

Thermodynamically consistent dynamic formulation of discrete thermoviscoelastic elements

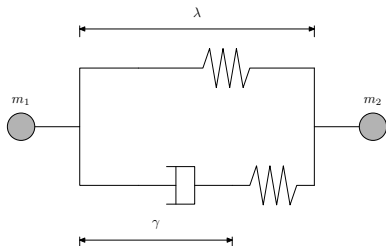
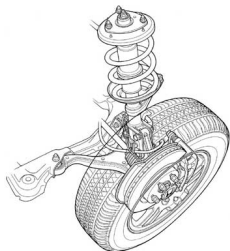
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Universidad Politécnica de Madrid
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ECCOMAS MBD 2011, Brussels

Motivation

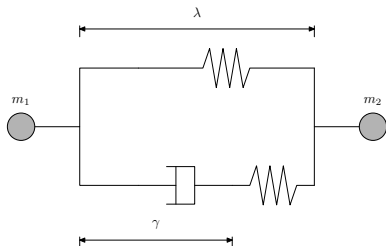
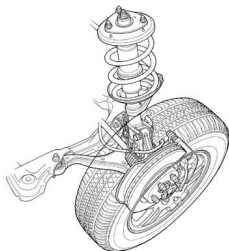
- Multibody systems incorporating discrete viscoelastic elements. Example: vehicle shock absorber devices, spatial mechanisms, etc.



- Temperature effects
 - ▶ Heat generated by viscous dissipation can flow to the environment and/or change the temperature of the element itself
 - ▶ Damping and stiffness may depend on temperature
 - ▶ Thermal expansion
 - ▶ ...

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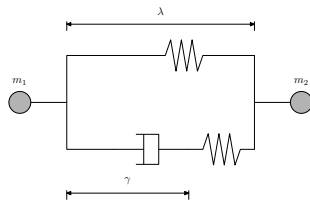
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Formulation

Positions:	$\mathbf{q}_1, \mathbf{q}_2$
Length:	$\lambda = \mathbf{q}_1 - \mathbf{q}_2 $
Viscous stretch:	γ
Element temperature:	θ
Ambient temperature: (constant)	θ_r



- Free energy function ¹ :

$$\psi(\lambda, \gamma, \theta) = \psi^\infty(\lambda, \theta) + \Gamma(\lambda, \gamma, \theta)$$

- Legendre transform: $\psi(\lambda, \gamma, \theta) \rightarrow e(\lambda, \gamma, s)$

$$\text{with } s = -\frac{\partial \psi(\lambda, \gamma, \theta)}{\partial \theta} \rightarrow \theta = \hat{\theta}(\lambda, \gamma, s)$$

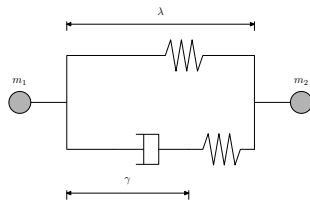
- Internal energy:

$$e(\lambda, \gamma, s) = \psi(\lambda, \gamma, \hat{\theta}(\lambda, \gamma, s)) + \hat{\theta}(\lambda, \gamma, s)s$$

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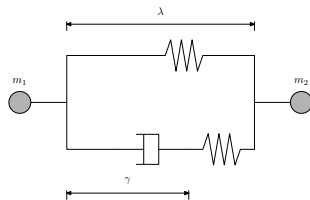
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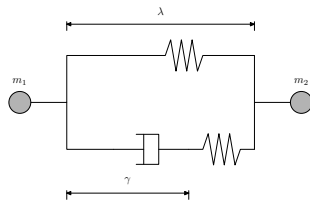
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Formulation

- Evolution equations in entropy form

$$\dot{\mathbf{q}}_1 = \frac{1}{m_1} \mathbf{p}_1$$

$$\dot{\mathbf{q}}_2 = \frac{1}{m_2} \mathbf{p}_2$$

$$\dot{\mathbf{p}}_1 = -f \frac{\mathbf{q}_1 - \mathbf{q}_2}{\lambda}$$

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$$\dot{\gamma} = \frac{1}{\eta} g$$

$$\dot{s} = \frac{1}{\theta} \left(\frac{g^2}{\eta} - h \right)$$

with

$$f = \frac{\partial e}{\partial \lambda}, \quad g = -\frac{\partial e}{\partial \gamma}, \quad \theta = \frac{\partial e}{\partial s}, \quad h = c(\theta - \theta_r)$$

damping of dashpot: $\eta(\theta)$

thermal conductivity: $c > 0$

Formulation

This continuous mathematical model satisfies:

