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1 Experimental Optimisation of Power for Large Arrays of
2 Cross-Flow Tidal Turbines

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10 **Abstract**

11 As commercial scale tidal energy devices are shortly to be deployed in the
12 first arrays, the knowledge of how different array layouts perform is a key
13 and under-examined field. Here, the Momentum Reversal Lift (MRL) tur-
14 bine, developed by the University of Exeter, is deployed in five different array
15 layouts utilising up to 15 devices. The use of dynamic turbines allows the
16 inclusion of analysis of the effects of flow direction in the wake.

17
18 The layouts investigated explore the effect of lateral and stream-wise tur-
19 bine spacings as well as differences between staggered and in-line layouts on
20 power. The staggered array with decreased streamwise spacing is shown to
21 have the highest total power per ‘footprint’ area among the layouts tested.
22 For the staggered arrays, increased downstream separation had little effect
23 on total power generated, while decreasing the lateral spacing below 2 rotor
24 diameters decreased the power. The in-line arrays showed a lower power
25 per device but similar total power. It was also shown that increased in-

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Belmont), www.flowavett.co.uk/ (Thomas Davey)

26 flow into a turbine didn't necessarily lead to an increased power extraction.
27 The decrease in power with a decrease in streamwise spacing is in-line with
28 theoretical and CFD predictions.

29 *Keywords:* Renewables, Tidal Energy, Arrays, Scale Testing, Wake
30 Interactions, Physical Modelling

31 **1. Introduction**

32 Tidal energy is considered a potentially significant contributor to the
33 UK's energy mix, with estimates ranging from 15.7 TWh/year [1] and 20.6
34 TWh/year [2] which would account for 4.6% to 6.1% of the UKs electricity
35 requirements [3]. With several commercial scale prototypes tested in isola-
36 tion, the focus of the hydrodynamic research has shifted towards both second
37 generation technologies optimised for specific environments and the interac-
38 tion of devices in arrays. This work focuses on the novel Momentum Reversal
39 Lift (MRL) turbine designed by the University of Exeter in conjunction with
40 Aquascientific Ltd, using up to 15 scale models in a variety of array con-
41 figurations to assess the effect of layout and spacing on power output. The
42 optimum spacing for turbines is critical to extract maximum power and to
43 predict loadings in arrays and is a field which has had limited experimental
44 testing given the stage of commercial array projects. This work builds upon
45 the work by: Janssen and Belmont [4] and Ordonez *et al.* [5] assessing the ex-
46 tractable power and wake evolution in both an individual turbine and a four
47 turbine array, as well as CFD work on both device and array optimisation
48 [6, 7, 8, 9].

49 **2. Background**

50 *2.1. Tidal Turbine Arrays*

51 Flow through an array of turbines is highly complex, due to the nature
52 of tidal energy sites [10, 11] and the interaction of turbine wakes [12, 13, 14].
53 Energy extraction devices in tidal channels can in theory utilise high global
54 blockage ratios, i.e. the ratio of total turbine swept area divided by the
55 channel cross-sectional area. By doing so they are in theory able to extract a
56 greater percentage of available power than in an open channel, increasing the
57 Lanchester-Betz ratio of 0.593 to 0.798 [15]. Staggering devices in rows use
58 upstream turbines, which provide local blockage, to accelerate flow between
59 them, so that the downstream turbines have a higher inflow velocity. The use
60 of these arrays has been theorised to increase extractable power for certain
61 downstream spacings [16]. To this end, several studies have focused on wake
62 evolution and downstream mixing [17, 12] with the goal of maximisation
63 of the local available power in the flow, which is expected to increase with
64 the cube of the flow speed. In addition, increasing the downstream spacing
65 between rows allows the wake after the first row to mix with the bypass and
66 free-stream flow to recover to a higher value, increasing the inflow to the
67 subsequent row. In principle it is possible that as a turbine causes bypass
68 flow acceleration around it, the downstream flow can be higher than the
69 upstream flow despite kinetic energy being extracted by the turbine, as the
70 total energy is conserved through a loss of head.

71 Local inflow velocity is not the only factor that will effect the extractable
72 power for a turbine. It is generally agreed that a more turbulent flow for the
73 equivalent velocity will induce less lift which in turn will reduce the power

74 that a turbine can extract. However, these effects are complex and depend
75 upon the scales of turbulence in question [18, 11]. Turbines are also sensitive
76 to flow direction which changes the effective angle of attack of the lifting
77 surfaces used as the prime mover to extract power. A theoretical exploration
78 of array layouts is given in Draper *et al.* [19].

79 Tank testing of tidal arrays to date has been limited, due in part to
80 the difficulty in finding appropriate testing facilities. Myers and Bahaj [20]
81 investigated array layouts through porous disks in a shallow tank with the
82 aim of maximising the flow acceleration through the array. Draper *et al.*
83 [17] conducted a similar study with the focus on the evolution of the wakes.
84 However, these studies investigated flow acceleration and not the extracted
85 power. Cooke *et al.* [21] used the thrust on the disk along with a near wake
86 velocity measurement to infer the power, finding power coefficients per disk
87 of ≈ 0.1 based on the global channel flow.

88 More complex flow effects caused by a dynamic turbine model (such as
89 rotational effects) are not present in porous disk experiments. Stallard *et al.*
90 [12] have investigated the layouts of up to ten three-bladed Horizontal Axis
91 Turbines (HATs), with lateral spacings of $1.5D$ and $2D$ (where D is a rotor
92 diameter) over two rows. They showed velocity deficits of 80% across the
93 turbines, $2D$ downstream by which point the wakes had begun merging and
94 were fully merged by $4D$. The widths of the individual wakes were seen to
95 expand to a maximum of $2D$ by $10D$ downstream. Power was measured via
96 a dynamometer but the variation of power with array layout is not reported
97 as the work focuses on the wake evolution.

98 Another HAT device study using two in-line three-bladed dynamic tur-

99 bine models at the deeper (2m) IFREMER tank was presented by Mycek *et*
100 *al.* [22]. They showed a drop in turbine performance in the downstream de-
101 vice compared with a single turbine for a series of downstream spacings. They
102 also show that in high turbulence environments, increased downstream mix-
103 ing leads to lower velocity deficits increasing downstream device performance
104 suggesting that velocity magnitude is more important than turbulence.

105 2.2. The Momentum Reversal Lift (MRL) turbine

106 The momentum reversal lift turbine, shown in Figure 1, was conceived by
107 the University of Exeter and Aqua Scientific Ltd. This cross flow horizontal
108 axis turbine has three symmetrical blades, each of which rotate through 180°
109 for a full rotation of the shaft. The turbine is unique in that it utilises both
110 lift and drag (momentum reversal) in order to generate rotational velocity
111 in the prime mover. The turbine is designed primarily for shallow estuaries
112 where the cross flow design will allow for high blockage ratios relative to a
113 circular swept area device thus increasing power output. For a comprehensive
114 overview of the turbine design see [4] and [7].

115 Initial experiments utilised both a balsa wood model in a wind tunnel and
116 a metal turbine in a flume. These devices both showed promising maximum
117 c_p values of ~ 0.5 , however these were in high blockage environments, 0.66 in
118 the case of the flume [4]. The flume results were compared with a Immersed
119 Body Force (IBF) CFD model utilising Large Eddy Simulation (LES), which
120 showed good similarity with the experimental data, particularly for the lower
121 torque range [6, 7].

