# EUR 5049 e

COMMISSION OF THE EUROPEAN COMMUNITIES

### CODE TAFEST

# NUMERICAL SOLUTION TO TRANSIENT HEAT-CONDUCTION PROBLEMS USING FINITE ELEMENTS IN SPACE AND TIME

by

J. DONEA and S. GIULIANI

1974



Joint Nuclear Research Centre Ispra Establishment - Italy

**Materials Division** 

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Commission of the European Communities Joint Nuclear Research Centre - Ispra Establishment (Italy) Materials Division Luxembourg, February 1974 - 42 Pages - 9 Figures - B.Fr. 60.—

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#### ABSTRACT

The present report describes the computer code TAFEST that has been developed for the purpose of solving two-dimensional transient heatconduction problems. The concept of finite elements in space and time is used as a means of obtaining numerical responses.

### **KEYWORDS**

T-CODES IBM COMPUTERS FORTRAN FINITE ELEMENT METHOD TRANSIENTS THERMAL CONDUCTION TIME DEPENDENCE SPACE TWO-DIMENSIONAL CALCULATIONS ACCURACY USES

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### 1. INTRODUCTION

Within the frame of the finite element method, solutions to the transient heat-conduction equation are governed by a system of first-order linear differential equations of the form<sup>1</sup>:

$$\begin{bmatrix} K \end{bmatrix} \{T(t)\} + \begin{bmatrix} C \end{bmatrix} \{T^{*}(t)\} = \{F(t)\}$$
(1.a)  
$$\{T(0)\} = \{T_{0}\}$$
(1.b)

where  $\{T(t)\}$  denotes the temperature vector,  $\{F(t)\}$ is the 'load' vector,  $\_K\_7$  is the conductivity matrix and  $\_C\_7$  the heat-capacity matrix. The vector  $\{To\}$  specifies the initial values of the temperature.

The differential system (1) can be integrated numerically with the aid of a digital computer. The most critical step is, of course, to choose an integration method that combines efficiency and accuracy.

In this report, the concept of finite elements in space and time is used as a means of integrating the differential system (1). The basic variational formulation involving both time and space variables is described by reference to the Galerkin process.

Although various elements of the space-time domain can easily be derived<sup>2</sup>, the only element described here is a right triangular prism of the (x, y, t) domain. This element is shown to lead to a better short-time accuracy than the Crank-Nicholson scheme used by Wilson-Nickell<sup>3</sup>, Zienkiewicz-Parekh<sup>4</sup> and Fullard<sup>5</sup>.

The main features of the computer code TAFEST are described in the last part of the report. This code was developed for the purpose of solving two-dimensional transient heat-conduction problems by means of the indicated space-time element. A typical example has been included in order to show the type of results that can be obtained on using the code.

### 2. BASIC VARIATIONAL EQUATION

Let it be required to solve the transient heat-conduction equation

k div(grad T)+Q(x,y,z,t) - gc 
$$\frac{\partial T}{\partial t} = 0$$
 (2)

in a domain V bounded by a surface S

In order to formulate a variational problem associated with eq. (2), we multiply it by an arbitrary admissible temperature variation  $\delta T$  and use the property

div 
$$(a\vec{B}) = a div \vec{B} + \vec{B} grad a$$
 (3)

Such a manipulation indicates that

div (k grad T 
$$\delta$$
T) - k grad T grad  $\delta$ T + Q  $\delta$ T -  $c \frac{\partial I}{\partial t} \delta$ T = 0 (4)

We now integrate eq. (4) over the domain V and the time t and transform the volume integral for the first term by means of the divergence theorem. This enables the order of the partial derivatives to be reduced and yields :

$$-\iint_{t \in S} k \frac{\partial T}{\partial n} \delta T \, dSdt + \iint_{t \in V} k \, grad T \, grad \delta T \, dV \, dt -$$

$$-\iint_{t} Q \ \delta T \ dV \ dt + \iint_{t} Q c \ \frac{\partial T}{\partial t} \ \delta T \ dV \ dt = 0$$
(5)

with n denoting the outward unit normal to S

Since eq. (5) holds for an arbitrary temperature-variation
6T , eq. (1) is also satisfied. The variational equation
(5) can thus be used as a basis for a numerical solution of
transient-conduction problems.

