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Frederic Monteverde
ISTEC-CNR, Italy

Diletta Sciti
ISTEC-CNR, Italy

Laura Silvestroni
ISTEC-CNR, Italy

Davide Alfano
Italian Research Aerospace Centre

Raffaele Savino
University of Naples

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Fabrication, properties and arc-jet testing of ZrB_2 -based composite containing short SiC fibers

Frédéric MONTEVERDE*, Diletta SCITI*, Laura SILVESTRONI*
Davide ALFANO #, Raffaele SAVINO ^

(*) National Research Council of Italy - Institute of Science and Technology for Ceramics

(#) Italian Research Aerospace Centre

(^) Dept. of Aerospace Engineering – University of Naples

Ultra-High Temperature Ceramics: Materials for Extreme Environment Applications II
May 14th 2012, Hernstein (Austria)

Re-entry flight trajectories



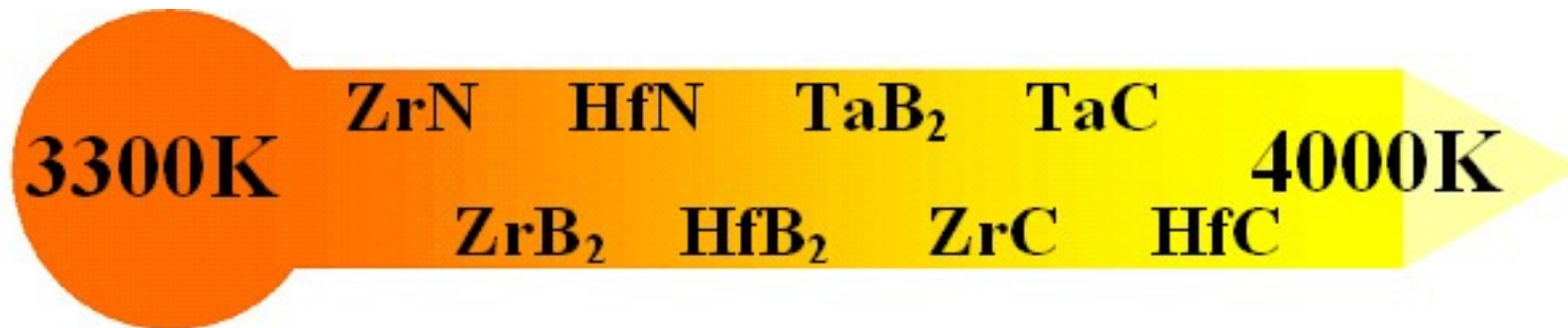
To re-enter Earth's atmosphere in an alternative way, flying with lower aerodynamic drag, larger cross range and better maneuverability along more gentle trajectories slender vehicles with sharp edges flying at moderate angles of attack are necessary.

The reduced leading edge's curvature radius implies surface temperature higher than that of the actual blunt vehicles and that could not be withstood by the conventional thermal protection system materials.

Refractory and oxidation resistant aero-surfaces (with sharp profile)

ARE REQUIRED

Ultra-high temperature ceramics



Melting point range of UHTCs (nitrides, borides and carbides of Zr, Hf, Ta)

UHTCs are currently investigated as TPS
of fuselage nosecone and wing leading edges

Key requisites for materials

Mechanical requisites: high resistance to thermal shock, high fracture toughness

Thermal requisites: high thermal conductivity, high emittance, low catalytic efficiency, high resistance to oxidation

MB_2 (M=Zr, Hf) ceramics alone possess

❖ **high temperature capabilities and high thermal conductivity**

❖ poor oxidation resistance

❖ low fracture toughness

} Add SiC (particle, whiskers, fibers)

Aim of this work

Study of the effects of SiC chopped fibers on ZrB₂-based massive ceramic

- properties
- resistance to oxidation under static conditions
- stability under aerothermal heating

Raw materials

- **ZrB₂** - (H.C. Starck) grade B

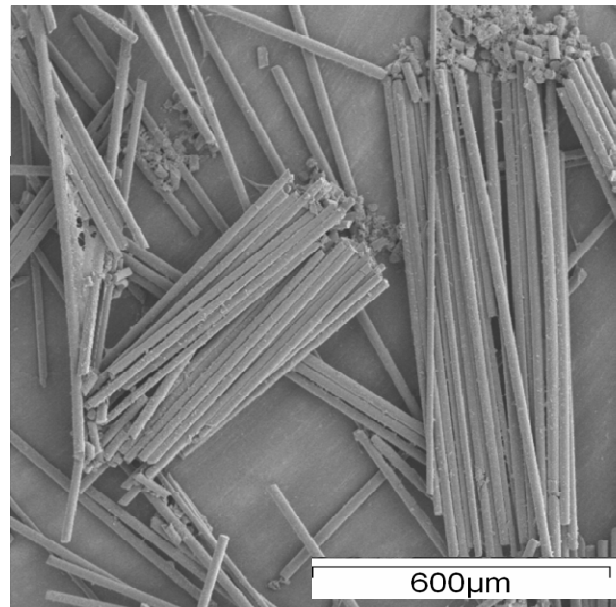
- **Si₃N₄** – Baysind

- **SiC** – HI Nicalon (COI Ceramics)

- length: 0.5-1 mm + debris

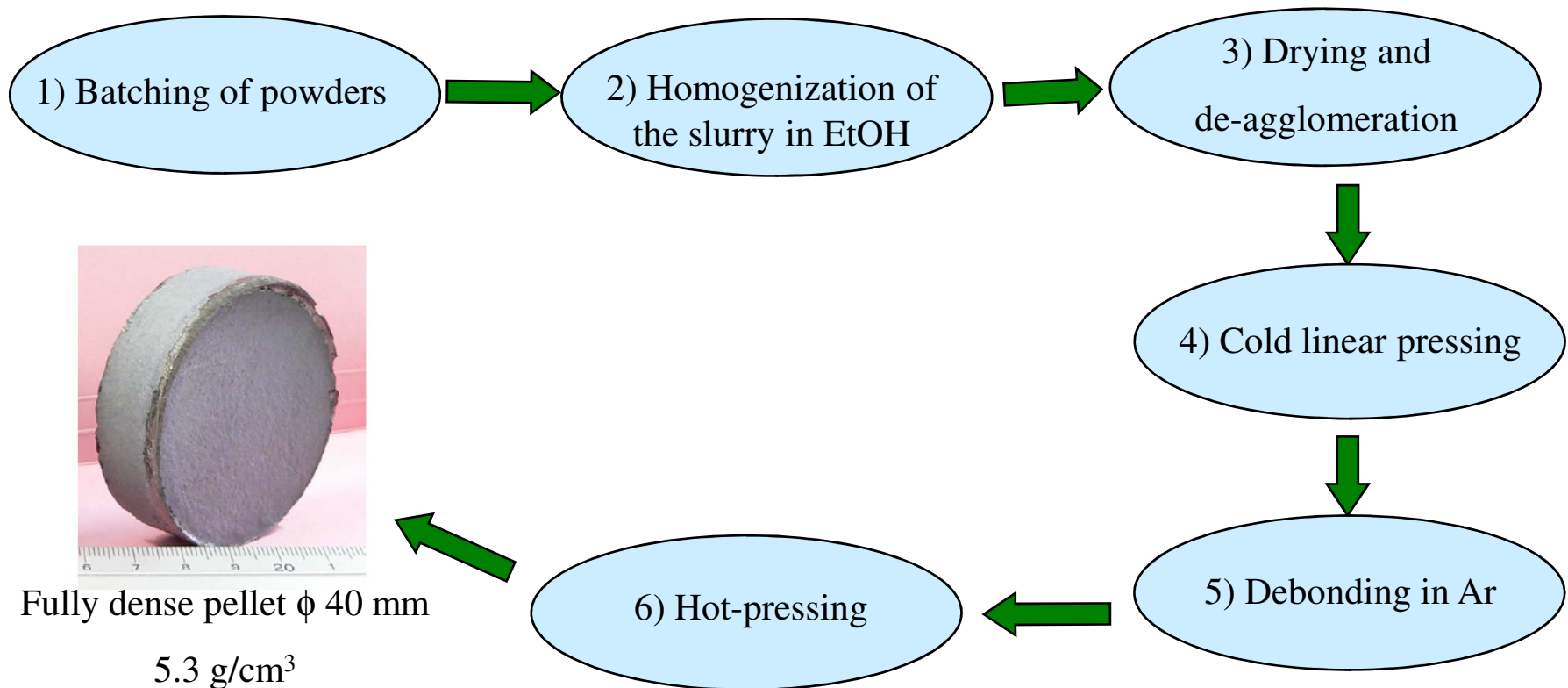
- density: 2.74 g/cm³

- composition: 55% β-SiC crystallites (2 nm), 5% C, embedded in 40% Si-C-O intergranular phase

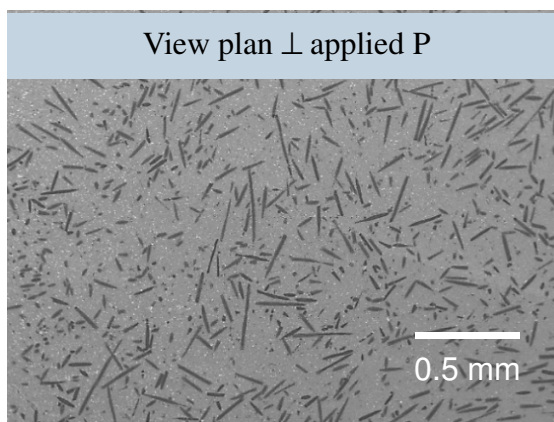


Production of $\text{ZrB}_2\text{-15\%SiC}_F$

- ✓ Starting composition (vol%): $\text{ZrB}_2 + 15\% \text{SiC}_{\text{Fiber}} + 5\% \text{Si}_3\text{N}_4$ (s.a)
- ✓ Material processing



