

# Predicting the filling of ventilated cavities behind spillway aerators

## Prediction du remplissage des cavités ventilés derrière des aérateurs d'évacuateurs de crûes



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### SUMMARY

Cavitation damage to spillway surfaces may be prevented with the use of aeration devices. These serve to introduce air into the layers close to the channel bottom in order to reduce cavitation erosion. Under some circumstances, the aerator can be drowned out, will cease to protect the spillway surface, and act potentially as a cavitation generator. This article analyses the conditions of filling. Then experimental data for ten aerator geometries are reviewed. Depending upon the aerator geometry, the cavity filling occurs when the Froude number is less than a critical value or when the ratio of the flow depth over the total offset is larger than a characteristic value.

### RÉSUMÉ

Les aérateurs de fond sont installés sur les évacuateurs de crûes pour empêcher l'érosion par cavitation. Ils servent à introduire de l'air dans les filets d'eau proches de la surface du coursier, afin d'empêcher les dommages par cavitation. Dans certaines conditions, un aérateur peut être noyé, auquel cas il cesse de protéger le coursier, et peut agir comme un générateur de cavitation. Cet article décrit les conditions de remplissage (noyade) des aérateurs. Puis plusieurs résultats expérimentaux sont présentés. En fonction de la géométrie des aérateurs, le remplissage est possible pour des nombres de Froude inférieurs à une valeur critique, ou pour des profondeurs adimensionnelles supérieures à une valeur limite.

### 1 Introduction

At large head chutes and bottom outlets, the risks of cavitation damage on chute spillways can be significant. For velocities greater than 35 m/s, severe tolerances of surface finish are required to avoid cavitation (e.g. FALVEY 1982). The cost of cavitation resistant materials to protect surfaces is prohibitive. For these reasons, the spillway surface is protected from cavitation erosion by introducing air next to the spillway surface. Air is introduced artificially by aeration devices located on the spillway floor and sometimes on the sidewalls.

A small deflection in a spillway structure (e.g. ramp, offset) deflects the high velocity flow away from the spillway surface (Fig. 1). In the cavity formed below the nappe, a local pressure depression (i.e. subpressure)  $\Delta P$  is produced by which air is entrained into the flow.

#### *Aerator filling*

When the local pressure within the ventilated cavity drops severely, the cavity downstream of the deflector may disappear and be filled or drowned. This is called the aerator filling: the process is

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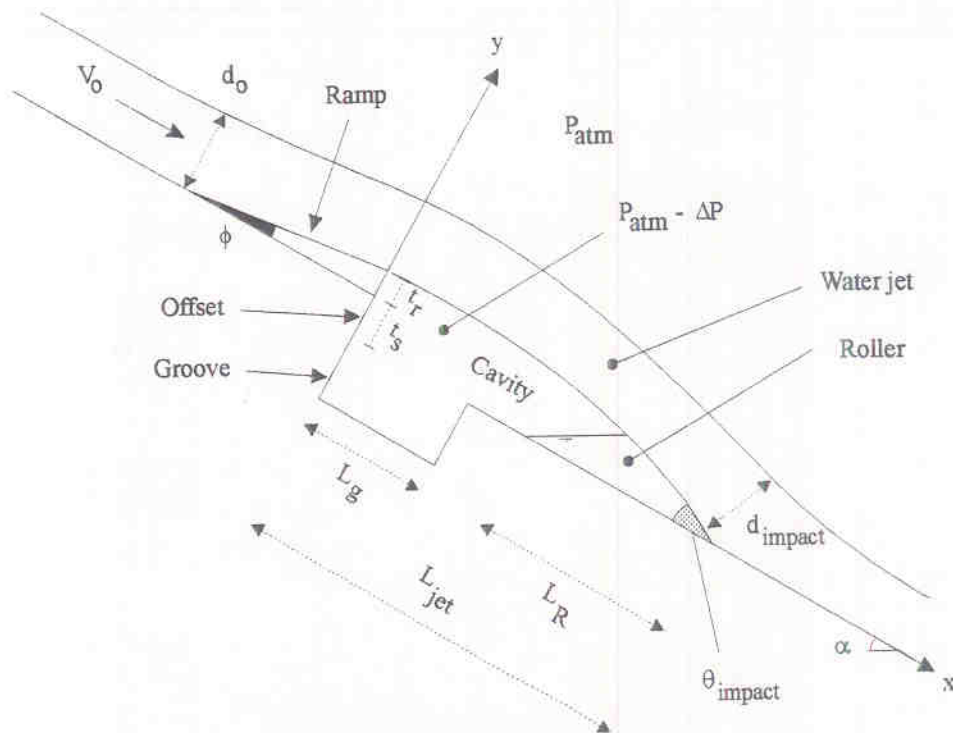


Fig. 1. Geometry of an aeration device.

named "cavity disappearance" by LAALI and MICHEL (1984), or "aerator submergence" by CHANSON (1990) and RUTSCHMANN and HAGER (1990). When the air cavity becomes filled with water, it stops playing its protecting role. At the limit the aerator becomes a large roughness and a cavitation generator (LI 1988, CHANSON 1990). For small Froude numbers, a partial or complete obstruction of the ventilation system can be caused by ice or debris blocking of the air inlets, man-made intervention or water filling of inlets. Such obstruction can reduce the local pressure ( $P_{atm} - \Delta P$ ) in the nappe below a critical limit ( $P_{atm} - \Delta P^{sub}$ ) and the cavity could be filled (CHANSON 1990, RUTSCHMANN and HAGER 1990).

