathbf{u}, \mathbf{u}]. \tag{95}$$

Since  $\mathbf{J}$  and  $\mathbf{H}$  are real matrices and the eigenvalues of a  $(i\mathbf{J})$ -Hamiltonian matrix are symmetric with respect to the imaginary axis, the spectrum of the matrix  $\mathbf{J}^{-1}\mathbf{H}$  is symmetric with respect to both real and imaginary axes of the complex plane.

**Theorem 4.1** *Let  $\lambda$  be an eigenvalue of the eigenvalue problem (94). Then so is its complex conjugate,  $\bar{\lambda}$ , and  $-\lambda$ . Hence, for a canonical Hamiltonian linear equation (91) the eigenvalues come in singlets  $\{0\}$ , doublets  $\{\lambda, -\lambda\}$  with  $\lambda \in \mathbb{R}$  or  $\lambda \in i\mathbb{R}$ , or quadruplets  $\{\lambda, -\lambda, \bar{\lambda}, -\bar{\lambda}\}$ . The algebraic multiplicity of the eigenvalue  $\lambda = 0$  is even.*

Consequently, the equilibrium  $\mathbf{x} = 0$  of the system (91) is Lyapunov stable, if and only if the eigenvalues  $\lambda$  of the eigenvalue problem (94) are pure imaginary and semi-simple (Yakubovich and Starzhinskii 1975).

## 4.2 Krein Signature of Eigenvalues

Let  $\lambda$  ( $\text{Re} \lambda = 0$ ) be a simple pure imaginary eigenvalue of a  $\mathbf{G}$ -Hamiltonian matrix  $\mathbf{A}$  and  $\mathbf{u}$  be a corresponding eigenvector:



$$\mathbf{A}\mathbf{u} = \lambda\mathbf{u}. \quad (96)$$

**Definition:** A simple pure imaginary eigenvalue  $\lambda = i\omega$  with the eigenvector  $\mathbf{u}$  is said to have positive Krein signature if  $[\mathbf{u}, \mathbf{u}] > 0$  and negative Krein signature if  $[\mathbf{u}, \mathbf{u}] < 0$ .

Let, further,  $\lambda$  ( $\operatorname{Re}\lambda = 0$ ) be a multiple pure imaginary eigenvalue of a  $\mathbf{G}$ -Hamiltonian matrix  $\mathbf{A}$ , and let  $\mathbf{L}_\lambda$  be the eigensubspace of  $\mathbf{A}$  belonging to the eigenvalue  $\lambda$ , i.e. the set of all  $\mathbf{u} \in \mathbb{C}^n$  satisfying Eq. (96). If  $[\mathbf{u}, \mathbf{u}] > 0$  for any  $\mathbf{u} \in \mathbf{L}_\lambda$  ( $\mathbf{u} \neq 0$ ), then  $\lambda$  is a multiple eigenvalue with positive Krein signature and the eigensubspace  $\mathbf{L}_\lambda$  is positive definite; if  $[\mathbf{u}, \mathbf{u}] < 0$ ,  $\lambda$  is a multiple eigenvalue with negative Krein signature and the eigensubspace  $\mathbf{L}_\lambda$  is negative definite. In such cases the multiple eigenvalue is said to have *definite Krein signature*. If there exists a vector  $\mathbf{u} \in \mathbf{L}_\lambda$  ( $\mathbf{u} \neq 0$ ) such that  $[\mathbf{u}, \mathbf{u}] = 0$ , the multiple pure imaginary eigenvalue  $\lambda$  is said to have *mixed Krein signature* (Yakubovich and Starzhinskii 1975).

Note that in case when a multiple pure imaginary eigenvalue  $\lambda_0$  of  $\mathbf{A}$  has geometric multiplicity that is less than its algebraic multiplicity, then there is an eigenvector  $\mathbf{u}_0$  at  $\lambda_0$  such that  $[\mathbf{u}_0, \mathbf{u}_0] = 0$ , i.e.  $\lambda_0$  has mixed Krein signature. Indeed, there exists at least one *associated vector*  $\mathbf{u}_1$ :  $\mathbf{A}\mathbf{u}_1 = \lambda_0\mathbf{u}_1 + \mathbf{u}_0$ , where  $\mathbf{A}\mathbf{u}_0 = \lambda_0\mathbf{u}_0$ . Taking into account the property (88), we obtain (Kirillov 2013a)

$$\begin{aligned} [\mathbf{A}\mathbf{u}_0, \mathbf{A}\mathbf{u}_1] &= -[\mathbf{u}_0, \mathbf{A}^2\mathbf{u}_1] = -\bar{\lambda}_0^2[\mathbf{u}_0, \mathbf{u}_1] - 2\bar{\lambda}_0[\mathbf{u}_0, \mathbf{u}_0] \\ &= \lambda_0\bar{\lambda}_0[\mathbf{u}_0, \mathbf{u}_1] + \lambda_0[\mathbf{u}_0, \mathbf{u}_0], \end{aligned} \quad (97)$$

which yields

$$[\mathbf{u}_0, \mathbf{u}_0] = 0 \quad (98)$$

since  $\lambda_0 = -\bar{\lambda}_0$  (Yakubovich and Starzhinskii 1975). On the other hand, if  $\lambda_0 \neq -\bar{\lambda}_0$  then  $[\mathbf{u}_0, \mathbf{u}_0] = 0$  for any eigenvector  $\mathbf{u}_0$  at  $\lambda_0$ , which follows from the identity

$$[\mathbf{A}\mathbf{u}_0, \mathbf{A}\mathbf{u}_0] = \lambda_0\bar{\lambda}_0[\mathbf{u}_0, \mathbf{u}_0] = -[\mathbf{u}_0, \mathbf{A}^2\mathbf{u}_0] = \bar{\lambda}_0^2[\mathbf{u}_0, \mathbf{u}_0].$$

Therefore, a multiple pure imaginary eigenvalue can have definite Krein signature only if it is semi-simple.

### 4.3 Krein Collision or Linear Hamiltonian-Hopf Bifurcation

Let in the eigenvalue problem (94) the matrix  $\mathbf{H}$  smoothly depend on a vector of real parameters  $\mathbf{p} \in \mathbb{R}^m$ :  $\mathbf{H} = \mathbf{H}(\mathbf{p})$ . Let at  $\mathbf{p} = \mathbf{p}_0$  the matrix  $\mathbf{H}_0 = \mathbf{H}(\mathbf{p}_0)$  has a double pure imaginary eigenvalue  $\lambda = i\omega_0$  ( $\omega_0 \geq 0$ ) with the Jordan chain consisting of the eigenvector  $\mathbf{u}_0$  and the associated vector  $\mathbf{u}_1$ . Hence,

$$\mathbf{H}_0\mathbf{u}_0 = i\omega_0\mathbf{J}\mathbf{u}_0, \quad \mathbf{H}_0\mathbf{u}_1 = i\omega_0\mathbf{J}\mathbf{u}_1 + \mathbf{J}\mathbf{u}_0. \quad (99)$$

812 Transposing Eq. (99) and applying the complex conjugation yields

$$813 \quad \bar{\mathbf{u}}_0^T \mathbf{H}_0 = i\omega_0 \bar{\mathbf{u}}_0^T \mathbf{J}, \quad \bar{\mathbf{u}}_1^T \mathbf{H}_0 = i\omega_0 \bar{\mathbf{u}}_1^T \mathbf{J} - \bar{\mathbf{u}}_0^T \mathbf{J}. \quad (100)$$

814 As a consequence,  $\bar{\mathbf{u}}_1^T \mathbf{J} \mathbf{u}_0 + \bar{\mathbf{u}}_0^T \mathbf{J} \mathbf{u}_1 = 0$ , i.e.

$$815 \quad [\mathbf{u}_0, \mathbf{u}_1] = -[\mathbf{u}_1, \mathbf{u}_0]. \quad (101)$$

816 Varying the vector of parameters along the curve  $\mathbf{p} = \mathbf{p}(\varepsilon)$  ( $\mathbf{p}(0) = \mathbf{p}_0$ ) and  
 817 applying the perturbation formulas for double eigenvalues that can be found e.g.  
 818 in Kirillov (2013a, 2017), we obtain

$$819 \quad \lambda_{\pm} = i\omega_0 \pm i\omega_1 \sqrt{\varepsilon} + o(\varepsilon^{1/2}), \quad \mathbf{u}_{\pm} = \mathbf{u}_0 \pm i\omega_1 \mathbf{u}_1 \sqrt{\varepsilon} + o(\varepsilon^{1/2}) \quad (102)$$

820 under the assumption

$$821 \quad \omega_1 = \sqrt{\frac{\bar{\mathbf{u}}_0^T \mathbf{H}_1 \mathbf{u}_0}{\bar{\mathbf{u}}_1^T \mathbf{J} \mathbf{u}_0}} > 0, \quad (103)$$

822 where

$$823 \quad \mathbf{H}_1 = \sum_{s=1}^m \left. \frac{\partial \mathbf{H}}{\partial p_s} \frac{dp_s}{d\varepsilon} \right|_{\varepsilon=0}. \quad (104)$$

824 When  $\varepsilon > 0$ , the double eigenvalue  $i\omega_0$  splits into two pure imaginary ones accord-  
 825 ing to the formulas (102). Calculating the indefinite inner product for the perturbed  
 826 eigenvectors  $\mathbf{u}_{\pm}$  by Eq. (92) and taking into account the conditions (98) and (101),  
 827 we find (Kirillov 2013a, 2017)

$$828 \quad [\mathbf{u}_{\pm}, \mathbf{u}_{\pm}] = \pm 2\omega_1 \bar{\mathbf{u}}_1^T \mathbf{J} \mathbf{u}_0 \sqrt{\varepsilon} + o(\varepsilon^{1/2}), \quad (105)$$

829 i.e. the simple pure imaginary eigenvalues  $\lambda_+$  and  $\lambda_-$  have opposite Krein signatures.  
 830 When  $\varepsilon$  decreases from positive values to negative ones, the pure imaginary eigen-  
 831 values of opposite Krein signatures merge at  $\varepsilon = 0$  to the double pure imaginary  
 832 eigenvalue  $i\omega_0$  with the Jordan chain of length 2 that further splits into two complex  
 833 eigenvalues, one of them with the positive real part.

