

# Root Cause Analysis of a vibration problem in a propylene turbo compressor

Pieter van Beek, Jan Smeulers



# Problem description

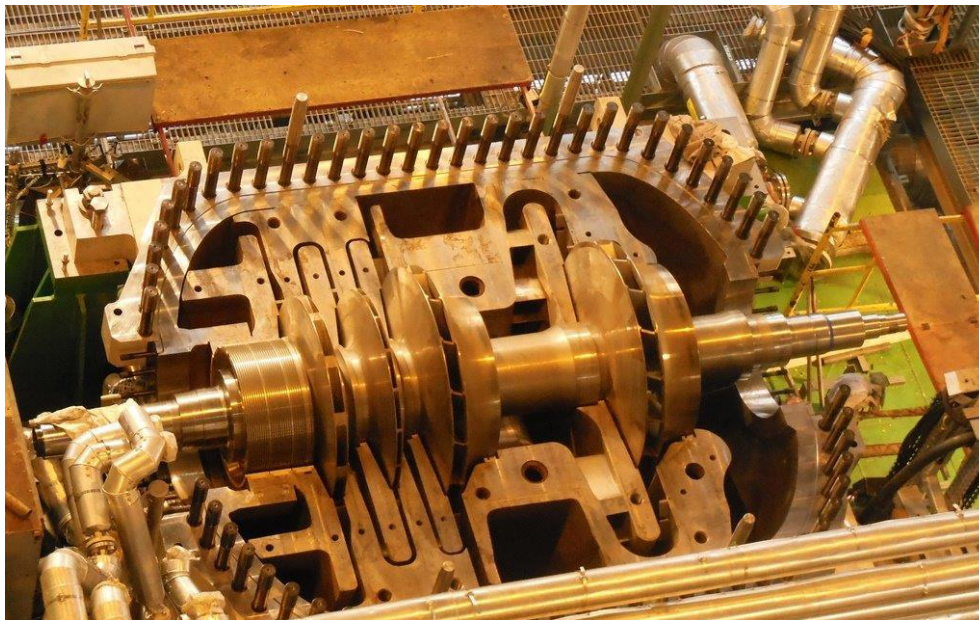
- A newly installed turbo compressor system for propylene showed vibrations in the piping system and rotor.
- After that supporting layout was significantly improved measurements showed that vibrations were within the allowable range.
- Still the rotor vibrations were not acceptable.
- A root cause analysis was carried that showed two likely causes.

# The installation

- Large diameter piping (60 inch suction).
- Reducer to 48 inch just upstream of the inlet of the compressor.
- Large flows ~ 1500 tonnes/h (25 m/s).
- Heavy gas ~ 44 kg/kmol @ 6.6 barg.
- 2 phase flow after 2<sup>nd</sup> stage condenser: liquid separation via large K.O. drum at the suction side (15 m height).

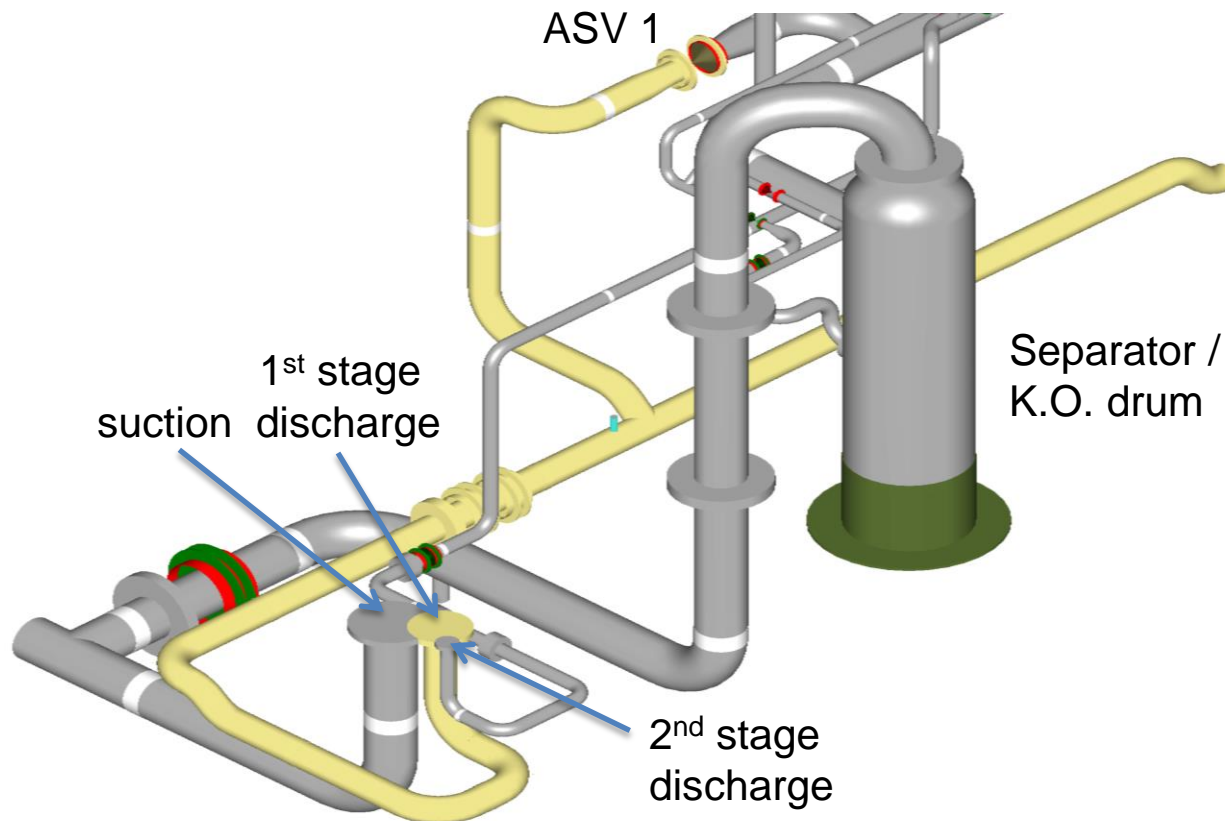
# The compressor

- 2 stage radial turbo compressor ~ 2800 rpm & 23 MW (horizontally split).
- 1 inlet and 2 outlets underneath compressor.
- Discharge stages connected with suction via anti-surge valves (ASV 1 and 2).



# The installation

- Essential part: separator, suction system, 1<sup>st</sup> stage discharge (yellow), turbo-compressor.