Observations on a spillway model (CHANSON 1988) indicated that the passage from a ventilated to a filled cavity is preceded by the appearance of a large amount of spray falling from the underside of the nappe. Further as the jet curvature increases with a local pressure reduction, the upper free-surface of the flow downstream of the nappe becomes completely modified (Fig. 2). The presence of a spray along the lower air-water interface was observed by other researchers (PINTO et al. 1982, VOLKART and RUTSCHMANN 1984) for moderate to large cavity depressions, but their studies did not consider the critical case of cavity filling.

In this paper, it is shown that the onset of cavity filling can be predicted in term of a critical sub-pressure  $\Delta P^{sub}$  for given aerator geometry and flow conditions. Then model and prototype data are examined. The analysis provides practical criterion to predict the risk of cavity filling.

## 2 Onset of cavity filling

### 2.1 Presentation

For low Froude numbers and large cavity subpressures, visual observations indicate that the water jet above the aerator is almost unaerated and is characterised by the presence of a roller where the



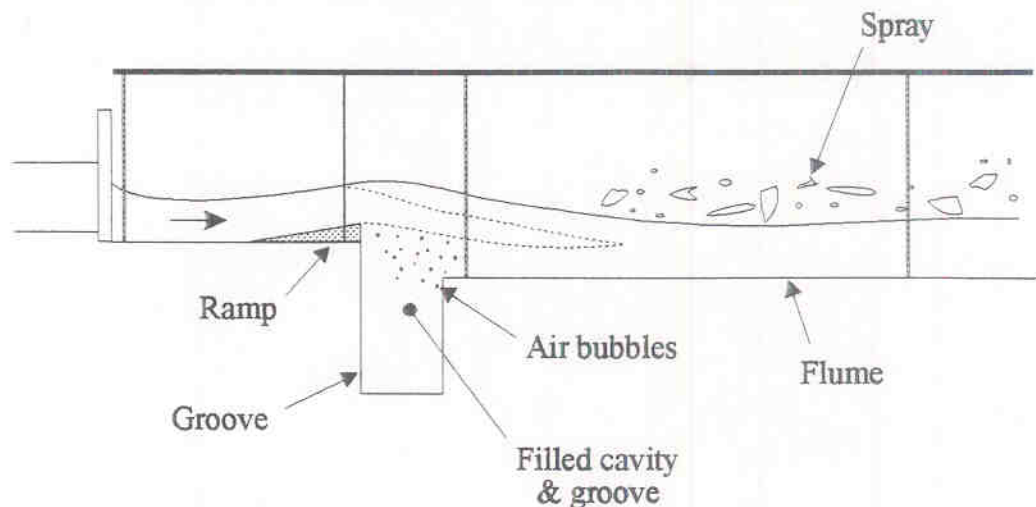
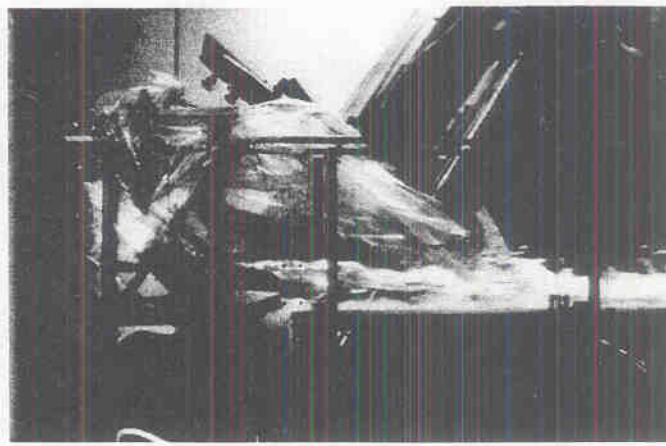


Fig. 2. Flow above the aerator of the Clyde dam spillway model during the filling of the aerator –  $Fr_o = 4.0$ ,  $d_o = 105$  mm, Geometry CH1 – Water flows from the left to the right.

water jet impacts on the spillway surface (Fig. 1). The filling of the cavity occurs when the upstream edge of the roller laps against the groove edge: i.e., if:

$$L_{jet} - L_R = L_g \quad (1)$$

where  $L_g$  is the groove length (Fig. 1). If the aeration device is grooveless,  $L_g$  must be replaced by the distance from the deflector edge to the downstream edge of the air inlets in equation (1).

The jet length  $L_{jet}$  and roller length  $L_R$  can be calculated as functions of the upstream flow conditions, the aerator geometry and the cavity subpressure (see below, eqs. (3), (4) & (10)). The critical subpressure at the onset of the cavity filling,  $P_N^{sub}$ , satisfies equation (1), where  $P_N^{sub} = \Delta P^{sub}/(\rho_w g d_o)$ , and cavity filling occurs for  $[L_{jet} - L_R < L_g]$  or  $[P_N > P_N^{sub}]$ , where  $P_N$  is the pressure gradient number ( $P_N = \Delta P/(\rho_w g d_o)$ ).

