Towards acoustic condition monitoring for detection and characterisation of laser induced breakdown in a gas turbine laser ignition system

Journal:	International Journal of Condition Monitoring
Manuscript ID	IJCM-12-2017-TN-0131.R1
Manuscript Type:	Technical Note
Date Submitted by the Author:	20-Feb-2018
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Keyword:	Condition monitoring, Lasers, Acoustic emission methods (AE)



1	Towards	acoustic	condition	monitoring	for detection	and
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- 2 characterisation of laser induced breakdown in a gas turbine
- 3 laser ignition system
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16 ABSTRACT:

17 Acoustic detection and characterisation of laser induced breakdown is an attractive

18 proposition in laser ignition systems in which condition monitoring is necessary but

- 19 where optical access for monitoring purposes is impractical. Presented is a signal
- 20 processing method based on wavelet decomposition for the non-invasive detection
- 21 of acoustic emissions resulting from laser induced breakdown in an atmospheric
- 22 pressure combustion test rig, representative of a single combustion chamber in a sub
- 23 15 MW industrial gas turbine. The probability and consistency of laser induced
- 24 breakdown is determined from the acoustic signal and used to characterize the
- 25 operating conditions and identify abrupt and incipient or slowly developing faults.
- 26

27 **KEYWORDS**:

- 28 Lasers; Condition monitoring; Acoustic emission methods (AE)
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31 1. INTRODUCTION

1 Laser ignition (LI) offers the potential to address the durability issues associated with 2 conventional high energy electrical ignition systems by removing the ignition 3 apparatus from the hot gas path [1,2]. LI typically utilizes tightly focussed visible to 4 near infrared nanosecond pulsed laser radiation to induce optical breakdown in 5 combustible mixtures via a multi-photon ionization process for the purpose of 6 initiating combustion [3,4]. In this process, multi-photon ionization of atoms in the 7 target medium results in the release of electrons which readily absorb more 8 photons, thereby increasing their kinetic energy. Subsequent collision of these 9 electrons with neighbouring electrons results in avalanche ionization and the 10 generation of a plasma in the focal region.

11

Condition monitoring is essential to the development of an LI system for gas 12 13 turbines. Unsatisfactory operation of the LI system may be indicative of an incipient 14 or slowly developing fault such as damage to optical elements in the beam path or 15 an abrupt fault such as failure of a light guide. Both of the aforementioned fault 16 types would require the activation of an interlock which is capable of reliably 17 reducing the beam strength to safe levels within eye protection delay times, as 18 defined in IEC 60825-2 [5]. Due to the nature of the application and associated 19 hardware, there is limited optical access to the combustion chamber, making the 20 effective use of optical means of condition monitoring impractical. There is potential 21 for use of acoustic emissions to monitor the LI process, as has been applied to laser 22 peening in recent years, a process with many similarities to LI [9,10]. A component of 23 the energy within the plasma generated as a result of the multi-photon ionization 24 process manifests itself as an acoustic emission in the audible frequency range. This 25 is the result of a shockwave which is formed and expands around the plasma. 26 Initially, expansion of this shockwave is driven by the high temperature plasma, 27 whose energy increases due to absorption of laser radiation during the laser pulse 28 [6]. After the laser pulse ends the temperature of the plasma reduces, resulting in a 29 deceleration in the expansion of the shockwave. From this point, the blast wave 30 model of Jones can be employed to describe the propagation of the shockwave into 31 the acoustic region [7,8].

