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Research Article

A Study on Estimating the Next Failure Time of Compressor Equipment in an Offshore Plant

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The offshore plant equipment usually has a long life cycle. During its O&M (Operation and Maintenance) phase, since the accidental occurrence of offshore plant equipment causes catastrophic damage, it is necessary to make more efforts for managing critical offshore equipment. Nowadays, due to the emerging ICTs (Information Communication Technologies), it is possible to send health monitoring information to administrator of an offshore plant, which leads to much concern on CBM (Condition-Based Maintenance). This study introduces three approaches for predicting the next failure time of offshore plant equipment (gas compressor) with case studies, which are based on finite state continuous time Markov model, linear regression method, and their hybrid model.

1. Introduction

In general, maintenance is defined as all technical and managerial actions taken during usage period to maintain or restore the required functionality of an asset or equipment. There have been various classifications of maintenance policies. Simply, maintenance policies can be divided into breakdown maintenance (corrective maintenance), preventive maintenance, and CBM (Condition-Based Maintenance). Unlike breakdown maintenance and preventive maintenance, the CBM focuses on not only fault detection and diagnostics of equipment but also degradation monitoring and failure prediction. Generally, CBM can be treated as a method used to reduce the uncertainty of maintenance activities and is carried out according to the requirements indicated by equipment condition [1]. Thus, the CBM enables us to identify and solve problems in advance before equipment damage occurs. In industry systems, any equipment damage can lead to serious results. Since a critical failure or degradation of the equipment during its operation can seriously damage the belief of customers on the equipment reliability, the maintenance enhancement for preventing this kind of failure or degradation in advance has precedence over any other things in a company. Since oil and gas industries are particularly capital-intensive, careful management on equipment is very important. In this respect, the CBM is a very attractive method for oil and gas industries. CBM is currently being utilized in the petrochemical industry, with condition monitoring of both on and offshore oil and gas wells [2].

Until now it has been difficult to achieve the effectiveness of maintenance operations because there is no information visibility during equipment's usage period. However, recently, with emerging technologies such as various sensors, SCADA (Supervisory Control And Data Acquisition) and PEID (Product Embedded Information Devices) are expected to be rapidly used for gathering and monitoring the status data of equipment during equipment usage period. Advancements in information communication technology have added accelerated growth in the CBM technology area

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by enabling network bandwidth, data collection and retrieval, data analysis, and decision support capabilities for large data sets of time series data [3]. Under the new environment, we can gather the equipment status data related to usage conditions, failure, maintenance or service events, and so on. These data sets enable us to diagnose the degradation status of the equipment in a more exact way. Therefore, using this information gives us new challenging issues for improving the efficiency of equipment maintenance operations. We can diagnose equipment status, predict equipment abnormality, and execute proactive maintenance, that is, doing CBM.

Although there have been some relevant research works so far, the CBM is still challenging area. Current approaches have the limitation in detailed methods or validated predictive models, in particular, in the offshore plant domain. This study deals with the approaches that can predict the next failure time of offshore plant equipment (gas compressor) used in LNG FPSO (Liquefied Natural Gas Floating Production Storage and Offloading vessel).

An LNG FPSO is an offshore plant of delivering liquefied gas from a gas field to customers. Recently, the demand for LNG FPSO is highly increasing and the demand for LNG FPSO projects will grow along with the increased demand for natural gas [4]. The O&M phase of LNG FPSO requires heavy charges and more efforts to optimize the cost and to reduce the risks than the construction phase because of its long life cycle. Nowadays due to the fact that an accident of LNG FPSO in operation causes catastrophic damage, many studies have focused on a maintenance system. In this vein, this study focuses on the prognostic approach for the gas compressor equipment in LNG FPSO, which is one of the main results for the Korean government supported project that is being currently developed since 2013 with the objective of implementing the predictive maintenance system for LNG FPSOs. The objective of this study is to develop the algorithm for estimating the next failure time of a gas compressor based on gathered vibration sensing and failure data. To this end, in this study, finite state Markov model based approach, linear regression model based approach, and their hybrid model have been introduced. To evaluate the proposed approaches, the case study and computational experiments for compressor equipment have been carried out.

The rest of this study is organized as follows. First, relevant previous studies are reviewed and their limitations are discussed in Section 2. Then, Section 3 describes the equipment for CBM focused on this study and two approaches for estimating the next failure time of a compressor, and Section 4 introduces relevant case studies and computational experiments. In Section 5, the hybrid approach combining two approaches is proposed with computational experiment. Finally this study is concluded with further research issues in Section 6.

2. Literature Review

There are several maintenance policies: corrective maintenance, preventive maintenance, opportunistic maintenance, Condition-Based Maintenance, and predictive maintenance.

Corrective maintenance is the unplanned maintenance. However, preventive maintenance (such as constant interval maintenance, age based maintenance, and imperfect maintenance) and predictive maintenance (such as RCM (Reliability Centered Maintenance) and CBM) are the types of planned maintenance. For more details, please refer to Bevilacqua and Braglia [5] or Kothamasu et al. [6]. Among various maintenance policies, this study focuses on the CBM.

The term, CBM, is often used with other terms such as PdM (Predictive Maintenance), PHM (Prognostic and Health Management), and on-condition maintenance which comes from the US Department of Defense and Department of Energy. In this study, we define CBM as a maintenance policy which does maintenance action before equipment failures happen, by assessing equipment condition including operating environments and predicting the risk of equipment failures in a real-time way, based on gathered data. The benefits of a successful CBM strategy are expected to include less regular maintenance, the reduction of unscheduled maintenance, and improved supply chain management [7].

Until so far, there have been several research works about CBM. For example, Lee [8] introduced the fundamental technologies for remote maintenance (called teleservice engineering) and CBM: machine performance assessment and remote diagnosis. Lee [9] introduced a new methodology of CBM, called machinery dynamics and data fusion through remote machinery monitoring. He presented an example of a remote wireless application currently in use for monitoring machinery in industrial plants. Dieulle et al. [10] dealt with the problem related to CBM policy for a single-unit deteriorating system and proposed the approach to determine the optimal inspection schedule and replacement threshold with renewal processes theory. Furthermore, Grall et al. [11] dealt with a condition-based inspection/replacement problem for a stochastically and continuously deteriorating single-unit system. With regenerative and semiregenerative processes theory, they tried to find two maintenance decision variables: preventive replacement threshold and inspection schedule, with the objective of minimizing the long run expected maintenance cost per unit time. In addition, Lin and Tseng [12] combined traditional reliability modelling methods with vibration-based monitoring techniques and artificial neural network technologies in an integrated system to determine the health status of machinery, namely, CMAC-PEM (Cerebellar Model Articulation Controller neural network-based machine Performance Estimation Model). They developed a WPHM (Weibull Proportional Hazards Model) and carried out a bearing deterioration experiment to test both the CMAC-PEM and the WPHM. Moore and Starr [13] have reviewed the methods and functionality of criticality assessments in condition-based monitoring and proposed the CBC (Cost Based Criticality) algorithm to rank all the alarms arising from condition monitoring, which could allow optimized prioritization of maintenance activities. Wu et al. [14] proposed an intelligent decision support system for the optimal CBM policy. They developed a neural network model that uses bearing vibration information to predict the life percentage of a machine and the remaining life of the

