

CSL *COORDINATED SCIENCE LABORATORY*

**SEGMENT METHOD
FOR THE CONTROL
OF SYSTEMS IN THE
PRESENCE OF UNCERTAINTY**

J. MEDANIC
CHENG-I CHEN

UNIVERSITY OF ILLINOIS - URBANA, ILLINOIS

SEGMENT METHOD FOR THE CONTROL OF SYSTEMS IN THE PRESENCE
OF UNCERTAINTY

J. Medanic and Cheng-I Chen
Coordinated Science Laboratory
University of Illinois
Urbana, Illinois

This work was supported in part by the Joint Services Electronics Program under Contract DAAB-07-67-C-0199; also in part by the Aid Force Office of Scientific Research under Grant AFOSR 931-67 and Grant AFOSR 68-1579; and by the National Science Foundation Grant NSF GK-3893.

Reproduction in whole or in part is permitted for any purpose of the United States Government.

This document has been approved for public release and sale; its distribution is unlimited.

SEGMENT METHOD FOR THE CONTROL OF SYSTEMS IN THE PRESENCE OF UNCERTAINTY[†]

J. Medanic^{††} and Cheng-I Chen
Coordinated Science Laboratory
University of Illinois
Urbana, Illinois

ABSTRACT

This paper develops the segment method for the problem of control with uncertainty. It is shown that due to disturbances a segment of values of the performance functional is associated with each admissible control. It is proposed that the optimal control for the problem with uncertainty be defined on the basis of an ordering of associated segments. The ordering is achieved by introducing a segment index. Two particular segments, the pessimistic and the optimistic segment, are shown to be important in the analysis of the problem while minimization of the introduced segment index is the basic synthesis problem. The approach is applied to a linear control problem where uncertainty is caused by external disturbances and specific results are presented. They show the optimal control to retain the advantages of the usual, minimax type designs and ensure better system performance for disturbances other than the worst.

INTRODUCTION

The control problem with uncertainty is commonly treated by stochastic methods where a distribution function is associated to the set of possible variations of plant parameters or initial conditions and where the optimal control is that which minimizes the expected value of the selected perform-

[†]This work was supported in part by the Joint Services Electronics Program under Contract DAAB-07-67-C-0199, by the Air Force Office of Scientific Research under Grant AFOSR 931-67 and Grant AFOSR 68-1579 and by the National Science Foundation Grant NSF GK-3898.

^{††}On leave from Mihailo Pupin Institute, Belgrade, Yugoslavia.

ance functional [1]. Unfortunately, it is difficult to find solutions in all but the simplest problems. Another commonly employed approach is based on the sensitivity of the system outputs and the performance functional to parameter variations where usually a correction δu to the nominal optimal control u^0 is sought. For a discussion of such methods and some recent results see Perkins et al [2] and Kokotovic et al [3]. In many problems with uncertainty neither of these two approaches can be fully justified. It is possible that a distribution function does not even exist, but more commonly that it is unknown, while methods based on first order sensitivities are valid only for small parameter variation.

The problem with large parameter variations has not received as much attention. The only approach employed in practice in the minimax, or worst case design. Recently performance sensitivity-based approaches which consist in finding the minimax solution of a selected performance sensitivity index [4,5] have been proposed. In order to approach the problem where uncertainties due to large parameter variations or external disturbances are present, this paper introduces the notion of performance segments and develops the segment method for the design of controls for uncertain systems. It is shown that a segment of a real line (instead of a point as in the problem without uncertainty) is associated with each control. The optimal control for the problem with uncertainty is obtained by ordering the set of segments by employing a selected segment index.

In this paper the basic features of the segment approach are presented. Specifically, the set of segments is defined, the particular significance of the pessimistic and the optimistic segment in characterizing the nature

of the problem with uncertainty is stressed and the segment index is introduced. The approach is then applied to a linear control problem with external disturbance inputs and results are presented.

SEGMENTS AND THE SEGMENT INDEX

Consider the system

$$\dot{x} = f(x, u, v), \quad x(t_0) = x_0(v), \quad u \in U, \quad v \in V, \quad (1)$$

and an associated functional

$$J(u, v) = \int_{t_0}^T L(x, u, v) dt + g[x(T)] \quad (2)$$

where $x = [x_1(t), \dots, x_n(t)]$ is the state and $u = [u_1(t), \dots, u_m(t)]$ is the control. The perturbation vector $v = [v_1, \dots, v_r]$ may be a parameter variation, or a variation in the initial conditions; it may also be an external disturbance input $v = [v_1(t), \dots, v_r(t)]$. The only specification on the perturbation v is that it belongs to a given compact set V . It is assumed that $f(x, u, v)$, $L(x, u, v)$ and $g(x)$ are sufficiently smooth in all variables and that $J(u, v)$ is continuous on the compact sets U and V .

The problem of control under uncertainty is to design a system which will perform "satisfactorily" with respect to the functional (2) in the presence of any $v \in V$. The more specific problem is to define the best, in this paper also called the optimal control. To achieve this aim some preliminary concepts concerning the structure of the set $\{J(u, v) \mid u \in U, v \in V\}$ of all attainable values of the associated functional, are introduced.

Notice that for each given control \tilde{u} and for all perturbations $v \in V$ equation (2) defines a set, denoted by $E(\tilde{u})$, of attainable values of the

performance functional:

$$E(\tilde{u}) = \{J(\tilde{u}, v) \mid v \in V\}. \quad (3)$$

Because of continuity of J , the set $E(\tilde{u})$ is a segment determined by its upper and lower bounds:*

$$J_h(\tilde{u}) = \max_{v \in V} J(\tilde{u}, v) \quad (4)$$

$$J_l(\tilde{u}) = \min_{v \in V} J(\tilde{u}, v). \quad (5)$$

Hence, the set $\{J(u, v) \mid u \in U, v \in V\}$ may be considered to be composed of segments $E(u)$ corresponding to all controls $u \in U$. In this way, the set of controls U is mapped into a set of segments $I(U)$,

$$I(U) = \{E(u) \mid u \in U\}. \quad (6)$$

The ordering of segments leads to an optimal segment and it is proposed that as optimal be selected the control corresponding to the optimal segment.

The optimal segment is dependent on the appropriate meaning of optimality when considering the set of associated segments. The nature of the problem with uncertainty and the role of the associated functional, (2), suggests that a reasonable way to define optimality is by associating the index

$$S(u) = \sqrt{J_h^2(u) + J_l^2(u)} \quad (7)$$

to the segment $E(u)$; optimal is the segment minimizing this index. Intuition suggests that if an entire segment of values characterizes the performance of the system for a selected control \tilde{u} , the feature desired of the optimal

*Only bounds, inf and sup, in the general case exist, but here it is assumed that all extremas are attained. Most important is that the assumption is satisfied in many problems of interest.

control is to displace and "squeeze" the associated segment of values downward. This is also the motivation behind minimax type designs; in such designs, however, what is achieved by selecting the control associated with the minimal upper bound is to squeeze down only the upper bound of the segment. In the segment method, on the other hand, the intention is to displace the complete segment of values downward and this is accomplished by selecting the control which minimizes the sum of squares of both bounds of the associated segment. The values of the two bounds of a segment are mutually dependent and this minimization must be joint.

To justify the segment approach and the employment of the introduced segment index, two particularly important segments, the pessimistic and the optimistic segment, are introduced in the following section and their influence on the solution of the problem of control with uncertainty is examined. The approach is then illustrated on a specific linear control problem where uncertainty is due to the presence of external disturbances. Some computer results are presented and compared with those attainable by other methods. Computation of the control that minimizes $S(u)$ is the basic synthesis problem in the segment approach. A gradient procedure is utilized in this paper to solve the linear control problem but it cannot be considered general.

THE PESSIMISTIC AND THE OPTIMISTIC SEGMENT

Recall that a segment $E(\tilde{u})$ is associated with a specific control \tilde{u} , and is defined by its upper and lower bounds (4) and (5). To define the pessi-

mistic segment consider u^s and v^s satisfying

$$J(u^s, v^s) = \min_{u \in U} \max_{v \in V} J(u, v). \quad (8)$$

The segment associated with u^s is called the pessimistic segment, $E(u^s)$.

