University of Southern Queensland Faculty of Engineering & Surveying

Structural Performance of Cross Laminated Timber Panels as Walls

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Abstract

Cross Laminated Timber (CLT) is made from dimensional lumber, laminated in layers with the grain in each layer placed orthogonal to the layer before it. The resultant product is not dissimilar to thick plywood.

CLT is made from timber, which is an environmentally attractive option in comparison to the reinforced concrete or steel structures which dominate the multistorey building market sector. Timber is one of the very few sustainable carbon neutral building materials available.

New technology such as advanced Building Information Modeling (BIM) software and high accuracy CNC machinery are giving CLT the advantage over traditional on-site construction methods. Prefabrication is the key to CLT's improved construction time-frames, far greater quality control, than on-site construction can deliver.

There are no CLT manufacturers currently producing in Australia. Hyne and Son Pty Ltd. is a leading Australian manufacturer and distributor of engineered, structural and decorative softwood products to the building industry. This project forms part of an initial investigation into cross laminated timber wall panels, manufactured from non-structural low grade pine. Initially, the investigations will be directed solely at the low-rise multistorey residential market sector, which Hyne already services with other products.

The steps required to firstly select a suitable CLT panel using Finite Element Analysis through to fabrication of a sample CLT panel, experimental design and testing of structural properties, and finally comparing results with accepted formula from the Australian standards for the design of plywood. Considerations such as fire, and deflection under load are also made. Results from initial investigations show that, even though fabricated from low grade structural timber, CLT has enormous axial capacity. Conclusions are drawn as to the application of certain parts of AS1720 Timber Structures to cross laminated timber design.

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Nomenclature

BCA	Building Code of Australia
CLT	Cross Laminated Timber
EWP	Engineered Wood Product
FEA	Finite Element Analysis
LVL	Laminated Veneer Lumber
USDA	United States Department of Agriculture

Chapter 1

Introduction

1.1 Outline of the Study

Cross Laminated Timber (CLT) panels are fabricated from strips of timber, laminated together in layers with the grain direction of each lamella orthogonal to that before.

According to B&K (*B & K Cross Laminated Timber* 2010), existing CLT wall panels currently on the market in Europe range from 57 mm to 300 mm in thickness, and consist of between three and nine laminates. The overall size of each panel is dependent on fabrication facilities and transportable size limitations.

Typically each CLT panel is prefabricated with penetrations for doors and windows already cut, ready for installation by way of simple screwed connections on site.

Prefabrication of wall panels reduces construction time and, therefore, costs in comparison with other commonly used construction materials. Costs may be further reduced by utilising non-structural grade timber in the manufacture of the CLT panels.

Hyne and Son Pty Ltd have identified Cross Laminated Timber panels (CLT) made from low grade Caribbean and Slash Pine as a product which may have significant market potential to replace existing concrete and masonry as wall systems for 'low rise' multi-storey buildings and commercial tilt-panel structures in Australia.

1.2 Introduction

This project investigates the structural performance of CLT panels manufactured from nonstructural grade Hyne timber for use in both internal and external load bearing wall applications. The performance of the CLT panels for wall applications is measured in terms of ultimate strength and serviceability.

External and internal load bearing walls serve to transfer the load from the structure above to the structure below and to resist raking loads from earthquake and wind load actions. Serviceability of CLT panels encompasses the effects of moisture, allowable deflection and fire.

The challenge is to identify firstly an 'optimal' CLT lay-up for Hyne and Son Pty Ltd in terms of the specific said application, and then fabricate and test samples to ascertain if the selected layup performs as predicted.

Wall loading for strength includes in-plane axial compression, bracing capacity (in-plane shear) and load transfer around penetrations. Serviceability aspects investigated include deflection, fire resistance and the effects of moisture.

1.3 Aim and Scope

Broadly, the aim of this project is to analyse the structural performance of CLT panels manufactured from non-structural grade timber, for internal and external wall applications in terms of serviceability and strength.

The scope of the project is limited to the application of CLT wall panels for low-rise multistorey buildings up to three storeys high, and to commercial and industrial buildings, including tilt-slab concrete buildings. Loads applied to the CLT panel will be those applied in the design of a typical three storey residential building in Australia.

Identification of the most suitable CLT cross section for the application was required such that further investigation could proceed into fabrication methods, cost of production and possible regulatory restraints on the introduction of the product to the Australian market. As no Australian Standard existed for the fabrication or use of CLT wall panels, a comparison between the accepted existing standards for plywood in plane diaphragm loading with the results of the laboratory testing, is included for assessment of implementation using existing standards.

The method of manufacture of CLT wall panels, and fabrication of the sample CLT panels does not form part of the scope for this project, as it is mainly concerned with the structural performance of the CLT panels. Wall raking resistance is specifically excluded from the scope of this project.

1.4 Research Objectives

The specific objectives of the project are to:

- 1. Research background information relating to the structural performance of cross laminated timber for wall applications. The research will include fire resistance, deflection and ultimate strength prediction of CLT panels.
- Undertake Finite Element Analysis (FEA) to investigate the structural properties of a number of CLT cross sections which Hyne can feasibly produce, followed by a subsequent evaluation and selection of a CLT panel cross section for fabrication and strength testing.
- 3. Undertake strength testing and compare theoretical results with actual load test results, and also with accepted formulae from AS1720.1 Appendix I Buckling Strength of Plywood Diaphragms. For serviceability, the aim is to provide theoretical internal and external CLT wall load charts for various common wall heights and report on the effects of fire.

1.5 Thesis Overview

This dissertation is organized as follows:

- **Chapter 2** is an in-depth literature review, which investigates all available literature on the design and serviceability aspects of CLT wall panels. This chapter provides background on existing regulations, timber serviceability with respect to moisture, design for fire, and accepted buckling and stability theory.
- **Chapter 3** is involved with the selection of a trial CLT layup for experimental testing. Finite Element Analysis (FEA) is used to ascertain the buckling strength of a number of different CLT panel layups, and the results compared.
- **Chapter 4** describes the experimental designs, how they were selected and developed for use in this project. Experiments were developed to determine the axial strength and section properties of the CLT sample selected in the previous chapter.
- Chapter 5 Covers the results and discussion of experiments conducted at the University of Southern Queensland. This chapter also reviews AS1720.1 Appendix I - Buckling Strength of Plywood Diaphragms to see if the accepted formula for the design of plywood diaphragms can be applied to CLT. A theoretical wall load vs height chart, for various load and material eccentricities is included.

Chapter 6 Concludes the dissertation and suggests areas of further work.

Chapters 7, 8 and 9 are appendices.

Chapter 2

Literature Review

2.1 History of CLT Panels

Cross Laminated Timber (CLT) panels provide many new possibilities for the efficient and environmentally sustainable construction of houses and buildings. The product has been available overseas since it's development in the 1970's (*Building with Timber - Nine Storeys and Beyond* 2010), and is still considered an emerging technology as different construction methodologies are tested and new capabilities identified.

A recent swing away from prescriptive building codes has allowed the introduction of alternative building materials such as CLT. One example of CLT implementation in the UK is the Stadhaus building, Hoxton London. Standing at nine storeys, and completed in January 2009, it is the tallest pure timber building in the world. The completed Stadhaus building is show in figure 2.1.

Current design practice for CLT, as employed by the engineering group, Techniker, is to prefabricate as large as is transportable panels and install on site directly with simple connection details.

According to Techniker, this style of CLT construction is economically viable up to fifteen storeys without the use of local strengthening to bearing points (*Tall Timber Buildings The Stradhaus, Hoxton, London* 2010). Simplicity of construction is key, and connections consist



Figure 2.1: Stadhaus Building, London 2010 (*Tall Timber Buildings The Stradhaus, Hoxton, London* 2010)

mostly of angle cleats and long screws. This type of connection facilitates rapid installation.

2.2 Australian Regulatory Requirements

For implementation in Australia, the material must be designed such that it complies with the Building Code of Australia (BCA), which is the governing regulatory Code to which buildings must be constructed.

The BCA is a not a prescriptive code and, therefore, it is possible that, assuming the material can be shown capable of achieving the performance levels stipulated, it may be used as a structural wall element. There is no Australian Standard for the manufacture or use of CLT products.

2.3 Timber Properties and Stress Grading

There are large variations between the structural properties of clear wood and the structural properties of timber. In this paper, timber is the term given to structural elements which includes the irregularities and defects such as knots, zones of compression wood, oblique fibre orientation etc.

For the characteristics of the timber to be determined in a reliable way, multiple samples must be tested in accordance with standardised test procedures, and statistical methods used to determine structural properties (Thelandersson & Larsen 2003, p.30). The properties can then be used in the calculation of the moment, tension, compression and shear capacity of the structural element by way of elastic theory.

The theory of elasticity, of course, is only representative if the strength properties used are accurate for the load type applied. This is of particular relevance to timber due to it's anisotropic and non-homogenized nature, and explains, for example, why the characteristic bending tensile strength is used for bending, and characteristic tensile strength is used for pure axial tensile loading. The influence of the defects within the timber is included in the strength properties used.

Knots are the cause of the largest portion of failures, however other defects such as top rupture, compression wood, grain slope, decay, bark and gum pockets, wane, holes and splits or a combination of same also contribute. Determining the modulus of elasticity is the single most effective way of determining the strength characteristics of timber because it includes information regarding the clear wood properties, knots, slope of grain and other characteristics.

Additional grading information which can be incorporated with the modulus of elasticity to more accurately grade timber includes narrow and wide face knot size and locations, ring width and density among others (Thelandersson & Larsen 2003, p.30).

In timber production, non-destructive strength grading of the timber is required to define the strength properties used. Non-destructive strength grading can be either visual or by machine grading methods, with machine grading being the more accurate method, and hence offering a better yield of higher grade timber. Visual grading has limitations in terms of the thick-

ness of the timber which can be graded because faults or irregularities within the timber are undetectable (Thelandersson & Larsen 2003, p.30).

Typical machine grading usually incorporates a flat-wise bending test to determine the modulus of elasticity, and may include density measurement by radiation and/or knot detection by optical detection. The stiffness can then be accurately determined using the following formula:

$$EI = \frac{P \cdot l^3}{48 \cdot \delta}$$

Where P is the applied load, l is the span between supports and δ is the deflection (Thelandersson & Larsen 2003, p.34).

Machine bend testing of CLT wall panels is unrealistic due to the size of prefabricated wall panels. Another method of determining the modulus of elasticity is to measure the natural frequency of the timber, and to this end, a fully automated procedure for determining the global elastic properties of full-scale CLT panels has been developed by Gsell et.al (Gsell, Feltrin, Schubert, Steiger & Motavalli 2007).

2.3.1 Homogenisation and Engineered Wood Products

The term Engineered Wood Product (EWP) is a broad term that includes material typically made from smaller wood elements such as veneers and laminates. Commonly available EWP's include Laminated Veneer Lumber (LVL), plywood, glue laminated timber and I-beams.

CLT also falls into the category of EWP because it is made from laminations which have been glued together. The advantages of EWP over dimensioned timber are that they are typically dimensionally stable, and can be produced in required dimensions not limited by the individual element sizes.(Thelandersson & Larsen 2003, p.18)

The design strength of EWP is defined as the fifth percentile similar to dimensioned lumber, however due to the homogenisation of material, the fifth percentile characteristic strength value is generally higher than for timber. Figure2.2shows the probability functions for EWP and Timber (Thelandersson & Larsen 2003, p.19).



Figure 2.2: Probability Density Functions for Glulam and Solid Timber (Thelandersson & Larsen 2003, p.19)

Considering that CLT panels are made from dimensioned timber, it is not unreasonable to assume that there is a relationship between the strength of the individual elements and the effective strength of the CLT panel.

Two options available for predicting the strength properties of CLT panels of various thicknesses and lamella are firstly to produce sufficient samples such that the properties can be determined statistically, and secondly, to use a method which uses the strength properties of the individual elements to determine the strength characteristics.



Figure 2.3: Glulam Beam

The second method is more difficult to accurately achieve for CLT panels than, for example, glue laminated (Glulam) beams indicated in figure 2.3, because of the composite nature of the orthogonally placed lamella.

For Glulam beams, the typical failure mode for bending is tensile failure of the outer laminate. Laminating increases the bending strength above the pure tensile strength of the laminates. The ratio of tensile bending strength to tensile strength is called the laminating factor (Thelandersson & Larsen 2003, p.69).

The Glulam strength increase is attributed to homogenisation of defects, and by reinforcement of defective areas by adjacent laminates. Some CLT panels are constructed in only three layers, and this presents a different case than for Glulam, which typically comprises many more layers.

To substantially increase the bending tensile strength by laminating would require outside lamella thickness to be reduced to less than 10mm (Thelandersson & Larsen 2003, p.71). CLT wall elements effectively load share along the length of the panel when loaded in plane and in bending, however unless more than three laminates are used, it is unlikely that the effective strength will be higher than that of the timber elements from which it is made.

In Australia, Glue Laminated Timber is fabricated to AS/NZS 1328, and establishment of characteristic properties of glulam is performed using AS/NZS 4063.