machine. They also computed the optimal replacement strategies by proposing a cost matrix method and suggested an optimal replacement time for assisting maintenance decisionmaking. In addition, in their other work [15], they proposed a prognostic method for machine degradation tracking using ARIMA (AutoRegressive Integrated Moving Average) time series model in order to predict the pending failures and the RUL (Remaining Useful Life) of machines. They developed a forecasting strategy and an automatic prediction algorithm of ARIMA models and carried out the analysis on the vibration severity data collected from rotating machines. They compared the performance of the proposed approach with the basic Box-Jenkins ARIMA method. Recently, Hashemian and Bean [16] discussed the limitations of time-based equipment maintenance methods and the advantages of predictive or online maintenance techniques in identifying the onset of equipment failure. Gruber et al. [17] suggested a CBM framework that is based on system simulations and a targeted Bayesian network model. Simulations are used for exploring various CBM policies under different scenarios and the Bayesian network is used for failure prediction based on simulation data. The proposed framework has been applied to a freight rail fleet case. In addition, Lee et al. [18] carried out a comprehensive review of the PHM field. They introduced a systematic PHM design methodology, 5S methodology, for converting data to prognostics information. They also presented a systematic methodology for conducting PHM as applied to machinery maintenance.

In particular, some research works focused on Markovbased model for CBM. For example, Bunks et al. [19] introduced an application of HMMs (Hidden Markov Models) to CBM for the helicopter gearbox case and presented examples of torque level, defect level, and defect-type classification. They concluded that HMMs have a strong potential for constructing practical and robust algorithms for CBM. Furthermore, Ambani et al. [20] developed a continuous time Markov chain degradation model and a cost model to quantify the effects of maintenance on a multiple machine system. To evaluate the effectiveness of the proposed methods, they introduced a case study of an automotive assembly line. In their model, a Markov-based degradation model is used to represent discrete state degradation process, under the assumption that the future degradation state of the machine depends only on the current degradation state and not on the history of the degradation states. Si et al. [21] mentioned that Markovian-based models have been widely applied to RUL estimation and to maintenance decision-making support. They said that the main reason for Markovian-based model is that the plant operation condition can be divided into several meaningful states, such as Good, OK, and Minor defects only, Maintenance required, and Unserviceable, so that the state definition is closer to what is used in industry than other stochastic models and therefore is easy to understand.

On the other hand, some previous works have focused on the maintenance of offshore plant. For example, Wang and Majid [22] carried out a case study of the analysis of reliability and maintenance data on five gas turbines and of the modelling for determining the appropriate preventive maintenance/inspection intervals of offshore oil platform

plant. Arthur and Dunn [23] introduced an application of an optimized CBM approach to large reciprocating compressors on an offshore installation. Caselitz and Giebhardt [24] introduced results of work in the field of condition monitoring and fault prediction in offshore wind energy converters. Their work included not only development hardware and software solution but also prototype tests and integration of fault prediction and maintenance and repair scheduling techniques. Furthermore, Dey et al. [25] developed a risk based maintenance model using a combined multiple-criteria decisionmaking and weight method for offshore oil and gas pipelines. Eleye-Datubo et al. [26] and Eleye-Datubo et al. [27] applied Bayesian network methods for examining the system safety of FPSOs. Migueláñez and Lane [28] presented the recovery system of offshore wind turbines. The recovery system takes a broad view of events and sensor values across the complete turbine system and subsystems. In addition, Hussin et al. [29] dealt with a systematic methodology for analyzing the maintenance data of gas compression train system on an offshore platform to gain insight about the system reliability performance and identify the critical factors influencing the performance. Telford et al. [2] explored the existing literature on the development and applications of CBM in the oil and gas industry. de Andrade Melani et al. [30] proposed a method for risk analysis of LNG carriers operations based on Bayesian network method. Recently, Griffith et al. [7] addressed initial development and integration of SHPM (Structural Health and Prognostics Management) system into the O&M process for offshore wind power plants. They developed a multiscale simulation-based methodology to investigate the sensitivity (or effects) of damage of blades. Cho et al. [31] proposed a linear regression-based approach for estimating the remaining life time of compressor. Cho et al. [32] reviewed previous studies associated with CBM of offshore plants and introduced case studies of prognosis system development predicting performance and failures of offshore plant equipment such as compressor and pump

Although not a few previous research works dealt with various CBM issues, little attention has been paid to the research that deals with offshore plant equipment and has the limitation in estimating the next failure time (remaining life time from the current time) based on gathered sensor data. Estimating the next failure time with sensor data is still the undeveloped area in an offshore plant equipment. To cope with the limitations, this study proposes three approaches to estimate the next failure time of offshore equipment based on finite state continuous time Markov model, linear regression model, and their hybrid model.

3. Target Equipment and Proposed Approaches

LNG FPSOs are used when an oil platform is in a remote or deepwater location where seabed pipelines are not cost effective [33]. Nowadays due to the fact that an accident of LNG FPSO in operation causes catastrophic damage, many studies dealt with the improvement of operating a maintenance

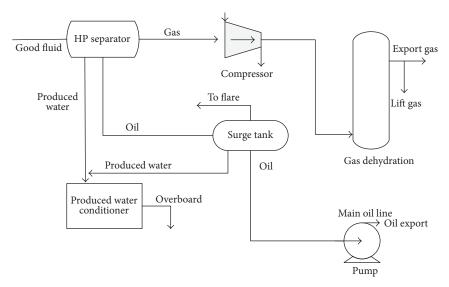


FIGURE 1: Process flow diagram of water, oil, and gas separation in LNG FPSO.

system for LNG FPSO. The LNG FPSO is composed of lots of facilities and equipment. Depending on location, they can be classified into top-side, hull-side, and subsea-side. Among them, this case study focuses on a gas compressor in the topside of LNG FPSO. Figure 1 shows the process flow diagram of water, oil, and gas separation in LNG FPSO. The compressor is used in liquefaction processes of offshore plants. It is important equipment in not only offshore but also onshore plants. It is a mechanical device to increase the pressure of gas and to reduce its volume. It spends most of the energy in offshore plants. Offshore gas compressors are used for various tasks including reservoir management, production enhancement, and the transmission of gas. Any unexpected or prolonged downtime of these units has a large impact on plant availability, as a loss of compression capability drastically affects the oil and gas production of the asset [23]. There are several kinds of compressors. Among them, this study deals with the centrifugal compressor. According to Hussin et al. [29], the number of failures of centrifugal compressor used in an offshore plant is about 26 during six years. For this reason, it preferentially needs to do the development of a prognosis system for the compressor.