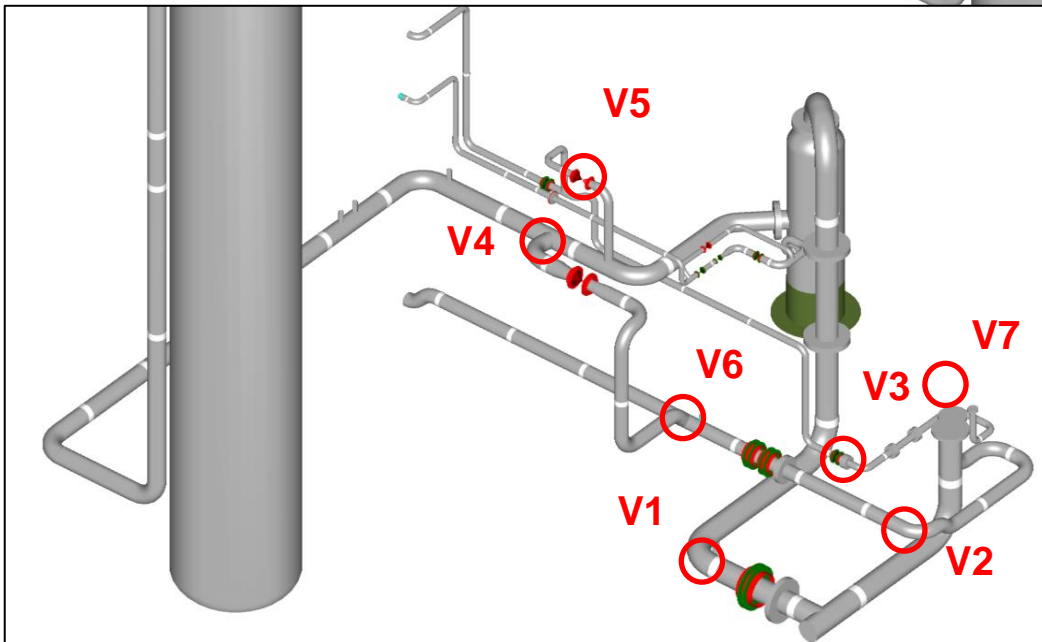
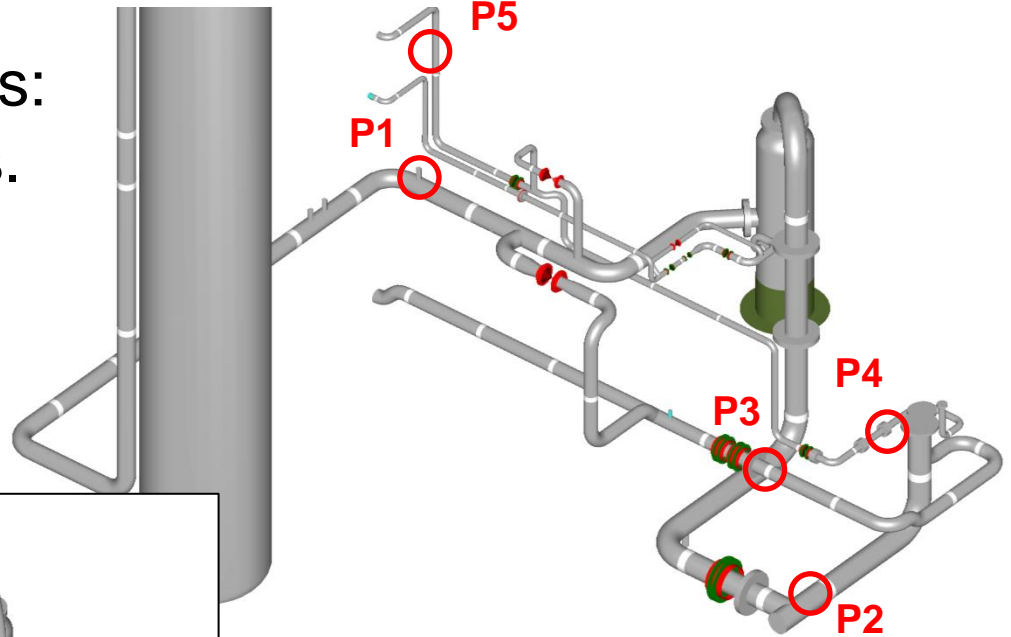
# 1 - Pipe system design

- Original design did not account for pulsations and vibrations.
- Flexible / spring pipe supports with gaps.
- After first start-up large vibrations.
- Pipe supports have been reinforced significantly.
- Verification measurements showed acceptable vibration levels.
- Rotor vibrations still present.
- Root cause unknown.



# Pulsation and vibration measurements

- Fixed measurement points: 5 puls. & 7 vibr. Locations.
- Also measurements with hand held equipment: 27 locations (tri-axial).

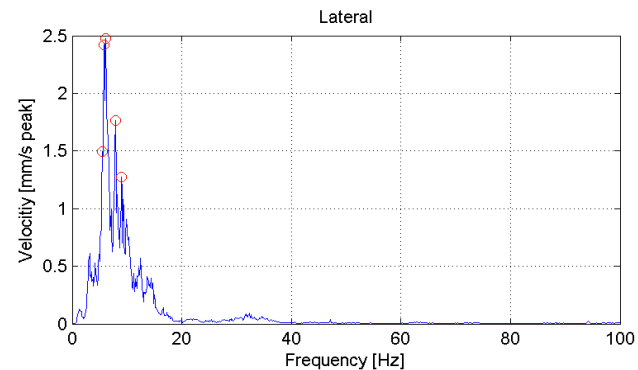
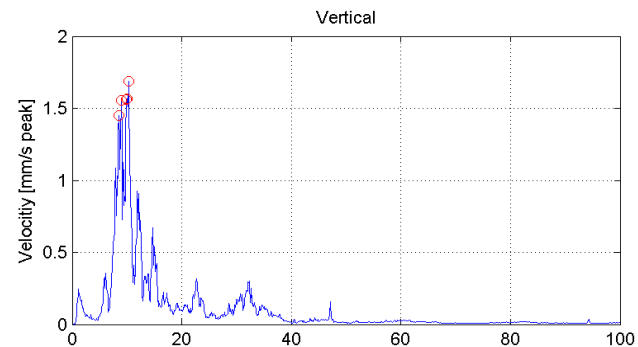
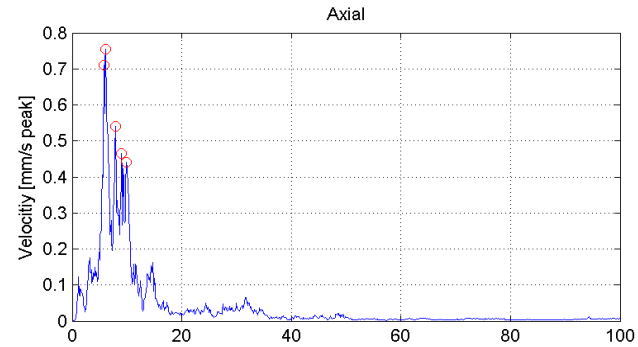


- Measurement program:
- Varying ASV settings.
- Varying RPM / load.