122 The scale model turbines used here were previously tested in a wider
123 flume at the IFREMER facility where some initial array configurations were

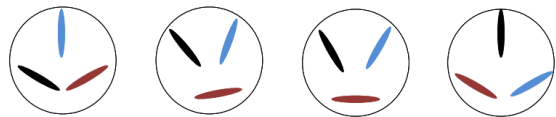
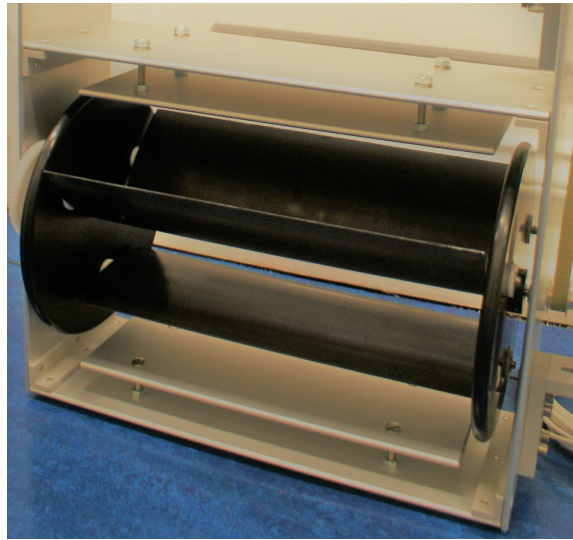


Figure 1: MRL turbine prototype used in the experiments described in this paper (top). The three bladed rotates around the shaft as it can be observed in the bottom figure. In the initial set up one blade is set completely flat while the others two start with a set angle.

Table 1: Array spacings in multiples of turbine diameters (D) and c_p of the centre turbine from CFD [8].

Downstream spacing	Lateral spacing	c_p
10D	3D	0.40
10D	6D	0.30
15D	3D	0.45
15D	6D	0.32

124 trialled [5]. The c_p values in a relatively open channel dropped to 0.14. In
 125 these tests, the wake had not recovered to the upstream velocity magnitudes
 126 by $20D$ downstream. In addition there was evidence of asymmetry in the
 127 wake of a single turbine which is not evident in similar three bladed HAT
 128 testing [12].

129 There have also been studies using the IBF model to investigate trends
 130 in array layouts. These have investigated both changing spacings [8] and
 131 varying resistive body force [9]. The former looked at a test matrix of two
 132 lateral and two downstream turbine separations for three rows of turbines
 133 in a 2-3-2 formation, with a blockage ratio of 0.044. Table 1 presents the
 134 spacings and c_p values. It found the highest c_p values were in the narrower
 135 (lateral) and longer (stream-wise) arrangement, with an increase in lateral
 136 spacing causing a decrease in c_p . In addition, while it was found that the
 137 blockage of the first row of turbines caused an acceleration into the middle
 138 turbine in the second row, this did not necessarily mean an increase in c_p .

139 *2.3. Aims and objectives of current study*

140 A key goal in tidal energy research is to determine the extent to which
141 array layouts affect the extractable power. Arrays can be staggered or in-line
142 and the lateral and downstream spacings between devices can all be varied.

143 In this work five array layouts are trialled with the aim of maximising
144 power output and exploring the extent of the influences of spacing and lay-
145 out on both power and flow. The evolution of the wake is also investigated
146 through two sets of streamwise lines of velocity measurements, to allow com-
147 parison with previous studies. Array layouts are chosen to highlight the effect
148 of changing a single metric and to align with previous CFD modelling work.

149 **3. Test Set-up and Methodology**

150 *3.1. Overview of the FloWave test tank*

151 The array testing was conducted at the FloWave Test Tank facility. This
152 25 m diameter circular tank has the facility to provide combined wave and
153 current, with wavemakers located around the entire circumference. The nom-
154 inal test area has a diameter of 10 m. The tank is capable of generating
155 currents upwards of 1.6 ms^{-1} , using 28 drive units mounted in a plenum
156 chamber below the test floor. Turning vanes mounted below and in front
157 of the wavemakers direct the current across the tank. These turning vanes
158 incorporate porous screens to provide flow conditioning and prevent debris
159 ingress to the plenum chamber [23]. This facility was selected due to the
160 large test area required for array testing.

161 In order to create an approximately uniform current across the test area
162 of the circular tank, the impeller units on either side of the required current

163 direction (i.e. both the upstream and downstream) are utilised. These are
164 driven at varying speeds to produce the required current corresponding to
165 the desired test velocity. This results in an ‘hour-glass’ shaped flow profile in
166 the xy plane [24]. Previous measurements in the tank have shown the flow to
167 be highly symmetrical about the stream-wise (x) axis [25]. However, in the
168 streamwise direction there is some variation in both the mean and turbulent
169 flow parameters. The velocity varies approximately linearly with depth but
170 has a very shallow gradient compared with measurement at full scale sites
171 [26].

172 3.2. Turbine Models

173 The small scale model utilised here is shown in Figures 1(bottom) and
174 2. The model has three 300mm wide (L) and 95mm chord length blades
175 mounted on a planetary gear system. The distance from the primary shaft
176 to the centre of rotation of each blade is 164mm. The cross sectional height
177 of the turbine (D) is 200mm giving a ‘swept area’ (A) of 0.06 m². Note
178 that this cross sectional area is not entirely swept by the blades due to the
179 change of angle through the rotation, but since the design prohibits mounting
180 another device within this cross sectional area the adopted definition was
181 deemed the most appropriate. Ground force and Pelton effect plates, which
182 act to increase the flow rate through the swept area, were added during early
183 testing in order to increase rotational velocity [4]. Power take off is provided
184 by a 2.5 kSt oil-filled dash-pot connected to the primary shaft by a 2:1 geared
185 pulley and the angular velocity (ω) is measured via a 24 tabbed disk mounted
186 on the primary shaft which passes through a Hall effect sensor. Part of the
187 previous work focused on finding the gear ratio and damper which produced

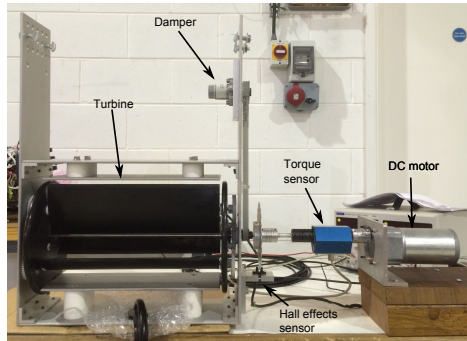


Figure 2: Model used in testing attached to torque calibration rig.

188 the highest power coefficient (c_p) [5].

189 The scale models utilised in these tests were by necessity relatively inex-
 190 pensive to allow a relatively large number to be constructed and as such, there
 191 was some variation in angular friction from one turbine to the next. This is
 192 detailed further in Section 4.1. Therefore, it was necessary to calibrate the
 193 torque for each individual turbine and its damper for a range of rotational
 194 speeds. The rig for calibrating this torque curve is shown in Figure 2. It
 195 features a 2 Nm rotary torque transducer attached to the primary shaft and
 196 the system is driven by 27 W DC motor. For each turbine a measurement
 197 was taken every 2 V from 4 V to 24 V which provided a range of rotational
 198 speeds up to approximately 140 rpm. At each setting the rotational velocity
 199 (ω) and torque (τ) were collected via a National Instruments data acquisition
 200 system and recorded via Labview.