2.3.2 Effects of Moisture

The dimension stability of the timber is relative to the moisture shrinkage. Shrinkage strains are much larger perpendicular to the grain than parallel to the grain, and also much larger tangentially than radially. Characteristic shrinkage and distortion of timber elements in relation to their location in the parent log from which they were sawn are indicated in figure 2.4

Drying of the timber to a moisture content less than 15 % results in the timber being classed as seasoned timber. The humidity of air inside a building is similar to the moisture content of seasoned timber and therefore equilibrium results in good dimensional stability for interior and dry applications. The dimensional stability is greatly enhanced for CLT panels due to the orthogonal lay-up of lamella.



Figure 2.4: Shrinkage and Distortion Relative to Parent Log (U.S. Department of Agriculture, Forest Service, Forest Products Laboratory 2010, p.4-5)

The CLT samples provided by Hyne and Son Pty Ltd were backsawn, and therefore are subject to cupping due to the moisture gradient between the air and the timber, unless sufficiently sealed against moisture ingress. It is likely that moisture cracks will occur parallel with the grain on the exterior lamella of the CLT wall panels, and therefore the shear capacity should not be relied upon parallel to the grain.

2.3.3 Duration of Load (DOL) Effects

The strength of timber also depends on the duration and rate of loading. AS1720.1 1997 defines duration load factors for five seconds, five minutes, five weeks, five months and fifty years, with load factors ranging from 1.0 for five second wind gusts to 0.57 for permanent loads. The long term load factor is a result of the significant loss of strength timber undergoes when subject to load for long periods of time.

2.4 Design for Fire

The use of timber for multistorey residential buildings has not been fully embraced since the great fire of London in 1666, where 373 acres of London was burned to the ground, and 100,000

people were left homeless in a dreadful and uncontrollable fire (Robinson 2010). Prior to that time, most of the buildings in Europe were of timber construction.

After the fire, new legislation was enacted (The Rebuilding Act 1667), which prescribed that the outsides of all buildings should be made of bricks or stone (Museum of London 2010)



Figure 2.5: Timber Building of The Era (Robinson 2010)

It should be noted that there were a number of contributing factors to the disaster, apart from the use of timber in the building structure, which the Rebuilding Act 1667 recognises. Houses prior to the great fire encouraged the spread of fire by close placement of dwellings, having disorderly design with no partitions between dwellings, overhangs out over the small laneways and overhangs over other properties. A typical Low-rise multistorey dwelling which was claimed by the great fire is shown in figure 2.5.

The main points of the Act called for wider roads and lane-ways, fire separating brick walls around properties with minimum separation distances specified, no overhangs or balconies outside the boundary of the property unless protruding over a street, and construction of the boundary walls to be made from stone or masonry.

The Rebuilding Act 1667 is a prescriptive code, however if it were a non-prescriptive code then it could in part be interpreted that if a major fire could be contained to an individual dwelling,

then it has met the requirements of the Act. The Building Code of Australia (BCA) is a nonprescriptive code, which shares a remarkable resemblance to the Rebuilding Act 1667, in that it recognises classes of buildings and contains similar requirements which limit the spread of fire.

The BCA requires that compliance of the Fire Resistance Level (FRL) be provided in either a report from a Registered Testing Authority, stating that the material complies with the standard fire test, or alternatively is designed in accordance with AS1720.4 if it is a solid or glued-laminated timber structure (Australian Building Codes Board 2010, p.25).

Much published research is already available on the effects of fire to exposed CLT. In nature, large trees and logs can resist the effects of fire by charring. Pyrolysis is a chemical decomposition reaction timber undergoes when exposed to temperatures around 150° deg C to 200° deg C. The gases produced by the pyrolysis reaction are combustible. The char-layers which form once the timber is ignited have excellent insulating properties which protect the timber.

Designing for fire typically involves adding a redundant thickness to the CLT wall panel. The additional thickness of material required is dependant on the required FRL.

A charring rate is the rate at which char is formed and is proportional to the temperature. According to AS1720.4 (Standards Australia 2010*c*, p.8), the notional charring rate for untreated timber exposed to fire is:

$$c = 0.4 + \frac{280^2}{\delta}$$
(2.1)

and

$$d_{\rm c} = ct + 7.5 \tag{2.2}$$

where:

c = notional charring rate in millimeters per minute

 δ = timber density at 12% moisture content

 $d_{\rm c}$ = effective depth of charring

t = time period in minutes

It should be noted that AS1720.4 only applies to Glued Laminated Timber fabricated with phenol, resorcinol, phenol resorcinol or poly-phenolic glues.

Results from a recent study by Wilinder (Wilinder 2009, p.38) indicate that the charring rate of CLT is similar to that of solid wood as calculated by EN1995 1-2 for a solid beam burning on one side only.

It follows that for CLT constructed of Pine, with a density of 600 kg/m^3 , the charring rate *c* is approximately 0.61 millimeters per minute. Further, for a nominal building constructed of CLT requiring a 30 min fire resistance level, a 26 mm additional thickness is required.

2.5 Wall Panel Axial In-Plane Loading

Timber framed wall elements of buildings are typically between 2.7m and 3.0m tall for single and multistorey construction alike, and so depending on the CLT geometry, the panel may be considered slender in some circumstances.

For low-rise multistorey construction, the floor is typically placed directly over the timber walls beneath (platform construction), to ensure in-plane loading is achieved(*Tall Timber Buildings The Stradhaus, Hoxton, London* 2010). The top and bottom restraint conditions of the panel can be modeled as being restrained at each end in position only (i.e pin end top and bottom).

According to R.H. Leicester of the CSIRO, Victoria (Leicester 2009), two parameters are used in the calculation of the buckling strength of dimensioned timber structures when designing to Australian Standard - Timber Structures Part 1: Design methods (AS1720.1).

The parameters are the stability parameter, and the stiffness parameter. Steiger and Gulzow (Steiger & Gulzow n.d.) note that verification of CLT panel properties must be undertaken to account for the effects of homogenisation of the low grade timber with laminating, when considering elastic and strength properties of the panel. One of the objectives of this project is to identify if the formula in Appendix I of AS1720.1 is applicable to CLT wall panels. Formulae are provided for diaphragms with lateral edges free and subjected to uniformly loaded edge forces. This is appropriate as the ends of the CLT wall sections will not necessarily have lateral support, as is the case for a wall section between two wall openings.

Ing, Blass and Fellmoser (Blass, Fellmoser & Ingenieurholzbau 2004) identify the composite theory as the current design method for calculating plywood properties. They describe the composite theory as only taking the modulus of elasticity of elements of the plies acting parallel to the load into consideration. This leads to inaccuracies because plies perpendicular to the load contribute to the bending rigidity of CLT panels.

Contrary to Ing, Blass and Fellmoser, AS1720.1 (Standards Australia 2010*b*, Appendix I) takes into consideration a small (3%) contribution for the plies acting perpendicular to the load, when determining the bending rigidity.

Unfortunately, however the equation for calculating the buckling strength of plywood is independent of the bending rigidity of the section. The slenderness coefficient is based on the ply lay-up and the stability parameter used is similar to that for solid timber.

Further, Ing, Blass and Fellmoser (Blass et al. 2004) note that when taking the plies in the X-axis direction into consideration, rolling shear of the laminate will detract from the panel stiffness, i.e. the thicker the laminate running in the X-direction, the higher will be the shear deformation.

To fabricate Hyne CLT panels with more than one laminate running in the X-direction would be expensive due to the additional fabrication complexity. One option to overcome rolling shear deformation of the central laminate, for a three ply CLT wall panel, would be to reduce the thickness of the central laminate.

2.6 Wall Panel Raking Resistance (In-Plane Shear Capacity)

Besides axial loading, walls are also used to transfer shear forces from wind loads to the ground. Vessby et al. (Vessby, Enquist, Petersson & Alsmarker 2009) recommend CLT panels as an alternative to timber framing for narrow tall houses. This recommendation is because narrow tall houses require a lot of bracing capacity, and often there is insufficient wall length for the placement of bracing plywood. CLT panels are similar to plywood panels in that they have lamella with grain direction in both axis, which allows them to function as a wall panel in compression as well as a bracing wall in shear.

2.7 Accepted Theory - Stability and Buckling

The theoretical buckling load capacity can be determined linearly using Eulers formula. Three defined states of loading for a system are when the system is stable, in neutral equilibrium and unstable. An axially loaded system is considered in neutral equilibrium when the applied load can be removed, and the system returns to it's original position. Euler's formula for the critical buckling load is...

$$P_{cr} = \frac{\pi^2 EI}{L^2}$$

where...

 P_{cr} = critical buckling load

E =modulus of elasticity

I = second moment of area

L = height of column or wall

When the restoring force of a column or wall is less than the applied load F as indicated in figure 2.6, the system is termed unstable and will collapse.

The load at which the system changes from stable to unstable is called the critical load. When a system is stable, the stress in the column or wall can be determined directly using the following



Figure 2.6: Stable and Unstable Systems

equation...

$$\sigma = \frac{P}{A}$$

where σ is the stress in the column or wall cross section, P is the applied load and A is the cross sectional area.

CLT load bearing panels can be described as slender columns with pin ends. Lateral support is only provided to the CLT wall panels at locations where a wall intersects. The trend in recent years is for large open span type living areas, and lateral support, therefore, cannot be assumed.

The important thing to note about the critical buckling load is that the material characteristic which alters the critical load is the stiffness, and not the strength of the material. Thus the modulus of elasticity and second moment of inertia are material variables which will determine the critical buckling load. Obviously the height of the wall is the most critical factor.

Localised buckling within the CLT material is not likely because each laminate is laterally restrained by adjacent laminates. As this project specifically deals with wall panels as opposed to columns failure, further calculations assume failure about the weak axis.

It must be assumed that the CLT wall is going to be subjected to eccentric axial loads, due to material eccentricities within the wall because timber is an inhomogeneous material.

Deflection of an eccentrically loaded wall increases proportionally to the applied load. Bending stresses or deflection limits for serviceability may define the allowable load, as opposed to the critical load. According to Gere and Timoshenko (Timoshenko 1991, p.596), the following formula is used to calculate the maximum compressive stress on the inside face of a column...

$$\sigma_{max} = \frac{P}{A} + \frac{M_{max}}{S} \tag{2.3}$$

Where

 σ_{max} = maximum stress on the compressive face

- p = axial compressive load
- M = moment induced by the load eccentricity

e = load eccentricity

S =section modulus.

The moment induced by the axial compressive load on an eccentric column is related to the induced curvature of the beam-column. The mid-height deflection of a beam-column of certain eccentricity can be calculated using the following formula...

$$\delta = e\left(sec\frac{KL}{2} - 1\right) \tag{2.4}$$

Combining equations (2.3) and (2.4), the following equation is formed, which is termed the secant equation and can be used to calculate the maximum compressive stress on the compressive face of a beam column...

$$\sigma_{max} = \frac{P}{A} \left[1 + \frac{ec}{r^2} sec\left(\frac{L}{2r}\sqrt{\frac{P}{EA}}\right) \right]$$

Where r is the radius of gyration, and c is the distance from the centroid to the compressive face. It should be noted that the above secant equation only applies to small deflections and

when Hooke's Law holds.

Although Gere and Timoshenko specifically investigate compressive failure of the outermost fibre of the inner face, and derive the secant formula on this basis, it should be noted that other flexure failures need also be considered.

Australian Standards for design of Compressive Members - Design for strength AS1720.1 presents a formula similar to that presented by Timoshenko. The following formula is the design compressive strength capacity parallel to the grain...

$$N_{d,c} = \sigma k_1 k_4 k_6 k_{12} f'_c A_c$$

Where k_1 is the Duration of Load (DOL) factor, k_4 is a factor relating to moisture content and k_6 relates to temperature. The remaining factor k_{12} is the stability factor.



Figure 2.7: Stability Factor VS Slenderness Factor (Standards Australia 2010b, p.32)

The k_{12} stability factor is calculated on the basis of the three ranges similar to that described by Gere and Timoshenko (Timoshenko 1991, p.603), and as indicated in Plot 2.7. The slenderness factor ρS indicated in the figure is a factored slenderness ratio, and takes into account column geometry and restraint.

Having a k_{12} value which falls within the first area of plot 2.7, indicates column fails will be due to pure axial compression as opposed to buckling. In this range the compressive strength capacity is not dependent on the material stiffness, or the slenderness ratio.

In the second area of plot 2.7, the strength of the outer compressive fibre on the inner face is dominant because the slenderness ratio is not high enough for the column to fail by buckling within the proportional limit.

Finally for the third area of plot 2.7, the formula used in calculation of k_{12} is similar to the euler buckling formula, and buckling of the column is due to high slenderness ratio of the column and the stiffness of the material. According to Gere and Timoshenko (Timoshenko 1991, p.603), failure which occurs in the third range is independent of the strength of the material.

Chapter 3

Finite Element Analysis

3.1 Introduction

This chapter outlines the technical details and processes of Finite Element Analysis (FEA) to determine which CLT panel layup best suits the requirements of the low-rise multistorey residential building sector, and which can be economically produced by Hyne. A CLT panel layup refers to the number, thickness and direction of each lamella within the CLT panel.

