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**Prediction of fracture loads in PMMA specimens using the Equivalent Material Concept and the Theory of Critical Distances combined criterion**

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## Prediction of fracture loads in PMMA specimens using the Equivalent Material Concept and the Theory of Critical Distances combined criterion

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### Abstract

This paper provides a methodology for the prediction of fracture loads in notched materials that combines the Equivalent Material Concept with the Theory of Critical Distances. The latter has a linear-elastic nature, and requires material (critical distance) calibration in those cases where the non-linear material behaviour is significant. The calibration may be performed by fracture testing on notched specimens, finite elements modelling or a combination of fracture and simulation. In any case, it may constitute a major issue when applying the Theory of Critical Distances on an industrial level. The proposed methodology sets out to define an equivalent linear-elastic material on which the Theory of Critical Distances may be applied through its basic formulation and without any previous calibration of the corresponding critical distance. It has been applied to PMMA Single Edge Notch Bending specimens, providing accurate predictions of fracture loads.

**Keywords:** notch, fracture load, Polymethyl-methacrylate

### 1. Introduction

The analysis of fracture processes on materials and structural components containing notches is the subject of an extensive pool of research work<sup>1-41</sup>. Understanding notches as any kind of macroscopic stress risers in the material, these may be responsible for structural failures caused by static fracture-plastic collapse processes, or the initiators of fatigue processes which may cause a crack to initiate, propagate, and eventually lead to failure. In other words, there are many practical situations where the defects responsible for structural failures are not necessarily crack-like defects. In such cases, if the defects are blunt, it is generally over-conservative to proceed on the assumption that the defects

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3 behave like sharp cracks, given that notched components develop a load-bearing  
4 capacity that is greater than that developed by cracked components.  
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7 Consequently, the particular nature of notches makes it necessary to develop specific  
8 approaches for the fracture analysis of this type of defects. In this sense, the analysis of  
9 the fracture behaviour of notches can be performed using different criteria, some of  
10 these being related to each other. Some examples are the different methodologies  
11 included within the Theory of Critical Distances (TCD)<sup>1-8</sup>, the Global Criterion<sup>9,10</sup>,  
12 Cohesive Zone models<sup>11-15</sup>, statistical models<sup>16-17</sup>, mechanistic models<sup>18</sup>, the Strain  
13 Energy Density (SED) criterion<sup>19-36</sup>, etc. The TCD methodologies have been  
14 successfully applied to different failure mechanisms (e.g., fracture, fatigue) and  
15 materials, and are particularly simple to implement in structural integrity assessments<sup>7,</sup>  
16 <sup>37-41</sup>. The TCD is based on linear-elastic assumptions, although it has been successfully  
17 applied to elastic-plastic situations, either through the direct consideration of elastic-  
18 plastic stress fields<sup>2</sup>, or through the assumption of linear-elastic behaviour (stress field)  
19 and the corresponding calibration of the inherent strength (see section 2)<sup>4,5</sup>. In any case,  
20 when the material behaviour is not completely linear-elastic, the application of the TCD  
21 requires the fracture testing of notched specimens, finite elements modelling, or both, in  
22 order to calibrate the material parameters involved (the critical distance -L-, and the  
23 inherent strength,  $\sigma_0$ ). This complicates the application of the TCD on an industrial  
24 level.  
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30 At the same time, when analysing an elastic-plastic material, Torabi<sup>42,43</sup> has proposed  
31 the use of the Equivalent Material Concept (EMC) to define an equivalent linear-elastic  
32 material that develops the same fracture behaviour. This proposal has been combined  
33 with the TCD<sup>44-50</sup> or the Strain Energy Density (SED)<sup>51-55</sup>, providing accurate analyses  
34 of the fracture behaviour of different materials, such as Al 6061-T6 and Al 7075-T6.  
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37 This paper analyses the fracture behaviour of Polymethyl-methacrylate (PMMA) Single  
38 Edge Notched Bending (SENB) fracture specimens containing U-notches. The fracture  
39 behaviour of PMMA in notched conditions is well known, and has been previously  
40 analysed through different methodologies<sup>4,5,14,23</sup>, all of them having significant  
41 complexity for common engineering practice. This work provides additional analyses of  
42 the fracture behaviour of this material, and verifies whether or not the straightforward  
43 combination of EMC and TCD (from now on, the EMC-TCD criterion), provides  
44 fracture assessment results with comparable accuracy to that provided by other  
45 methodologies (e.g., TCD, SED criterion, Cohesive Zone models, etc).  
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49 With all this, section 2 provides a theoretical overview of the Equivalent Material  
50 Concept (EMC), the Theory of Critical Distances (TCD) and the EMC-TCD criterion,  
51 section 3 describes the experimental programme, section 4 provides the fracture load  
52 predictions obtained by using the EMC-TCD criterion and the corresponding  
53 discussion, and section 5 gathers the main conclusions.  
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## 2. Theoretical background

### 2.1. The Theory of Critical Distances

The Theory of Critical Distances (TCD) is in essence a set of methodologies, all of which use a material length parameter (the critical distance,  $L$ ) when performing fracture or fatigue assessments<sup>1</sup>. The origin of the TCD is located in the works of Neuber<sup>56</sup> and Peterson<sup>57</sup>, but it has been in the last two decades that this theory has been thoroughly developed for the analysis of different types of materials, failure processes and conditions (e.g., linear-elastic vs. elastoplastic)<sup>1</sup>.

The aforementioned critical distance is generally referred to as  $L$  and its expression, in fracture analyses, is:

$$L = \frac{1}{\pi} \left( \frac{K_c}{\sigma_0} \right)^2 \quad (1)$$

$K_c$  being the material fracture toughness and  $\sigma_0$  being a material strength parameter usually referred to as the inherent strength. This parameter is usually larger than the ultimate tensile strength ( $\sigma_u$ ), in case it requires calibration. Only in certain materials where there is a linear-elastic behaviour at both the micro and the macro scale (e.g., fracture of ceramics) does  $\sigma_0$  coincide with  $\sigma_u$ . In such cases, the application of the TCD does not require calibration, given that  $L$  is directly obtained from equation (1), the material fracture toughness and the material ultimate tensile strength.

Two of the methodologies included within the TCD are especially simple to apply: the Point Method (PM) and the Line Method (LM). Both of them are based on the stress field at the defect tip and, as stated by Taylor<sup>1</sup>, the corresponding predictions are very similar.