## 2.2 Jet length calculations

The jet trajectory and the cavity geometry can be computed as a function of the aerator geometry, the flow properties at the edge of the deflector and the cavity subpressure using analytical methods or numerical methods, including the finite element method.

For engineering applications, TAN (1984) developed a simple method to compute the jet length, assuming that the sub-nappe pressure acts only in the direction perpendicular to the spillway surface: i.e. the effect of the subpressure in the direction parallel to the spillway surface is ignored. At

the end of the ramp, the velocity parallel to the spillway surface is ( $V_o \cos \phi$ ) and the velocity perpendicular to the spillway surface ( $V_o \sin \phi$ ). Once the fluid leaves the deflector, the acceleration towards the surface is:  $-g(\cos \alpha + P_N)$  and the acceleration parallel to the spillway surface is:  $+g \sin \alpha$ . The time  $t$  taken to reach the spillway surface is deduced from the trajectory equation:

$$t_r + t_s = (\cos \alpha + P_N) \frac{gt^2}{2} - V_o (\sin \phi) t \quad (2)$$

where  $t_s$  is the offset height,  $t_r$  the ramp height,  $V_o$  the velocity at the end of the deflector, and  $\phi$  the ramp angle. The solution in term of  $t$  is:

$$t = \frac{V_o \sin \phi}{g (\cos \alpha + P_N)} \left( 1 + \sqrt{1 + 2 (t_r + t_s) \frac{g (\cos \alpha + P_N)}{(V_o \sin \phi)^2}} \right) \quad \text{aerator with ramp (3a)}$$

or

$$t = \sqrt{\frac{2t_s}{g (\cos \alpha + P_N)}} \quad \text{aerator without ramp (3b)}$$

The distance travelled during the time  $t$  is the jet length:

$$L_{jet} = \frac{g \sin \alpha}{2} t^2 + V_o (\cos \phi) t \quad (4)$$

where  $t$  is computed from equation (3). TAN (1984) compared his results with SCHWARTZ and NUTT's (1963) calculations and experimental values observed on the Clyde dam spillway model. The results showed no significant difference between TAN's and SCHWARTZ and NUTT's methods but all calculated values were larger than the observed jet lengths (TAN 1984).

At the impact of the water jet with the spillway bottom, the jet velocity and the jet thickness are respectively:

$$\frac{V_{impact}}{V_o} = \left( 1 + 2 \frac{gt}{V_o} (\cos \phi \sin \alpha - \sin \phi (\cos \alpha + P_N)) + \left( \frac{gt}{V_o} \right)^2 (1 + P_N (P_N + 2 \cos \alpha)) \right)^{1/2} \quad (5)$$

$$\frac{d_{impact}}{d_o} = \left( 1 + 2 \frac{gt}{V_o} (\cos \phi \sin \alpha - \sin \phi (\cos \alpha + P_N)) + \left( \frac{gt}{V_o} \right)^2 (1 + P_N (P_N + 2 \cos \alpha)) \right)^{-1/2} \quad (6)$$

The impact angle of the jet with the spillway bottom can be deduced from the jet trajectory equations. It yields:

$$\theta_{impact} = \tan^{-1} \left( \frac{\frac{gt}{V_o} (\cos \alpha + P_N) - \sin \phi}{\frac{gt}{V_o} \sin \alpha + \cos \phi} \right) \quad (7)$$

Note that the jet length calculations are developed neglecting the effects of the pressure from the roller. These effects are minor compared to the errors on experimental data.

### 2.3 Roller calculation

At low Froude numbers, a roller is observed when the jet reaches the spillway surface (Fig. 1). The roller is important as its weight provides the force parallel to the spillway surface which is required to prevent the flow recirculating into the cavity below the jet. At higher Froude numbers these rollers are no longer discernible but the hydrostatic pressure engendered by the pool is still necessary to change the jet from an angle to the spillway surface to parallel to this surface.

With the knowledge of the jet trajectory, the roller length can be calculated. The momentum equation resolved down the plane is:

$$\begin{aligned}
 & -\frac{\Delta P}{2} d_{\text{impact}} \cos(\theta_{\text{impact}}) - \frac{\rho_w g}{2} (d_{\text{impact}})^2 \cos \alpha + \left( \frac{\rho_w g}{2} \frac{(L_R \sin \alpha)^2}{\sin(\alpha + \theta_{\text{impact}})} - \frac{\Delta P L_R \sin \alpha}{\sin(\alpha + \theta_{\text{impact}})} \right) \sin(\theta_{\text{impact}}) \\
 & = \rho_w \frac{q_w^2}{d_{\text{impact}}} [1 - \cos(\theta_{\text{impact}})] + \rho_w g [\text{CV}] \sin \alpha
 \end{aligned} \tag{8}$$