The main problem in solving eq. (5) consists in the definition of suitable finite elements in the space-time domain. For any such element, the local field will be represented in the form

$$T(x,y,z,t) = \sum_{i=1}^{M} N_i(x,y,z,t) T_i$$
 (6)

where the modes Ni depend on space and time, while the M nodal values Ti are independent on the coordinates x, y,z,t. The characteristic equations for the element are obtained by introduction of the local representation (6) into the basic variational equation (5).

The assembly of the various elements appearing in the discretization of the space-time domain follows the usual rules of the finite element method.

### 3. A RIGHT TRIANGULAR PRISM IN THE (x, y, t) DOMAIN

Although various elements in space and time can easily be derived<sup>2</sup>, we shall concentrate on the particular element that has been choosen for the computer code TAFEST to be described in section 5.

### 3.1. Choice of the local temperature field

The right triangular prism represented in Fig. 1 has six degrees of freedom. In order to ensure the continuity of the temperature on the interfaces between the various elements, the local temperature field is choosen in the form :

$$T^{e}(x,y,t) = a + bx + cy + dt + ext + fyt$$
(7)

In function of the nodal parameters (Fig. 1)

$$\{T^{e}\}^{e} = (T_{i}, T_{j}, T_{k}, T_{l}, T_{m}, T_{n})$$
 (8)

the local field can be written as

$$T(x,y,t) = \left[N_{i}, N_{j}, N_{k}, N_{l}, N_{m}, N_{n}\right] \left\{T^{e}\right\}$$
(9)

with

$$N_i = M_i(t_l - t)$$
;  $N_j = M_j(t_l - t)$ ;  $N_k = M_k(t_l - t)$   
 $N_l = M_i(t - t_i)$ ;  $N_m = M_i(t - t_i)$ ;  $N_m = M_k(t - t_i)$ 

 $M_i = \frac{a_i + b_i x + c_i y}{2 V}$ ; V = Volume of the prismatic element.

 $a_i = x_j y_k - x_k y_j$ ;  $b_i = y_j - y_k$ ;  $c_i = x_k - x_j$ 

The modes  $M_j$  and  $M_k$  are obtained by cyclic permutation of the indexes in the order i, j, k. The parameters  $t_i$  and  $t_i$  define the time interval spanned by the element.

### 3.2. Element characteristics

The governing equations for the element are obtained through introduction in eq. (5) of independent variations on the six nodal parameters (8). Such an operation yields the following relationship :

$$\begin{bmatrix} H^e \end{bmatrix} \{T^e\} = \{F^e\}$$

(10)

$$\left[\kappa^{e}\right] = \frac{k V}{12 A^{2}} \begin{bmatrix} ([B]+[C]) & \frac{1}{2}([B]+[C]) \\ \frac{1}{2}([B]+[C]) & ([B]+[C]) \end{bmatrix}$$
(11)

The matri trix

- 9

(15)

(12)

$${F^{e}}^{*} = (F_{i}, F_{j}, F_{k}, F_{l}, F_{m}, F_{n})$$

A = Area of triangle i,j,k.

where

and a heat-capacity matrix

 $\left[\mathfrak{P}^{\mathsf{e}}\right] = -\frac{\mathfrak{P}^{\mathsf{c}}}{\mathfrak{P}^{\mathsf{a}}} \begin{bmatrix} \left[\mathsf{P}\right] & -\left[\mathsf{P}\right] \\ \left[\mathsf{P}\right] & -\left[\mathsf{P}\right] \end{bmatrix}$ 

$$\begin{bmatrix} b_i b_i + c_i c_i \end{pmatrix} (b_i b_j + c_i c_j) (b_j b_k + c_i c_k) \\ \begin{bmatrix} B \end{bmatrix} + \begin{bmatrix} C \end{bmatrix} = (b_j b_j + c_j c_j) (b_j b_k + c_j c_k)$$
(13)   
 Symmetric  $(b_k b_k + c_k c_k)$ 