Instead of fabricating and testing many different CLT panel layup samples, an FEA was conducted to predict a single suitable CLT Panel layup. The layup was selected based on a number of considerations including strength, available dimensional timber sizes which could be economically used in the CLT fabrication, and available test equipment capacities.

A nominal load case was developed for a low-rise multistorey residential building. The nominal building wall height is also identified. Further considerations such as available dimensional timber sizes are developed. This forms a criteria for selecting the CLT panel Layup for experimental analysis.

A total of six different CLT layups were modeled and analysed using the Strand7 Finite Element Analysis (FEA) package. Two additional models were also developed to test the sensitivity and boundary condition assumptions. Each wall panel modeled was 3m tall and 1m wide so that a load per unit meter could be directly calculated. The actual CLT panels for experimental
testing were limited to a width of 300 mm due to the machine axial load capacity.

Density and acoustic modulus of elasticity of the timber were provided by Hyne prior to fabrication of the sample panels, and the remainder were taken from published data. Orientation of the orthotropic elastic properties for each lamella was undertaken graphically.

Actual restraint conditions at the top and bottom of a building vary between construction methods. Restraint and loading conditions applied to the modeled wall sections are modeled as being a pin connection at the top and bottom of the wall. The walls were modeled using quad eight brick elements, and analysed using the linear buckling solver.

In summary, the following steps were required to select a suitable layup for experimental testing...

- 1. Determine the material properties of the dimensional timber used to fabricate the CLT panels
- 2. Define a nominal load case applicable to the low rise multistorey residential building sector
- 3. Identify a number of CLT layups which Hyne could economically produce
- 4. Undertake Finite Element Analysis (FEA) for all six of the identified CLT layups
- 5. Rationalise the results and select a suitable layup to fabricate samples

3.2 Properties of Dimensional Timber

The CLT sample panels were fabricated from a mixture of non-structural grade Slash Pine (Pinus Elliottii) and Caribbean Pine (Pinus Caribaea) dimensional timber. Mechanical properties, including orthotropic elastic properties of the timber elements were required to fully describe the behavior of the CLT trial panels by FEA modeling.

In Australia, published timber properties are usually those properties used to design structures in accordance with Australian Standard AS1720 - Timber structures, and other associated Australian Standards. As there is such a large variation in mechanical properties of different tree species, and also within species due to discontinuities such as knots and sloping grain etc., a grading system was developed.

For Slash and Caribbean pine, grading systems could be either by Machine Grade Pine (MGP) stress grading or grading under the general F grade system. The MGP grading system for structural pine is a result of the 'in-grade testing program of Australian Pine (radiata pine, pinaster pine, slash pine and Caribbean pine), undertaken by CSIRO and State Forests of NSW' in 1996 (*Education Resources Structural Grading* n.d.). The program aimed to achieve a better yield of pine timber with a higher stress grade by implementing more rigorous machine testing techniques.

The properties refered to in the Australian Standards for timber design are fifth percentile characteristic mechanical properties as opposed to mean mechanical properties, which may be significantly higher. If fifth percentile characteristic strengths were used, then the resulting FEA model would not be comparable with the laboratory test results.

Hyne provided mean density and acoustic modulus of elasticity properties for the dimensional timber as measured prior to fabrication refer to Appendix 8. Where published mean mechanical properties for Australian grown Slash and Caribbean Pine could not be found, properties for American Slash and Caribbean Pine were substituted where required, on the basis that the modulus of elasticity and density of the American variety was similar to those measured by Hyne.

Because the test data received from Hyne was not specific as to the species of pine, the mechanical properties of Slash Pine were used where no other data existed. According to the United States Department of Agrictulture (USDA), 'the mechanical properties of Slash Pine and Caribbean pine are rather similar' (U.S. Department of Agriculture, Forest Service, Forest Products Laboratory 2010, p.2-39).

The mechanical properties for Slash Pine and Caribbean Pine from the USDA Forest Products Laboratory Wood Handbook, Hyne test data, and the MGP15 strength group are summarised on Table 3.1. The properties which were adopted for this project are also identified.

In addition to the timber properties listed in Table 3.1, FEA modeling requires a full description of the elastic properties of the timber to populate the compliance matrix. Timber is orthotropic

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Table

	Pine Timber	USDA Forest	Hyne Test Data	MGP 15	Adopted
ies	Tvpe	Service	(Mean Values)	Characteristic	Properties
	- J /-			Strengths	· · · · · ·
icity	Caribbean	15.4	t T	0.21	ŗ
	Slash	13.7	17	7.01	17
ture	Caribbean	115	NIA	VIIV	ç
	Slash	112	N/N	INA	112
3)	Caribbean	680	U27	200	027
<u>_</u>	Slash	590	000	noc	000
allel	Caribbean	58.9	VII	30	1 7 2
	Slash	56.1	N/A	CC	1.00
	Caribbean	Not Listed	NIA	VII	ſ
1pa)	Slash	7	A/M	INA	-
	Caribbean	14.4	NIA	-	11 C
	Slash	11.6	N/A	9.1	0.11
Stress	Caribbean	Not Listed	Lo	11	L0
	Slash	Not Listed	10	41	10

because it has different elastic properties in different orthogonal directions.

Twelve constants describe the elastic properties of an orthotropic material, including three elastic moduli, E, three moduli of rigidity G, and six Poisson's ratios v (U.S. Department of Agriculture, Forest Service, Forest Products Laboratory 2010, p.5-2). Three of the Poisson ratios are dependent, and related via the following equation...

$$\frac{v_{ij}}{E_i} = \frac{v_{ji}}{E_j}, \quad i \neq j \quad i, j = L, R, T$$
(3.1)

Orthotropic elastic properties of timber are approximately related to the longitudinal modulus of elasticity, and were therefore published as ratios to the longitudinal modulus of elasticity (U.S. Department of Agriculture, Forest Service, Forest Products Laboratory 2010, p.5-2).

The calculated orthotropic elastic properties for Slash Pine are provided in Table 3.2. No orthotropic data could be found for Caribbean Pine, however as previously stated, the material properties of Slash and Caribbean Pine are similar, and Slash Pine properties will therefore be substituted.

According to the Strand7 online manual (Strand7 Pty Ltd 2010), the compliance matrix for an orthotropic material is defined as ...

$$C = \begin{bmatrix} \frac{1}{E_1} & \frac{-v_{21}}{E_2} & \frac{-v_{31}}{E_3} & 0 & 0 & 0 \\ \frac{-v_{12}}{E_1} & \frac{1}{E_2} & \frac{-v_{32}}{E_3} & 0 & 0 & 0 \\ \frac{-v_{13}}{E_1} & \frac{-v_{23}}{E_2} & \frac{1}{E_3} & 0 & 0 & 0 \\ \frac{-v_{13}}{E_1} & \frac{-v_{23}}{E_2} & \frac{1}{E_3} & 0 & 0 & 0 \\ 0 & 0 & 0 & \frac{1}{G_{12}} & 0 & 0 \\ 0 & 0 & 0 & 0 & \frac{1}{G_{23}} & 0 \\ 0 & 0 & 0 & 0 & 0 & \frac{1}{G_{31}} \end{bmatrix}$$

Where E is the modulus of elasticity, G is the modulus of rigidity or shear modulus, and v_{ij} are Poisson's Ratios. The first letter in the Poisson Ratio subscript refers to the direction of applied stress, and the second refers to the resulting direction of deformation.

USDA Material	Material Property		
Property Notation	Value		
Modulus of Elasticity (Mpa)			
$E_{\rm L}$	17000		
E_{T}	1258		
E _R	935		
Shear Mod	ulus (Mpa)		
G _{LT}	901		
G _{TR}	170		
G_{RL}	935		
Poisson's Ratio			
v _{LT}	0.444		
v _{TR}	0.387		
$v_{\rm RL}$	0.02		

 Table 3.2: Orthotropic Elastic Properties of Slash Pine

A discrepancy between the relationship of the six Poisson's Ratios published in the Strand7 Online Manual (Strand7 Pty Ltd 2010) and Equation 3.1 was identified. The Strand7 Online Manual identifies one of the Poisson's Ratio relationships as being...

$$v_{23}E_3 = v_{23}E_2 \tag{3.2}$$

however, according to the USDA formula 3.1, the equation should read \cdots

$$v_{23}E_3 = v_{32}E_2 \tag{3.3}$$

The USDA published formula is adopted for this thesis, and according to the Strand7 Manual, v_{23} is interchangeable with v_{32} (Strand7 Pty Ltd 2010), so in any event the FEA model should still be accurate regardless.

Roof Dead Loads	Mass (kN/m ²)
Terracotta tiles (french pattern)	0.12
Timber trussed metal roof	0.12
Gypsum plaster (13 mm)	0.13
Total	0.37

Table 3.3: Roof Dead Loads

3.3 Calculation of Nominal Wall Loads

Three storey low-rise apartments constructed in brick, rendered masonry and framed timber are popular in Australia. Hyne already services the low-rise residential market with other timber products and so, in the short term, this market presents a logical entry point for CLT wall products.

In the late 1970's, a standardised three storey brick block of flats emerged which was very economical to build and popular with investors. Three storey residential units are economical for developers because in general, they do not require the additional expense of services such as pressure boost water pumps and elevators, which are required by medium rise buildings.

In more recent times, architectural flare has seen a trend toward more complex and aesthetically pleasing buildings. As architectural flare is difficult to predict, a standardised rectangular building of similar proportions to the three storey brick block of flats will be used for the purposes of this project.

A cross section of the nominal unit block is indicated in figure 3.1. The CLT wall under consideration will be the external wall on the ground floor. This wall supports floor load widths and half of the roof span.

Standard construction systems producing the highest axial in-plane wall loading case are assumed. The floor is therefore a 300 mm thick tiled timber floor system, consisting of Hyne timber I-Beams, 15 mm thick cement fibre board and tiles over, which is the industry standard for bathrooms on timber framed floors.

The floors on all levels are assumed to be bathroom floors, which are often located similarly



Figure 3.1: Nominal Residential Unit Building

Floor Dead Loads	Mass (kN/m ²)
Tiles	0.27
Fibre cement sheet (15 mm)	0.23
Hyne I-beam joists at 450 mm crs.	0.18
Total	0.68

3.3 Calculation of Nominal Wall Loads

on each level. The roof system is a timber trussed roof spanning between external walls, with terracotta tiles. Roof and floor dead loads used in the following calculation were taken from AS/NZS 1170.1 Structural Design Actions Part 1: permanent, imposed and other actions (Standards Australia 2002). Dead loads are summarised in tables 3.3 and 3.4.

As the load case considered is for a wall, a universally distributed load (UDL) is appropriate. The following equations 3.4 to 3.6 yield the required axial in plane ultimate load.

Dead Load...

$$G = (M_{\text{Roof}} \times RLW) + (M_{\text{Floor}} \times FLW) + (M_{\text{Wall}} \times H_{\text{Wall}})$$

= 0.37 × 7.0 + 0.68 × 6 + 0.63 × 5.7
= 10.26 kN/m (3.4)

Where G is the dead load, RLW is the roof load width, FLW is the floor load width, H_{Wall} is the wall height and M is the mass of the associated element. The mass of the wall assumes a 100 mm thick CLT wall section.

Live Load...

$$Q = (Q_{\text{Roof}} \times RLW) + (Q_{\text{Floor}} \times FLW)$$
$$= 0.25 \times 7 + 1.50 \times 6.0$$
$$= 10.75 \text{ kN/m}$$
(3.5)

Where Q is the uniformly distributed live load on the wall. Applied live loads $Q_{\text{Roof}} = 0.25$ kPa and $Q_{\text{Floor}} = 1.5$ kPa were taken from Table 3.2 of AS/NZS 1170.1 Structural Design Actions Part 1 (Standards Australia 2003, pg.506).

Ultimate Axial Load...

$$N^* = 1.2G + 1.5Q$$

= 1.2 × 10.26 + 1.5 × 10.75
= 28.44 kN/m (3.6)

where N^* is the ultimate axial compressive wall load.

According to AS1720.1-1997 (Standards Australia 2003, pg307), the required design compressive capacity of timber parallel to the grain (N_c) can be calculated from the following equation...

$$N_{c} = \frac{N^{*}}{\phi}$$

= $\frac{28.44}{0.65}$
= 43.7 kN/m (3.7)

Where ϕ is the capacity factor for compressively loaded timber for the nominal building.

There is no suggested serviceability limit in AS/NZS 1170.2 for timber walls, however as the walls would likely be lined with plaster or gypsum board, Table C1 specifies an appropriate element response of wall height/300.

3.4 Trial CLT Layups

Apart from strength and serviceability, a number of considerations were taken into account when deciding the most suitable CLT layup including:

- Available timber sizes produced at the Hyne timber mill
- Complexity of CLT panel manufacture
- Capacity of the University of Southern Queensland axial load test rig

As the timber mill is optimised for production of timber with specific dimensions, it was decided on the basis of economics that only currently produced dimensional timber sizes would be considered for CLT manufacture. The preferred non-structural grade sizes provided by Hyne were 25x90 mm boards and 35x90 mm and 45x90mm utility grade studs. As resurfacing

3.4 Trial CLT Layups

is required prior to bonding to ensure an even surface, an allowance of between two and four millimeters was provided for all bonded surfaces.