# Measured vibrations

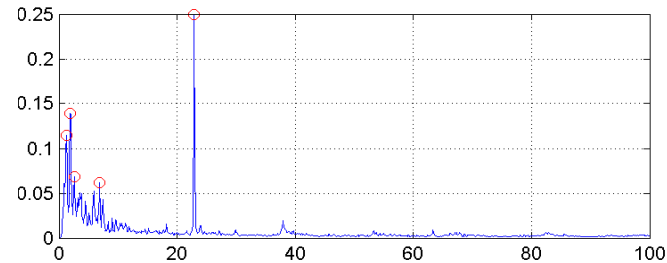
- Typical vibration spectra on V2 (close to compressor) show vibrations at low frequencies:
  - 0 – 15 Hz.
  - 20 – 40 Hz.
  - 45 Hz (compr. speed).
- Due to the improved pipe supports vibration levels are acceptable, both displacements and velocities.



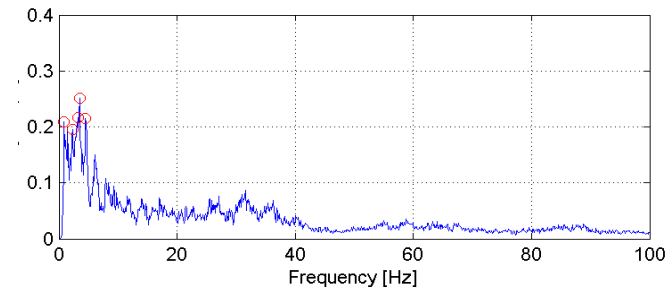


# Measured pulsations

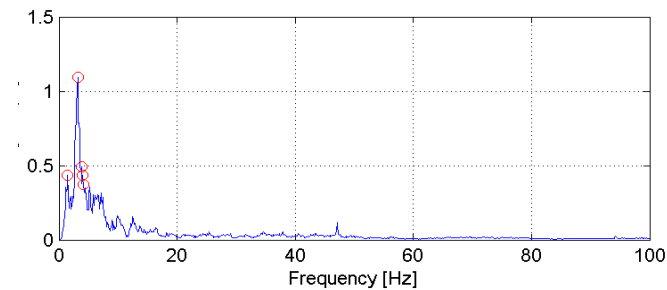
- Typical pressure spectra show acoustic resonances at low frequencies → flow induced pulsations (FIPs).
- Pulsations reach the allowable pulsation levels of API 618 .... For reciprocating compressors!
- Pulsation levels up to approximately 16 kPa → vibration forces in the order of 10 kN on the piping @ 3.2 Hz!



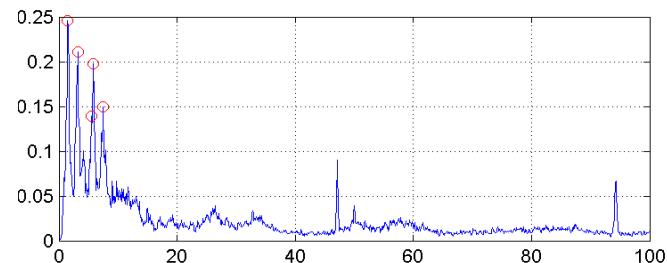
Freq. [Hz]	Pulsations [kPa pk]
22.9	0.25
1.9	0.14
1.3	0.11
2.6	0.07
6.9	0.06
2.7 [kPa pkpk]	



Freq. [Hz]	Pulsations [kPa pk]
3.5	0.25
3.2	0.22
4.5	0.22
0.9	0.21
2.3	0.20
8.3 [kPa pkpk]	



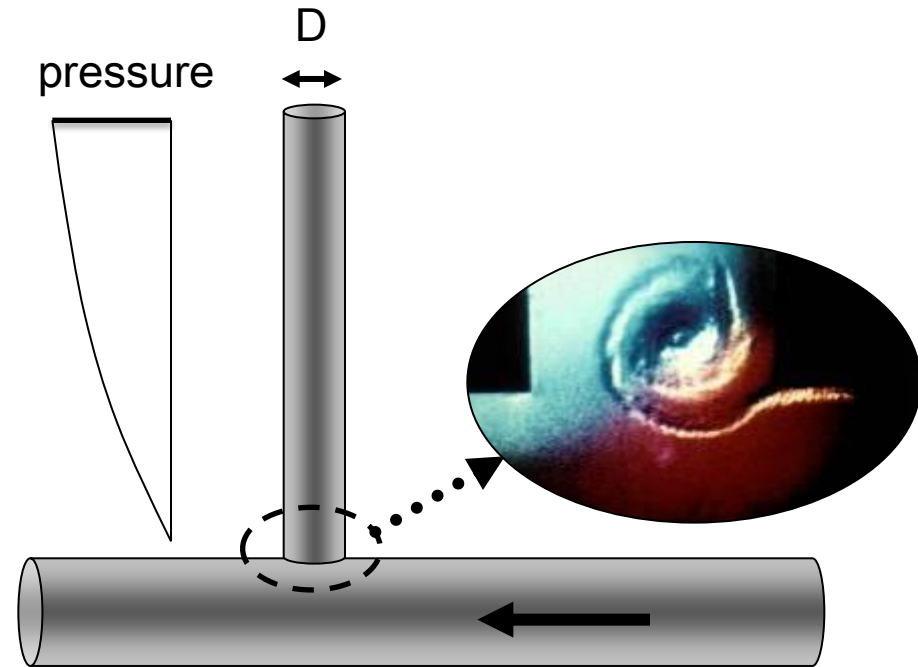
Freq. [Hz]	Pulsations [kPa pk]
3.2	1.09
3.8	0.49
1.5	0.44
4.0	0.43
4.2	0.37
15.8 [kPa pkpk]	



Freq. [Hz]	Pulsations [kPa pk]
1.5	0.25
3.2	0.21
5.8	0.20
7.5	0.15
5.5	0.14
7.8 [kPa pkpk]	

# Flow Induced Pulsations

- Pulsations are caused by vortex shedding in a T of a closed side branch.
- The vortex frequency depends linearly on the flow velocity and diameter of the side branch.
- Pulsations are amplified if the vortex frequency is equal to the resonance frequency of the side branch.