assuming that the edges of the jet fluid have not disintegrated into spray, neglecting the shear forces on the surfaces and assuming that the magnitude of the velocity entering the control volume approximately the same as leaving it. For a large momentum inflow, it is possible to select a control volume [CV] such that the volume force is small compared to the momentum change. With this assumption, it yields:

$$\begin{aligned}
 & \frac{1}{2} \frac{\sin(\theta_{\text{impact}})}{\sin(\alpha + \theta_{\text{impact}})} \left( \frac{L_R \sin \alpha}{d_{\text{impact}}} \right)^2 - \left( \frac{\sin(\theta_{\text{impact}})}{\sin(\alpha + \theta_{\text{impact}})} P_{N_{\text{impact}}} \right) \left( \frac{L_R \sin \alpha}{d_{\text{impact}}} \right) \\
 & - \left( \frac{1}{2} \cos \alpha (1 + P_{N_{\text{impact}}}) + (Fr_{\text{impact}})^2 [1 - \cos(\theta_{\text{impact}})] \right) = 0
 \end{aligned} \tag{9}$$

where  $Fr_{\text{impact}} = q_w / \sqrt{g d_{\text{impact}}^3}$  and  $P_{N_{\text{impact}}} = \Delta P / (\rho_w g d_{\text{impact}})$ . The solution of the second-degree polynomial (eq. (9)) is:

$$\begin{aligned}
 & \left( \frac{L_R \sin \alpha}{d_{\text{impact}}} \right) = P_{N_{\text{impact}}} + P_{N_{\text{impact}}} \\
 & \times \sqrt{1 + \frac{\sin(\theta_{\text{impact}}) \{ \cos \alpha (1 + P_{N_{\text{impact}}}) + 2 (Fr_{\text{impact}})^2 [1 - \cos(\theta_{\text{impact}})] \}}{(P_{N_{\text{impact}}})^2 \sin(\alpha + \theta_{\text{impact}})}}
 \end{aligned} \tag{10}$$

For a zero pressure gradient (i.e. the same air pressure above and below the jet:  $\Delta P = 0$ ), equation (10) yields:

$$\left( \frac{L_R \sin \alpha}{d_{\text{impact}}} \right) = \sqrt{\frac{\sin(\theta_{\text{impact}})}{\sin(\alpha + \theta_{\text{impact}})} \{ \cos \alpha + 2 (Fr_{\text{impact}})^2 [1 - \cos(\theta_{\text{impact}})] \}} \tag{11}$$



The author (CHANSON 1988) compared successfully equation (10) with visual observations of the roller length  $L_R$  at low Froude numbers.

Note that both the jet length (eq. (3) & (4)) and the roller length (eq. (10)) are functions of the cavity subpressure  $\Delta P$ . Hence equation (1) is a function the cavity subpressure.

#### 2.4 Discussion

Equations (1), (4) and (10) form a system of non linear equations in terms of  $P_N^{sub}$ ,  $L_{jet}$  and  $L_R$ . For a given aerator geometry ( $\alpha$ ,  $\phi$ ,  $t_r$ ,  $t_s$ ,  $L_g$ ) and flow conditions ( $V_o$ ,  $d_o$ ), the cavity filling will occur if the pressure gradient number  $P_N$  is larger than  $P_N^{sub}$ .

The author measured cavity subpressures during some cavity filling events (Table 2): in one case ( $Fr_o = 4.02$ ), immediately before and during the filling, and immediately before in the second case ( $Fr_o = 4.63$ ). The data are compared with the calculated values of  $P_N^{sub}$  on Figure 3. Details of the aerator geometry are reported in Table 1.

TAN (1984), RUTSCHMANN (1988) and the author recorded also large cavity subpressures for flow conditions close to the occurrence of cavity filling. Their data were obtained with non-filled cavities (table 2). They are compared also with the critical subpressure for cavity filling on figure 3. During the experiments, the error on the subpressure measurement for non-filled cavities (TAN 1984, RUTSCHMANN 1988, CHANSON 1988) is about:  $\delta(\Delta P)/\Delta P \sim \pm 10\%$ . During filling events (CHANSON 1988), the error was about:  $\delta(\Delta P)/\Delta P \sim \pm 30\%$  because of the large pressure fluctuations and the difficulties to read accurately the subpressure readings.

On Figure 3, it is interesting to note that: 1- all the subpressure data for non-filled cavity are smaller than the calculated subpressure for onset of cavity filling. And: 2- the subpressure data recorded immediately before or during cavity filling events are nearly equal or larger than the calculated  $P_N^{sub}$ . Such results, shown on Figure 3, indicate that the calculation of  $P_N^{sub}$  provides a reasonable estimate of the onset of cavity filling on spillway model. The calculation enables a reasonable prediction method of the risks of cavity filling, if the local cavity pressure can be predicted. The complete set of data and calculations are reported on Table 2.