201 In order to mount the turbines in the tank, an adaptable frame design
 202 was developed that would allow for relatively quick changes between array
 203 configurations. Ideally the turbines would have occupied a greater percentage
 204 of the channel depth increasing the global blockage ratio. However, with the

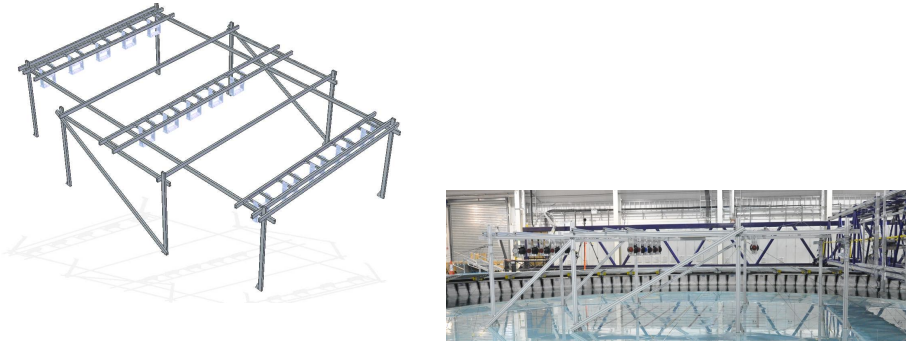


Figure 3: The frame for mounting the turbines, which was designed for high stiffness and for ease of reconfiguring turbine locations.

205 working depth of FloWave being 1.9 m this was not possible. The turbines
 206 were mounted close to the free surface as this would provide realistic blockage
 207 effects at one boundary as per the design specification [6]. Figure 3 shows
 208 the frame which was constructed from 45×90 mm cross section aluminium
 209 extrusion and Figures 5 and 6 highlight its location in the tank. In order to
 210 increase the rigidity of the frame, guy lines were tied to the tank floor from
 211 each of the vertical poles. In order to track any vibration of the frame and also
 212 the position of the turbines, a Qualisys tracking system was employed. This
 213 showed that under load the maximum movement in the frame was < 4 mm.

214 3.3. Array Layouts

215 Five different array layouts were tested in order to assess the effect of:
 216 lateral separation, stream-wise separation and in-line and staggered rows of
 217 turbines. To this end a base-case array layout was selected. Each of the other
 218 arrays vary one parameter from this base-case. The spacing are relative to
 219 the turbine dimensions where D is the cross flow height (equal to 200mm)
 220 and L is the cross-stream width (equal to 300mm).

Table 2: Overview of the array layout edge to edge spacings used in the tests. Bracketed numbers indicate the number of devices in each row.

Layout code	Staggered /In-line	x spacing	y spacing
Baseline	Staggered (4-5-4)	$14D$	$2L$
A	Staggered (4-5-4)	$10D$	$2L$
B	Staggered (4-5-4)	$14D$	$1.5L$
C	In-line (5-5-5)	$14D$	$2L$
D	In-line (5-5-5)	$14D$	$1.5L$

221 Each of the arrays featured three rows of turbines. All of these configu-
 222 rations included five turbines in the middle row. The staggered layouts had
 223 four turbines in the front and back with these spaced at the mid points in
 224 the transverse (y) between the middle row turbine locations. For the in-line
 225 arrays the turbines were mounted in the same y locations in each of the three
 226 rows, each containing five devices. The staggered arrays featured 13 turbines
 227 and the in-line 15 devices to accommodate this. Table 2 provides an overview
 228 of the array configurations which were investigated.

229 3.4. Measurement strategy

230 In addition to the rotational velocity of each turbine, the flow velocity
 231 within each array was investigated. In order to do this a Nortek Vectrino was
 232 utilised to measure the flow velocities. This instrument is capable of measur-
 233 ing at 100 Hz resolving the velocity into three Cartesian components. The
 234 Vectrino was mounted to the tank’s instrumentation gantry on an adjustable
 235 frame, allowing it to translate in the x , y and z directions. As with any

236 acoustic sensor, measurements are subject to uncertainty due to noise. This
237 Doppler noise is generally agreed to be zero mean, thus mean velocities are
238 computed over a large number of samples (6000) to reduce the uncertainty.
239 This noise will bias the turbulence intensity value high but this is corrected
240 for by measuring the variance due to noise using the method described by
241 Richard *et al.* [27].

242 To maximise the extent of velocity information from throughout the array,
243 symmetry about the x axis was assumed, based on research that showed the
244 undisturbed tank flow to be symmetrical [25] and that the frame and turbine
245 array layouts were also symmetrical. All measurements were taken at a
246 fixed depth at the midpoint of the turbine swept area. The co-ordinates
247 for measurements are given in terms of their (x,y) position in mm from the
248 centre point of the tank, where x is positive upstream of the centre.

249 For each array, one measurement (u_0) was taken significantly far upstream
250 as to not be affected by the array as a reference. From there, one measure-
251 ment was taken $2D$ upstream of each turbine on the $y \leq 0$ side of the array,
252 this is referred to as the u_{in} measurement. In addition the development of
253 the flow along the x axis is measured at $y = 0$ and at $y_{spacing}/2$. These were
254 taken every $2D$ downstream (or as close as possible where the horizontal
255 frame beams were obstructing access) from the first row to the last row of
256 turbines. The positions of the array relative to the stream-wise direction of
257 the tank were varied to minimise effects of the support struts on the flow.

258 Tests were carried out at the nominal, scaled current speed of 1.2 ms^{-1} ,
259 although higher speeds were used on occasion to ensure all turbines cut-in,
260 before being reduced to 1.2 ms^{-1} again for testing.

261 Each velocity vector is the result of one minutes measurement. This
 262 magnitude has previously been shown to be a statistically stationary period
 263 [26] and a mean value for each vector (e.g. \bar{u}) is reported. In addition to this,
 264 other metrics are used to give more information about the behaviour of the
 265 flow: the mean stream-wise velocity deficit (Δu), the turbulence intensity for
 266 each vector (i.e., I_u), the heading (θ), and the pitch (ψ) which are defined in
 267 the following equations:

$$\Delta u = 100 \cdot \left(1 - \frac{\bar{u}}{u_0} \right) \quad (1)$$

$$I_u = \frac{\sqrt{\sigma_u}}{\bar{u}} \quad (2)$$

$$\theta = \tan^{-1} \left(\frac{\bar{v}}{\bar{u}} \right) \quad (3)$$

$$\psi = \tan^{-1} \left(\frac{\bar{w}}{\bar{u}} \right) \quad (4)$$

268 3.4.1. Power

269 In order to compare the different arrays an appropriate metric must be
 270 defined. For the individual turbines the power extracted is given in Equation
 271 5, where ω is the angular speed (in rad/s) and τ the torque of the turbine at
 272 that angular velocity calculated via the calibration curve detailed in section
 273 4.1. The local power available in the flow (i.e. available to a specific device)
 274 is defined via Equation 6, where ρ is the fluid density and A the swept area
 275 of the turbine.

$$P_{turbine} = \omega\tau \quad (5)$$

$$P_{available} = 0.5\rho Au_{in}^3 \quad (6)$$

276 This allows the calculation of the power coefficient of the turbines, defined
 277 via Equation 7:

$$c_p = \frac{P_{turbine}}{P_{available}} \quad (7)$$

278 Note that the central goal of this study is to maximise power output.
 279 Thus, in order to compare the power captured by each of the array layouts,
 280 three power metrics are used: P_{total} i.e. the sum of the mean power from each
 281 turbine, P_{mean} the mean output from each individual turbine and finally $\frac{P_{total}}{m^2}$
 282 the power per square meter of the array based on the total xy ‘footprint’ area
 283 of the array configuration.