By (4) and (5), with $\tilde{u} = u^s$, its upper and lower bounds are

$$P_h = J_h(u^s) = \max_{v \in V} J(u^s, v) = \min_{u \in U} \max_{v \in V} J(u, v), \quad (9)$$

$$P_l = J_l(u^s) = \min_{v \in V} J(u^s, v). \quad (10)$$

Analogously, to define the optimistic segment consider u^b and v^b satisfying

$$J(u^b, v^b) = \min_{u \in U} \min_{v \in V} J(u, v). \quad (11)$$

The segment associated with u^b is called the optimistic segment, $E(u^b)$,

and from (4) and (5), with $\tilde{u} = u^b$, has bounds

$$O_h = J_h(u^b) = \max_{v \in V} J(u^b, v), \quad (12)$$

$$O_l = J_l(u^b) = \min_{v \in V} J(u^b, v) = \min_{u \in U} \min_{v \in V} J(u, v). \quad (13)$$

A relationship between the two segments is given by

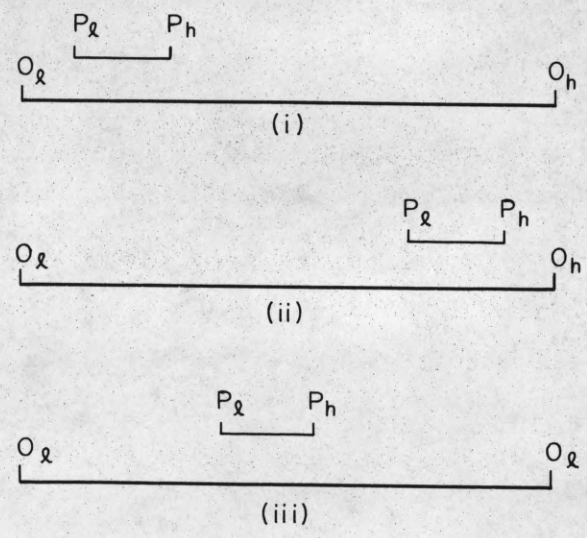
Theorem 1: The pessimistic segment is contained in the optimistic segment,

$$E(u^s) \subset E(u^b). \quad (14)$$

Proof: By definition $J(u^b, v^b) \equiv O_l \leq P_l$.

Moreover, $P_h \equiv J(u^s, v^s) \leq J_h(u)$ for all $u \in U$ and therefore for $u = u^b$.

Hence, $P_h \leq J_h(u^b) \equiv O_h$.



FP-1324

Fig. 1a

The pessimistic segment and the optimistic segment are characteristics of the system, independent of the particular segment index used and out of influence of design. They characterize the nature of the problem of control with uncertainty. To further reveal their role consider, in view of the theorem, the three representative situations depicted in Fig. 1a, where:

- (I) $d_1 \ll d_h, 0_h - 0_1 \gg P_h - P_h$; (II) $d_1 \gg d_h, 0_h - 0_1 \gg P_h - P_1$; (III) $d_1 \approx d_h, 0_h - 0_l \gg P_h - P_l$,
 with $d_1 \equiv P_1 - 0_1$ and $d_h \equiv 0_h - P_h$.

In (I) the position of $E(u^s)$ in $E(u^b)$ shows the pessimistic segment to be satisfactory: u^s achieves the minimal value of the upper bound of the associated segment while a comparatively low value of P_1 , as compared to 0_1 , indicates satisfactory system performance for all $v \in V$. In (II) the position of $E(u^s)$ in $E(u^b)$ shows the optimistic segment to be satisfactory: u^b secures the minimal value of the lower bound of the associated segment while a comparatively low value of 0_h , as compared to P_h , indicates satisfactory system performance even for worst perturbations in V .

While $E(u^s)$ in (I) or $E(u^b)$ in (II) are not the segments that strictly minimize (7) in the given situations, there is negligible gain in system performance to be expected by further sophistication of design. Design can, therefore, be terminated by selecting the pessimistic design in (I) and the optimistic design in (II).

It is because of the general situation in (III) that there is need of more sophisticated design. The pessimistic design displays a short-coming in the high value of P_1 while the optimistic design displays a short-coming in the high value of 0_h . While now the position of $E(u^s)$ within $E(u^b)$ does not indicate the optimal mode of system control. It is stressed, however, that even in this general case the pessimistic and the optimistic segment retain their significance due to the following result stated as a corollary to the above theorem:

Corollary: The segment $E(u^*)$ associated to the optimal control u^* minimizing $S(u)$ must satisfy the condition

$$E(u^s) \subset E(u^*) \subset E(u^b). \quad (15)$$

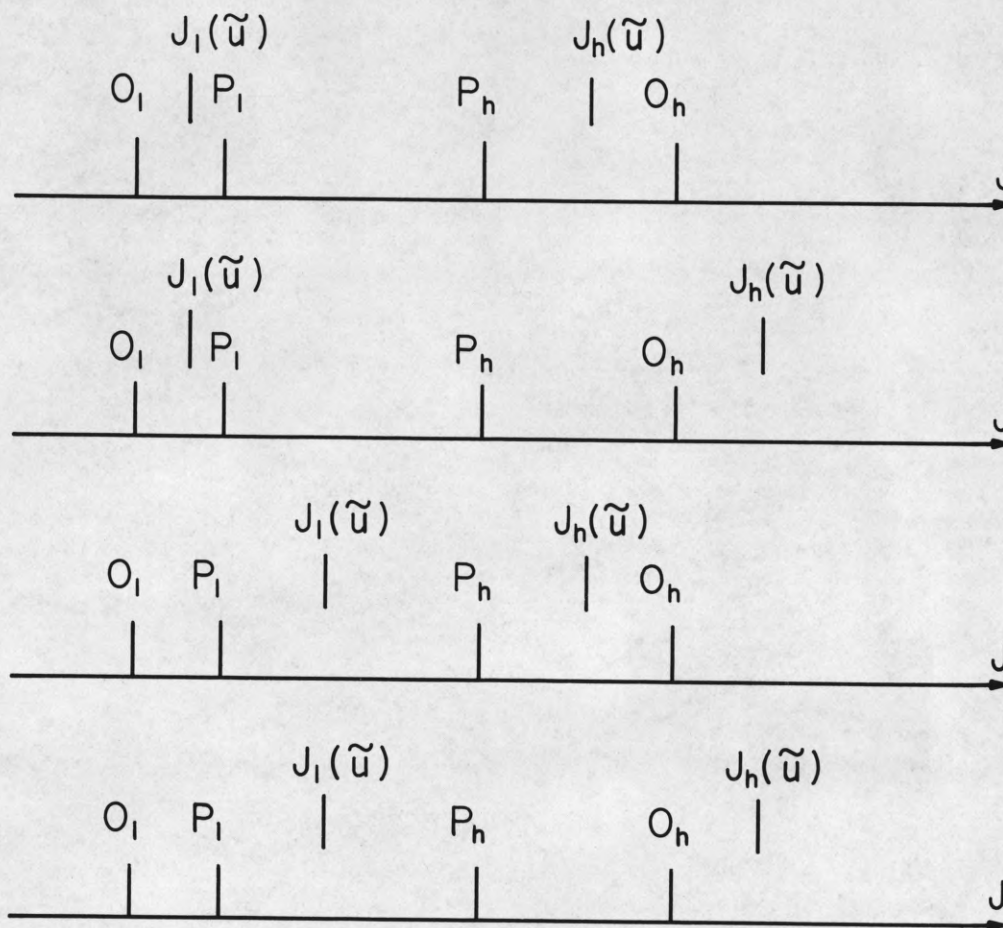
Or, in expanded form,

$$0_1 \leq J_1(u^*) \leq P_1 < P_h \leq J_h(u^*) \leq 0_h. \quad (16)$$

Proof is trivial: Suppose a control \tilde{u} violates (16). This is possible in one of the following ways: (i) let $J_h(\tilde{u}) > 0_h$, $J_1(\tilde{u}) \leq P_1$. Necessarily $J_1(\tilde{u}) \geq 0_1$ and therefore $S(\tilde{u}) > S(u^b)$ and \tilde{u} cannot be optimal; (ii) $J_h(\tilde{u}) \leq 0_h$, $J_1(\tilde{u}) > P_1$. Necessarily $J_h(\tilde{u}) \geq P_h$ and therefore $S(\tilde{u}) > S(u^s)$ and again \tilde{u} cannot be optimal; (iii) the case $J_h(\tilde{u}) > 0_h$, $J_1(\tilde{u}) > P_1$ is evident from the first two.