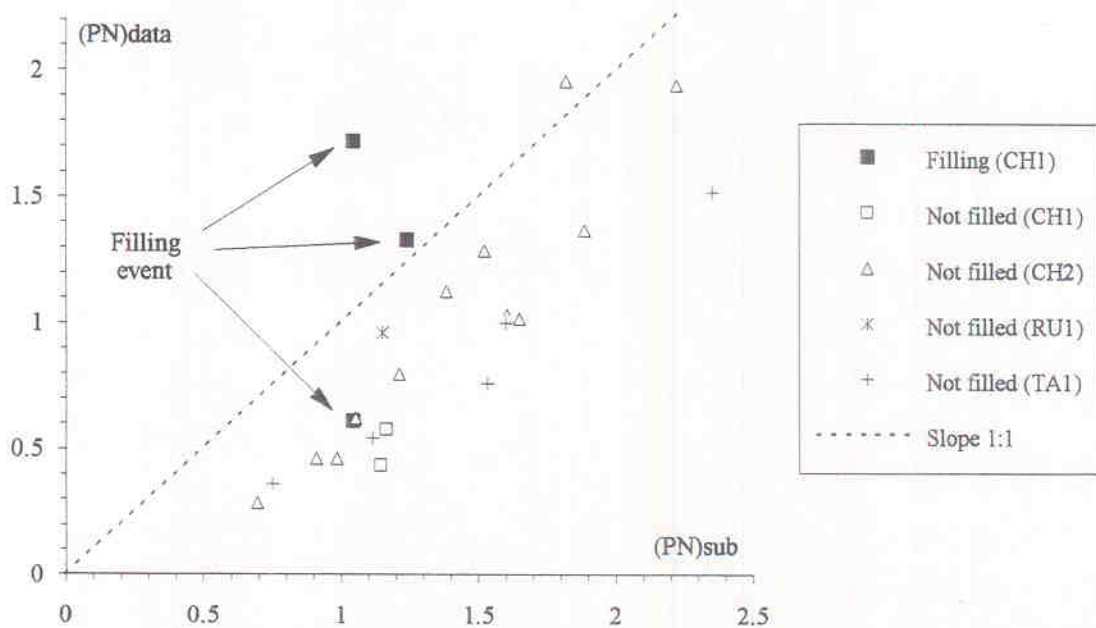


Fig. 3. Comparison between large subpressure measurements (Table 2) and the critical subpressure for onset of cavity filling  $P_N^{sub}$  (eq. (1)) – References are given in Table 1.

In practice, the cavity subpressure is a function of the duct head losses and the air entraining capacity of the flow above the aerator. The air entraining capacity of the flow is called the air demand. Although extensive investigations were performed on both model and prototype (e.g. PINTO et al. 1982, CHANSON 1990, RUTSCHMANN and HAGER 1990), there is still a lack of clear understanding of the air entrainment mechanisms. Presently, no analytical nor theoretical method exists to predict 'a priori' the air demand.

### 3 Experimental observations of cavity filling

#### 3.1 Experimental data

Experimental observations of aerator filling were obtained by COLEMAN et al. (1983), SHI et al. (1983), TAN (1984), BRETSCHNEIDER (1986) and CHANSON (1988). The experimental flow conditions and aerator geometries of these experiments are given in table 1. The critical flow conditions for cavity filling are summarised in table 3. For the experiments of SHI et al. (1983), TAN (1984) and CHANSON (1988), the filling occurred for a complete closure of the ventilation system (i.e.  $Q_{air}^{inlet} = 0$ ).

Table 1. Flow conditions and aerator geometry.

Aerator geometry	Ref.	Channel slope $\alpha$ (deg.)	Ramp angle $\phi$ (deg.)	Ramp height $t_r$ (m)	Offset height $t_s$ (m)	Groove length $L_g$ (m)	Froude number $Fr_o$	Flow depth $d_o$ (m)	Remarks
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)	(9)	(10)
Aerator with ramp									
SH1	(I)	0	5.7	0.01	0.0	(--)	2.9 to 21.9	0.06 to 0.18	Fengjiashan spillway model. W = 0.2 m.
SH2	(I)	10	5.7	0.01	0.0	(--)	2.9 to 21.9	0.06 to 0.18	
SH3	(I)	30	5.7	0.01	0.0	(--)	2.9 to 21.9	0.06 to 0.18	
SH4	(I)	49	5.7	0.01	0.0	(--)	2.9 to 21.9	0.06 to 0.18	
CO1	(II)	7.125	1.9	0.005	0	(--)	(--)	(--)	Urbante spillway model
RU1	(IV)	34.5	4	0.0095	0	0.2	8.7	0.10	Configuration I.E.
CH1	(V)	52.3	5.7	0.03	0.03	0.19	3.4 to 23.4	0.023 to 0.11	Clyde dam spillway model. W = 0.25 m.
Aerator with offset									
TA1	(III)	51.3	0	0.0	0.03	0.19	5.0 to 15.4	0.05 to 0.11	Clyde dam spillway model. W = 0.25 m.
CH2	(V)	52.3	0	0	0.03	0.19	5.4 to 23.6	0.023 to 0.083	
Horizontal groove									
BR1	(IV)	0	0	0	0	0.02 to 0.10	(--)	0.004 to 1	Flow above a groove only

(I): Shi et al. (1983)

(II): Coleman et al. (1983)

(III): Tan (1984)

(IV): Bretschneider (1986)

(V): Chanson (1988)

(VI): Rutschmann (1988)

(--): Unavailable information

For aerator geometries with a ramp, the experiments of SHI et al. (1983), COLEMAN et al. (1983) and CHANSON (1988) suggest that cavity filling occurs for Froude numbers less than a critical value:

$$Fr_o < Fr^{sub} = f_1\left(\phi; \frac{t_r}{d_o}; \alpha; \frac{t_s}{d_o}\right) \quad (12)$$

where  $Fr^{sub}$  is the critical Froude number at which filling occurs. Figure 4 shows: 1) the critical conditions of filling  $Fr^{sub}$  at which cavity filling starts, obtained by SHI et al. (1983) and CHANSON (1988), and 2) two experimental observations of filled cavities reported by COLEMAN et al. (1983) (i.e.  $Fr_o < Fr^{sub}$ ).