The PM is the simplest methodology, and it proposes that fracture takes place when the stress at a distance of  $L/2$  from the defect tip reaches the inherent strength ( $\sigma_0$ ):

$$\sigma \left( \frac{L}{2} \right) = \sigma_0 \quad (2)$$

On the other hand, the LM proposes that fracture takes place when the average stress along a distance equal to  $2L$  (starting from the defect tip) reaches the inherent strength  $\sigma_0$ :

$$\frac{1}{2L} \int_0^{2L} \sigma(r) dr = \sigma_0 \quad (3)$$

The TCD (and therefore, both the PM and the LM) allows the fracture assessment of components containing notches to be performed. However, for those materials on which  $\sigma_0$  does not coincide with  $\sigma_u$  (e.g., most polymers, metals, etc), the former parameter requires calibration. This may be performed by undertaking an experimental programme

on notched specimens with different notch radii, and defining  $L$  as that value providing the best fit to the experimental results<sup>1,5</sup>, by finite elements simulation of specimens with different notch radii (the superposition of the corresponding stress fields at failure directly provides  $L$  and  $\sigma_0$ , see Figure 1)<sup>1,5,6</sup>, or by a combination of experimental programme and finite elements modelling. In any case, the calibration process constitutes a major issue when applying the TCD methodologies and it is a clear obstacle to their extensive application in industrial practice.

## 2.2. The Equivalent Material Concept

In this subsection, the Equivalent Material Concept (EMC) proposed originally by Torabi<sup>42</sup> is presented with the aim of equating a real ductile material with elastic-plastic behaviour to a virtual brittle material with perfectly elastic behaviour. A summary of the concept is presented in the following.

The famous power-law equation indicating the tensile stress-strain relationship in the plastic region can be found in equation (4) in which the parameters  $\sigma$ ,  $\epsilon_p$ ,  $K$ , and  $n$  are the true stress, the true plastic strain, the strain-hardening coefficient, and the strain-hardening exponent, respectively.

$$\sigma = K\epsilon_p^n \quad (4)$$

As seen in figure 2, which is the typical engineering stress-strain curve for a ductile material, the Strain Energy Density (SED) is the area under the curve until the beginning of the necking (peak point). Considering the total SED as the summation of the SEDs in elastic and plastic regions, one can write

$$(SED)_{tot.} = (SED)_e + (SED)_p = \frac{1}{2}\sigma_Y\epsilon_Y + \int_{\epsilon_p^Y}^{\epsilon_p} \sigma d\epsilon_p \quad (5)$$

where  $\sigma_Y$ ,  $\epsilon_Y$ , and  $\epsilon_p^Y$  are the yield strength, the elastic strain at yield point, and the true plastic strain at yield point, respectively.

By substituting equation (4) into equation (5) and considering Hooke's Law ( $\sigma_Y = E\epsilon_Y$ ), we get

$$(SED)_{tot.} = \frac{\sigma_Y^2}{2E} + \int_{\epsilon_p^Y}^{\epsilon_p} K\epsilon_p^n d\epsilon_p = \frac{\sigma_Y^2}{2E} + \frac{K}{n+1}[(\epsilon_p)^{n+1} - (\epsilon_p^Y)^{n+1}] \quad (6)$$

Assuming that the offset yield point is equal to 0.2% (i.e.  $\epsilon_p^Y = 0.002$ ), then

$$(SED)_{tot.} = \frac{\sigma_Y^2}{2E} + \frac{K}{n+1}[(\epsilon_p)^{n+1} - (0.002)^{n+1}] \quad (7)$$

The crack initiation in the ductile material will take place just when the ultimate load is reached. Therefore, the total SED (the area under the curve) should be calculated until

this point, which is called the necking instance. Consequently, the  $\varepsilon_P$  is substituted by  $\varepsilon_{u, True}$  (true plastic strain at maximum load) in the following equation:

$$(SED)_{necking} = \frac{\sigma_Y^2}{2E} + \frac{K}{n+1} [(\varepsilon_{u, True})^{n+1} - (0.002)^{n+1}] \quad (8)$$

A common stress-strain curve for the virtual brittle material is illustrated in Figure 3. As shown in this figure, the total strain energy absorbed until fracture is computed as  $\sigma_f^* \varepsilon_f^* / 2$ , where  $\sigma_f^*$  and  $\varepsilon_f^*$  are the tensile stress and the strain at crack initiation for the virtual brittle material, respectively. Since the main assumption of EMC is to have the same Young modulus and  $K$ -based fracture toughness ( $K_{Ic}$  or  $K_c$ ) for both ductile and virtual brittle materials, one can write

$$(SED)_{EM} = \frac{\sigma_f^{*2}}{2E} \quad (9)$$

where  $E$  is the Young modulus for both the original ductile and the virtual brittle materials.

As mentioned above, the Equivalent Material Concept (EMC) equates a ductile material having valid  $K$ -based fracture toughness and elastic modulus to a virtual brittle material having the same values but with a different tensile strength. Therefore, setting equations 8 and 9 to be equal leads to:

$$\frac{\sigma_f^{*2}}{2E} = \frac{\sigma_Y^2}{2E} + \frac{K}{n+1} [(\varepsilon_{u, True})^{n+1} - (0.002)^{n+1}] \quad (10)$$

Finally, the following equation is proposed by EMC for calculating the  $\sigma_f^*$ :

$$\sigma_f^* = \sqrt{\sigma_Y^2 + \frac{2EK}{n+1} [(\varepsilon_{u, True})^{n+1} - (0.002)^{n+1}]} \quad (11)$$

where  $\varepsilon_{u, True}$  (the true plastic strain at peak point) can be calculated from the  $\varepsilon_u$  (engineering plastic strain) by the following expression:  $\varepsilon_{u, True} = \ln(1 + \varepsilon_u)$ .

The  $\sigma_f^*$  calculated by equation (11) and a valid fracture toughness can be used conveniently in different brittle fracture criteria, e.g. TCD, to predict the crack initiation in ductile components containing a notch.

In the following sections, the experimental programme is presented and the corresponding results are utilized to verify the validity of the EMC-TCD criterion.