Moreover, in view of theorem I, there are only four basically distinct positions a segment $E(\tilde{u})$ corresponding to a $\tilde{u} \in U$ can have with respect to the position of $E(u^s)$ and $E(u^b)$. The first is given in the statement of the corollary, the other three in its proof. When the four positions are displayed graphically, Fig. 1b, intuitive feeling agrees with the selection of a control satisfying (16).

In conclusion, the segments $E(u^s)$ and $E(u^b)$ characterize the nature of the problem with uncertainty and moreover confine a range of possible improvements in system design by confining the range within which the optimal segment must be located. In certain cases the pessimistic and optimistic segment directly furnish the solution making further design unnecessary.



FP-1935

Fig. 1b

APPLICATION TO THE DESIGN OF LINEAR CONTROL SYSTEMS

Consider now the application of the segment approach to the problem of designing controls for a linear control system in which uncertainty exists and is caused by external disturbance inputs. The associated functional is assumed to be quadratic in the state and control variables.

Hence, (1) and (2) take the form

$$\dot{x} = A_0(t)x + B(t)u + v, \quad x(t_0) = x_0 \quad (17)$$

$$J(u, v) = \frac{1}{2} \int_{t_0}^T (x^*Qx + u^*Ru) dt + \frac{1}{2} x(T)^*F x(T). \quad (18)$$

The corresponding problem without uncertainty ($v \equiv 0$) is the linear regulator problem whose solution is well known and the details are therefore omitted [7,8], with the remark that the relevant conditions are imposed on $A_0(t)$, $B(t)$, Q , R , and F . The optimal control is

$$u^0 = -R^{-1}B^*K_0 x \quad (19)$$

where the optimal gain matrix, $K_0(t)$, is the solution of the Riccati equation

$$\dot{K}_0 + A_0^*K_0 + K_0A_0 - K_0S_0K_0 + Q = 0, \quad K_0(T) = F, \quad (20)$$

$$S_0 = BR^{-1}B^*, \quad (21)$$

and the minimal value of the performance index is

$$J(u^0) = \frac{1}{2} x_0^* K_0(t_0) x_0. \quad (22)$$

When disturbances from a specified set V act on the system it may be necessary to modify the optimal control. The first step is to analyze the problem by obtaining the pessimistic and the optimistic segment as well as to obtain the position of the segment $E(u^0)$ corresponding to the optimal control u^0 , (19). If necessary, the optimal control for the problem with uncertainty is then modified and obtained by minimizing (7).

The nature of the problem with uncertainty depends on the set of disturbances considered as admissible. Here and in the following section

admissible disturbances are implicitly defined as those satisfying

$$\int_{t_0}^T v^* v dt \leq \gamma \int_{t_0}^T x^* x dt, \quad (23)$$

where γ is a specified constant. Such a characterization of the set V indicates a relative dependence of disturbances on the system trajectory.

Consider now the problem of finding the bounds of the pessimistic segment. P_h the minimax of $J(u,v)$ is simply found since it is at the same time a saddle point [9,10]. In view of (23) the Hamiltonian for the saddle point problem becomes

$$H = \frac{1}{2}[x^*(Q + \gamma m_1 I)x + u^* R u - m_1 v^* v] + p^*(A_0 x + B u + v). \quad (24)$$

From the necessary conditions for a saddle point, the minimizing control and the maximizing disturbance may be written in the form[†]

$$u^s = -R^{-1} B^* K_s x, \quad v^s = \frac{1}{m_1} K_s x, \quad (25)$$

where $K_s(t)$ is the solution of the Riccati equation

$$\dot{K}_s + A^* K_s + K_s A_0 - K_s S_s K_s + Q + \gamma m_1 I = 0, \quad K_s(T) = F \quad (26)$$

with

$$S_s = B R^{-1} B^* - \frac{1}{m_1} I = S_0 - \frac{1}{m_1} I. \quad (27)$$

[†]It is not implied by (25) that the disturbances appear in the system in some feedback realization but that the worst disturbance $v(t)$, given u^s is expressible in this way.

The multiplier $m_1 > 0$ is such that equality in (23) is satisfied. The value of the upper bound of the pessimistic segment is

$$P_h = \frac{1}{2} x_0^* K_s(t_0) x_0. \quad (28)$$

The lower bound P_1 is obtained by closing the loop with the feedback control u^s and finding the most favorable disturbance in the resulting system. The system equation becomes

$$\dot{x} = (A_0 - S_0 K_s) x + v, \quad x(t_0) = x_0. \quad (29)$$

with

$$J(u^s, v) = \frac{1}{2} \int_{t_0}^T x^* (Q + K_s S_0 K_s) x dt + \frac{1}{2} x(T)^* F x(T). \quad (30)$$

The most favorable, minimizing, disturbance is

$$v^1 = -\frac{1}{m_2} N_1 x, \quad (31)$$

where $N_1(t)$ is the solution of the Riccati equation

$$\dot{N}_1 + (A_0 - S_0 K_s)^* N_1 + N_1 (A_0 - S_0 K_s) - \frac{1}{m_2} N_1^2 + Q + K_s S_0 K_s - \gamma m_2 I = 0, \quad N_1(T) = F, \quad (32)$$

where m_2 is a positive multiplier such that equality in (23) is satisfied.

The value of the lower bound of the pessimistic segment is

$$P_1 = \frac{1}{2} x_0^* N_1(t_0) x_0. \quad (33)$$

A similar set of equations can be shown to define the bounds of the optimistic segment:

$$u^b = -R^{-1} B^* K_b x, \quad (34)$$

$$v^b = \frac{1}{m_3} K_b x, \quad (35)$$

$$O_1 = \frac{1}{2} x_0^* K_b(t_0) x_0, \quad (36)$$

$$v^h = -\frac{1}{m_4} N_h x, \quad (37)$$

$$O_h = \frac{1}{2} x_o^* N_h(t_o) x_o. \quad (38)$$

$K_b(t)$ and $N_h(t)$ are solutions of (26) and (32) respectively, where m_3 , m_4 are now, respectively, negative multipliers. The values of the multipliers m_1 , m_2 , m_3 , and m_4 are initially unknown and are found by the iterative solution of the corresponding Riccati equation together with the system equation and the constraint condition (23) until equality in (23), in each case, becomes satisfied.

Two examples are presented for a discussion of the results. Both examples are of second order with the same associated functional characterized by

$$Q = \begin{vmatrix} 1 & 0 \\ 0 & 1 \end{vmatrix}, \quad R = 1,$$

with $T = \infty$. In the first example, denoted by S, the system is characterized by

$$A_o = \begin{vmatrix} -1 & 1 \\ 0 & -1 \end{vmatrix}, \quad B = \begin{vmatrix} 0 \\ 2 \end{vmatrix}.$$

In the second example, denoted by C, the system is characterized by

$$A_o = \begin{vmatrix} 0 & 1 \\ 0 & -1 \end{vmatrix}, \quad B = \begin{vmatrix} 1 \\ 1 \end{vmatrix}.$$

The results were obtained on a digital computer and have been plotted in function of the parameter γ . Hence, they represent an analysis of a family of problems and show how the systems react under increasing level of external disturbances. As $\gamma \rightarrow 0$, both the pessimistic and the optimistic segment degenerate to the point $J(u^0)$ while the controls u^s , u^b converge to the optimal control u^0 . Both examples, Fig. 2, show the typical position

of $E(u^s)$ in $E(u^b)$ characteristic to this problem. The supremacy of the pessimistic over the optimistic segment, particularly for greater values of γ , is evident. As expected, O_1 decreases with the increase of γ indicating the presence of disturbances not adverse to system performance. On the other side, $O_h \rightarrow \infty$ as $\gamma \rightarrow \gamma_{crit}$, where γ_{crit} is smaller for less stable plants. Concerning the pessimistic segment, it is noted that although the sufficiency condition for the existence of a saddle point (positive semi-definiteness of S_s) is violated even for arbitrary small m_1 , a saddle point with a finite value of P_h exists in some systems for relatively large γ . In other examples, particularly those where the plant is less susceptible to control action, P_h tends to infinity as $\gamma \rightarrow \gamma_{max}^\dagger$ (e.g., in system C but with $B^* = \begin{bmatrix} 0 & 1 \end{bmatrix}^*$, $P_h \rightarrow \infty$ at about $\gamma \approx 1$). The dependence of P_1 on γ is also characteristic. As expected, its value decreases as γ increases but only until a minimal value of P_1 is reached for some γ_{min} . Increasing γ further causes an increase in the value of P_1 . This manifests the necessity of employing a rigid control to give the system relatively good performance characteristics even for worst disturbances.