834 When  $\omega_0 \neq 0$ , this process is known as the linear Hamiltonian-Hopf bifurcation  
 835 (Langford 2003), the onset of flutter, non-semi-simple 1 : 1 resonance or the Krein  
 836 collision (Kirillov 2013a).

837 When  $\omega_0 = 0$ , a pair of pure imaginary eigenvalues of opposite Krein signatures  
 838 colliding at zero and splitting then into a pair of real eigenvalues of different sign  
 839 means the onset of the non-oscillatory instability or divergence known also as the  
 840 linear *steady-state bifurcation*.

## 5 Dissipation-Induced Instabilities of Hamiltonian Systems

### 5.1 The Kelvin-Tait-Chetaev Theorem

Potential system of the form  $\mathbf{M}\ddot{\mathbf{x}} + \mathbf{K}\mathbf{x} = 0$  with the mass matrix,  $\mathbf{M} = \mathbf{M}^T$ , and the stiffness matrix,  $\mathbf{K} = \mathbf{K}^T$ , can be transformed to the Hamiltonian form (91). Furthermore, this can be done also in the presence of velocity-dependent gyroscopic forces with the matrix  $\mathbf{G} = -\mathbf{G}^T$  for the gyroscopic system  $\mathbf{M}\ddot{\mathbf{x}} + \mathbf{G}\dot{\mathbf{x}} + \mathbf{K}\mathbf{x} = 0$ . Gyroscopic forces can stabilize the otherwise unstable static equilibrium. This gyroscopic stabilization can be lost in the presence of dissipation, as we all know from observing the behavior of rotating tops.

This dissipation-induced instability of gyroscopic systems is formalized by the Kelvin-Tait-Chetaev theorem (Thomson and Tait 1879; Bloch et al. 1994; Krechetnikov and Marsden 2007).

**Theorem 5.1** (Kelvin-Tait-Chetaev Theorem) *Stability of solutions of the equation*

$$\mathbf{M}\ddot{\mathbf{x}} + (\mathbf{G} + \mathbf{D})\dot{\mathbf{x}} + \mathbf{K}\mathbf{x} = 0, \quad (106)$$

where  $\mathbf{M} > 0$ ,  $\mathbf{D} = \mathbf{D}^T > 0$  and  $\mathbf{K}$  nondegenerate is the same as the stability of solutions of the corresponding potential system,  $\mathbf{M}\ddot{\mathbf{x}} + \mathbf{K}\mathbf{x} = 0$ . In particular, if all the eigenvalues of the real symmetric matrix  $\mathbf{K}$  are positive (negative) then the system (106) is asymptotically stable (unstable).

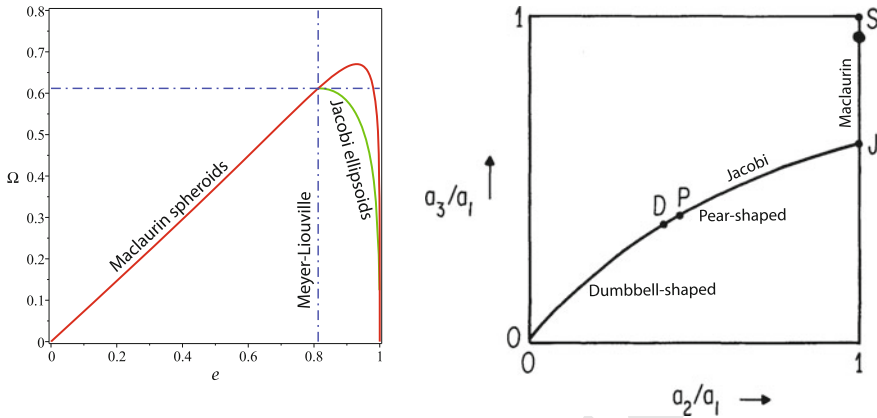
The number of eigenvalues with positive real parts of the system (106) is equal to the number of negative eigenvalues of the matrix  $\mathbf{K}$  (Zajac Theorem, 1964). If the number of negative eigenvalues of  $\mathbf{K}$  (known also as the *Poincaré instability degree*) is even, then the equilibrium of the corresponding potential system can be stabilized by the gyroscopic forces. However, this gyroscopic stabilization is destroyed when dissipative forces with full dissipation ( $\mathbf{D} > 0$ ) are added, no matter how weak they are Kirillov (2013a).

Remarkably, the origin of the Kelvin-Tait-Chetaev theorem is in the centuries-old problem (going back to Newton) on the stability of rotating and self-gravitating masses of fluid motivated by the question of the actual shape of the Earth (Lebovitz 1998; Borisov et al. 2009).

### 5.2 Secular Instability of the Maclaurin Spheroids

In 1742 Maclaurin has found that an oblate spheroid

$$\frac{x^2}{a_1^2} + \frac{y^2}{a_2^2} + \frac{z^2}{a_3^2} = 1, \quad a_3 < a_2 = a_1$$



**Fig. 16** (Left) Families of Maclaurin spheroids and Jacobian ellipsoids in the plane of angular velocity versus eccentricity with the common point at  $e \approx 0.8127$ . (Right) Sequence of bifurcations proposed by the fission theory of binary stars (Lebovitz 1987)

871 is a shape of relative equilibrium of a self-gravitating mass of inviscid fluid in a  
 872 solid-body rotation about the  $z$ -axis, provided that the rate of rotation,  $\Omega$ , is related  
 873 to the eccentricity  $e = \sqrt{1 - \frac{a_3^2}{a_1^2}}$  through the formula (Lebovitz 1998)

874 
$$\Omega^2(e) = 2e^{-3}(3 - 2e^2) \sin^{-1}(e) \sqrt{1 - e^2} - 6e^{-2}(1 - e^2). \quad (107)$$

A century later, Jacobi (1834) has discovered less symmetric shapes of relative equilibria in this problem that are tri-axial ellipsoids

$$\frac{x^2}{a_1^2} + \frac{y^2}{a_2^2} + \frac{z^2}{a_3^2} = 1, \quad a_3 < a_2 < a_1.$$

875 Later on Meyer (1842) and Liouville (1846) have shown that the family of Jacobi's  
 876 ellipsoids has one member in common with the family of Maclaurin's spheroids at  
 877  $e \approx 0.8127$ , see Fig. 16. The equilibrium with the Meyer-Liouville eccentricity is  
 878 neutrally stable, Fig. 17.

879 In 1860 Riemann established neutral stability of inviscid Maclaurin's spheroids  
 880 on the interval of eccentricities ( $0 < e < 0.9529$ ). At the Riemann point with the  
 881 critical eccentricity  $e \approx 0.9529$  the Hamilton-Hopf bifurcation sets in and causes  
 882 dynamical instability with respect to ellipsoidal perturbations beyond this point.

883 A century later Chandrasekhar (1969) proposed a virial method to reduce the problem  
 884 to a finite-dimensional system, which stability is governed by the eigenvalues  
 885 of the matrix polynomial

Editor Proof

$$\mathbf{L}_i(\lambda) = \lambda^2 \begin{pmatrix} 1 & 0 \\ 0 & 1 \end{pmatrix} + \lambda \begin{pmatrix} 0 & -4\Omega \\ \Omega & 0 \end{pmatrix} + \begin{pmatrix} 4b - 2\Omega^2 & 0 \\ 0 & 4b - 2\Omega^2 \end{pmatrix}, \quad (108)$$

where  $\Omega(e)$  is given by the Maclaurin law (107) and  $b(e)$  is as follows

$$b = \frac{\sqrt{1-e^2}}{4e^5} \left\{ e(3-2e^2)\sqrt{1-e^2} + (4e^2-3) \left( \frac{\pi}{2} - \tan^{-1}(e^{-1}\sqrt{1-e^2}) \right) \right\}. \quad (109)$$