### 3. Experimental programme

The experimental programme covers the definition of the stress-strain tensile curve of the material (following ASTM D638<sup>58</sup>), which is necessary for the application of the EMC method, and the fracture tests performed on SENB specimens containing U-shaped notches (see Figure 4). These fracture tests (32 in total) were performed following ASTM D5045<sup>59</sup>, with the notch radii varying between 0 mm (crack-like defect) and 2.5 mm. Details on the experimental procedures are gathered in Cicero et al.<sup>5</sup>.

Figure 5 shows the obtained stress-strain curve (engineering variables) used in this work, revealing a clear non-linear behaviour. The main material parameters are gathered in Table 1,  $E$  being the Young's modulus,  $\sigma_{0.2}$  being the 0.2% proof strength,  $\sigma_u$  the ultimate tensile strength and  $e_{\max}$  the maximum strain. This curve is used in Section 4 to derive  $\sigma_f^*$  and, thus, the tensile behaviour of the equivalent linear-elastic material.

Concerning the fracture tests, a total of eight sets of tests were performed, corresponding to eight different notch radii (from 0 mm up to 2.5 mm), each set being tentatively composed of five tests. The notches were performed by machining, except for those whose notch radius was close to zero, which were generated by sawing a razor blade across an initial notch root. Table 2 gathers the different tests with the corresponding geometries and the resulting fracture loads. Some of the sets do not include the initial five intended tests, given that some of the specimens were incorrectly machined. Details of the experimental procedure and the obtained load-displacement curves may be consulted in Cicero et al.<sup>5</sup>, with some examples of the above being shown in Figure 6.

The results of the fracture tests reveal that there are sets in which there is significant scatter in the fracture loads (e.g., specimens with 0.50 mm radius). It can also be observed that there is an evident loss of linearity in the load-displacement curves obtained in specimens with higher radii, although such losses are noticeably less pronounced than that observed in the tensile test.

Finally, the results obtained in the three cracked specimens have been used to derive the material fracture toughness ( $K_c$ )<sup>59</sup>. The fracture toughness is easily derived from the critical load and both the specimen and crack geometries (SENB specimen):

$$K_c = \left( \frac{P_{\max}}{B \cdot W^{0.5}} \right) \cdot 6 \left( \frac{a}{W} \right)^{0.5} \left( \frac{1.99 - \left( \frac{a}{W} \right) \left( 1 - \frac{a}{W} \right) \left( 2.15 - 3.93 \left( \frac{a}{W} \right) + 2.7 \left( \frac{a}{W} \right)^2 \right)}{\left( 1 + 2 \frac{a}{W} \right) \left( 1 - \frac{a}{W} \right)^{1.5}} \right) \quad (12)$$

The average value of  $K_c$  derived from the three tests is 2.04 MPa·m<sup>1/2</sup> (see Table 1).

#### 4. EMC-TCD fracture load predictions

##### 4.1. Calibration of the Equivalent Material

The tensile curve shown in Figure 5 has been used to define the equivalent linear-elastic material following the Equivalent Material Concept formulation gathered in Subsection 2.2. The equivalent material maintains the same elastic modulus as that observed in the real material (3.40 GPa, see Table 1), but the tensile strength of the equivalent material ( $\sigma_f^*$ ) is 129.4 MPa, which is significantly higher (1.73 times higher) than that observed experimentally. These two parameters (E and  $\sigma_f^*$ ) are sufficient to define the tensile behaviour of the equivalent material, and allow the fracture behaviour of the real (non-linear) material to be determined based on linear-elastic assumptions.

#### 4.2. Derivation of fracture load predictions

Once the material properties of the equivalent linear-elastic material are known, the linear elastic formulation of the TCD can be directly applied. Assuming a perfectly linear-elastic behaviour implies that the value of the critical distance (L) can be directly obtained from equation (1) and considering that the inherent strength ( $\sigma_0$ ) is equal to the tensile strength of the equivalent material ( $\sigma_f^*$ ). Thus, the calibration process required to define L (and  $\sigma_0$ ) in the real material is avoided. In this case, L is 0.079 mm, which is slightly lower than that obtained by Cicero et al.<sup>5</sup> (L = 0.105 mm) through finite elements calibration.

As mentioned above, one of the main purposes of this work is to provide a simple methodology for the assessment of notched components. For this reason, instead of using finite elements modelling to determine the fracture load predictions, the use of well known accurate analytical solutions is proposed. In the case of U-shaped notches, the Creager-Paris solution<sup>60</sup> for the stress field at the notch tip is widely accepted<sup>1</sup>. Creager and Paris state that the stress field ahead of the notch tip is equal to that ahead of the crack tip but displaced a distance equal to  $\rho/2$  along the x-axis:

$$\sigma(r) = \frac{K_I}{\sqrt{\pi}} \frac{2(r + \rho)}{(2r + \rho)^{3/2}} \quad (13)$$

where  $K_I$  is the mode I stress intensity factor in cracked conditions,  $\rho$  is the notch radius and  $r$  is the distance existing from the notch tip to the point being assessed. Equation (13) may be used to derive the estimations of the critical loads through the TCD.

If the PM is considered, the corresponding fracture condition for a particular notch radius ( $\rho$ ) would be:

$$\sigma(L/2) = \frac{K_I}{\sqrt{\pi}} \frac{2(L/2 + \rho)}{(L + \rho)^{3/2}} = \sigma_f^* \quad (14)$$

Thus, equation (14) allows the value of  $K_I$  at fracture to be obtained. Finally, the estimation of the critical load ( $P_{est}^{PM}$ ) is easily derived from:



$$K_I = \left( \frac{P_{est}^{PM}}{B \cdot W^{0.5}} \right) 6 \left( \frac{a}{W} \right)^{0.5} \left( \frac{1.99 - \left( \frac{a}{W} \right) \left( 1 - \frac{a}{W} \right) \left( 2.15 - 3.93 \left( \frac{a}{W} \right) + 2.7 \left( \frac{a}{W} \right)^2 \right)}{\left( 1 + 2 \frac{a}{W} \right) \left( 1 - \frac{a}{W} \right)^{1.5}} \right) \quad (15)$$

If the LM is considered, it is necessary to determine the average stress ( $\sigma_{av}$ ) over the distance  $r = 0$  to  $2L$ , giving<sup>1</sup>:

$$\sigma_{av} = \frac{K_I}{2L\sqrt{2\pi}} \left( 2\sqrt{\frac{\rho}{2} + 2L} - \frac{\rho}{\sqrt{\frac{\rho}{2} + 2L}} \right) \quad (16)$$