SYNTHESIS OF THE OPTIMAL CONTROL

The examples demonstrate that the pessimistic design might be a satisfactory choice in this problem, particularly for large γ . For smaller γ the supremacy of the pessimistic over the optimistic design is not as

[†]The limiting values γ_{crit} and γ_{max} depend on the system, were obtained experimentally and have not been investigated at the present time.

pronounced and it is meaningful to undertake further design.

Consider therefore the synthesis problem of finding the linear feedback control that minimizes $S(u)$. The gradient technique developed below restricts further discussion to time-invariant linear systems and an infinite integration interval, as in the examples. The synthesis problem reduces to finding the optimal feedback control u^* characterized by the optimal gain matrix K^* . The procedure consisted in the following. Let the feedback control

$$\tilde{u} = -R^{-1} B^* K \tilde{x} \quad (39)$$

be implemented where \tilde{K} is an initial guess for K^* . Elements of \tilde{K} are adjusted in accordance with the usual algorithm,

$$\delta k_{ij} = -\frac{1}{\rho} \frac{\partial S(\tilde{u})}{\partial k_{ij}} \quad (40)$$

where

$$\frac{\partial S}{\partial k_{ij}} = \frac{1}{S} \left[J_h \frac{\partial J_h}{\partial k_{ij}} + J_l \frac{\partial J_l}{\partial k_{ij}} \right]$$

Implementation of the algorithm demands that for each \tilde{u} , the values of $J_h(\tilde{u})$, $J_l(\tilde{u})$, $S(\tilde{u})$, $\frac{\partial J_h(\tilde{u})}{\partial k_{ij}}$, $\frac{\partial J_l(\tilde{u})}{\partial k_{ij}}$ be computed.

Bounds, $J_h(\tilde{u})$, $J_l(\tilde{u})$, of the segment associated with \tilde{u} are found by solving the Riccati equation (32) (to find the steady state solution); the lower bounds with a negative multiplier m_l , the upper bound with a positive multiplier m_h . $S(\tilde{u})$ is then found from (7). The gradients are found from

$$\frac{\partial J_h}{\partial k_{ij}} = \frac{1}{2} x_o^* \frac{\partial N_h}{\partial k_{ij}} x_o, \quad (42)$$

$$\frac{\partial J_1}{\partial k_{ij}} = \frac{1}{2} x_o^* \frac{\partial N_h}{\partial k_{ij}} x_o. \quad (43)$$

where the sensitivity coefficients $\frac{\partial N_h}{\partial k_{ij}}$, $\frac{\partial N_1}{\partial k_{ij}}$ are the (steady state) solutions of the sensitivity equation of the Ricatti equation (32):

$$\begin{aligned} \frac{\partial N_s}{\partial k_{ij}} + (A_o - S_o K - \frac{1}{m_s} N_s)^* \frac{\partial N_s}{\partial k_{ij}} + \frac{\partial N_s}{\partial k_{ij}} (A_o - S_o K - \frac{1}{m_s} N_s) \\ + \frac{\partial K}{\partial k_{ij}} S_o (K - N_s) + (K - N_s) S_o \frac{\partial K}{\partial k_{ij}} = 0, \end{aligned}$$

$$\frac{\partial N_s}{\partial k_{ij}} (T) = 0, \quad s = 1, h. \quad (44)$$

The pairs (N_h, m_h) , (N_1, m_1) are known and have been obtained as solutions of (32) in the course of determining $J_h(\tilde{u})$, $J_1(\tilde{u})$. The gradient procedure is easily implemented on a digital computer and has been employed to determine the optimal control gains for the examples considered earlier.

The results illustrate the type of results expected and obtained from the segment approach. They have again been plotted for the whole family of problems when γ is allowed to lie in the range $0 \leq \gamma \leq 1$. For example, in system C, for $\gamma = .9$ the optimal control gain matrix K_o , for the problem without disturbances, and the pessimistic gain matrix K_s , serving as the initial guess for K^* were found in the analysis to be

$$K_o = \begin{vmatrix} .813 & .174 \\ .174 & 1.157 \end{vmatrix}, \quad \tilde{K} = K_s = \begin{vmatrix} 1.109 & .274 \\ .274 & .561 \end{vmatrix}$$

After the iterative procedure was judged to have approached the minimum within prescribed bounds, the result was

$$K^* = \begin{vmatrix} 1.108 & .237 \\ .237 & .508 \end{vmatrix}.$$

It is most appropriate to compare these results with corresponding results in minimax (wrost case) designs. Such designs minimize the upper bound of segments, correspond, therefore, to the pessimistic design and result in the selection of the pessimistic control. The optimal control from the segment approach as demonstrated, Fig. 3, 4, retains the advantages of the worst case designs while improving the system performance for perturbations other than the worst. With the optimal control from the segment approach the upper bound of the associated segment remains virtually unchanged (an increase of .8% from P_h) while the lower bound undergoes a substantial decrease (about 10% decrease from the value of P_1 or about a 20% decrease of the range $d_1 = P_1 - 0_1$). The implementation of the optimal control must be made at the risk of greater expenditure of control energy. It should, however, be noted that with minimax type designs generally even more energy expenditure must be allowed.

The problem with uncertainty considered above is of course specific, its nature depending on the set V of admissible disturbances as defined by (23). It is, however, emphasized that this study does characterize the reaction of linear system to increasing level of external disturbances. Moreover, with simple modifications it is possible to consider the following

frequently appearing problems with uncertainty: (i) a more general statement of the above problem is obtained if in (17) the term Cv is put in place of v . Moreover, the approach is not restricted to problems with complete state feedback; (ii) energy constrained external disturbances may be considered by assuming V is given by

$$\int_{t_0}^T v^* v dt \leq \gamma \quad (45)$$

With slight modifications in the computational algorithm it is possible to apply the gradient procedure to obtain the optimal feedback control. The authors have considered such problems for time invariant systems and infinite integration interval and have obtained results corresponding to those presented here for the constraint set (23); (iii) a frequently discussed problem with uncertainty occurs when amplitude bounded external disturbances act on the system. The set V is then assumed to consist of piecewise continuous time functions such that for all $t \in [0, T] \mid v_i \mid \leq 1$, $i = 1, \dots, r$ and the computational requirements become more complex. The authors do not have sufficient experience with such problems at the present time; (iv) Another common problem with uncertainty occurs in linear systems when some plant parameters are known and the set V consists of a closed and bounded domain in the plant parameter space. For any $v \in V$ there exists an optimal linear feedback controller and it is again advantageous to look for the optimal linear feedback control when applying the segment approach; (v) Finally, one of the most common problems, due to variations in the initial conditions is solved trivially since the optimal

control (19) is independent of the initial conditions and modifications of the optimal control is unnecessary.

CONCLUSIONS

In conclusion, it is hoped that the results presented in this paper will contribute to better understanding of control problems in the presence of uncertainty. The intention of this paper was to introduce the basic notions and justify the approach by presenting specific results for a particular problem. Hence, a large portion of the paper was devoted to a specific linear control problem although the method can, with slight modifications, be used to solve other commonly discussed problems with uncertainty in linear control systems with a quadratic associated functional. If the obtained results are representative of the general situation and acceptable from the design point of view the basic problem is the synthesis problem consisting in the determination of the optimal control u^* which minimizes $S(u)$. Methods used to minimize $S(u)$ depend on the problem, in particular on the nature of the sets U and V . Gradient and other local procedures may be used when the solution is local [5]. It is however, possible that the solution is not local in which case global algorithms such as presently being developed for the minimax problem must be employed [5,13,14]. A complete discussion of all computational aspects even for this type of problems is out of the scope of this paper.