According to OREDA [34], frequently observable failure modes of compressor are low output, overheating, spurious stop, external leakage, and so on. The failure mode is usually generated from some failure causes. Compressor failure causes (frequency) are shown as follows:

Rotation/shaft (22%); instrumentation (21%); radial bearing (13%)

Blade/impeller (8%); thrust bearing (6%); compressor seal (6%)

Motor winding (3%); diaphragm (1%); and so forth (20%)

Hence, in this study, we focus on one main cause, rotation/shaft for CBM.

3.1. Vibration Analysis: Compressor. There are some kinds of parameters to monitor the status of the compressor. This study deals with a vibration parameter, because it is widely used in detecting the status of rotating equipment. Generally speaking, vibration is the value with time of the magnitude of a quantity that is descriptive of the motion or position of a mechanical system. Because most normal plant equipment is mechanical, vibration monitoring provides the best tool for routine monitoring and identification of incipient problems [35]. The increasing amplitude of vibration may be an indication of a deteriorating machine condition and the rate of increase is proportional to the degree of damage. Therefore, it is possible to predict the trend of deterioration of a machine by monitoring the amplitudes of its fault related vibration features [36].

Relative shaft vibration and bearing vibration data are usually used to evaluate the status of a compressor of an LNG FPSO. In this study we monitor the status of a gas compressor through relative shaft vibration data. ISO 7919 (international standard for relative shaft vibration of rotating machines) suggests a way for the measurement of a vibration parameter. According to ISO 7919, the max value among two peak-peak values measured by two sensors located in x-axis and y-axis as 90 degree is expressed as $S_{(p-p)max} = [S_{X(p-p)}, S_{Y(p-p)}]_{max}$.

A common practice in industry is to set up various warning levels instead of maintenance stages. The warning levels can be classified as alert, high alert, alarm, serious alarm, and breakdown. General guidelines for setting up warning levels for different types of machines are recommended by various national and international committees [36]. ISO 7919 recommends four vibration limits: limit of start-up performance (A), limit of good vibration performance (B), limit for warning alarm (C), and limit for trip (D). The values of limits are calculated as follows: A/B ($4800/\sqrt{\pi}$), B/C ($9000/\sqrt{\pi}$), and C/D ($13200/\sqrt{\pi}$) where π denotes the RPM (Revolution Per Minute) of a compressor.

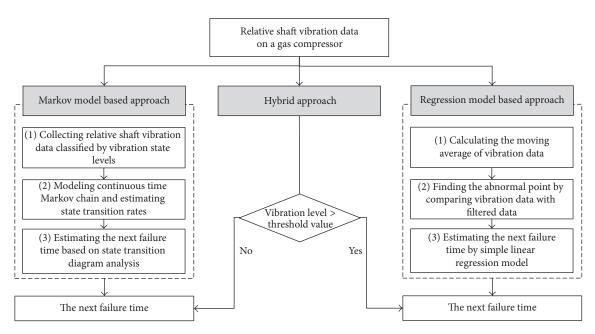


FIGURE 2: Overview of proposed approaches.

In this study, we propose three approaches for estimating the next failure time of a compressor: (1) Markov model based approach, (2) regression model based approach, and (3) hybrid approach combining (1) and (2). In the Markov model based approach, we divide the vibration value into some levels and stochastically predict the next failure time based on the finite state continuous time Markov model theory. In the regression model based approach, when the vibration data evidently differs from the moving average filter and shows increasing trends, we predict the next failure time using simple linear regression model. The hybrid approach takes the advantage of two approaches. Depending on the vibration level, one of two approaches is applied to estimate the next failure time. Figure 2 depicts the detailed process flow of these approaches.

3.2. Markov Model Based Approach. In order to apply the CBM policy of an LNG FPSO compressor, among various prognostics methods such as wavelets, artificial neural networks, and Bayesian network, in this study, we use the finite state continuous time Markov model since it has the benefit in the amount of data required for analysis, compared to other methods. Furthermore, it allows an exact computation of system reliability.

In this study, it is assumed that LNG FPSO operation system records $S_{(p-p)max}$ and its timestamp data whenever the level of relative shaft vibration exceeds the predefined limit (e.g., warning level and trip level). Furthermore, the amplitude of vibration signal of the compressor will remain the limit of tolerance range unless it has abnormal symptoms for faults or failures. In addition, we assume that the vibration state evolves continuously over time and state transition rate does not depend on time based on the interview result with compressor engineers.

To maintain brevity and consistency, this study defines the following notation.

s: index of compressor state

 $\tau_{s,s'}$: the number of changes from s to s'

 $T_{s,s'}$: transition rate from s to s'

 $t_{s,s'}^n$: the time interval of the nth status transition from status s to status s'

 $L_{s,s'}$: the expected time to reach the state s' from the state s

 $E_{s,s'}$: the expected time from status s to the very next state s'

 Pr_s : the probability to transit from state s to state s + 1

 P_s^j : the probability of the *j*th return to the state *s* after going through state s-1

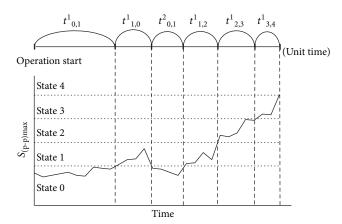
 π : the RPM (Revolution Per Minute) of a compressor

 λ : the deteriorating transition rate

 μ : the recovery transition rate

The detailed procedures of the proposed approach are as follows.

Step 1 (collecting relative shaft vibration data classified by vibration state levels). We assumed that LNG FPSO operation system records $S_{(p-p)\max}$ and its timestamp data whenever the level of relative shaft vibration exceeds the predefined limit. Regarding the predefined limit, based on ISO 7919, we set the vibration levels into good vibration level and alarm level. And we divide the alarm level into low, middle, and high levels, in detail, because it is more important to predict the next failure time in an alarm level than in a good vibration



--- Event occurrence

FIGURE 3: Example of vibration state transition plot in Markov model based approach.

performance level. Thus, whenever the vibration data exceeds over the predefined levels, the state values (denoted as *s*) for representing each level and its timestamp data are recorded. The value of state *s* has the following meaning:

s = 0: a good vibration level under $9000/\sqrt{\pi}$

s = 1: a low alarm level under $10400/\sqrt{\pi}$

s = 2: a middle alarm level under 11800/ $\sqrt{\pi}$

s = 3: a high alarm level under $13200/\sqrt{\pi}$

s = 4: a trip level above $13200/\sqrt{\pi}$

Figure 3 depicts an example of transition plot representing the state values of vibration data changed over time.