The eigenvalues of the matrix polynomial (108) are

$$\lambda = \pm \left( i\Omega \pm i\sqrt{4b - \Omega^2} \right). \quad (110)$$

Requiring  $\lambda = 0$  we can determine the critical Meyer-Liouville eccentricity by solving with respect to  $e$  the equation (Chandrasekhar 1969)

$$4b(e) = 2\Omega^2(e).$$

The critical eccentricity at the Riemann point follows from requiring the radicand in (110) to vanish:

$$4b(e) = \Omega^2(e).$$

Remarkably, when

$$\Omega^2(e) < 4b(e) < 2\Omega^2(e) \quad (111)$$

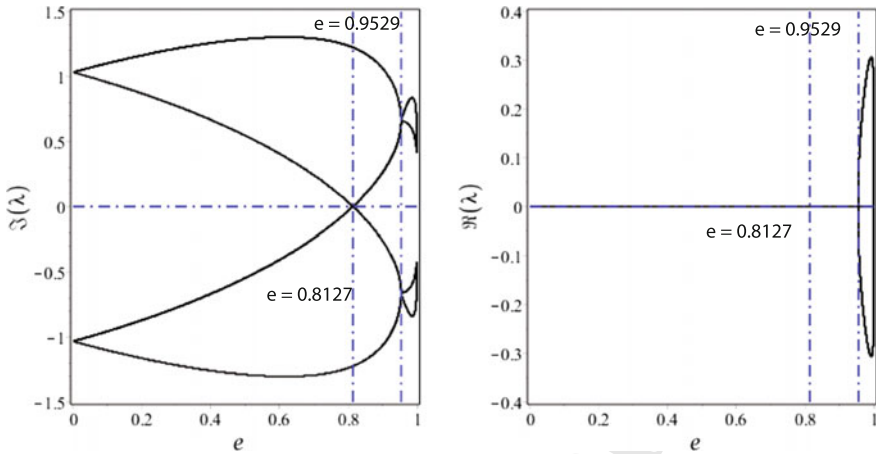
both eigenvalues of the stiffness matrix

$$\begin{pmatrix} 4b - 2\Omega^2 & 0 \\ 0 & 4b - 2\Omega^2 \end{pmatrix}$$

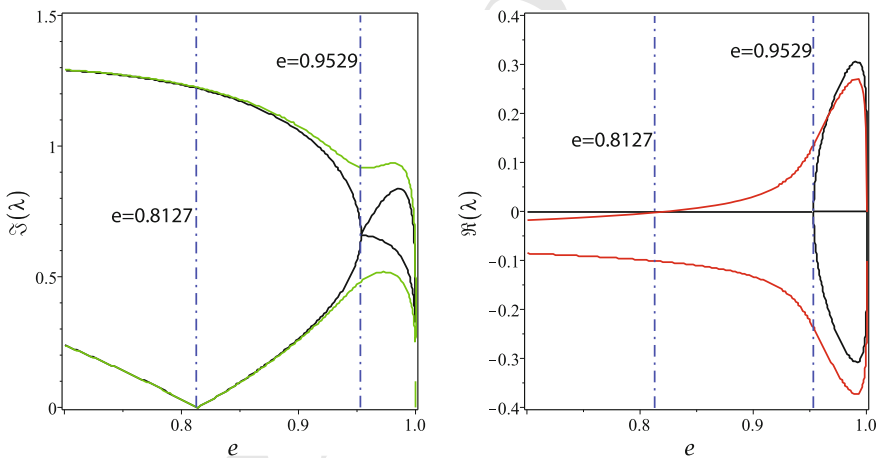
are negative, i.e. the Poincaré instability degree of the equilibrium is even and equal to 2. Hence, the interval (111) corresponding to  $0.8127.. < e < 0.9529..$ , which is stable according to Riemann, is, in fact, the interval of gyroscopic stabilization of the Maclaurin spheroids, Fig. 17.

According to the Theorem 5.1 the gyroscopic stabilization of the equilibrium with nonzero Poincaré instability degree can be destroyed even by the infinitely small dissipation with the positive-definite damping matrix. In the words by Thomson and Tait (1879), “If there be any viscosity, however slight, in the liquid, the equilibrium [beyond  $e \approx 0.8127$ ] in any case of energy either a minimax or a maximum cannot be secularly stable”.

The prediction made by Thomson and Tait (1879) has been verified quantitatively only in the XX-th century by Roberts and Stewartson (1963). Using the virial approach Chandrasekhar (1969) reduced the linear stability problem to the study of eigenvalues of the matrix polynomial



**Fig. 17** (Left) Frequencies and (right) growth rates of the eigenvalues of the inviscid eigenvalue problem  $\mathbf{L}_i(\lambda)\mathbf{u} = 0$  demonstrating the Hamilton-Hopf bifurcation at the Riemann critical value of the eccentricity,  $e \approx 0.9529$  and neutral stability at the Meyer-Liouville point,  $e \approx 0.8127$

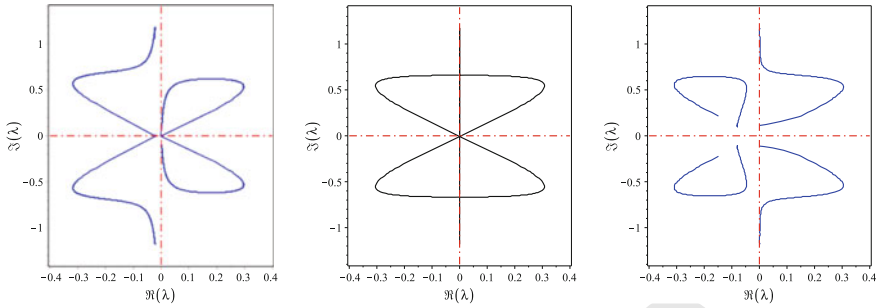


**Fig. 18** (Left) Frequencies and (right) growth rates of the (black lines) inviscid Maclaurin spheroids and (green and red lines) viscous ones with  $\mu = \frac{\nu}{a_1^2} = 0.01$ . Viscosity destabilizes the gyroscopic stabilization of the Maclaurin spheroids on the interval  $0.8127 \dots < e < 0.9529 \dots$ , which is stable in the inviscid case (Roberts and Stewartson 1963; Chandrasekhar 1969; Chandrasekhar 1984)

$$\mathbf{L}_v(\lambda) = \lambda^2 \begin{pmatrix} 1 & 0 \\ 0 & 1 \end{pmatrix} + \lambda \begin{pmatrix} 10\mu & -4\Omega \\ \Omega & 10\mu \end{pmatrix} + \begin{pmatrix} 4b - 2\Omega^2 & 0 \\ 0 & 4b - 2\Omega^2 \end{pmatrix}, \quad (112)$$

where  $\mu = \frac{\nu}{a_1^2}$  and  $\nu$  is the viscosity of the fluid. The operator  $\mathbf{L}_v(\lambda)$  differs from the operator of the ideal system,  $\mathbf{L}_i(\lambda)$ , by the matrix of dissipative forces  $10\lambda\mu\mathbf{I}$ , where  $\mathbf{I}$  is the  $2 \times 2$  unit matrix.

Editor Proof



**Fig. 19** Paths of the eigenvalues in the complex plane for (left) viscous Maclaurin spheroids with  $\mu = \frac{\nu}{\Omega^2} = 0.002$ , (centre) Maclaurin spheroids without dissipation, and (right) inviscid Maclaurin spheroids with radiative losses for  $\delta = 0.05$ . The Krein collision of two modes of the non-dissipative Hamiltonian system shown in the centre occurs at the Riemann critical value  $e \approx 0.9529$ . Both types of dissipation destroy the Krein collision and destabilize one of the two interacting modes at the Meyer-Liouville critical value  $e \approx 0.8127$

911 The characteristic polynomial written for  $\mathbf{L}_v(\lambda)$  yields the equation governing the  
 912 growth rates of ellipsoidal perturbations in the presence of viscosity:

913 
$$25\Omega^2\mu^2 + (\text{Re}\lambda + 5\mu)^2(\Omega^2 - \text{Re}\lambda^2 - 10\text{Re}\lambda\mu - 4b) = 0. \quad (113)$$

914 The right panel of Fig. 18 shows that the growth rates (113) become positive beyond  
 915 the Meyer-Liouville point. Indeed, assuming  $\text{Re}\lambda = 0$  in (113), we reduce it to  
 916  $50\mu^2(\Omega^2 - 2b) = 0$ , meaning that the growth rate vanishes when  $\Omega^2 = 2b$  no mat-  
 917 ter how small the viscosity coefficient  $\mu$  is. But, as we already know, the equation  
 918  $\Omega^2(e) = 2b(e)$  determines exactly the Meyer-Liouville point,  $e \approx 0.8127$ .

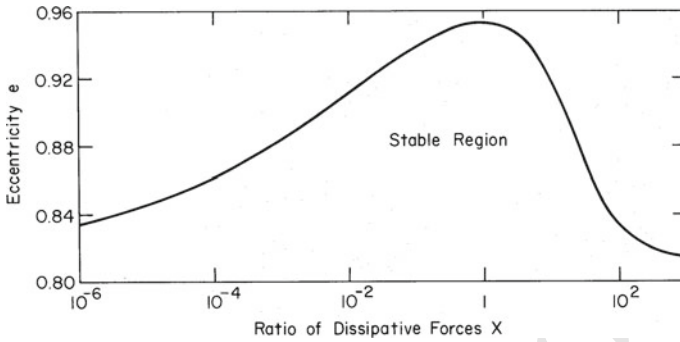
919 It turns out, that the critical eccentricity of the viscous Maclaurin spheroid is equal  
 920 to the Meyer-Liouville value,  $e \approx 0.8127$ , even in the limit of vanishing viscosity,  
 921  $\mu \rightarrow 0$ , and thus does not converge to the inviscid Riemann value  $e \approx 0.9529$ . This is  
 922 nothing else but the Ziegler–Bottema destabilization paradox in a near-Hamiltonian  
 923 dissipative system (Langford 2003; Krechetnikov and Marsden 2007; Kirillov 2007,  
 924 2013a).

925 Viscous dissipation destroys the Krein interaction of two modes at the Riemann  
 926 critical point and destabilizes one of them beyond the Meyer-Liouville point, showing  
 927 a typical for the destabilization paradox *avoided crossing* in the complex plane,  
 928 Fig. 19(left).

929 Thomson and Tait (1879) hypothesised that the instability, which is stimulated by  
 930 the presence of viscosity in the fluid, will result in a slow, or *secular*, departure of  
 931 the system from the unperturbed equilibrium of the Maclaurin family at the Meyer-  
 932 Liouville point and subsequent evolution along the Jacobi family, as long as the latter  
 933 is stable (Lebovitz 1998).