ACKNOWLEDGMENT

The authors would like to extend their appreciation to Professors J. B. Cruz, Jr., P. Kokotović, and W. R. Perkins for stimulating discussions and many helpful comments in writing this paper.

REFERENCES

1. M. Aoki, Optimization of Stochastic Systems, Academic Press, New York, 1967.
2. W. R. Perkins, J. B. Cruz, Jr., and R. Gonzales, "Design of Minimum Sensitivity Systems," IEEE Trans. on Automatic Control, Vol. AC-13, No. 2, pp. 159-167, April, 1968.
3. P. Kokotovic, J. B. Cruz, Jr., J. E. Heller, and P. Sannuti, "Synthesis of Optimally Sensitive Systems," Proc. IEEE, Vol. 56, No. 8, pp.
4. R. A. Rohrer and M. Sobral, "Sensitivity Considerations in Optimal Control Systems," IEEE Trans. on Automatic Control, Vol. AC-10, No. 1, pp. 43-48, January, 1965.
5. D. M. Salmon, "Minimax Controller Design," IEEE Trans. on Automatic Control, Vol. AC-13, No. 4, pp. 369-376, August, 1968.
6. J. Medanic, "Three Segment Method in the Design of Control Systems," Fifth Annual Allerton Conf. on Circuit and System Theory, Monticello, Ill., October, 1967.
7. R. E. Kalman, "Contribution to the Theory of Optimal Control," Bol. Soc. Mat. Mexico, Vol. 5, pp. 102-119, 1960.
8. M. Athans and P. L. Falb, Optimal Control: An Introduction to Theory and Applications, McGraw-Hill, New York, 1966.
9. Y. C. Ho, A. E. Bryson, Jr., and S. Baron, "Differential Games and Optimal Pursuit and Evasion Strategies," IEEE Trans. on Automatic Control, Vol. AC-10, pp. 385-389, October, 1965.
10. J. Medanic, "Bounds on the Performance Index and the Riccati Equation in Differential Games," IEEE Trans. on Automatic Control, Vol. AC-12, October, 1967.
11. J. Danskin, The Theory of Max-Min, Series in Econometrics and Operations Research, Vol. V, Springer-Verlag, New York, 1967.
12. V. F. Demjanov, Kibernetike, Vol. 2, No. 6, 1966 (will be translated into English).
13. J. Medanic, "An Elimination Algorithm for Computation on the minimax," Sixth Annual Allerton Conf. on Circuit and System Theory, Monticello, Ill., October, 1968.
14. J. E. Heller, Ph.D. thesis at the Coordinated Science Laboratory, University of Illinois, Urbana, Illinois.

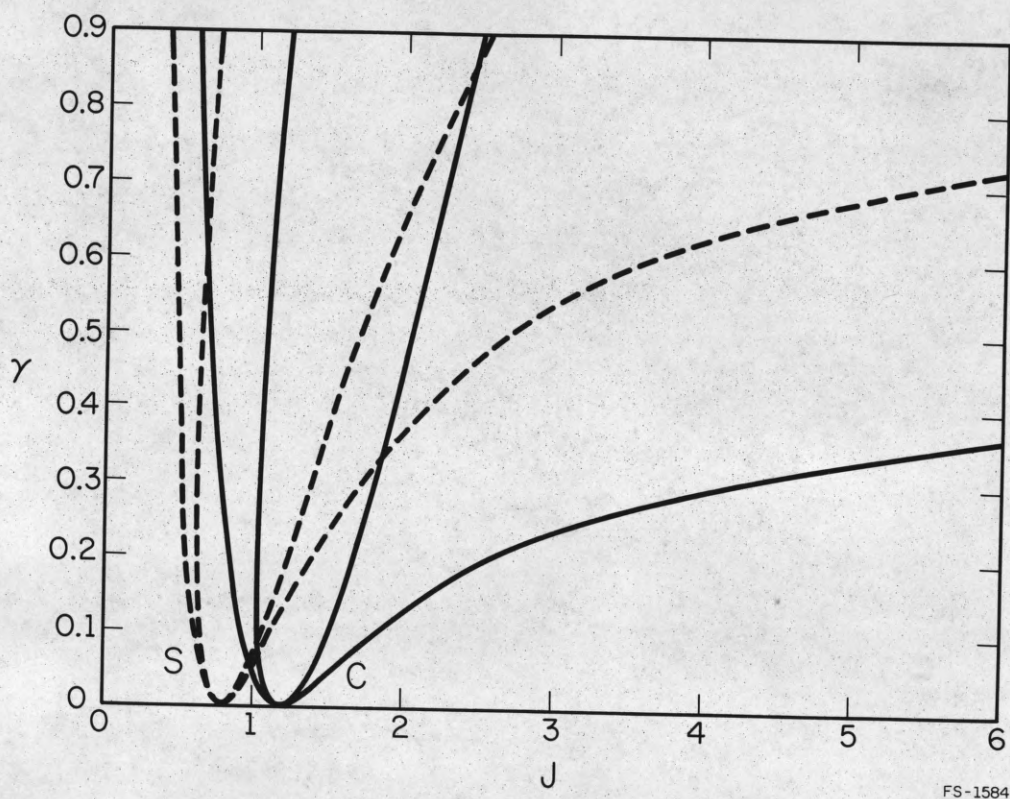


Fig. 2. The dependence of bounds O_1 , P_1 , and O_h on the amount of external disturbances, i.e. on the parameter γ . Full lines belong to system C, dotted lines to system S.

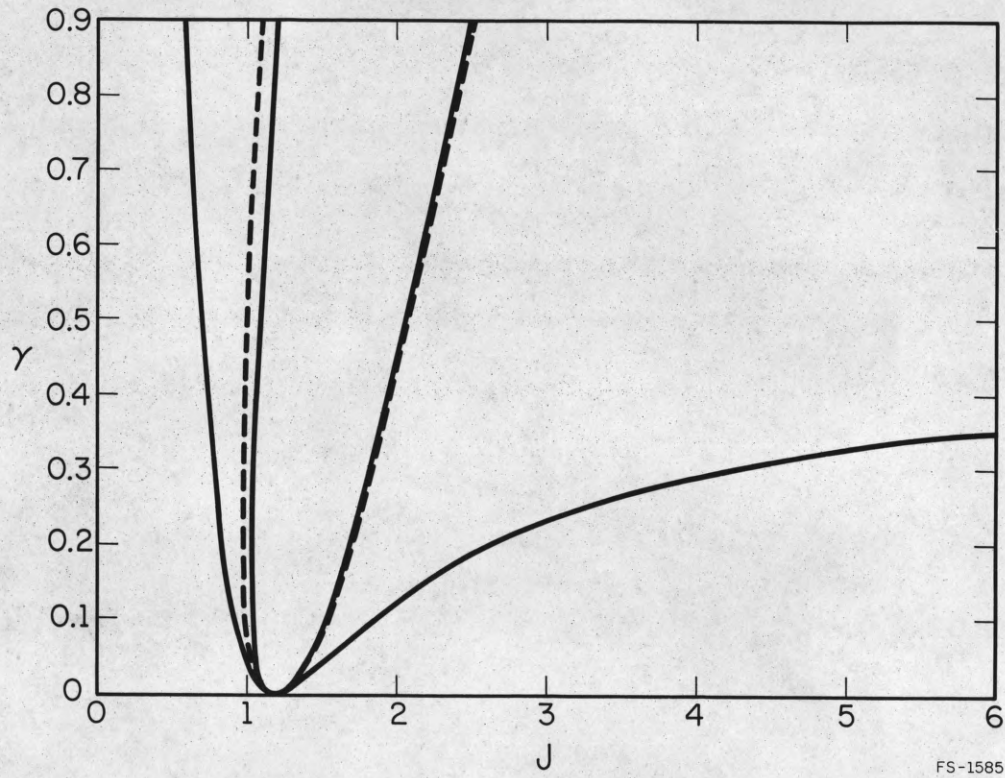


Fig. 3. Dotted lines indicate the location of the optimal segment, corresponding to the optimal control u^* , for system C.