The preferred layup is three layers, because this reduces manufacturing costs. The advantages of additional layers is increased dimensional stability and homogenisation. To what extent the disadvantages outweigh the advantages or vice-versa is difficult to determine due to the complexity of modeling timber discontinuities using FEA.

Due to the technical difficulties of gluing and applying bonding pressure for curing, Hyne specified that short edges would not be bonded. Applying the required 1 MPa of pressure to the short edge of the boards, as well as the long edge, requires a specialist CLT press, not available at the time of manufacture.



Figure 3.2: Trial CLT Cross Sections

Testing was carried out at the University of Southern Queensland Center of Excellence in Engineered Fibre Composites. The available test rig was limited to a wall height of 2.7 m and a maximum axial load of 200 tonnes. This is much less than the required building nominal load.

Six trial CLT layups were identified for FEA modeling, and a cross section of each is indicated in figure 3.2.

3.5 The Finite Element Model

Strand7 is a general purpose finite element analysis package developed by G+D Computing, and offered by Strand7 Pty Ltd. The software has the capacity to analyse beam, plate and brick elements using a variety of solvers. The two solvers used for analysis in this project are linear static solver and the Linear buckling analysis solver.

The following steps were undertaken for each trial layup...

- 1. Create the geometric model
- 2. Apply material properties
- 3. Apply restraint conditions
- 4. Apply loads
- 5. Solve for linear static and linear buckling analysis
- 6. Check model quality

3.5.1 Model Geometry and Material Properties

Each geometric model was constructed of quad eight brick elements. If smaller than necessary element sizes are used, the model becomes inefficient to solve, and if the elements are too coarse, then the results may be inaccurate.

For simplicity, all quad-eight block elements were modeled with overall dimensions 100 mm \times 100 mm \times the thickness of the respective lamella. A quality check was undertaken after each analysis to ensure that the brick elements were sufficiently sized to provide accurate results. More details of the quality check are provided later in this section.

The model was created by firstly using the grid command to create a quad eight plate element with the same overall dimensions as the wall panel. The plate was then extruded into brick elements. As there are three lamella, the brick element material properties were incremented such that the outside lamella had the same material property and the central lamella had a different material property.

A single material property could have been assigned to brick elements on a local axis basis, however this would have resulted in a complex FEA model, and differentiation between horizontal and vertical lamella in the Strand7 graphical environment would have been difficult.

For simplicity, two material properties were used, one for each lamella grain direction. The first material property (Material Property 1) was applied to lamella with a vertical (global Y-Axis) grain direction, and the second material property (Material Property 2) was applied to lamella with a horizontal (global X-Axis) grain direction. In Strand7, if no local axis are assigned, material property directions 1,2 and 3 correspond to global directions X,Y and Z respectively.



Figure 3.3: Transformed Material Properties for Vertical CLT Lamella

Instead of using a transformation tensor to re-orient the elastic material properties provided in section 3.2, the transformation of axis for material properties was done graphically as shown in figures 3.3 and 3.4.

The resulting Strand7 block element orthotropic material inputs are shown in Figures 3.5 and 3.6, and summarised in Table 3.5.



Figure 3.4: Transformed Material Properties for Horizontal CLT Lamella



Figure 3.5: Strand7 Orthotropic Block Element Material 1

USDA Material	Strand7 Material	Strand7 Material	Material	
Property Notation	Property 1	Property 2	Property Value	
	Modulus of Ela	sticity (Mpa)		
EL	E_2	E_1	17000	
E _T	E_1	E_2	1258	
E _R	E_3	E_3	935	
Shear Moduli (Mpa)				
$G_{ m LT}$	G_{21}	G_{12}	901	
G_{TR}	G_{13}	G_{23}	170	
G_{RL}	G_{32}	G_{31}	935	
Poissons Ratio				
$v_{ m LT}$	v_{21}	v_{12}	0.444	
v_{TR}	v_{13}	v ₂₃	0.387	
v _{RL}	v_{32}	v_{31}	0.020	
v _{TL}	v_{12}	v_{21}	0.033	
v _{RT}	v_{31}	v_{32}	0.288	
v _{LR}	v ₂₃	v ₁₃	0.364	

Table 3.5: FEA Material Properties Summary



Figure 3.6: Strand7 Orthotropic Block Element Material 2

3.5.2 Forces and Restraints

Understanding the restraint conditions to apply to the wall requires knowledge of the construction method to be used. In general, low-rise multistorey CLT buildings are constructed using the platform method. By definition, walls of buildings constructed using the platform method are discontinuous between levels. Manufacturing, transporting and handling walls taller than three meters is not practical.

The flooring system adopted for the nominal building structure is the Hyne I-Beam floor joist product. I-beams are engineered beams consisting of a top and bottom chord connected with a shear web. Given the depth of the nominated floor joists (300 mm), it is not realistic that the floor joists bear directly on top of the walls. In this situation, the connection detail should have the floor joists supported on joist hangers to the side of the wall as shown in Figure 3.7.

The linear analysis solver in Strand7 does not account for load eccentricities, so modeling a load offset to the edge of the wall, even though it is a better representation of the physical connection detail, would not alter the calculated critical buckling load.

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Figure 3.7: I-Joist to CLT Wall Panel Connection(Pryda Australia 2009)

The sides of the walls were modeled as being unrestrained, and no global restraints were applied. Obviously for a continuous wall, there would be some interaction with the continuing wall, and a check of the model quality is discussed in subsequent sections.



Figure 3.8: Wall Top Edge Local Deflection

Initially as the wall was modeled as block elements, the obvious choice was to apply a global vertical pressure. On closer inspection of the deformations to the top and bottom edges of the wall, the applied global pressure resulted in a pronounced vertical displacement of the central lamella, in relation to the exterior lamella as indicated in Figure 3.8. This was because the

3.5 The Finite Element Model

modulus of elasticity of timber parallel to the grain is much lower than that perpendicular to the grain.

Applying a global pressure to the top of the wall is inaccurate for two reasons, firstly because the central lamella cannot displace further than the exterior lamella due to the joist attachment geometry, and secondly, the central horizontal lamella was not pressed along the short edge during fabrication and therefore, may not effectively transfer the applied load between horizontal dimensional timber elements.



Figure 3.9: Strand7 Rigid Link Example (Strand7 Pty Ltd 2010)

A better method of applying the load to the top of the wall was to rigidly link all of the nodes to a master node in the center of the wall at the top. This is akin to placing an infinitely rigid top and bottom plate to the wall.The concept of applying rigid links is best demonstrated in figure 3.9. Applying rigid links to the top and bottom edges also simplifies the model because node restraints other than the two master nodes were not required.

The rigid links, master node, and applied loads for the top of Trial 1 CLT layup are shown in figure 3.10. The bottom of the wall is similar except that instead of a load being applied to the node, the node is restrained in the global Y-Axis. A summary of node restraints is provided in Table 3.6, where Node 1 refers to the Node located at the bottom of the wall, and Node 2 refers to the node at the top of the wall.

In order to undertake a nonlinear buckling analysis, the linear solver first had to be run to



Figure 3.10: Trial 1 Rigid Links, Node Restraints and Unit Load

calculate the element geometric stiffness (Strand7 Pty Ltd 2010). The linear static analysis was undertaken with an applied load of -1 kN in the Y axis direction. This load was applied for the linear static analysis because the output from the linear buckling solver is a single buckling load factor for each buckling mode, which can then be multiplied by the applied load to determine the critical bucking load.

3.6 Model Sensitivity and Boundary Conditions Check

A simple check to identify if the element sizes are sufficiently small, is to subdivide all of the brick elements and re-run analysis. When subdivided, each brick element is divided into four new brick elements.

If the difference between the buckling capacities is insignificant, say less than 1% for example, then element size is assumed to be suitable. If the resulting difference in buckling capacities is large, then the wall needs to be broken down into additional elements.

The model quality check was conducted on only one trial CLT layup. This can be justified firstly because all of the CLT layups have similar overall dimensions and slenderness ratios, and secondly, the wall shape is not complex and does not include sharp corners or discontinuities,

3.7 Results of Finite Element Analysis

Clobal Avia	Restraint		
Giobal Axis	Lateral	Rotational	
Node 1			
X	F	R	
Y	F	F	
Z	F	F	
Node 2			
X	F	R	
Y	R	F	
Z	F	F	

Table 3.6: Node Restraint Conditions (F=Fixed R=Released)

which require a higher resolution of elements.

One further quality check was undertaken regarding boundary conditions at the sides of the 1m wall section. The Strand7 model of the wall is 1m long, however in reality, the wall could be as long as the building.

To check the assumption that the linear buckling load of the modeled 1m wall is accurate to be used as a unit length buckling load, and that the boundary conditions on the side of the 1m wall have little effect, the wall was modeled as being two meters long, and the buckling load results compared.

Load comparison is only for the first buckling mode because modes above the first mode are not physically possible for the vertically discontinuous CLT wall section modeled with pin ends. The boundary condition check was only undertaken for the Trial 1 CLT layup.

3.7 Results of Finite Element Analysis

This section discusses the results of the linear buckling analysis for each of the CLT trial sections, as well as the sensitivity analysis and the boundary condition check.

3.7.1 Trial CLT Finite Element Analysis Results

The linear buckling solver calculated four modes of buckling failure. Though the graphical output indicates displacement and strain, only the critical buckling loads and associated modes of failure are obtained from the linear static buckling analysis.

Deflection and stress data is not attainable unless a non-linear analysis is conducted. A nonlinear analysis involves applying loads at load intervals until convergence cannot be obtained. At this point the wall would be said to have buckled.

All of the CLT trial samples produced similar buckling mode shapes. The buckling mode shapes provide an insight into possible failure modes which would need to be considered if the walls were continuous over two or more levels. They also provide insight into the effects of the boundary conditions. The four modes of failure for Trial Layup 2 are shown in figure 4.10.

As an example of how the boundary conditions effect the buckling mode, if the wall was modeled as being longer, or the nodes on the sides of the wall were restrained from lateral movement in the Z-direction, the buckling mode shape of figure 3.13b, and the associated critical buckling load would be significantly different.

	Linear Static Buckling Loads (kN)			
CLT Layup	Buckling Mode			
	1	2	3	4
Trial 0	367.8	1373.01	2119.71	2777.58
Trial 1	545.28	1974.33	3056.93	3838.1
Trial 2	758.49	2666.64	4153.66	49998.97
Trial 3	845.78	3057.44	4638.57	5930.67
Trial 4	1150.12	3998.16	6037.28	7401.39
Trial 5	1501.5	5026.91	7576.37	8925.38

Table 3.7: Trial CLT Layup Linear Buckling Capacity Results

In this case though, being a platform construction type building, the first mode buckling load is the load of interest, and higher buckling modes are not physically possible. The results of the



Figure 3.11: Critical Buckling Mode Shapes

buckling analysis for each of the five trial Layups are provided in Table 3.7.

In addition to the capacity factor applied to the design action effect, modification factors apply, specifically k_1, k_4, k_6, k_{11} , which are for the effects of duration of load, moisture condition, temperature, and size respectively.

CLT Layup	CLT Critical Buckling	CLT Design Buckling
	Load (kN)	Capacity, (ϕN) (kN)
Trial 0	367.8	209.65
Trial 1	545.28	310.81
Trial 2	758.49	432.34
Trial 3	845.78	482.09
Trial 4	1150.12	655.57
Trial 5	1501.5	855.86

Table 3.8: Trial CLT Layup Design Buckling Load Capacities

Assuming that the building has a design life of 50 years, and is built in normal conditions of temperature and humidity, all of the modification factors are unity, apart from k_1 . k_1 is the duration of load factor and is equal to 0.57. The calculated design compressive buckling capacity of the CLT panels is summarised in table3.8.

3.7.2 Sensitivity Analysis and Boundary Condition Check Results

The first mode critical buckling load for the Trial 1 CLT layup is 545.28 kN. Subdividing the model resulted in an increase in the number of brick elements from 300 to 3600. The model was then re-solved, and the resulting critical buckling load decreased to 545.13 kN, which is a net percentage decrease of 0.03 %.

This result indicates that the model has a sufficient element resolution to provide consistent results regardless of increasing the number of brick elements. The Trial 1 CLT layup first mode buckling shape for the normal and subdivided models are shown in figure 3.12.

The boundary condition check is used to identify if a wall length of 1m is sufficient to determine a unit length load, or if the wall length needs to be increased so that effects of local conditions



Figure 3.12: Trial 1 CLT Layup Model and Associated Subdivided Model



Figure 3.13: Trial 1 CLT Layup Model and Associated Boundary Condition Check Model

at the edge of the wall are made insignificant. To check the boundary condition, the the length of the Trial 1 CLT wall model was increased from 1 m to 2 m, and the results compared.