- Non-negative dissipation (Clausius-Plank) condition:

$$\mathcal{D} = f\dot{\lambda} - \dot{\psi} - s\dot{\theta} = g^2/\eta \geq 0$$

- First law of thermodynamics: total energy of a closed system (element + ambient) must be constant
- Second law of thermodynamics: total entropy of a closed system (element + ambient) must not decrease.
- Symmetries: Conservation of linear and angular momentum (under free motion)

We say that a **discrete** model that satisfies the previous relations is **thermodynamically consistent**

Standard time integration methods may not be thermodynamically consistent for moderate time steps.

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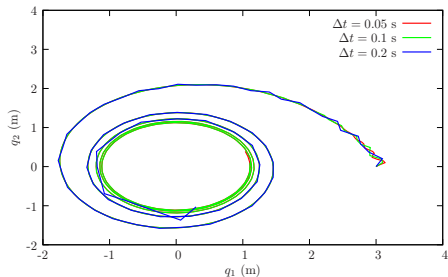
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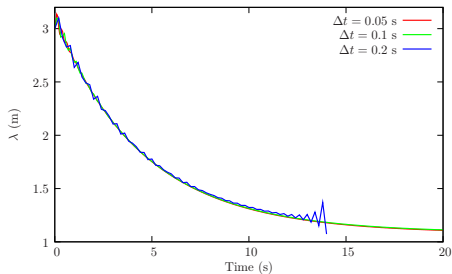
Numerical experiment

Standard integrator (midpoint rule)

Single particle with mass m connected by a thermoviscoelastic element with length 1 m to a fixed point in space.



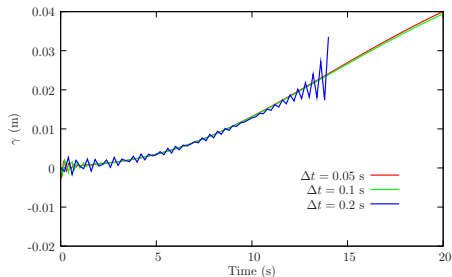
Trajectory



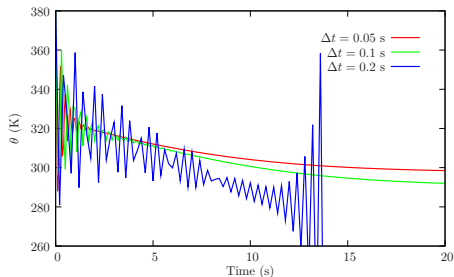
Length (λ) vs. time

Numerical experiment

Standard integrator (midpoint rule)



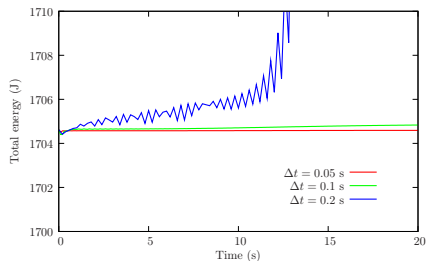
Viscous stretch (γ) vs. time



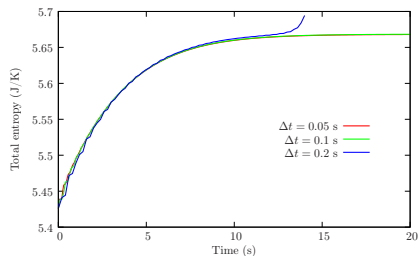
Temperature (θ) vs. time

Numerical experiment

Standard integrator (midpoint rule)



Total energy (E) vs. time



Total entropy (S) vs. time

Objective and methodology

- **Objective:** For discrete thermoelastic elements, define a time integration scheme accurate and thermodynamically consistent

Expected superior stability and long term accuracy compared standard integrators.