32

1	The principal challenge in the development of an acoustic emission based condition
2	monitoring system for detection and characterisation of laser induced breakdown
3	concerns resolving the acoustic emissions of interest in the relatively noisy
4	environment associated with engine operation [11]. Relatively little literature exists
5	concerning detection of acoustic emissions that result from laser induced
6	breakdown. Yaacob et al. proposed utilising changes in the refractive index of
7	multimode optical fibres to detect acoustic emissions resulting from laser induced
8	breakdown in isolation oil [12]. It was shown that this method produced a noise free
9	signal which required no signal processing. The linear dependence of acoustic
10	pressure on laser pulse energy for optical breakdown in water was characterized by
11	Bulanov et al. using a hydrophone and acoustic receiver [13]. Krasnenko et al.
12	studied the sound generation in the atmosphere resulting from optical breakdown in
13	aerosol sprays using a microphone, with a view to its practical application as a
14	diagnostic tool for the influence of atmospheric conditions on laser induced
15	breakdown regimes [14]. Hosoya et al. studied the time responses of sound pressure
16	generated by laser induced breakdown in air in an anechoic box [15]. The acoustic
17	signals resembled impulse responses, characterised by a high peak pressures, and
18	were found to be highly reproducible in terms of pulse width and peak pressure. This
19	reproducibility was attributed to the consistency of the laser as a source. No
20	literature exists regarding detection of acoustic emissions from optical breakdown
21	events in noisy environments associated with engine operation which are typical in
22	LI applications.
23	
24	Proposed in this study is a new approach to the condition monitoring of an LI system
25	in realistic, engine-like environments. Information regarding the probability and
26	consistency of laser induced breakdown is discerned from associated acoustic
27	emissions using wavelet decomposition; a signal processing methodology which is
28	uniquely suited to the extraction of impulse-like features heavily masked by noise.
29	The results of this investigation reveal significant potential for the practical
30	application of acoustic-based condition monitoring techniques in LI systems.
31	

32 2. METHODOLOGY

- 1 In this section, details of the experimental arrangement used to record acoustic
- 2 signals are given followed by a description of the signal processing technique used to
- 3 analyse the data.
- 4

5 2.1 Experimental arrangement

- 6 An atmospheric pressure combustion test rig was utilized in this study, which could
- 7 be fitted with a single combustor can from a range of Siemens industrial gas
- 8 turbines. For the purpose of this investigation, the rig was fitted with a combustion
- 9 can and pilot burner from a small to medium (that is, sub 15 MW output power)

10 Siemens industrial gas turbine.

- 11
- 12 A laser ignition system was developed utilizing a Q-switched Nd:YAG TEM₀₀ laser
- 13 (Brilliant; Quantel, Ltd.) with an M² of 1.84, pulse duration of 4 ns and operating at
- 14 10 Hz repetition rate and 1064 nm wavelength. The output pulse energy (E_p) of the
- 15 laser source was limited from its maximum of 360 mJ to approximately 21 mJ using
- 16 by flashlamp/Q-switch delay time to avoid damage to optical elements in the
- 17 system. With the flashlamp/Q-switch delay kept constant, a polarization based
- 18 optical attenuator consisting of a ¹/₂ wave plate and polarizing beam splitting cube
- 19 was used to control the laser pulse energy. The laser pulse energy for a given ½ wave
- 20 plate rotation was measured (UP19K-VR; Gentec Electro-Optics, Inc.) and averaged
- 21 over 100 pulses. The experimental arrangement is shown in Figure 1.
- 22





Figure 1: Experimental arrangement with 1) laser source, 2) ½ wave plate, 3) polarizing beam
splitting cube, 4) beam dump, 5) combustion chamber, 6) laser igniter assembly, 7) microphone, 8)
data acquisition computer and 9) camera.

5

6 A custom laser igniter assembly was utilized, consisting of a clear aperture for

7 transmission of the laser beam, a plano-convex focusing optic with an effective focal

8 length of 27 mm and an anti-reflective coated N-BK7 output window. The optical

9 path length from the laser source to the igniter assembly was approximately 3 m.

10 The tip of the igniter assembly was sealed with red silicone around the edge of the

- 11 output window.
- 12

13 The presence or otherwise of laser induced breakdown within the combustion

- 14 chamber could be verified by monitoring the camera feed. Figure 2 shows the
- 15 camera feed for both flashlamp operation and laser induced breakdown.
- 16



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Figure 2: Camera feed showing (a) flashlamp operation and (b) laser induced breakdown (10 Hz
 repetition rate, 1064 nm wavelength).