L L

$$\begin{bmatrix} P \end{bmatrix} = \begin{bmatrix} P_{ij} & P_{ij} & P_{ik} \\ P_{ij} & P_{jj} & P_{jk} \\ P_{ik} & P_{jk} & P_{kk} \end{bmatrix}$$
(14)

$$P_{ij} = \int (a_i + b_i x + c_i y) (a_j + b_j x + c_j y) dx dy$$

$$= \int (a_i + b_i x + c_i y) (a_j + b_j x + c_j y)$$

If the internal heat-generation Q is independent of space but varies linearly from  $Q_i$  to  $Q_1$  during the time interval  $t_1 - t_i$ , the contributed nodal loads are easily shown to be

$$\left\{F_{Q}^{e}\right\}^{*} = \frac{V}{9} \left(F_{Q}^{1}, F_{Q}^{1}, F_{Q}^{1}, F_{Q}^{2}, F_{Q}^{2}, F_{Q}^{2}\right)$$
(16)

where

$$F_Q^1 = Q_i + \frac{1}{2}Q_l$$
;  $F_Q^2 = \frac{1}{2}Q_i + Q_l$ 

3.3. Nodal loads due to the boundary conditions

### Prescribed normal heat-flux

Suppose we impose between nodal points i and j (Fig. 1) a uniform normal heat-flux which varies linearly from  $\overline{\varphi_i}$  to  $\overline{\varphi_i}$  during the time interval  $t_1 - t_i$ . The first term in eq. (5) shows that such a condition induces the nodal loads

$$\left\{F_{\varphi}^{e}\right\}^{*} = -\frac{L(t_{l}-t_{i})}{6} \left(F_{\varphi}^{1}, F_{\varphi}^{1}, 0, F_{\varphi}^{2}, F_{\varphi}^{2}, 0\right)$$
(17)

where

$$F_{\varphi}^{1} = \overline{\varphi}_{i} + \frac{1}{2} \overline{\varphi}_{l} \qquad F_{\varphi}^{2} = \frac{1}{2} \overline{\varphi}_{i} + \overline{\varphi}_{l}$$

L = Length of side i-j

### Convective heat-transfer

Suppose now we have a convective heat-transfer between nodes i and j. The heat-transfer coefficient h as well as the fluid temperature  $T_f$  vary linearly during the time interval  $t_1 - t_j$ . This type of boundary condition yields a contribution to both the matrix  $\_\ensuremath{\mathsf{-H}^e}\ensuremath{\mathsf{-7}}$  and the nodal loads  $\{F^e\}$  .

The additional terms in the matrix  $/ H^{e} / J$  are

$$\begin{bmatrix} H_{conv}^{e} \end{bmatrix} = \frac{L(t_{l} - t_{i})}{36} \begin{bmatrix} T_{1} & \frac{1}{2}T_{1} & 0 & T_{2} & \frac{1}{2}T_{2} & 0 \\ T_{1} & 0 & \frac{1}{2}T_{2} & T_{2} & 0 \\ 0 & 0 & 0 & 0 \\ T_{3} & \frac{1}{2}T_{3} & 0 \\ Symmetric & T_{3} & 0 \\ 0 \end{bmatrix}$$
(18)

where

$$T_1 = 3h_i + h_i$$
;  $T_2 = h_i + h_i$ ;  $T_3 = h_i + 3h_i$   
 $h_i = h(t_i)$ ;  $h_i = h(t_i)$ 

The nodal loads contributed by the condition of convection are

$$\left\{F_{\text{conv}}^{e}\right\}^{*} = \frac{L(t_{1} - t_{1})}{24} \left(S_{1}, S_{1}, 0, S_{2}, S_{2}, 0\right)$$
(19)

where

$$S_{1} = h_{i} (3 T_{f}^{i} + T_{f}^{l}) + h_{l} (T_{f}^{i} + T_{f}^{l})$$

$$S_{2} = h_{l} (3 T_{f}^{l} + T_{f}^{i}) + h_{i} (T_{f}^{i} + T_{f}^{l})$$

$$T_{f, I}^{l} = Fluid \text{ temperature at times } t_{i} \text{ and } t_{l}.$$

The one-dimensional example of a constant heat-flux applied to a semi-infinite solid has been analyzed in order to illustrate the achievable accuracy with the space-time element previously described.

