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# Composites on fire at reduced scale: evaluation, characterization and modeling

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**COMPOSITES ON FIRE AT  
REDUCED SCALE:  
EVALUATION,  
CHARACTERIZATION AND  
MODELING**

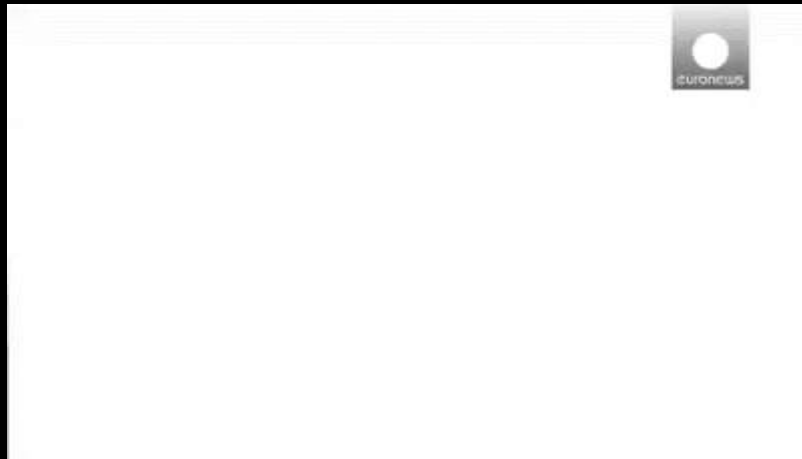
**Serge BOURBIGOT**

**R<sub>2</sub>FIRE@UMET-UMR/CNRS 8207**

➔ CFRP in aircraft structure has introduced potential fire threats



- ❑ engine compartments (fuel leakage can occur)
- ❑ fuselage (post-crash fire)



**Jet fuel fire: heat flux between 110 and 200 kW/m<sup>2</sup>**

➔ Fire resistance of fuselage and other parts of aircraft: **full scale test or burnthrough test (jet fuel fire at ~186 kW/m<sup>2</sup>)**