284 For tidal and wind turbines the power is often expressed as a function
 285 of the Tip Speed Ratio which is the ratio of the velocity at the blade tip to
 286 the velocity of the inflow fluid. As the MRL turbine tip speed is difficult to
 287 define, here the Blade Speed Ratio (BSR) is used as an equivalent. BSR as
 288 defined in [5] as:

$$BSR = \frac{\omega R}{u_{in}} \quad (8)$$

289 where ω is the angular velocity in rad/s and R is the radius to the axis of
 290 blade rotation.

291 3.5. *Scaling and Blockage*

292 As there is no prototype scale device against which to scale the MRL
 293 turbine, the depth and flow speeds of a typical tidal site of 50 m and 3 ms^{-1}

294 are considered respectively. These values do not represent estuary conditions,
 295 but given the depth of the test tank relative to the model turbine, this was
 296 deemed appropriate for this test, where the tank depth is significantly greater
 297 than the turbine diameter.

298 The two main scaling factors in tidal arrays are the Reynolds number, a
 299 ratio of the momentum to viscous forces, and the Froude number, a ratio of
 300 the inertia to the gravitational effects on the flow. These ratios are defined
 301 in equations 9 and 10 where ρ is the density, l is a characteristic length and
 302 g is the gravitational field strength [17]. For the same fluid and gravitational
 303 forces these two dimensionless quantities can not be equally scaled. How-
 304 ever, flow conditions are required to be within the same regimes, i.e., fully
 305 developed turbulence and sub-critical [28].

$$Re = \frac{\rho ul}{\mu} \quad (9)$$

$$Fr = \frac{u}{\sqrt{gl}} \quad (10)$$

306 If l is taken to be the channel depth and g is taken to be constant at
 307 9.81 ms^{-1} , the Re and Fr numbers for this test and for the nominal site are
 308 given in Table 3. These numbers are in the range of those in similar work
 309 [17].

310 Whilst some authors have envisaged that Froude number has minor in-
 311 fluence in power and thrust (both increase about 3% according to [29]), the
 312 discrepancy between Reynolds numbers between prototypes and full scale
 313 devices has large effects in the performance of a tidal turbine. In Mycek
 314 *et al.* [30], Reynolds numbers from 1.4×10^5 to 4.2×10^5 were used in the

Table 3: Comparison of scaling parameters

	Re	Fr
FloWave	2.4×10^6	0.32
Full Scale	1.3×10^8	0.09

315 experimental campaign. It was demonstrated that the c_p of a turbine can in-
 316 crease by about 10% when working at larger flow velocities and hence larger
 317 Reynolds numbers. This increase is somehow to be expected, Mason-Jones
 318 *et al.* [31] suggested that in order to reach Reynolds independence, Reynolds
 319 numbers higher than 3×10^5 should be contemplated in small scale test cam-
 320 paigns. However, this insensitivity of Reynolds number could be dependent
 321 on the aerofoil shape but according to the authors knowledge there is no ev-
 322 idence to prove it. As it has been envisaged by Selig *et al.* [32], wind tunnel
 323 tests have demonstrated that the magnitude of lift on thick aerofoils can be
 324 increased slightly when increasing Reynolds numbers from 1×10^5 to 5×10^5 .
 325 However, the effects on drag will be severely, in some cases an increase of
 326 50 - 80% was observed at angles of attack between 0 to 10 degrees. This
 327 proportion depends on the type of aerofoil shape, in this case the S822 was
 328 taken as an example. Wind tunnel tests studies will need to be considered
 329 in the next development stages of the MRL turbine due to the constant and
 330 different changes in angle of attack related to each of the blades.

331 As previously stated the blockage for this test is relatively low compared
 332 with early tests of this device and other arrays. The swept area of the
 333 staggered arrays are $9 \times A$ which is: $0.2 \times 0.3 \times 9 = 0.54 \text{ m}^2$. The tank area
 334 at the mid point as $25 \times 1.93 = 48.25 \text{ m}^2$. This leads to a global blockage

335 ratio of 0.010 for the staggered arrays and of 0.005 for the in-line arrays with
336 a swept area of $5A$.

337 **4. Results**

338 *4.1. Turbine Calibration*

339 The calibration results from each of the turbines with dampers installed
340 are presented in Figure 4. The dashpot's resistance changes with temperature
341 which is related to ω hysteretically. Thus, the calibration was repeated in
342 ascending and descending ω to capture this effect. As can be seen, there is
343 a spread of values with a maximum difference at the highest voltage used
344 of 0.26 Nm indicating a significant degree of variation in damping between
345 devices. It can be noticed that the turbine 3A, which used an older damper
346 of the same specification, showed the lowest resistance, indicating that the
347 performance of these devices in this installation were decreasing over time or
348 with use.

349 Multiple types were trialled to this data (using the downward calibration
350 curve as all turbine measurements were taken at established speeds). The
351 power law gave the highest goodness of fit values thus this fit type was
352 adopted. The form of this curve is given in Equation 11, where a and b are
353 constants defined individually for each turbine.

$$\tau = a \cdot \omega^b \quad (11)$$

354 *4.2. Base-Case*

355 As a large quantity of data was collected for each array, greater detail
356 is provided for the base-case layout, which will provide values which can be

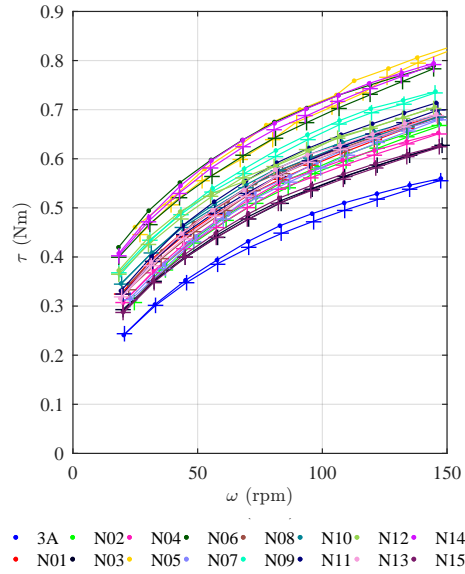


Figure 4: Results of individual turbine torque calibrations. Dots mark the curve for increasing ω and crosses for decreasing.

357 contrasted with other layouts.