Establishing the fracture condition proposed by the LM,  $K_I$  is easily derived from equation (17) for any given notch radius:

$$\frac{K_I}{2L\sqrt{2\pi}} \left( 2\sqrt{\frac{\rho}{2} + 2L} - \frac{\rho}{\sqrt{\frac{\rho}{2} + 2L}} \right) = \sigma_f^* \quad (17)$$

Once  $K_I$  is obtained, the estimations of the fracture loads ( $P_{est}^{LM}$ ) are straightforward:

$$K_I = \left( \frac{P_{est}^{LM}}{B \cdot W^{0.5}} \right) 6 \left( \frac{a}{W} \right)^{0.5} \left( \frac{1.99 - \left( \frac{a}{W} \right) \left( 1 - \frac{a}{W} \right) \left( 2.15 - 3.93 \left( \frac{a}{W} \right) + 2.7 \left( \frac{a}{W} \right)^2 \right)}{\left( 1 + 2 \frac{a}{W} \right) \left( 1 - \frac{a}{W} \right)^{1.5}} \right) \quad (18)$$

Here, it is important to notice that the whole process only requires the calibration of the equivalent material, which is easily completed from a tensile test, with no need for finite elements modelling and/or calibration fracture tests.

#### 4.3. Results and discussion

Table 2 shows the fracture load predictions obtained through the application of the EMC and the TCD (both the PM and the LM methodologies). Figure 7 shows the same results graphically. It can be observed that the predictions provided when using the Point Method are very accurate, with a maximum deviation (when compared to the average fracture load for each notch radius) of -11.7%, which is obtained for a notch radius of 1.0 mm. It can be observed that, when using the PM, there is not a clear tendency of overestimation or underestimation of the fracture loads, with the points in Figure 7 being located indistinctly over and below the 1/1 line. Overall, the average error is +3.3%. The predictions are good even for the higher radii, for which the Creager-Paris equation validity range is questionable (the Creager-Paris equation is defined for narrow defects, on which  $\rho \ll a$ ).

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3 When using the LM, the error of the predictions is still reasonable, considering the high  
4 scatter of the experimental results, but there is a clear tendency towards the  
5 overestimation of the fracture loads. The maximum deviation regarding the average  
6 experimental fracture load for a particular notch radius is +23.1%, the average value  
7 being +11.7%. Again, the results for higher radii do not seem less accurate than those  
8 obtained for notch radii for which the Creager-Paris assumptions are completely  
9 fulfilled.  
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13 In order to determine the type of failure regime for the tested notched PMMA  
14 specimens, i.e. the small-scale yielding (SSY), moderate-scale yielding (MSY), or  
15 large-scale yielding (LSY), a set of elastic-plastic finite element (FE) analyses were  
16 performed on the SENB specimen, shown in figure 4, in ABAQUS software under  
17 plane-strain conditions. As with the material properties, the true tensile stress-strain  
18 curve of the tested PMMA was given to the FE software point-by-point. Meanwhile, to  
19 reach the size of the plastic region around the notch at the onset of crack initiation from  
20 the notch tip, the mean experimentally obtained maximum load (i.e. average of the four  
21 values presented in 4<sup>th</sup> column of Table 2) was applied to each FE model. The FE  
22 models were meshed by quad shaped elements (see figure 8) and the distribution of  
23 Von-Mises stress around the U-notch tip at the onset of crack initiation is illustrated in  
24 Figure 9. The results for two cases with different notch radii (one for near-crack  
25 condition (0.25 mm) and the other for higher radius (2.5 mm)) indicate that the size of  
26 the plastic zone increases as the radius of the U-notch increases (see Figure 9). This can  
27 be attributed to the stress gradient near the notch tip. For the lower notch radius, the  
28 stress gradient at the notch neighbourhood is significantly higher and hence, the plastic  
29 zone is more localized and its size is relatively small. In contrast, for the higher notch  
30 radius, the stress gradient at the notch tip vicinity is lower, meaning a larger plastic zone  
31 size. For the notch radii equal to 0.25 mm and 2.5 mm, about 8% and 25% of the  
32 ligaments experience plastic deformations at failure, respectively, meaning that the  
33 small notch radius fails by the SSY regime, while the large notch radius by the MSY  
34 regime. The results of the elastic-plastic FE analyses presented in Figure 9 strongly  
35 confirm the experimentally obtained load-displacement curves presented in Figure 6, in  
36 which the curves for the small radius are almost linear while those for the large radius  
37 exhibit a moderate nonlinear portion as a result of the moderate plastic deformations  
38 around the notch tip.  
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## 46 **5. Conclusions**

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48 This paper provides a methodology for the predictions of fracture loads in PMMA  
49 containing U-shaped notches. This material has no fully linear-elastic material, neither  
50 on its tensile curve nor on the fracture specimens with higher radii. This means that a  
51 calibration process is required when analysing fracture processes using the Theory of  
52 Critical Distances (TCD). This calibration requires finite elements modelling, fracture  
53 tests on specimens with different notch radii, or a combination of finite elements with  
54 fracture testing. In order to avoid such a calibration, it is proposed to combine the TCD  
55 with the Equivalent Material Concept (EMC), on which the non-linear material is  
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3 substituted by a perfectly linear-elastic material. This leads to the EMC-TCD criterion,  
4 with fully linear-elastic formulation and without any need for calibration processes  
5 beyond the equivalent material itself which, in any case, is a straightforward calibration  
6 performed from the material stress-strain tensile curve. Moreover, in order to avoid any  
7 finite elements modelling for the estimation of fracture loads, analytical stress fields are  
8 used (Creager-Paris, in this case).  
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11 Under all these assumptions, and considering the scatter associated to the fracture  
12 processes being analysed, the obtained predictions of fracture loads have been  
13 noticeably accurate, especially when using the Point Method (PM) as the TCD  
14 methodology. In such a case, the average deviation between the predicted fracture load  
15 and the corresponding average experimental fracture load has been +3.3 %, with a  
16 maximum deviation of -11.7%. When using the Line Method (LM), the average  
17 deviation has been +11.7%, with a maximum of +23.1%.  
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21 Both the load-displacement curves of the SENB PMMA specimens recorded  
22 experimentally and the plastic zone size determined numerically confirmed the ductile  
23 failure of the U-notched specimens by considerable plastic deformations around the  
24 notch tip (particularly for higher notch radii). For such notched components for which  
25 the plastic zone effects on the fracture behaviour cannot be ignored, the failure criteria  
26 in the context of strictly linear elastic notch fracture mechanics (LENFM) could not  
27 accurately be utilised without employing EMC.  
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37 based.  
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**Tables****Table 1.** Main mechanical (tensile and fracture) properties of the PMMA