*Post-crash fire simulation in full scale indoor at FAA*



- ❑ Time consuming
- ❑ Expensive
- ❑ Slow development

*Burnthrough test (NexGen)*



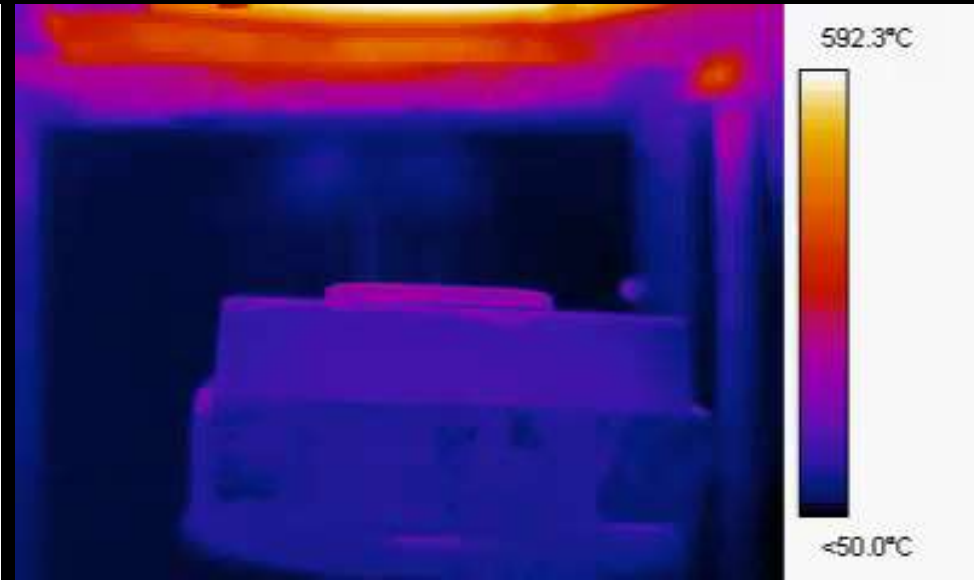
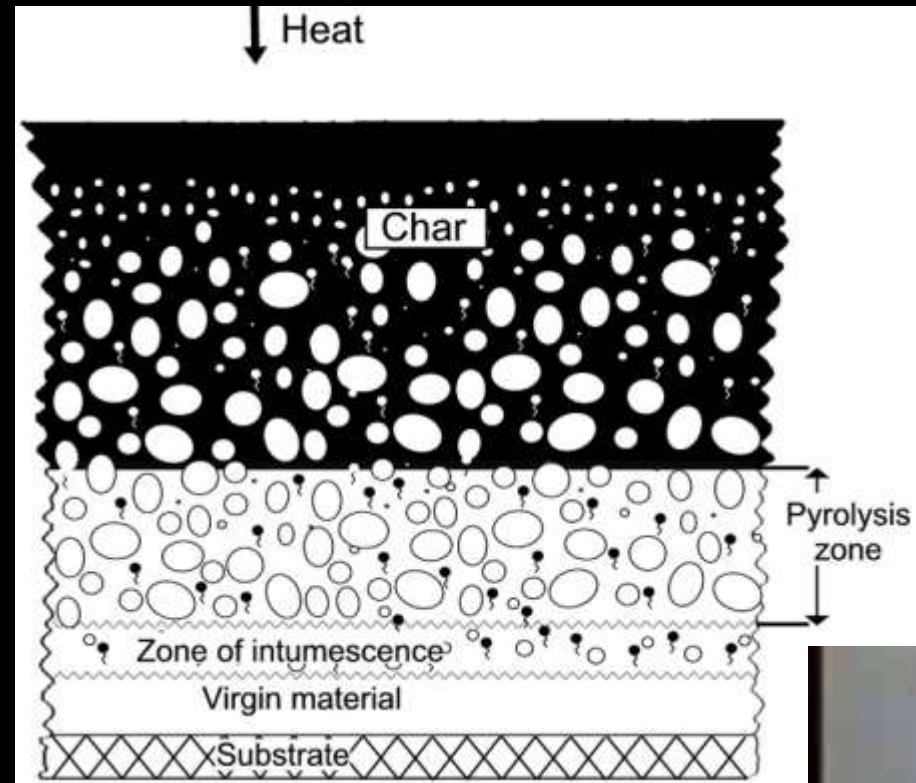
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# *Fire protection of CFRP*

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# Intumescence?



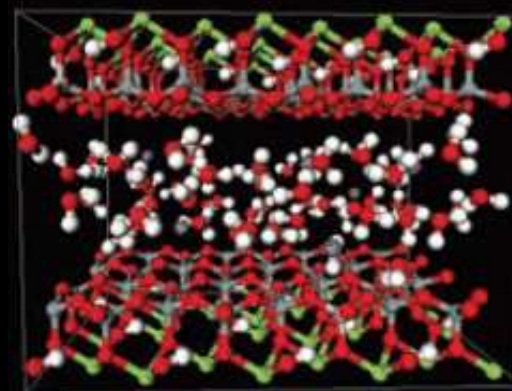
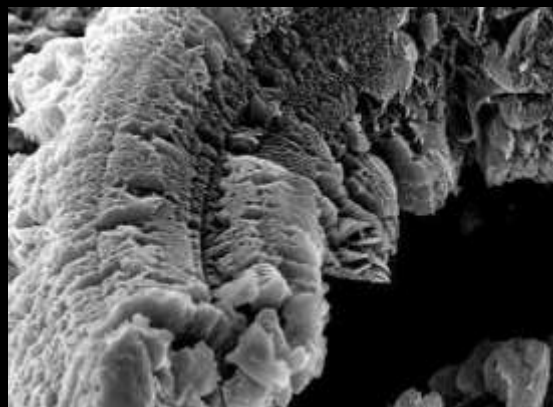
- ❑ Formation of heat barrier
- ❑ Fire protection of materials?