934 Therefore, a rotating, self-gravitating fluid mass, initially symmetric about the axis  
 935 of rotation, can undergo an axisymmetric evolution in which it first loses stability

Editor Proof



**Fig. 20** Critical eccentricity in the limit of vanishing dissipation depends on the damping ratio,  $X$ , and attains its maximum (Riemann) value,  $e \approx 0.9529$  exactly at  $X = 1$ . As  $X$  tends to zero or infinity, the critical value tends to the Meyer-Liouville value  $e \approx 0.8127$ , (Lindblom and Detweiler 1977; Chandrasekhar 1984)

936 to a nonaxisymmetric disturbance, and continues for a while evolving along a non-  
 937 axisymmetric family toward greater departure from axial symmetry, Fig. 16; then it  
 938 undergoes a further loss of stability to a disturbance tending toward splitting into two  
 939 parts (Lebovitz 1998).

940 Rigorous mathematical treatment of the fission theory of binary stars proposed  
 941 by Thomson and Tait (1879) by Lyapunov and Poincaré has laid a foundation to  
 942 modern nonlinear analysis. In particular, it has led Lyapunov to the development of a  
 943 general theory of stability of motion (Borisov et al. 2009). As we remember, it is the  
 944 Lyapunov stability theory that helped Nicolai and Ziegler to shed light on stability  
 945 of nonconservative systems under circulatory forces.

946 Chandrasekhar (1970) demonstrated that there exists another mechanism making  
 947 the Maclaurin spheroid unstable beyond the Meyer-Liouville point of bifurcation,  
 948 namely, the radiative losses due to emission of gravitational waves. However, the  
 949 mode that is made unstable by the radiation reaction is not the same one that is made  
 950 unstable by viscosity, Fig. 19(right).

In the case of the radiative damping mechanism stability is determined by the spectrum of the following matrix polynomial (Chandrasekhar 1970)

$$\mathbf{L}_g(\lambda) = \lambda^2 + \lambda(\mathbf{G} + \mathbf{D}) + \mathbf{K} + \mathbf{N}$$

that contains the matrices of gyroscopic,  $\mathbf{G}$ , damping,  $\mathbf{D}$ , potential,  $\mathbf{K}$ , and nonconservative positional,  $\mathbf{N}$ , forces

$$\mathbf{G} = \frac{5}{2} \begin{pmatrix} 0 & -\Omega \\ \Omega & 0 \end{pmatrix}, \quad \mathbf{D} = \begin{pmatrix} \delta 16\Omega^2(6b - \Omega^2) & -3\Omega/2 \\ -3\Omega/2 & \delta 16\Omega^2(6b - \Omega^2) \end{pmatrix}$$

$$\mathbf{K} = \begin{pmatrix} 4b - \Omega^2 & 0 \\ 0 & 4b - \Omega^2 \end{pmatrix}, \quad \mathbf{N} = \delta \begin{pmatrix} 2q_1 & 2q_2 \\ -q_2/2 & 2q_1 \end{pmatrix},$$

Editor Proof



951 where  $\Omega(e)$  and  $b(e)$  are given by Eqs. (107) and (109). Explicit expressions for  $q_1$   
 952 and  $q_2$  can be found in Chandrasekhar (1970).

953 Lindblom and Detweiler (1977) studied the combined effects of gravitational  
 954 radiation reaction and of viscosity on the stability of the Maclaurin spheroids. As we  
 955 know, each of these dissipative effects induces a secular instability in the Maclaurin  
 956 sequence past the Meyer-Liouville point of bifurcation. However, when both effects  
 957 are considered together, the sequence of stable Maclaurin spheroids therefore reaches  
 958 past the bifurcation point to a new point determined by the *ratio* of the strengths of  
 959 the viscous and the radiative forces.

960 Figure 20 shows the limit of the critical eccentricity as a function of the damping  
 961 ratio in the limit of vanishing dissipation. This limit coincides with the inviscid  
 962 Riemann point only at a particular damping ratio. At any other ratio, the critical  
 963 value is below the Riemann one and tends to the Meyer-Liouville value as this ratio  
 964 tends either to zero or infinity. Lindblom and Detweiler (1977) correctly attributed  
 965 the cancellation of the secular instabilities to the fact that viscous dissipation and  
 966 radiation reaction cause different modes to become unstable, see Fig. 19.

967 Andersson (2003) relates the mode destabilized by the fluid viscosity to the pro-  
 968 grade moving spherical harmonic that appears to be retrograde in the frame rotating  
 969 with the fluid mass and the mode destabilized by the radiative losses to the retrograde  
 970 moving spherical harmonic when it appears to be prograde in the inertial frame. This  
 971 gives a link to destabilization of positive- and negative energy modes (Ostrovsky  
 972 et al. 1986; Kirillov 2009, 2013a) as well as to the theory of the anomalous Doppler  
 973 effect (Nezlin 1976; Ginzburg and Tsytovich 1979; Vesnitskii and Metrikin 1996).  
 974 It is known (Nezlin 1976) that to excite the positive energy mode one must provide  
 975 additional energy to the mode, while to excite the negative energy mode one must  
 976 extract energy from the mode. The latter can be done by dissipation and the former  
 977 by the nonconservative positional (curl) forces. Both are presented in the model by  
 978 Lindblom and Detweiler (1977).

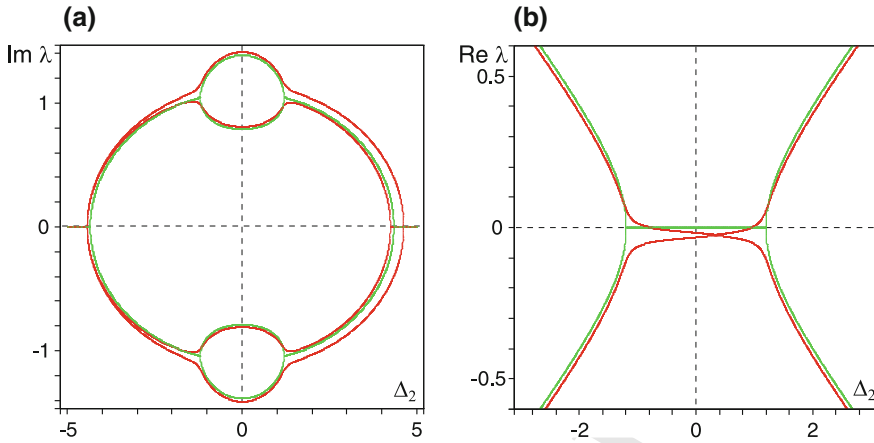
979 The destabilization of a Hamiltonian system in the presence of two different types  
 980 of non-Hamiltonian perturbations can be understood on the example of the general  
 981 two-dimensional system

$$982 \quad \ddot{\mathbf{x}}(t) + (\delta \mathbf{D} + \Omega \mathbf{G}) \dot{\mathbf{x}}(t) + (\mathbf{K} + \nu \mathbf{N}) \mathbf{x}(t) = 0, \quad \mathbf{x} \in \mathbb{R}^2 \quad (114)$$

where  $\delta$ ,  $\Omega$ ,  $\nu$  are scalar coefficients and matrices  $\mathbf{D} > 0$ ,  $\mathbf{K} > 0$  are real and sym-  
 metric, while matrices  $\mathbf{G}$  and  $\mathbf{N}$  are skew-symmetric as follows

$$\mathbf{G} = \mathbf{N} = \begin{pmatrix} 0 & -1 \\ 1 & 0 \end{pmatrix}.$$

983 This system is a conservative Hamiltonian system if  $\delta = 0$ ,  $\Omega = 0$ , and  $\nu = 0$ ,  
 984 which is statically unstable for  $\mathbf{K} < 0$  with the even Poincaré instability degree equal  
 985 to 2. Adding gyroscopic forces with  $\Omega > 0$ , keeps this system Hamiltonian and yields  
 986 its stabilization if  $\Omega > \Omega_f = \sqrt{-\kappa_1} + \sqrt{-\kappa_2}$ , where  $\kappa_{1,2} < 0$  are eigenvalues of  $\mathbf{K}$ .



**Fig. 21** Given  $\Omega = 0.3$ , the green lines depict (a) imaginary and (b) real parts of the eigenvalues of the PT - symmetric problem with indefinite damping (116) as functions of the parameter  $\Delta_2 = \mu_1 - \mu_2 = 2\mu$  when  $k = 1$ . Red lines correspond to the eigenvalues of the problem (41) with  $k_1 = 1$ ,  $\kappa = k_2 - k_1 = 0.1$  and  $\Delta_1 = \mu_1 + \mu_2 = 0.1$

987 Owing to the ‘reversible’ symmetry of its spectrum (MacKay 1991; Bloch et al.  
 988 1994), the Hamiltonian system displays flutter instability via the collision of imagi-  
 989 nary eigenvalues at  $\Omega = \Omega_f$  and their subsequent splitting into a complex quadruplet  
 990 as soon as  $\Omega$  decreases below  $\Omega_f$ . This is already familiar to us linear Hamilton-  
 991 Hopf bifurcation.

If  $\delta > 0$ ,  $\nu > 0$  the gyroscopic stability is destroyed at the threshold of the classical-Hopf bifurcation (Kirillov 2007, 2013a)

$$\Omega_H \approx \Omega_f + \frac{2\Omega_f}{(\omega_f \text{tr} \mathbf{D})^2} \left( \frac{\nu}{\delta} - \frac{\text{tr}(\mathbf{KD} + (\Omega_f^2 - \omega_f^2)\mathbf{D})}{2\Omega_f} \right)^2,$$

992 where  $\omega_f^2 = \sqrt{\kappa_1 \kappa_2}$  and  $\mathbf{D} > 0$ .