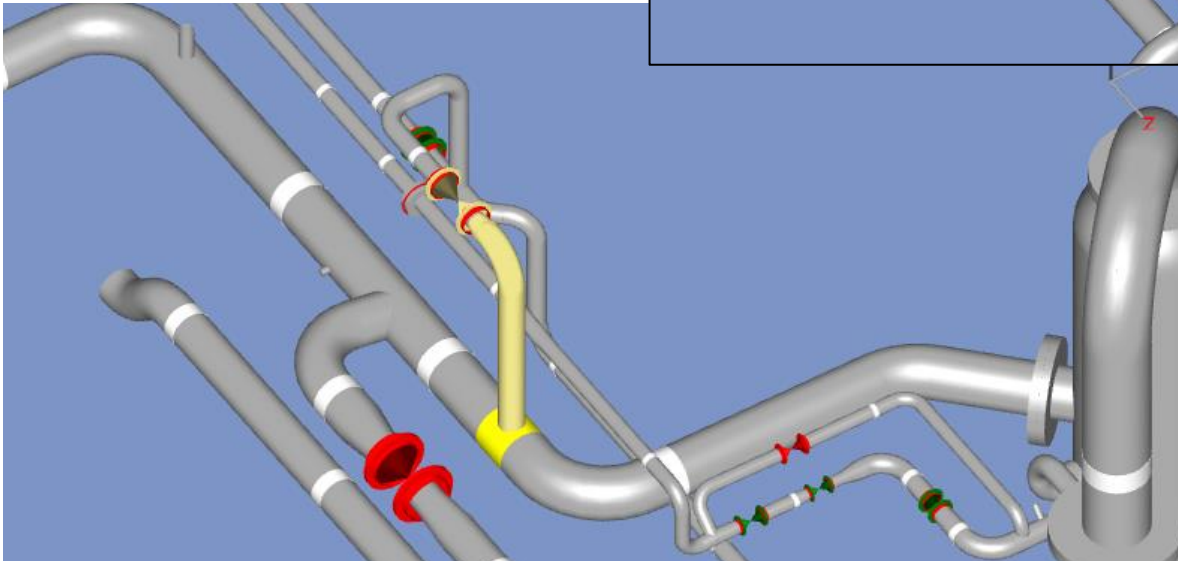
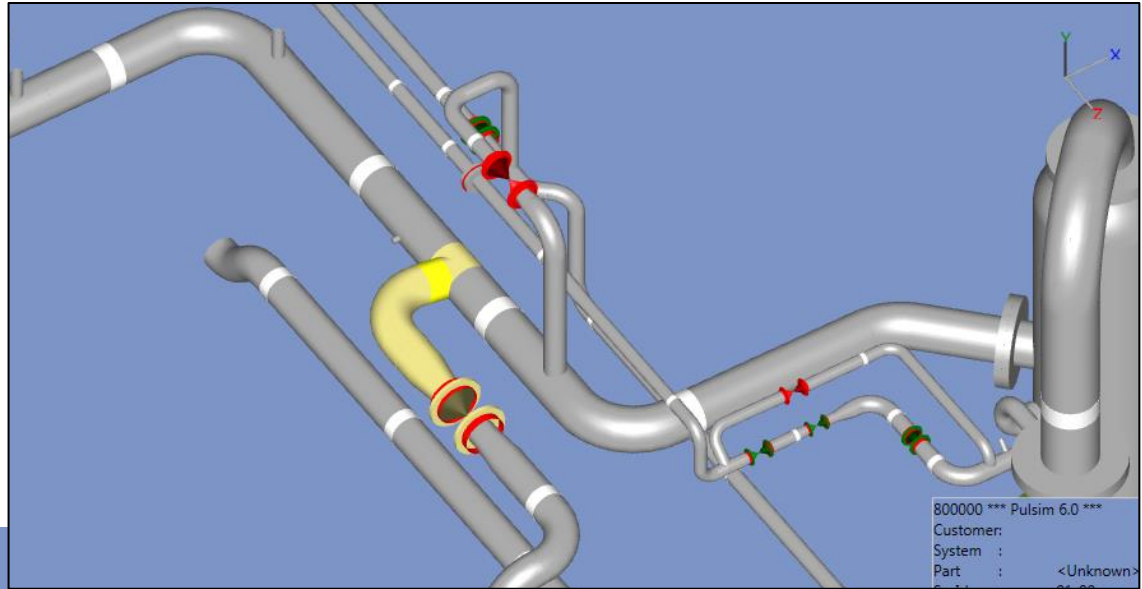


$$f = Sr \cdot \frac{U_0}{D}$$

$$Sr \approx 0.4$$

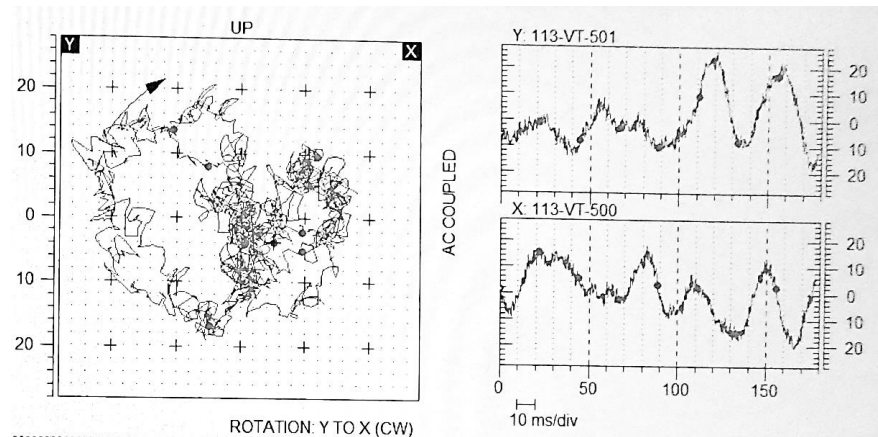
# Flow Induced Pulsations

- Examples for the present system:
- 1<sup>st</sup> and 2<sup>nd</sup> stage ASV lines, when valves (partially) closed.

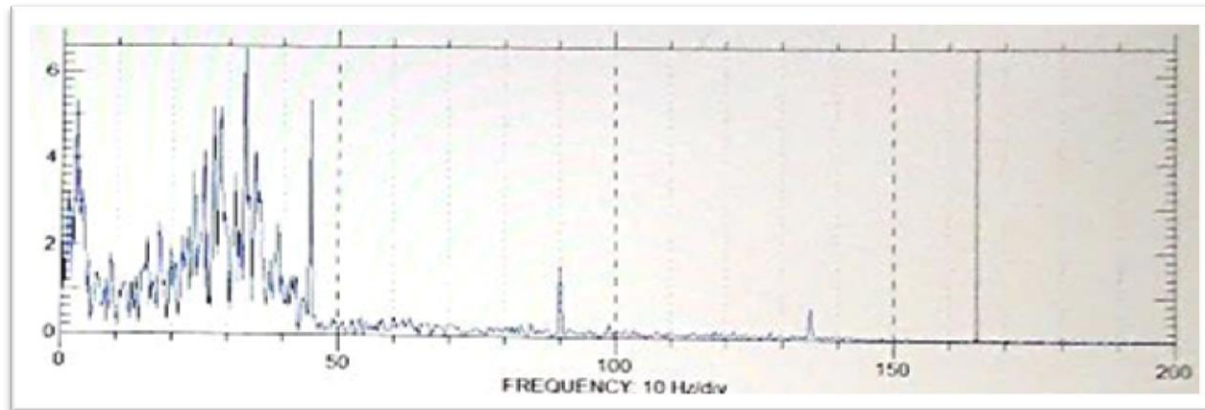


# Rotor vibrations

- Proximity probes on rotor show instability in orbits at  $> 90\%$  load.