358 Figure 6 shows the positions of the turbines, the positions of the vertical
 359 pillars, the velocity measurements taken during the test, as well as the ve-
 360 locity measurements in the tank with no devices installed taken from Noble
 361 *et al.* [25]. This gives an illustration of where measurements were taken and
 362 the effect of the turbines on the flow velocity. These measurement locations
 363 do not capture the effects of the vertical support poles of the frame on the
 364 flow during the testing, which were observed, visually, to be significant.

365 4.2.1. Velocity deficit and turbulence intensity

366 Focusing on the two sets of stream-wise velocity measurements, Figure 7
 367 shows the evolution of the flow through the turbines. As can be seen there is

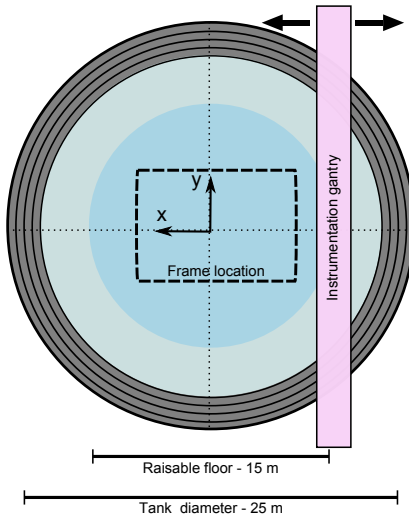


Figure 5: Illustration of test area, detailed in Figure 6, in dotted line and its position within the FloWave tank. Co-ordinate system is given from the tank centre.

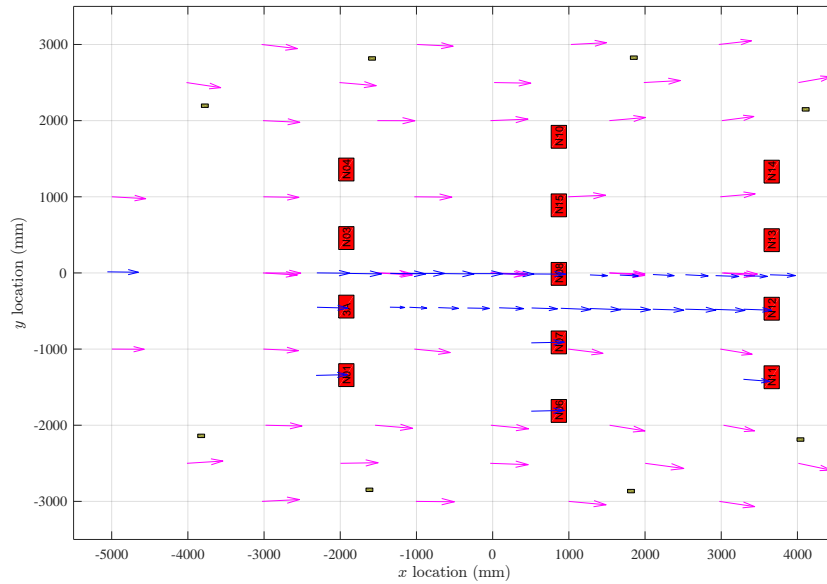


Figure 6: Quiver plot for the base-case array. Blue arrows represent in-situ velocity measurements, the pink arrows represent the ‘natural’ flow in the tank were there no obstructions. The large red rectangles represent the turbines and the small grey ones the vertical frame poles.

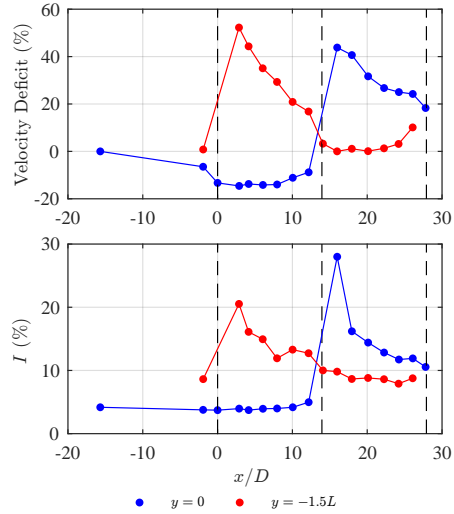


Figure 7: Velocity deficit and turbulence intensity of flow as it propagates through the base-case array. The dashed lines indicate the location of the turbine rows. Flow moving from left to right.

368 an acceleration from the upstream measurement point due to the tank flow
 369 geometry then an acceleration through at the centreline through the first row
 370 of turbines. There is a large decrease in velocity of 52% as the flow passes
 371 through a turbine with a similar decrease at $y = -1.5L$ over the turbine in
 372 the middle row of 54%. There is a corresponding jump in turbulence intensity
 373 at each row of 12% and then 26%. It is interesting to note that, although the
 374 velocity deficits are approximately equal, in each case the downstream jump
 375 in I is significantly greater. Downstream of the turbines, the velocity and I
 376 start to recover towards their upstream values.

377 4.2.2. Velocity direction

378 A phenomenon which became apparent during testing was a standing
 379 surface wave downstream of the first row (Figure 8). This effect had been

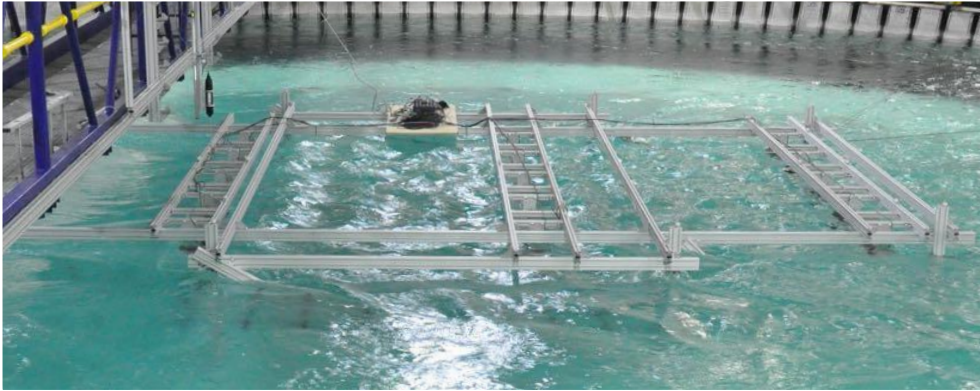


Figure 8: Evidence of ‘standing wave’ effects in wake. Flow moving from left to right.

380 previously noted in other experiments and predicted by CFD [17, 4].

381 The pitch and heading angles as the flow propagates through the rows of
382 turbines are presented in Figure 9. The heading angles are relatively small
383 ($< 5^\circ$) throughout. In the majority of cases, there is a $\sim 4^\circ$ shift of the flow
384 to the left each time the flow passes through a row of turbine. This was
385 only observed for the centre-line measurements in the base-case array. The
386 measurements of the pitch angle of the flow are an order of magnitude greater
387 than those of the heading. The measurement resolution is not high enough
388 to capture properly the sinusoidal pattern visually observed. However, there
389 is evidence of this effect between the first and second turbine rows. Each
390 time the flow passes the turbine there is an upward shift in the pitch angle,
391 which corresponds to the rotational direction of the turbine. The flow in the
392 main has a small negative pitch angle, i.e. a downward trajectory.