A finite element solution for this problem is given by Wilson and Nickell <sup>3</sup> using a regular mesh with  $\Delta x = 0.2$ . The time integration is performed on the basis of a recurrence relation which can be shown to be a generalization of the Crank-Nicholson scheme<sup>2</sup>. Constant time steps  $\Delta t = 0.1$  are used. We solved the same problem by means of finite elements in space and time, i.e. with eq. (10) as the integration formula. Fig. 2 compares both numerical solutions to the exact one. As can be seen, a much better short-time accuracy is achieved with the space-time element. The reasons for this better behaviour with respect to the Crank-Nicholson scheme are fully explained elsewhere<sup>6</sup>.

### 5. THE COMPUTER CODE TAFEST

In this section we describe the main features of the computer code TAFEST. This code was developed for the purpose of solving two-dimensional transient heat-conduction problems by means of finite elements in space and time. Starting from known temperatures at time t, the last three equations in relation (10) are used as an integration scheme to yield the temperatures at time t +  $\Delta$ t. The assembled equations are solved by means of Choleski's method. A general flow chart of the programme is given on Fig. 3.

TAFEST has been written in Fortran IV language and compiled on the IBM 370/165 computer of CETIS (EURATOM C.C.R. - Ispra). In the present version, the code has a size of about 200K bytes, so that no auxiliary storage space is needed.

### 5.1.Description of input data

The input data required by TAFEST are defined here in the sequence in which they occur. References to card numbers will be found in the listing of data and formats which follows this section.

### CARD (1)

TIT The problem title in 72 alpha-numeric characters. This information is used to identify the problem in the printed output.

### CARD (2)

NUMEL	The number of triangular-shaped elements in the structure (max. 700)
NUMNP	The number of nodal points (max. 400)
NUMTM	Number of points used in the discretization of the time (Initial time included) (max. 50)
N1	Option to define the coordinate system used for in- put
	0 means Cartesian
	1 means Polar
N2	Option to define the type of the heat flow
	0 means plane
	1 means axisymmetric
CARD (3)	
N3 (I)	Option to punch temperature cards at time TM (I) $(I = 1, NUMTM)$

- 0 Print nodal temperatures but do not punch;
- 1 Print nodal point and element temperatures
  Punch the element temperatures;
- 2 Print and punch nodal point temperatures;
- 3 Print and punch element and nodal point temperatures.

### CARD (4)

- NTI Number of nodal points with prescribed temperatures (max. 100)
- NTB Number of elements with one side subject to convection (max. 100)
- NTF Number of elements with a non-zero normal heat-flux prescribed on one side (max. 100)
- NTQ Number of groups of elements with internal heat-generation (max. 100)

### CARD (5)

- COND Main thermal conductivity (w/cm-°C)
- CAPA Main heat capacity gc (Joule/cm<sup>3</sup> °C)

### CARD (6)

TM (I) Location (expressed in seconds) of the various points used in the discretisation of the time. (I = 1, NUMTM) (TM (1) is the initial time for the transient problem). Six time stations are given per card.

### CARD(7)

One card is required for each element (N = 1, NUMEL)

N The element index number

NPI (N)

NPJ (N) Index numbers of the element nodal points NPK (N)

CT (N) The effective thermal conductivity of element N is COND - CT (N) (See card (5))

CP (N) The effective heat-capacity of element N is CAPA - CP (N) (See card (5))

### CARD(8)

One card is required to describe each nodal point (M = 1 , NUMNP)

M The nodal point index number

XORD(M) The x or r-coordinate of nodal point M (mm)

YORD(M) The y or  $\Theta$ -coordinate of nodal point M (angles are given in degrees)

### $\underline{C}$ ARD (9)

J Non processed index that may be used to number the nodal points if J = I.