- **Methodology:** Geometric or structure preserving integrator.
(*warning: non Hamiltonian system !*)

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Methodology

Hamiltonian systems

→ Hamilton equations ($\dot{z} = \mathbf{J}\nabla H$)

→ Energy-Momentum methods

Non-Hamiltonian systems

→ GENERIC ² ($\dot{z} = \mathbf{L}\nabla E + \mathbf{M}\nabla S$)

→ Energy-Entropy-Momentum methods ³

First time applied to a dissipative system with internal variables.

²Ottinger, H. *Beyond equilibrium thermodynamics*. Wiley, 2005

³Romero, I. *Thermodynamically consistent time-stepping algorithms for non-linear thermomechanical systems*. IJNME, 79(706-732), 2009

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Energy-Entropy-Momentum (EEM)

- Discrete derivative operator.⁴ in a partitioned case, for a smooth $f : \mathbb{R}^N \rightarrow \mathbb{R}$:

$$\mathbf{D}f(\mathbf{x}, \mathbf{y}) \cdot \mathbf{u} = \sum_{i=1}^N \mathbf{D}^i f(x_i, y_i) \cdot u_i \quad , \quad \text{for } \mathbf{x}, \mathbf{y} \in \mathbb{R}^N, \mathbf{u} \in \mathbb{R}^N$$

verifying:

$$\text{Directionality} \quad \mathbf{D}f(\mathbf{x}, \mathbf{y}) \cdot (\mathbf{y} - \mathbf{x}) = f(\mathbf{y}) - f(\mathbf{x})$$

$$\text{Consistency} \quad \mathbf{D}f(\mathbf{x}, \mathbf{y}) = Df\left(\frac{\mathbf{x} + \mathbf{y}}{2}\right) + \mathcal{O}(\|\mathbf{y} - \mathbf{x}\|)$$

Operator D denotes the standard derivative.

⁴González, O. *Design and analysis of conserving integrators for nonlinear hamiltonian systems with symmetry*. PhD Thesis, Stanford Univ., Dep. of Mechanical Engineering (1996)

EEM scheme

$$\dot{\mathbf{q}}_1 = \frac{1}{m_1} \mathbf{p}_1$$

$$\dot{\mathbf{q}}_2 = \frac{1}{m_2} \mathbf{p}_2$$

$$\dot{\mathbf{p}}_1 = -f \frac{\mathbf{q}_1 - \mathbf{q}_2}{\lambda}$$

$$\dot{\mathbf{p}}_2 = -f \frac{\mathbf{q}_2 - \mathbf{q}_1}{\lambda}$$

$$\dot{\gamma} = \frac{1}{\eta} g$$

$$\dot{s} = \frac{1}{\theta} \left(\frac{g^2}{\eta} - h \right)$$

with

$$f = \partial e / \partial \lambda$$

$$g = -\partial e / \partial \gamma$$

$$\theta = \partial e / \partial s$$

and $\eta = \eta(\theta)$, $h = c(\theta - \theta_r)$

→

→

→

$$\frac{\mathbf{q}_{1,n+1} - \mathbf{q}_{1,n}}{\Delta t} = \frac{1}{m_1} \mathbf{p}_{1,n+1/2}$$

$$\frac{\mathbf{q}_{2,n+1} - \mathbf{q}_{2,n}}{\Delta t} = \frac{1}{m_2} \mathbf{p}_{2,n+1/2}$$

$$\frac{\mathbf{p}_{1,n+1} - \mathbf{p}_{1,n}}{\Delta t} = -f^* \frac{(\mathbf{q}_1 - \mathbf{q}_2)_{n+1/2}}{\lambda_{n+1/2}}$$

$$\frac{\mathbf{p}_{2,n+1} - \mathbf{p}_{2,n}}{\Delta t} = -f^* \frac{(\mathbf{q}_2 - \mathbf{q}_1)_{n+1/2}}{\lambda_{n+1/2}}$$

$$\frac{\gamma_{n+1} - \gamma_n}{\Delta t} = \frac{1}{\eta^*} g^*$$

$$\frac{s_{n+1} - s_n}{\Delta t} = \frac{1}{\theta^*} \left(\frac{g^{*2}}{\eta^*} - h^* \right)$$

