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Effects of span-to-depth ratios on moment connection damage evolution under catenary action

Abstract

This paper proposes an improved method for determining the gravity resistance of a moment resisting beam-column assembly following an interior column loss. The proposed method accounts for the connection's damage evolution and for the catenary mechanism developed by the assembly as it deflects downward. Through a full-scale laboratory test and finite element simulations, the complete responses of moment resisting beam-column assemblies including the connection's damage evolution are investigated under different beam span-to-depth ratios. The welded unreinforced flange-bolted web (WUF-BW) connection method is used for its robustness in developing the catenary action. It is found that, under the same span-to-depth ratio, beam-column assemblies exhibit similar normalized load-rotation relationships, even with different beam depths. The assembly with a larger span-to-depth ratio is able to develop the gravity resistance earlier, and provides a higher ultimate resistance by developing a more effective catenary mechanism. On the other hand, the assembly with a smaller span-to-depth ratio exhibits a more ductile response. A simplified curve model of the gravity resistance development of a moment beam-column assembly with damage evolution has been proposed for a convenient assessment of the progressive collapse resistance following a central column loss.

Disciplines

Engineering | Science and Technology Studies

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1 **Effects of Span-to-Depth Ratios on Damage Evolution of Moment Connections** 2 **in Column Removal Scenario**

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7 **Abstract:** This paper proposes an improved method for determining the gravity resistance of a moment
8 resisting beam-column assembly following an interior column loss. The proposed method accounts for the
9 connection's damage evolution and for the catenary mechanism developed by the assembly as it deflects
10 downward. Through a full-scale laboratory test and finite element simulations, the complete responses of
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16 resistance earlier, and provides a higher ultimate resistance by developing a more effective catenary
17 mechanism. On the other hand, the assembly with a smaller span-to-depth ratio exhibits a more ductile
18 response. A simplified curve model of the gravity resistance development of a moment beam-column assembly
19 with damage evolution has been proposed for a convenient assessment of the progressive collapse resistance
20 following a central column loss.

21 **Keywords:** progressive collapse; span-to-depth ratio; steel moment connection; gravity resistance
22 development; damage evolution; catenary mechanism; flexural mechanism.

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26 **1. Introduction**

27 There have been several guidelines [1-3] for the progressive collapse design and analysis
28 of building structures under extreme or abnormal load, all of which employ basically the
29 same principles and analysis methods. According to UFC 4-023-03 “Design of building to
30 resist progressive collapse” [2], a progressive collapse design may use different methods
31 depending on the occupancy category of the building, including the Tie Force (TF) method
32 for the entire structure, the Alternate Path (AP) method and the Enhanced Local Resistance
33 (ELR) method for some specific structure regions.

34 The Alternate Path method [4], as both the design and the analysis methods, is the most
35 popular for the study of progressive collapse prevention [1-3]. A structure must be able to
36 bridge over vertical load-carrying elements notionally removed from itself by satisfying the
37 requirements of the Alternate Path method, otherwise it must be re-designed or retrofitted to
38 increase the structural bridging capacity [2, 3]. In this method, any further failure of
39 structural components (connections, beams and columns) following the notional column
40 removal is prevented by ensuring the components meet certain criteria for various building
41 materials including reinforced concrete, structural steel, masonry and wood [2, 3].

42 It has been found [5-10] that the structural bridging capacity depends on the performance
43 of the connections. There have been a number of experimental tests and numerical
44 simulations focusing on the behaviour of various connections [11-21] following an interior
45 column loss. The moment connections were found to work firstly by flexural action and later
46 by catenary action [6, 14, 15, 18-20]. It was found [15, 18-21] that a steel moment
47 connection usually acquires a meaningful contribution to the gravity resistance from the

48 catenary mechanism at chord rotations greater than 0.03 radians.

49 When the nonlinear static analysis procedure is employed, nominally rigid moment
50 connections must deform within the prescribed deformation limits so as to meet the
51 acceptance criteria [2]. The acceptance criteria for moment connections are given in terms of
52 the plastic rotation, whose values for a primary component correspond to its plastic
53 deformation limit prior to capacity degradation [2-3]. Moment connections are permitted to
54 deform within a small range of plastic rotations, below 0.025 radians for the typical
55 “improved welded unreinforced flange-bolted web” (WUF-BW) connection [2, 3], which
56 does not allow any significant catenary action to be developed [15, 18-22]. However, the
57 capacity degradation does not usually occur until a much larger rotation, typically greater
58 than 0.06 radians [15, 18-21].

59 In traditional seismic structural designs, the occurrence of fracture signifies the ultimate
60 limit state of a moment connection due to the loss of its flexural capacity. However, in an
61 interior column removal scenario, catenary action can still be developed by the tensioning of
62 the connected beam members under large deflection following fracture, provided the
63 connections are designed appropriately [18-22]. Two types of moment connection failure
64 modes, being the beam-end interrupted failure mode and the column-wall failure mode,
65 have been identified [18-20] as being able to allow the assembly to obtain a higher gravity
66 resistance (from the catenary mechanism) in the post-fracture stage than its previous peak
67 resistance (under the flexural mechanism). It is therefore rational to explore new design
68 criteria that take advantage of the catenary mechanism that develops following an interior
69 column loss.

70 Among the various levels of sub-structure idealisation in the simplified framework
71 proposed by Izzuddin et al. [23] for multi-storey buildings, the double-span beam-column
72 assembly within the bays above the lost column is the lowest level of sub-structure whose
73 response is used for composing the higher level sub-structures. The beam's span-to-depth
74 ratio has been found to significantly affect the response of the double-span beam-column
75 assembly following the column removal [24-27]. However, these investigations did not
76 account for the damage evolution of the beam-to-column connections.

77 In this paper, the complete responses of the moment resisting beam-column assemblies
78 under the column removal scenario are investigated. The welded unreinforced flange-bolted
79 web (WUF-BW) is used to connect the beams and the column as such a connection
80 facilitates the development of the catenary mechanism following an initial fracture. The
81 development of the assembly's gravity resistance in the post-fracture stage and the effects of
82 the span-to-depth ratio are studied in detail.

83 A full-scale laboratory test is conducted where a pushdown action at the central column is
84 applied in order to simulate the column removal scenario. The test results are used to verify
85 the refined finite element model incorporating material fracture, which is employed in
86 subsequent parametric analyses of the effects of the beam span-to-depth ratio on the gravity
87 resistance of the beam-column assemblies. Based on the parametric analyses results, an
88 improved development model will be proposed for the structural gravity resistance taking
89 into account the damage evolution of the connection region.

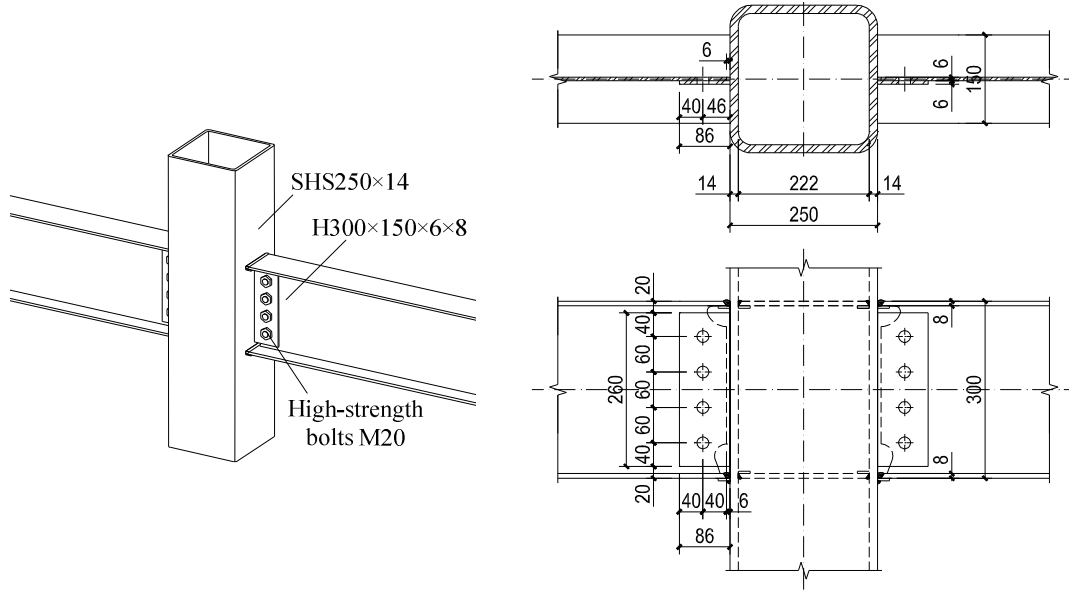
90 **2. Full-scale laboratory test**

91 **2.1. Test specimen**

92 Due to its robustness during the beam-end interrupted failure and column-wall failure
93 under a central column removal scenario [18, 19], the welded unreinforced flange-bolted
94 web (WUF-BW) connection was used for the test specimen whose details are given in Fig. 1.
95 The double-span assembly consisted of two I-section beams (H300×150×6×8) and a square
96 hollow section column (SHS250×14) with two inner diaphragms (thickness $t = 8\text{mm}$) at
97 locations corresponding to the top and the bottom flanges of the beam, as illustrated in Fig. 1
98 (b).