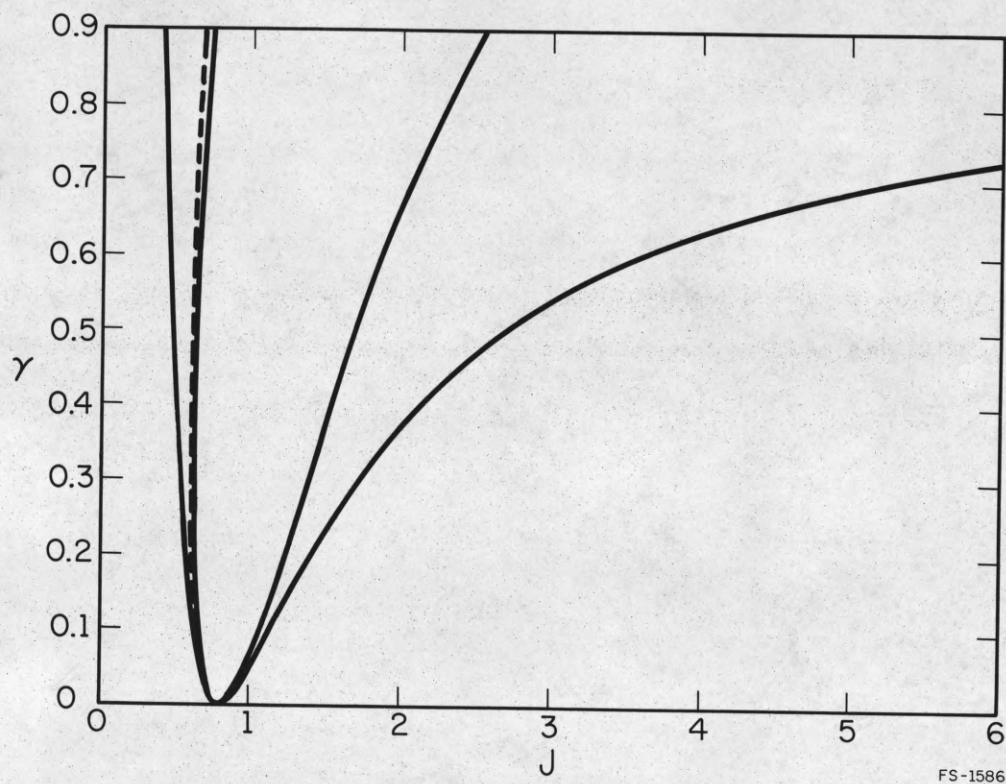


Fig. 4. Dotted lines indicate the location of the optimal segment for system S. The upper bound virtually coincides with P_h .

Distribution List as of 1 October, 1969

Dr A.A. Dougal
Asst Director (Research)
Ofc of Defense Res & Eng
Department of Defense
Washington, D.C. 20301

Office of Deputy Director
(Research and Information, Rm 3D1037)
Department of Defense
The Pentagon
Washington, D.C. 20301

Director, Advanced Research Projects
Agency
Department of Defense
Washington, D.C. 20301

Director for Materials Sciences
Advanced Research Projects Agency
Department of Defense
Washington, D.C. 20301

Headquarters
Defense Communications Agency (340)
Washington, D.C. 20305

Defense Documentation Center
Attn: DDC-PCA
Cameron Station
Alexandria, Virginia 22314 (50 Copies)

Director
National Security Agency
Attn: TRL
Fort George G. Meade, Maryland 20755

Weapons Systems Evaluation Group
Attn: Colonel Blaine D. Vogt
400 Army-Navy Drive
Arlington, Virginia 22202

Central Intelligence Agency
Attn: OCR/DD Publications
Washington, D.C. 20505

Hq USAF (AFRDD)
The Pentagon
Washington, D.C. 20330

Hq USAF (AFRDC)
The Pentagon
Washington, D.C. 20330

Hq USAF (AFRSD)
The Pentagon
Washington, D.C. 20330

Colonel E.P. Gaines, Jr.
ACMA/PO
1901 Pennsylvania Ave N.W.
Washington, D.C. 20451

Lt Col R.B. Kalisch (SREE)
Chief, Electronics Division
Directorate of Engineering Sciences
Air Force Office of Scientific Research
Arlington, Virginia 22209

Dr I.R. Mirman
AFSC (SCT)
Andrews Air Force Base, Maryland 20331

AFSC (SCTSE)
Andrews Air Force Base, Maryland 20331

Mr Morton M. Pavane, Chief
AFSC Scientific and Technical Liaison Office
26 Federal Plaza, Suite 1313
New York, New York 10007

Rome Air Development Center
Attn: Documents Library (EMFLD)
Griffiss Air Force Base, New York 13440

Mr H.E. Webb (EMHIS)
Rome Air Development Center
Griffiss Air Force Base, New York 13440

Dr L.M. Hollingsworth
AFCL (CRB)
L.G. Hanscom Field
Bedford, Massachusetts 01730

AFCL (CRPLR), Stop 29
AFCL Research Library
L.G. Hanscom Field
Bedford, Massachusetts 01730

Hq ESD (ESTI)
L.G. Hanscom Field
Bedford, Massachusetts 01730 (2 copies)

Professor J. J. D'Asso
Dept of Electrical Engineering
Air Force Institute of Technology
Wright-Patterson AFB, Ohio 45433

Dr H.V. Noble (CAVI)
Air Force Avionics Laboratory
Wright-Patterson AFB, Ohio 45433

Director
Air Force Avionics Laboratory
Wright-Patterson AFB, Ohio 45433

AFAL (AVTA/R.D. Larson)
Wright-Patterson AFB, Ohio 45433

Director of Faculty Research
Department of the Air Force
U.S. Air Force Academy
Colorado Springs, Colorado 80840

Academy Library (DFSLR)
USAF Academy
Colorado Springs, Colorado 80840

Director
Aerospace Mechanics Division
Frank J. Seiler Research Laboratory (OAR)
USAF Academy
Colorado Springs Colorado 80840

Director, USAF PROJECT RAND
Via: Air Force Liaison Office
The RAND Corporation
Attn: Library D
1700 Main Street
Santa Monica, California 90045

Hq SANSO (SMTA/Lt Nelson)
AF Unit Post Office
Los Angeles, California 90045

Det 6, Hq OAR
Air Force Unit Post Office
Los Angeles, California 90045

AULST-9663
Maxwell AFB, Alabama 36112

AFTR Technical Library
(STV, HQ-133)
Patrick AFB, Florida 32925

ADDC (ADBF-13)
Eglin AFB, Florida 32542

Mr B.R. Locks
Technical Adviser, Requirements
USAF Security Service
Kelly Air Force Base, Texas 78241

Hq AMD (AMR)
Brooks AFB, Texas 78235

USAFSAM (SMKOR)
Brooks AFB, Texas 78235

Commanding General
Attn: STEMS-RE-1, Technical Library
White Sands Missile Range
New Mexico 88002 (2 copies)

Hq AEDC (AETS)
Attn: Library/Documents
Arnold AFB, Tennessee 37389

European Office of Aerospace Research
APO New York 09667

Physical & Engineering Sciences Division
U.S. Army Research Office
3045 Columbia Pike
Arlington, Virginia 22204

Commanding General
U.S. Army Security Agency
Attn: IARD-T
Arlington Hall Station
Arlington, Virginia 22212

Commanding General
U.S. Army Materiel Command
Attn: AMED-TF
Washington, D.C. 20315

Technical Director (SMIFA-A2000-107-1)
Frankford Arsenal
Philadelphia, Pennsylvania 19137

Redstone Scientific Information Center
Attn: Chief, Document Section
U.S. Army Missile Command
Redstone Arsenal, Alabama 35809

Commanding General
U.S. Army Missile Command
Attn: AMEMI-REX
Redstone Arsenal, Alabama 35809

Commanding General
U.S. Army Strategic Communications Command
Attn: SCC-CG-SAE
Fort Huachuca, Arizona 85613