993 The dependency of the new gyroscopic stabilization threshold just on the ratio  
 994  $\nu/\delta$  implies that the limit of  $\Omega_H$  as both  $\nu$  and  $\delta \rightarrow 0$  is higher than  $\Omega_f$  for all ratios  
 995 except a unique one. Similarly to the case of nonconservative reversible systems,  
 996 this happens because the classical Hopf and the Hamilton-Hopf bifurcations meet in  
 997 the Whitney umbrella singularity that exists on the stability boundary of a nearly-  
 998 Hamiltonian dissipative system and corresponds to the onset of the Hamilton-Hopf  
 999 bifurcation (Bottema 1956; Arnold 1972; Langford 2003; Kirillov 2007, 2013a;  
 1000 Krechetnikov and Marsden 2007; Kirillov and Verhulst 2010).

Editor Proof

## 6 Stability in the Presence of Potential, Circulatory, Gyroscopic and Dissipative Forces

Beletsky (1995) remarked that when potential, circulatory and gyroscopic forces are present simultaneously, it becomes nontrivial to judge about stability. “The pairwise interaction of arbitrary two of these [forces] results in the existence of stable domains in the parameter space. However, the simultaneous action of all three effects always results in instability!” (Beletsky 1995). Addition of dissipation entangles stability analysis even more (Kirillov 2013a; Hagedorn et al. 2014; Kliem and Pommer 2017). Here we present several examples illustrating these statements.

### 6.1 Rotating Shaft by Shieh and Masur (1968)

Let us return once again to the model (20) of a rotating shaft by Shieh and Masur (1968) with damping but without circulatory forces

$$\begin{aligned} m\ddot{u} + \mu_1\dot{u} - 2\Omega\dot{v} + (k_1 - \Omega^2)u &= 0 \\ m\ddot{v} + \mu_2\dot{v} + 2\Omega\dot{u} + (k_2 - \Omega^2)v &= 0. \end{aligned} \quad (115)$$

Although the literal meaning of the word ‘damping’ prescribes the coefficients  $\mu_1$  and  $\mu_2$  to be nonnegative, it is instructive to relax this sign convention Kirillov (2013b). Therefore, we consider the gyroscopic system (115) where the negative sign of the damping coefficient corresponds to a gain and the positive one to a loss (Karami and Inman 2011; Schindler et al 2011).

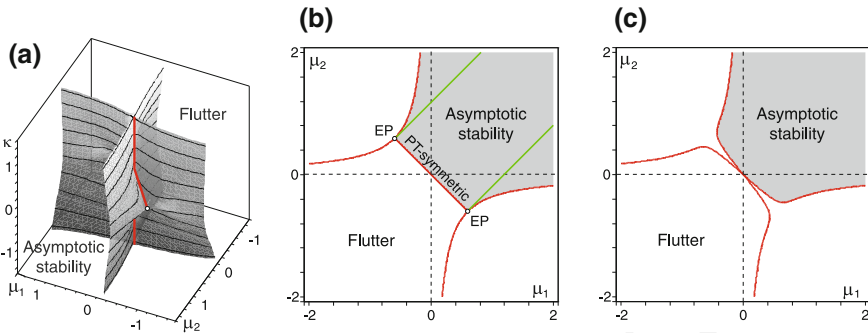
In mechanics, negative damping terms enter the equations of motion of moving continua in frictional contact when the dependence of the frictional coefficient on the relative velocity has a negative slope, which can be observed already in the tabletop experiments with the singing wine glass (Kirillov 2009, 2013a). In physics, a pair of coupled oscillators, one with gain and the other with loss, can naturally be implemented as an LRC-circuit (Schindler et al 2011).

When  $\mu_1 = -\mu_2 = \mu > 0$  the gain and loss in Eq. (41) are in perfect balance. Let us further assume that  $k_1 = k_2 = k$ :

$$\begin{aligned} m\ddot{u} + \mu\dot{u} - 2\Omega\dot{v} + (k - \Omega^2)u &= 0 \\ m\ddot{v} - \mu\dot{v} + 2\Omega\dot{u} + (k - \Omega^2)v &= 0. \end{aligned} \quad (116)$$

Let us look at what happens with these equations when we change the direction of time, assuming  $t \rightarrow -t$ . Then,

$$\begin{aligned} m\ddot{u} - \mu\dot{u} + 2\Omega\dot{v} + (k - \Omega^2)u &= 0 \\ m\ddot{v} + \mu\dot{v} - 2\Omega\dot{u} + (k - \Omega^2)v &= 0 \end{aligned} \quad (117)$$



**Fig. 22** Stability domain of the rotating shaft by Shieh and Masur for  $k_1 = 1$ ,  $\Omega = 0.3$ , and  $\beta = 0$ . **a** The Plücker conoid in the  $(\mu_1, \mu_2, \kappa)$ -space and its slices in the  $\mu_1, \mu_2$ -plane with **b**  $\kappa = 0$  and **c**  $\kappa = 0.1$ . Open circles show locations of exceptional points (EPs) where pure imaginary eigenvalues of the ideal PT-symmetric system (116) experience the nonsemisimple 1 : 1 resonance; green lines are locations of the exceptional points where double nonsemisimple eigenvalues have negative real parts

1034 and we see that Eq. (116) are not invariant to the time reversal transformation (T).  
 1035 The interchange of the coordinates as  $x \leftrightarrow y$  in Eq. (116) results again in Eq. (117),  
 1036 which do not coincide with the original. Hence, the Eq. (116) are not invariant with  
 1037 respect to the parity transformation (P).

1038 Nevertheless, two negatives make an affirmative, and the combined PT-  
 1039 transformation leaves the Eq. (116) invariant despite the T-symmetry and P-symmetry  
 1040 not being respected separately. The spectrum of the PT-symmetric system (116) with  
 1041 indefinite damping is symmetrical with respect to the imaginary axis on the complex  
 1042 plane as it happens in Hamiltonian and reversible systems.

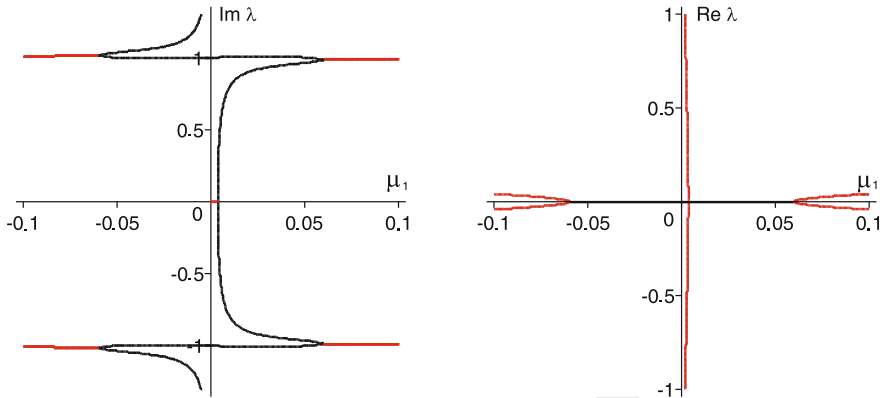
To see this, let us consider the eigenvalues  $\lambda$  of the problem (41) introducing  
 the new parameters  $\Delta_1 = \mu_1 + \mu_2$ ,  $\Delta_2 = \mu_1 - \mu_2$  and  $\kappa = k_2 - k_1$ . At  $\Delta_1 = 0$  and  
 $\kappa = 0$  they represent the spectrum of the problem (116)

$$\lambda = \pm \frac{1}{4} \sqrt{2\Delta_2^2 - 16k_1 - 16\Omega^2 \pm 2\sqrt{(16\Omega^2 - \Delta_2^2)(16k_1 - \Delta_2^2)}}$$

1043 where  $k_1 = k$  and  $\Delta_2 = 2\mu$ . In Fig. 21 the eigenvalues are shown by the green lines.  
 1044 They are pure imaginary when  $|\Delta_2| < 4|\Omega|$ . At the exceptional points (EPs),  $\Delta_2 =$   
 1045  $\pm 4\Omega$ , the pure imaginary eigenvalues collide into a double defective one which with  
 1046 the further increase in  $\Delta_2$  splits into a complex-conjugate pair (flutter instability).

1047 PT-symmetry can be violated by the asymmetry both in the stiffness distribution  
 1048  $\kappa \neq 0$  and in the balance of gain and loss  $\Delta_1 \neq 0$ . In such a situation, the merging  
 1049 of eigenvalues that was perfect for the PT-symmetric system (116) is destroyed. The  
 1050 red eigencurves in Fig. 21 demonstrate the imperfect merging of modes that causes  
 1051 a decrease of the stability interval with respect to that of the symmetric system (the  
 1052 effect similar to the Ziegler–Bottema destabilization paradox in circulatory systems).