99 The flanges of the beam and the inner diaphragms were joined to the column wall using
100 complete joint penetration (CJP) groove welds, and weld access holes of the beam were cut
101 from the beam web in accordance with the standard recommendation [28]. The beam webs
102 were bolted to the shear tab welded to the column via four M20 Grade 10.9 frictional type
103 high-strength bolts arranged in one vertical row. The tightening torque applied on the bolts
104 was 440 N-m according to standard requirements [29]. All the contact surfaces were treated
105 with sand blasting. The measured material properties of the specimen are summarized in
106 Table 1.

107



108

(a) Components.

(b) Dimensions (in mm).

109

Fig. 1. Details of the WUF-BW connection.

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Table 1. Material properties of test specimen.

Components	Yield strength	Tensile strength
	f_y (MPa)	f_u (MPa)
Plate of SHS250×14	410	655
Corner of SHS 250×14	415	750
Beam flange ($t_f = 8$ mm)	400	670
Beam web ($t_w = 6$ mm)	405	640

111

The Beam-Joint-Beam (B-J-B) assembly [18] was employed for the specimen, as

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illustrated in Fig. 2 (a). A relatively small span of the beam $l_0 = 2400$ mm was used, giving a

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gross span-to-depth ratio of $l_0/H = 8$, in order to obtain the complete response of the

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beam-to-column connection including the damage evolution since there was a limited

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vertical displacement range (approximately 400 mm). The length of the central column was

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1100 mm. The design of beam-column assembly was based on the strong column-weak

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beam seismic design philosophy according to Chinese codes [30, 31].

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2.2. Test setup and instrumentation

119

The test specimen, mounted on a purpose-built test rig as illustrated in Fig. 2 (b), was

120 loaded vertically at the unsupported central column by the actuator at a stroke rate less than
 121 7 mm/min. The central column was guided at the bottom end using a sliding support so that
 122 only vertical movement of the column is possible. The two pin supports at the outer ends of
 123 the beams were designed using latch-type rollers for free rotation in the assembly plane,
 124 with their distance matching the span of 2,400 mm. The test was terminated when the
 125 connection totally lost its bearing capacity on either side.

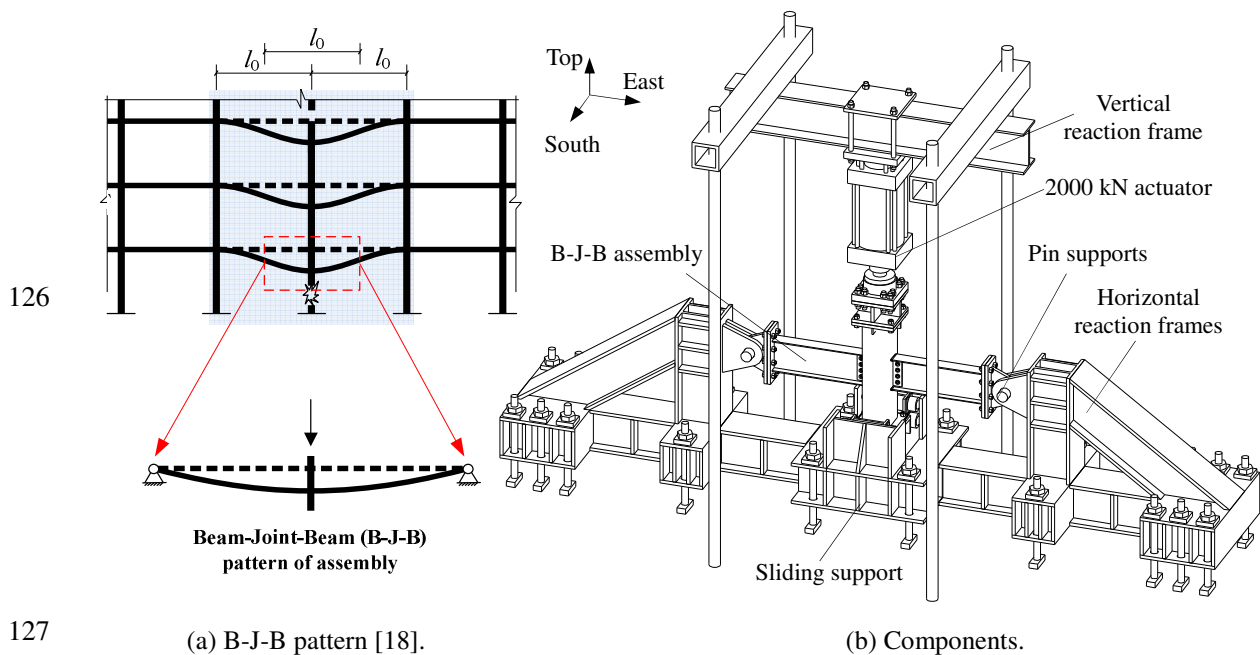
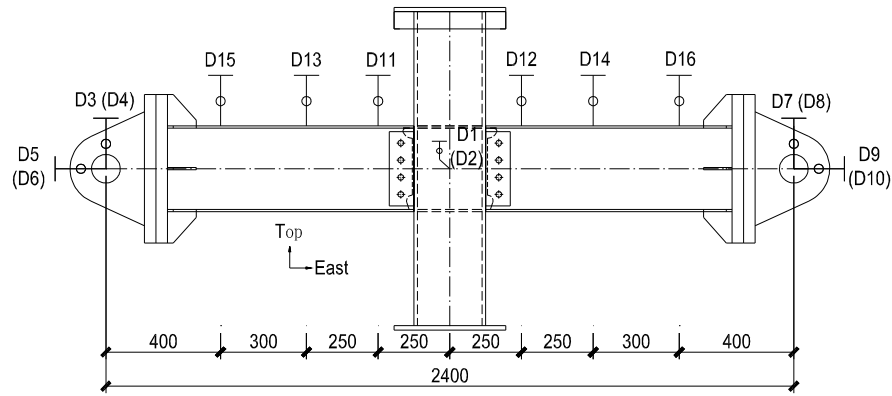


Fig. 2. Test setup.

129 Instrumentations were arranged as shown in Fig. 3 to measure the displacement of the
 130 assembly and strains at the critical regions during the test. Sixteen displacement transducers
 131 (see Fig. 3 (a)) were used to measure the assembly deflection along the beam length and any
 132 possible movements of the two pin supports. Strain gauges were arranged at six beam
 133 sections as shown in Fig. 3 (b).

134

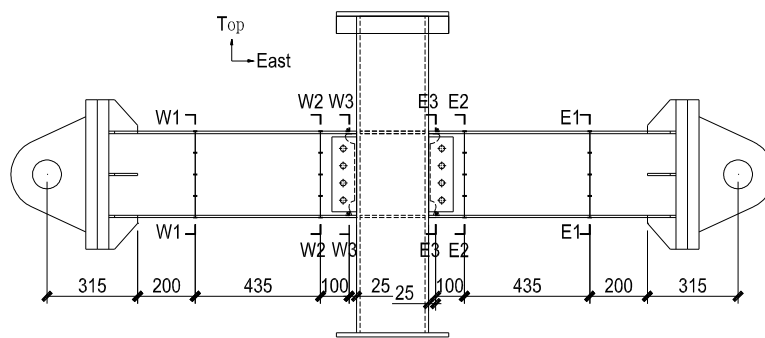


135

(a) Displacement transducers.