ZrB₂-15 SiC_F: microstructure and properties



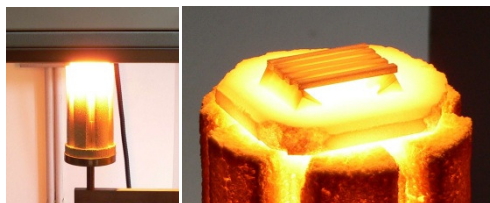
max fiber length: ~ 0.3 mm

Polished section, by SEM

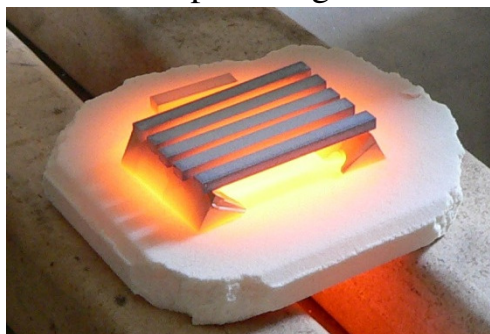
Property	Method	Specimen size	Notes	Mean value (± 1 s.d)
Elastic modulus	Resonance frequency	30 x 8 x 0.8 mm	RT	402 GPa
Flexure strength	4-pt bending	25 x 2.5 x 2 mm	RT	(453\pm 19) MPa
Flexure strength	4-pt bending	25 x 2.5 x 2 mm	1200°C air	(336 \pm 29) MPa
Fracture toughness	Chevron notched beam	25 x 2.5 x 2 mm	RT	(5.3\pm0.05) MPa\sqrt{m} (ΔK_{Ic} 50%)
Thermal shock resistance	Water quenching method	25 x 2.5 x 2 mm	RT	400-500°C (in progress)
Thermal conductivity	Laser flash	Φ 12.7 mm	RT - 1500°C	65 - 50 W/m K

Resistance to oxidation in static air

Iso-thermal run for 30 min at 1700°C



Bottom-up loading furnace



Visual appearance before oxidation

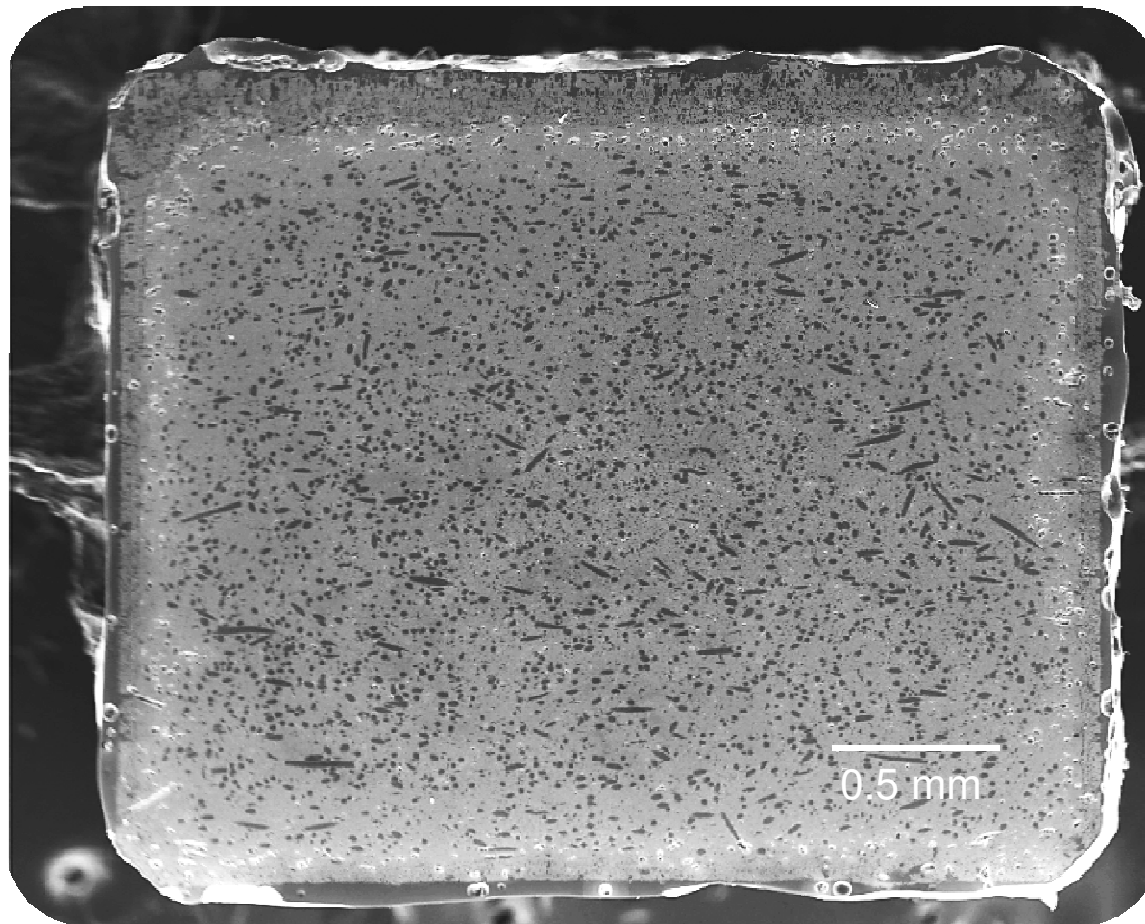


Visual appearance after oxidation

Specific mass change (w/S)
(7.5 ± 0.1) mg/cm²

Resistance to oxidation in static air

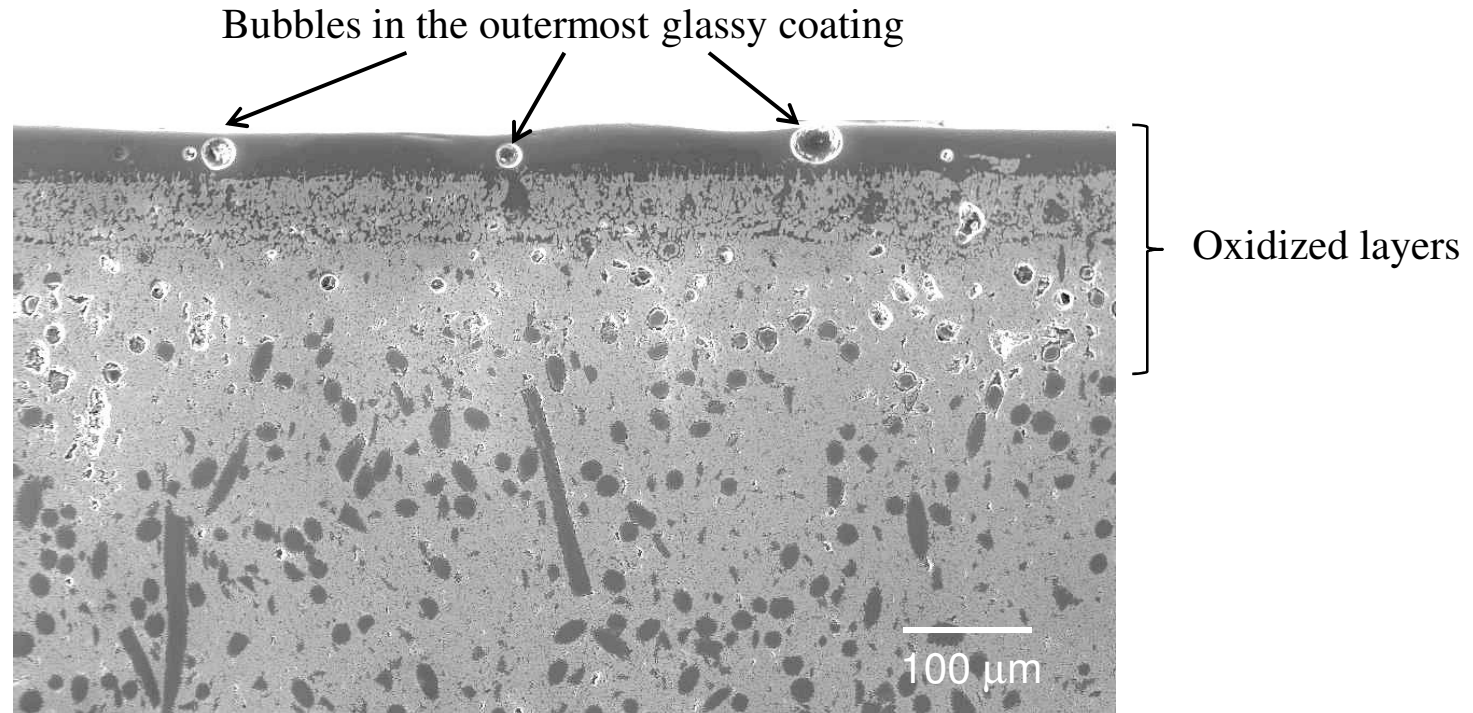
Iso-thermal run for 30 min at 1700°C



Polished cross-section of oxidized bar (SEM micrograph)