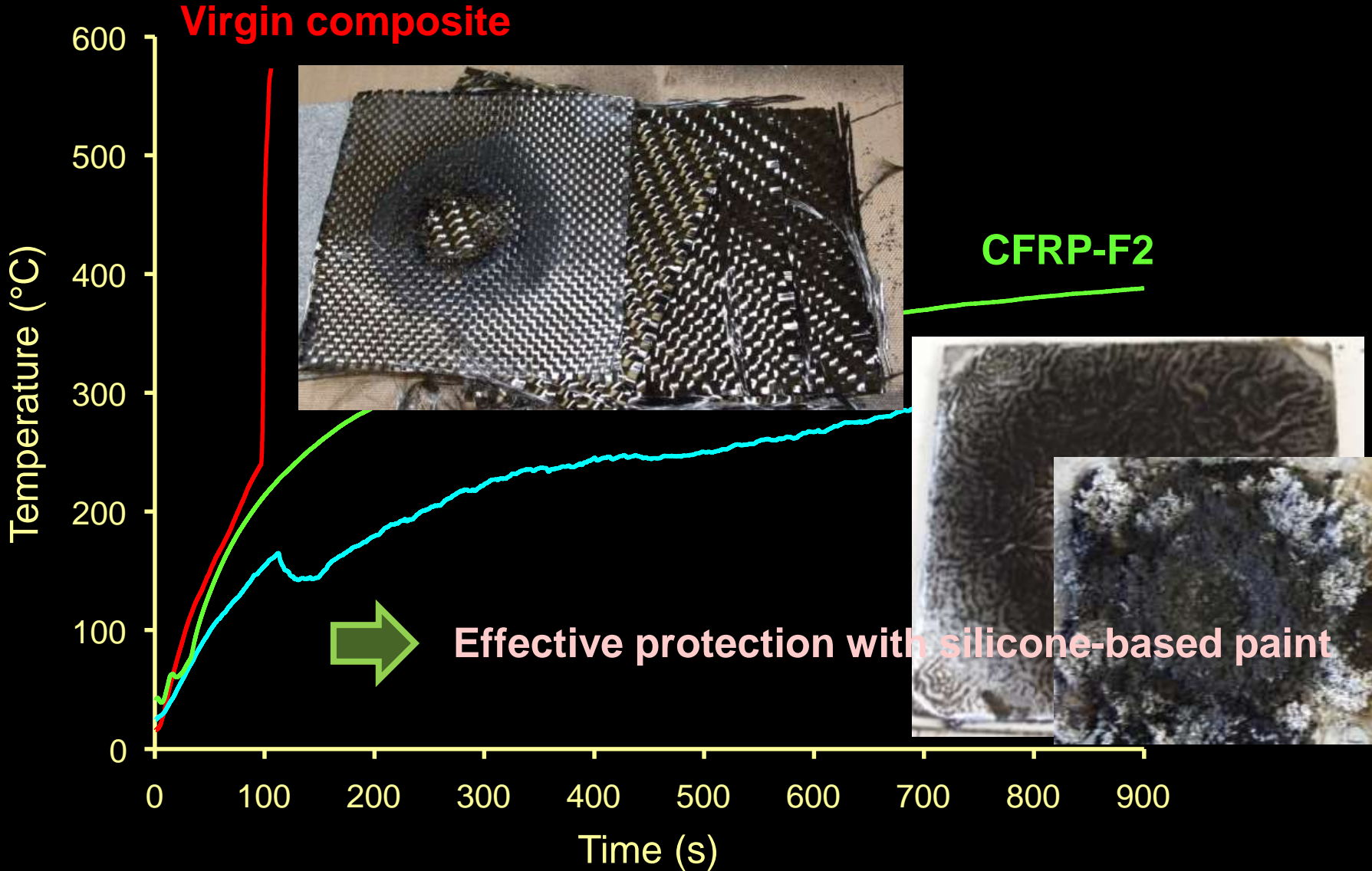
**Intumescent paint on CFRP:** silicone-based coating containing expandable graphite\* compared to low intumescenting paint



Silicone formulation	F1 – High intumescenting coating	F2- Low intumescenting coating
Silicone matrix	56%	56%
Expandable graphite	25%	-
Calcium carbonate	12%	37%
Clay	7%	7%