$$f^* = \mathbf{D}^\lambda e(\lambda_n, \lambda_{n+1})$$

$$g^* = -\mathbf{D}^\gamma e(\gamma_n, \gamma_{n+1})$$

$$\theta^* = \mathbf{D}^s e(s_n, s_{n+1})$$

$$\eta^* = \eta(\theta^*), \quad h^* = c(\theta^* - \theta_r)$$

EEM Scheme

A closer look at the partitioned discrete derivative:

$$\begin{aligned} D^\lambda e(\lambda_n, \lambda_{n+1}) &= \frac{e(\lambda_{n+1}, \gamma_{n+1}, s_{n+1}) - e(\lambda_n, \gamma_{n+1}, s_{n+1})}{2(\lambda_{n+1} - \lambda_n)} \\ &+ \frac{e(\lambda_{n+1}, \gamma_n, s_n) - e(\lambda_n, \gamma_n, s_n)}{2(\lambda_{n+1} - \lambda_n)} \end{aligned}$$

$$D^\gamma e(\gamma_n, \gamma_{n+1}) = \dots$$

$$D^s e(s_n, s_{n+1}) = \dots$$

EEM Scheme

It can be shown that this scheme satisfies:

- Non-negative viscous dissipation: $\mathcal{D}_n = g_n^2/\eta_n \geq 0$, $\forall n$
- First law of thermodynamics in discrete form: $E_{n+1} - E_n = 0$

$$\text{with } E = \frac{1}{2m_1} \mathbf{p}_1^2 + \frac{1}{2m_2} \mathbf{p}_2^2 + e(\lambda(\mathbf{q}_1, \mathbf{q}_2), \gamma, s) + \sigma\theta_r$$

- Second law of thermodynamics in discrete form: $S_{n+1} - S_n \geq 0$

$$\text{with } S = s + \sigma$$

- Symmetries: linear and angular momentum (under free motion):
 $\mathbf{L}_{n+1} - \mathbf{L}_n = \mathbf{0}$, $\mathbf{J}_{n+1} - \mathbf{J}_n = \mathbf{0}$

$$\text{with } \mathbf{L} = \mathbf{p}_1 + \mathbf{p}_2 \quad \text{and} \quad \mathbf{J} = \mathbf{q}_1 \times \mathbf{p}_1 + \mathbf{q}_2 \times \mathbf{p}_2$$

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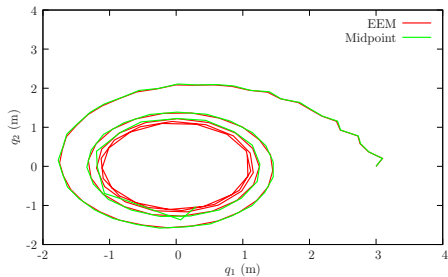
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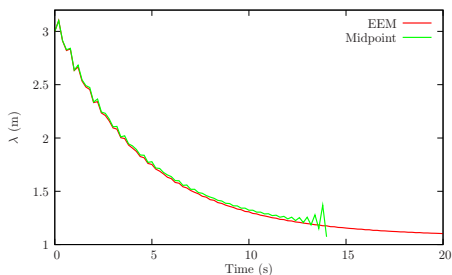
Numerical experiment