Resistance to oxidation in static air

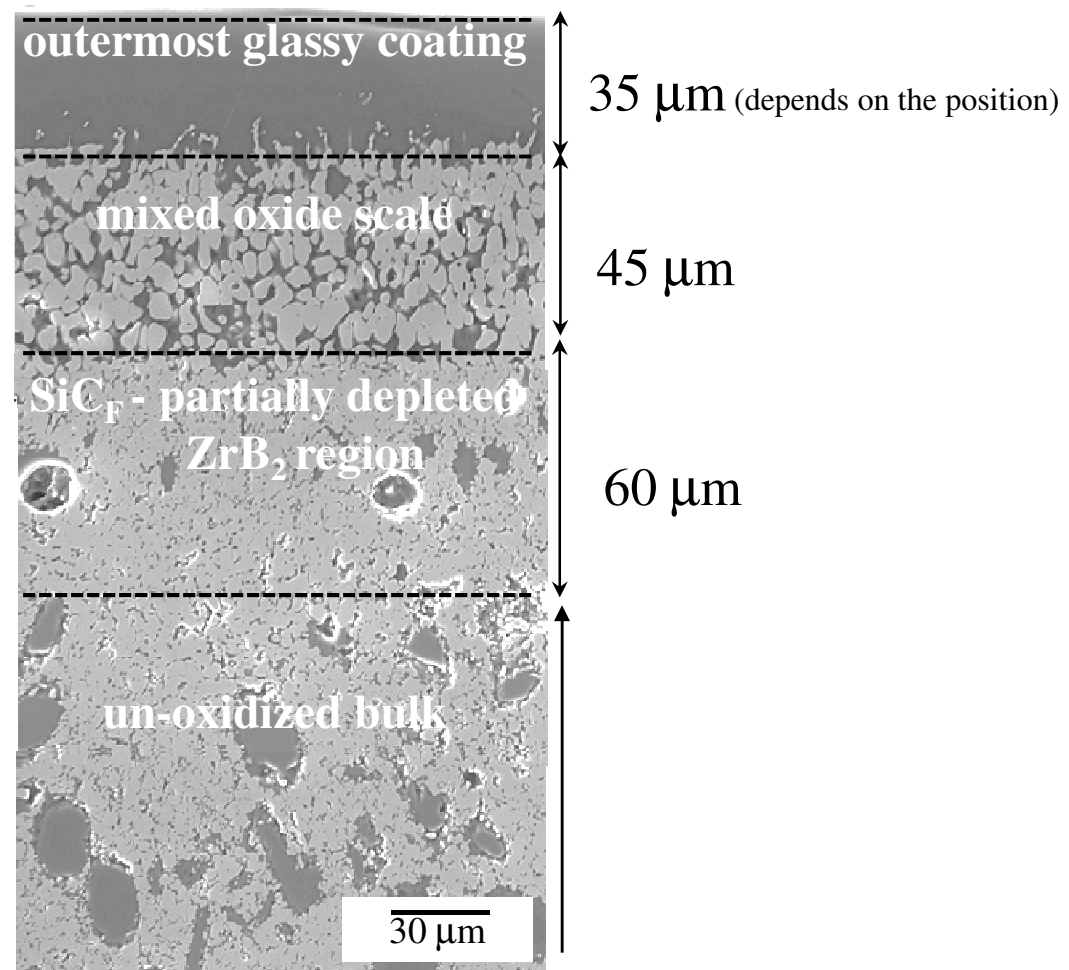
Iso-thermal run for 30 min at 1700°C



Polished cross-section: details of the external oxide scale
(SEM micrograph)

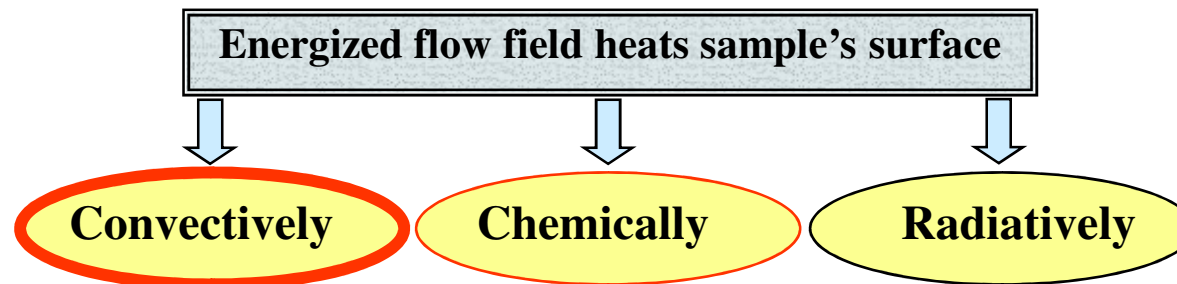
Resistance to oxidation in static air

Iso-thermal run for 30 min at 1700°C



Arc-jet testing

- Simulate re-entry conditions in a ground-based facility
- Heat a test gas (air) to plasma temperature by an electric arc, then accelerate into a pressure-controlled chamber, and onto a stationary test article



In a re-entry environment:

- Oxygen and Nitrogen may be dissociated, thus their recombination adds to surface heating
- stagnation pressure may be below 1 atm, thus active oxidation of SiC plays an important role

Arc-jet testing in high enthalpy supersonic flows



Small Planetary Entry Simulator

**Dept. of Aerospace
Engineering - University of
Naples “Federico II”, Italy**

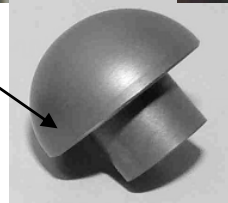
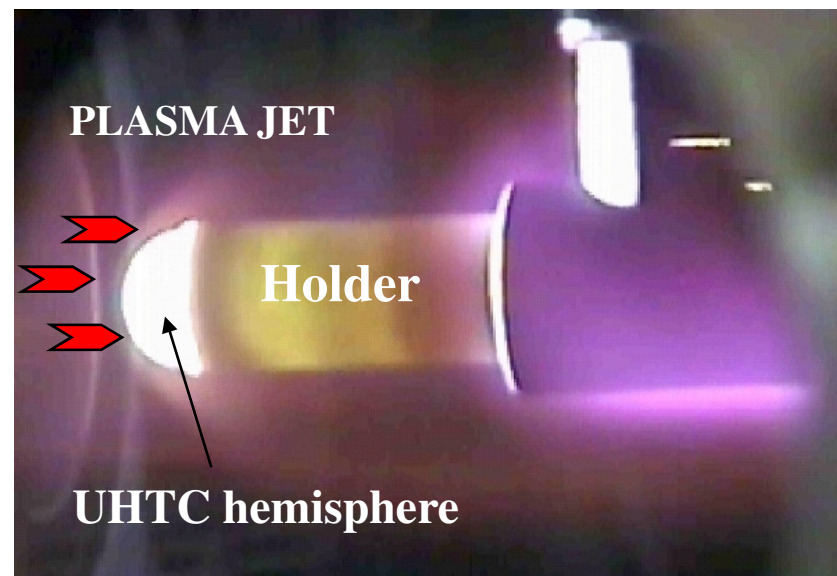
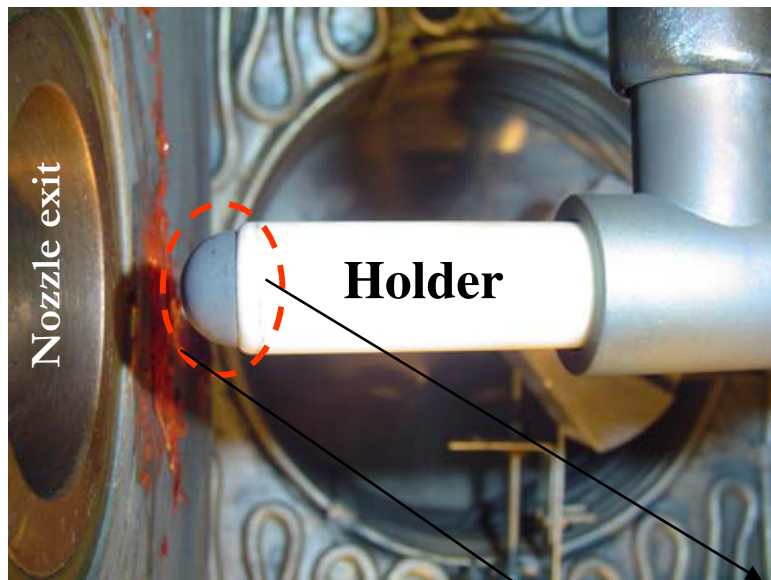
Main Technical Specifications

- Mean free stream total enthalpy (MJ/kg) 3 - 35
- Stagnation-point pressure (Pa) 500 – 10000
- Mach number up to 5
- Mass flow rate (g/s) 0.5 - 5
- Gas mixtures **air**, Ar, N₂, CO₂
- Nozzle exit diameter (mm) 22, 60
- Model size (mm) up to 40

Main Measurement Techniques

- Total enthalpy **energy balance**
- Stagnation-point pressure water-cooled Pitot probe
- Stagnation-point heat flux Gardon gauges and slug calorimeters
- Pressures precision vacuum transducers
- Mass flowthermal mass flowmeters
- Surface temperatures **IR thermography, pyrometer**

Arc-jet testing_setup



Two UHTC hemispheres (nominal $R=5$ mm) were machined

- $M \approx 3$ (supersonic regime)
- 1 g/s of 80% N_2 + 20% O_2 (air)
- static chamber pressure ≈ 200 Pa
- maximum stagnation point pressure 8.5-10 kPa

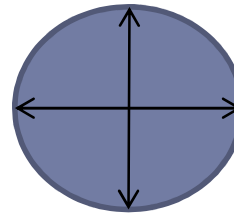
Two arc-jet tests were executed

1st test: max $H_0=13$ MJ/kg for 350 s

2nd test: max $H_0=17$ MJ/kg for 270 s

Arc-jet testing_setup

- After conventional machining, UHTC hemispheres were cleaned, and weighed before and after arc-jet testing (precision 10^{-4} g)
- The profile of the curved surface was measured before and after arc-jet test using a contact-stylus profilometer

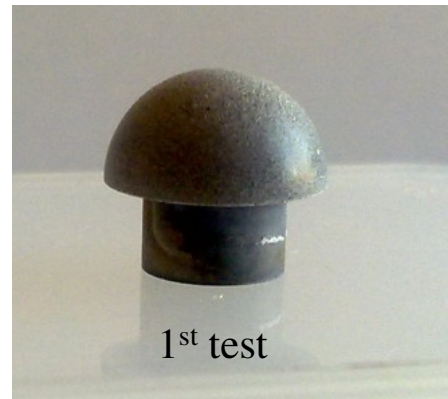


- Best fitted R^* was estimated along two orthogonal directions

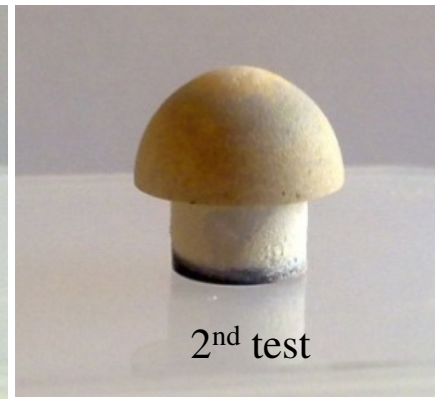
Visual appearance of hemispheres after



As-machined



1st test



2nd test

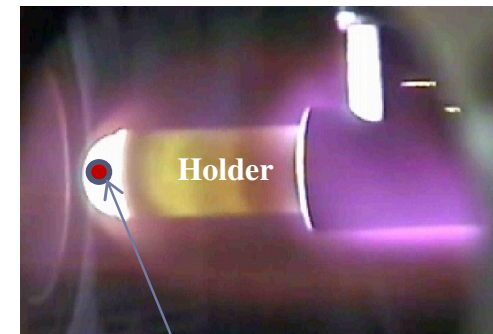
1st arc-jet test_results

- Max H_0 : 13 MJ/kg, duration (t): 350 s
- Equilibrium temperature (by two-color pyrometer)