<b>E (GPa)</b>	<b><math>\sigma_{0.2}</math>(MPa)</b>	<b><math>\sigma_u</math>(MPa)</b>	<b><math>e_{max}</math>(%)</b>	<b><math>K_c</math> (MPa·m<sup>1/2</sup>)</b>
3.40	47.0	74.5	4.7	2.04

**Table 2.** Experimental programme, experimental fracture loads, and fracture load estimations:  $P_{est}^{PM}$  (EMC-TCD (PM)),  $P_{est}^{LM}$  (EMC-TCD (LM)).

Specimen	Notch length, a (mm)	Notch radius, $\rho$ (mm)	Max. Load, $P_{max}$ (N)	$P_{est}^{PM}$ (N)	Error (%)	$P_{est}^{LM}$ (N)	Error (%)
0-1	5.50	0	130.0	-	-	-	-
0-2	4.72		83.0				
0-3	5.32		131.2				
0.25-1	5	0.25	124.9	111.0	-2.6	128.1	+12.4
0.25-2			119.9				
0.25-3			104.0				
0.25-4			107.1				
0.32-1	5	0.32	117.4	119.4	+8.2	135.8	+23.1
0.32-2			112.6				
0.32-3			102.5				
0.32-4			108.7				
0.5-1	5	0.5	90.0	139.0	+9.4	153.8	+21.1
0.5-2			85.2				
0.5-3			170.3				
0.5-5			162.6				
1.0-1	5	1.0	212.8	183.6	-11.7	195.3	-6.1
1.0-2			213.6				
1.0-3			204.8				
1.0-4			202.8				
1.0-5			202.6				
1.5-2	5	1.5	215.5	219.4	+9.9	229.4	+14.9
1.5-3			165.9				
1.5-4			219.0				
1.5-5			197.9				
2.0-1	5	2.0	258.5	250.2	-0.9	259.1	+2.6
2.0-2			261.1				
2.0-3			237.8				
2.5-1	5	2.5	253.8	277.7	+10.3	285.7	+13.5
2.5-2			259.9				
2.5-3			250.4				
2.5-5			251.3				
2.5-6			243.2				

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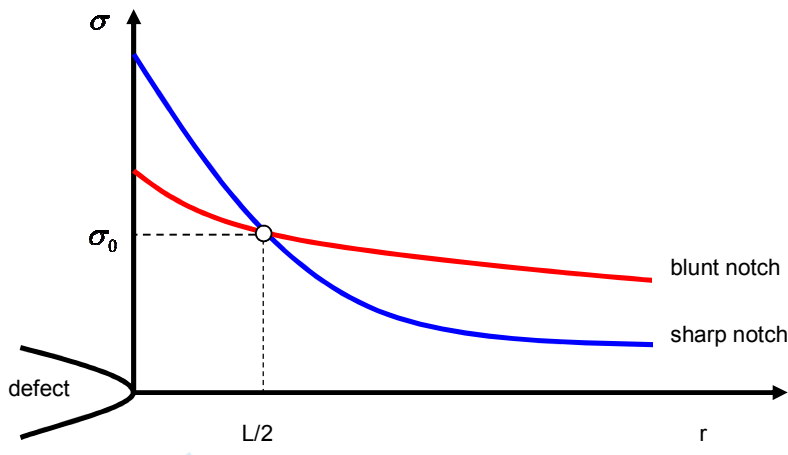
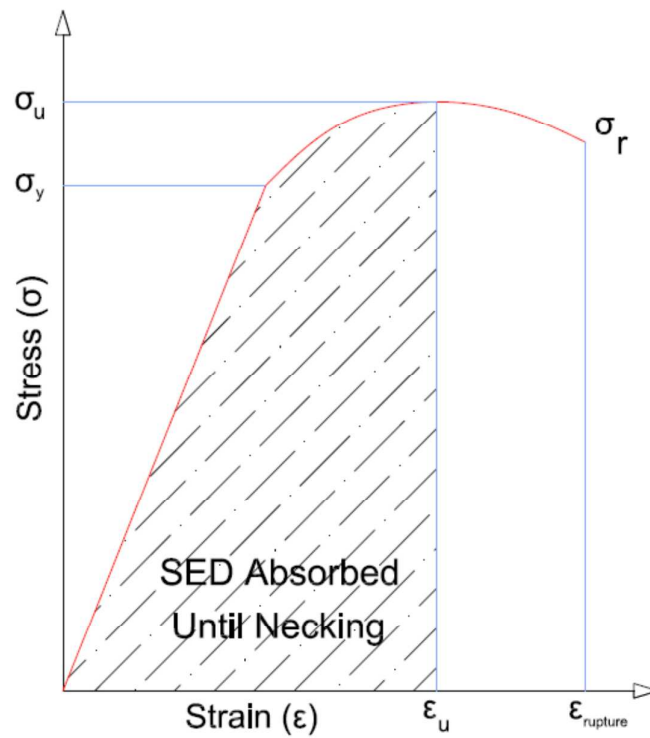


Figure 1. Obtaining  $L$  and  $\sigma_0$  parameters based on the PM definition.

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**Figure 2.** A typical stress-strain curve for a ductile material.