Editor Proof



**Fig. 23** Imaginary and real parts of the roots of the characteristic equation (118) as a function of the damping coefficient  $\mu_1$  under the constraints (119) for  $k_1 = 1$ ,  $\Omega = 0.03$  and  $\beta = 0.03$

1053 The Routh–Hurwitz conditions applied to the characteristic polynomial of the  
 1054 system (1.60) yield the domain of the asymptotic stability

1055 
$$\mu_1\mu_2\kappa^2 + (\mu_1 + \mu_2)(\mu_1\mu_2 + 4\Omega^2)(\mu_1\kappa + (\mu_1 + \mu_2)(k_1 - \Omega^2)) > 0$$

1056 
$$\mu_1 + \mu_2 > 0,$$

1057 shown in Fig. 22a in the  $(\mu_1, \mu_2, \kappa)$ -space. The surface has a self-intersection along  
 1058 the  $\kappa$ -axis that corresponds to a marginally stable conservative gyroscopic (Hamil-  
 1059 tonian) system. More intriguing is that in the  $(\kappa = 0)$  - plane there exists another  
 1060 self-intersection along the interval of the line  $\mu_1 + \mu_2 = 0$  with the ends at the excep-  
 1061 tional points  $(\mu_1 = 2\Omega, \mu_2 = -2\Omega)$  and  $(\mu_1 = -2\Omega, \mu_2 = 2\Omega)$ , see Fig. 22b. This  
 1062 is the interval of marginal stability of the oscillatory damped (PT-symmetric) gyro-  
 1063 scopic system (116) with the perfect gain/loss balance. At the exceptional points, the  
 1064 stability boundary has the Whitney umbrella singularities.

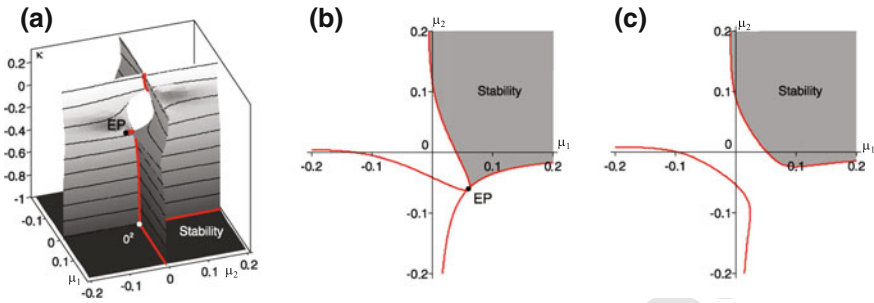
In the  $(\kappa = 0)$  - plane the range of stability is growing with the increase of the distance from the line  $\mu_1 + \mu_2 = 0$ , which is accompanied by detuning of the gain/loss balance, Fig. 22b. Indeed, in this slice the boundary of the domain of asymptotic stability is the hyperbola

$$(\mu_1 - \mu_2)^2 - (\mu_1 + \mu_2)^2 = 16\Omega^2.$$

1065 At  $\mu_1 + \mu_2 = 0$  it touches the two straight lines  $\mu_1 - \mu_2 = \pm 4\Omega$ , every point of  
 1066 which corresponds to a pair of double defective complex-conjugate eigenvalues with  
 1067 real parts that are negative when  $\mu_1 + \mu_2 > 0$ , positive when  $\mu_1 + \mu_2 < 0$ , and zero  
 1068 when  $\mu_1 + \mu_2 = 0$ :

$$\lambda = -\frac{\mu_1 + \mu_2}{4} \pm \frac{1}{4}\sqrt{(\mu_1 + \mu_2)^2 - 16(k_1 - \Omega^2)}$$

Editor Proof



**Fig. 24** Stability domain of the rotating shaft by Shieh and Masur for  $k_1 = 1, \Omega = 0.03, \beta = 0.03$ . **a** The ‘Viaduct’ in the  $(\mu_1, \mu_2, \kappa)$ -space and its slices in the  $(\mu_1, \mu_2)$ -plane with **b**  $\kappa = 0.06$  and **c**  $\kappa = 0.03$  (Kirillov 2011a, b)

1069 The two lines of exceptional points stem from the end points of the interval of  
 1070 marginal stability of the PT - symmetric system and continue inside the asymptotic  
 1071 stability domain of the near-PT-symmetric one (green lines in Fig. 22b).

1072 The proximity of a set of defective eigenvalues to the boundary of the asymptotic  
 1073 stability, that generically is characterized by simple pure imaginary eigenvalues, plays  
 1074 an important role in modern nonconservative physical and mechanical problems.  
 1075 Near this set the eigenvalues can dramatically change their trajectories in the complex  
 1076 plane. For this reason, encountering double eigenvalues with the Jordan block and  
 1077 negative real parts is considered as a precursor to instability.

1078 The full model of Shieh and Masur (20) provides even more non-trivial example.  
 1079 Indeed, its characteristic equation has the form

$$1080 \lambda^4 + (\mu_1 + \mu_2)\lambda^3 + (\mu_1\mu_2 + k_1 + k_2 + 2\Omega^2)\lambda^2 \quad (118)$$

$$1081 + (k_1\mu_2 + \mu_1k_2 + 4\Omega\beta - (\mu_1 + \mu_2)\Omega^2)\lambda + (\Omega^2 - k_1)(\Omega^2 - k_2) + \beta^2 = 0.$$

1082 Equation (118) is biquadratic in the case when

$$1083 \mu_1 + \mu_2 = 0, \quad \kappa = -\frac{4\Omega\beta}{\mu_1}, \quad (119)$$

1084 with  $\kappa = k_2 - k_1$ . If  $k_1 > \Omega^2$  and  $\beta > 0$  then all the roots of Eq. (118) are imaginary  
 1085 when

$$1086 2\Omega \leq \mu_1 < 0, \quad \frac{4\Omega\beta(k_1 - \Omega^2)}{\beta^2 + (k_1 - \Omega^2)^2} < \mu_1 \leq 2\Omega. \quad (120)$$

1087 In Fig. 23 the imaginary eigenvalues are shown by black lines as functions of the  
 1088 damping parameter  $\mu_1$ . At

$$1089 \mu_1 = \mu_d := \frac{4\Omega\beta(k_1 - \Omega^2)}{\beta^2 + (k_1 - \Omega^2)^2}, \quad \kappa = \kappa_d := -k_1 + \Omega^2 - \frac{\beta^2}{k_1 - \Omega^2} \quad (121)$$

Editor Proof

there exists a double zero eigenvalue with the Jordan block, see Figs. 23 and 24a. In the interval  $0 < \mu_1 < \mu_d$  there exist one positive and one negative real eigenvalue. In Fig. 23 the eigenvalues with non-zero real parts are shown in red. In the  $(\mu_1, \mu_2, \kappa)$ -space the exceptional points (EPs)

$$(-2\Omega, 2\Omega, 2\beta), \quad (2\Omega, -2\Omega, -2\beta)$$

correspond to the double imaginary eigenvalues with the Jordan block

$$\lambda_{-2\Omega} = \pm i\sqrt{k_1 - \Omega^2 + \beta}, \quad \lambda_{2\Omega} = \pm i\sqrt{k_1 - \Omega^2 - \beta},$$

1090 for  $\mu_1 = -2\Omega$  and  $\mu_1 = 2\Omega$ , respectively.

1091 We see in Fig. 23 that changing the damping parameter  $\mu_1$  we migrate from the  
1092 marginal stability domain to that of flutter instability by means of the collision of  
1093 the two simple pure imaginary eigenvalues as it happens in gyroscopic or circulatory  
1094 systems without dissipation. It is remarkable that such a behavior of eigenvalues is  
1095 observed in the gyroscopic system in the presence of dissipative and non-conservative  
1096 positional forces.

1097 Let us now establish how in the  $(\mu_1, \mu_2, \kappa)$ -space the domain of marginal stability  
1098 given by the expressions (119) and (120) is connected to the domain of asymptotic  
1099 stability of the Eq. (118). Writing the Liénard and Chipart conditions for asymptotic  
1100 stability of the polynomial (118) we find

$$\begin{aligned} 1101 \quad p_1 &:= \mu_1 + \mu_2 > 0, \\ 1102 \quad p_2 &:= \mu_1\mu_2 + k_1 + k_2 + 2\Omega^2 > 0, \\ 1103 \quad p_4 &:= (\Omega^2 - k_1)(\Omega^2 - k_2) + \beta^2 > 0, \\ 1104 \quad H_3 &:= (\mu_1 + \mu_2)(\mu_1\mu_2 + k_1 + k_2 + 2\Omega^2) \\ 1105 \quad &\quad \times (k_1\mu_2 + \mu_1k_2 + 4\Omega\beta - (\mu_1 + \mu_2)\Omega^2) \\ 1106 \quad &\quad - (\mu_1 + \mu_2)^2((\Omega^2 - k_1)(\Omega^2 - k_2) + \beta^2) \\ 1107 \quad &\quad - (k_1\mu_2 + \mu_1k_2 + 4\Omega\beta - (\mu_1 + \mu_2)\Omega^2)^2 > 0. \end{aligned} \quad (122)$$

1108 The surfaces  $p_4 = 0$  and  $H_3 = 0$  are plotted in Fig. 24a. The former is simply  
1109 a horizontal plane that passes through the point of the double zero eigenvalue with  
1110 the coordinates  $(\mu_d, -\mu_d, \kappa_d)$  and thus bounds the stability domain from below. The  
1111 surface  $H_3 = 0$  is singular because it has self-intersections along the portions of  
1112 the hyperbolic curves (119) selected by the inequalities (120). The curve of self-  
1113 intersection that corresponds to  $\kappa > 0$  ends up at the EP with the double pure imag-  
1114 inary eigenvalue  $\lambda_{-2\Omega}$ .

1115 Another curve of self-intersection has at its ends the EP with the double pure  
1116 imaginary eigenvalue  $\lambda_{2\Omega}$  and the point of the double zero eigenvalue,  $0^2$ . In Fig. 24a  
1117 the curves of self-intersection are shown in red and the EP and  $0^2$  are marked by the  
1118 black and white circles, respectively. At the point  $0^2$  the surfaces  $p_4 = 0$  and  $H_3 = 0$   
1119 intersect each other forming a trihedral angle singularity of the stability boundary

with its edges depicted by red lines in Fig. 24a. The surface  $H_3 = 0$  is symmetric with respect to the plane  $p_1 = 0$ . Thus, a part of it that belongs to the subspace  $p_1 > 0$  bounds the domain of asymptotic stability.