Commanding Officer
Army Materials and Mechanics Res. Center
Attn: Dr H. Priest
Watertown Arsenal
Watertown, Massachusetts 02172

Commandant
U.S. Army Air Defense School
Attn: Missile Science Division, C&S Dept
P.O. Box 9390
Fort Bliss, Texas 79916

Commandant
U.S. Army Command & General Staff College
Attn: Acquisitions, Library Division
Fort Leavenworth, Kansas 66027

Commanding Officer
U.S. Army Electronics R&D Activity
White Sands Missile Range, New Mexico 88002

Mr Norman J. Field, AMSEL-RD-8
Chief, Office of Science & Technology
Research and Development Directorate
U.S. Army Electronics Command
Fort Monmouth, New Jersey 07703

Commanding Officer
Harry Diamond Laboratories
Attn: Dr Berthold Altman (AMXDD-TI)
Connecticut Avenue and Van Ness St N.W.
Washington, D.C. 20438

Director
Walter Reed Army Institute of Research
Walter Reed Army Medical Center
Washington, D.C. 20012

Commanding Officer (AMKRD-BAT)
U.S. Army Ballistics Research Laboratory
Aberdeen Proving Ground
Aberdeen, Maryland 21005

Technical Director
U.S. Army Limited War Laboratory
Aberdeen Proving Ground
Aberdeen, Maryland 21005

Commanding Officer
Human Engineering Laboratories
Aberdeen Proving Ground
Aberdeen, Maryland 21005

U.S. Army Munitions Command
Attn: Science & Technology Br. Bldg 59
Picatinny Arsenal, SMPTA-VA6
Dover, New Jersey 07801

U.S. Army Mobility Equipment Research
and Development Center
Attn: Technical Document Center, Bldg 315
Fort Belvoir, Virginia 22060

Director
U.S. Army Engineer Geodesy,
Intelligence & Mapping
Research and Development Agency
Fort Belvoir, Virginia 22060

Dr Herman Robl
Deputy Chief Scientist
U.S. Army Research Office (Durham)
Box 04, Duke Station
Durham, North Carolina 27706

Richard O. Ulsh (CRDARD-1FO)
U.S. Army Research Office (Durham)
Box CM, Duke Station
Durham, North Carolina 27706

Mr Robert O. Parker, ANSEL-RD-8
Executive Secretary, JSTAC
U.S. Army Electronics Command
Fort Monmouth, New Jersey 07703

Commanding General
U.S. Army Electronics Command
Fort Monmouth, New Jersey 07703

Attention: ANSEL-SC
RD-GF
RD-MT
XL-D
XL-E
XL-C
XL-S (Dr R. Buser)
HL-CT-DD
HL-CT-R
1 copy to HL-CT-L (Dr W.S. McAfee)
each sym- HL-CT-O
bol listed HL-CT-I
individually HL-CT-A
ally HL-D
addressed HL-A
NI-P
NI-P-2 (Mr D. Haratz)
NI-R (Mr R. Kulinyi)
NI-S
KI-D
KI-E
KI-S (Dr H. Jacobs)
KI-SM (Mrs Schiel/Hieselmaier)
KI-T
VI-D
VI-P (Mr R.J. Niemela)
WI-D

Dr A.D. Schnitzler, ANSEL-HL-WVII
Night Vision Laboratory, USAECOM
Fort Belvoir, Virginia 22060

Dr G.M. Jamney, ANSEL-HL-NVOR
Night Vision Laboratory, USAECOM
Fort Belvoir, Virginia 22060

Atmospheric Sciences Office
Atmospheric Sciences Laboratory
White Sands Missile Range
New Mexico 88002

Missile Electronic Warfare,
Technical Area, ANSEL-VT-MT
White Sands Missile Range
New Mexico 88002

Project Manager
Common Positioning & Navigation Systems
Attn: Harold H. Bahr (ANCFM-NS-TH), Bldg 439
U.S. Army Electronics Command
Fort Monmouth, New Jersey 07703

Director, Electronic Programs
Attn: Code 427
Department of the Navy
Washington, D.C. 20360

Commander
U.S. Naval Security Group Command
Attn: 643
3901 Nebraska Avenue
Washington, D.C. 20390

Director
Naval Research Laboratory
Washington, D.C. 20390
Attn: Code 2027 6 copies
Dr W.C. Hall, Code 7000 1 copy
Dr A. Brodzinskiy, Sup.Elec Div. 1 copy

Dr G.M.R. Winkler
Director, Time Service Division
U.S. Naval Observatory
Washington, D.C. 20390

Naval Air Systems Command
AIR 03
Washington, D.C. 20360 2 copies

Naval Ship Systems Command
Ship 031
Washington, D.C. 20360

Naval ship Systems Command
Ship 035
Washington, D.C. 20360

U.S. Naval Weapons Laboratory
Dahlgren, Virginia 22448

Naval Electronic Systems Command
ELEX 03, Room 2046 Munitions Building
Department of the Navy
Washington, D.C. 20360 (2 copies)

Commander
Naval Electronics Laboratory Center
Attn: Library
San Diego, California 92152 (2 copies)

Deputy Director and Chief Scientist
Office of Naval Research Branch Office
1030 Est Gree Street
Pasadena, California 91101

Library (Code 2124)
Technical Report Section
Naval Postgraduate School
Monterey, California 93940

Glen A. Myers (Code 52Nv)
Assoc Professor of Elec. Engineering
Naval Postgraduate School
Monterey, California 93940

Commanding Officer and Director
U.S. Naval Underwater Sound Laboratory
Fort Trumbull
New London, Connecticut 06840

Commanding Officer
Naval Avionics Facility
Indianapolis, Indiana 46241

Dr H. Harrison, Code RBE
Chief, Electrophysics Branch
National Aeronautics & Space Admin.
Washington, D.C. 20546

NASA Lewis Research Center
Attn: Library
21000 Brookpark Road
Cleveland, Ohio 44135

Los Alamos Scientific Laboratory
Attn: Report Library
P.O. Box 1663
Los Alamos, New Mexico 87544

Federal Aviation Administration
Attn: Admin Stds Div (MS-110)
800 Independence Ave S.W.
Washington, D.C. 20590

Head, Technical Services Division
Naval Investigative Service Headquarters
4420 North Fairfax Drive
Arlington, Virginia 22203

Commander
U.S. Naval Ordnance Laboratory
Attn: Librarian
White Oak, Maryland 21502 (2 copies)

Commanding Officer
Office of Naval Research Branch Office
Box 39 PFD
New York, New York 09510

Commanding Officer
Office of Naval Research Branch Office
219 South Dearborn Street
Chicago, Illinois 60604

Commanding Officer
Office of Naval Research Branch Office
495 Summer Street
Boston, Massachusetts 02210

Commander (ADL)
Naval Air Development Center
Johnsville, Warminster, Pa 18974

Commanding Officer
Naval Training Device Center
Orlando, Florida 32813

Commander (Code 753)
Naval Weapons Center
Attn: Technical Library
China Lake, California 93555

Commanding Officer
Naval Weapons Center
Corona Laboratories
Attn: Library
Corona, California 91720

Commander, U.S. Naval Missile Center
Point Mugu, California 93041

W.A. Eberspacher, Associate Head
Systems Integration Division
Code 5340A, Box 15
U.S. Naval Missile Center
Point Mugu, California 93041

Mr M. Zane Thornton, Chief
Network Engineering, Communications
and Operations Branch
Lister Hill National Center for
Biomedical Communications
8600 Rockville Pike
Bethesda, Maryland 20014

U.S. Post Office Department
Library - Room 1012
12th & Pennsylvania Ave, N.W.
Washington, D.C. 20260

Director
Research Laboratory of Electronics
Massachusetts Institute of Technology
Cambridge, Massachusetts 02139

Mr Jerome Fox, Research Coordinator
Polytechnic Institute of Brooklyn
55 Johnson Street
Brooklyn, New York 11201

Director
Columbia Radiation Laboratory
Columbia University
338 West 120th Street
New York, New York 10027

Director
Coordinated Science Laboratory
University of Illinois
Urbana, Illinois 61801