Step 2 (estimating status transition rates). From the interview with compressor maintenance engineers, we identified that the failures of the compressor shaft are not much related to the deterioration process and occur randomly. Hence, in this study, we assume that the time when status transition occurs follows the homogeneous Poisson process having the state transition rate as follows:

$$T_{s,s'} = \frac{\tau_{s,s'}}{\sum_{k=1}^{\tau_{s,s'}} t_{s,s'}^k}.$$
 (1)

Then, since the state transition rate is constant and does not depend on the time, we assume that it has the Markovian property.

Step 3 (estimating the next failure time). Then, $L_{0,s}$ could be estimated by the following formula:

$$L_{0,s} = \sum_{j=0}^{\infty} \left[P_{s-1}^{j} \cdot Pr_{s-1} \cdot \left\{ j \cdot \left(L_{s-2,s-1} + E_{s-1,s-2} \right) + E_{s-1,s} \right\} \right]$$
(2)
+ $L_{0,s-1}$,

where $L_{0,0} = 0$ and $L_{0,1} = 1/\lambda_0$, s' = 2, 3, 4.

Let j be the number of first passage from s'-1 to s'-2. The first term of formula (2) indicates the expected time until arriving at state s after the jth return to state s-1, calculated by multiplying the duration time; that is, $(j \cdot (L_{s-2,s-1} + E_{s-1,s-2}) + E_{s-1,s})$ and its probability $(P_{s-1}^j \cdot Pr_{s-1})$. The second term (i.e., $L_{0,s-1})$ of formula (2) denotes the expected time from state 0 to state s-1.

Let $\lambda_s = T_{s,s+1}$, $0 \le \lambda_s \le 1$, for s = 0, 1, 2, 3, which indicates the deteriorating transition rate, and $\mu_s = T_{s+1,s}$, $0 \le \mu_s \le 1$, for s = 0, 1, 2, 3, which indicates the recovery transition rate, respectively. Then, according to Anderson [37], $E_{s-1,s} = 1/\lambda_{s-1}$, $E_{s,s-1} = 1/\mu_s$, $P_s^j = (\mu_s/(\lambda_s + \mu_s))^j$, and $Pr_{s-1} = (\lambda_s/(\lambda_s + \mu_s))$ in the continuous time Markov model. As a result, formula (2) could be unfolded as follows:

$$\begin{split} L_{0,s} &= \sum_{j=0}^{\infty} \left[P_{s-1}^{j} \cdot Pr_{s-1} \right. \\ &\cdot \left\{ j \cdot \left(L_{s-2,s-1} + E_{s-1,s-2} \right) + E_{s-1,s} \right\} \right] + L_{0,s-1} \\ &= \sum_{j=0}^{\infty} \left[\left(\frac{\mu_{s-1}}{\lambda_{s-1} + \mu_{s-1}} \right)^{j} \cdot \left(\frac{\lambda_{s-1}}{\lambda_{s-1} + \mu_{s-1}} \right) \\ &\cdot \left\{ j \cdot \left(L_{s-2,s-1} + \frac{1}{\mu_{s-1}} \right) + \frac{1}{\lambda_{s-1}} \right\} \right] + L_{0,s-1} \\ &= \sum_{j=0}^{\infty} \left[\left(\frac{\mu_{s-1}}{\lambda_{s-1} + \mu_{s-1}} \right)^{j} \cdot \left(\frac{\lambda_{s-1}}{\lambda_{s-1} + \mu_{s-1}} \right) \\ &\cdot \left\{ j \cdot \left(L_{s-2,s-1} + \frac{1}{\mu_{s-1}} \right) \right\} \right] + \sum_{j=0}^{\infty} \left[\left(\frac{\mu_{s-1}}{\lambda_{s-1} + \mu_{s-1}} \right)^{j} \\ &\cdot \left(\frac{\lambda_{s-1}}{\lambda_{s-1} + \mu_{s-1}} \right) \cdot \frac{1}{\lambda_{s-1}} \right] + L_{0,s-1} \\ &= \sum_{j=0}^{\infty} \left[\left(\frac{\mu_{s-1}}{\lambda_{s-1} + \mu_{s-1}} \right)^{j} \cdot \left(\frac{\lambda_{s-1}}{\lambda_{s-1} + \mu_{s-1}} \right) \\ &\cdot \left\{ \left(j + 1 \right) \cdot \left(L_{s-2,s-1} + \frac{1}{\mu_{s-1}} \right) \right\} \right] \\ &- \sum_{j=0}^{\infty} \left[\left(\frac{\mu_{s-1}}{\lambda_{s-1} + \mu_{s-1}} \right)^{j} \cdot \left(\frac{\lambda_{s-1}}{\lambda_{s-1} + \mu_{s-1}} \right) \\ &\cdot \left(\frac{\lambda_{s-1}}{\lambda_{s-1} + \mu_{s-1}} \right) \cdot \frac{1}{\lambda_{s-1}} \right] + L_{0,s-1} = \left(\frac{\lambda_{s-1}}{\lambda_{s-1} + \mu_{s-1}} \right) \\ &\cdot \left(L_{s-2,s-1} + \frac{1}{\mu_{s-1}} \right) \cdot \sum_{j=0}^{\infty} \left[\left(\frac{\mu_{s-1}}{\lambda_{s-1} + \mu_{s-1}} \right)^{j} \cdot \left(j + 1 \right) \right] \\ &- \left(L_{s-2,s-1} + \frac{1}{\mu_{s-1}} \right) \cdot \sum_{j=0}^{\infty} \left[\left(\frac{\mu_{s-1}}{\lambda_{s-1} + \mu_{s-1}} \right)^{j} \cdot \left(j + 1 \right) \right] \\ &- \left(L_{s-2,s-1} + \frac{1}{\mu_{s-1}} \right) \cdot \sum_{j=0}^{\infty} \left[\left(\frac{\mu_{s-1}}{\lambda_{s-1} + \mu_{s-1}} \right)^{j} \cdot \left(j + 1 \right) \right] \\ &- \left(L_{s-2,s-1} + \frac{1}{\mu_{s-1}} \right) + \frac{1}{\lambda_{s-1}} + L_{0,s-1}. \end{split}$$

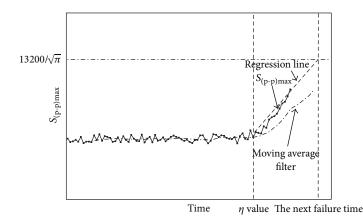


FIGURE 4: Example of the trend plot of vibration data in regression model based approach.