At the EPs, the boundary of the asymptotic stability domain has singular points that are locally equivalent to the Whitney umbrella singularity. Between the two EPs the surface  $H_3 = 0$  has an opening around the origin that separates its two sheets. This window allows the flutter instability to exist in the vicinity of the origin for small damping coefficients and small separation of the stiffness coefficients  $\kappa$ .

In Fig. 24b a cross-section of the surface  $H_3 = 0$  by the horizontal plane that passes through the lower exceptional point is shown. The domain in grey indicates the area of asymptotic stability. Its boundary has a cuspidal point singularity at the EP. Although the very singular shape of the planar stability domain is typical in the vicinity of the EP with the pure imaginary double eigenvalue with the Jordan block, the unusual feature is the location of the EP that corresponds to non-vanishing damping coefficients, Fig. 24b.

According to the theorems of (Bottema, 1955; Lakhadanov, 1975) the undamped gyroscopic system with non-conservative positional forces is generically unstable, see e.g. Beletsky (1995), Kirillov (2013a). By examining the slices of the surface  $H_3 = 0$  at various values of  $\kappa$  one can see that the origin is indeed always unstable, Fig. 24b, c. At  $\kappa = 0$  the origin is unstable in the presence of the non-conservative positional forces even when the rotation is absent ( $\Omega = 0$ ) according to the Merkin theorem. Contrary to the situation known as the Ziegler–Bottema destabilization paradox, in the Shieh–Masur model the tending of the damping coefficients to zero along a path in the  $(\mu_1, \mu_2)$ -plane cannot lead to the set of pure imaginary spectrum of the undamped system because in this model such a set corresponds to the non-vanishing damping coefficients.

Therefore, the Shieh–Masur model provides a nontrivial example of a gyroscopic system that can have all its eigenvalues pure imaginary in the presence of dissipative and circulatory forces. The highly non-trivial shape of the discovered stability boundary illustrates the peculiarities of stability of a system loaded by non-conservative positional forces in their interplay with the dissipative, gyroscopic and potential ones.

## 6.2 Two-Mass-Skate (TMS) Model of a Bicycle

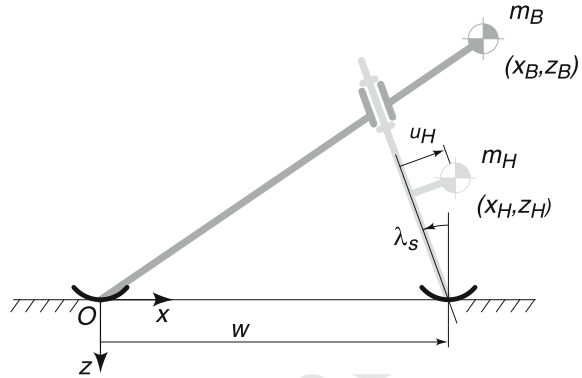
Kooijman et al. (2011) considered a reduced model of a bicycle with vanishing radii of the wheels (that are replaced by skates), known under the name of the two-mass-skate (TMS) bicycle. The deviation from the straight vertical equilibrium is described by the leaning angle of the frame and the steering angle of the front wheel/skate that are governed by the following system of two linear equations

$$\mathbf{M}\ddot{\mathbf{x}} + v\mathbf{D}\dot{\mathbf{x}} + g\mathbf{K}\mathbf{x} + v^2\mathbf{N}\mathbf{x} = 0, \quad (123)$$

where dot denotes time differentiation,



**Fig. 25** The two-mass-skate bicycle model (Kooijman et al. 2011)



$$\begin{aligned}
 \mathbf{M} &= \begin{pmatrix} m_B z_B^2 + m_H z_H^2 & -m_H u_H z_H \\ -m_H u_H z_H & m_H u_H^2 \end{pmatrix}, \\
 \mathbf{D} &= \begin{pmatrix} 0 & -(m_B x_B z_B + m_H x_H z_H)/\hat{w} \\ 0 & (m_H u_H x_H)/\hat{w} \end{pmatrix}, \\
 \mathbf{K} &= \begin{pmatrix} m_B z_B + m_H z_H & -m_H u_H \\ -m_H u_H & -m_H u_H \sin \lambda_s \end{pmatrix}, \\
 \mathbf{N} &= \begin{pmatrix} 0 & -(m_B z_B + m_H z_H)/\hat{w} \\ 0 & (m_H u_H)/\hat{w} \end{pmatrix}, \quad (124)
 \end{aligned}$$

$u_H = (x_H - w) \cos \lambda_s - z_H \sin \lambda_s$ ,  $\hat{w} = w / \cos \lambda_s$  and  $g$  denotes the gravity acceleration.

The model (123), (124) is nonconservative, containing dissipative, gyroscopic, potential and circulatory forces. Curiously enough, Eq. (123) has a form that is typical in many *fluid-structure interactions* problems, where the parameter  $v$  would correspond to the velocity of the flow either inside of a flexible pipe or around a flexible structure (Mandre and Mahadevan 2010; Paidoussis 2016). This similarity in the mathematical description suggests an analogy between the *weaving bicycle* and *fluttering flag*, which is not very obvious.

In fact, Eq. (123) depends on 9 dimensional parameters:

$$w, v, \lambda_s, m_B, x_B, z_B, m_H, x_H, z_H$$

that represent, respectively, the wheel base, velocity of the bicycle, steer axis tilt, rear frame assembly ( $B$ ) mass, horizontal and vertical coordinates of the rear frame assembly center of mass, front fork and handlebar assembly ( $H$ ) mass, and horizontal and vertical coordinates of the front fork and handlebar assembly center of mass.

Choosing the wheelbase,  $w$ , as a unit of length, and introducing the Froude number,  $Fr$ , we find that, actually, the model depends on the following *seven* dimensionless parameters:

$$\text{Fr} = \frac{v}{\sqrt{gw}}, \quad \mu = \frac{m_H}{m_B}, \quad \xi_B = \frac{x_B}{w}, \quad \xi_H = \frac{x_H}{w}, \quad \zeta_B = \frac{z_B}{w}, \quad \zeta_H = \frac{z_H}{w}, \quad \lambda_s.$$

1176 We can assume that for realistic bicycles  $0 \leq \mu \leq 1$ . Notice that  $\zeta_B \leq 0$  and  $\zeta_H \leq 0$   
1177 due to choice of the system of coordinates, Fig. 25.

Assuming the solution  $\sim \exp(\sigma t)$  and introducing the dimensionless time  $\tau = \sqrt{\frac{g}{w}}t$  such that the dimensionless eigenvalue is  $s = \sigma \sqrt{\frac{w}{g}}$ , we write the characteristic polynomial of the TMS bicycle model:

$$p(s) = a_0s^4 + a_1s^3 + a_2s^2 + a_3s + a_4,$$

1178 with the coefficients

$$\begin{aligned} 1179 \quad a_0 &= -(\zeta_H \tan \lambda_s - \xi_H + 1)\zeta_B^2, \\ 1180 \quad a_1 &= \text{Fr}(\zeta_B \xi_H - \zeta_H \xi_B)\zeta_B, \\ 1181 \quad a_2 &= \text{Fr}^2(\zeta_B - \zeta_H)\zeta_B - \zeta_B(\zeta_B + \zeta_H) \tan \lambda_s - (\xi_H - 1)(\mu\zeta_H - \zeta_B), \\ 1182 \quad a_3 &= -\text{Fr}(\xi_B - \xi_H)\zeta_B, \\ 1183 \quad a_4 &= -\zeta_B \tan \lambda_s - \mu(\xi_H - 1). \end{aligned} \quad (125)$$

Notice that in the case when the coordinates of the masses  $m_B$  and  $m_H$  coincide:

$$\xi_H = \xi_B, \quad \zeta_H = \zeta_B$$

the characteristic polynomial simplifies and factorizes as

$$p(s) = -(s^2\zeta_B + 1)(\zeta_B(\zeta_B \tan \lambda_s - \xi_B + 1)s^2 + \zeta_B \tan \lambda_s + \mu(\xi_B - 1)).$$

1184 Since  $\zeta_B < 0$  by definition, this immediately yields static instability (growth of the  
1185 leaning angle yielding *capsizing* of the bike).