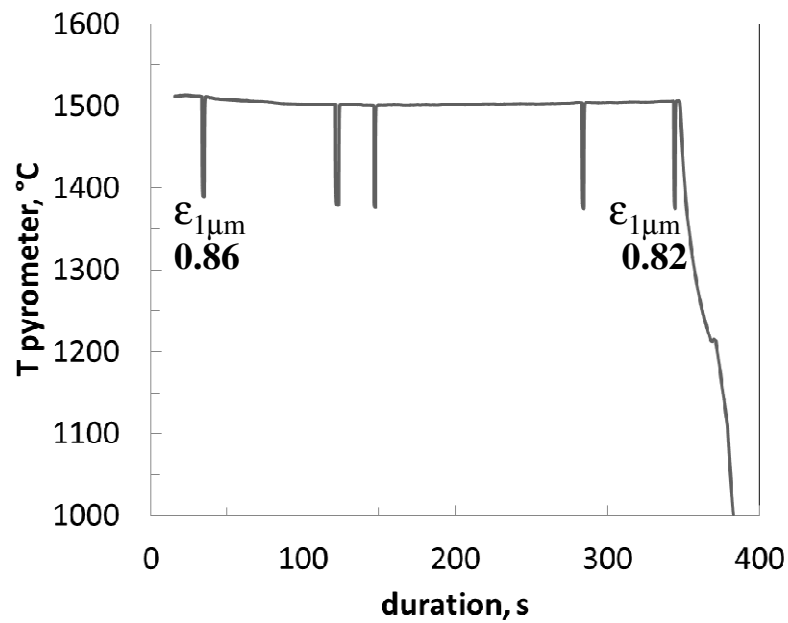
1500 °C at t_{0s} 1505 °C at t_{350s}

- Spectral emissivity at 1 μm ($\epsilon_{1\mu\text{m}}$)

$t_{0s} = 0.86$, $t_{350s} = 0.82$



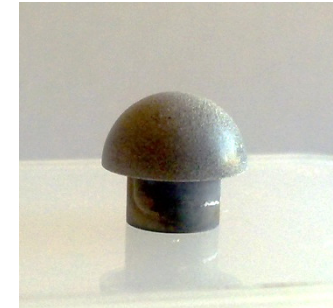
Two-color pyrometer



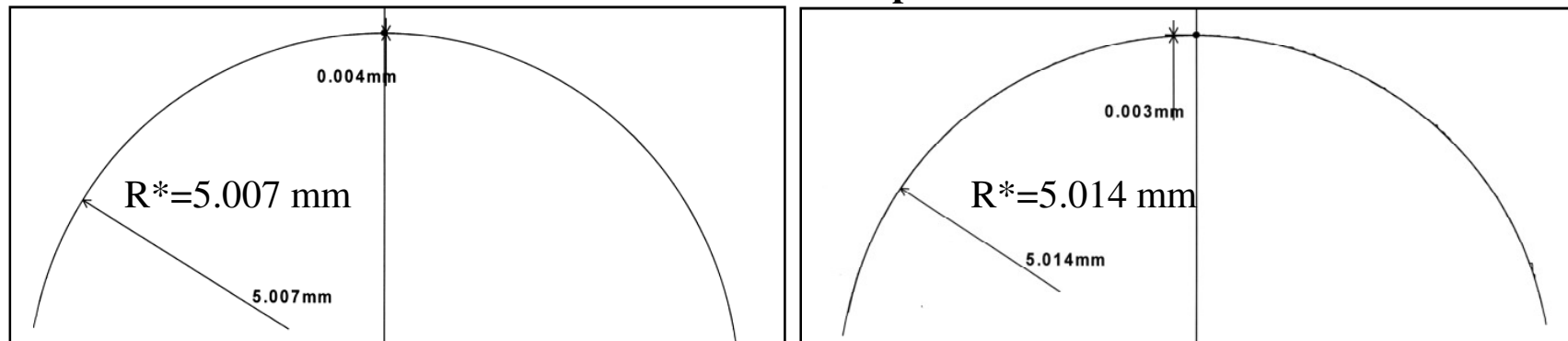
As-cut cross section of the hemisphere, by OM

1st arc-jet test_results

- Mass change ($\Delta m/m_0$): 0.4% (all modifications upon the curved surface)
- R^* (best fitted radius of curvature) change: + 0.007 mm (profilometer)



Profile of the round hemispheric surface

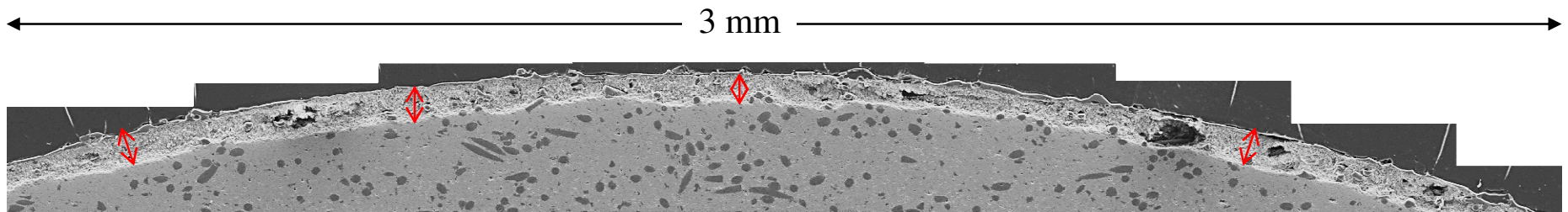


before arc-jet

after arc-jet test

1st arc-jet test_results

- Max H_0 : 13 MJ/kg, duration: 350 s
- Equilibrium temperature range (by two-color pyrometer): 1500 - 1505 °C

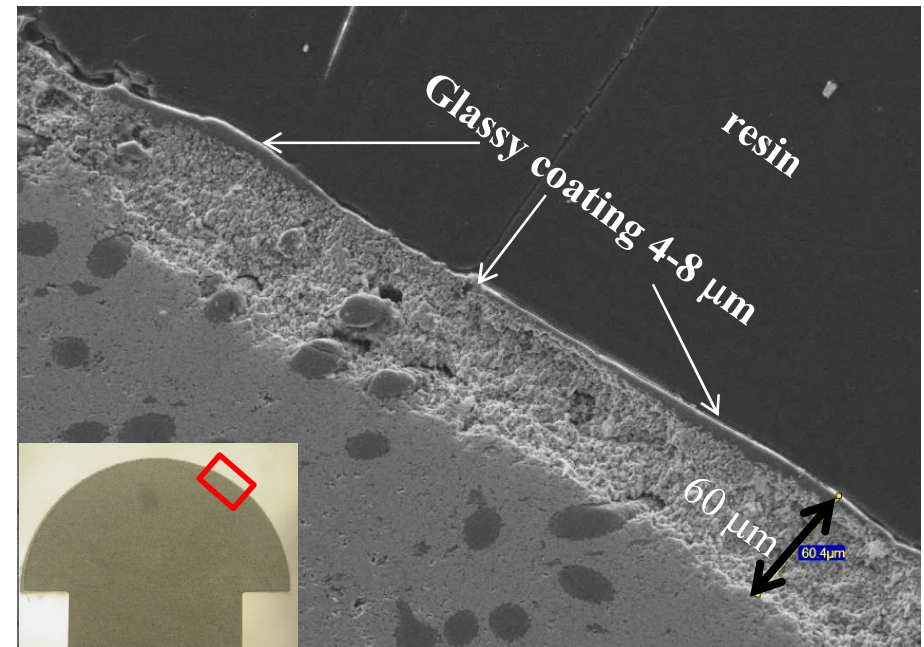
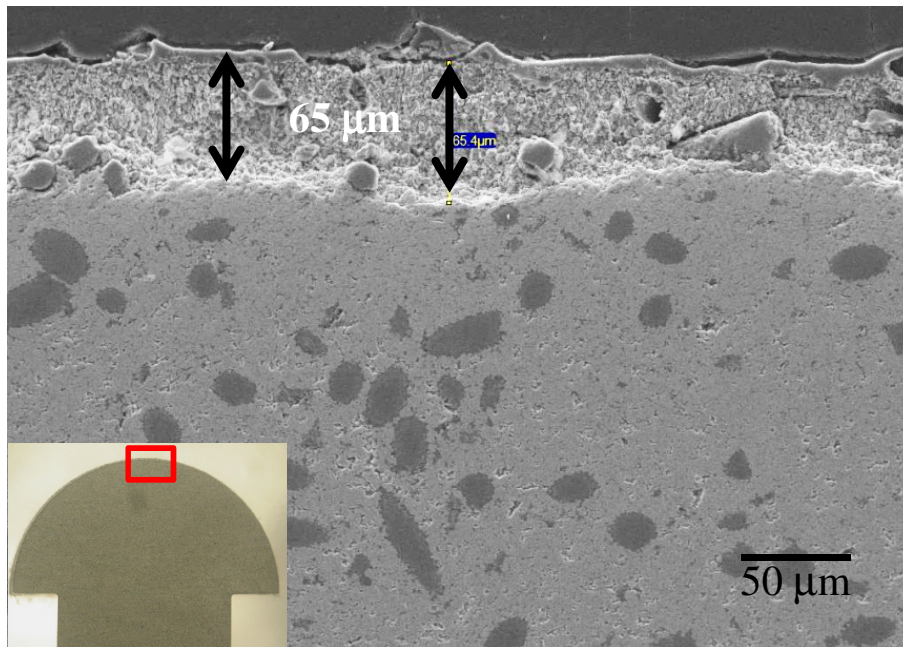