Since

$$\sum_{j=0}^{\infty} \left[K^{j} \cdot (j+1) \right] = \frac{d}{dK} \sum_{j=0}^{\infty} K^{j+1} = \frac{d}{dK} \left(\frac{K}{1-K} \right)$$

$$= \frac{1}{(1-K)^{2}},$$
(4)

where $K = \mu_{s-1}/(\lambda_{s-1} + \mu_{s-1})$, the above equation could be expressed as follows.

$$= \left(\frac{\lambda_{s-1}}{\lambda_{s-1} + \mu_{s-1}}\right) \cdot \left(L_{s-2,s-1} + \frac{1}{\mu_{s-1}}\right)$$

$$\cdot \frac{1}{\left[1 - (\mu_{s-1}/(\lambda_{s-1} + \mu_{s-1}))\right]^{2}} - \left(L_{s-2,s-1} + \frac{1}{\mu_{s-1}}\right)$$

$$+ \frac{1}{\lambda_{s-1}} + L_{0,s-1}$$

$$= \frac{(\lambda_{s-1}/(\lambda_{s-1} + \mu_{s-1})) - \left[1 - \mu_{s-1}/(\lambda_{s-1} + \mu_{s-1})\right]^{2}}{\left[1 - \mu_{s-1}/(\lambda_{s-1} + \mu_{s-1})\right]^{2}}$$

$$\cdot \left(L_{s-2,s-1} + \frac{1}{\mu_{s-1}}\right) + \frac{1}{\lambda_{s-1}} + L_{0,s-1}$$

$$= \frac{(\lambda_{s-1} \cdot \mu_{s-1})/(\lambda_{s-1} + \mu_{s-1})^{2}}{\left[1 - \mu_{s-1}/(\lambda_{s-1} + \mu_{s-1})\right]^{2}} \cdot \left(L_{s-2,s-1} + \frac{1}{\mu_{s-1}}\right)$$

$$+ \frac{1}{\lambda_{s-1}} + L_{0,s-1}$$

$$= \frac{\mu_{s-1}}{\lambda_{s-1}} \cdot \left(L_{s-2,s-1} + \frac{1}{\lambda_{s-1}}\right) + \frac{1}{\lambda_{s-1}} + L_{0,s-1}.$$

Along formula (5), it is possible to estimate $L_{0.4}$ as follows:

$$L_{0,4} = \left[\frac{\mu_3}{\lambda_2} \cdot \left\{ L_{2,3} + \frac{1}{\mu_2} \right\} \right] + \frac{1}{\lambda_2} + L_{0,3}. \tag{6}$$

Also, we can know that the expected time to reach state s from state s-1 is simply calculated as follows.

$$L_{s-1,s} = L_{0,s} - L_{0,s-1}. (7)$$

Along formulae (5) and (7), $L_{0,3}$ and $L_{2,3}$ in formula (6) could be estimated as follows:

$$L_{0,3} = \left[\frac{\mu_2}{\lambda_2} \cdot \left\{ L_{1,2} + \frac{1}{\mu_2} \right\} \right] + \frac{1}{\lambda_2} + L_{0,2}, \tag{8}$$

$$L_{2,3} = L_{0,3} - L_{0,2}. (9)$$

 $L_{0,s}$ and $L_{s-1,s}$ in a series of calculations like $L_{0,2}$ and $L_{1,2}$ in formula (8) could be estimated using formulae (5) and (7) in the same way.

If the current state is s, then $L_{s,4}$ could be estimated as follows:

$$L_{s,4} = L_{0,4} - L_{0,s}. (10)$$

Then, finally the next failure time could be estimated as follows:

The next failure time = the current time +
$$L_{s,4}$$
. (11)

3.3. Regression Model Based Approach. In addition to the Markov model based approach, in this study, the regression model based approach for estimating the next failure time of the compressor is proposed. The regression model based approach is based on the $S_{(p-p)max}$ trend plot (please refer to Figure 4). The trend plot is a method to record the variation of the magnitude that is descriptive of motion or positions of the equipment with time. The trend plot helps engineers figure out the status of the equipment at a glance. After the raw data of vibration is recorded in the trend plot, the regression model based approach analyzes $S_{(p-p)max}$ trend plot with moving average filter and abnormal indicator variable (denoted as v_i). According to ISO 7919, we let the limit for trip be a critical limit for the failure of a compressor. Then, it calculates the point where the simple linear regression line intersects with the limit for trip and considers it as the next failure time.

To maintain brevity and consistency, this study defines the following notation.

k: the number of y_i 's *m*: index for calculating η

n: the number of entities used in the moving average filter

r: the number of comparisons with y_i and \overline{y}_i

 v_i : the 0-1 binary variable that indicates whether $\overline{y}_i < y_i$ or not (i.e., whether abnormal situation occurs or not)

 x_i : the time when the *i*th $S_{(p-p)max}$ value (relative shaft vibration data) is recorded

 y_i : the *i*th $S_{(p-p)\max}$ value

 \overline{y}_i : moving average filter of y_i

 η : the time point just before the vibration value is continuously over the moving average filter value with r times

 π : the RPM (Revolution Per Minute) of a gas compressor

The detailed procedure is as follows.