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137



138

(b) Beam sections for strain gauges.

139

Fig. 3. Schematic arrangements of test instrumentation.

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2.3. Test results

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The tested specimen exhibited a complete failure process at the beam-to-column connection, where the beam on the east side totally separated from the central column. The final condition of the beam-column assembly and the detailed view of the WUF-WB connection at the end of the test are shown in Fig. 4.

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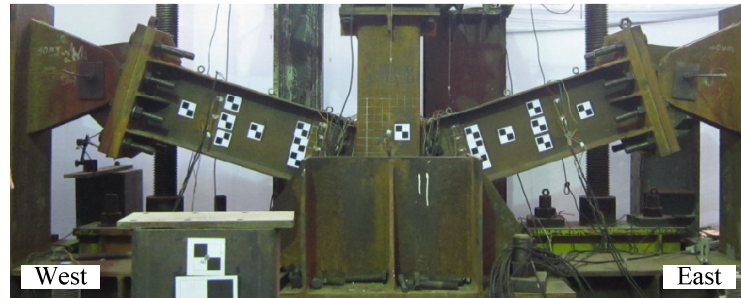
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The load-displacement curve of the central column is shown in Fig. 5. A few key stages are identified on the curve, and the associated damage evolutions are depicted in the corresponding photographs in Fig. 6. The nominal plastic load F_p is the vertical load causing the formation of plastic hinges at the critical locations (Sections W3 and E3), which is 359 kN. The beam chord rotation θ is obtained by dividing the vertical displacement of the

150 central column by the distance of 1,200 mm between the column and the pin support
 151 (effectively the half-span length).

152



153

(a) Southern view of the test assembly

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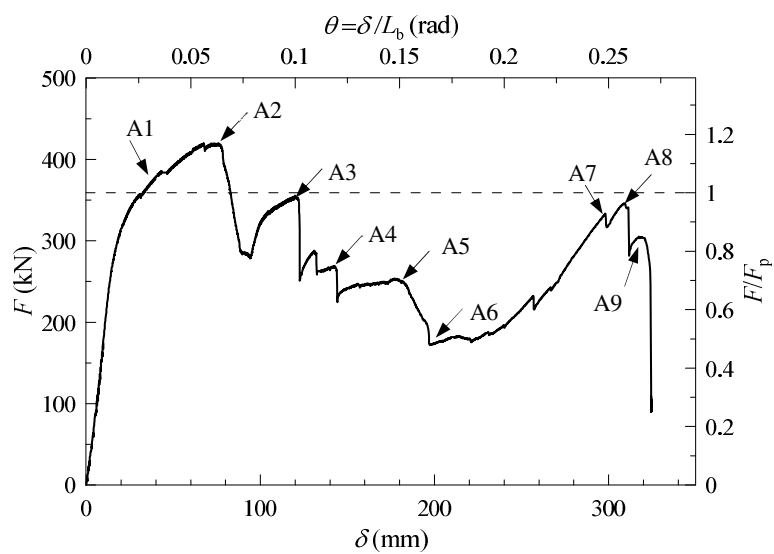
(b) Northern view of the tested WUF-WB connection

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Fig. 4. Photographs of the specimen at the end of the test.



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Fig. 5. Load-displacement curve of test specimen.

 <p>A1: Local buckling of top flanges near Sections W3 and E3</p>	 <p>A2: Bottom flange fractured at Section E3 (1.17F_p, 0.061 rad)</p>	 <p>A3: Column wall cracked near the southern end of bottom flange on the west side (0.99F_p, 0.100 rad)</p>
 <p>A4: The lowest bolt was torn out of web on the east side (0.75F_p, 0.120 rad)</p>	 <p>A5: Column wall cracked near the northern end of bottom flange on the west side (0.70F_p, 0.150 rad)</p>	 <p>A6: Column wall completely fractured near the bottom flange and cracks entended upwards on the west side (0.48F_p, 0.164 rad)</p>
 <p>A7 : Shear tab fractured at the middle and top parts across the bolt holes on the east side (0.92F_p, 0.248 rad)</p>	 <p>A8: Column wall cracked along the weld between the shear tab and column on the west side (0.96 F_p, 0.259 rad)</p>	 <p>A9: Top flange of Section E3 fracture and the eastern beam totally separated from the column (0.84 F_p, 0.268 rad)</p>

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Fig. 6. Damage evolutions at key stages of test specimen.

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As demonstrated in Fig. 5 and Fig. 6, the first significant event (point “A1” on the load-displacement curve) took place when local buckling occurred at the top flanges near Sections W3 and E3 with the displacement reaching about 40 mm, which corresponded to the beam chord rotation θ of 0.033 rad.

The applied load kept increasing until the specimen reached the first peak load (point “A2”) when the bottom flange near the access hole at Section E3 fractured at a displacement

169 of 73 mm ($\theta = 0.061$ rad). The fracture caused a drastic drop of the applied load from the
170 peak value of 419 kN ($1.17F_p$) to 281 kN ($0.78F_p$).

171 However, the flexural capacity of the beam on the other side (west side) enabled the
172 applied load to reach a second peak value of 355 kN ($0.99F_p$) at a displacement of 120 mm
173 ($\theta = 0.100$ rad), when the column wall fractured near the southern end of the bottom flange
174 on the west side (point “A3”), which induced an abrupt drop of load to about 250 kN
175 ($0.70F_p$).

176 With the increasing displacement of the central column, the specimen saw two small
177 fluctuations of the applied load from the peak value of 287 kN ($0.80 F_p$) at a displacement of
178 132 mm ($\theta = 0.110$ rad) and from 268 kN ($0.75F_p$) at 144 mm ($\theta = 0.120$ rad). In the latter
179 event, the load suddenly reduced to about 250 kN ($0.70F_p$) due to the tear-out of the lowest
180 bolt on the east side out of the web (point “A4”).

181 When the displacement reached 180 mm ($\theta = 0.150$ rad), the column wall fractured near
182 the northern end of the bottom flange on the west side (point “A5”), after which the load
183 decreased due to the crack propagation across the entire width of the bottom flange on the
184 west side, until a complete fracture through its thickness formed below the bottom flange
185 (point “A6”). The displacement at this point was 197 mm ($\theta = 0.164$ rad) and the load
186 reached the lowest value of 172 kN ($0.48 F_p$).

187 Thereafter the west-side column wall tore up from the two ends of the bottom flange as
188 the applied load gradually recovered, on account of the development of the catenary
189 mechanism. At a displacement of 298 mm ($\theta = 0.248$ rad), the shear tab fractured vertically
190 at the middle and top parts through the bolt holes on the east side (point “A7”) following the

191 horizontal crack below the third bolt, causing a slight drop in the applied load from 330 kN
192 ($0.92 F_p$) to 317 kN ($0.88 F_p$).

193 The load quickly increased and reached another peak value of 345 kN ($0.96 F_p$) at a
194 displacement of 311 mm ($\theta = 0.259$ rad) when the column wall fractured along the weld
195 connecting the shear tab and column on the west side (point “A8”) with an abrupt drop of
196 load to 280 kN ($0.78 F_p$). Although the load was able to slightly pick up to 303 kN ($0.84 F_p$),
197 the beam-column assembly virtually lost its bearing capacity due to the complete fracture of
198 the top flange at Section E3 and hence the separation between the eastern beam and the
199 column (point “A9”). At this point, the displacement of central column was 321 mm ($\theta =$
200 0.268 rad) and the test was terminated.

201 Two failure modes, the interrupted beam-end failure mode and the column-wall failure
202 mode [19], took place during the test. A complete process of the interrupted beam-end
203 failure mode covering the entire damage evolution was present for the WUF-BW connection
204 on the east side. The fracture took place initially at the bottom flange, then at the bottom of
205 the web and the middle-top part of the shear tab, and eventually at the top flange. The
206 fracture of the top flange signified the end of the damage evolution on this side.