T (I) Initial temperature (°C) at nodal point I
 (I = 1, NUMNP; 4 nodal point temperatures are gi ven per card)

Cards (10) and (11) are repeated NTI times (I = 1, NTI).

CARD (10)

NTT (I) Index number of a node with prescribed temperature

NTMI (I) Number of points in time which are given to describe the evolution of the prescribed temperature. (Piecewise linearization of the effective

CARD(11)

TI (I, N) Nodal point temperature (°C) at time TIMI(I,N)
TIMI (I,N) Time in seconds

Three groups TI, TIMI are given per card (N = 1, NTMI (I))Cards (12) and (13) are repeated NTB times (I = 1, NTB)

### CARD (12)

М	Index number of an element subject to convec- tion heat-transfer on one side.
NTMB (I)	Number of points in time which are given to describe the evolution of the convective heat-transfer.
LI ( I)	Nodal points defining the element side
LJ (I)	Subject to convection.

### CARD (13)

H (I, N) Value of the heat-transfer coefficient  $(W/cm^2 - °C)$  at time TIMB (I, N)

temperature)

- TF (I,N) Temperature of the reference fluid (°C) at time TIMB (I, N)
- TIMB (I,N) Time in seconds

Two groups H, TF, TIMB are given per card (N = 1, NTMB (I)) Cards (14) and (15) are repeated NTF times (I = 1, NTF).

- CARD (14)
- M Index number of an element with a prescribed normal heat-flux on one side.
- NTMX (I) Number of points in time which are given to describe the evolution of the prescribed heat-flux.
- MI (I) Nodal points defining the element side with pre-MJ (I) scribed heat-flux.
- CARD (15

FLUX (I,N) Prescribed heat-flux (W/cm<sup>2</sup>) at time TIMX (I,N) TIMX (I,N) Time in seconds

Three groups FLUX, TIMX are given per card (N = 1, NTMX (I)). Cards (16) and (17) are repeated NTO times (I = 1, NTO).

- CARD (16)
- IFIRST (I) Index number of the first element in a group with internal heat-generation.
- ILAST (I) Index number of the last element in a group with internal heat-generation.

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NTMQ (I) Number of points in time which are given to describe the evolution of the internal heat generation.

### CARD(17)

Q (I,N) Heat generation  $(W/cm^3)$  at time TIMQ (I,N)

TIMQ (I,N) Time in seconds

Three groups Q-TIMQ are given per card (N = 1, NTMQ (I))



Cand	Column	1 - 7 2	
Card	Format	18 A 4	
	Symbol	TIT	

Card	Column	1-6	7 - 12	13 - 18	19 - 24	25 — 30	
	Format	16	16	16	16	16	
2	Symbol	NUMEL	NUMNP	NUMTM	N 1	N 2	

Card	Column	1	2	3	4	 79	80
caru	Format	11	11	11	I 1	 11	11
3	Symbol	N3 (1)	N3 (2)	N3 (3)	N3 (4)	 N3(79)	N3(80)

Card	Column	1-6	7 - 12	13 - 18	19-24	
Lard 4	Format	16	16	16	16	
	Symbol	NTI	NTB	NTF	NTQ	
		••••••••••••••••••••••••••••••••••••••	······································		·	TAFEST

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Card	Column	1 - 12	13 - 24	· · · · · · · · · · · · · · · · · · ·
5 Caru	Format	E12.5	E 12.5	
5	Symbol	COND	CAPA	

Card	Column	1 - 12	13 - 24	25 - 36	37 - 48	49 - 60	61 - 72
Gard	Format	E12.5	E 12.5				
U	Symbol	TM (I)	TM (I)	TM (I)	TM (I)	TM (1)	TM (1)

Card	Column	1 - 6	7 - 12	13 - 18	19 - 24	25 - 36	37 - 48	
	Format	16	16	16	16	E12.5	E 12.5	
	Symbol	N	NPI (N)	NPJ(N)	NPK (N)	CT(N)	CP(N)	