Step 1 (calculating the moving average of vibration data). In this study, in order to predict the next failure time, it is necessary to carefully monitor the trend of vibration data over time points. Since the vibration value itself has the limitation in giving the trend of time series values, the moving average filter is applied to catch the trend of vibration values. If there are y_1, y_2, \ldots, y_k , a moving average filter \overline{y}_i for $S_{(p-p)\max}$ is calculated as follows:

$$\overline{y}_i = \frac{y_{i-n+1} + y_{i-n+2} + \dots + y_i}{n}, \quad n \le i \le k.$$
 (12)

Step 2 (finding the abnormal time point by comparing vibration data with filtered data). If there is the time point showing abnormal situations continuously compared to previous data, we assume that the failure propagation evolves in a fast way after that point. In this study, to find the abnormal time point, η is calculated by formula (13). Here, η indicates the time point just before showing the abnormal performance (i.e., $\overline{y}_i < y_i$) continuously in r times.

$$\eta = \left\{ \arg\min_{m} \left(\prod_{i=m}^{m+r-1} v_i \right) \right\}, \quad n \le m \le k-r+1, \quad (13)$$

where

$$v_i = \begin{cases} 1 & \text{(if } \overline{y}_i < y_i) \\ 0 & \text{(otherwise)} \end{cases} \quad \text{for } n \le i \le k. \tag{14}$$

Step 3 (estimating the next failure time by a linear regression model). The next failure time could be estimated based on the values of $x_{\eta+1}, x_{\eta+2}, \ldots, x_k$ by a linear regression model. If the regression equation is expressed as $y_i = \widehat{\beta}_0 + \widehat{\beta}_1 \cdot x_i$ where $\widehat{\beta}_0 = \overline{y} - \widehat{\beta}_1 \cdot \overline{x}$ and $\widehat{\beta}_1 = \sum_{i=\eta+1}^k (x_i - \overline{x})(y_i - \overline{y}) / \sum_{i=\eta+1}^k (x_i - \overline{x})^2$, then the next failure time could be estimated by formula (15):

The next failure time =
$$\frac{1}{\hat{\beta}_1} \cdot \left(\frac{13200}{\sqrt{\pi}} - \hat{\beta}_0 \right)$$
. (15)

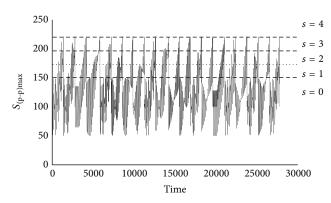


FIGURE 5: Example of vibration data.

4. Case Study

For a case study, we used the data on shaft vibration of a gas compressor generated based on the sample vibration data obtained from a compressor manufacturing company in South Korea. The generated data for about three years includes 20 failures and about 564 state transitions, where RPM is 3600 (i.e., $\pi = 3600$). The vibration data used in this study is depicted by Figure 5.

4.1. Case Study of Markov Model Based Approach. From the example data of the case study, we obtain the number of changes from s to s', that is, $\tau_{s,s'}$ as follows: $\tau_{0,1}=135$, $\tau_{1,0}=115$, $\tau_{1,2}=108$, $\tau_{2,1}=88$, $\tau_{2,3}=59$, $\tau_{3,2}=39$, and $\tau_{3,4}=20$. Here, we describe how to calculate the transition rate, λ_3 , that is, $T_{3,4}$. Other calculations for $T_{s,s'}$ are omitted for the convenience.

 λ_3 could be estimated as follows:

$$\lambda_3 = \frac{\tau_{3,4}}{\sum_{k=1}^{\tau_{3,4}} t_{3,4}^k} = \frac{20}{15 + 14 + 15 + \dots + 13} = 0.0738. \quad (16)$$

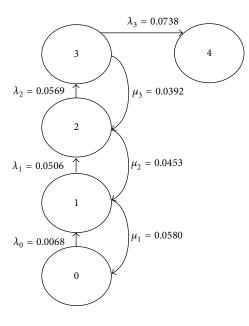
We could obtain the transition rates in the same way as follows: $\lambda_0 = 0.0068$, $\lambda_1 = 0.0506$, $\lambda_2 = 0.0569$, $\mu_1 = 0.0580$, $\mu_2 = 0.0453$, and $\mu_3 = 0.0392$. The state transition diagram with these transition rates is depicted in Figure 6.

Where the current state is the state 0, the expected time to reach the failure from the state 0 ($L_{0,4}$) could be estimated along formula (6) as follows:

$$L_{0,4} = \left[\frac{0.0392}{0.0738} \cdot \left\{ L_{2,3} + \frac{1}{0.0392} \right\} \right] + \frac{1}{0.0738} + L_{0,3}. \quad (17)$$

In formula (17), $L_{2,3}$ and $L_{0,3}$ could be estimated in a series of calculations as follows:

$$\begin{split} L_{2,3} &= L_{0,3} - L_{0,2}, \\ L_{0,3} &= \left[\frac{0.0453}{0.0569} \cdot \left\{ L_{1,2} + \frac{1}{0.0453} \right\} \right] + \frac{1}{0.0569} + L_{0,2}, \end{split}$$



State

- (0) Good vibration performance
- (1) A warning alarm (low)
- (2) A warning alarm (middle)
- (3) A warning alarm (high)
- (4) A trip

FIGURE 6: Example of state transition diagram.

$$\begin{split} L_{0,2} &= \left[\frac{0.0580}{0.0506} \cdot \left\{L_{0,1} + \frac{1}{0.0580}\right\}\right] + \frac{1}{0.0506} + L_{0,1},\\ L_{0,1} &= \frac{1}{0.0068} = 147.9778. \end{split} \tag{18}$$

Table 1 shows a result of a series of the above formulae. The time point when we estimated the next failure time is 27628 (hours). Since $S_{(p-p)\max}$ value is 197.8346 at that time, the vibration state is the state level 3. Thus, the next failure time is as follows: the current time point + $L_{3,4}=27628+134.17=27762.17$ (hours). The real failure time is known as 27776. The residual life time at 27628 time point is only about 134 (hours). Hence, we could see that the estimated next failure time by Markov model based approach is close to the real one.

4.2. Computational Experiments of Markov Model Based Approach. In order to evaluate the performance of the Markov model based approach, we have carried out computational experiments based on the vibration history data (refer to Figure 5). To quantify the degree of the performance of the proposed approach, this study uses the MAPE (Modified Absolute Percentage Error [38]) measure represented as follows:

MAPE =
$$\left| \frac{R - R^*}{(R + R^*)/2} \right| \cdot 100,$$
 (19)

Table 1: $L_{s,s'}$ calculations.

n	$t_{3,4}^n$
$L_{0,4}$	693.3072
$L_{0,3}$	559.1403
$L_{0,2}$	357.3453
$L_{0,1}$	147.9778
$L_{3,4}$	134.1669
$L_{2,3}$	201.7950
$L_{1,2}$	209.3675

TABLE 2: Test results of Markov model based approach.

$\begin{array}{cc} \text{Failure event} & & \text{Average} \\ & & \text{MAPE}^* \end{array}$		# obs. less than 30% [†]	♯ obs. less than 50%♣	
1	0.308	6	8	
2	0.310	5	8	
3	0.498	2	6	
4	0.407	5	6	
5	0.653	3	4	
6	0.422	5	7	
7	0.377	3	5	
8	0.667	5	5	
9	0.438	6	8	
10	0.644	2	5	
11	0.485	4	5	
12	0.387	5	5	
13	0.258	6	10	
Average	0.450	4.380	6.310	

^{*}The average on 10 MAPEs for each failure event.

where R^* indicates residual time to the next failure and R is the estimated residual time to the next failure.