207 On the other side (west side), the column-wall failure mode did not present a complete
208 damage evolution, with the cracks extending upwards to one third of the beam’s depth. As
209 discussed in previous papers [19, 20], the occurrence of fracture at the column wall was
210 preceded by the separation between the inner diaphragm and the column inside wall as
211 shown in Fig. 7 (a).

212



213

(a) Separation on the west side.



214

(b) Well fabricated welding on the east side.

Fig. 7. Final conditions between the bottom inner diaphragm and column inside wall.

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3. Verification of numerical simulations

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Numerical analyses were carried out using the explicit time integration approach in the general-purpose finite element (FE) analysis software ABAQUS [32]. The verification of the FE model was firstly made by comparing the FE simulation results against the present laboratory test results. The verified FE analysis method was subsequently used in parametric analyses for studying the performance of moment resisting beam-column assemblies under different span-to-depth ratios.

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3.1. FE modelling of test specimen

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The present test assembly was modelled in whole to enable the simulation of the asymmetric damage evolutions on the two sides of the WUF-BW connection. The actuator's load was simulated by a prescribed vertical displacement of the central column. The geometric, boundary and material nonlinearities including material fracture were taken account into the FE simulation. The stress-strain constitutive relationships of the steel material were defined based on the coupon test results (see Table 1).

229

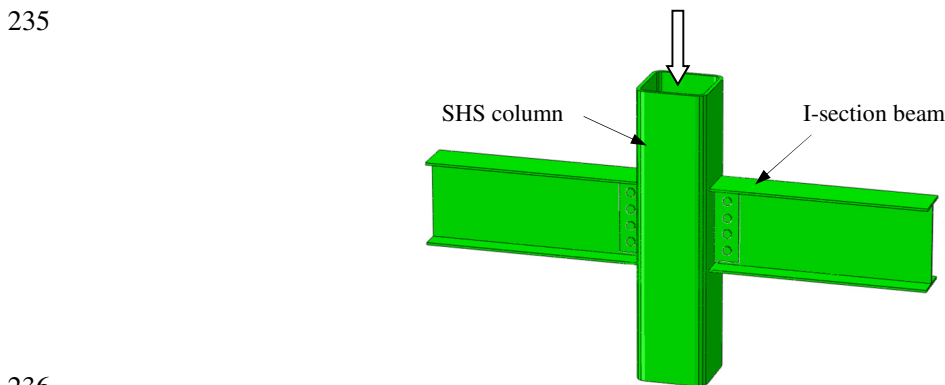
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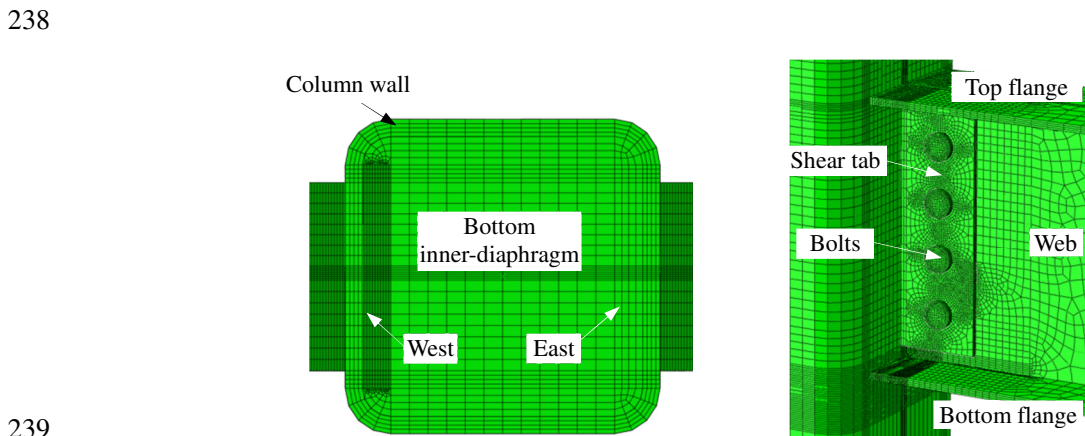
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All components were created using solid elements of the 8-node linear brick elements with reduced integration (C3D8R). In order to capture the fracture at the connection region, sufficiently fine mesh of solid elements was employed at the parts where fracture may occur, with an element size of approximately 1.0 mm, as shown in Fig. 8 (b), including the

233 I-section at the beam end segment together with the bolted shear tab on the east side, and the
234 bottom inner diaphragm together with the connected column wall on the west side.



236
237 (a) FE model of the test assembly.



239
240 (b) Meshes in the connection region.

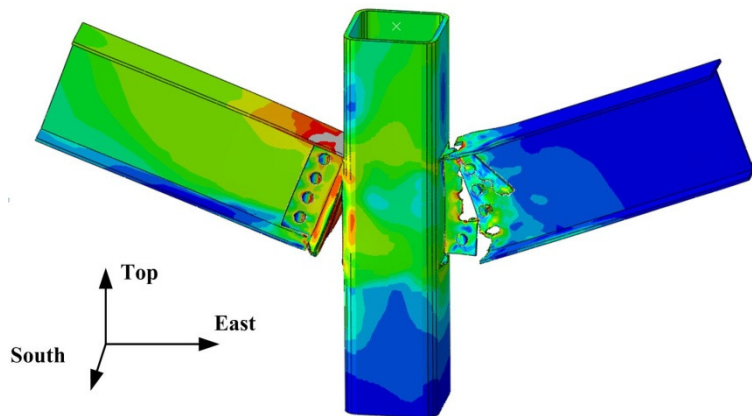
241 Fig. 8. Finite element models composed of solid elements.

242 The “Damage for Ductile Metals” approach was employed to activate the deletion of
243 elements whose strain responses reach the pre-specified fracture threshold [32]. The fracture
244 strain limits ranged from 0.2 to 0.8 for the column wall, I-section (bottom and top flanges
245 and web) and the shear tabs surrounding the bolts. In order to simulate the column-wall
246 failure mode on the west side, i.e. the separation between the bottom inner diaphragm and
247 the column inside wall, the western edge of the bottom inner diaphragm was given a
248 relatively small fracture strain limit of 0.04.