For aerators with an offset and no ramp (i.e.  $t_r = 0$ ,  $\phi = 0$ ), the experiments of TAN (1984) and CHANSON (1988) indicate that filling occurs for flow depths larger than a critical value: i.e. if

$$\frac{t_s}{d_0} < \frac{t_s}{(d_0)_{sub}} = f_2(\alpha) \quad (13)$$

where  $(d_0)_{sub}$  is a characteristic depth above which filling occurs. For a channel slope of about 52 degrees, the data of TAN (1984) and CHANSON (1988) indicate that:  $t_s/(d_0)_{sub} = 0.61$ . Their experiments were performed in the same channel but with different approach flow conditions (CHANSON 1990). The upstream flow conditions were more turbulent for TAN's experiments. The identical results obtained by TAN (1984) and CHANSON (1988) show a good consistency in the data and reproductibility of the submergence process for that particular slope.

Table 2. Experimental data of extreme cavity subpressures.

Aerator geometry	$d_0$ (m)	$Fr_0$	$\frac{Q_{air}^{inlet}}{Q_w}$	$P_N$	$P_N^{sub}$ (a)	Comments
(1)	(2)	(3)	(4)	(5)	(6)	(7)
Cavity filling CH1	0.105	4.02	0.0	0.611 <sup>(b)</sup>	1.044	Before filling.
		4.02	0.0	1.728 <sup>(c)</sup>	1.044	During filling.
	0.107	4.63	0.0	1.33 <sup>(b)</sup>	1.240	Before filling.
Cavity not filled CH1	0.060	4.63	- 0	0.307	1.195	Cavity not filled.
	0.077	4.39	- 0	1.145	1.145	filled. (d)
	0.77	4.44	- 0	1.165	1.165	
CH2	0.051	10.72	0.033	1.032	1.605	Cavity not filled.
	0.051	11.01	0.039	1.017	1.650	
	0.051	12.43	0.044	1.369	1.888	
	0.051	14.56	0.041	1.941	2.220	
	0.067	5.60	0.029	0.286	0.695	
	0.067	6.73	0.023	0.461	0.911	
	0.067	7.53	0.024	0.621	1.052	
	0.067	8.44	0.028	0.796	1.212	
	0.067	10.36	0.037	1.287	1.522	
	0.067	12.32	0.035	1.955	1.821	
0.082	7.15	0.027	0.463	0.985		
0.082	9.64	0.031	1.125	1.383		
TA1	0.05	10.3	0.0	1.52	2.35	Cavity not filled.
	0.05	10.7	0.0	1.0	1.6	
	0.05	15.4	0.07	0.760	1.534	
	0.10	5.80	0.037	0.360	0.752	
	0.10	8.05	0.032	0.540	1.115	
RU1	0.10	8.71	0.058	0.960	1.152	Cavity not filled.

(a): computed using equations (1), (4) and (10), (b): recorded immediately before cavity filling, (c): recorded during the filling of the cavity for the same run as (b), (d): air inlets not perfectly sealed

For ventilated cavities behind an offset in horizontal channels, LAALI and MICHEL (1984) reported cavity filling when the air inlets are fully-closed (i.e.  $Q_{air}^{inlet} = 0$ . No quantitative measurements near cavity disappearance were reported).

### 3.2 Discussion

The filling of an aerator cavity can occur by an increase of the flow depth or a decrease of the Froude number. As a result, the filling of the cavity may occur for discharges lower than the design



discharges as observed by COLEMAN et al. (1983). The data of COLEMAN et al. (1983) (for filled cavities) indicate Froude numbers that are larger than the other model data. Experimental investigations performed with a wide range of channel slopes and aerator configurations (e.g. LOW 1986, RUTSCHMANN 1988) showed that the cavity subpressure number  $P_N$  for small or zero air discharges increases with decreasing ramp angle  $\phi$ , for identical flow conditions. As a result, cavity filling of aerators with flat ramps (e.g. COLEMAN et al. 1983) is expected to occur at larger Froude numbers than for aerators with a large ramp angle (e.g. SHI et al. 1983, CHANSON 1988).

Table 3. Conditions for cavity filling.