Polished cross section, by SEM, near the stagnation point

oxide scale thickness 60-65 μm

1st arc-jet test_results

- Max H_0 : 13 MJ/kg, duration: 350 s
- Equilibrium temperature range (by two-color pyrometer): 1500 - 1505 °C

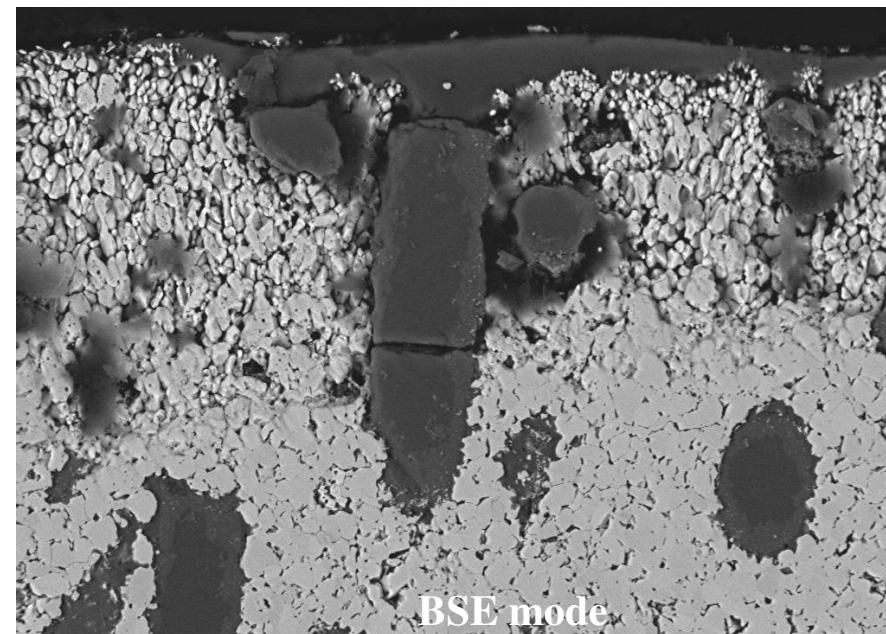
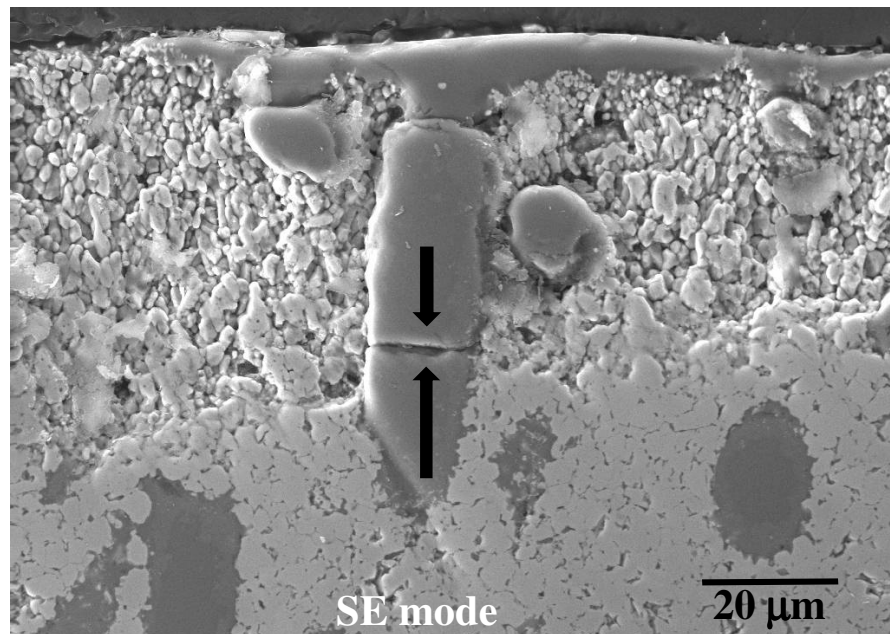


Polished cross section, by SEM: the external oxide layers

thickness of zirconia scale 60-65 μm ; thickness of glassy coating 4-8 μm

1st arc-jet test_results

- Max H_0 : 13 MJ/kg, duration: 350 s
- Equilibrium temperature range (by two-color pyrometer): 1500 - 1505 °C

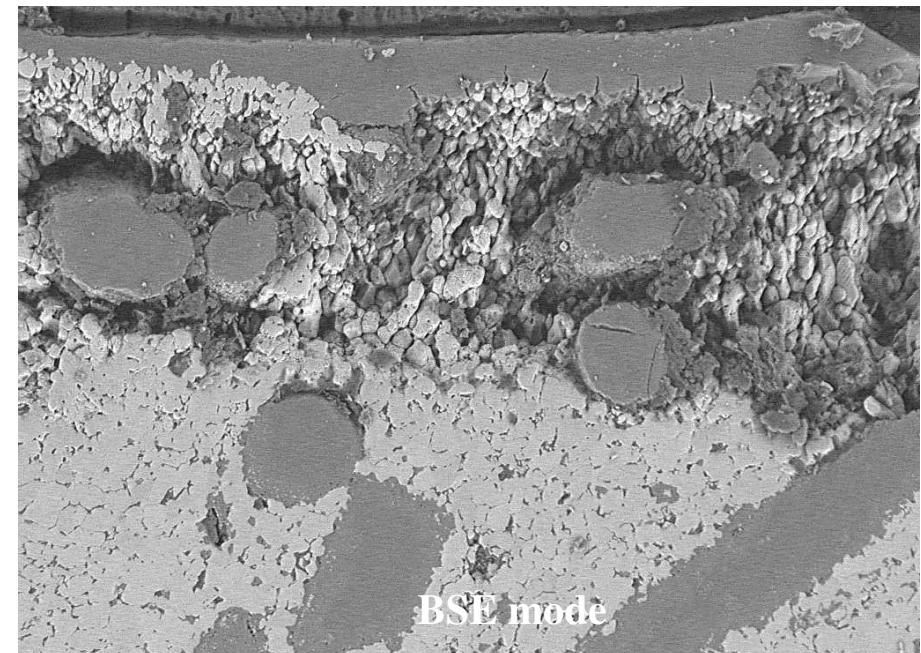
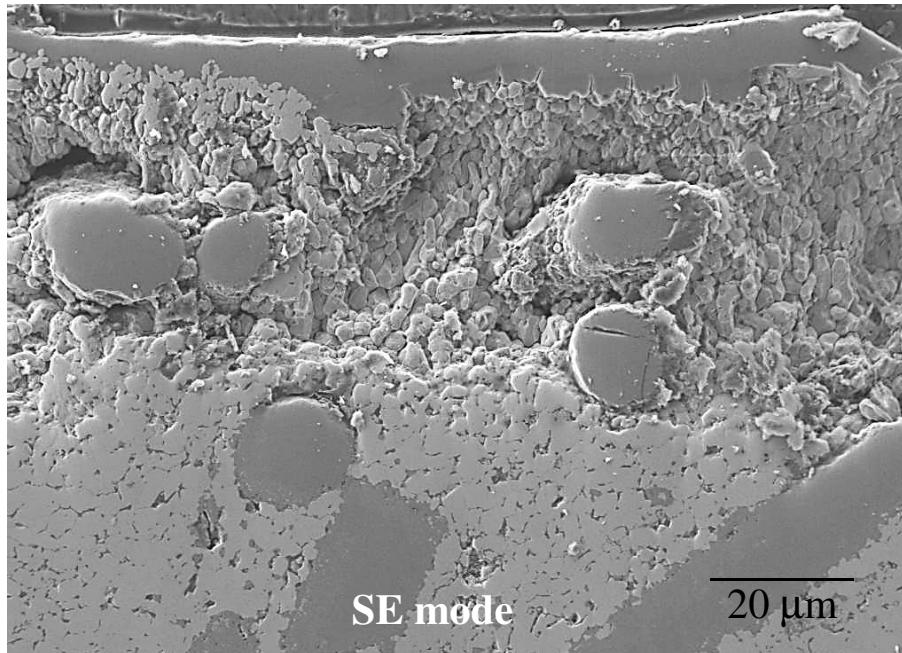


Polished cross section, by SEM: details on the fibers inside the oxide scale

SE: secondary electron, BSE: backscattered electron

1st arc-jet test_results

- Max H_0 : 13 MJ/kg, duration: 350 s
- Equilibrium temperature range (by two-color pyrometer): 1500 - 1505 °C