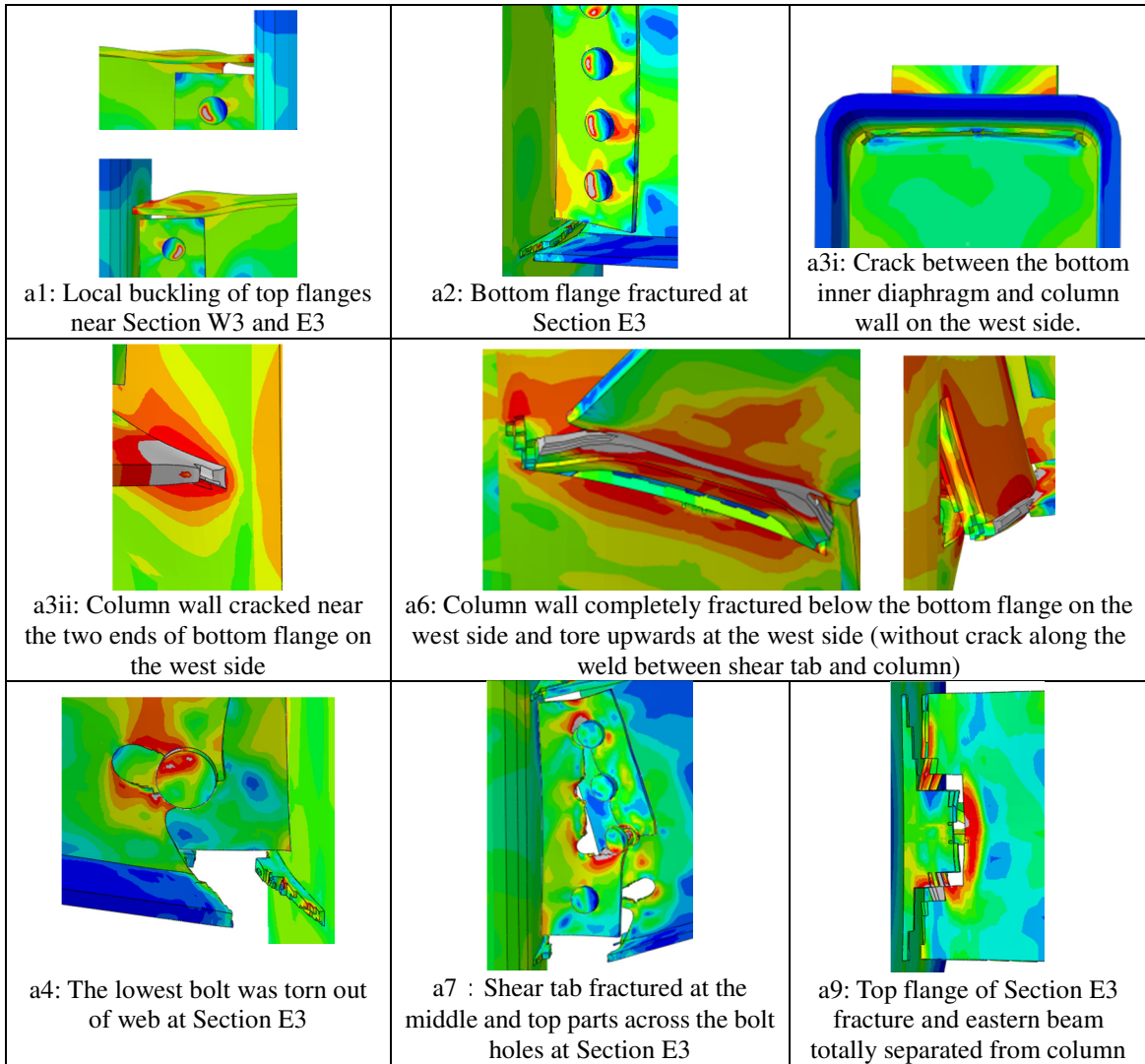
249 **3.2. Simulation results**

250 The final state of the test specimen in the FE simulation is shown in Fig. 9 (a), involving a
251 beam-end interrupted failure at Section E3 on the east side and a column-wall failure on the
252 west side. The key stages in the simulated failure process shown in Fig. 9 (b) agreed
253 reasonably well with the experimental results presented earlier in Fig. 6, and are labelled in
254 the same manner with respect to the fracture mode as the experimental key stages using the
255 lower case “a” in lieu of the upper case. The numerals for the simulated key stages are not
256 always consecutive, indicating that the sequence of fractures do not necessarily match the
257 experimental sequence.

258 The FE load-displacement curve is compared against the experimental curve in Fig. 10,
259 with the indicated key events corresponding to Fig. 6 and Fig. 9 (b). The comparison shows
260 a reasonable agreement between the two sets of data in terms of the load development and
261 the damage evolution.



263 (a) Final state of the beam-column assembly.
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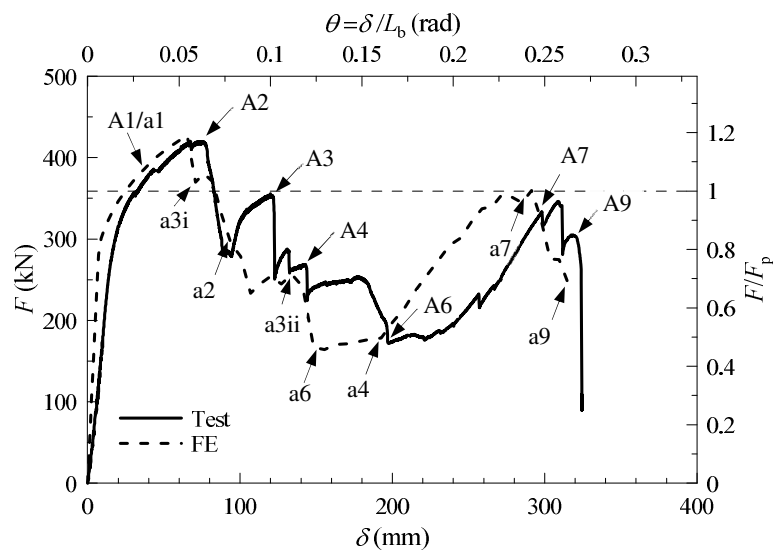
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(b) Key stages in the failure process

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Fig. 9. Simulated failure modes.

267



268

Fig. 10. Comparison of load-displacement curves between FE simulation and test for specimen.

269 **4. Parametric analyses on span-to-depth ratios**

270 In this section, thirty-two double-span beam-column assemblies of four different
271 configurations shown in Table 2 were analysed under varying span-to-depth ratios. The
272 fourth configuration in the table is the same as that of the test specimen depicted in Fig. 1. As
273 can be seen from the table, all connections are of the WUF-BW type.

274 Due to symmetry, only one half of each assembly was modelled. Four span-to-depth
275 ratios(R) of 18, 15, 12 and 8 were employed in the parametric analyses, which cover the
276 commonly used range in design codes [33]. The beam-end interrupted failure mode and the
277 column-wall failure mode were separately simulated (refer to Section 3.1). The label of each
278 specimen indicates its span-to-depth ratio, failure mode (“BF” or “CF”) and beam depth, in
279 that order. The “BF” designation refers to the beam-end interrupted failure mode, and the
280 “CF” designation refers to the column wall failure mode. For example, Specimen
281 R18-BF-H600 is the beam-column assembly with a span-to-depth ratio R of 18, composed
282 of beam section H600×300×12×20 connected to column section SHS 500×25 by M30×10
283 bolts (see Table 2), and fails by the beam-end interrupted failure mode.

284 Table 2. Components of four groups for beam-column assemblies in the parametric analyses.

Beam section	Column section	WUF-BW connection
H600×300×12×20	SHS 500×25	M30×10
H500×200×9×14	SHS 400×20	M24×10
H400×200×7×9	SHS 300×16	M24×8
H300×150×6×8	SHS 250×14	M20×4

285 **4.1. Assemblies having the same span-to-depth ratio**

286 As explained in [27], a normalized chord rotation over the plastic hinge rotation θ_p is
287 more appropriate to use as the generalized displacement variable for the purpose of

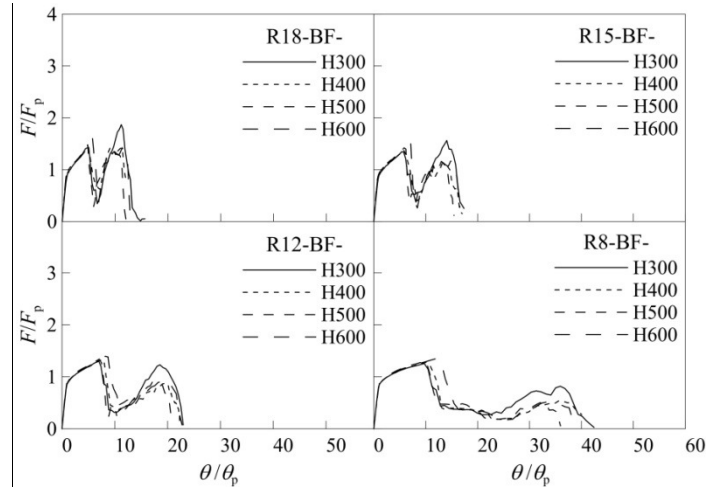
288 comparing the progressive collapse resistance performance between double-span moment
 289 resisting assemblies. The plastic hinge rotation θ_p is defined [27] as

$$290 \quad \theta_p = \frac{\delta_p}{l_0/2} = \frac{F_p}{K_e} \frac{2}{l_0} = \frac{4W_p f_y / l_0}{48EI_b} \frac{2}{l_0} = \frac{W_p f_y l_0}{6EI_b} \quad (1)$$

291 where K_e is the elastic stiffness of a simply supported beam under a concentrated force at
 292 midspan, and I_b is the second moment of area of the beam section.

293 The normalized load-rotation curves of the assemblies having different beam depths but
 294 the same span-to-depth ratio are shown in Fig. 11 (a) and (b), corresponding to the beam-end
 295 interrupted failure mode and the column-wall failure mode, respectively. It can be seen that
 296 the different assemblies behave similarly to each other if their span-to-depth ratios and
 297 failure modes are the same, irrespective of their beam depths. The slight differences in the
 298 post-fracture stage of the beam-end interrupted failure mode are mostly caused by the
 299 different connection geometry (see Table 2). For the column-wall failure mode, the different
 300 capacities of the column-wall (thickness) of the assemblies relative to their respective beam
 301 section's plastic capacities may lead to some differences in their progressive collapse
 302 behaviour. However, such differences are much smaller than those between the assemblies
 303 having different span-to-depth ratios, as demonstrated in the following subsection.

