Sheet classification : SECRET

SUCOS AGI	ional onautical oratory	Documentation Sheet	Document Classification SECRET
Title	STABILITY	OF THE LONGITUDINAL CHARACTERISTICS OF MIG-21BIS ROM FLIGHT DATA	Document No. PD SE 8622 Date of issue: 11-12-86
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Division	SYSTEMS ENG	INEERING DIVISION	No. of copies: 30
External participation	: Aircraft & ment (ASTE)	Systems Testing Establish-	NAL Project No. SE-8-223
Sponsor	:	-	Sponsor's Project No. -
Approval	: Head, Syste	ms Engineering Division	B
Remarks	:	-	n an an an Anna an Anna Anna an Anna an
<i>Keywords</i> : Flight Test, Parameter Estimation, Stability derivatives			
Abstract	: This monor	rt presents the parameter e	stimation nocults denived

This report presents the parameter estimation results derived from flight test data of a MIG-21BIS aircraft. The flight test program was executed by Aircraft & Systems Testing establishment (ASTE) and the flight data analysis was carried out at NAL. The aircraft was tested in four configurations to evaluate the effect of i) a vortex plate attached to the leading edge of the wing to improve the drag characteristics and ii) a larger saddle tank to improve the range.