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Abstract :			
	Random load sequence for the full scale fatigue test of Mig-2lM airframe is derived from flight data. Loads corresponding to 'g' peak/trough are derived from information on Mach number, angle of attack and weight of the aircraft. The random load sequence will be applied flight-by-flight on the airframe during the full scale test. These loads will be simulated with the help of 16 servo-hydraulic actuators. The loading actions of each of these actuators are derived. This report describes, in detail, the derivation of the random load sequence and the loading action on the multi-actuator test rig to be used in the Mig-2lM airframe. A mini-load sequence is derived by omitting lower g-ranges on the basis of their cumulative damage contribution.		

It is suggested that this mini-load sequence be used in the full scale test of Mig-21M airframe.