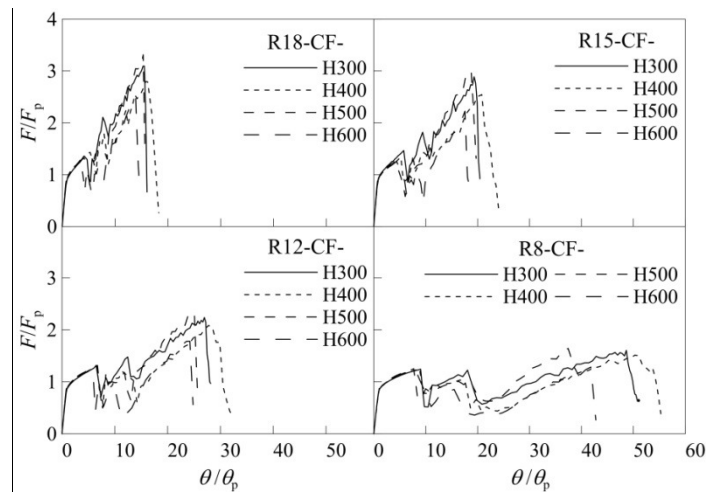
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305

(a) Beam-end interrupted failure mode

306



307

(b) Column-wall failure mode

308

Fig. 11. Normalized load-displacement curves for assemblies having the same span-to-depth ratio.

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4.2. Assemblies having different span-to-depth ratios

310

In order to study the effects of span-to-depth ratio, the normalized load-rotation curves

311

of assemblies configured with H300×150×6×8 beam under different span-to-depth ratios

312

are compared to each other in Fig. 12 and Fig. 13, for the beam-end interrupted failure mode

313

and the column-wall failure mode, respectively. Certain key stages of the damage evolution

314

are identified on the curves and depicted in the accompanying figures of FE simulation. As

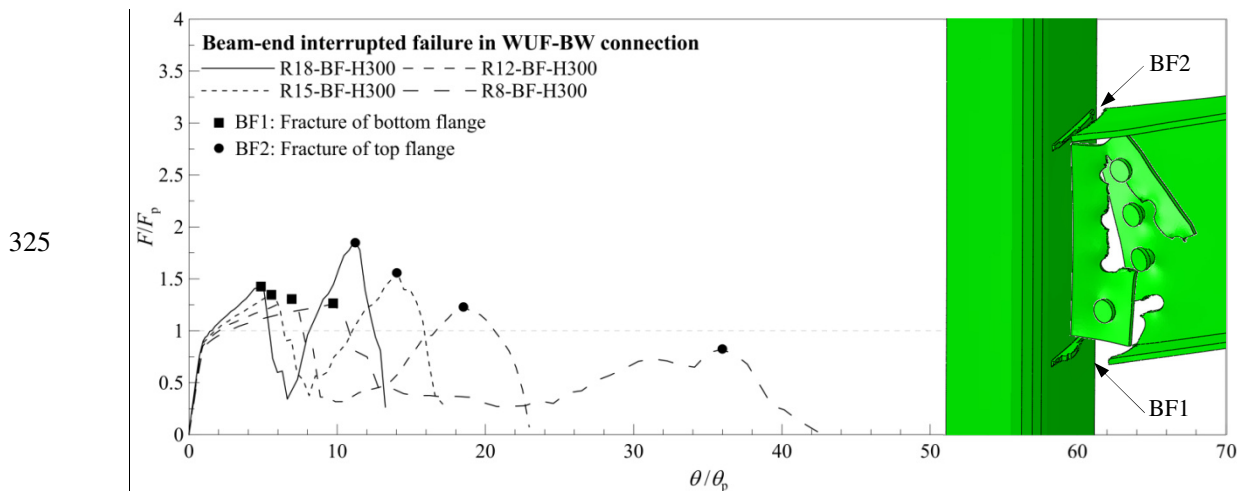
315

shown in Fig. 12, each assembly experiencing the beam-end interrupted failure mode has

316

two peak resistances associated with fractures of the bottom and the top flanges. The bottom

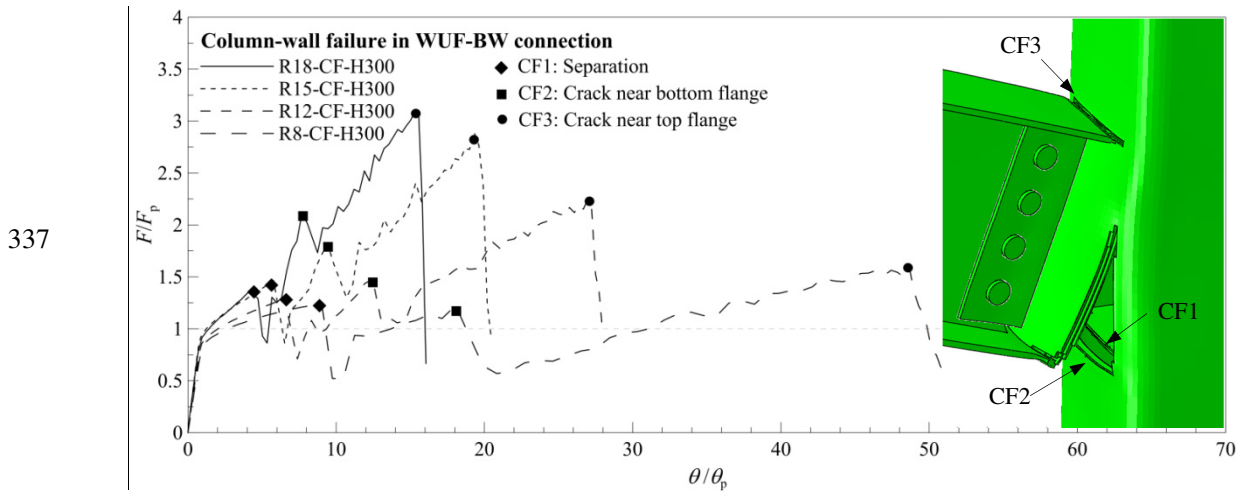
317 flanges (Step “BF1” in Fig. 12) fracture when the resistances are equal to $1.25F_p$ to $1.43F_p$ at
 318 normalized chord rotations θ/θ_p ranging from 4 to 10 (at an approximately constant chord
 319 rotation of 0.07 rad), each of which is followed by a drop in the resistance to about half the
 320 plastic hinge load F_p . The resistance then recovers on account of the interaction between the
 321 bolts and the web as well as the shear tab, before it is eventually lost when the top flange
 322 fractures (Step “BF2”). The second peak value of F/F_p range from 0.8 to 1.8, reached at θ/θ_p
 323 ranging from 11 to 36. The smaller the span-to-depth ratio, the lower the peak resistance and
 324 the larger the normalized rotation demand.



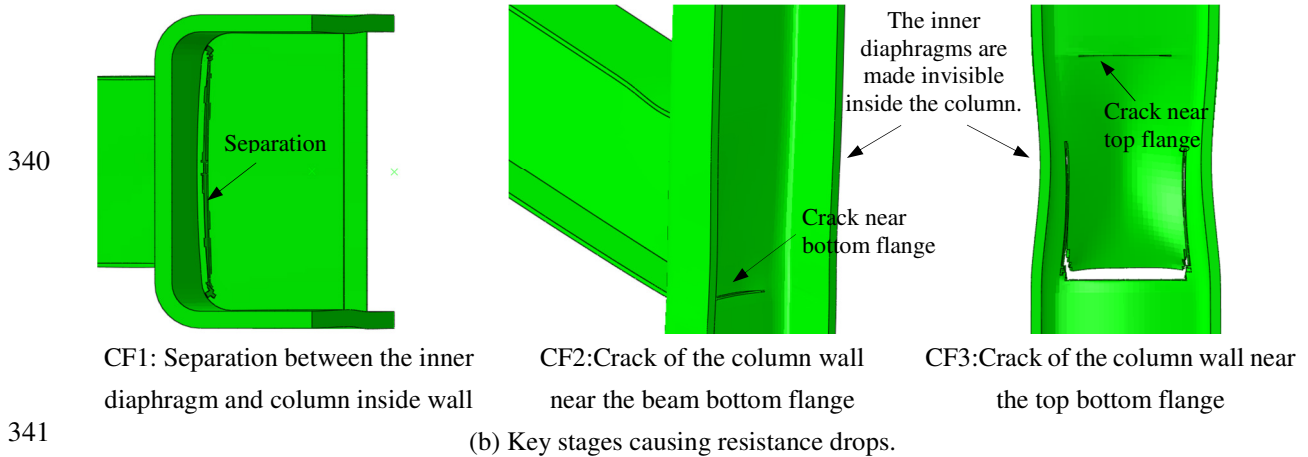
326 Fig. 12. Responses of assemblies having different span-to-depth ratios experiencing beam-end
 327 interrupted failure.