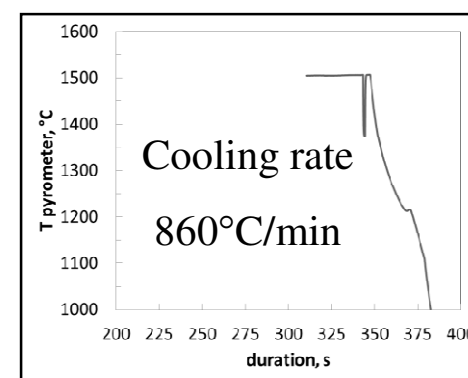
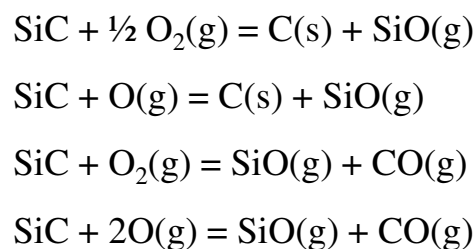
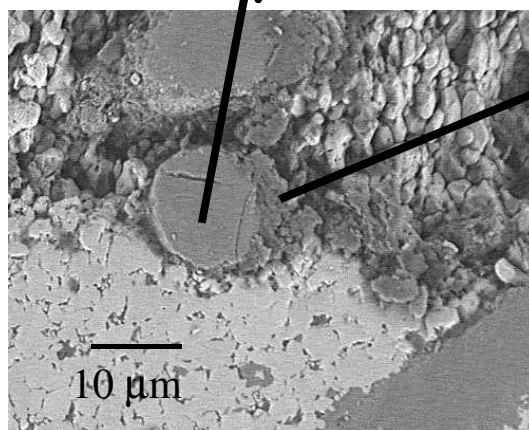
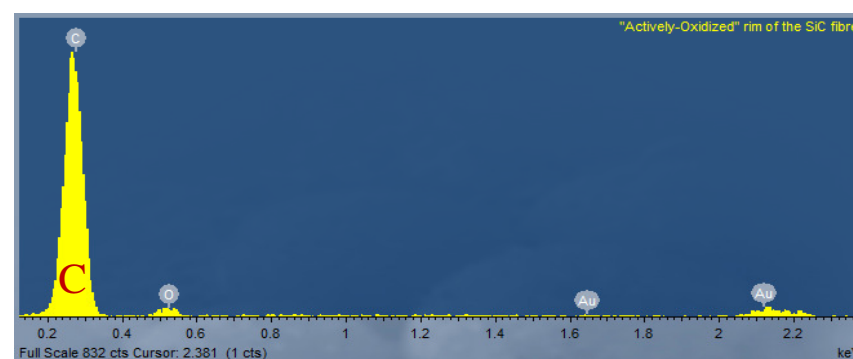
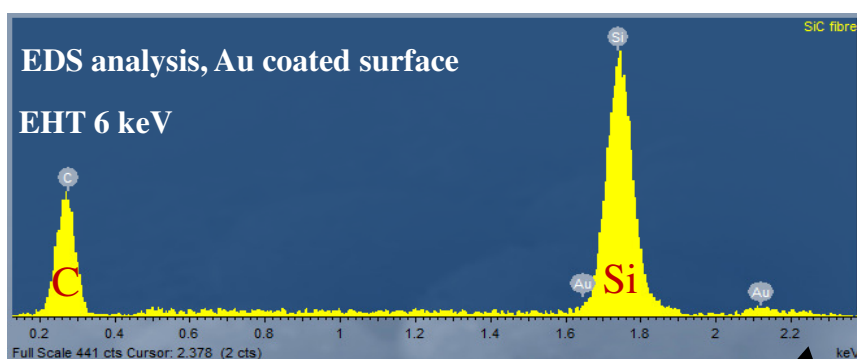
Polished cross section, by SEM: details on the fibers inside the oxide scale

The partially –consumed SiC fibers via mechanism of active oxidation

SE: secondary electron, BSE: backscattered electron

1st arc-jet test_results

- Max H_0 : 13 MJ/kg, duration: 350 s
- Equilibrium temperature range (by two-color pyrometer): 1500 - 1505 °C



Detail of a partially oxidized SiC fiber, by BSE-SEM

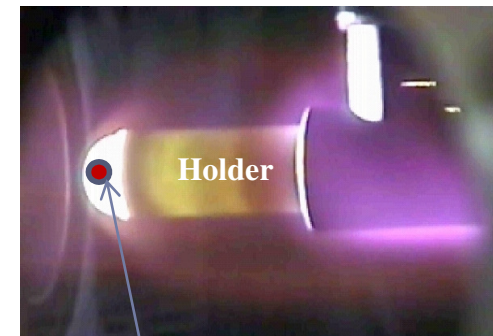
2nd arc-jet test_results

- Max H_0 : 17 MJ/kg, duration: 270 s
- **Equilibrium temperature (by two-color pyrometer)**

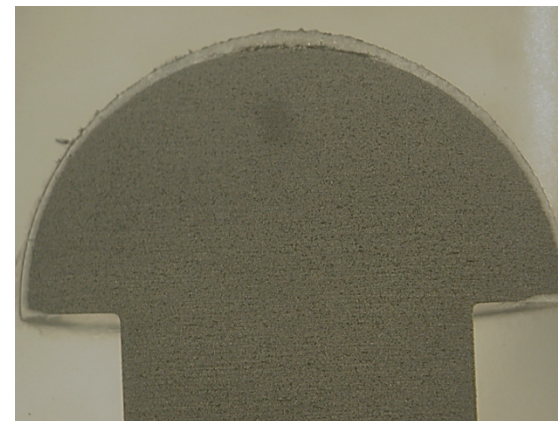
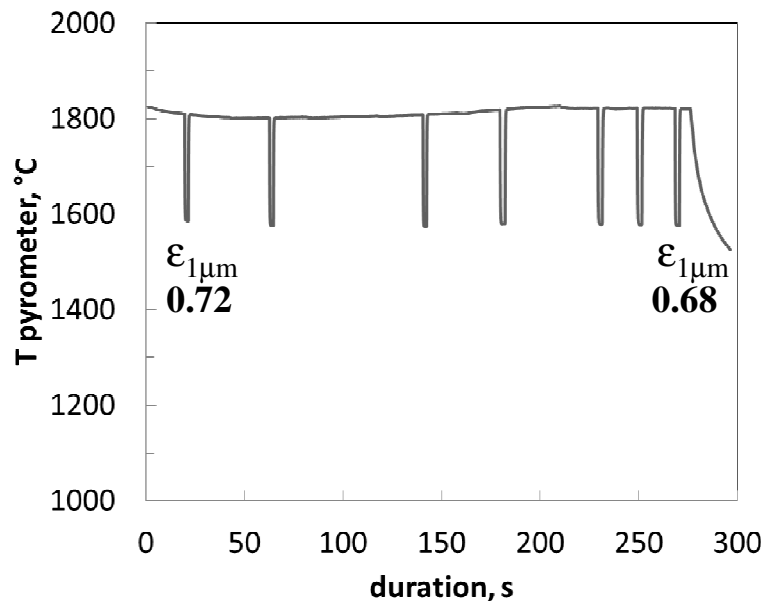
$$t_{0-360} = (1800-1820) ^\circ\text{C}$$

- **Spectral emissivity at 1 μm**

$$t_0 = 0.72, t_{270s} = 0.68$$



Two-color pyrometer

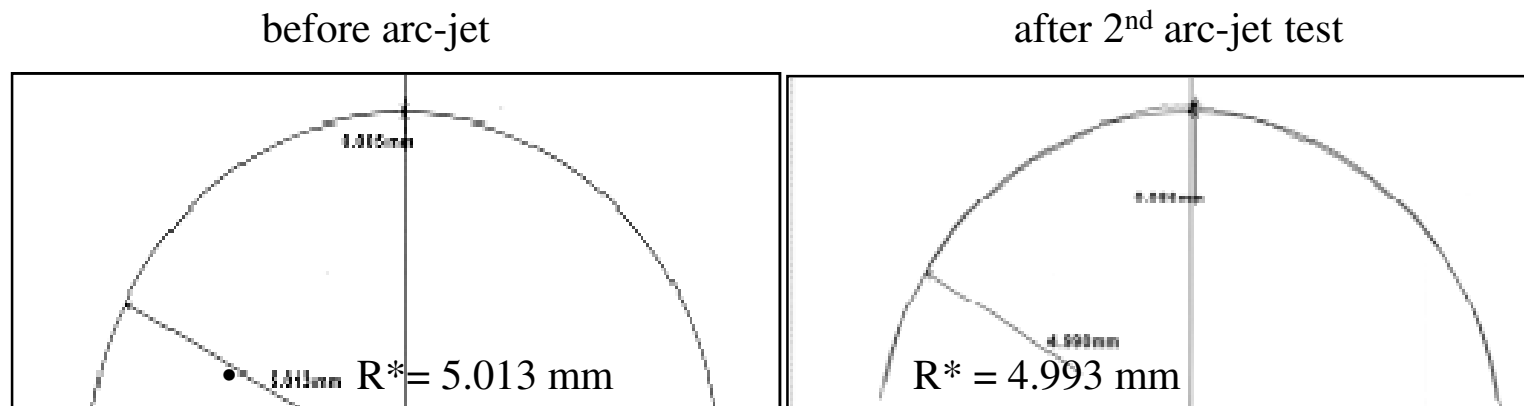


As-cut cross section of the hemisphere, by OM

2nd arc-jet test_results



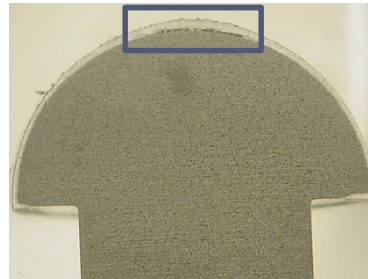
- Mass change ($\Delta m/m_0$): + 0.145 % (all modifications upon the curved surface)
- R^* (best fitted radius of curvature) change: - 0.02 mm (profilometer)



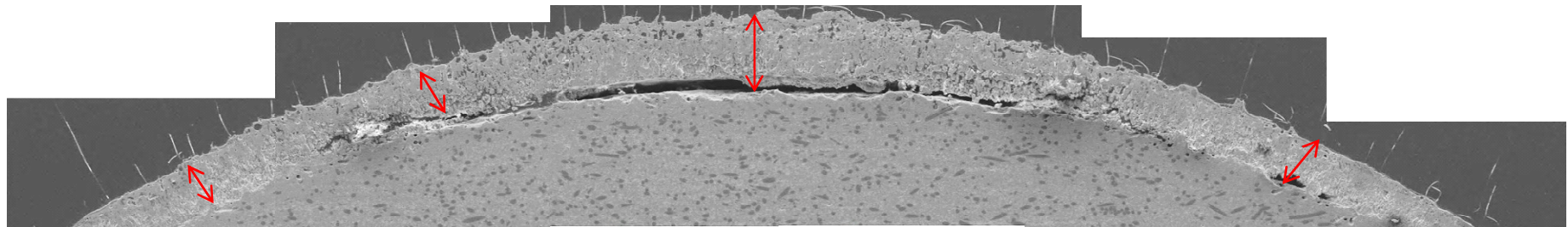
Profile of the round hemispheric surface