EEM vs. midpoint

$$\Delta t = 0,2 \text{ s}$$



Trajectory

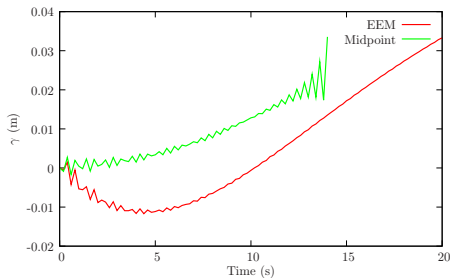


Length (λ) vs. time

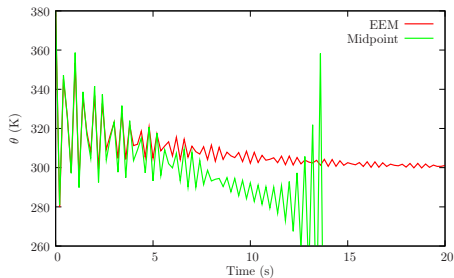
Numerical experiment

EEM vs. midpoint

$$\Delta t = 0,2 \text{ s}$$



Viscous stretch (γ) vs. time

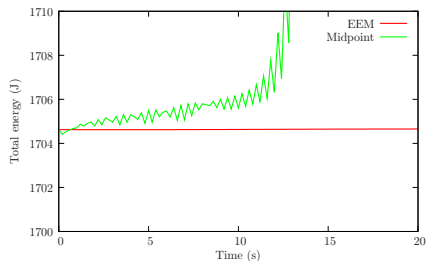


Temperature (θ) vs. time

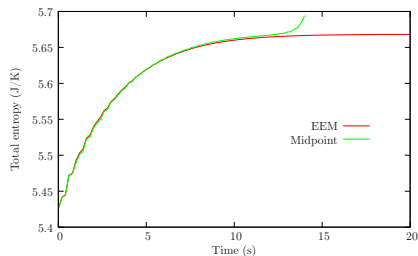
Numerical experiment

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$$\Delta t = 0,2 \text{ s}$$

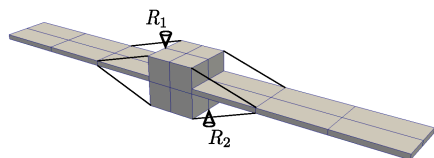


Total energy (E) vs. time

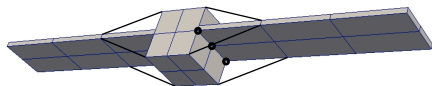


Total entropy (S) vs. time

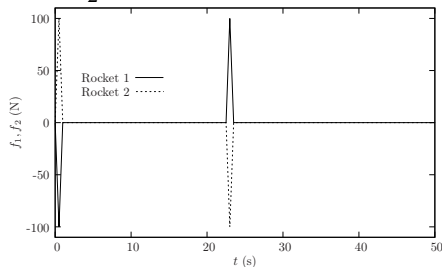
Application: satellite maneuver



(a) Upper view, showing the rockets R_1 and R_2



(b) Lower view, showing the hinges of the right solar panel with the satellite body.



Rocket forces vs. time

[movie]