328 Fig. 13 (a) shows that, for each of the four assemblies undergoing the column-wall failure
 329 mode, the resistance quickly recovers after the first two interruptions, and the peak
 330 resistances generally exhibit an increasing trend. The first two interruptions are due to the
 331 separation between the bottom inner diaphragm and the column wall, and the fracture of the
 332 column wall, respectively, as illustrated in Fig. 13 (b). The resistance is only lost when crack
 333 takes place near the top flange. The maximum normalized resistances F/F_p range from 1.5 to
 334 3.1, reached at θ/θ_p ranging from 16 to 49. As in the case of the assemblies undergoing the

335 beam-end interrupted failure mode, the smaller the span-to-depth ratio, the lower the peak
 336 resistance and the larger the normalized rotation demand.



338 (a) Normalized load - rotation curves.
 339



342 Fig. 13. Responses of assemblies having different span-to-depth ratios experiencing column-wall failure.

343 5. Flexural and catenary mechanisms under different span-to-depth ratios

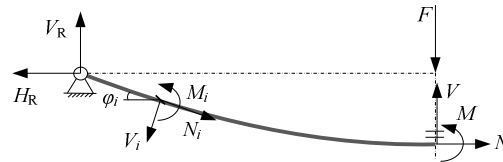
344 Under the central column removal scenario, the gravity resistance of a moment
 345 beam-column assembly is contributed by the flexural and the catenary mechanisms. As
 346 discussed in reference [18], the vertical reaction V_R in Fig. 14, can be calculated from the
 347 following equation

$$348 \quad V_R = V_i \cos \varphi_i + N_i \sin \varphi_i = F_f + F_c \quad (2)$$

349 where V_i , N_i and φ_i are the transverse shear force, axial force and rotation of the deflected

350 beam section, respectively. The internal forces V_i and N_i can be determined from the strain
 351 readings located at some distances from the supports [18].

352



353

354 Fig. 14. Analysis of resistance and internal force for the beam-column assembly (modified from [18]).

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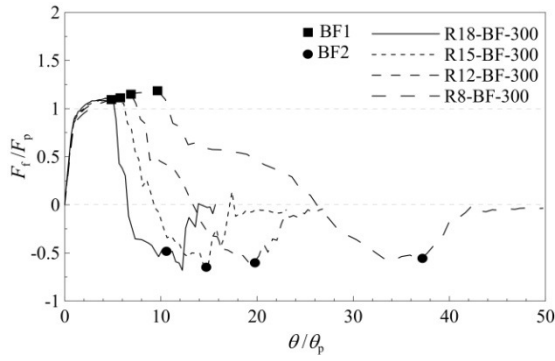
356 The first term on the right hand side of Equation (2), F_f , is the resistance component due
 357 to the flexural mechanism, and the second term, F_c , is due to the catenary mechanism. The
 358 developments of these two resistance components of assemblies in Section 4.2 as computed
 359 from the equation at certain sections of the beams are shown in Fig. 15, normalized by the
 360 corresponding plastic hinge load F_p and plotted against the normalized chord rotation.

361 It is demonstrated in Fig. 15 (a) and (b) that the flexural resistances F_f of all assemblies
 362 develop in the same manner during the elastic stage until they exceed the plastic hinge load
 363 F_p , following which the respective initial damages (step “BF1” or “CF1”) cause drastic
 364 declines of the flexural resistances. The negative zone of each flexural resistance is due to
 365 the rapidly growing horizontal reaction force at the pin support, associated with the
 366 development of the catenary mechanism.

367 Fig. 15 (c) and (d) show that, although the catenary resistances F_c are affected by the early
 368 damages (step “BF1” or “CF1” and “CF2”) to drop temporarily, thereafter they increase to
 369 peak values ranging from $1.4F_p$ to $3.6F_p$.

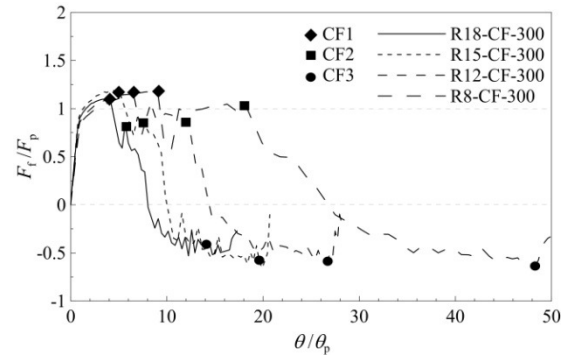
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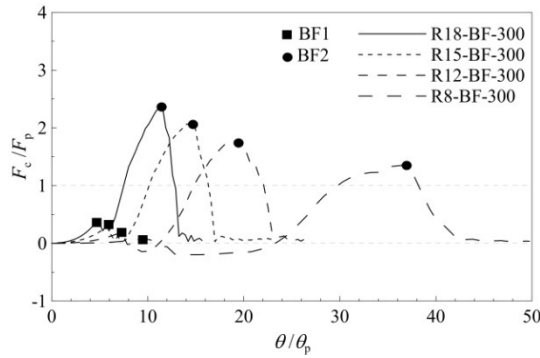
372

(a) Flexural mechanism for “BF” models.



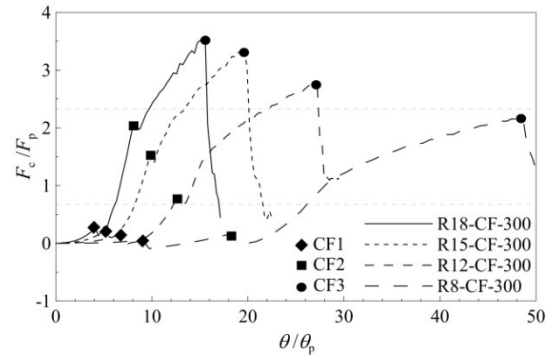
(b) Flexural mechanism for “CF” models.

373



374

(c) Catenary mechanism for “BF” models.



(d) Catenary mechanism for “CF” models.

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Fig. 15. Developments of gravity resistances contributed by flexural and catenary mechanisms.