\*S. Bourbigot et al. "Protecting substrates against damages by fire", WO 2013/150121 - Dow Corning, 2013

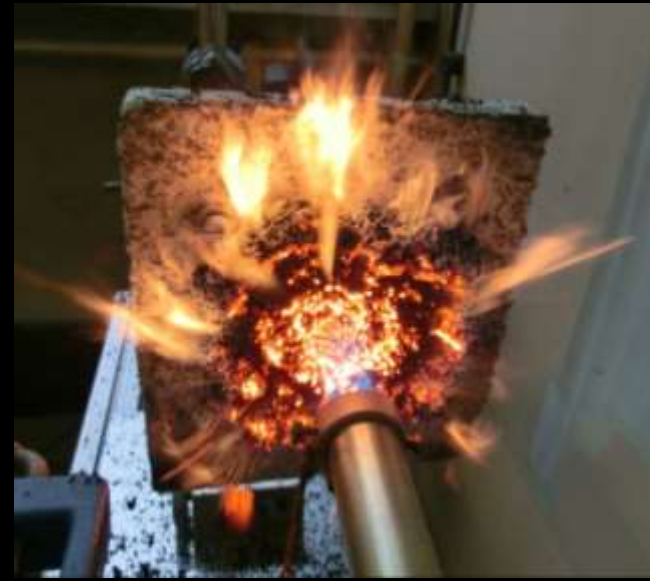




# Protection by intumescence



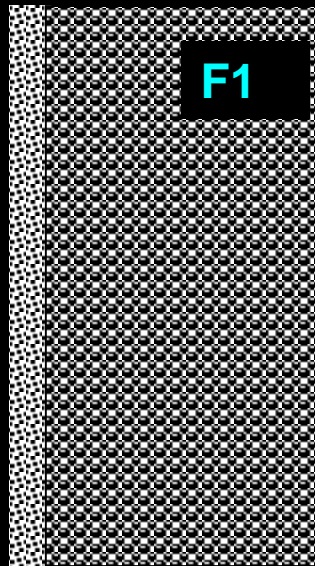
**Virgin composite**



**CFRP-F1**

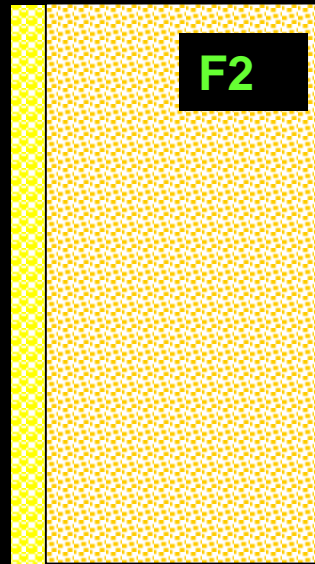
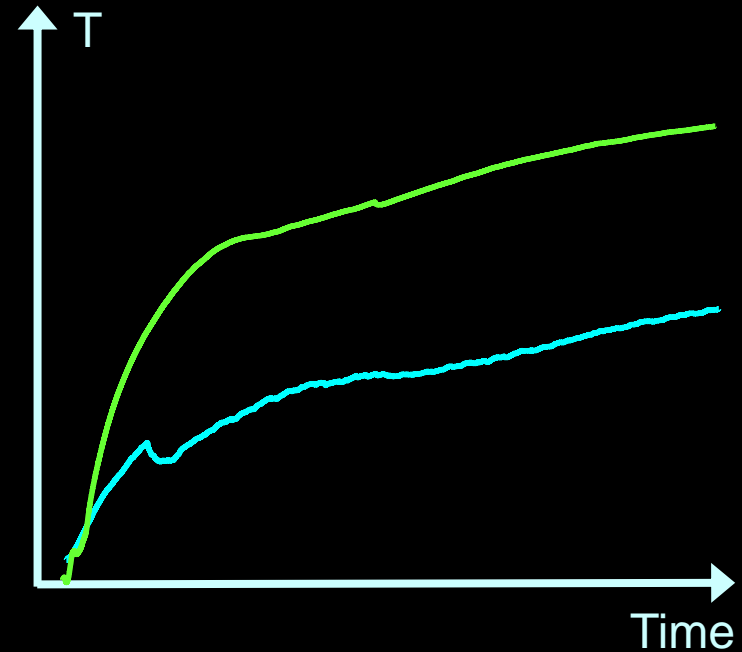