Boundary condition check results indicate that increasing the length of the Trial 1 CLT wall from 1 m to 2 m increases the critical buckling load from 545.13 kN to 546.58 kN. This increase represents a net percentage increase of 0.26%, which is insignificant. The Trial 1 CLT wall layup 1m and 2m long wall models are shown in figure 3.13.

3.8 Summary of FEA and Selection of Trial Experimental CLT Layup

Six trial CLT wall panel layups or 'cross sections', which Hyne could feasibly produce, were modeled and analysed in the Strand7 Finite Element package to determine their linear buckling capacities.

3.8 Summary of FEA and Selection of Trial Experimental CLT Layup

Results of the FEA indicate that the trial CLT layups have a much higher design buckling capacity (ϕN) than is required for the nominal load case (N^*). The thinnest of the six CLT layups, which comprises of three lamella, all 21 mm thick, has a design buckling capacity of 209.65 kN/m, whilst the design action effect for the nominal load case requires a load bearing capacity of only 43.7 kN/m. This represents a safety factor of 4.78.

The design buckling capacity (ϕN), has been modified to include the effects of moisture, temperature and duration of load in accordance with AS1720.1-1997,*Timber structures*, Part 1: Design Methods. It should be noted that the design buckling capacity derived from the FEA analysis does not take into account inhomogeneities in timber such as knots and sloping grain, and therefore is not conservative.

The CLT layup which best fits the criteria for experimental testing in terms of meeting the requirements of the intended market, being constructed of dimensional timber sizes currently being produced by Hyne and which can be tested on available test equipment, is the layup which consists of three lamella, each 22 mm thick (22/22/22).

Sensitivity analysis and boundary condition checks were undertaken on the finite element model to ensure its validity in terms of mesh resolution and wall length.

Chapter 4

Experimental Design

4.1 Introduction

In order to meet the requirements of the project specification, two experiments were required to be conducted, those being the in-plane load capacity and a bending test to determine the modulus of elasticity of the CLT panel. This chapter is intended to provide details of the two test procedures developed to determine the required properties.

Sample preparation for axial load testing at the University of Southern Queensland involved trimming three samples provided by Hyne to 2.7 m long so that they would fit between the load plates of the axial load test rig. No further preparation of the samples was required for either of the tests.

Though the bending test was not specifically included in the project specification, it was required to check the validity of the accepted formula for the buckling strength of plywood diaphragms. Also included in this chapter is an overview of the fabrication of the experimental test samples prepared by Hyne. Six test samples were produced to specification, each measuring 275 mm \times 68 mm \times 3000 mm.

4.1 Introduction

4.1.1 Materials

Machine grading equipment at the Hyne & Son Tuan Mill in Maryborough, QLD, optimizes the yield of high grade timber by detecting and removing sections of lower grade timber, such as knots and timber with sloping grain, from otherwise high grade stock. The 25 mm \times 90 mm boards used to fabricate the CLT samples were tested, but not graded, during milling.



Figure 4.1: Locations of 25 mm \times 90 mm Boards within Parent Log

Typical commercial application of the 25 mm \times 90 mm timber boards produced by Hyne is for use as fence posts and other similar non-structural applications. The boards are quartersawn from the outside of the parent log as shown indicatively in figure 4.1. For Australian Slash and Caribbean Pine, boards cut from positions close to the outside of the log typically have a higher modulus of elasticity than timber from the rest of the log.

Prior to fabricating the sample panels, tests were conducted to determine the density and acoustic modulus of elasticity of the dimensional timber. The moisture content was tested using a Wagner Hand Held Moisture Meter similar to the one shown in figure 4.2, whilst the dimensions, mass and density were drawn from records taken by the proprietary testing equipment used at the Hyne mill.



Figure 4.2: Wagner Moisture Meter

4.1.2 Sample Fabrication Procedure

The sample width was dictated by the requirement for handling during testing at the University, and also dimensions of the boards used in the sample fabrication. The dimensions of the boards were $22 \text{ mm} \times 90 \text{ mm}$. The samples were fabricated three boards wide (270 mm) for simplicity of fabrication, and to reduce boundary condition effects on the sides of the wall samples.

The process of fabricating the CLT samples panels is as follows...

- The boards for the first CLT lamella were assembled on an LVL platen press
- Glue was evenly applied to the exposed upper surface of the first lamella
- The second lamella was placed over the first lamella noting orthogonal grain orientation
- Glue was evenly applied to the exposed upper surface of the second lamella
- A final check that all of the boards were assembled as closely as possible was undertaken
- A 0.5 MPa pressure was applied using the press, and the assembly was left to cure

4.1 Introduction



• After a curing time of 25 min, the sample was removed from the press

Figure 4.3: Assembly of CLT Panel

During production of the first sample CLT panel, a load was applied to the sides in order to compress the individual timber elements together on their short edge, however, this resulted in the upper lamella arching up in the center. Subsequent samples were produced without any side loads. An exploded view of the CLT sample in the LVL platen press is provided in figure 4.3.

The glue used in the production of the CLT panels was Purbond HB S109, which is a new generation single component isocyanate adhesive developed by Purbond. The adhesive has an open time (or working time) of 10 min, and a press time of 25 min before curing. A chemical reaction between moisture in the atmosphere and the glue causes the glue to cure (Purbond 2010).

Given the short time frame for assembly of the dimensional timber elements, it was possible that some areas of glue on the first laminate tacked-off prior to pressing. KLH Massivholz GmbH, a recognised CLT fabricator in Europe, recommends pressing laminates with a pressure of 60 kPa to achieve a high quality adhesion (KLH Massivholz GmbH 2010). The LVL press used to fabricate the samples could achieve a maximum pressure of 0.5 MPa.

To reduce the possibility of moisture content variation between measuring the moisture content between fabrication and testing, the samples were kept undercover at all times prior to testing at the University of Southern Queensland. The fabricated CLT samples are shown in figures 4.4.



Figure 4.4: CLT Samples Fabricated by Hyne

4.2 Axial Load Capacity of CLT Wall Panels

4.2.1 Introduction

This section provides an overview of the experimental design of the axial load test. A review of the technical aspects of the experiment followed by development of detailed experimental procedures is undertaken.

A number of samples fabricated by Hyne to the same specification were tested. Ideally the samples would have been tested to the full 3.0 m wall height as defined by the nominal residential building case, however the maximum height which could be tested in the axial load test rig at the University of Southern Queensland was 2.7 m.

When timber is loaded axially, typical failure modes are normal compressive failure, beamcolumn failure and buckling failure, depending on the geometry of the sample and applied load conditions. Other failure modes may result from flexure of the wall.

The slenderness ratio and stiffness of the wall determine the critical elastic buckling load. Due to assumptions made in the development of Euler's theory, the actual ultimate buckling load can be much less than the theoretical critical buckling load. The critical buckling load determined by Linear Static FEA is the theoretical upper limit of axial strength within the elastic limit of

the material.

The test setup was chosen to provide an indication of the wall's load capacity in terms of serviceability and ultimate strength. The serviceable limit for mid-height deflection as defined for the nominal residential multistorey building is $\frac{h}{300}$, where *h* is the height of the wall. Therefore, the mid height deflection of the CLT wall panel was measured and logged in relation to the applied load.



Figure 4.5: Axial Load Test, Loads of Interest

Axial deflection, though not specifically defined in AS1170.2 as a serviceability criterion for walls, was measured and logged in relation to the applied load. All three CLT wall samples were loaded to destruction to determine their ultimate axial buckling load capacity. The loads and displacements required to be recorded in the axial load test are shown diagrammatically in figure 4.5.

4.2.2 Test Procedure

The following equipment was used to undertake the axial load testing:

- 1. 240V Circular Saw
- 2. 2000kN Axial load test rig fixed to a concrete strong floor

- 3. 300t Hydraulic ram and associated hydraulic power-pack (Enerpac models AC CLRG 30012 and Enerpac ZE5 class respectively)
- 4. 20t Load cell (Anyload Model 363YH)
- 5. Safety cage and associated fasteners for fixing to the test rig columns
- 6. Load top and bottom plates, and associated chain restraints
- 7. $2 \times$ Cable extension position transducers (Celesco PT Series)
- 8. Screw to attach cable to sample for mid-height deflection measurement
- 9. Sturdy stand to mount the mid height deflection cable extension position transducer
- 10. Numerous G-Clamps
- Signal processor and computer with proprietary laboratory calibration and data recording software

Steps involved in setting up for axial load test are as follows:

- As the CLT wall samples requested from Hyne were 3.0m high, and the maximum height which would fit into the test rig was only 2.7m, the three samples which were to be used for Axial load testing were measured and docked to length using a circular saw.
- 2. The axial load test rig also needed to be setup to accept the 2.7m tall samples. This involved repositioning the cross beam of the test rig to the maximum height possible. Access to fasten the bolts for the test rig was achieved using a dog box on a forklift. The bolts retaining the cross beam were tensioned loosely with a hand wrench.
- 3. Next, the top load plate was installed. The top load plate distributed the load from the hydraulic ram evenly across the top of the CLT sample. Short chains were installed at either end of the load plate, so that when the sample failed, the load plate was restrained. The chains were then shackled to lugs mounted on the sides of the hydraulic ram, with sufficient additional length that the full travel of the ram was still achievable. The top plate and chains are shown in figure 4.6.



Figure 4.6: Load Application and Vertical Deflection Measurement



Figure 4.7: Safety Cage, load cell and Sample Panels

4.2 Axial Load Capacity of CLT Wall Panels

- 4. The Y-axis Cable Extension Position Transducer was then securely mounted to the underside of the test rig cross beam with two G-Clamps. The cable from the position transducer was then attached to a lug welded to the top load plate as shown in figure 4.6. Attaching the cable to the top load plate isolated the extension cable from the sample. Isolating the extension cable from the sample was necessary to limit damage to the instrument upon failure of the sample.
- 5. A mesh safety cage was then installed between the test rig columns. The mesh on one side of the safety cage was removable, allowing the samples to be positioned inside the cage. Rather than remove the cage each time a new sample was required, all of the samples were positioned inside the cage, and restrained against the test rig column as indicated in figure 4.7.
- 6. The bottom load plate and 200t load cell were then positioned at the bottom of the test rig.
- 7. The Z-axis Cable Extension Position Transducer was then bolted to a specially fabricated rigid steel stand as indicated in figure 4.8a. The stand was positioned in front of the wall and the base packed to ensure stability. The height of the position transducer was then adjusted to match the mid-height location of the wall sample.
- 8. After all of the instruments were connected to the signal processor/datalogger, which is shown in figure 4.8c, the first sample was positioned in place, and a small starting load applied to hold it.
- A small self tapping screw was then installed at mid-height of the wall sample, and the Z-axis Extension Position Transducer attached to it.
- 10. The removable safety cage mesh was then installed as shown in figure 4.8b, and all of the instruments zeroed, ready for testing.
- 11. Load was applied over a couple of minutes to achieve a good data capture resolution, and to achieve failure within a five minute period. Once the sample had failed, the mid-height deflection anchor point (self tapping screw) was removed.
- 12. This procedure was repeated for three samples



(a) Z-Axis Transducer Stand





(c) Signal Processor and Computer

(b) Refitting of Mesh



(d) Hydraulic Power Pack

Figure 4.8: Test Equipment Setup
4.3 Effective Modulus of Elasticity of CLT Wall Panels

4.3.1 Introduction

For clarity, the term 'apparent modulus of elasticity' will be used in this report to describe the modulus of elasticity of solid timber including knots and other discontinuities, whereas the 'effective modulus of elasticity' will be used to describe the modulus of elasticity of the CLT panels.

The modulus of elasticity is the single most useful piece of information to describe the strength properties of dimensional timber. The effective modulus of elasticity of CLT panels includes additional information about the homogenising effects of laminating, laminate geometry and grain direction.



Figure 4.9: Four Point Bending Test (Standards Australia 2010a)

The method for determining the apparent modulus of elasticity of solid timber is the four point bending test, as described in chapter two of AS4063.1-2010, Characterization of structural timber. This method was adopted, with some modification, for testing the CLT wall panel effective modulus of elasticity. The four point bending moment test according to AS4063.1-2010 is shown in Figure 4.9.

The sample is to the same specification as that used for the axial load capacity test, which



(a) AS4063.1 Sample Cross Section





Figure 4.10: CLT VS AS4063.1 Sample Cross Section Notation (Standards Australia 2010a, p.5)

4.3 Effective Modulus of Elasticity of CLT Wall Panels

is a layup of 22 mm thick x 90mm boards in three layers, with the central lamella placed orthogonally to the outside lamella (i.e. 22/22/22 layup). The breadth of the sample tested was 270 mm, and the length of the test piece was 3.0m, using the notation provided in figure 4.9.

Modification of the four point bending test was required to ensure a bending type failure of the test samples as opposed to shear failure. According to AS4063.1 (Standards Australia 2010a), the breadth of the test sample is the lesser cross-sectional dimension of the test piece, as indicated in Figure 4.10a.

The CLT wall panel sample has a very different cross section as shown in figure 4.10b, and when using three point test setup ratios indicated in figure 4.9, indicates a span between centres of approximately 1.2m.



