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## Numerical and Experimental Results of a Passive Free Yawing Downwind Wind Turbine

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# Numerical and Experimental Results of a Passive Free Yawing Downwind Wind Turbine

David R.S. Verelst



**DTU Wind Energy** Department of Wind Energy

## The WINDFLOWER project

- EU Marie Curie Industry Academia Partnerships and Pathways (IAPP) co-funded PhD project consisting out of the following consortium:
  - o 3E/XANT (Brussels, renewable energy consultant)
  - DTU Wind Energy (formerly known as Risø)
  - TU Delft (Netherlands)
- Focus of the PhD research:
  - o Numerical investigation of the feasibility of the free yawing downwind concept
  - Wind tunnel tests at the TU Delft Open Jet Facility (OJF):
    - Comparing different degrees of blade flexibility
    - Free yawing, downwind turbine
  - o Comparison HAWC2 simulations with wind tunnel tests
  - Close link with industry
- This PhD project contributed to:
  - PhD thesis and presentation
  - Patent application (3E/XANT), wind turbine in development
  - $\circ\,$  Three conference papers on free yawing and the wind tunnel experiments
  - o Technical Risø report on blade sweep for the NREL 5MW reference turbine
  - Journal publication as co-author on extreme load extrapolation techniques

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- TU Delft OJF support: Jan-Willem van Wingerden, Roeland De Breuker, Kees Slinkman, Hans Weerheim
- Colleagues from AED
- The Office
- Friends and family

#### **Presentation Overview**

- Numerical studies: static and free yaw stability for a 140kW wind turbine
- Small 300 Watt experimental wind turbine:
  - o Wind Tunnel experiments: design, production, measurement techniques
  - Measurements and results
  - o Simulation input data: a numerical representation of the experiment in HAWC2
  - o Comparing numerical and experimental results
- Conclusions and future work

## The Basics



## **Numerical Studies with HAWC2**

- Coupled aerodynamic-structural time domain wind turbine simulation code
- Structure:
  - Multi-body formulation
  - Flexible bodies with Timoshenko beam elements
  - o Orthotropic material properties: no structural couplings
- Aerodynamics:
  - o Blade Element Momentum theory
  - Tip correction: Prandtl
  - Dynamic stall: Beddoes-Leishman
  - Dynamic inflow
  - $\circ\,$  Skewed and sheared inflow corrections

#### **Numerical Studies: Baseline Design**

Configuration	3 blades, downwind, stall controlled	Cut in, cut out wind speeds	3–25 m/s
Rated power	140 kW	Rated wind speed	12 m/s
Blade length	10 m	Hub radius	0.5 m
Tower height	30 m	Rated rotor speed	57 RPM



## Yaw Moments and Rotor Coning



- Static yaw stability
- Sheared inflow conditions



## Varying Rotor Configurations in Free Yaw

- A practical and applied approach
- Standard: straight blade, no coning angle
- Coned: straight blade with a  $10^\circ$  coning angle (coned downwind)
- Swept: swept blade, no coning angle
- Swept and coned: swept blade with a  $10^\circ$  coning angle (coned downwind)
- Uniform and standard sheared wind profiles

• Blade sweep curve: 
$$x = a \left(\frac{z-z_0}{z_e-z_0}\right)^b$$

• Evaluate both static and dynamic yaw stability

#### **Free Yaw Response**



#### **Free Yaw Response**



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#### Free Yaw Response Overview



#### Wind Tunnel Experiments





## The TU Delft Open Jet Facility

- Wind speeds: 3 35 m/s (wind force 11, 70 knots)
- 500 kW fan
- 2.8m by 2.8m exit nozzle



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## Scaling: Blade Length from 12m to 1m



- Full scale starting point (XANT)
  - o rated power: 100kW
  - o 24m rotor diameter
  - $\circ\,$  Optimal tip speed ratio (TSR)  $\cong 6$
  - $\circ\,$  Typical Reynolds numbers at optimal TSR  $\cong 0.50e6 1.50e6$
- Scaled down model, very simple scaling rules:
  - $\circ\,$  Rotor diameter  $\leq$  1.8m (wind tunnel size restriction)
  - Maintain TSR, consequently optimal RPM's / wind speeds are:
    - 300 RPM @ 4 m/s
    - 750 RPM @ 10 m/s
  - $\circ\,$  Typical Reynolds number similarity is not maintained  $\simeq 0.10e6 0.15e6$
- High rotor speeds result in significant centrifugal stiffening. Achieving blade flexibility is challenging.