With MAPE, we could avoid the problem of large errors when the residual life time to the next failure is close to zero. To calculate the value of MAPE, in the computational experiment, we use the data after 10000 hours, because Markov model based approach needs enough history data. And then, we randomly chose ten time points for each failure time and estimated the failure time at those times. Table 2 shows that the average MAPE of Markov model based approach is 45.03%. Observation results for the number of solutions less than 30% or 50% showed us that a few solutions have high MAPE values so that they seemed to affect the overall performance of the approach. Furthermore, we could find that Markov model based approach has more large MAPEs as the time point becomes close to the real failure time.

4.3. Case Study of Regression Model Based Approach. The case study of regression model based approach has been carried

[†]Number of observations such that MAPE < 30% among 10 test examples.

^{*}Number of observations such that MAPE < 50% among 10 test examples.

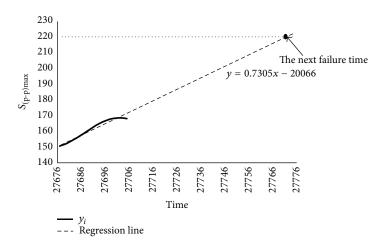


FIGURE 7: Example of regression model based approach.

out based on 100 time point data sets as shown in Table 3. Note that y_i is the maximum $S_{(p-p)\max}$ value at time i whose unit time is an hour. We set parameters as follows: n=10, r=5, and k=27705. Table 3 shows moving average filter \overline{y}_i , indicator variable v_i , and $\prod_{i=m}^{m+r-1} v_i$ for the time point x_i . Along formula (13), η is 27676 in this case study.

Figure 7 depicts how to estimate the next failure time. The linear regression model in this case study is calculated as y = 0.7305x - 20066 based on the values where (x_{27677}, y_{27677}) , $(x_{27678}, y_{27678}), \dots, (x_{27705}, y_{27705})$. Along formula (15), the next failure time could be estimated as follows:

The next failure time =
$$\frac{1}{0.7305} \left(\frac{13200}{\sqrt{3600}} + 20066 \right)$$
 (20)
 $= 27756.33 \text{ (hours)}.$

The real failure time is 27776. Since the time point when we estimated the next failure time is 27705, the remaining life time is 51.33 (hours). Hence, we could also see that the estimated next failure time by regression model based approach is close to the real one.

4.4. Computational Experiments of Regression Model Based Approach. In order to evaluate the performance of regression model based approach, we have also carried out computational experiments. Table 4 shows the values of MAPEs when comparing the residual life time estimated from proposed regression model based approach to the real residual life time. According to this table, we could find the trend that regression model based approach has good performance as the residual life time becomes close to zero, compared to Markov model based approach. However, we could also see that the regression model based approach has the limitation in estimating the next failure time when the residual life time is relatively long since the linear regression model has the tendency in being highly affected by abnormal data.

5. Hybrid Approach

Throughout case study and computational experiments for two approaches, we could find that Markov model based approach has more better performance than regression model based approach when the remaining residual life time is relatively long, while regression model based approach has more better performance in the opposite case. The Markov model based approach could have stable performance compared to the regression model based approach although it seems to give not good performance as the residual life time becomes closed to the failure time. On the contrary, the linear regression method is straightforward and practical, and it gives us the relative good performance when the residual life time is closed to the failure time compared to the Markov model based approach. However, it is so sensitive on some abnormal data that it often overestimates or underestimates them. For this reason, in this study we suggest the hybrid approach that applies both Markov model based approach and regression model based approach into estimating the next failure time. In the hybrid approach, until a certain vibration level, Markov model based approach is applied, and then, after over the certain vibration level, the regression model based approach is used to estimate the next failure time. The below formula shows us how to apply both approaches depending on the level of vibration, in the hybrid approach.

The next failure time =
$$\begin{cases} EFT_1, & (y_i < \omega) \\ EFT_2, & Otherwise, \end{cases}$$
 (21)

where EFT₁ and EFT₂ represent the estimated next failure time by the Markov model based approach and that by the regression model based approach, respectively, and ω indicates the vibration index.

In order to minimize the MAPE, it is necessary to find the best ω . To this end, we have carried out the experiments with increasing ω from $3000/\sqrt{\text{RPM}}$ to $12600/\sqrt{\text{RPM}}$ by

Table 3: Data and variables used in the regression model based approach.

Table 3: Continued.

approach.					x_i	y_i	\overline{y}_i	ν_i	$\prod_{i=m}^{m+r-1} v_i$
x_i	y_i	$\overline{\mathcal{Y}}_i$	ν_i	$\prod_{i=m}^{m+r-1} v_i$	27656	169.394	178.775	0	0
27606	136.703				27657	167.483	176.727	0	0
27607	139.811				27658	165.574	174.680	0	0
27608	143.092				27659	163.767	172.647	0	0
27609	146.458				27660	162.018	170.651	0	0
27610	149.736				27661	160.327	168.693	0	0
27611	153.115				27662	158.764	166.788	0	0
27612	156.455				27663	157.409	164.966	0	0
27613	159.753				27664	156.072	163.223	0	0
27614	163.064				27665	154.883	161.569	0	0
27615	166.381	151.457	1	1	27666	153.795	160.009	0	0
27616	169.561	154.742	1	1	27667	152.810	158.542	0	0
27617	172.655	158.027	1	1	27668	151.905	157.175	0	0
27618	175.649	161.283	1	1	27669	151.295	155.928	0	0
27619	178.551	164.492	1	1	27670	150.824	154.808	0	0
27620	181.348	167.653	1	1				0	0
27621	183.963	170.738	1	1	27671	150.363	153.812		
27622	186.444	173.737	1	1	27672	150.066	152.942	0	0
27623	188.752	176.637	1	1	27673	149.992	152.200	0	0
27624	190.983	179.429	1	1	27674	149.918	151.585	0	0
27625	193.030	182.094	1	1	27675	150.113	151.108	0	0
27626	194.859	184.623	1	1	27676	150.297	150.758	0	0
27627	196.451	187.003	1	1	27677	150.689	150.546	1	1
27628	197.835	189.222	1	1	27678	151.243	150.480	1	1
27629	198.999	191.266	1	1	27679	151.726	150.523	1	1
27630	199.979	193.129	1	1	27680	152.399	150.680	1	1
27631	200.711	194.804	1	1	27681	153.234	150.968	1	1
27632	201.280	196.288	1	1	27682	154.061	151.367	1	1
27633	201.615	197.574	1	1	27683	154.907	151.859	1	1
27634	201.790	198.655	1	0	27684	155.768	152.444	1	1
27635	201.793	199.531	1	0	27685	156.715	153.104	1	1
27636	201.542	200.199	1	0	27686	157.694	153.843	1	1
27637	201.101	200.664	1	0	27687	158.639	154.638	1	1
27638	200.427	200.924	0	0	27688	159.699	155.484	1	1
27639	199.639	200.924	0		27689	160.672	156.379	1	1
27640	199.696	200.859	0	0	27690	161.683	157.307	1	1
	198.696	200.553		0	27691	162.687	158.252	1	1
27641	197.032		0		27692	163.702	159.217	1	1
27642		200.063	0	0	27693	164.565	160.182	1	1
27643	194.990	199.400	0	0	27694	165.367	161.142	1	1
27644	193.359	198.557	0	0	27695	166.084	162.079	1	1
27645	191.658	197.544	0	0	27696	166.677	162.977	1	1
27646	189.901	196.380	0	0					
27647	187.956	195.065	0	0	27697	167.268	163.840	1	1
27648	186.044	193.627	0	0	27698	167.799	164.650	1	1
27649	184.097	192.073	0	0	27699	168.114	165.395	1	1
27650	181.980	190.401	0	0	27700	168.363	166.063	1	1
27651	179.913	188.627	0	0	27701	168.493	166.643	1	1
27652	177.806	186.770	0	0	27702	168.592	167.132	1	
27653	175.629	184.834	0	0	27703	168.541	167.530	1	
27654	173.501	182.849	0	0	27704	168.372	167.830	1	
27655	171.427	180.825	0	0	27705	168.135	168.035	1	

