

Author(s)

Division

External

Sponsor

Approval

Remarks

:

participation

RESTRICTED

Document No.

Aeronautical Documentation Sheet Title

FORCE MEASUREMENTS AT LOW SPEEDS ON BODY WING TAIL COMBINATIONS (KAHA

PD EA 8916 Date of issue: DEC.1989

; Document Classification

MODEL) AT HIGH REYNOLDS NUMBERS AND INCIDENCES FROM -2° TO 70° V.Kanagarajan, K. Hartmann

EXPERIMENTAL AERODYNAMICS

Contents 146 Pages Text 11

64

Table Figs.

No. of copies: 10

NAL Project No.

NT-0-134 Sponsor's Project No.

: Head, Experimental Aerodynamics Division



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Six component measurements, Slender Body Model, Reynolds Number.

: Six component measurements were carried out on a slender body

model with lifting surfaces (KaHa Model) to study the vortex flows of such configurations at High Reynolds numbers upto very high incidences. The basic body consists of a 105 mm (3D) long tangent ogive nose and a 420 mm (12D) long circular cylinder afterbody with a diameter of 35mm (D). To this slender body cruciform lifting surfaces of 105mm (3D) and 140mm (4D) span and a constant chord length of 35mm were added to realize Body+Wing, Body+Tail and Body+Wing+Tail combinations The experiments have been carried out in the 0.6m x 0.6m pressurized low speed

wind tunnel of the DFVLR in Gottingen. The test Reynolds number, based on body diameter and a test section speed of 31 m/sec, varied from 0.7 million to 5.7 The angle of attack was varied from -2° to 70° in two ranges using a straight sting for incidences upto 30° and a 40° cranked sting at higher inci-With respect to the pitch plane the cruciform lifting surfaces were arranged in "+" and "x" position. In this report the complete test results are given in form of tables and plots with a limited analysis. At high incidences

and at higher Reynolds numnbers the model entered into severe vibrations. Thus data could not be obtained for most of the configurations around an angle of attack of 50°.