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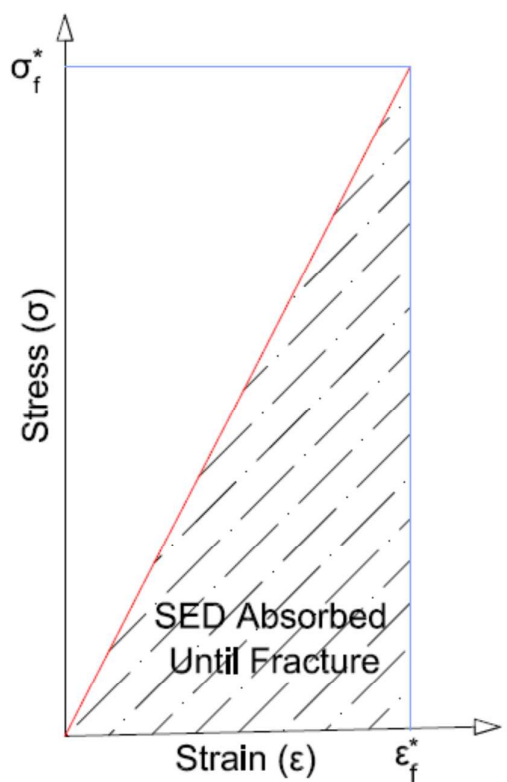
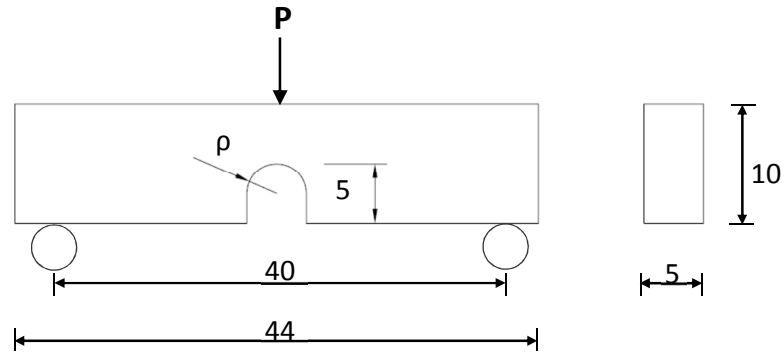


Figure 3. Stress-strain curve for the equivalent brittle material.



**Figure 4.** Schematic showing the geometry of the SENB test specimens. Dimensions in mm,  $\rho$  varying from 0 mm to 2.5 mm. Thickness ( $B$ ) = 5 mm; Width ( $W$ ) = 10 mm.

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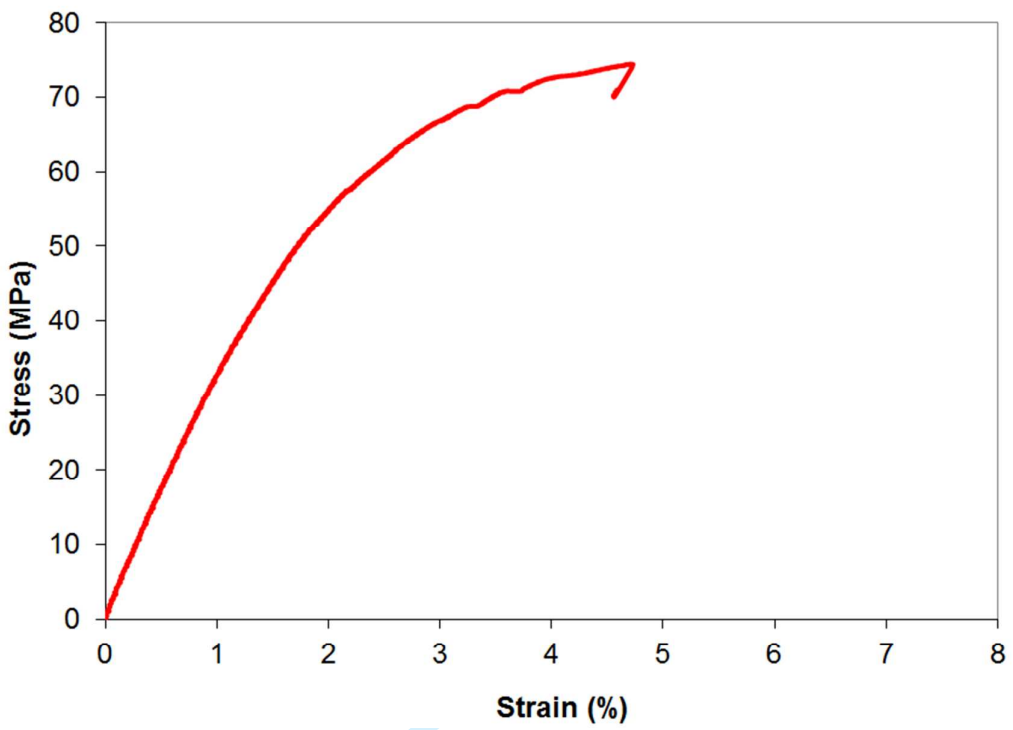


Figure 5. Stress-strain tensile curve of the PMMA being analysed (engineering variables).