393 *4.3. Turbine Performance*

394 The mean and standard deviation of individual turbine ω values for the
395 base-case array are given in Figure 10. A large variation in values which

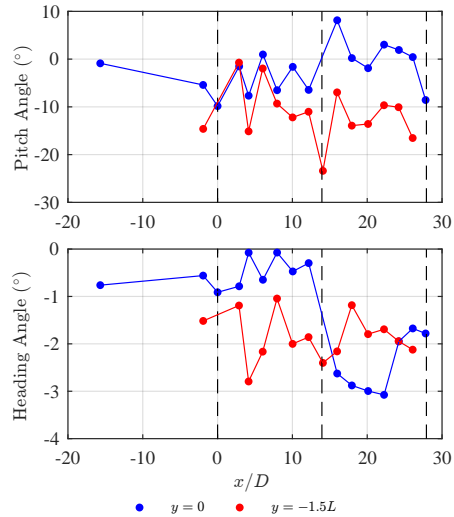


Figure 9: Results of velocity direction through the base-case array. Flow moving from left to right.

396 is due to the variability of both inflow velocity through the array and the
 397 damping of the individual turbines can be seen.

398 In order to examine the variation of damping between devices, the angular
 399 velocity is converted to a c_p value through equations 5, 6, 7 and 11. Limiting
 400 the results to those for which the inflow is directly measured (rather than
 401 inferred through the assumption of a symmetrical array) the c_p for the Blade
 402 Speed Ratio (BSR) is shown in Figure 11. This analysis uses Array A as
 403 there was an additional inflow measurement available.

404 It can be seen that the c_p values for the turbines in the front and mid-
 405 dle rows are approximately linear on a positive gradient, suggesting power
 406 capture increases as turbine damping decreases. However, the two back row
 407 values do not conform to this trend. These results suggest that the down-
 408 stream turbines with inflows from the wakes of the upstream turbines are

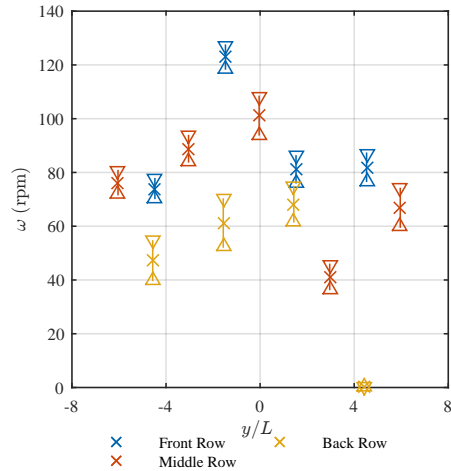


Figure 10: Base-case array mean and standard deviations of ω .

409 less predictable than those upstream.

410 The $\overline{u_{in}}$, $\overline{\omega}$ and \overline{P} for each turbine in the base-case array are presented
 411 in Figure 12. It is reiterated that the $\overline{u_{in}}$ measurements are only measured
 412 for half of each array layout and the $y > 0$ values are inferred by assuming
 413 symmetry about the x axis. One aspect that is evident is the lack of symme-
 414 try in the measured ω and P values. In the top plot it can be seen that the
 415 middle row has the highest inflow velocities due to the blockage of the flow
 416 through the upstream turbine row. However, this higher u_{in} does not corre-
 417 spond to higher rotational velocities in the middle plot. As the turbulence
 418 intensity is also constant into both rows it is inferred that it is the vertical
 419 component of the velocity (i.e. the pitch of the flow as per Figure 9) that is
 420 affecting the reduction in ω in the middle row devices. The power is highly
 421 variable between devices in the two front rows, with the back row presenting
 422 the lowest variation and generally lowest response.

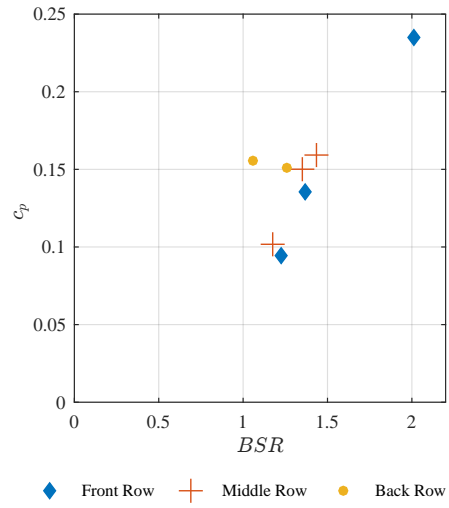


Figure 11: c_p variation between turbines for array A.

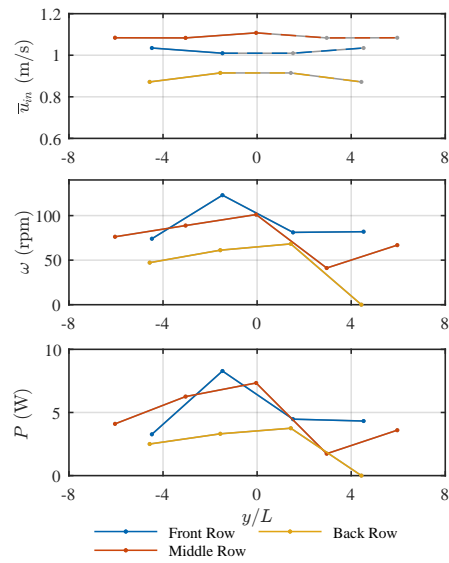


Figure 12: $\overline{u_{in}}$, $\overline{\omega}$ and \overline{P} for each turbine in the base-case array.

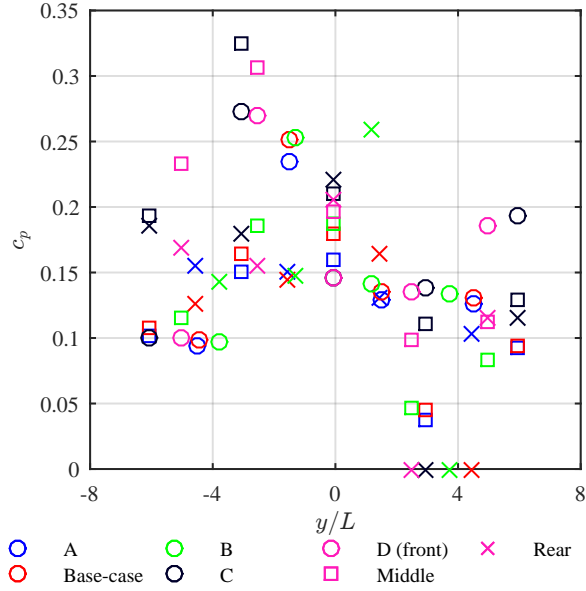


Figure 13: All c_p values across the 5 arrays. Circles mark the front row, squares the middle and crosses the rear row, with colours indicating the array layout.

4.4. Inter Array Comparisons

The $\overline{u_{in}}$, $\overline{\omega}$ and \overline{P} values for each turbine for the four additional array layouts are given in Figures 14, 15, 16 and 17, with the base-case included in grey for comparison.