- Rotor vibrations have similar frequencies as both the vibrations and pulsations:  $< 10\text{Hz}$ , **20 – 40 Hz** and 45 Hz.



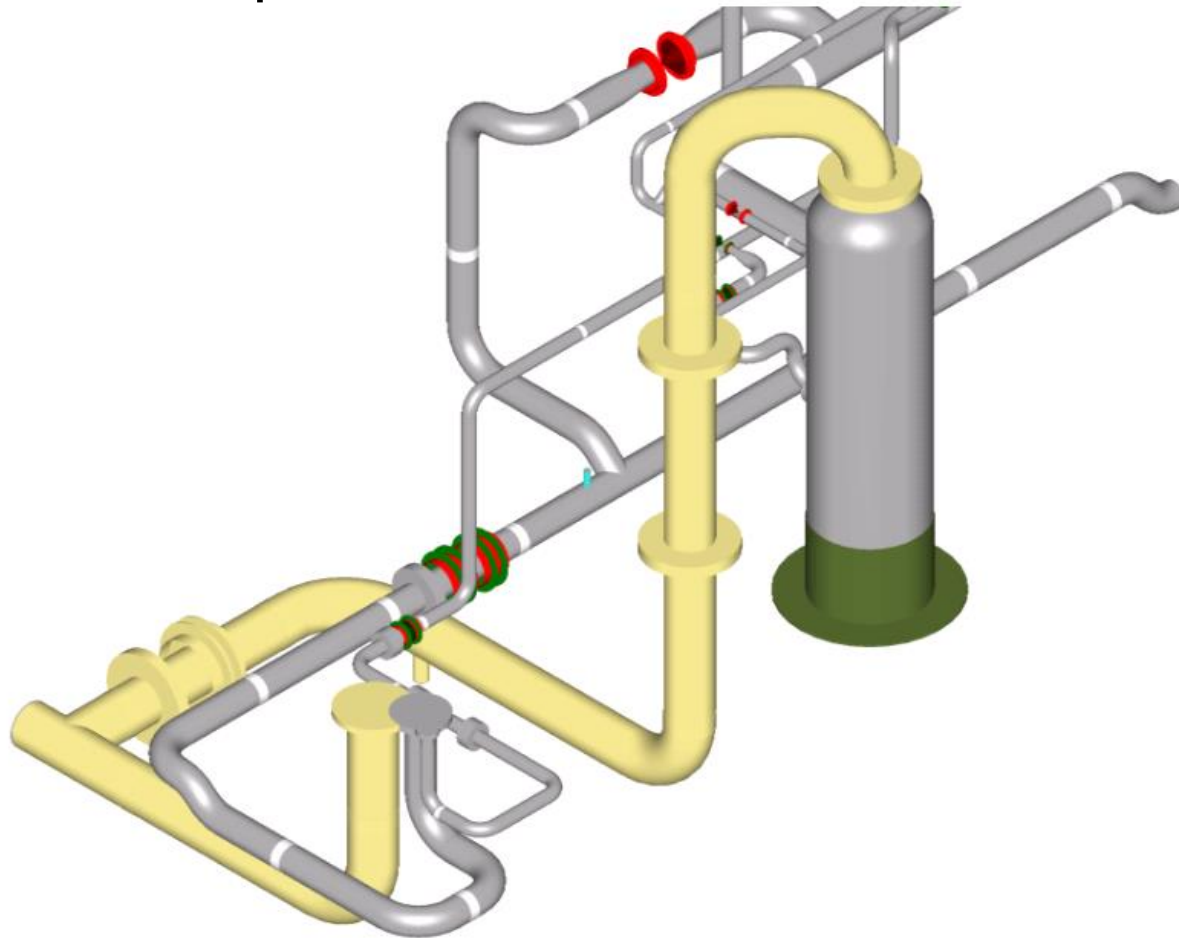
Typical rotor displacement spectrum

# Rotor vibrations

- Although the API617 Level II stability criteria are met (log. dec.  $>0.1$ ), still high rotor vibration occur.
- First critical of the rotor around 36 Hz (on site mech. run test)
  - first lateral resonance mode excited by a broadband source around this frequency.
- Vibrations in 20-40 Hz range increase with increasing compressor speed / flow.
- A Root Cause Analysis has been made.

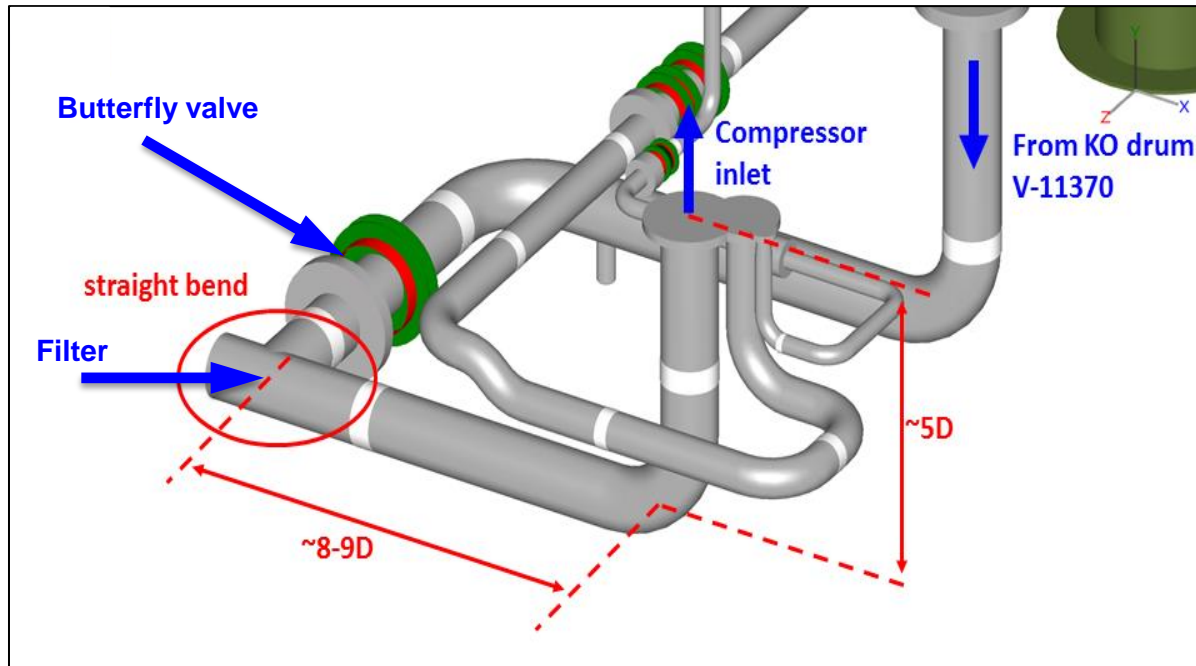
# Root Cause Analysis (RCA) rotor vibrations