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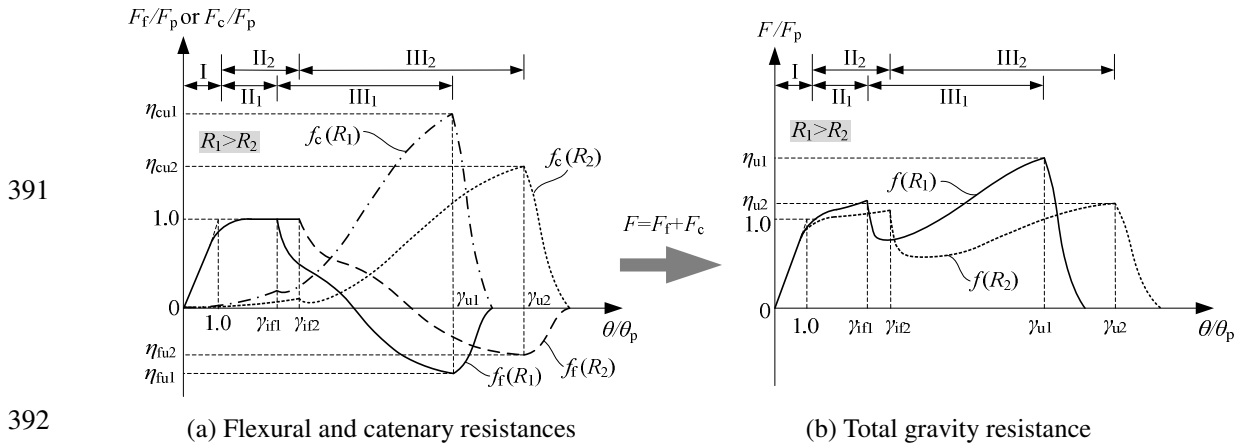
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386

Based on the parametric analyses (Fig. 11, Fig. 12, Fig. 13, and Fig. 15), a schematic illustration is provided in Fig. 16 to outline the development of the progressive collapse resistance of the moment beam-column assemblies having the same beam section but two span-to-depth ratios R_1 and R_2 ($R_1 > R_2$). The two components of the gravity resistance due to the flexural and the catenary mechanisms are separately plotted in Fig. 16 (a), denoted ' f_f ' and ' f_c ', respectively, and their resultant is plotted in Fig. 16 (b). Three distinctive stages are identified as indicated in the graphs, being the flexure dominated stage “I”, the combined flexure-catenary stage “II” and the catenary dominated stage “III”. The three stages are separated from each other by the plastic hinge formation and the initial fracture of the connection (such as “BF1” and “CF1” when $\theta/\theta_p = \gamma_{if1}$ or γ_{if2}). Stage “III” ends when the last fracture takes place in the connection (such as “BF2” and “CF3” when $\theta/\theta_p = \gamma_{uf1}$ or γ_{uf2}).

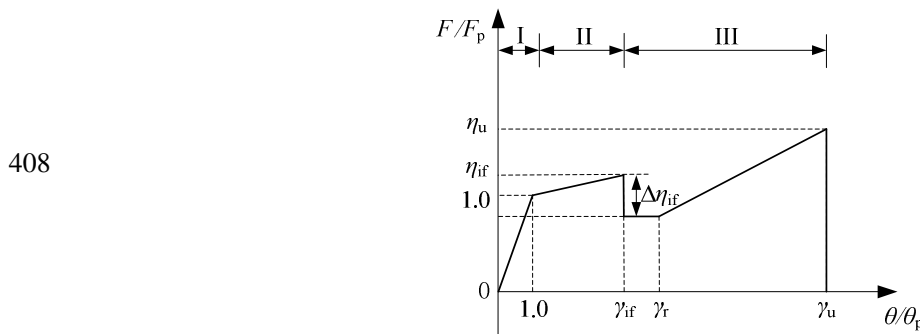
387 It can be seen that the assembly with a larger span-to-depth ratio R_1 is able to provide a
 388 higher ultimate gravity resistance ratio η_{u1} due to its more effective facilitation of the
 389 catenary mechanism. However, the smaller span-to-depth ratio R_2 enables the assembly to
 390 resist the ultimate load at a greater chord rotation ratio γ_{u2} .



393 Fig. 16. Schematic illustration of gravity resistance development for beam-column assembly.

394 For a convenient assessment of the beam-column assembly directly affected by the
 395 removed column [23], a simplified curve for the gravity resistance development is proposed
 396 in Fig. 17. It is suitable for the connection methods exhibiting failure modes that facilitate an
 397 effective development of the catenary mechanism in the post-fracture stage, such as the
 398 beam-end interrupted failure mode and the column-wall failure mode. The assembly has a
 399 gravity resistance of F_p when a plastic hinge forms at the beam-end section at chord rotation
 400 θ_p (refer to equation (1)). Afterwards, the gravity resistance grows to $\eta_{if}F_p$ (at a slower rate)
 401 until the initial fracture occurs at chord rotation $\eta_{if}\theta_p$, which causes a loss of gravity
 402 resistance equal to $\Delta\eta_{if}F_p$. The gravity resistance may then plateau, a response which is most
 403 pronounced for the assembly having a small span-to-depth ratio undergoing the beam-end
 404 interrupted mode (see Fig. 12), and which can be neglected otherwise. The assembly reaches
 405 the ultimate gravity resistance $\eta_u F_p$ when the damage has extended upwards close to the top

406 flange, with corresponding chord rotation of $\gamma_u\theta_p$, after which the gravity resistance is
 407 deemed to be lost completely.



409 Fig. 17. Simplified curve model for the development of gravity resistance.

410 The values of the parameters in the proposed simplified curve model in Fig. 17, including
 411 the gravity resistance ratios and the chord rotation ratios, depend on the span-to-depth ratio
 412 and connection methods as well as the failure modes. Further research is required to quantify
 413 them.

414 6. Conclusions

415 The full response of moment resisting beam-column assemblies, extracted from the bays
 416 directly affected by a failed interior column in a typical steel framing system, have been
 417 investigated under different span-to-depth ratios covering the commonly used range through
 418 an experimental test and thirty-three numerical simulations.

419 The tested specimen, a B-J-B assembly with a beam span-to-depth ratio of 8, experienced
 420 failures at the beam-end section and in the column wall on the two sides of the WUF-BW
 421 connection, respectively. Both the beam-end interrupted failure mode and the column-wall
 422 failure mode enabled the assembly to effectively facilitate the development of the catenary
 423 mechanism in the post-fracture stage, which is important for structure bridging over a failed
 424 interior column so as to prevent progressive collapse.

425 Parametric analyses of beam-column assemblies having four span-to-depth ratios (18, 15,
426 12 and 8) have been conducted, using validated finite element (FE) models which took
427 account of material fracture. It has been demonstrated that assemblies having the same
428 span-to-depth ratio behave similarly in terms of their normalized load-rotation relationships
429 even though they are configured with different beam depths. Conversely, assemblies having
430 the same beam and column sections but different span-to-depth ratios behave differently in
431 terms of their normalized load-rotation relationships.

432 Nevertheless, for a particular failure mode of the moment connection that is capable of
433 facilitating an effective development of the catenary mechanism, the gravity resistance
434 developments of all assemblies share a common trend despite their different span-to-depth
435 ratios (and different beam sections). The three development stages, being the flexure
436 dominated stage, the combined flexure-catenary stage and the catenary dominated stage, are
437 separated from each other by the plastic hinge formation at the critical beam section and the
438 initial fracture in the connection region.

439 In general, the beam-column assembly with a larger span-to-depth ratio is able to develop
440 the gravity resistance earlier, and provide a higher ultimate resistance by facilitating a more
441 effective catenary mechanism. However, the assembly with a smaller span-to-depth ratio
442 exhibits a more ductile response.

443 A simplified curve model of the gravity resistance development of a moment
444 beam-column assembly with damage evolution has been proposed for a convenient
445 assessment of the progressive collapse resistance following a central column loss. Further
446 research is required to quantify the model parameters.

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List of figures

Fig. 1. Details of the WUF-BW connection.

Fig. 2. Test setup.

Fig. 3. Schematic arrangements of test instrumentation.

Fig. 4. Photographs of the specimen at the end of the test.

Fig. 5. Load-displacement curve of test specimen.

Fig. 6. Damage evolutions at key stages of test specimen.

Fig. 7. Final conditions between the bottom inner diaphragm and column inside wall.

Fig. 8. Finite element models composed of solid elements.

Fig. 9. Simulated failure modes.

Fig. 10. Comparison of load-displacement curves between FE simulation and test for specimen.

Fig. 11. Normalized load-displacement curves for assemblies having the same span-to-depth ratio.

Fig. 12. Responses of assemblies having different span-to-depth ratios experiencing beam-end interrupted failure.

Fig. 13. Responses of assemblies having different span-to-depth ratios experiencing column-wall failure.

Fig. 14. Analysis of resistance and internal force for the beam-column assembly (modified from [18]).

Fig. 15. Developments of gravity resistances contributed by flexural and catenary mechanisms.

Fig. 16. Schematic illustration of gravity resistance development for beam-column assembly.

Fig. 17. Simplified curve model for the development of gravity resistance.

List of tables

Table 1. Material properties of test specimen.

Table 2. Components of four groups for beam-column assemblies in the parametric analyses.