For the decreased stream-wise separation (Figure 14) the trends are very similar to those in the base-case array, with highest flow velocities being into the middle row of turbines due to acceleration between the first row turbines and similar ω values. In the decreased lateral spacing case (Figure 15) the velocity into the second row is very similar to the front row, suggesting that at this spacing the first turbine row as a whole has a blockage ratio that is causing more flow to divert round the sides of the array. The reason of this diversion is because the flow has reached its maximum choking capacity

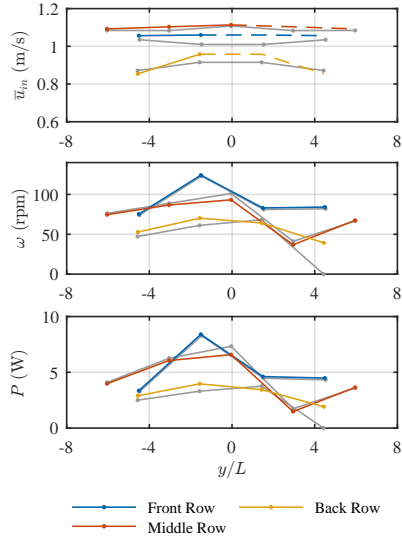


Figure 14: \bar{u}_{in} , $\bar{\omega}$ and \bar{P} for array layout A, with base-case results in grey.

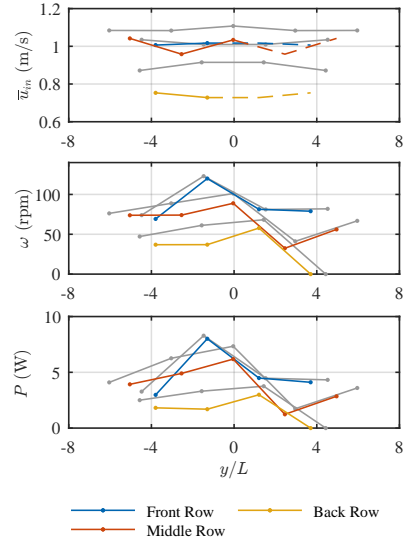


Figure 15: \bar{u}_{in} , $\bar{\omega}$ and \bar{P} for array layout B, with base-case results in grey.

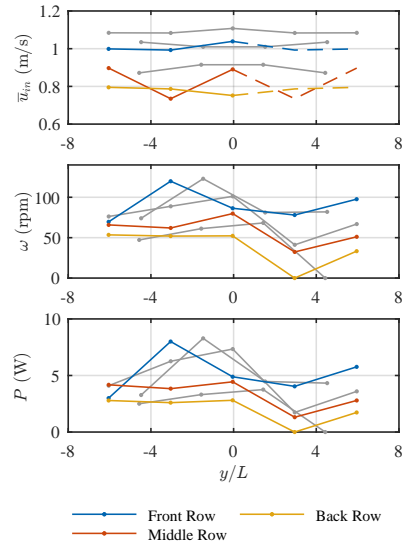


Figure 16: \bar{u}_{in} , $\bar{\omega}$ and \bar{P} for array layout C, with base-case results in grey.

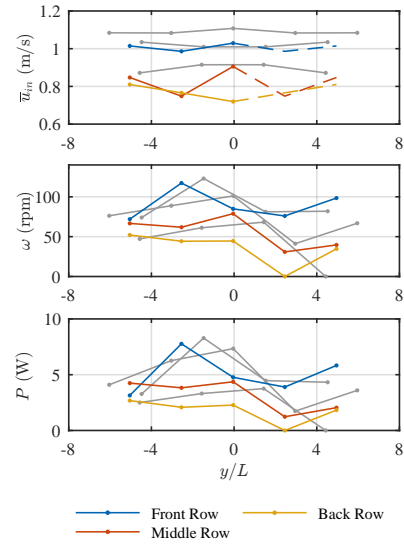


Figure 17: \bar{u}_{in} , $\bar{\omega}$ and \bar{P} for array layout D, with base-case results in grey.

435 which as defined by Nishino and Willden [15], is the reduced ow through an
436 entire array. For all three of the staggered arrays there is little asymmetry of
437 P and ω results around the stream-wise axis in any of the rows. The trends
438 in these metrics are, however, consistent between the three staggered array
439 layouts.

440 In the two in-line arrays (Figures 16 and 17), u_{in} is highest in the front
441 row as would be expected. However, there is evidence of the wake from the
442 upstream posts affecting the inflow to the middle row at the second turbine
443 from each edge in both array spacings, leading to those turbines having lower
444 u_{in} than the back row. This is not directly reflected in the ω and P results
445 which decrease row by row.

446 This asymmetry of power and angular velocity across the 5 arrays (seen in
447 Figures 14 to 17), may in part be due to the direction of the flow in the tank
448 which forms an hour glass shape [26]. Although through the majority of the
449 test area the flow is uniform, at the edges a combination of this inward flow
450 and the wake of the vertical poles may cause flow velocities from different
451 directions to affect the turbines. However, the effect is relatively constant
452 across the arrays suggesting that power comparisons are valid.

453 It has been observed in the literature that there are some effects on the
454 power captured depending on the direction of the flow. However, Figure
455 9 shows that the heading angles in these tests are less than 4 degrees and
456 therefore it was envisaged that such small angles will not have a significant
457 impact in the power calculations. This supposition was due to the research
458 presented by Galloway *et al.* [33] who showed that power reductions only
459 become apparent with heading angles above 7.5 degrees. Also, the turbulence

460 intensities at the outer turbines are greater than that at the u_{in} locations for
461 the inner turbines, with range of values increasing from 4 - 14% to 14 - 25%.
462 Figure 18 shows how the velocity deficit and I values vary in the in-line
463 Array C. For the first and second turbine row there is a Δu of 51% and 45%
464 respectively, with a recovery to within 13% of the upstream value by $12D$.
465 The sharp peak in I midway between the front and centre row exists at both
466 spacing but it is not clear what the driving factor is. It may be a mixing point
467 for the turbine wakes, but the spatial resolution of velocity measurements is
468 insufficient to analyse.

469 In order to assess which array is the best, three power related metrics are
470 employed. The results of power for each configuration are given in Table 4.
471 In terms of total power, four of the five arrays show similar values with the
472 narrow staggered Array B being the lowest and the base-case the highest.
473 The total power per device is dominated by the two 0.6m spaced staggered
474 arrays, which both have thirteen devices, two less than the in-line arrays. In
475 terms of power for a given footprint-area, it is Array A (decreased stream-
476 wise spacing) that gives the best results.

477 Figure 13 shows all the c_p values across the five array configurations.
478 The highest value is 0.32 which is for the middle row of Array C. The mean
479 value, discounting any non-rotating devices, is 0.15. There is evidence of
480 the asymmetry in the arrays with the values on the left ($y/L < 0$) being
481 significantly higher than those on the right, which is relatively consistent
482 across the array layouts.

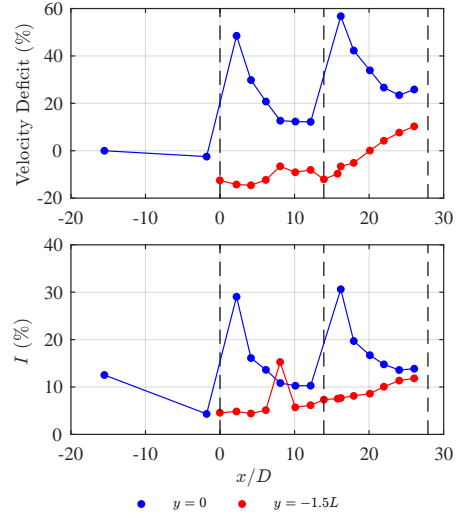


Figure 18: In-line array ‘C’ showing the Velocity deficit and I across two sets of turbines.