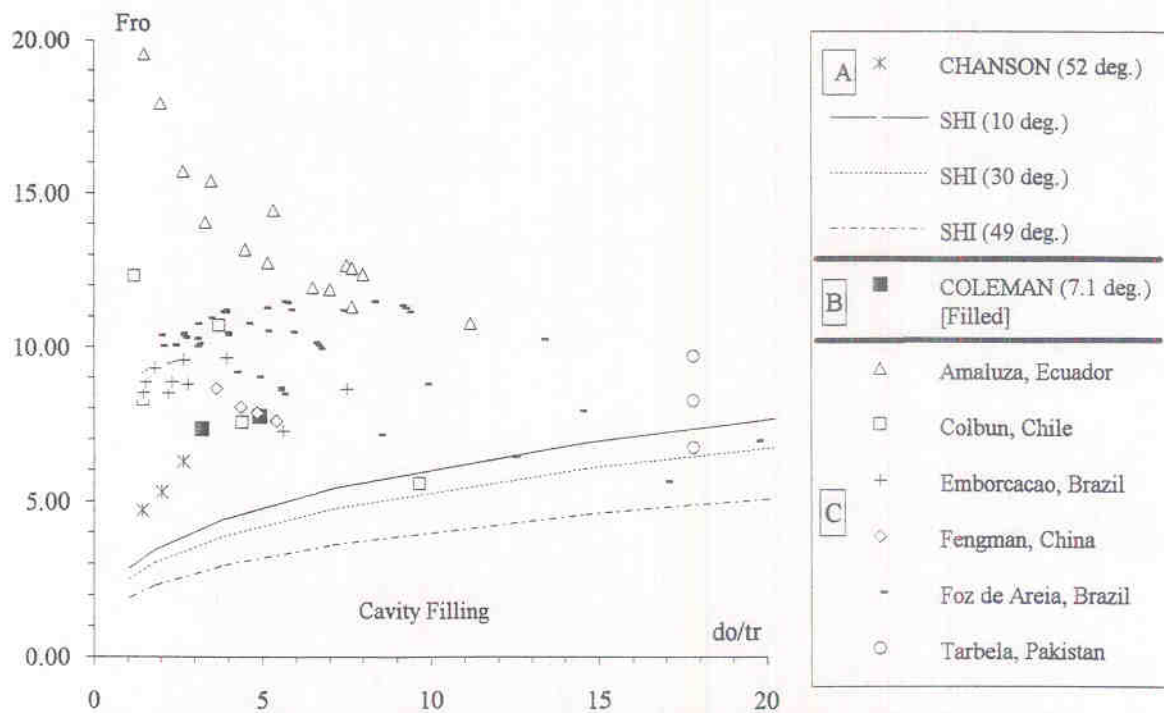
Ref. (1)	Aerator geometry (2)	Condition for filling (3)	Remarks (4)
SHI et al. (1983)	Aerator with ramp SH1, SH2, SH3, SH4	$\frac{t_r}{(R_H)_0} < 23.5 \left( \frac{\cos\alpha \cos\phi}{\frac{V_0}{\sqrt{g(R_H)_0}}} \right)^3$	For a wide channel it yields: $Fr_0 < 2.86 \cos\alpha \cos\phi \sqrt[3]{d_0/t_r}$
CHANSON (1988)	CH1	$Fr_0 < 2.77 + 0.94 \frac{d_0}{t_s}$	
TAN (1984)	Aerator without ramp TA1	$\frac{t_s}{d_0} < 0.6$	Experiments performed in 1983.
CHANSON (1988)	CH1	$\frac{t_s}{d_0} < 0.62$	Experiments performed in 1986-87.
BRETSCHNEIDER (1986)	Horizontal groove BR1	$Fr_0 < 5.8$	
ROBERTSON (1965)	Cavity behind blunt body Very blunt bodies (normal discs and plates)	$\sigma < 0.5$	Note: $\sigma = 2 \frac{\cos\alpha + P_{atm}/(\rho_w g d_0)}{Fr_0^2}$

Notes:  $(R_H)_0$ : hydraulic radius in the approach flow conditions

For aerator geometries with a ramp and no offset (i.e.  $t_s = 0$ ), operating conditions of existing prototype aerators (Table 4) are presented also on Figure 4. The flow conditions for the spillways of Amaluza, Colbun, Emborcacao, Fengman, Foz de Areia and Tabela are operating flow conditions for which no cavity filling is observed. Figure 4 would suggest some discrepancies between the observations on spillway models (Tables 1 and 3) and the operating conditions of prototypes. On prototypes, the air supply system of the aerator was either fully open or half-open while the data of SHI et al. (1983) and CHANSON (1988) were obtained with fully closed air ducts. Therefore these model results are more pessimistic than prototype conditions.

Table 4. Geometry of prototype spillway aerators.