2nd arc-jet test_results

- Max H_0 : 17 MJ/kg, duration: 270 s
- **Equilibrium temperature range (by two-color pyrometer): 1800-1820 °C**



6 mm

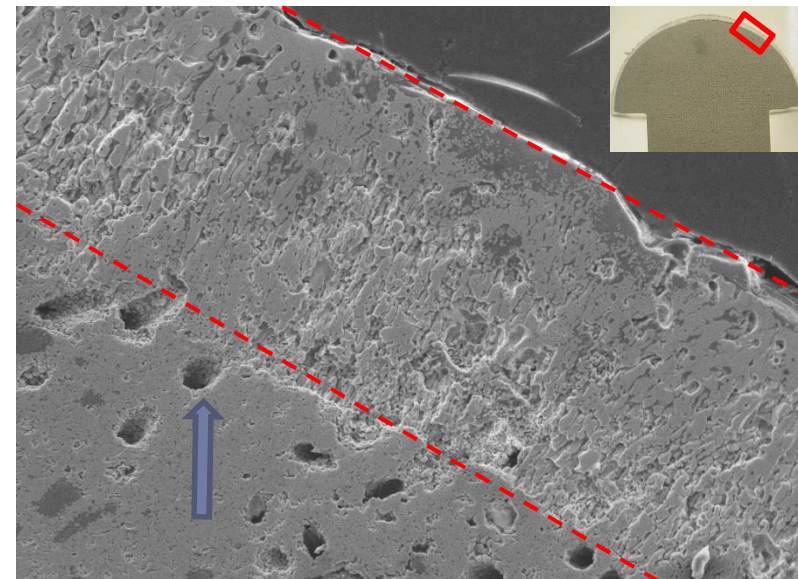
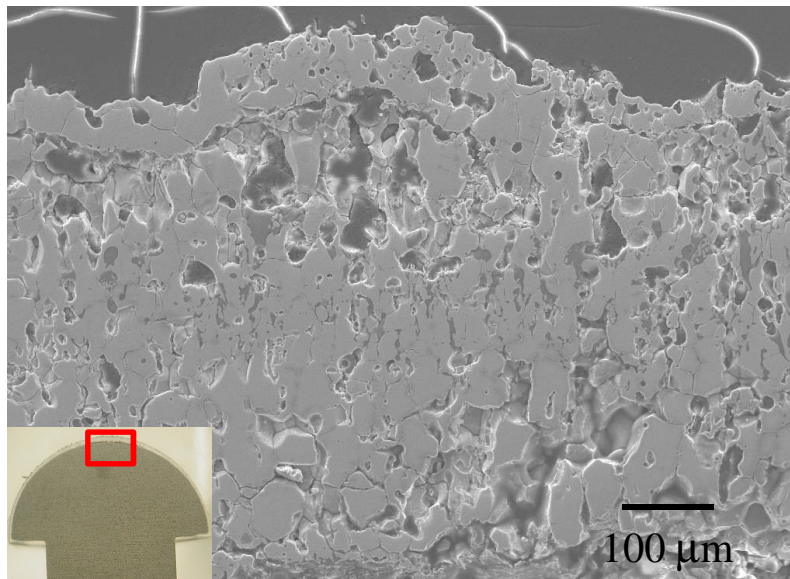
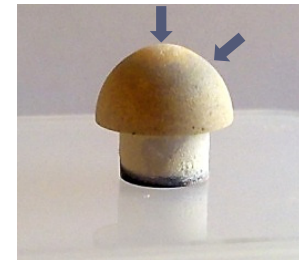


Polished cross section, by SEM, near the stagnation point

oxide scale thickness 280-170 μm ; no external glassy coating was observed

2nd arc-jet test_results

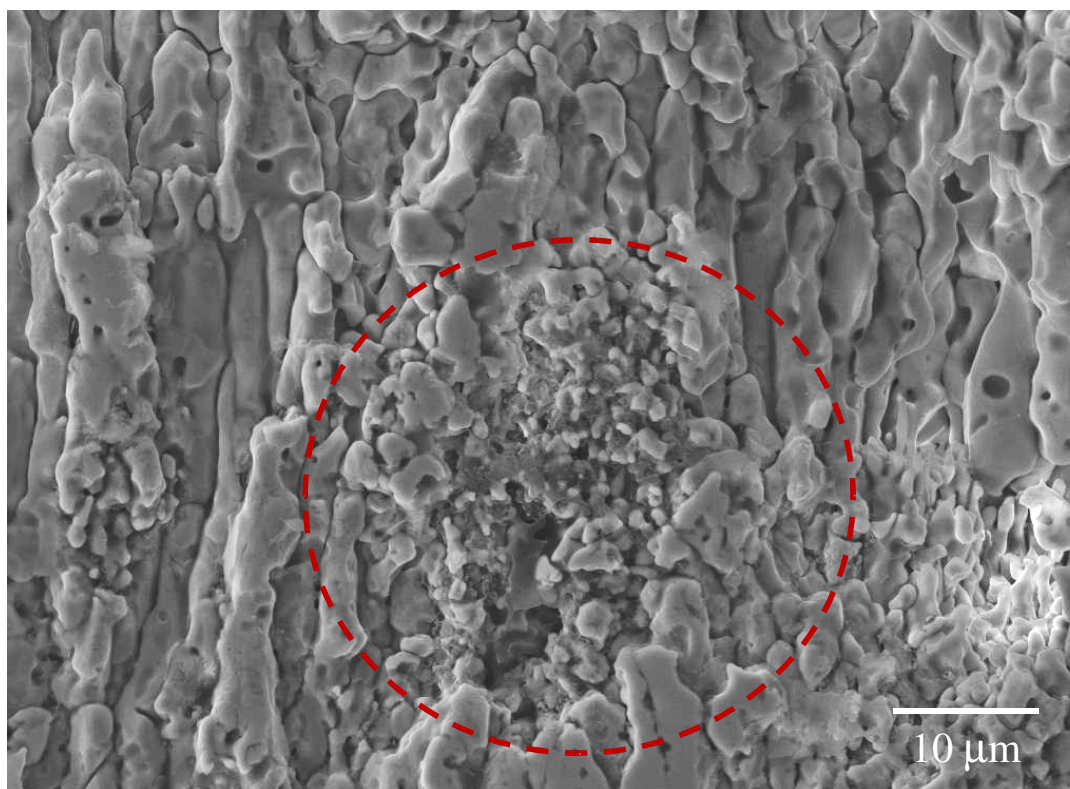
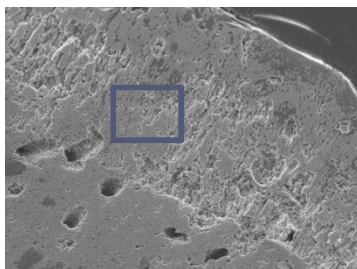
- Max H_0 : 17 MJ/kg, duration: 270 s
- **Equilibrium temperature range (by two-color pyrometer): 1800-1820 °C**



Zoomed views near (left) and far (right) from the stagnation point
(polished cross section, by SE-SEM)

2nd arc-jet test_results

- Max H_0 : 17 MJ/kg, duration: 270 s
- **Equilibrium temperature range (by two-color pyrometer): 1800-1820 °C**

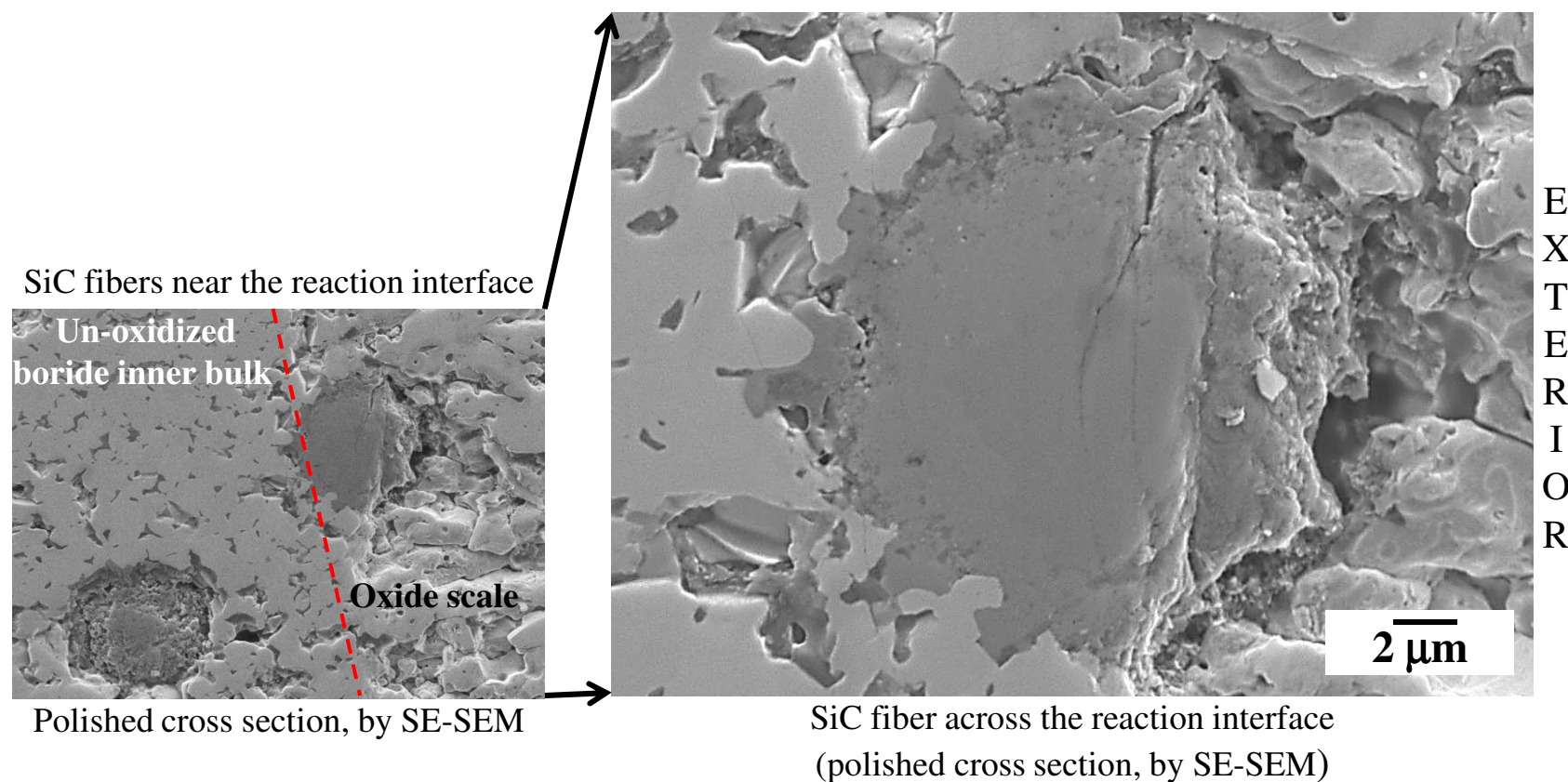


Residue of the SiC fiber inside the zirconia scale
(polished cross section, by SE-SEM)