Director
Stanford Electronics Laboratories
Stanford University
Stanford, California 94305

Director
Microwave Physics Laboratory
Stanford University
Stanford, California 94305

Director, Electronics Research Laboratory
University of California
Berkeley, California 94720

Director
Electronic Sciences Laboratory
University of Southern California
Los Angeles, California 90007

Director
Electronics Research Center
The University of Texas at Austin
Austin Texas 78712

Division of Engineering and Applied Physics
210 Pierce Hall
Harvard University
Cambridge, Massachusetts 02138

Dr G.J. Murphy
The Technological Institute
Northwestern University
Evanston, Illinois 60201

Dr John C. Hancock, Head
School of Electrical Engineering
Purdue University
Lafayette, Indiana 47907

Dept of Electrical Engineering
Texas Technological College
Lubbock, Texas 79409

Aerospace Corporation
P.O. Box 95083
Los Angeles, California 90045
Attn: Library Acquisitions Group

Professor Nicholas George
California Inst of Technology
Pasadena, California 91109

Aeronautics Library
Graduate Aeronautical Laboratories
California Institute of Technology
1201 E. California Blvd
Pasadena, California 91109

The John Hopkins University
Applied Physics Laboratory
Attn: Document Librarian
8621 Georgia Avenue
Silver Spring, Maryland 20910

Raytheon Company
Attn: Librarian
Bedford, Massachusetts 01730

Raytheon Company
Research Division Library
28 Seyon Street
Waltham, Massachusetts 02154

Dr Sheldon J. Wells
Electronic Properties Information Center
Mail Station E-175
Hughes Aircraft Company
Culver City, California 90230

Dr Robert E. Fontana
Systems Research Laboratories Inc.
7001 Indian Ripple Road
Dayton, Ohio 45440

Nuclear Instrumentation Group
Bldg 29, Room 101
Lawrence Radiation Laboratory
University of California
Berkeley, California 94720

Sylvania Electronic Systems
Applied Research Laboratory
Attn: Documents Librarian
40 Sylvan Road
Waltham, Massachusetts 02154

Hollander Associates
P.O. Box 2276
Fullerton, California 92633

Illinois Institute of Technology
Dept of Electrical Engineering
Chicago, Illinois 60616

The University of Arizona
Dept of Electrical Engineering
Tucson, Arizona 85721

Utah State University
Dept of Electrical Engineering
Logan, Utah 84321

Case Institute of Technology
Engineering Division
University Circle
Cleveland, Ohio 44106

Hunt Library
Carnegie-Mellon University
Schenley Park
Pittsburgh, Pennsylvania 15213

Dr Leo Youns
Stanford Research Institute
Menlo Park, California 94025

School of Engineering Sciences
Arizona State University
Tempe, Arizona 85281

Engineering & Mathematical Sciences Library
University of California at Los Angeles
605 Hilgard Avenue
Los Angeles, California 90024

The Library
Government Publications Section
University of California
Santa Barbara, California 93106

Carnegie Institute of Technology
Electrical Engineering Department
Pittsburgh, Pennsylvania 15213

Professor Joseph E. Rose
Chairman, Dept of Electrical Engineering
The University of Michigan
Ann Arbor, Michigan 48104

New York University
College of Engineering
New York, New York 10019

Syracuse University
Dept of Electrical Engineering
Syracuse, New York 13210

Yale University
Engineering Department
New Haven, Connecticut 06520

Airborne Instruments Laboratory
Deerpark, New York 11729

Raytheon Company
Attn: Librarian
Bedford, Massachusetts 01730

Lincoln Laboratory
Massachusetts Institute of Technology
Lexington, Massachusetts 02173

The University of Iowa
The University Libraries
Iowa City, Iowa 52240

Lenkurt Electric Co, Inc
1105 County Road
San Carlos, California 94070
Attn: Mr E.K. Peterson

Philco Ford Corporation
Communications & Electronics Div.
Union Meeting and Jolly Roads
Blue Bell, Pennsylvania 19422

Union Carbide Corporation
Electronic Division
P.O. Box 1209
Mountain View, California 94041

Electromagnetic Compatibility Analysis Center
(ECAC), Attn: ACLP
North Severn
Annapolis, Maryland 21402

Director
U. S. Army Advanced Materiel Concepts Agency
Washington, D.C. 20315

Dept of Electrical Engineering
Rice University
Houston, Texas 77001

Research Laboratories for the Eng. Sc.
School of Engineering & Applied Science
University of Virginia
Charlottesville, Virginia 22903

Dept of Electrical Engineering
College of Engineering & Technology
Ohio University
Athens, Ohio 45701

Project Mac
Document Room
Massachusetts Institute of Technology
545 Technology Square
Cambridge, Massachusetts 02139

Lehigh University
Dept of Electrical Engineering
Bethelam, Pennsylvania 18015

Commander Test Command (TCD-)
Defense Atomic Support Agency
Sandia Base
Albuquerque, New Mexico 87115

Materials Center Reading Room 13-2137
Massachusetts Institute of Technology
Cambridge, Massachusetts 02139

Professor James A. Cadzow
Department of Electrical Engineering
State University of New York at Buffalo
Buffalo, New York 14214

Director, Naval Research Laboratory
Attn: Library, Code 2029 (ONRL)
Washington, D.C. 20390

Commanding Officer (Code 2064)
Navy Underwater Sound Laboratory
Fort Trumbull
New London, Connecticut 06320

ERRATUM

Mr Jerome Fox, Research Coordinator
Polytechnic Institute of Brooklyn
55 Johnson St (should be 333 Jay St)
Brooklyn, N.Y. 11201

DELETE

Mr Morton M. Pavane, Chief
AFSC Scientific & Tech. Liaison Office
26 Federal Plaza, Suite 1313
New York, New York 10007

Commanding Officer
Office of Naval Research Branch Office
Box 39 FPO
New York, N.Y. 09510

DOCUMENT CONTROL DATA - R & D

(Security classification of title, body of abstract and indexing annotation must be entered when the overall report is classified)

1. ORIGINATING ACTIVITY (Corporate author) University of Illinois Coordinated Science Laboratory Urbana, Illinois 61801		2a. REPORT SECURITY CLASSIFICATION	
		2b. GROUP	
3. REPORT TITLE SEGMENT METHOD FOR THE CONTROL OF SYSTEMS IN THE PRESENCE OF UNCERTAINTY			
4. DESCRIPTIVE NOTES (Type of report and inclusive dates)			
5. AUTHOR(S) (First name, middle initial, last name) MEDANIC, J. & CHEN, Cheng-I			
6. REPORT DATE November, 1969		7a. TOTAL NO. OF PAGES 25	7b. NO. OF REFS 14
8a. CONTRACT OR GRANT NO. DAAB -07-67-C-0199; also AFOSR 931-67;		9a. ORIGINATOR'S REPORT NUMBER(S) R-442	
b. PROJECT NO. AFOSR 68-1579 and by NSF GK-3893.		9b. OTHER REPORT NO(S) (Any other numbers that may be assigned this report)	
c.			
d.			
10. DISTRIBUTION STATEMENT This document has been approved for public release and sale; its distribution is unlimited.			
11. SUPPLEMENTARY NOTES		12. SPONSORING MILITARY ACTIVITY Joint Services Electronics Program thru U.S. Army Electronics Command Fort Monmouth, New Jersey 07703	
13. ABSTRACT This paper develops the segment method for the problem of control with uncertainty. It is shown that due to disturbances a segment of values of the performance functional is associated with each admissible control. It is proposed that the optimal control for the problem with uncertainty be defined on the basis of an ordering of associated segments. The ordering is achieved by introducing a segment index. Two particular segments, the pessimistic and the optimistic segment, are shown to be important in the analysis of the problem while minimization of the introduced segment index is the basic synthesis problem. The approach is applied to a linear control problem where uncertainty is caused by external disturbances and specific results are presented. They show the optimal control to retain the advantages of the usual, minimax type designs and ensure better system performance for disturbances other than the worst.			

14

KEY WORDS

Near Optimal Control
 Segment Method
 Uncertainty
 Minimax Design
 Sensitivity

LINK A

LINK B

LINK C

ROLE

WT

ROLE

WT

ROLE

WT