Figure 4.11: Modified Four Point Bending Test Dimensions

Given that the shear capacity of the glue was unknown, and considering the sample width to depth ratio, the test setup was rationalised by increasing the span between supports from 1.2 m to 2.0 m to induce a bending type. The rationalised dimensions of the modified four point bending test are shown in Figure 4.11.

Considering the application of the CLT panels will have a span between centres of 3.0 m

or greater in most cases, it follows that increasing the span to 2.0 m would provide more representative results.

In any case, the modulus of elasticity could still be determined even if the failure mode was shear failure, provided a clear linear elastic load range could be determined from the load vs displacement plot. According to AS4063.1 (Standards Australia 2010*a*, p.8), the equation for the modulus of elasticity is...

$$E = \frac{23}{108} \left(\frac{L}{d}\right)^3 \left(\frac{\Delta F}{\Delta e}\right) \frac{1}{b}$$
(4.1)

where $\frac{\Delta F}{\Delta e}$ is the linear elastic slope of the load-displacement graph.

As the four point bending test is a standard test, no further theory or procedures are provided, and the reader is directed to chapter two of AS4063.1 (Standards Australia 2010*a*) for further details.

4.3.2 Test Equipment

Specific equipment required for the Modified Bending Test at the University of Southern Queensland is shown in figure 4.12, and itemised as follows...

- 1. Universal beam test rig with concrete strong floor
- 300t Hydraulic ram and associated hydraulic power-pack (Enerpac models AC CLRG 30012 and Enerpac ZE5 class respectively)
- 3. 20t Load cell (Anyload Model 363YH)
- 4. Load spreader beam with rollers and load plates to limit perpendicular to grain local crushing
- 5. Universal beam supports with load plates to limit perpendicular to grain local crushing
- 6. Cable Extension Position Transducer (Celesco PT Series)
- 7. PC with proprietary laboratory calibration and data recording software (not shown)



Figure 4.12: Four Point Bending Test Equipment

Chapter 5

Results and Discussion

5.1 Axial Load Test Results and Discussion

Due to the high axial loads, and hence large stored potential energy during testing, it was initially decided that the deflection should be limited to around 150 mm. Exceeding this limit posed a safety risk.

The first sample tested reached it's ultimate axial load capacity, then retained a degree of post failure strength right up until the 150 mm deflection safety limit was reached. Some loud cracking noises were heard as the load was applied.

It was not clear at exactly when the ultimate load was reached without monitoring the load during the testing because the failure was not rapid or violent. The first sample was then removed from the test rig and the testing continued with the second and third samples. A photo of the sample loaded showing post ultimate load deflection is provided in figure 5.1.

Apart from a small deflection, the second sample reached it's ultimate axial load capacity without indication of impending failure. The failure was very rapid, and a loud cracking noise was emitted.

The second sample buckled in the opposite direction to the first and third test samples. The second sample was removed and replaced with the third test sample for the final axial load test.



Figure 5.1: Sample 1, Post Ultimate Load Deflection



(a) Just Prior to Failure



(b) Post Failure

Figure 5.2: Sample 2. Photo Sequence

Sample No.	Maximum Service	Ultimate Axial
	Load (kN)	Load (kN)
1	350	370
2	200	270
3	340	345
Mean Average	297	328

Table 5.1: Service and Ultimate Axial Load Summary

The failure mode of the third test sample was similar to the second.

A two photo sequence of the load testing of sample two is provided in Figure 5.2. The first photo was taken just prior to failure.

A plot of axial load versus the displacement in the Y-direction (i.e. change in height of the wall) for all three samples is included in figure 5.3. The plot indicates that the ultimate axial load for the three samples varies between 275 kN and 370 kN.

All three samples had similar elasticity, however, the first sample showed signs of slipping and load redistribution resulting in a 1mm initial displacement. Axial deflections at ultimate axial loads ranged from approximately six to eight millimeters.

A plot of the axial load versus mid-height deflection is shown in figure 5.4. From the plot, it can be seen that the mid-height horizontal displacements for the samples at ultimate axial load ranged between 10 mm and 28 mm. All of the plot shapes were consistent.

The second and third sample post ultimate load strength terminated at approximately 35 mm deflection, as indicated by the straight line thereafter, whilst sample one retained some post ultimate failure load capacity at nearly 40mm deflection.

In meeting the aims of the axial load test, Table 5.1 provides a summary of the ultimate and service loads achieved by the three CLT samples.

Failed samples two and three were retained and disassembled with a pry bar to ascertain information on what may have caused the failure. Failure was in general along the glue line. It is not clear if the failure of samples two and three emanated from the top of the sample, or from





Figure 5.4: Axial Load VS Mid-Height Horizontal Displacement



Figure 5.5: Fracture Path for Samples Two and Three

the inner compressive fibre, mid height.

In each case, the timber fibres were observed to be retained in over 30 % of the delaminated glue area, indicating insufficient glue adhesion to approximately 70 %. A diagram showing the general path of the fracture line for samples two and three is included in figure 5.5.

A photo of the the delaminated glue surface showing retained timber fibres is indicated in figure 5.6. It was also noted that the timber which was retained in the glue came from the central laminate.

5.2 Four Point Bending Test Results and Discussion

The four point bending test was undertaken in accordance with AS4063.1-2010, Characterization of structural timber, with modification to the span of the supports, as indicated in Figure 4.11 at the University of Southern Queensland Centre of Excellence in Engineered Fibre Composites (CEEFC) laboratory, Toowoomba.

The test was required so that an effective modulus of elasticity for the CLT panel could be de-



Figure 5.6: Sample Three Delaminated Glue Surface

5.2 Four Point Bending Test Results and Discussion

termined. The results will be used for comparison with the modulus of elasticity as determined from the plywood composite theory, which is discussed in more detail in Chapter 5.3.

On application of the load a noticeable mid-span deflection increased linearly, until failure. There was the occasional cracking sound at the sample flexed, followed by a clear cracking sound when the ultimate load was reached.

The CLT lamella had delaminated between the load application points and the supports at either end of the panel, as indicated in figure 5.7, and the convex shape of the sample during loading changed to include contra-flexure at the ends of the sample as shown in figure 5.8.

Though the test setup aimed to avoid a shear type failure, it was immediately evident that shear failure along the glue line was the failure mode as indicated in figure 5.9. This is verified by observing the locations of maximum shear in the shear force diagram for a four point bending test shown in figure 5.10.



Figure 5.7: Four Point Bending Test:Elastic Region

The linearity of the elastic region of the load versus displacement plot as indicated in figure 5.11, was surprisingly smooth apart from one small glitch at approximately 16 kN of load,



Figure 5.8: Four Point Bending Test: Post Failure



Figure 5.9: Delamination of CLT Due To Flexural Shear



Figure 5.10: Shear Force and Bending Moment Diagram (American Forest and Paper Association, Inc., American Wood Council 2007, p.8)

possibly caused by load redistribution in the CLT panel. The effective modulus of elasticity was determined using equation 4.1, for points one and two indicated in 5.11, as follows...

$$E = \frac{23}{108} \left(\frac{L}{d}\right)^3 \left(\frac{\Delta F}{\Delta e}\right) \frac{1}{b}$$
$$= \frac{23}{108} \left(\frac{2.0}{0.066}\right)^3 \left(\frac{(14.5 - 10) \times 10^3}{(25 - 6.5) \times 10^{-3}}\right) \frac{1}{0.27}$$
$$= 5.34 \text{ GPa}$$

Therefore, the sample CLT effective modulus is much less than that of the timber elements from which it is composed. The timber elements tested by Hyne wee found to have a mean acoustic modulus of elasticity of 17 MPa.





5.3 AS1720.1 Appendix I - Buckling Strength of Plywood Diaphragms

Initially, it was thought that section five of AS1720.1, which deals with the design of plywood, would be the most likely formula to apply to the design of CLT panels, however, given the large panel area to thickness ratio of CLT wall panels, investigations moved toward using AS1720.1 Appendix I - Buckling Strength of Plywood Diaphragms. Appendix I takes into consideration the size of the diaphragm with a plywood assembly factor g_{61} .

According to AS1720.1, Appendix I - Buckling Strength of Plywood Diaphragms is applicable where large sheets of thin plywood are used in composite construction, and it is possible for buckling distortions to cause a reduction in the load capacity of the plywood membrane (Standards Australia 2010*b*, p.161). Typical examples are webs of I-beams and plywood box beams.

Section I2.3 of AS1720.1 Appendix I, refers to the calculation of buckling strengths of plywood 'diaphragms with lateral edges free and subjected to uniformly loaded edge forces' (Standards Australia 2010*b*, p.163).

In the case of a CLT wall panel, the load is typically considered uniformly distributed by evenly spaced joists and continuously from walls above. No consideration is given to concentrated edge loads, such as floor bearers and roof beams. Point load considerations are outside the scope of this project.

The buckling strength of plywood diaphragms is determined using the same method as that used for the design of timber compression members, with the exception being the calculation of the slenderness coefficient S. The slenderness coefficient is required for determination of the stability factor K_{12} . The reader is referred to section 2.7 of this document for background information on the design of timber compression members.

The buckling strength modification factor for diaphragms with lateral edges free and subjected to uniformly loaded edge forces is given by the following equation...

$$S = g_{61} \frac{d_{\rm w}}{t_{\rm w}} \tag{5.1}$$

Where g_{61} is a factor describing the layup of the plywood, in terms of the number of lamella, direction of grain, ratio of thickness of outer plies to that of the inner plies, d_w is the depth of the web, and t_w is the thickness of the web. For CLT wall sample panels, the depth of the web is actually the height of the sample which was 2.70 m, and the thickness of the web is the overall thickness of the CLT wall panel sample, which was .066 m.

The slenderness coefficient for the Hyne CLT sample panel, according to equation 5.1 is...

$$S = g_{61} \frac{d_{w}}{t_{w}}$$

= 2.2 × $\frac{2.7}{0.066}$
= 90 (5.2)

Where $g_{61} = 2.2$ for a three ply CLT panel with equal lamella thickness, as per table I3 (Standards Australia 2010*b*, p.165)

The stability factor k_{12} is calculated from equation 3.3(11c)of AS1720 (Standards Australia 2010*b*, p.38) as follows...

$$k_{12} = \frac{200}{(\rho_{\rm c}S)^2}$$
$$= \frac{200}{(1.0 \times 90)^2}$$
$$= 2.47 \times 10^{-2}$$
(5.3)

and putting this into equation 2.7 yields a design compressive strength of...

$$\frac{N_{d,c}}{\phi} = k_1 k_4 k_6 k_{12} f'_c A_c$$

= 1.0 × 1.0 × 1.0 × 0.025 × 0.01782 × 56.1 × 10⁶
= 24.95kN (5.4)

It is interesting to note that in the calculation of the axial in-plane loading for plywood in accordance with Section Five of AS1720.1, the area used is the cross sectional area of plies with their grain parallel to the load, whereas for S1720.1 Appendix I - Buckling Strength of Plywood Diaphragms, the cross sectional area is that of the entire plywood cross section (Standards Australia 2010*b*, p.163).

Further, according to section I3.2 of Appendix I of AS1720.1, two second moments of area for plywood are required to describe plywood section properties. The first is used when determining the bending capacity, and the second for determining the bending rigidity.

Plies which run perpendicular to the direction of load are not considered when determining the bending capacity. When determining the bending rigidity of plywood, an additional amount of rigidity is included to account for the stiffness of plies which run perpendicular to the load direction. Non-inclusion of the additional second moment of area when calculating bending rigidity would lead to conservative design.



Figure 5.12: Plywood Cross Section and Notation(Standards Australia 2010b, p.167)

Though the section modulus for bending rigidity includes information about rigidity of plies running both parallel and perpendicular to the grain, neither the calculation of the compressive strength of plywood in section 5, nor buckling strength of plywood diaphragms in Appendix I, use the modulus of bending rigidity to determine the buckling capacity. The following equation is for the second moment of area for calculation of bending rigidity (Standards Australia 2010*b*,

p167), and should be read in conjunction with figure 5.12...

$$I_{\rm R} = 2\left(\frac{1}{12}bd_1^3 + A_1\bar{y}_1^2\right) + 2 \times 0.003\left(\frac{1}{12}bd_2^3 + A_2\bar{y}_2^2\right) + \frac{1}{12}bd_3^3 \tag{5.5}$$

Where $I_{\rm R}$ = second moment of area about the neutral axis for bending rigidity calculations for a width b = 1 mm, in mm⁴. $A_1 = d_1 b$, $A_2 = d_2 b$ and 0.03 = factor applied for plies at right angles to the span.

Considering that the buckling capacity, especially for slender columns, is dominated by the stiffness of the section, and not the section capacity, the calculations for in-plane compressive strength as indicated in AS1170.2 are inconsistent with the requirements for determining the axial in-plane capacity of CLT wall panels of buildings, which are typically considered intermediate to slender.