# Mechanism of protection



F1

- Heat barrier: high expansion, low  $k$  ( $0.4 \text{ W/m.K@600}^\circ\text{C}$ )
- Structure: high cohesion thanks to chemical interactions (SiC, Ca-Si)



F2

- Heat barrier: low expansion, low  $k$  ( $0.4 \text{ W/m.K@600}^\circ\text{C}$ )
- Structure: cohesive porous structure (highly polymerized Si, Ca-Si)

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# *Dimensionnall analysis: reducing the scale*

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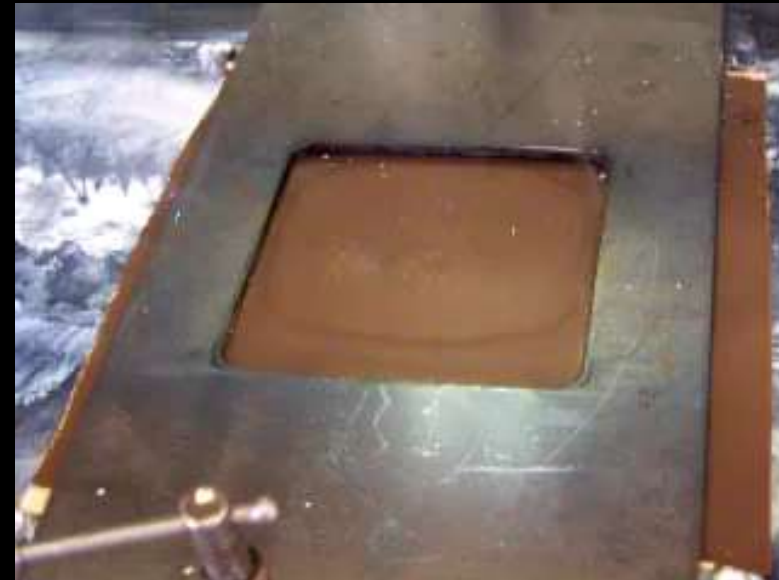


# ISO 2685: goal and test



Pass/fail test for equipment located in fire zone (engine, auxiliary unit):

- Heat flux of 116 kW/m<sup>2</sup>
- T<sub>flame</sub> of 1100°C
- Withstanding of the component for 5 min ⇒ fire proof
- Withstanding of the component for 15 min ⇒ fire resistant



$$\left\{ \begin{array}{l} \rho C \dot{T} - k \Delta \bar{T} = \frac{q_{av} - q_{ar}}{e_p} \\ q_{ar} = h_{ar}(T - T_{amb}) + \varepsilon \sigma (T^4 - T_{amb}^4) \\ q_{av} = h_{av}(x, y)(T_g - T) + C(x, y)\sigma(T_f^4 - T^4) - \varepsilon \sigma (T^4 - T_{amb}^4) \end{array} \right.$$

Dimensionless numbers are determined:

$\tau$  : duration of the experiment  
 $L$ : length of the plate

$$\tilde{x} = x/L \quad \tilde{y} = y/L \quad \tilde{t} = t/\tau \quad \tilde{T} = T/T_{amb}$$