RLT*		10	1	15	2	20	2	25
No. [†]	Est.‡	MAPE	Est.	MAPE	Est.	MAPE	Est.	MAPE
1	9.323	7.01%	18.692	21.91%	29.264	37.61%	45.582	58.32%
2	9.836	1.65%	21.562	35.89%	36.439	58.25%	87.836	111.38%
3	11.632	15.09%	24.609	48.52%	38.227	62.61%	58.742	80.59%
4	6.703	39.47%	13.807	8.28%	22.950	13.74%	37.241	39.33%
5	4.897	68.52%	10.003	39.97%	17.013	16.14%	26.157	4.52%
6	10.510	4.97%	21.993	37.81%	35.793	56.61%	59.961	82.30%
7	9.318	7.06%	18.093	18.69%	27.484	31.52%	40.181	46.58%
8	6.443	43.26%	14.471	3.59%	26.705	28.71%	181.786	151.64%
9	10.946	9.03%	22.764	41.12%	36.031	57.22%	56.923	77.93%
10	9.669	3.36%	18.323	19.94%	27.370	31.12%	39.112	44.02%
11	9.845	1.57%	20.124	29.17%	31.880	45.80%	50.038	66.73%
12	6.977	35.62%	14.004	6.87%	22.399	11.32%	33.762	29.82%
13	7.267	31.65%	14.912	0.59%	23.800	17.35%	36.211	36.63%
Average	20.	.64%	24.	03%	36.	00%	63.	83%
Total average	36.12%							

TABLE 4: Computational experiment results of regression model based approach.

TABLE 5: Test results depending on ω .

ω	Average MAPE
3000/√RPM	48.64%
$3600/\sqrt{RPM}$	47.88%
$4200/\sqrt{RPM}$	47.04%
$4800/\sqrt{RPM}$	45.85%
$5400/\sqrt{RPM}$	43.93%
$6000/\sqrt{RPM}$	42.13%
$6600/\sqrt{RPM}$	40.05%
$7200/\sqrt{\text{RPM}}$	38.02%
$7800/\sqrt{\text{RPM}}$	37.25%
$8400/\sqrt{RPM}$	37.09%
9000/ \sqrt{RPM}	37.39%
9600/ \sqrt{RPM}	37.32%
$10200/\sqrt{\text{RPM}}$	37.85%
$10800/\sqrt{RPM}$	37.98%
$11400/\sqrt{\text{RPM}}$	38.27%
$12000/\sqrt{\text{RPM}}$	38.48%
$12600/\sqrt{\text{RPM}}$	38.69%

 $600/\sqrt{\text{RPM}}$. Table 5 shows the result of experiments depending on the changes of ω . It shows us that $\omega = 8400/\sqrt{\text{RPM}}$ has the best performance.

6. Conclusion

This study has dealt with the approaches for estimating the next failure time of offshore plant equipment, gas compressor. How to estimate the next failure time based on vibration data

has been proposed by three approaches: regression model based approach, Markov model based approach, and their hybrid approach. In the Markov model based approach, based on the gathered vibration state-timestamp data, the next failure time of a gas compressor has been estimated by the finite state continuous time Markov model theory. In the regression model based approach, the linear regression model using moving average filter has been proposed. The hybrid approach takes the advantage of two approaches. Depending on the vibration level, one of two approach is applied to estimate the next failure time.

To show the usefulness of the proposed approaches, case examples and computational experiments based on shaft vibration sensor data were introduced in a case study. Although the proposed approaches have some limitations in fully evaluating the usability in the real field due to the limited examples used in the case study, we believe that it will provide offshore operation companies with a reference for doing the improved maintenance planning and decreasing equipment downtime due to unexpected failures.

We can think of several future research works. First, after suitably tuning the parameters (e.g., n, η , and regression parameters) used in our approaches based on lots of field data, our approaches could be applied to not only gas compressor but also other pieces of equipment, the failure of which could be being monitored by vibration signals. Second, the linear regression model could be improved to the more elaborate regression model by considering the trade-offs between generalizability and goodness of fit. In general case, since exponential model seems to be more suitable for the degradation model of mechanical system, it is valuable to consider more complex regression model rather than the first-order linear regression model. Third, it is possible to

^{*}Residual life time (unit: hour).

[†]Failure event number.

[‡]Residual life time estimated by regression model based approach (unit:hour).

specify the approaches in detail considering the segmentation of ocean environment and equipment operating mission profile. Fourth, in this study, based on engineers interview, we assumed that the state transition in the Markov model is time independent. However, as the future research issue, it is meaningful to consider the time dependency of deterioration process of compressor. In addition, we did not consider real-time interactive update of the parameters used in the Markov model. However, since the degradation process is changing over time due to various uncertainties, it is also valuable to consider the real-time update by using the methods such as MCMC (Monte Carlo Markov Chain) and classic control models. Finally, other probabilistic methods such as Bayesian network and artificial neural network could be applied to develop a more elaborated prognostic algorithm.

Competing Interests

The authors declare that there is no conflict of interests regarding the publication of this paper.

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