### 1186 Asymptotic Stability and Critical Froude Number

1187 We study linear stability of the TMS bicycle with the Lienard–Chipart version of the  
1188 Routh–Hurwitz criterion (Kirillov 2013a). First, compute the Hurwitz determinants  
1189 of the characteristic polynomial

$$\begin{aligned} 1190 \quad h_1 &= \text{Fr}(\zeta_B \xi_H - \zeta_H \xi_B)\zeta_B, \\ 1191 \quad h_2 &= \text{Fr}\zeta_B f, \\ 1192 \quad h_3 &= -\text{Fr}^2\zeta_B^2(\zeta_B - \zeta_H)h, \\ 1193 \quad h_4 &= \text{Fr}^2\zeta_B^2(\zeta_B - \zeta_H)(\tan(\lambda_s)\zeta_B + \mu\xi_H - \mu)h, \end{aligned} \quad (126)$$

1194 where

$$1195 \quad f = -\zeta_B(\zeta_B^2 \xi_H - \zeta_H^2 \xi_B) \tan \lambda_s - \zeta_H(\xi_H - 1)(\zeta_B \xi_H - \zeta_H \xi_B) \mu \\ 1196 \quad + \zeta_B(\zeta_B - \zeta_H)(\zeta_B \xi_H - \zeta_H \xi_B) \text{Fr}^2 + \zeta_B \xi_B (\xi_H - 1)(\zeta_B - \zeta_H) \quad (127)$$

1197 and

$$1198 \quad h = -\zeta_B \xi_B \xi_H (\zeta_B - \zeta_H) \tan \lambda_s - \xi_H (\xi_H - 1)(\zeta_B \xi_H - \zeta_H \xi_B) \mu \\ 1199 \quad + \zeta_B (\xi_B - \xi_H)(\zeta_B \xi_H - \zeta_H \xi_B) \text{Fr}^2 + \zeta_B \xi_B (\xi_H - 1)(\xi_B - \xi_H). \quad (128)$$

The Lienard–Chipart criterion requires that

$$a_4 > 0, \quad a_3 > 0, \quad a_1 > 0, \quad a_0 > 0, \quad h_1 > 0, \quad h_3 > 0.$$

1200 The relation  $h_1 = a_1$  eliminates one of the inequalities and in view of that  $\mu > 0$ ,  
1201  $\zeta_B < 0$ , and  $\xi_B > 0$  yields the following explicit conditions

$$1202 \quad \xi_H > 1 + \zeta_H \tan \lambda_s \\ 1203 \quad \xi_H < 1 - \frac{\zeta_B}{\mu} \tan \lambda_s \\ 1204 \quad \xi_H < \xi_B \\ 1205 \quad \zeta_H > \zeta_B \\ 1206 \quad \text{Fr} > \text{Fr}_c > 0, \quad (129)$$

1207 where the critical Froude number at the stability boundary is given by the expression

$$1208 \quad \text{Fr}_c^2 = \frac{\zeta_B - \zeta_H}{\xi_B - \xi_H} \frac{\xi_B \xi_H}{\zeta_B \xi_H - \zeta_H \xi_B} \tan \lambda_s + \frac{\xi_H - 1}{\xi_B - \xi_H} \frac{\xi_H}{\zeta_B} \mu - \frac{(\xi_H - 1) \xi_B}{\zeta_B \xi_H - \zeta_H \xi_B} \quad (130)$$

1209 that follows from the condition  $h = 0$ .

1210 At  $0 \leq \text{Fr} < \text{Fr}_c$  the bicycle is unstable by flutter demonstrating the *weaving*  
1211 motion (Kooijman et al. 2011)

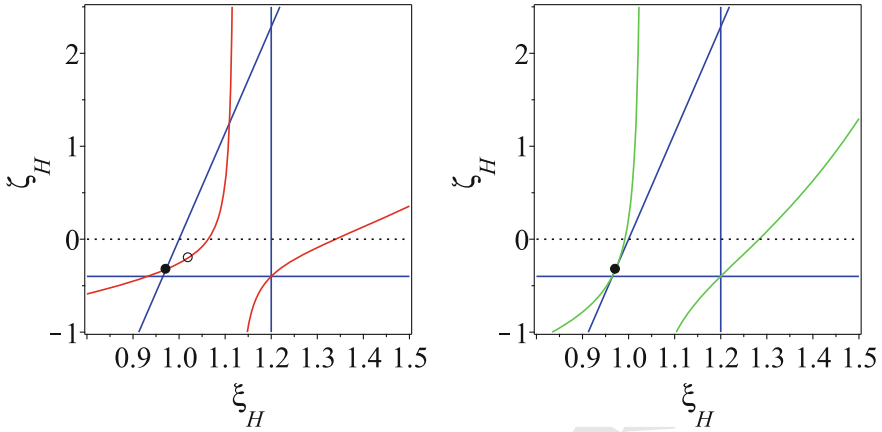
1212 **Critical Fr for the Benchmark Bikes of Kooijman et al. (2011)**

For the design determined by

$$w = 1 \text{ m}, \quad \lambda_s = \frac{5\pi}{180} \text{ rad}, \quad m_H = 1 \text{ kg}, \quad m_B = 10 \text{ kg}, \\ x_B = 1.2 \text{ m}, \quad x_H = 1.02 \text{ m}, \quad z_B = -0.4 \text{ m}, \quad z_H = -0.2 \text{ m}$$

1213 the critical Froude number is

$$1214 \quad \text{Fr}_1 = 0.9070641497, \quad (131)$$



**Fig. 26** For  $w = 1$  m,  $\lambda_s = \frac{5\pi}{180}$  rad,  $m_H = 1$  kg,  $m_B = 10$  kg,  $x_B = 1.2$  m,  $z_B = -0.4$  m (left) stability diagram at  $Fr = Fr_1 = 0.9070641497$  with the circle corresponding to  $x_H = 1.02$  m and  $z_H = -0.2$  m; (right) stability diagram at  $Fr = Fr_{min} = 0.6999527422$ . Black circle denotes a point with the coordinates  $(0.9716634870, -0.3238878290)$  given by (135)

1215 which corresponds to the critical velocity of weaving

1216 
$$v_1 = 2.841008324 \text{ m/s} \tag{132}$$

1217 in accordance with the original result by Kooijman et al. (2011).

For the alternative design determined by

$$w = 1 \text{ m}, \quad \lambda_s = -\frac{5\pi}{180} \text{ rad}, \quad m_H = 1 \text{ kg}, \quad m_B = 10 \text{ kg},$$

$$x_B = 0.85 \text{ m}, \quad x_H = 1 \text{ m}, \quad z_B = -0.2 \text{ m}, \quad z_H = -0.4 \text{ m}$$

1218 the critical Froude number is

1219 
$$Fr_2 = 0.8415708896, \tag{133}$$

1220 which corresponds to the critical velocity of weaving

1221 
$$v_2 = 2.635877411 \text{ m/s} \tag{134}$$

1222 in accordance with the original result by Kooijman et al. (2011). Notice that careful  
 1223 analysis of the Lienard-Chipart criteria for the TMS bicycle proves the existence of  
 1224 just two classes of self-stable TMS bikes that differ by the sign of  $\lambda_s$ , see Austin  
 1225 Sydes (2018).

Editor Proof

## 1226 Finding Designs that Minimize the Critical Fr

1227 Let us fix  $\lambda_s$ ,  $\xi_B$ ,  $\zeta_B$ , and  $\mu$  and plot the stability domain specified by Eq. (129) in  
1228 the  $(\xi_H, \zeta_H)$  - plane at different values of Fr, Fig. 26.

1229 This yields a vertical line  $\xi_H = \xi_B$ , a horizontal line  $\zeta_H = \zeta_B$  and an inclined  
1230 line  $\xi_H = 1 + \zeta_H \tan \lambda_s$  that form a rectangular triangle in the  $(\xi_H, \zeta_H)$  - plane,  
1231 Fig. 26. There is no stability outside of this triangle. On the other hand the condition  
1232  $Fr = Fr_c$  defines two hyperbola-like curves, one of which always passes through  
1233 the right lower corner of the triangle and the other one always passes through a  
1234 point on the hypotenuse of the triangle shown by a black circle in Fig. 26. Solving  
1235 simultaneously equations  $\xi_H = 1 + \zeta_H \tan \lambda_s$  and  $Fr = Fr_c$  we find the coordinates  
1236 of this point to be

$$1237 \quad \xi_H = \frac{-\xi_B}{\zeta_B \tan \lambda_s - \xi_B}, \quad \zeta_H = \frac{-\zeta_B}{\zeta_B \tan \lambda_s - \xi_B}. \quad (135)$$

1238 If we take, for instance

$$1239 \quad \begin{aligned} w &= 1 \text{ m}, & \lambda_s &= \frac{5\pi}{180} \text{ rad}, & m_H &= 1 \text{ kg}, & m_B &= 10 \text{ kg}, \\ x_B &= 1.2 \text{ m}, & z_B &= -0.4 \text{ m}, \end{aligned} \quad (136)$$

1240 the branch of the curve  $Fr = Fr_c$  passing through the point (135) with the coordinates  
1241  $(0.9716634870, -0.3238878290)$  lies partially inside the triangle, Fig. 26(left). The  
1242 area between this part and the hypotenuse is the stability domain, which for the TMS  
1243 bicycle is further restricted by the condition  $\zeta_H < 0$ .

1244 Can we change the design in order to minimize the critical Froude number? If we  
1245 plot the curve  $Fr_c(\xi_H, \zeta_H) = Fr$  at different values of Fr, we will see that the portion  
1246 of its branch passing through the point (135) and lying in the triangle tends to get  
1247 smaller as Fr decreases. At some  $Fr_{min}$  the branch is tangent to the hypotenuse at the  
1248 point (135), and the stability domain disappears, Fig. 26(right).

Therefore, the design specified by the conditions (135) gives the minimum possible Froude number, beyond which the TMS bike becomes stable:

$$Fr_{min}^2 = \frac{(\zeta_B \tan \lambda_s - \xi_B)^2 + \mu}{(\zeta_B \tan \lambda_s - \xi_B)(\zeta_B \tan \lambda_s - \xi_B + 1)} \tan \lambda_s.$$

For instance, if we take parameters as in (136) and use (135) to find  $x_H = 0.9716634870 \text{ m}$  and  $z_H = -0.3238878290 \text{ m}$ , then we obtain the minimal Froude number and the corresponding velocity of weaving

$$Fr_{min} = 0.6999527422, \quad v_{min} = 2.192316351 \text{ m/s}$$

1249 that indeed are smaller then that given by (131) and (132) for the benchmark TMS  
1250 bike in Kooijman et al. (2011).

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