First: overview suction side compressor, between K.O. drum and compressor inlet:



# Suction side compressor

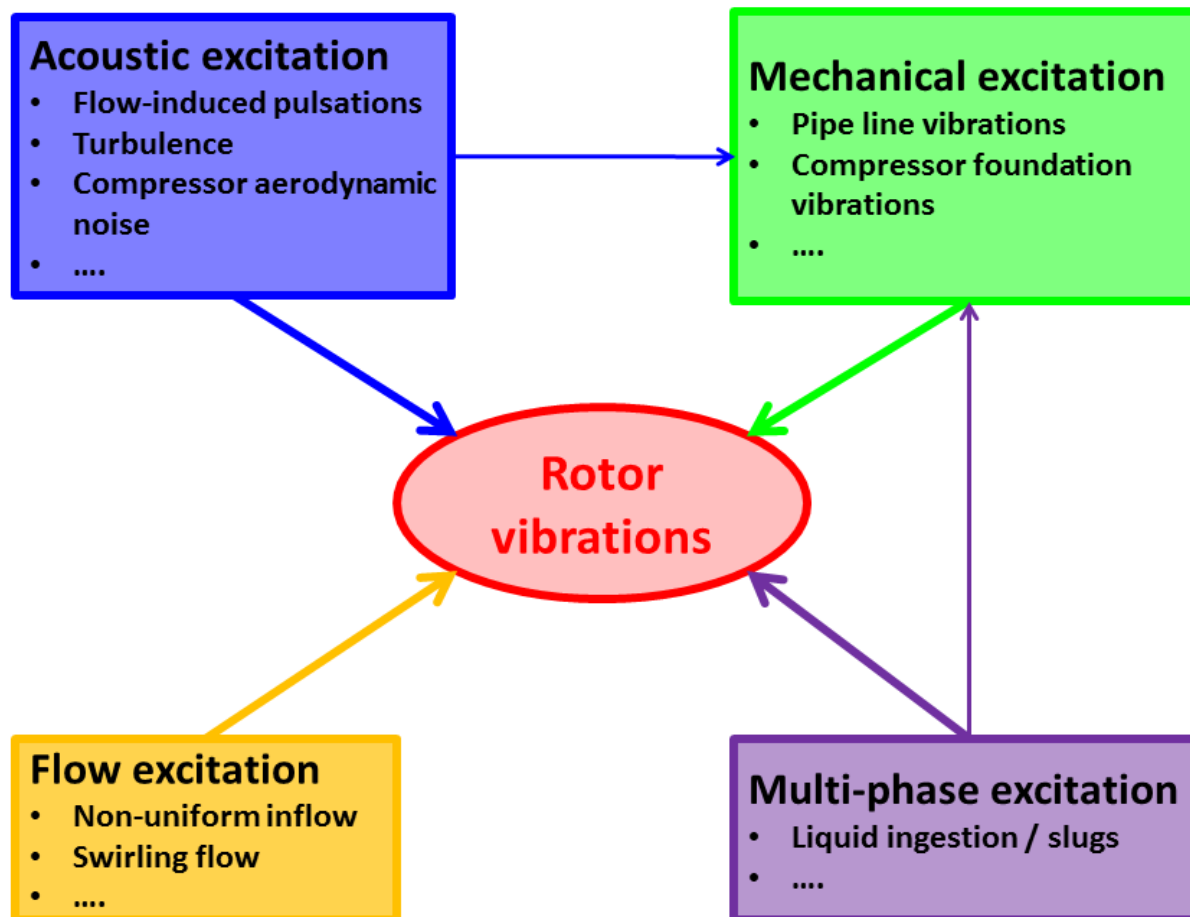
- Several out of plane sharp bends in the suction piping.
- Distance between sharp 90 degr. T joint and elbow  $<10D$ .
- Distance between elbow and inlet  $\sim 5D$ .
- Filter in T joint.
- Butterfly valve just upstream T joint.
- 60  $\rightarrow$  48 inch reduced just before compressor inlet.
- Low point in between K.O. drum and inlet: 20 inch draining boot.





# Root Cause Analysis (RCA) rotor vibrations

Schematic of possible mechanisms that can lead to rotor vibrations



# RCA matrix

	Mechanism	Description	Mitigation measures	Judgment
1	Acoustic – compressor aerodynamic noise	Tonal, high-frequent excitation, caused by rotor-stator interactions		Unlikely
2	Mechanical – piping vibrations	Connecting piping vibrations exciting the compressor and triggering the rotor instability		Unlikely
3	Mechanical – foundation vibrations	Concrete pedestal vibrations are mechanically exciting the compressor and rotor		Unlikely
4	Rotating stall in the compressor	Flow in impeller gets unstable at a certain load		Unlikely RS occurs at reduced flow
5	Mechanical malfunction in compressor	Run out of clearance → rubbing		Unlikely