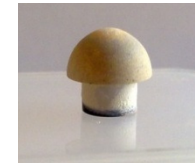
2nd arc-jet test_results



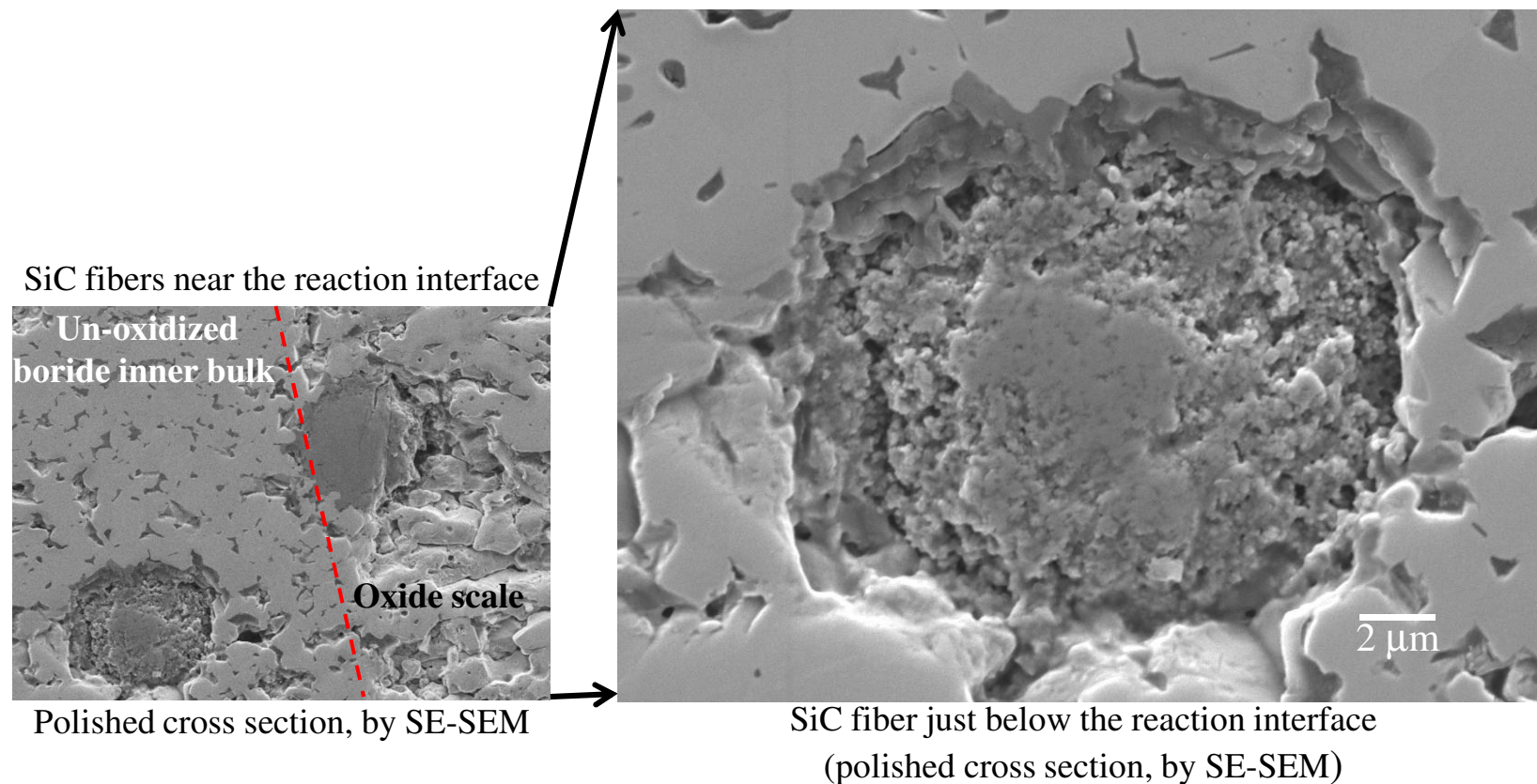
- Max H_0 : 17 MJ/kg, duration: 270 s
- **Equilibrium temperature range (by two-color pyrometer): 1800-1820 °C**



2nd arc-jet test_results

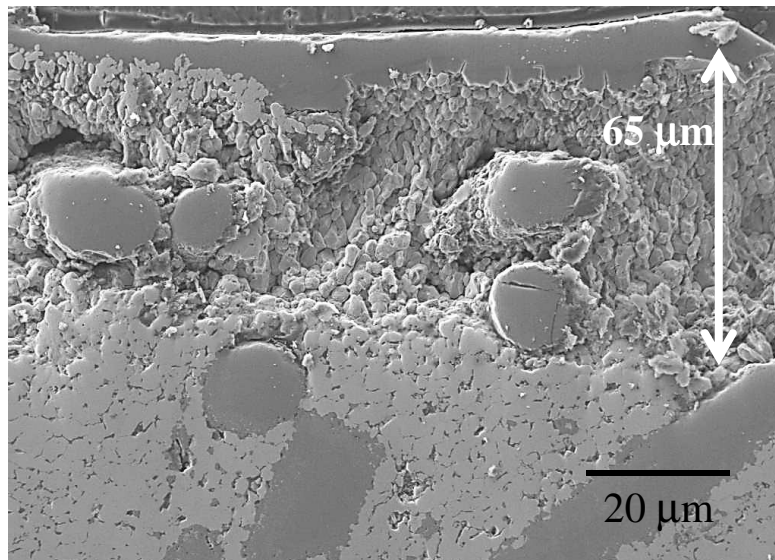


- Max H_0 : 17 MJ/kg, duration: 270 s
- **Equilibrium temperature range (by two-color pyrometer): 1800-1820 °C**

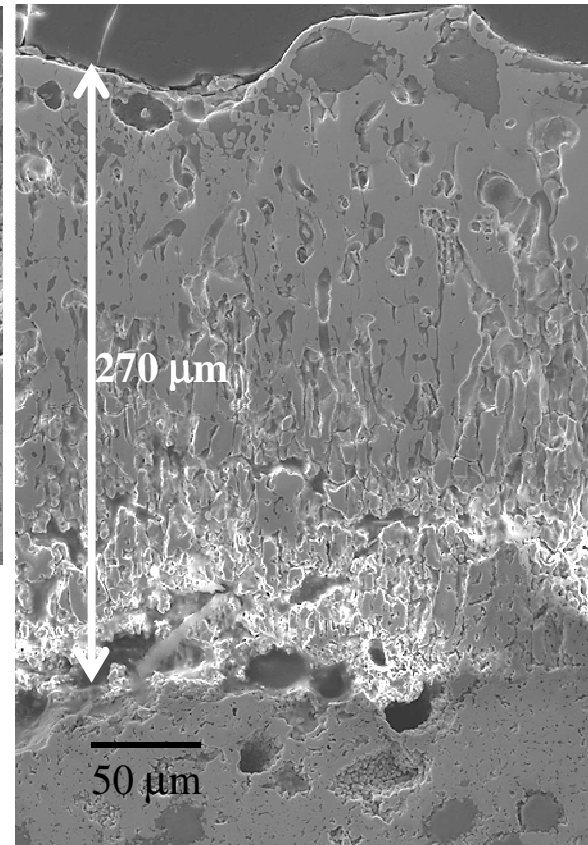


Arc-jet test_comparison

1st test



2nd test



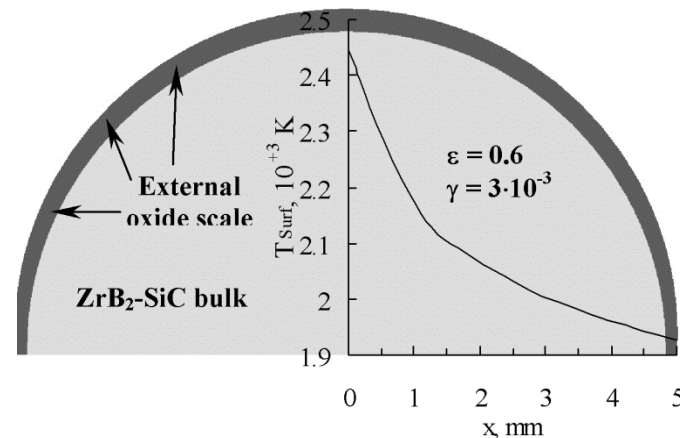
Comparison of the external oxide scale (by SE-SEM) near the stagnation point

Summary

- ZrB_2 -15% SiC_F were fully densified by hot pressing and characterized
- In simulated aero-thermal heating the material withstood rather well the severe environment
- The chopped SiC_F behaved differently depending on the temperature regimes: in all cases provided a significant benefit to the overall resistance to oxidation

Work in progress

- Use more stable SiC fibers
- Optimize the max SiC fiber length
- CFD modeling to calculate equilibrium temperature map upon the surface (and inside the bulk)



F. Monteverde, R. Savino, Dynamic oxidation of ultra-high temperature $\text{ZrB}_2\text{-SiC}$ under high enthalpy supersonic flows, *Corrosion Science* **53** (2011) 922