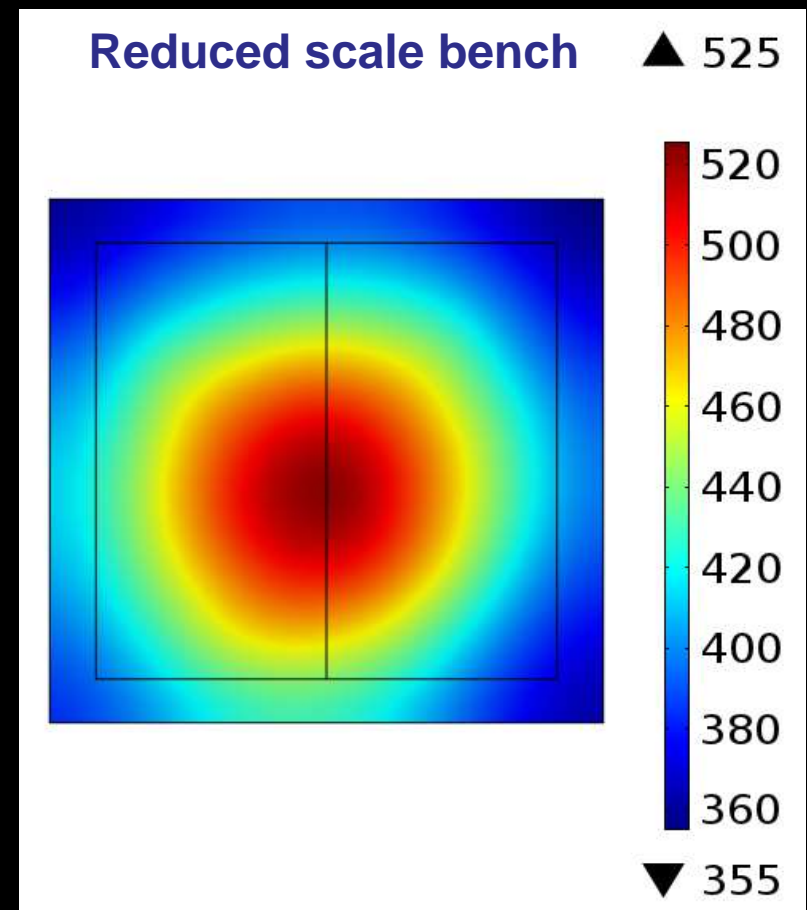
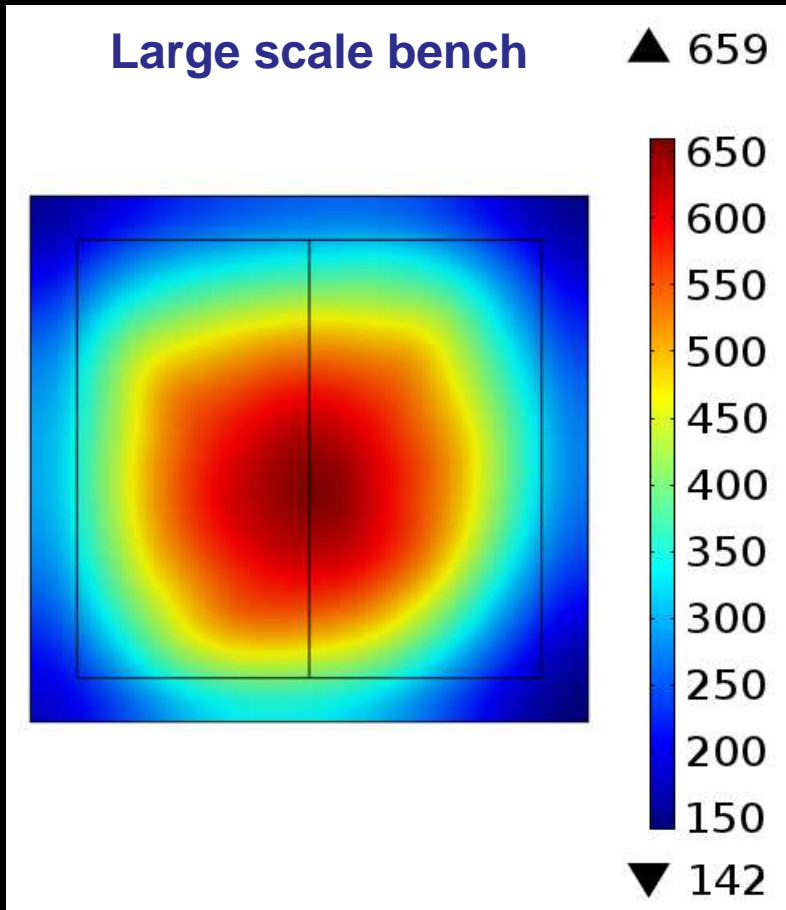
$$\frac{\tau e_p}{\tau} \frac{\partial \tilde{T}}{\partial \tilde{t}} - \left(\frac{e_p}{L}\right)^2 \Delta \tilde{T} = B_i^{av} \left(\frac{T_g}{T_{amb}} - \tilde{T}\right) + CN_r \left(\left(\frac{T_f}{T_{amb}}\right)^4 - \tilde{T}^4\right) - 2\varepsilon N_r (\tilde{T}^4 - 1) - B_i^{ar} (\tilde{T} - 1)$$