Spillway (1)	Channel slope $\alpha$ (deg.) (2)	Ramp angle $\phi$ (deg.) (3)	Ramp height $t_r$ (m) (4)	Offset height $t_s$ (m) (5)	Reference (6)
Amaluza, Ecuador	67.5	5.7	0.06	0	RUTSCHMANN (1988)
Colbun, Chile	27.1	9.8	0.22	0	ALVARDO et al. (1988)
Emborcacao, Brazil	10.2	7.12	0.30 0.20	0	RUTSCHMANN (1988)
Fengman, China	52.0	1.15	0.2	0	ZHOU and WANG (1988)
Foz de Areia, Brazil	14.5	7.12	0.19 0.14 0.10	0	PINTO et al. (1982)
Tabela, Pakistan	0	7.12	0.14	0	PINTO (1989)



A: critical conditions of cavity filling (i.e.  $Fr_o = Fr^{sub}$ ): SHI et al.(1983), CHANSON (1988)  
 B: operating condition with filling (i.e.  $Fr_o = Fr^{sub}$ ): COLEMAN et al. (1983)  
 C: operating conditions on prototype without filling (i.e.  $Fr_o > Fr^{sub}$ ): AMULAZA, COLBUN, EMBORCACOA, FENGMAN, FOZ DE AREIA, TARBELA

Fig. 4. Conditions of filled cavities for aerators with ramp.

In term of model-prototype agreement, the present investigation suggests that the mechanism of cavity filling depends upon the cavity subpressure and its effect on the jet and roller calculations, but not upon the air entrainment processes. For scaling purpose, the relevant dimensionless parameter is the pressure gradient number  $P_N$ .  $P_N$  is proportional to the average pressure gradient in the direction normal to the streamlines. With spillways located in high mountains, the effect of the atmospheric pressure must be taken also into account. Indeed, for atmospheric pressures ranging from  $1 \times 10^4$  Pa to  $2 \times 10^4$  Pa, LAALI and MICHEL (1984) showed that, for a given pressure gradient number  $P_N$  and Froude number  $Fr$ , the air inlet discharge decreases when the ambient pressure above the jet  $P_o$  decreases. Their results suggest that the performances of aerator located at high altitude may be affected by the lower atmospheric pressure.

In the previous section, we have shown that equation (1) would enable a reasonable prediction of cavity filling if the cavity subpressure could be predicted as a function of the aerator geometry and flow conditions. In practice, the cavity subpressure cannot be predicted theoretically. The information summarised on Figure 4 can be considered as a conservative set of guidelines for aerators with ramps; it indicates that flow conditions with Froude numbers less than 7 to 8 should be investigated with a physical model to verify the risks of cavity filling.

#### 4 Conclusion

For low flow rates and partial or complete obstruction of the air supply system, the air cavity above an aerator can fill and the aeration device stops preventing cavitation damage. Further, a filled-cavity deflector might act as a cavitation generator, even at low Froude numbers.

Cavity filling occurs when the cavity subpressure exceeds a critical value (eq. (1)). The critical pressure number  $P_N^{sub}$  for the onset of cavity filling can be calculated from the aerator geometry



and the flow conditions (eqs. (1), (4) and (10)). Comparison with experimental observations suggests that these calculations provide a reasonable prediction of cavity filling.

In practice, for the design of new aerators, it is not presently possible to predict the cavity subpressure without model experiments. In such cases, the experience of cavity filling presented in this paper, for ten aerator geometries, may provide information to avoid the filling. For an aerator with a ramp, the cavity filling occurs when the Froude number becomes less than a critical value of about 7 to 8 (Fig. 4). For an aerator with an offset and without a ramp, the filling of the cavity occurs above a critical value of the flow depth.

Most experimental observations of cavity filling were obtained on spillway models. Further experimental data on prototypes are required to check possible scale effects.

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## List of symbols

- $d$  1. flow depth (m) measured perpendicular to the channel bottom;  
2. water jet thickness (m) measured perpendicular to the jet streamlines;
- $d_{impact}$  water jet thickness (m) at the jet impact, measured perpendicular to the jet streamlines (Fig. 1);
- $d_o$  depth of flow at the end of the approach flow region (m);
- $Fr$  Froude number defined as:  $Fr = q_w \sqrt{gd^3}$ ;
- $g$  gravity constant ( $m/s^2$ );
- $L_g$  groove length (m);
- $L_{jet}$  jet length (m);
- $L_R$  roller length (m);
- $P_{atm}$  atmospheric pressure (Pa): i.e. pressure above the flow;
- $P_N$  pressure gradient number defined as:  $P_N = \Delta P / (\rho_w g d)$
- $Q_{air}^{inlet}$  air discharge ( $m^3/s$ ) supplied by the air ducts;
- $Q_w$  water discharge ( $m^3/s$ );
- $q_w$  water discharge per unit width ( $m^2/s$ );
- $(R_H)_o$  hydraulic radius (m) at the end of the approach flow region;
- $t$  time (s);
- $t_r$  ramp height (m);
- $t_s$  offset height (m);
- $V$  velocity (m/s);
- $V_o$  velocity (m/s) at the end of the approach flow region;
- $W$  channel width (m);
- $\alpha$  spillway slope;
- $\Delta P$  difference between the pressure above the flow and the air pressure in the cavity
- $\phi$  ramp angle;
- $\theta$  angle of the water jet with the channel bottom;
- $\rho_w$  water density ( $kg/m^3$ );
- $\sigma$  cavitation index (also called Thoma number);

### *Subscript*

- impact flow conditions at the end of the water jet, near the impact of the jet with the spillway bottom;
- o initial flow conditions;

### *Superscript*

- sub onset of cavity filling.

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