Card	Column	1-4	5-6	7 - 18	19 - 30	
Lard	Format	14	2 X	E 12.5	E 12.5	
0	Symbol	м		XORD (M)	YORD(M)	

TAFEST

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Card	Column	1 - 6	7 - 18	19 - 24	25-36	37 - 42	43-54	55-60	61 - 72	
	Format	16	E12.5	Ι6	E12.5	16	E12.5	16	E12.5	
3	Symbol	J	T(I)	J	T (I)	J	T (1)	J	T (I)	

Card 10	Column	1 - 6	7 - 12	
	Format	I 6	Ι6	
	Symbol	NTT (1)	NTMI(I)	

ſ	Card	Column	1 — 12	13 - 24	25-36	37 - 48	49 - 60	61 - 72	
	11	Format	E12.5	E12.5	E12.5	E12.5	E12.5	E12.5	
		Symbol	TI ( I,N)	TIMI (I,N)	TI(I,N)	TIMI(I,N)	TI ( I, N )	TIMI (I,N)	

Card	Column	1-6	7 - 12	13 – 18	19 – 24		e <sup>1</sup>
12	Format	16	16	Ι6	Ι6		
12	Symbol	м	NTMB(I)	LI (I)	LJ (I)		
				<u>-</u> -	<b>.</b>	TAFES	л <del>т</del>

Card	Column	1 – 12	13 - 24	25 - 36	37 - 48	49-60	61 - 72	
13	Format	E12.5	E12.5	E12.5	E12.5	E12.5	E12.5	
	Symbol	H (1,N)	TF(1,N)	TIMB(I,N)	H(I,N)	TF(1,N)	TIMB(I,N)	

Card	Column	1 - 6	7 - 12	13 - 18	19 - 24
	Format	16	16	16	16
14	Symbol	м	NTMX(I)	MI(I)	MJ (1)

Card	Column	1 12	13 - 24	25 - 36	37 - 48	49 - 60	61 72	
15	Format	E12.5	E12.5	E12.5	E12.5	E12.5	E12.5	
	Symbol	FLUX (1, N)	TIMX(I,N)	FLUX (I,N)	TIMX (I,N)	FLUX (I,N)	TIMX (I,N)	

Card	Column	1 - 6	7 – 12	13 - 18	
16	Format	16	16	16	
10	Symbol	IFIRST(1)	ILAST(I)	NTMQ(1)	
			•	·	TAF

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I Card L			13-24	25 - 36	37 - 48	49-60	61 - 72	
F 17	Format	E12.5	E12.5	E12.5	E 12.5	E12.5	E12.5	
s s	Symbol	Q (1,N)	TIMQ(I,N)	Q (I,N)	TIMQ(I,N)	Q (1,N)	TIMQ(I,N)	

TAFEST

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### 5.3. Description of the printed output

As an illustration of the printed output of TAFEST, we reproduce hereafter the results for the one-dimensional problem of a constant heat-flux applied to a semi-infinite solid (see section 4).

# 

### \*\*\* TEST TAFEST - CONSTANT HEAT FLUX APPLIED TO A SEMI-INFINITE SOLID \*\*\*\*\*\*\*\*

THIS PROBLEM IS SOLVED UNDER PLANE CONDITIONS

NUMBER UF ELEMENTS	=	38
NUMBER OF NODAL POINTS	=	40
NUMBER OF TIME POINTS	=	11
THERMAL CONDUCTIVITY (W/CMC)	=	1.0000E 00
THERMAL CAPACITY (J/CM3C)	=	1.0000E 00
NUMBER OF NODES WITH PRESCRIBED TEMPERATURE	=	0
NUMBER OF ELEMENTS WITH CONVECTION	=	0
NUMBER OF ELEMENTS WITH PRESCRIBED HEAT FLUX	#	1
NUMBER OF GROUPS OF ELEMENTS WITH HEAT GENERATION	=	0
N 1	±	0
N 2	=	0