# RCA matrix

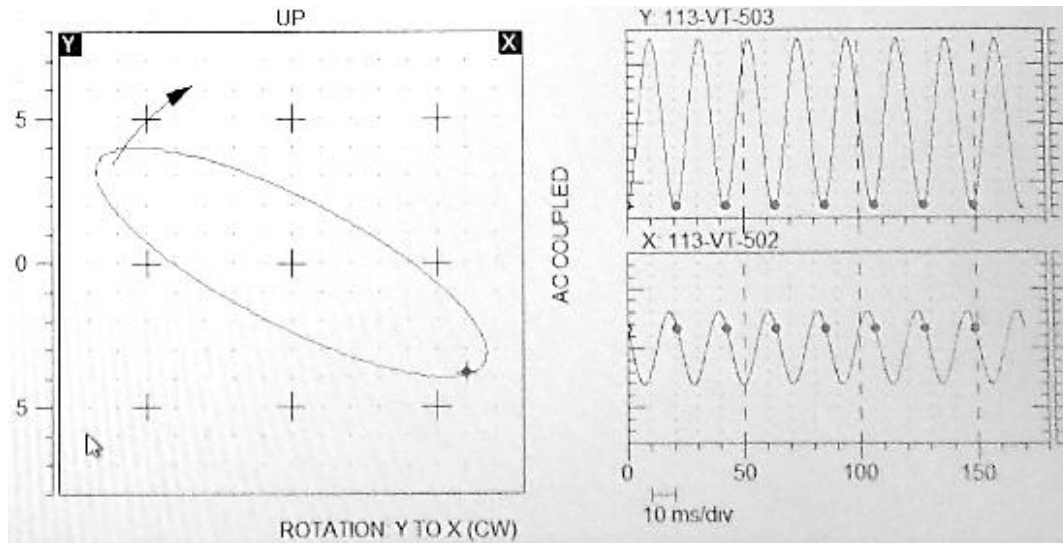
	Mechanism	Description	Mitigation measures	Judgment
6	Acoustic – flow-induced pulsations	Resonance in closed side branch; vortex shedding	Relocation of valve; reduce flow speed in main piping; apply restriction in branch	Likely to occur, but not the critical effect for rotor vibrations
7	Acoustic – pressure fluctuations caused by turbulence in flow	Broad-band, low-frequent excitation of impeller and rotor	Reduce flow speed	Likely
8	Multi-phase excitation – liquid ingestion	Accumulated liquid is entrained into the compressor; varying liquid → unsteady load on the rotor	Improve separator, avoid liquid accumulation in upstream piping; thermal insulation piping	Likely
9	Flow excitation – non uniform inflow	Short radius elbows → varying load on compressor	Apply large radius elbows, flow straightener	Likely
10	Flow excitation – swirling inflow	Double out-of-plane elbows induce swirling flow that may not be re-developed before impacting on the compressor	Increase distance between elbows and compressor, flow straightener	Likely

# RCA rotor vibrations

## 1. Liquid in suction flow.

The internals of the K.O. drum have been modified:

- Liquid carry over to compressor inlet mitigated.
- Compressor now runs stable up to 106% compressor speed!
- Rotor vibration amplitudes still high (50  $\mu\text{m}$  pk-pk). However, this is acceptable according to compressor manufacturer.



# RCA rotor vibrations

## 2. FIPs in combination with flow distortion.

The high vibration amplitudes can be caused by FIPs and flow distortion:

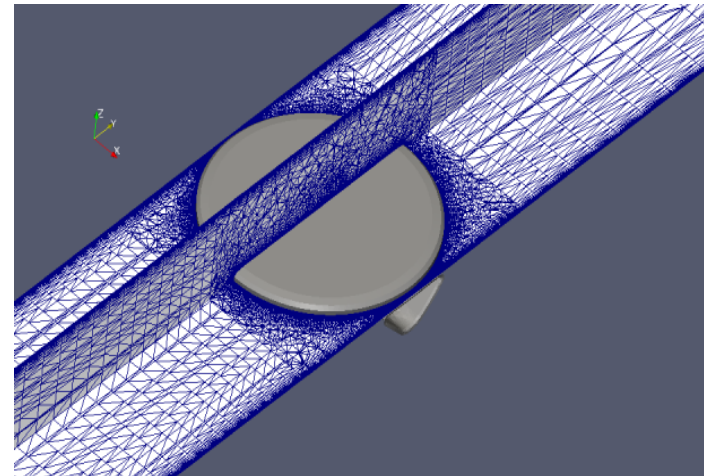
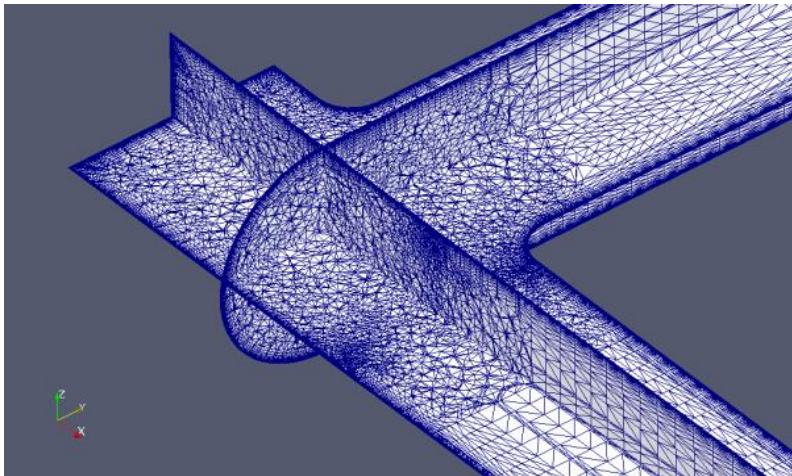
- Sharp bends in the suction piping can induce unsteady flow distortion.
- Double out-of-plane bend will cause (unsteady) swirl in flow.
- Reducer close to compressor inlet can increase flow distortion.

→ CFD analysis compressor inlet section performed.

Note: especially combination of rather undamped rotor and flow FIP / distortion can lead to high rotor vibration amplitudes.

# RCA rotor vibrations

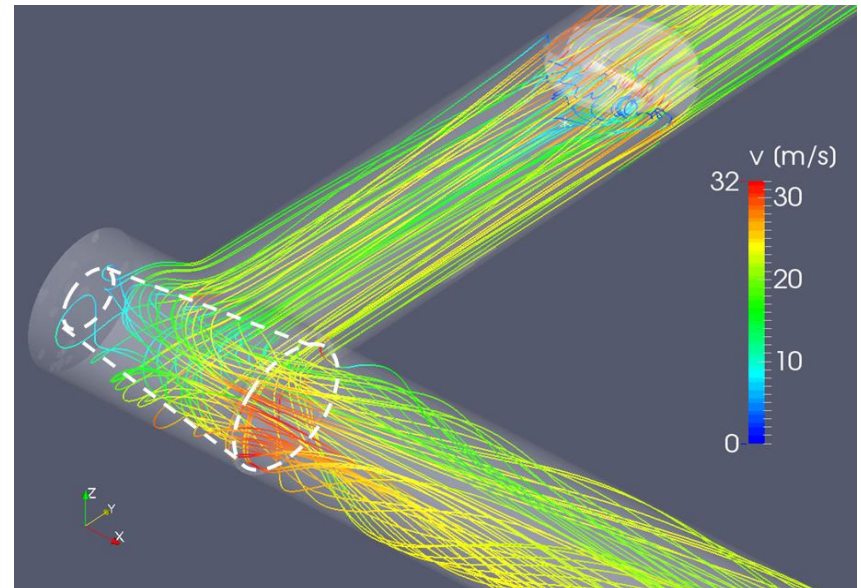
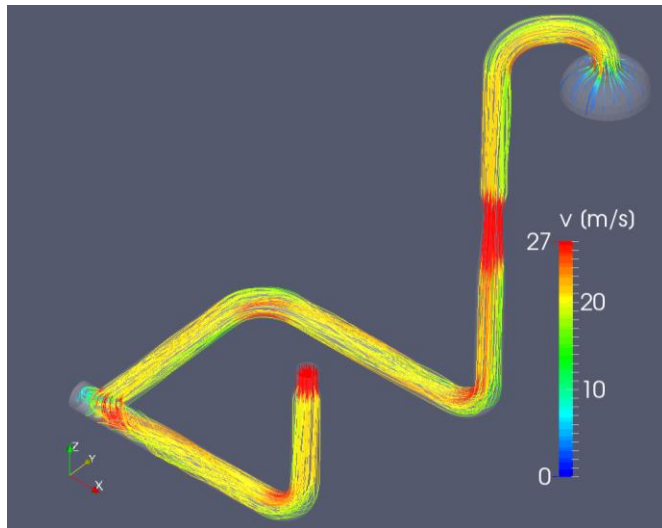
2. FIPs in combination with flow distortion – CFD analysis.
  - High Reynolds number and large geometry dimension require super-fine boundary-layer mesh.
  - Also very fine mesh needed at butterfly valve and filter section.
  - Code-to-code comparison carried out; separation behaviour checked.