Application: satellite maneuver

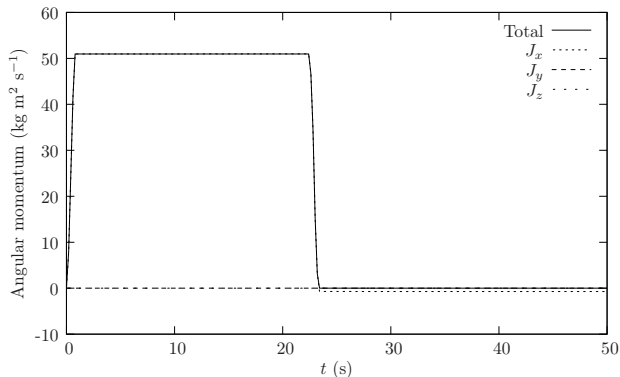


Figura: Angular momentum vs. time. EEM, $\Delta t = 0,2$ s

Application: satellite maneuver

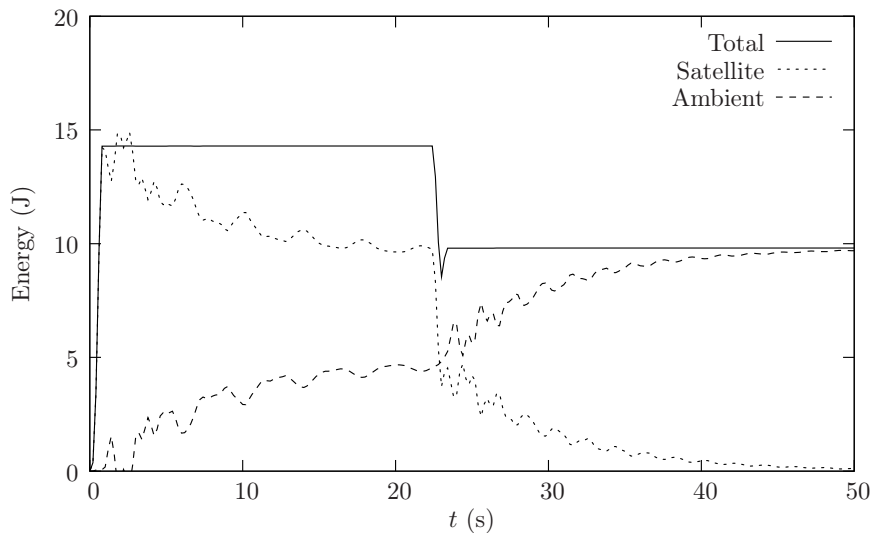


Figura: Energy vs. time. EEM, $\Delta t = 0,2$ s

Application: satellite maneuver

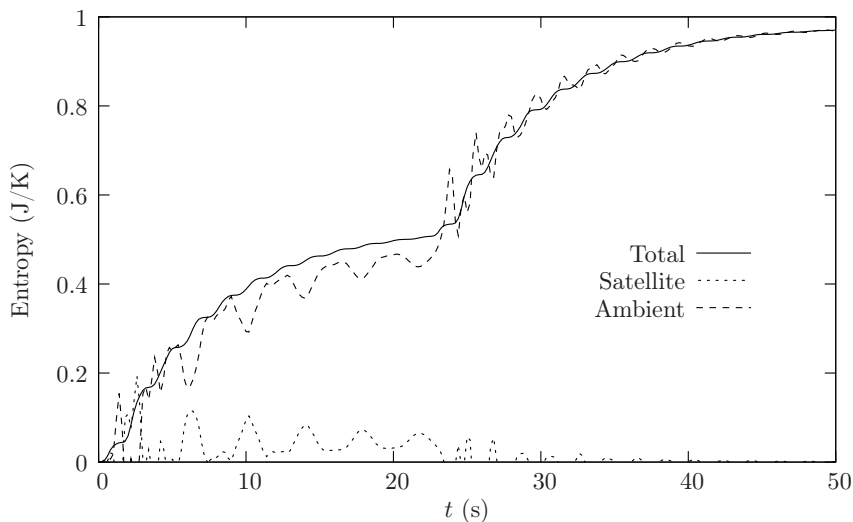


Figura: Entropy vs. time. EEM, $\Delta t = 0,2$ s

Application: satellite maneuver

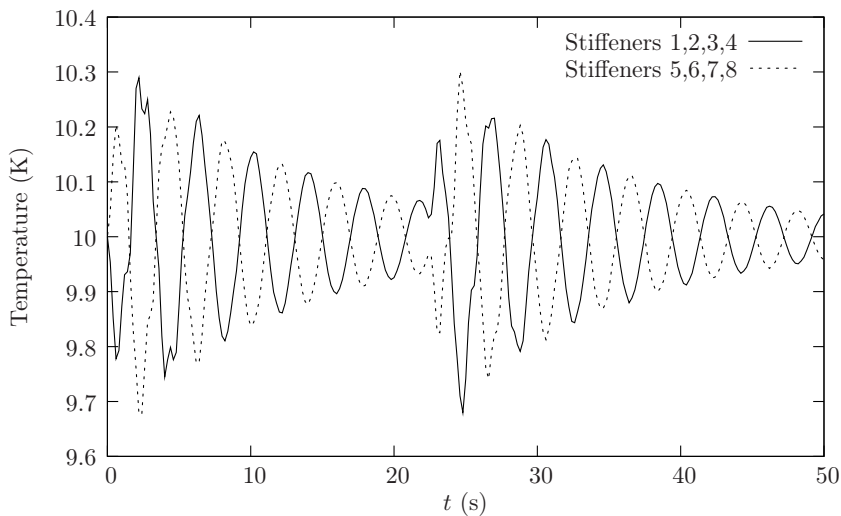


Figura: Stiffeners' temperature vs. time. EEM, $\Delta t = 0,2$ s

Conclusions

- Presented novel procedure to represent the nonlinear dynamics of a discrete viscoelastic element with temperature effects
- Large displacements, non-linear elastic behaviour and temperature-dependent mechanical properties
- Energy-entropy-momentum method: complies with first and second law of thermodynamics in a discrete form and preserves symmetries.
- First time applied to a dissipative system with internal variables.
- Numerical experiments suggest that proposed method possesses superior stability compared to typical implicit methods
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Thermodynamically consistent dynamic formulation of discrete thermoviscoelastic elements

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