The bending rigidity of the CLT sample is calculated according to equation 5.5, for comparison with the actual bending rigidity of the sample as follows...

$$\begin{split} I_{\rm R} &= 2\left(\frac{1}{12}bd_1^3 + A_1\bar{y}_1^2\right) + 0.003\left(\frac{1}{12}bd_2^3\right) \\ &= 2 \times \left(\frac{1}{12}\right) \times 270 \times 22^3 + 0.003 \times \left(\frac{1}{12}\right) \times 270 \times 22^3 \\ &= 6.23 \times 10^6 {\rm mm}^4 \end{split}$$

whereas the actual effective second moment of area for the CLT test sample, as calculated from the results of the four point bending test using accepted deflection formula (American Forest and Paper Association, Inc., American Wood Council 2007, p.8), is as follows...

$$I = \frac{Pa}{\triangle_{max} \times 24E} (3l^2 - 4a^2)$$
$$= \frac{20 \times 10^3 \times 0.67}{0.035 \times 24 \times 17 \times 10^9} \times (3 \times 2^2 - 4 \times 0.67^2)$$
$$= 9.575 \times 10^6 \text{ mm}^4$$

The load of 20 kN and deflection of 35 mm were taken from the load deflection plot of the four point bending test in figure 5.11.

This result indicates that for a CLT wall panel, using the plywood composite theory to calculate section properties underestimates the contribution of plies with grain perpendicular to the load direction.

5.4 Development of Load vs Height Charts for the CLT Sample Panel

The secant formula (equation 2.7) was used to develop a theoretical load chart for the CLT sample layup (22/22/22), based on the effective modulus of elasticity of the CLT cross section as determined from the four point bending test, and the maximum compressive strength of the timber.

The load chart is shown in figure 5.13 is a theoretical load chart which assumes pin connection at the top and bottom of the wall. Various load eccentricities are indicated on the plot.

These loads are based on the mean modulus of elasticity of the CLT samples, and should not be used for design purposes. Fifth percentile characteristic strengths are significantly less. The matlab code, including variables used to calculate and plot the graph is included in Appendix 9.





Chapter 6

Conclusions and Further Work

6.1 Summary

Investigation into serviceability aspects of CLT panels was undertaken, including preliminary calculations for required fire cover.

Finite element analysis (FEA) was used to specify a CLT panel layup to meet the requirements of the low rise multistorey residential market, as well as those of Hyne Pty Ltd. Samples of the proposed CLT panel were fabricated by Hyne and subsequently tested at the University of Southern Queensland.

Accepted formula from AS1720.1 for the buckling strength of plywood diaphragms were checked for application to CLT panels by comparing theoretical results with experimental results. Theoretical load charts were developed based on experimental results and accepted engineering theory.

6.2 Achievement of Project Objectives

The following is a brief discussion on each of the project objectives outlined in the project specification...

6.2 Achievement of Project Objectives

Research background information relating to the structural performance of cross laminated timber for wall applications. The research will include fire resistance, deflection and ultimate strength prediction of CLT panels.

Research indicates that fire resistance of CLT panels is similar to that of normal timber, and charring factors can be calculated directly from AS1720.4-2006 Timber Structures Part 4: Fire resistance for structural adequacy of timber members. For external CLT walls, only one side of the panel is exposed to elevated temperatures.

Axial load testing of the Hyne CLT samples indicate that mid-height deflection at ultimate load is well inside the advised limits of AS1170.1 for walls.

Use FEA to investigate structural properties of internal and external CLT wall panels for a number of CLT cross sections which Hyne can feasibly produce.

Finite element linear static buckling analysis was undertaken using Strand7, as well as sensitivity analysis to ensure adequate mesh sizing. The analysis extended to multiple CLT cross sectional layups which could be feasibly fabricated by Hyne.

Evaluate the results of the FEA modeling and select a CLT panel cross section for compression load testing.

A criteria was developed for selection of a suitable CLT panel, which included available dimensional timber element sizes, the proposed market to which the CLT panel is aimed. A nominal load case was developed based on a low-rise multistory residential building representing the proposed market.

Evaluation of the results from FEA concluded that the smallest CLT panel of all the trial sections was, in terms of strength, more than adequate for the intended purpose.

Hyne then fabricated a total of six CLT sample panels to the selected single specification, for destructive testing at the University of Southern Queensland. Each CLT sample panel consisted of three orthogonally placed lamella, each 22 mm thick. The sample wall panel overall dimensions were 3.0 m tall, and 270 mm long.

Undertake laboratory testing of the selected CLT panel cross section for buckling strength

6.2 Achievement of Project Objectives

when loaded in-plane. Compare theoretical results with actual load test results.

The two laboratory tests were performed including an axial load test and four point bending test. A third test (shear test) was also undertaken so that, if time permitted, a theoretical analysis of the effects of hole openings in CLT panels could be investigated.

A comparison between theoretical results from finite element modeling and actual load test results revealed large disparities. These are believed to be related to the wall end connections.

The connections were modeled as pin end connections in the Finite Element software, which is loosely correct, as there is no rigid connection at the ends.

The test setup did not accurately represent the pin ends of the Finite Element model, and though not rigidly connected, the ends of the wall panels were restrained between the load plates, resulting in higher than expected load capacities.

Compare theoretical and laboratory results with accepted formulae from AS1720.1 Appendix I - Buckling Strength of Plywood Diaphragms.

To get a full understanding of how the accepted theory for plywood in-plane buckling load capacity reflects the material properties of CLT, a four point bending test was first undertaken.

An effective second moment of area for the experimental sample was then calculated for comparison with section properties calculated using accepted formulae from AS1720.1 Appendix I - Buckling Strength of Plywood Diaphragms.

The buckling strengths and section properties determined using Appendix I significantly underestimated the bending stiffness and axial load capacity of the sample CLT panel.

Provide theoretical internal and external CLT wall load charts for various common wall heights and report on the effects of fire.

A theoretical lineal wall load vs wall height chart has been developed using the secant formula, for a number of different material or load eccentricities.

Undertake compression load testing of various common wall heights to verify results of theo-

retical analysis, as well as a theoretical analysis of the effects of hole openings in CLT wall panels

Time was not permitting to undertake further work set out in the project specification.

6.3 Conclusion

CLT wall panels are already being used overseas, to the point where standards for design and construction have been developed and implemented. The benefits of CLT as a building material have been proven by the many projects already completed overseas on a purely economic basis. Environmental benefits of CLT as compared to other construction materials are widely published, and as governments move towards carbon neutral development and use of sustainable resources, CLT construction will surely gain more popularity in Australia.

This project has identified that existing timber standards cannot be used directly for CLT design, as they do not accurately represent the strength of CLT panels. Though produced from low grade timber, the CLT sample panels have load capacities far in excess of what is typically required for residential low-rise buildings.

6.4 Recommendations and Further Work

Further work is required to develop FEA models which more accurately represent the end connection details of the walls when they are installed in buildings. Many floor systems and connections details are possible for floors and walls, and so a number of models need to be developed.

Standards specific to the fabrication, testing and design of CLT building elements are required so that mainstream adoption of CLT panels as an option to existing building materials is possible. Connection details, shear capacity, vibration and transmission, and weather protection methods are all areas which need to be developed in relation to walls.

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Chapter 7

Project Specification

ENG 4111/2 (or ENG8002) Research Project

Project Specification

For:	Alan Turner	
Topic:	STRUCTURAL PERFORMANCE OF CROSS LAMINATED TIMBE	
	PANELS AS WALLS (INTERNAL AND EXTERNAL)	
Supervisors:	Associate Prof. Karu Karunasena	
	Geoff Stringer, Hyne and Son Pty Ltd.	
Sponsorship:	HYNE & SON PTY. LIMITED	
Project Aim:	To analyse the structural performance of Cross Laminated Timber	
	(CLT) panels manufactured from non-structural grade timber, for	
	internal and external wall applications in terms of serviceability	

Program:

and strength.

- 1. Research background information relating to the structural performance of cross laminated timber for wall applications. The research will include fire resistance, deflection and ultimate strength prediction of CLT panels.
- 2. Use FEA analysis to investigate structural properties internal and external CLT wall panels for a number of CLT cross sections which Hyne can feasibly produce.
- 3. Evaluate the results of the FEA modeling and select a CLT panel cross section for compression load testing.
- 4. Undertake laboratory testing of the selected CLT panel cross section for buckling strength when loaded in-plane. Compare theoretical results with actual load test results.
- Compare theoretical and laboratory results with accepted formulae from AS1720.1 Appendix I - Buckling Strength of Plywood Diaphragms.
- 6. Provide theoretical internal and external CLT wall load charts for various common wall heights and report on the effects of fire.
- 7. Submit an academic dissertation on the research

As time and resources permit:

- 1. Undertake compression load testing of various common wall heights to verify results of theoretical analysis
- 2. Undertake a theoretical analysis of the effects of hole openings in CLT wall panels

Agreed:

Student Name:	Alan Turner
Date:	15/03/2010
~	
Supervisor Name:	A/Prof. Karu Karunasena
Date:	15/03/2010
Examiner/Co-Examiner:	Approved
Date:	

Chapter 8

Moisture Content and Acoustic Modulus of Elasticity
Figure 8.1: Moisture Content and Acoustic Modulus of Elasticity Data Received From Hyne

Ē

	MoistCon	Width	Thickness	Length	Mass	Frequency	Density	Acc E
Board #	(%)	(mm)	(mm)	(mm)	(kg)	(Hz)	(kg/m ³)	(GPa)
1	14.6	93	23	3024	4.637	855	717	19.170
2	12.3	93	23	3024	3.797	829	587	14.758
3	16.2	93	23	3019	4.510	879	698	19.672
4	13.9	93	23	3021	4.376	911	677	20.516
5	14.6	93	23	3025	4.475	831	692	17.481
9	17.3	93	23	3021	4.272	852	661	17.517
7	<u>1</u> 3.9	93	23	3024	4.033	814	623	15.111
8	13.7	93	23	3027	4.537	879	701	19.841
6	14.7	93	23	3008	4.407	856	685	18.163
10	15.4	93	23	3027	4.143	824	640	15.925
11	13.1	93	23	3023	3.819	788	591	13.407
12	10.9	93	23	3024	4.343	821	671	16.555
13	14.0	93	23	3024	3.682	784	569	12.796
14	15.2	93	23	3027	4.952	903	765	22.856
15	13.7	93	23	3022	3.178	809	492	11.754
Avg	14.2							
						AV	,G	17.035
						STD	EV	3.122
						00	N	0.183

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Chapter 9

MATLAB Source Code

9.1 Load Vs Displacement Experimental Data Plotting Programs

```
%function [ output_args ] = Untitled1( input_args )
%This function plots the axial and mid-height
%deformation of the sample under axial load
clear;
clc;
%Load and post process raw data files load compl.dat;
xdispcomp1=comp1(:,4).* -1;
ydispcomp1=comp1(:,3);
loadcomp1=comp1(:,5)./1000;
load comp2.dat;
xdispcomp2=comp2(:,4);
ydispcomp2=comp2(:,3);
loadcomp2=comp2(:,5)./1000;
load comp3.dat;
xdispcomp3=comp3(:,4).* -1;
ydispcomp3=comp3(:,3);
loadcomp3=comp3(:,5)./1000;
%Plotting - Axial Load VS Y-Axis Deflection
figure('MenuBar', 'none');
set(gcf, 'position', get(0, 'ScreenSize'));
h=plot(ydispcomp1, loadcomp1, ydispcomp2, loadcomp2...
,ydispcomp1, ioadcomp1, ydispcomp2, ioadcomp2...
,ydispcomp3, loadcomp3);
set (h, 'LineWidth',1);
set (gca, 'FontSize',13, 'LineWidth',1);
xlabel('Y-Axis_Displacement_(mm)');
ylabel('Load_(kN)');
%title('Axial Load VS Y-Axis Deflection', 'Fontsize',15);
arid.
grid:
set(gca, 'XTick', [0:1:10]);
```

9.1 Load Vs Displacement Experimental Data Plotting Programs

```
label1='Sample_1';
label2='Sample_2';
label3='Sample_3';
legend(label1,label2,label3);
orient landscape;
print('-djpeg','axialplot1.jpg');
close;
%Plotting - Axial Load VS Mid-Height Deflection
figure('MenuBar', 'none');
set(gcf, 'position ',get(0, 'ScreenSize'));
h=plot(xdispcomp1,loadcomp1,xdispcomp2,loadcomp2,...
xdispcomp3,loadcomp3);
set(h, 'LineWidth',1);
stabel('Mid-Height_Displacement_(mm)');
ylabel('Load_(kN)');
%title('Axial Load VS Mid-Height Deflection', 'Fontsize',15);
grid;
set(gca, 'XTick',[0:5:70]);
label1='Sample_1';
label2='Sample_2';
label3='Sample_3';
legend(label1,label2,label3);
orient landscape;
print('-djpeg','axialplot2.jpg');
close;
%EOF
```