PUINTS IN TH	E TIME	DOMAIN									
TIME (SEC)	N3	TIME (SEC)	N3	TIME (SEC)	N3	TIME (SEC)	N 3	TIME (SEC)	N3	TIME (SEC)	N3
C.O 6.(00E-01	0	1.000E-01 7.000E-01	0	2.000 E-01 8.000 E-01	0 0	3.000E-01 9.000E-01	0	4.000E-01 1.000E 00	0	5.000 E-01	0

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NODE	X-ORD (MM)	Y-ORD (MM)	NODE	X-ƏRD (MM)	Y-ORD (MM)	NODE	X-ORD (MM)	Y-ORD (MM)
14703692581470 122233470	0.0 2.00000E 00 6.0000E 00 8.00000E 00 1.20000E 01 1.40000E 01 2.00000E 01 2.00000E 01 2.6000E 01 3.00000E 01 3.00000E 01 3.80000E 01 3.80000E 01	0.0 4.00000E 00 4.00000E 00	2 58 11 14 17 23 26 29 35 38	0.0 4.0000E 00 6.00000E 01 1.20000E 01 1.60000E 01 1.60000E 01 2.20000E 01 2.40000E 01 2.40000E 01 3.00000E 01 3.40000E 01 3.60000E 01	4.00000E 00 0.0 4.00000E 00 4.00000E 00 4.00000E 00 4.00000E 00 4.00000E 00 4.00000E 00 4.00000E 00 4.00000E 00	3 92 15 18 21 27 33 36 39	2.00000E 00 4.00000E 00 8.00000E 00 1.00000E 01 1.40000E 01 2.00000E 01 2.00000E 01 2.60000E 01 2.80000E 01 3.40000E 01 3.80000E 01	0.0 4.00000E 00 4.00000E 00 4.00000E 00 4.00000E 00 4.00000E 00 4.00000E 00 4.00000E 00 4.00000E 00

ELEM.	1	J	К	CT (W/C1 C)	(J/CM3 C)	ELEM.	I	j	к	CT (W/CM C)	CP (j/cm3 c)
1 57 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 91357 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 9135577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577 913577	1357913579135791357 11179135791357 2033357	3 070145 892307 014589 11111122222333333	24 08 1024580245802468 1124580245802468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 2002468 20023 200268 200268 200268 200268 200268 200268 200268 200268 200268 200268 200268 200268 200268 200268 200268 200268 200268 200268 200268 200268 200268 200268 200268 200268 200268 200268 200268 200268 200268 20020000000000	0 0 0 0 0 0 0 0 0 0 0 0 0 0	1) • ( 0 • C 0 • C	2 4 6 8 0 2 1 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 4 6 8 0 2 2 4 6 8 0 2 2 4 6 8 0 2 2 4 6 8 0 2 2 4 6 8 0 2 2 4 6 8 0 2 2 4 6 8 0 2 2 4 6 8 0 2 2 4 6 8 0 2 2 4 6 8 0 2 2 4 6 8 0 2 2 4 6 8 0 2 2 4 6 8 0 2 2 4 6 8 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3	3877115599337711559 111112222253333	4589236701458923670	266004488226600448833333333333333333333333333333333	0,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,000,0000	0.0000000000000000000000000000000000000

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TOTAL AREA (CM 2) = 1.51999E 00

ELEM 1	TIME PUINTS 2	NODES 1 2			
FLUX (W/CM2)	TIME (SEC)	FLUX (W/CM2)	TIME (SEC)	FLUX (W/CM2)	TIME (SEC)
-1.000E 00	0.0	-1.000E 00	1.000E 02		

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NODE	TEMPERATURE (C)						
1	0.0	2	0.0	3	0.0	4	0.0
5	0. C	6	0.0	7	0.0	8	D.0
9	0.0	10	0.0	11	0.0	12	0.0
13	0.C	14	0.0	15	0.0	16	0.0
17	0.0	18	0.0	19	0.0	20	<b>0</b> .0
21	0.0	22	0.0	23	0.0	24	0.0
25	0.0	25	0.0	27	0.0	28	0.0
29	0.0	30	0.0	31	0.0	32	0.0
33	0.0	34	0.0	35	0.0	36	0.0
37	0.0	38	0.0	39	0.0	40	0.0

