Sheet classification : RESTRICTED

Site Ae	-	nal lautical atory	Documentation Sheet	Document Classification RESTRICTED	
Title	:		Vind Tunnel Testing ible Delta Wing	Document No. PD-FC-9307	
				Date of issue: March 1993	
Author(s)	•		murthy, Basappa, nagawan & G. Shubha	Contents Pages: 42 Figures: 18 Tables: 7	
Division	:	Flight Me	chanics & Control	No. of copies: 25	
External participation	:			NAL Project No. FC-0-102	
Sponsor	:	Aeronautio	cal Development Agency (ADA	Sponsor's Project No.	
Approval	:	Head, Flight Mechanics & Control Division 55/C			
Remarks	:				
Keywords	:	Aeroservoelasticity, Flexible Delta Wing, Dynamic Wind Tunnel Testing, Time Domain Identification, Sine Sweep Tests, Acceleration Transfer Function			
Abstract		Dynamic wind tunnel testing of a flexible delta wing			
model was carried out to generate exp				experimental data	
		to be used by; the Aero-servo-elastic analysis group			
of ADA. This report det				ils model construction,	
	instrumentation, and wind tunnel testing as well as			0	
			ysis carried out to obtain	transfer functions	
		from win	d tunnel test data.		