## **Practical Design Constraints**

- Platform: small 300 Watt turbine, designed and assembled in Canada (vpturbines.com)
- Refitted with custom build and in-house designed blades







## **Aerofoil Selection**



- source: University of Illinois Low Speed Aerodynamic test database (UIUC LSATs)
- aerofoil aerodynamic characteristics: uncertainties with measured data

	region	t/c	$Re_{design}$	$Re_{data}$	$C_{L_{max}}$
NREL S823	inboard	21%	4e5	1e5	1.184
NREL S822	outboard	16%	6e5	2e5	1.100



## Aerodynamic Rotor Design with HAWTOPT



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## Aerodynamic Rotor Performance (HAWTOPT)



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## **Blade Structural Design**

- Using steady state averaged HAWC2 simulations
- Maximize tip deflection  $\geq 10\%$  of rotor radius. Difficult duo to rotational stiffening
- Basic cross sectional modeller TU Delft
- · Basic failure criteria based on cross sectional area and HAWC2 loads





## **Test Setup Overview**

- Blade tip trajectory (HS camera)
- 3D accelerometer tower top
- Fixed data acquisition dSPACE
- Free yawing (tower base), control with wire
- Limited generator torque control (no active tracking of rotor speed)
- Blades made from injected PVC foam, internal glass fiber stiffener
- Rotor speed measurements
- Tower base strain FA, SS
- Blade strain (flapwise), wireless transmitted
- Yaw angle (laser)





#### **Tower Support Structure**





## Free Yaw: Locking and Range Limits





## Yaw Bearing and Generator Load





#### Blades made from injected PVC foam





#### **Rotor Speed on Extended Shaft**





## **Tower Strain Gauges**





#### Wireless Blade Strain Transmitter





## Yaw Angle with Laser Distance Meter





#### **Measurements and Results**





## **High Speed Camera Data Processing**





## **High Speed Camera Processing Results**



 Trailing Edge (TE) coordinates function of position on the lens: perspective deformation.

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- February, April results: camera positions slightly different, other lenses and lighting conditions
- Results used to establish coning imbalances, and blade pitch angles

## Synchronizing dSPACE and Wireless Strain



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## Synchronizing dSPACE and Wireless Strain



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## Trailer Time: Tower Eigenfrequency Passage



#### **Trailer Time: Free Yaw Stability**



#### **Simulation Input Data**



## Model Properties, System Identification

- Create a numerical model that corresponds to the experimental setup
- Blade structural properties (static, non rotating):
  - $\circ\,$  complex varying cross section geometry
  - $\circ\,$  optimize stiffness distribution to match measured static deflection curves
  - $\circ\,$  optimize mass distribution to match measured center of gravity, and eigenfrequency
  - o optimize damping to match measured frequency response decay tests
- Tower structural properties (static, non rotating):
  - $\circ\,$  simple tubular constant cross section geometry
  - $\circ\,$  stiffness affected by clamping at the yaw bearings
  - o optimize stiffness to match measured eigenfrequency
  - $\circ\,$  optimize damping to match measured frequency response decay tests
- Nacelle and hub are assumed stiff compared to the tower and blades
- Blades stiff in torsion: no measurable blade tip twist deformations (rotating, HS camera)
- Yaw bearing friction not measured, but very low
- Lacking: accurate generator torque-rpm curve, no torque measurements

## **Linear Generator Model**



#### **Simulations vs Measurements**





#### **Rotor Thrust Coefficients**





(b) Yawed flow. Triangles refer to measurements, crosses to simulations. Dotted lines are proportional to  $cos^2\psi$ .

#### Flapwise Blade Root Moment (high RPM)



#### Flapwise Blade Root Moment (low RPM)



#### Free Yaw Response: Deep Stall





#### Free Yaw Response: Deep Stall



#### Free Yaw Response: Optimal TSR





#### Free Yaw Response: Optimal TSR



## Conclusions

- Numerical studies for a 100 kW wind turbine:
  - $\circ\,$  Unstable in free yaw when blade close to maximum lift point
  - Combining blade sweep and rotor coning angle minimizes unstable operating points
- Wind tunnel experiments:
  - o High blade flexibility failed due to centrifugal stiffening
  - Hardware and sensor limitations (generator and control, torque measurements, synchronisation)
  - Documentation
  - Verified free yaw stability
  - Unstable regions not reached due to limited generator control
  - Recorded azimuthal blade load dependency for various inflow angles
- Simulations vs experiments:
  - $\circ\,$  Matching thrust coefficients for varying inflow angles
  - Data synchronisation issues
  - Similar trends for blade load azimuthal dependency for varying inflow angles
  - Comparable free yaw dynamics
  - o Difference in steady state free yaw angle while operating

## **Future Work**

- Detailed aerodynamic assessment of yaw moment contributions from different radial stations along the blade, and under varying operating conditions
- Blade design: formulate strategy which includes free yawing behaviour
- Yaw moment sensitivity to aerodynamic profile coefficient data, and modelling (3D stall delay)
- More data remains to be analysed/compared with simulations:
  - $\circ\,$  Improving high speed data footage analysis
  - Synchronization issues
  - o More accurate generator model, better torque estimates?
  - o Other blade configurations (sweep, coning)
- Follow up experiment:
  - Use practical experience gained to improve the experiment (measurements techniques and test definitions)
  - Use more extensive and robust/redundant system identification strategies
  - Sufficient torque control to test unstable free yawing conditions
  - o Design and built a truly flexible blade
  - Focus on yawed flow
  - o Influence of wind shear





Thank you for your attention.