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NALLA-2: STRAIN MEASUREMENTS .		
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Abstract:		
	Strains were monitored for NALLA2 Structural components at critical locations : 60 locations for wing, 50 for fuselage, 6 each for horizontal tail, vertical tail, landing gear and nose gear and 20 locations for wing fuselage joint. While the wing fuselage joint was tested upto the ultimate load, the rest of the components were loaded upto the limit load only. The measured strains have proved the linear behaviour upto the limit Load, with zero yield, validating the structural integrity, of the components.	