4

5 A condenser microphone (AT2020, Audio-Technica Ltd.) with a frequency response

6 of 0.02 to 20 kHz was used to capture the acoustic signal, which was sampled

7 directly using MATLAB at 48 kHz and 16-bit resolution. The sensor was placed

8 approximately 2 cm from the combustion chamber's external casing with the

9 sensitive plane aimed directly at the casing wall. In this investigation, the ambient

10 conditions were such that no additional steps were required to reduce wind noise on

11 the microphone.

12

13 **2.2 Signal processing methodology**

- 14 The signal processing work focused on acoustic identification of laser induced
- 15 breakdown in an atmospheric pressure combustion test rig, representative of an
- 16 industrial gas turbine during its start-up procedure. Audio recordings of 11 s in
- 17 duration were repeated ten times for a given set of experiment parameters. The raw
- 18 acoustic signal was sampled and imported directly into MATLAB. Each signal was
- 19 clipped to a length of 10 s to remove unwanted artefacts from the beginning and
- 20 end of the recording and then processed by two level wavelet decomposition,
- 21 allowing for the resolution of information in both the time and frequency domains.
- 22 Wavelet decomposition was chosen on the basis of both its sensitivity to abrupt
- 23 changes in frequency in the time domain and its computational efficiency. The base
- 24 wavelet best capable of resolving the laser induced breakdown events was selected
- using energy to Shannon entropy ratio at the first level of decomposition as the
- 26 criterion [16]. A total of 131 base wavelet types from the wavelet families

- 1 Daubechies, symlets, coiflets, biorthogonal, Fejer-Korovkin and reverse biorthogonal 2 were considered, with the base wavelet producing the maximum energy to Shannon 3 entropy ratio selected as the most appropriate wavelet; in this case, the 4 biorthogonal (Bior6.8) wavelet. The wavelet decomposition produced three sets of 5 coefficients, representing the approximation of the signal at low frequencies (a_1) and the detail of the signal at high frequencies (d_1 and d_2). Where present, laser induced 6 7 breakdown events are clearly visible at the first level of decomposition in the detail 8 coefficients d_1 , as shown in Figure 3 for raw acoustic signal s, representing consistent 9 laser induced breakdown in an atmospheric pressure combustion test rig. As such, 10 the d_1 coefficients (henceforth referred to as the detail signal) were extracted for 11 further processing.
- 12



13

Figure 3: Two level Bior6.8 wavelet decomposition of raw acoustic sound pressure signal *s* with approximation coefficients *a*₂, and detail coefficients *d*₁ and *d*₂ (21 mJ pulse energy, 10 Hz repetition rate, 1064 nm wavelength).

18 3 RESULTS AND DISCUSSION

1 The experimental work focused on (i) identification of abrupt faults and (ii) 2 identification of incipient or slowly developing faults in a gas turbine laser ignition 3 system. Acoustic signals were recorded during operation of the laser ignition system 4 on an atmospheric pressure combustion test rig with a constant air mass flow rate of 5 0.4 kg/s. It should be noted that this intake of air into the combustion chamber was the main source of noise when the test rig was in operation. Operation of intake and 6 7 exhaust fans also contributed to the noise. No fuel was introduced into the 8 combustion chamber in this investigation. This effectively recreates the conditions at 9 the point in the turbine start-up procedure at which the laser ignition system would 10 be fired for condition monitoring purposes. The acoustic signals were subsequently 11 processed according to the methodology outlined in Section 2.2. To simulate 12 unsatisfactory operation of the laser ignition system, the laser pulse energy (E_p) was 13 increased from a minimum to the maximum pulse energy used in this investigation 14 of 21 mJ by rotating the ½ wave plate in the experimental arrangement depicted in 15 Figure 1. A peak detection algorithm was applied to the detail signal to determine 16 the probability of laser induced breakdown as a function of laser pulses fired over a 17 given period of time, a representative output of which is shown in Figure 4. The 18 threshold above which the signal was considered to be a peak was dictated by the 19 maximum and minimum amplitude values for the detail signal at the first level of 20 decomposition for an ambient acoustic signal, recorded with test rig in operation 21 and the laser flashlamps on. The peak-to-peak separation was restricted using a 22 windowing function in accordance with the 10 Hz repetition rate of the laser source. 23