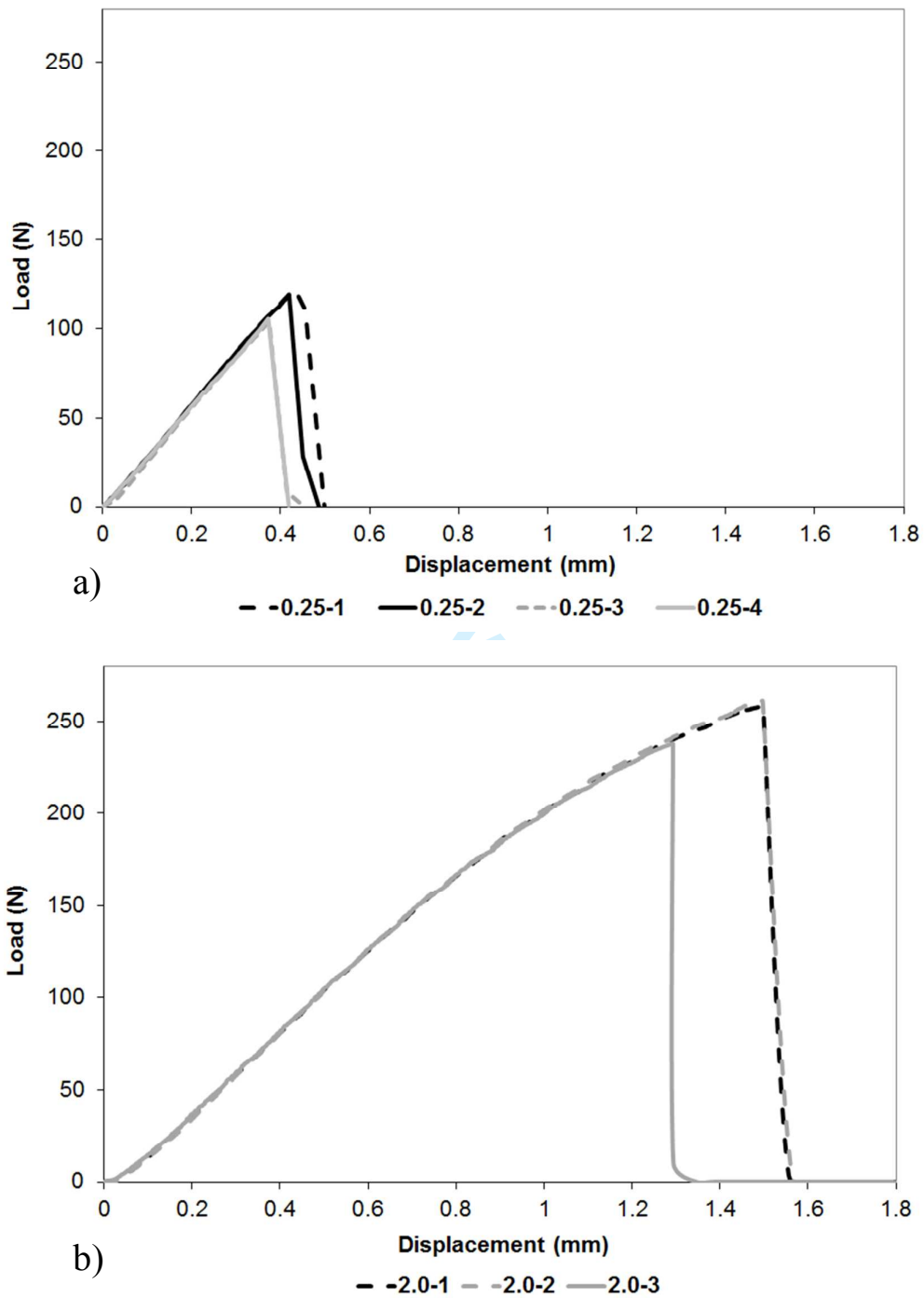
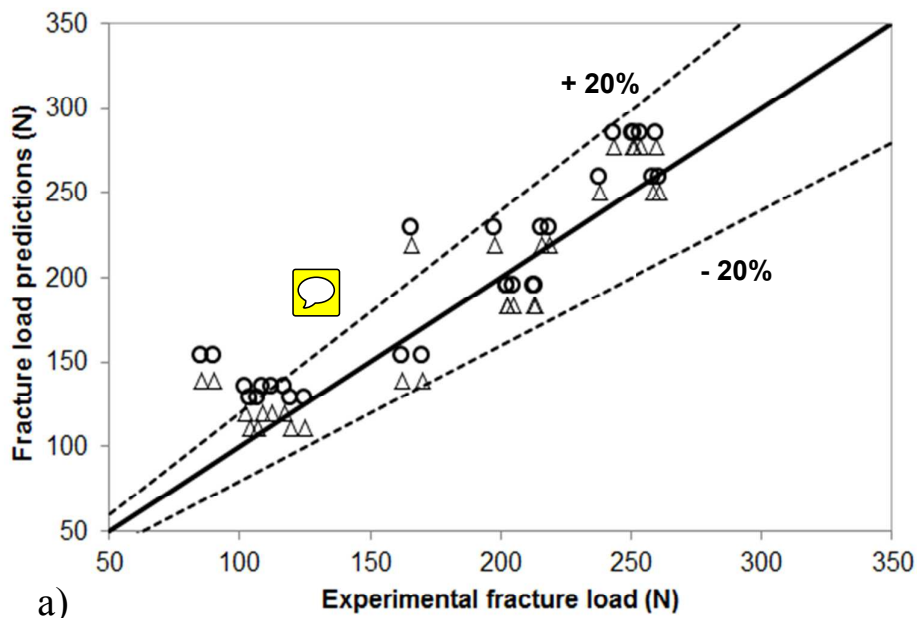


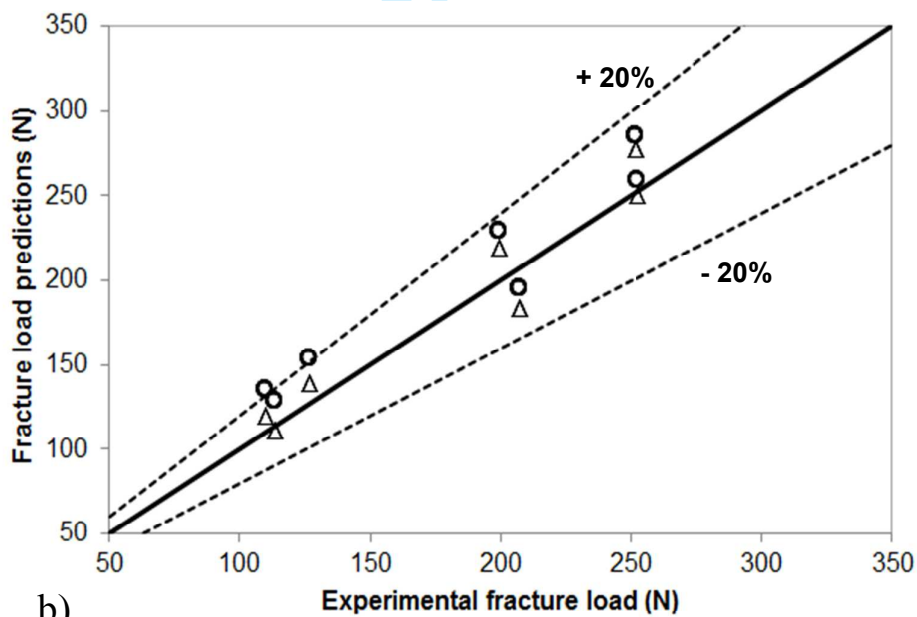
Figure 6. Examples of load-displacement curves obtained in the fracture tests: a) specimens with notch radius 0.25 mm; b) specimens with notch radius 2.0 mm.