# RCA rotor vibrations

## 2. CFD analysis - results.

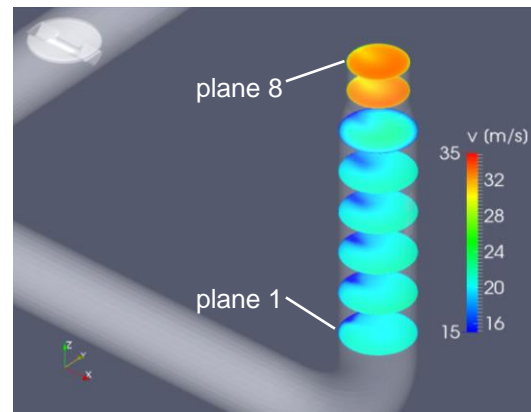
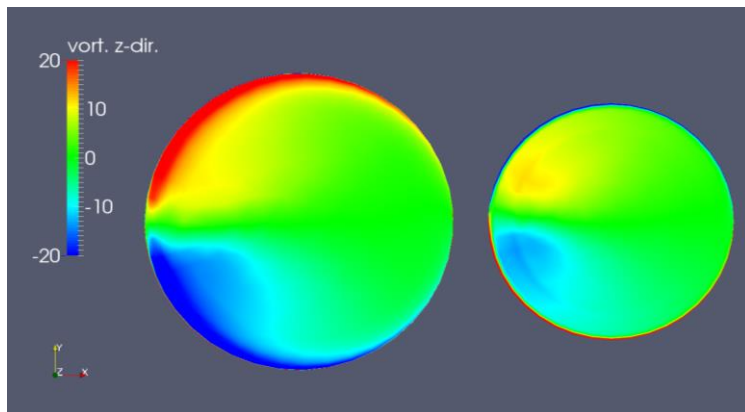
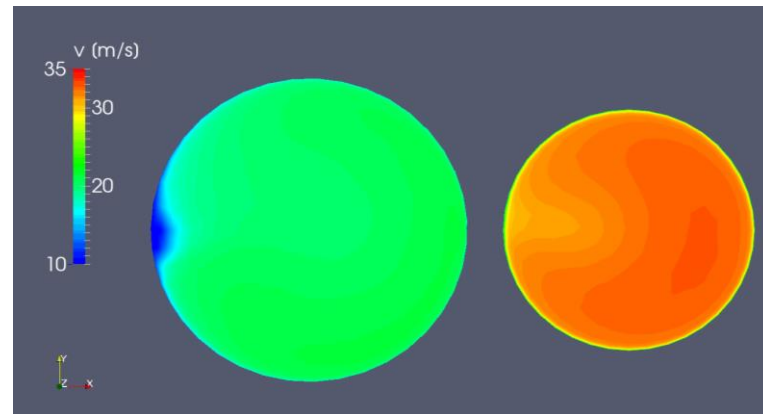
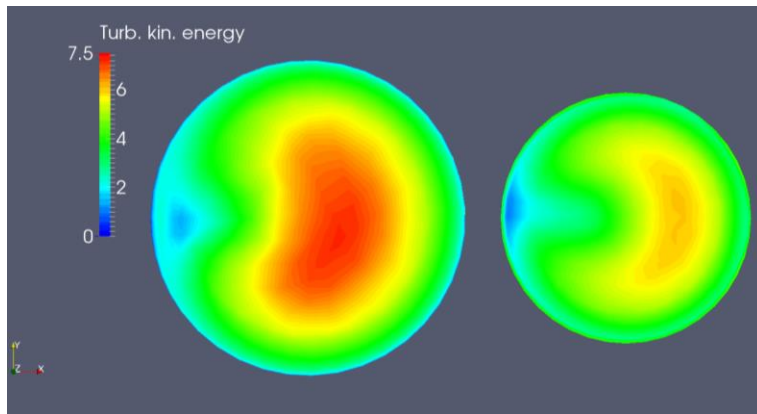
- Filter dominant obstacle:
  - Imposing the main pressure drop.
  - Redirection of flow at large scale vortical structures and small scale turbulence.
- K.O. drum inflow turbulence no significant impact on flow topology.
- Generally no flow separation.





# RCA rotor vibrations

- CFD analysis - Compressor inlet conditions; velocity, turbulence kinetic energy and vorticity (z-direction).



Only weak, counter rotating vortices at the inlet.

# Conclusions

- Design philosophy did not consider **Flow Induced Pulsations (FIPs)**.
- Improved pipe supports reduced vibrations but do not eliminate the source.
- Rotor instability mainly caused by liquid carry over K.O. drum to compressor inlet.
- No fluctuating swirling flow into the compressor, mainly due to pressure drop over filter and high Reynolds number flow.
- Rotor most likely too susceptible for disturbances. Not critical anymore, but additional improvements planned.

# Lessons learned

- FIPs can cause serious vibration problems at low frequencies.
- A pulsation and vibration analysis for this large diameter pipe systems should be part of the design.
- 3D pipe bend configuration in the suction piping can lead to flow distortion. To avoid this long radius bends should be applied or guide vanes could be installed in the bends.
- Take actual inlet flow into account in rotor damping (seals) and stiffness (bearing clearance) design.

Thank you for your attention!

Pieter van Beek

TNO

Heat Transfer & Fluid Dynamics

tel. +31 (0)88 8666366

[Pieter.vanbeek@tno.nl](mailto:Pieter.vanbeek@tno.nl)