1

Figure 4: Representative outputs from the peak detection algorithm applied to the detail signal for
 a single 10 s audio recording (10 Hz repetition rate, 1064 nm wavelength).

4

5 Figure 4 reveals the presence of three distinct energy regions as the pulse energy is 6 increased. At an incident pulse energy of 5.18 mJ no laser induced breakdown 7 occurred. A second region was observed for incident pulse energies of 8.55 mJ and 8 10.71 mJ, characterized by inconsistent laser induced breakdown with a probability 9 of 2.7% and 64.6%, respectively. Optical breakdown can only occur when the 10 incident laser radiation is above a threshold value; therefore, for pulse energies close 11 to this threshold value as opposed to those well in excess of it, a statistically smaller 12 fraction of the laser pulse is capable of causing optical breakdown, resulting in a 13 reduced probability of laser induced breakdown [17]. A third region of consistent 14 laser induced breakdown was observed for incident pulse energies in excess of 12.15 15 mJ, in accordance with observations made by Chen et. al [18]. The presence of these 16 three distinct energy regions was confirmed by reviewing the recorded camera feed. 17

- 1 In addition to the probability of laser induced breakdown, information relating to the
- 2 consistency of laser induced breakdown can be inferred from the detail signal. The
- 3 following sections detail methods for the identification of abrupt and incipient or
- 4 slowly developing faults from this information.
- 5

6 **3.1 Abrupt fault detection**

7 An abrupt fault would result in unsatisfactory operation of the laser ignition system 8 immediately after the fault develops. Possible causes for such a fault may include 9 misalignment, severe damage to an optical element in the beam path or failure of 10 the laser source to operate correctly. In the event of an abrupt fault, inconsistent or 11 no laser induced breakdown will occur. As such, the time interval between the peaks 12 detected in the detail signal is a useful parameter to consider. Figure 5 presents a 13 box plot of the peak intervals for those incident pulse energies which resulted in 14 laser induced breakdown. 15



Figure 5: Box plot of the peak intervals (d_{1interval}) from the peak detection algorithm applied to the detail signals in (a) the consistent laser induced breakdown region and (b) the inconsistent laser induced breakdown region (10 Hz repetition rate, 1064 nm wavelength).

1	
2	Referring to Figure 5, for incident pulse energies in excess of 12.15 mJ (that is, in the
3	consistent laser induced breakdown region) the peak intervals are centred on 0.1 s,
4	corresponding to the 10 Hz repetition rate of the laser source. As the incident pulse
5	energy decreases to 10.71 mJ and 8.55 mJ the median interval and distribution both
6	increase. This can be attributed to increasingly inconsistent spark formation as the
7	incident pulse energy decreases, therefore reducing the number of breakdown
8	events detected.
9	
10	3.2 Incipient fault detection
11	An incipient or slowly developing fault may result in unsatisfactory operation of the
12	laser ignition system at some point after the fault initially develops. A possible cause
13	for such a fault could be slight damage to an optical element in the beam path,
14	acting as a nucleation point for further damage upon successive irradiation. In the
15	event of an incipient fault occurring, it is important to identify and rectify the
16	underlying issue as it can eventually lead to inconsistent laser induced breakdown
17	and therefore unsatisfactory operation of the laser ignition system. Figure 6 shows
18	the distribution of the absolute value of peak amplitudes as a function of incident
19	pulse energy.
20	



1

Figure 6: Box plot of absolute values of peak amplitude from the peak detection algorithm applied
to the detail signal (*d_{1peak}*) and their relative standard deviation (circles) as a function of incident
pulse energy (10 Hz repetition rate, 1064 nm wavelength).