Table 4: Power parameters from each array layout.

	Array				
	Base-case	A	B	C	D
Total P (W)	55.4	54.8	47.0	54.8	52.0
P per Device (W)	4.3	4.2	3.6	3.7	3.5
P per xy area (W/m^2)	2.8	3.8	2.8	2.7	3.1

483 5. Discussion

484 The three power metrics show that Array A has the highest power per
485 footprint area and the base-case the highest total power. Comparing the
486 trends from the base-case with similar trends in the CFD work [7, 8]: the
487 decreasing lateral separation had the opposite effect on power in the CFD
488 study compared to the current experiments. It should be noted that in
489 the experimental work these separations are $2L$ and $1.5L$ (due to practical
490 limits of the tank) whereas in the CFD these are $2L$ and $4L$. Malki *et al.*
491 [34] suggests that a lateral spacing of $1D$ represents a critical ratio after
492 which decreased lateral separation decreased the bypass flow. This reverse of
493 trend between experimental and CFD results is likely due to the experiments
494 operating in the region of this the critical array ‘choking’ capacity. Only
495 two lateral separations were tested in this work and thus a useful follow
496 up study would focus on finding the exact point of lateral separation for
497 maximum downstream turbine power. Likewise the downstream separation
498 will allow mixing to occur (decreasing I and increasing u at the inflow to
499 downstream devices) but also requires a greater xy area and there will likely
500 be a maximum value of P/A in this trade off.

501 The c_p values recorded are similar to those in Ordonez *et al.* [5] but
502 lower than earlier studies [4] and lower than for other small-scale turbine
503 experiments [35] which both produced maximum values of $c_p > 0.45$. It
504 should be noted that the MRL turbines were designed for high blockage
505 ratio flows and thus the c_p results presented were expected to be lower than
506 these high blockage tests. If a suitable facility could be found, a repeat test in
507 shallow water, where the turbines could be mounted to the floor to avoid the

508 effects of structural struts on the flow would be of benefit. In addition, work
509 by Salter and Taylor [36] and Nishino *et al.* [15] has suggested that very high
510 global blockage ratios would increase c_p and allow the Lanchester-Betz limit
511 to be exceeded. This is due to the tidal channel being more analogous to a
512 duct than unbounded flow for which the limit was defined. This would require
513 larger turbine models or a shallower tank to effectively test this hypothesis,
514 which remains a key query for the industry.

515 An interesting wake effect was the standing wave at the free surface behind
516 the first row of turbines. This effect was also noted in the results of [17], who
517 speculated it could be due to critical bypass flow. However, with the Froude
518 numbers in this experiment of 0.32 being significantly below the critical value
519 of 1, it is far more likely the effect is due to vertical mixing and head loss
520 across as the flow travels through the turbine. This varying flow direction
521 will change the effective angle of attack of the blades in downstream devices.
522 There is likely to be a ‘sweet-spot’ within the wavelength of the standing
523 wave that will improve efficiency of downstream devices. This will be at the
524 angle relative to the blade acting primarily in lift (rather than drag) where
525 the flow direction has the effective angle of attack of highest lift coefficient.
526 It was also noted that the pitch angle had a small negative (downward)
527 velocity throughout the measurements. This may in part be due to a small
528 misalignment of the ADV in this plane.

529 A key source of uncertainty in this experimental work was the asymmetry
530 in the turbines’ in-flow velocity. A previous study validated the symmetry of
531 the flow in facility and the turbines and the frame were mounted symmetrically
532 about the centre line of the tank. Due to this the inflow to individual

533 turbines was only conducted for half the devices, in order to maximise the
534 number of arrays that could be trialled in the available test time. The results
535 presented show that there are changes in flow heading as it passes through
536 the arrays. While Ordonez et al. [5] observed evidence of asymmetry in the
537 wake, no measurements were made of the transverse velocity components.
538 Thus, no direct comparison can be made with the present tests and the
539 measured transverse flow velocities. It is recommended for future tests that
540 inflow measurements are taken for all devices. In addition advancing the
541 models to a higher degree of sophistication including active torque or speed
542 control could reduce variation of results between devices.

543 In addition to this, the wake from the support poles played a large role
544 in the results. Different turbine array layouts were affected differently due
545 to different turbine positions relative to these poles. While this increases the
546 uncertainty of these results, it is also a reminder of the sensitivity of machine
547 performance to likely complex local flows in a real field setting.

548 In summary there are three proposed optimising spacings for this type of
549 turbine in arrays:

- 550 1. A stream-wise spacing to maximise P_{total}/A of the order of multiple
551 rotor diameters;
- 552 2. A lateral spacing to maximise bypass flow, i.e., the array ‘choking’
553 capacity;
- 554 3. A refinement of the stream-wise spacing of the order of a single rotor
555 diameter, to find the optimum performance within the standing wave
556 wavelength.

557 It is still a point of debate as to which metric is best for comparing arrays.

558 Power per xy area is one logical choice but this is only worthwhile if the area
559 constraint is likely to be the dominant parameter in the array design. This
560 subject would require a site optimisation tool and is likely project specific.

561 As this work represents one of the largest array testing projects to use
562 rotating models at this scale, there is significant scope for future work, beyond
563 what has already been discussed in this section. An expansion to compare
564 different turbine types such as conventional three-bladed HATs would inform
565 differences in array spacings for different designs. It is obvious from the
566 results presented that flows through arrays are complex and it is best to
567 measure at as high a spatial resolution as resources allow. Finally, for many
568 tidal sites the direction of flow varies significantly [10]. Hence a test of array
569 sensitivity to off angle flow is a key metric to predict total power over a full
570 tidal cycle.

571 As the first commercial arrays of devices are shortly to become a reality,
572 the increased knowledge of flow interaction and array layout optimisation
573 represent essential knowledge. However, with many key question still asso-
574 ciated with a high level of uncertainty further work is needed to ensure the
575 successful progression of the industry.

576 **6. Conclusions**

577 Comparing the power extracted for each trend in the arrays the following
578 conclusions can be drawn:

- 579 • Power extraction changes the flow through the arrays, with the power
580 per turbine varying by up to 19% in the arrays presented. Thus posi-
581 tioning turbines is important to maximise power output as predicted.

- 582 • Increasing streamwise spacing increases the total power captured in the
583 staggered arrays (base-case and A). This is as predicted by [16] who
584 suggest that downstream rows will be less effected by the performance
585 of upstream rows.
- 586 • Staggering rows generally improves the power per device as predicted
587 by [16].
- 588 • Decreasing lateral spacing can increase or decrease power output as
589 there is an optimal local blockage to maximise power output [15]. The
590 results here show a decrease in power output for both staggered and
591 in-line arrays suggesting that the spacing in the narrow arrangements
592 (arrays B and D) are at a spacing less than this critical array ‘choking’
593 capacity.

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