$$B_i^{av} = \frac{e_p h_{av}}{k} \quad B_i^{ar} = \frac{e_p h_{ar}}{k}$$

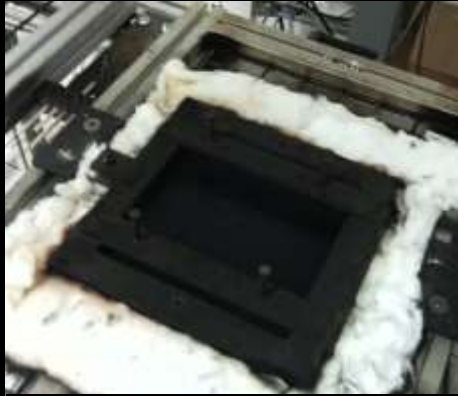
→ Biot numbers linked to the convection on the 2 faces

$$F_o^{ep} = \tau / \tau_{ep} \quad \tau_{ep} = \frac{\rho C_p e_p^2}{k} \quad N_r = \frac{e_p \sigma T_{amb}^3}{k} \quad \longrightarrow \text{Fourier, time and radiative numbers}$$

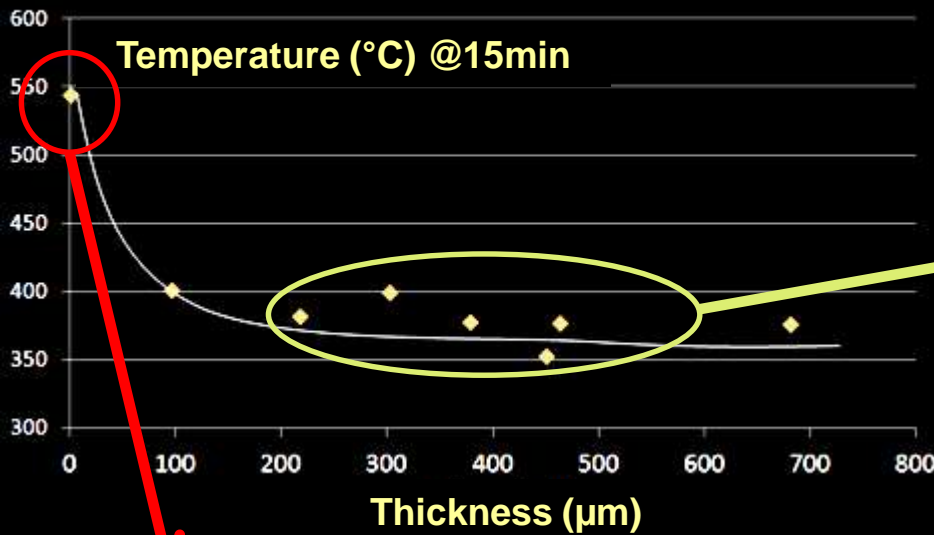
- ➔ **Simulated scenario:**
- Scale divided by 3 except sample thickness
  - Same heat transfer
  - Same duration



**Lower temperature field for the small- scale bench**



Evaluation of  
intumescent  
CFRP



Efficiency of the fire protection from  
250 µm via an intumescent behavior

# Summary and Conclusions

- **Similitude:** *scale reduction is not straight forward but correlation can be found simulating scale reduction*
- **Modeling:** *numerical simulation and optimization for the development of small scale bench*
- **Intumescence:** *efficient method to fire protect CFRP for aircraft and building applications*



# Research Group & Acknowledgement



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