5

Figure 6 reveals a decrease in the median absolute value of peak amplitude with 6 7 decreasing incident pulse energy, following a sigmoidal trend. In the consistent laser 8 induced breakdown region (that is, at incident pulse energies of 12.15 mJ and above) 9 there is also a narrowing of the distribution with decreasing incident pulse energy. 10 This narrowing can be attributed to the decrease in magnitude of the individual peak 11 amplitudes as less energy is deposited in the shock wave. As such, this narrowing is 12 not reflected in the relative standard deviation, which remains consistent in this 13 region. The transition from consistent to inconsistent laser induced breakdown is 14 characterized by an abrupt increase in the relative standard deviation at an incident 15 pulse energy of 10.71 mJ, corresponding to a breakdown probability of 64.6%. Long 16 term monitoring (that is, at every start-up sequence for the turbine) of the absolute 17 value of peak amplitude could be used to monitor for the sigmoidal trend shown in 18 Figure 6, indicative of an incipient or slowly developing fault.

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1	4 CONCLUSIONS
2	Noise robust acoustic condition monitoring for detection and characterisation of
3	laser induced breakdown in a laser ignition system has been demonstrated. Acoustic
4	detection of laser induced breakdown events under realistic engine-like conditions
5	was achieved through use of a signal processing method based on wavelet
6	decomposition using a Bior6.8 mother wavelet and extraction of the detail
7	coefficients at the first level of decomposition as a de-noised signal.
8	
9	The potential for the implementation of an acoustic sensor system for condition
10	monitoring of a gas turbine laser ignition system has been clearly demonstrated. A
11	peak detection algorithm was applied to determine the probability of laser induced
12	breakdown as a function of laser pulses fired over a given period of time.
13	Information relating to the consistency of laser induced breakdown was then used to
14	identify abrupt and incipient or slowly developing faults, simulated by decreasing the
15	incident pulse energy. The time interval between peaks was shown to increase from
16	a base of 0.1 s corresponding to the repetition rate of the laser as the energy region
17	transitioned from consistent to inconsistent laser induced breakdown with
18	decreasing incident pulse energy, indicative of an abrupt fault. The absolute value of
19	peak amplitude was shown to decrease with decreasing incident pulse energy,
20	following a sigmoidal trend which could be used to identify insipient faults. Further
21	work is required to address the instrumentation and integration challenges
22	associated with development of a real-time acoustic condition monitoring system for
23	gas turbine laser ignition systems.
24	
25	ACKNOWLEDGEMENTS:
26	
27	The authors would like to thank Siemens Industrial Turbomachinery Ltd (SITL) for
28	funding this investigation as part of a wider investigation into laser ignition for gas
29	turbines.
30	
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Figure 1: Experimental arrangement with 1) laser source, 2) ½ wave plate, 3) polarizing beam splitting cube, 4) beam dump, 5) combustion chamber, 6) laser igniter assembly, 7) microphone, 8) data acquisition computer and 9) camera.

248x180mm (96 x 96 DPI)

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Figure 2: Camera feed showing (a) flashlamp operation and (b) laser induced breakdown (10 Hz repetition rate, 1064 nm wavelength).

240x89mm (96 x 96 DPI)



Figure 3: Two level Bior6.8 wavelet decomposition of raw acoustic sound pressure signal s with approximation coefficients a2, and detail coefficients d1 and d2 (21 mJ pulse energy, 10 Hz repetition rate, 1064 nm wavelength).

2057x1543mm (72 x 72 DPI)



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Figure 6: Box plot of absolute values of peak amplitude from the peak detection algorithm applied to the detail signal (d1peak) and their relative standard deviation (circles) as a function of incident pulse energy (10 Hz repetition rate, 1064 nm wavelength).

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