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a)

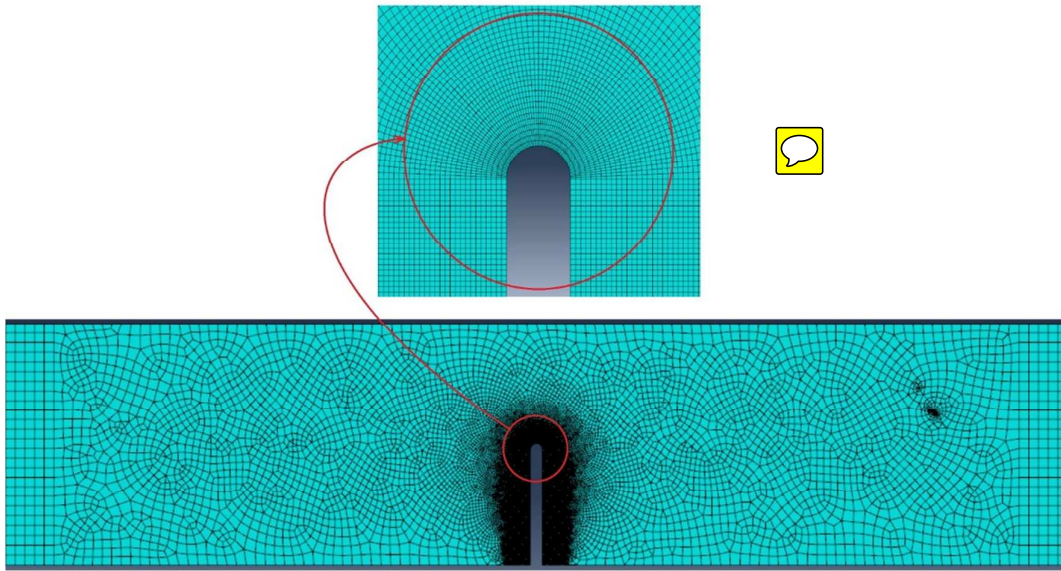
△ EMC-TCD (PM) predictions    ○ EMC-TCD (LM) predictions    — 1:1 line



b)

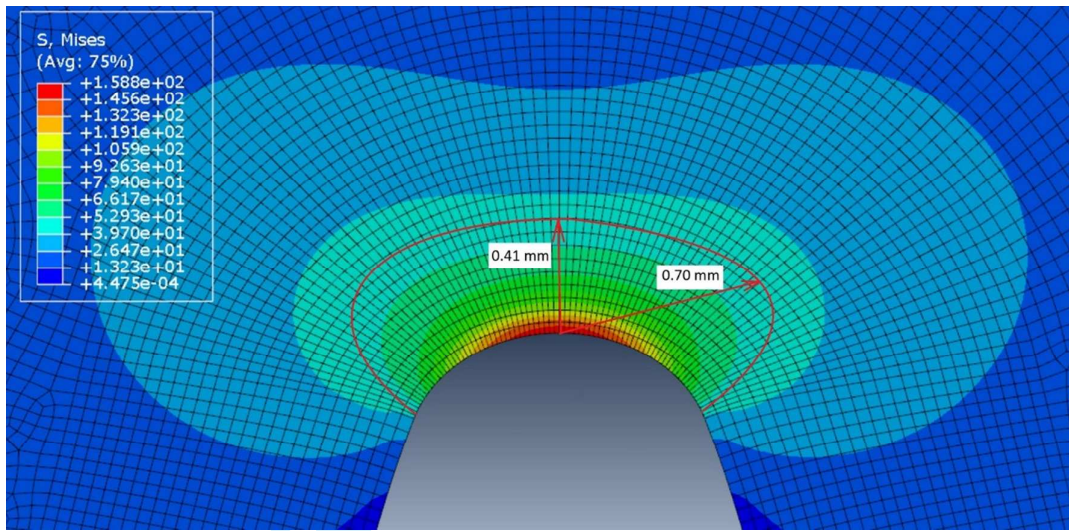
△ EMC-TCD (PM) predictions    ○ EMC-TCD (LM) predictions    — 1:1 line

**Figure 7.** Comparison between fracture load predictions and experimental fracture loads: a) individual tests; b) average values for each set of tests (notch radius).

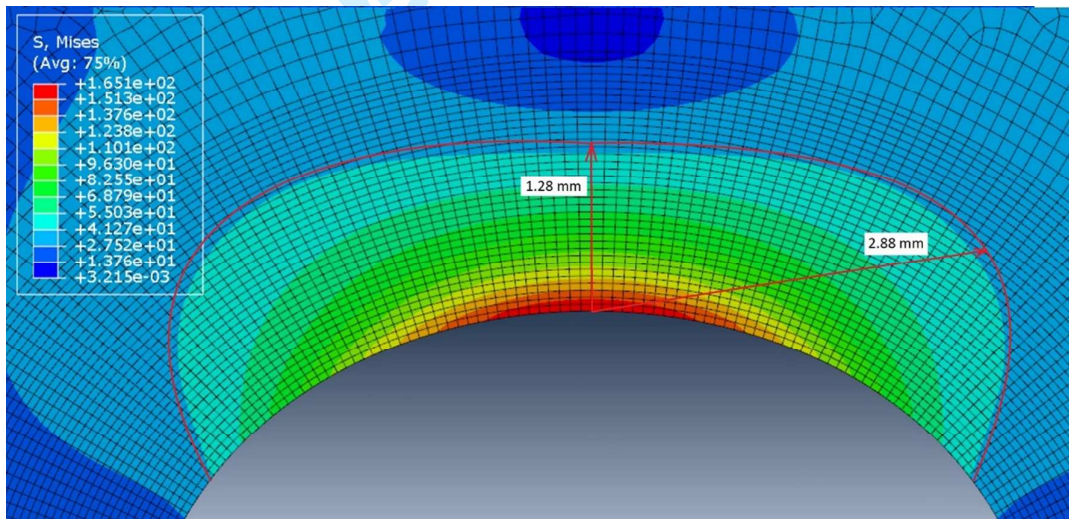


**Figure 8.** FEM model for SENB specimen containing a U-notch of 0.2 mm radius.

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a)



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**Figure 9.** Von-Mises stress distribution around the U-notch: a) notch radius equal to 0.25 mm subjected to the mean fracture load of 114 N and b) notch radius equal to 2.5 mm subjected to the mean fracture load of 252 N. The plastic zone is shown by a red curved line in both cases.