### INITIAL TEMPERATURES - TIME (SEC) = 0.0

DIMENSIONS OF THE MATRIX  $S = 40 \pm 4$ 

### TIME (SEC) = 1.000E-01

NDDE	TEMPERATURE (C)	NUDE	TEMPERATURE (C)	NODE	TEMPERATURE (C)	NODE	TEMPERATURE (C)
1	0.39	2	0.37	3	0.16	4	0.18
5	0.08	6	0.08	7	0.03	8	0.04
9	0.02	10	0.02	11	0.01	12	0.01
13	0.00	14	0.00	15	0.00	16	0.00
17	0.00	18	0.00	19	<b>0.00</b>	20	0.00
21	0.00	22	0.00	23	0.00	24	0.00
25	0.00	26	0.00	27	0.00	28	0.00
29	0.00	30	0.00	31	0.00	32	0.00
33	0.00	34	0.00	35	0.00	36	0.00
37	0.00	38	0.00	39	0.00	40	0.00

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#### TIME (SEC) = 2.000E-01

NODE	TEMPERATURE (C)						
1	0.49	2	6.49	3	0.33	4	0.33
5	0.20	6	0.19	7	0.11	8	0.11
9	0.06	10	0.06	11	0.03	12	0.03
13	0.02	14	0.02	15	0.01	16	0.01
17	0.00	18	0.00	19	0.00	20	0.00
21	0.00	22	0.00	23	0.00	24	0.00
25	0.00	26	0.00	27	0.00	28	0.00
29	0.00	30	0.00	31	0.00	32	0.00
33	0.00	34	0.00	35	0.00	36	0.00
37	0.00	38	0.00	39	0.00	40	0.00

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### TIME (SEC) = 3.000E-01

NODE	TEMPERATURE (C)	NUDE	TEMPERATURE (C)	NODE	TEMPERATURE (C)	NODE	TEMPERATURE (C)
1	0.62	2	0.61	3	0.43	4	0.43
5	0.29	6	0.29	7	0.19	8	0.19
9	0.12	10	0.12	11	0.07	12	0.07
13	0.04	14	0.04	15	0.02	16	0.02
17	0.01	19	0.01	19	0.01	20	0.01
21	0.00	22	0.00	23	0.00	24	0.00
25	0.00	26	0.00	27	0.00	28	0.00
29	0.00	30	0.00	31	0.00	32	0.00
33	0.00	34	0.00	35	0.00	36	0.00
37	<b>0.0</b> 0	33	0.00	39	0.00	40	0.00

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### 6. A PRACTICAL PROBLEM

We analyzed a transient heat-flow in the graphite matrix of a HTGR fuel element. As indicated by Fig. 4, the analysis is limited to the symmetric portion of the graphite matrix. Fig. 5 shows the finite element grid that has been used. The transient heat-flow is due to a power increase of the reactor. Along the interface between fuel and graphite, we prescribe a uniform normal heat-flux  $\varphi$  which increases with time as indicated on Fig. 6.

The same figure gives the evolution of the heat transfer coefficient h between graphite and coolant. The thermal properties of graphite are

k = 0.2 W/cm °C ; gc = 4.5 Joules/cm<sup>3</sup> - °C while the coolant temperature is 600°C.

The upper curves in Fig. 6 show the temperature evolution at points A and B of the symmetric cell. It can be noted that if the temperature at point B decreases as soon as the heat transfer coefficient h is increased, the temperature at point A reacts with a small phase-difference due to the effect of the heat capacity. Figures 7 - 8 - 9 show the isothermal curves in the graphite matrix at three typical instants of the transient problem.

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Fig. 1 A right triangular prism in the (x,y,t) domain.







Fig. 4 Graphite matrix of a HTGR fuel element and symmetric cell.



Fig. 5 Finite element grid.



Fig. 6 Boundary conditions and temperature evolution at points A and B.

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Alfred Nobel

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