9.1 Load Vs Displacement Experimental Data Plotting Programs

```
%function [ output_args ] = Untitled1( input_args )
%This function plots the mid-span delfection vs
%applied load for the flatwise bending test
clear;
clc;
load bend1.dat;
xb1=bend1(:,3).*-1;
yb1=bend1(:, 4) . * 1 e - 3;
%Plotting - Mid-Span Delfection VS Applied Load
figure('MenuBar', 'none');
set(gcf, 'position', get(0, 'ScreenSize'));
h=plot(xb1,yb1);
set (h, 'LineWidth', 1);
set (gca, 'FontSize', 13, 'LineWidth', 1);
ylabel('Force_(kN)');
xlabel('Mid-Span_Deflection_(mm)');
title('Lord_Displacement Couch for CLT
title ('Load Displacement Graph for CLT Flatwise ...
Bending Test', 'Fontsize', 15);
grid;
set (gca, 'YTick', [0:5:25]);
orient landscape;
print('-djpeg', 'bending.jpg');
close;
%EOF
```

9.2 Secant Formula Buckling Linear Load Calculation Program

```
%function [ output_args ] = Untitled1( input_args )
%This function plots linear elastic column behavior
%using the Secant Formula the resultant plot can be
%used to asess the axial capacity of the CLT panel
%within the elastic region. The capacity is based on
%the maximum fibre stress when subjected to both axial
% and moment loading caused by eccentricities of the
%load application as well as and those caused
%by imperfections.
clear;
clc;
%External Lamella Thickness t1(mm)
t1 = 22;
%Central Lamella Thickness t2(mm)
t_2 = 22;
%Wall length (m)
WLength = 1.0;
%Compressive strength of outer fibre (Pa)
L1Fc = 56.1e6;
%Effective Modulus of Elasticity Determined ...
% from Testing (Pa)
E=5.34e9;
%Calculate Section Properties
Width = ((2*t1) + t2)/1000;
Area=WLength * Width;
Height = [0, 1.0, 1.5, 2.0, 2.4, 2.7, 3.0, 3.5];
I = 2 * (((WLength * (Width)^3))/12)
RGyration=sqrt(I/Area);
SRatio=Height./RGyration
Centroid=Width/2
% Loading
LEccentricity = [0, 0.001, 0.005, 0.01, 0.025, 0.05, 0.1, 0.25];
ERatio = (LEccentricity .* Centroid )./(RGyration .^2);
i = 0;
j =0;
k = 0;
for j=1:length (SRatio)
     k=0
     for k=1:length(ERatio)
         x=0
         while x<L1Fc
              i = i + 1e4
              x=i * (1+(ERatio(k) * sec((0.5 * SRatio(j)))...
                   *(sqrt(i/E))));
         end
          Axial(k, j) = i
         i = 0;
    end
end
%Plotting – Secant Formula Graph
Axial=Axial
AxialMPa = Axial . *1e-6
figure('MenuBar', 'none');
set(gcf, 'position', get(0, 'ScreenSize'));
```

9.2 Secant Formula Buckling Linear Load Calculation Program

```
plot (SRatio, AxialMPa);
set (gca, 'FontSize',13);
xlabel('Slenderness_Ratio_L/R');
ylabel('Axial_Stress_P/A_(MPa)');
%title('Secant Formula Graph For CLT Laminate
% (Timber Fc=56.1 MPa) for Various Load Eccentricities '
%, 'Fontsize ',15);
grid;
set (gca, 'XTick', [0:10:140]);
set (gca, 'YTick', [0:5:22]);
axis ([0 140 0 22]);
labell = '0 mm';
1abel2 = '1 \mbox{.mm'}
1abe13 = '10 \mbox{mm'}
label4 = '25 \text{ mm}';
label5 = '50 \text{ mm}';
label6 = '100 \text{ mm}';
1abe17 = 200 \text{ Jmm}
1abe18 = '300 \text{ Jmm'}
legend (label1, label2, label3, label4, label5, label6, ...
       label7, label8);
pause;
orient landscape;
print('-djpeg', 'eccentric.jpg');
close;
%Plotting - Wall Height VS Axial Load
LinealLoad = (Axial. * Area)
LinealLoadkN=LinealLoad.*1e-3
figure('MenuBar', 'none');
set(gcf, 'position', get(0, 'ScreenSize'));
plot(Height, LinealLoadkN);
set (gca, 'FontSize', 13);
xlabel('Wall_Height');
ylabel('Lineal_Load_(kN/m)');
%title('Height VS Lineal Load of CLT Panel (Max.
%Timber Fc=56.1Mpa) For Various Load Eccentricities'
%, 'Fontsize ',15);
grid;
set (gca, 'XTick', [1:0.5:3.5]);
set (gca, 'YTick', [0:200:1300])
axis ([0 4 0 1400]);
                         ,[0:200:1300]);
label1 = '0 \mbox{.mm'};
1abel2 = '1 \mbox{mm'}
1abel3 = 10 \text{ mm}^{\circ}
1abel4 = 25 \text{ mm}^{2}
1abe15 = 50 \text{ mm}
1abe16 = '100 \text{ mm'}
label = 200 \text{ mm}^{\circ};
label = 300 \text{ mm}^{\circ};
legend (label1, label2, label3, label4, label5, label6, ...
       label7 , label8 );
pause:
orient landscape;
print('-djpeg', 'loadheight.jpg');
close:
%EOF
```

Chapter 10

Purbond Adhesive Datasheet



*** SAFETY DATA SHEET ***

1. IDENTIFICATION OF THE SUBSTANCE/PREPARATION AND OF THE COMPANY/UNDERTAKING

Adhesive

PRODUCT NAME PURBOND® HB S109

RECOMMENDED USE

SUPPLIER

Purbond AG CH-6203 Sempach-Station Switzerland

EMERGENCY TELEPHONE: Tel. +41-(0)41-469-6863 (business hours)

info@purbond.com

2. HAZARDS IDENTIFICATION

EMERGENCY OVERVIEW	SENSITIZER IRRITANT HARMFUL LIMITED EVIDENCE OF A CARCINOGENIC EFFECT
EYE	Will cause eye irritation.
SKIN CONTACT	Repeated and/or prolonged contact may cause irritation and skin sensitisation.
INHALATION	Vapors and/or aerosols may cause irritation. May cause allergic respiratory reaction.
	Danger of serious damage to health by prolonged exposure through
	inhalation.
INGESTION	Low oral toxicity.
PHYSICO-CHEMICAL	No unusual hazards are expected.
ENVIRONMENTAL	Not readily biodegradable.

3. COMPOSITION/INFORMATION ON INGREDIENTS

CHEMICAL FAMILY	Isocyanate Adhesive		
<u>COMPONENT</u>	EC NUMBER	CONCENTRATION %	<u>EC SYMBOL/</u> RISK PHRASES
Diphenylmethanediisocyanate, isomers and homologues		10 - 20	XN XI R20 R36/37/38 R40 R42/43 R48/20
Diphenylmethane-4,4'- diisocyanate	247-714-0	5 - 10	XN XI R20 R36/37/38 R40 R42/43 R48/20

For full text of risk phrases, please refer to Section 16.

4.	FIRST-	AID	MEASURES
	TTTTT		

EYE	Immediate medical attention is not required. Irrigate with eyewash solution or clean water until pain is relieved.
SKIN CONTACT	Immediate medical attention is not required. Wash skin with soap and water. Remove grossly contaminated clothing, including shoes, and launder before re-use. Discard shoes.
INHALATION	Immediate medical attention is not required. Remove to fresh air. If breathing is difficult, give oxygen. If breathing has stopped, give artificial respiration. Get medical attention.
INGESTION	Immediate medical attention is not required. Treat symptomatically and supportively.

5. FIRE-FIGHTING MEASURES

EXTINGUISHING MEDIA SPECIAL FIREFIGHTING PROCEDURES	CO2; Dry Chemical; Foam Fire fighters should be equipped with self-contained breathing apparatus to protect against potentially toxic and irritating fumes.; Cool exposed equipment with water spray.
FIRE & EXPLOSION HAZARDS	Combustion will evolve toxic and irritant vapours.
HAZARDOUS COMBUSTION PRODUCTS LOWER EXPLOSION LIMIT	Decomposes upon heating to release toxic fumes of nitrogen oxides, carbon monoxide, carbon dioxide, and hydrogen cyanide. Not applicable
(%) UPPER EXPLOSION LIMIT (%) AUTOFLAMMABILITY FLASH POINT	Not applicable 480 °C > 200 C (Pensky-Martens Closed Tester)

6. ACCIDENTAL RELEASE MEASURES

SPILL AND LEAK	Adsorb spillages onto sand, earth or any suitable adsorbent material. Sweep
PROCEDURES	up and shovel into waste drums.

For safety and environmental precautions, please review entire Safety Data Sheet for necessary information.

7. HANDLING AND STORAGE

STORAGE TEMPERATURE HANDLING/STORAGE	Ambient. Store at room temperature. Vapours can accumulate at potentially hazardous levels in the unvented headspace of drums or bulk storage vessels. Open drums in ventilated area. Avoid breathing vapours or mists.
SENSITIVE TO STATIC ELECTRICITY	No
SPECIAL SENSITIVITY	Keep away from moisture.

8. EXPOSURE CONTROLS/PERSONAL PROTECTION

COMPONENT

EXPOSURE LIMITS

Diphenylmethanediisocyanate,	0.02 mg/m3 TWA
isomers and homologues	0.07 mg/m3 STEL
Diphenylmethane-4,4'-diisocyanate	0.02 mg/m3 TWA
	0.07 mg/m3 STEL

OCCUPATIONAL EXPOSURE CONTROLS

VENTILATION	Provide adequate ventilation to ensure that the defined occupational
REQUIREMENTS	exposure limit is not exceeded.
EYE PROTECTION	Safety glasses, goggles or face shield to protect against splashing. Personal
REQUIREMENTS	eye protection should conform to EN 166.
GLOVE REQUIREMENTS	Gloves are recommended due to possible irritation. Gloves should conform
	to EN 374.
CLOTHING REQUIREMENTS	Appropriate protective clothing and equipment is recommended to minimize
	skin contact with this substance.
WASH REQUIREMENTS	Wash before eating, drinking, or using toilet facilities.
RESPIRATORY	Respiratory protection required if the exposure level is unknown or has been
REQUIREMENTS	measured and found to exceed the published exposure limits.
	Self-contained breathing apparatus with a full facepiece operated in
	pressure-demand or other positive pressure mode.

9. PHYSICAL AND CHEMICAL PROPERTIES

PURE SUBSTANCE OR PREPARATION	Preparation.
PHYSICAL FORM	Liquid
COLOUR	tan
ODOUR	Negligible
ODOUR THRESHOLD	Not available
OXIDIZING PROPERTIES	Not applicable
SOLUBILITY IN WATER	Insoluble
PARTITION COEFFICIENT (n-	Not applicable
octanol/water)	
VISCOSITY	20000 mPa.s
RELATIVE DENSITY	1.1
EVAPORATION RATE	Not applicable
VAPOUR PRESSURE (mmHg)	Not applicable
VAPOUR DENSITY (air=1)	Not applicable
VOLATILES	< 0.1 %
VOLATILE ORGANIC COMPOUNDS	< 1 g/litre
AUTOFLAMMABILITY	480 °C
FLASH POINT	> 200 C (Pensky-Martens Closed Tester)

10. STABILITY AND REACTIVITY

STABILITY	Stable
CONDITIONS TO AVOID	Avoid moisture contamination.
HAZARDOUS	Decomposes upon heating to release toxic fumes of nitrogen oxides, carbon
DECOMPOSITION PRODUCTS	monoxide, carbon dioxide, and hydrogen cyanide.

11. TOXICOLOGICAL INFORMATION

EYE	Will cause eye irritation.
SKIN	Repeated and/or prolonged contact may cause irritation and skin
	sensitisation.
INHALATION	Vapors and/or aerosols may cause irritation. May cause allergic respiratory
	reaction.
INGESTION	Low oral toxicity
I (OLD HOI)	Low order toxicity.
EFFECTS OF CHRONIC	Although this product has not been tested for chronic effects it is judged as
EXPOSURE	having a low order of toxicity based on component information. Use of good
	industrial hygiene practices is recommended.
TARGET ORGANS	Respiratory system. Skin
RESPIRATORY	Sensitizer.
SENSITIZATION	Sensitized
SKIN SENSITIZATION	Sensitizer
SIGN SENSITIZATION	Sensitizer.
PRODUCT INFORMATION	Not established
	Tot established.

12. ECOLOGICAL INFORMATION

POTENTIAL EFFECT ONNot readily biodegradable.ENVIRONMENTAQUATIC TOXICITYAQUATIC TOXICITYNone EstablishedPOTENTIAL TOThe product has low potential for bioaccumulation.BIOACCUMULATE

13. DISPOSAL CONSIDERATIONS

WASTE DISPOSAL METHODS	Waste disposal should be in accordance with existing Community, National
	and local regulations.
EMPTY CONTAINER	Empty containers may contain product residue; follow SDS and label
WARNINGS	warnings even after they have been emptied.

14. TRANSPORT INFORMATION (See also section 9)

IATA CLASSIFICATION	Not classified as dangerous.
IMDG CLASSIFICATION	Not classified as dangerous.
ADR/RID	Not classified as dangerous.

15. REGULATORY INFORMATION

EC INDICATION OF DANGER EC SYMBOL	HARMFUL			
RISK PHRASES	 R20 - Harmful by inhalation. R36/37/38 - Irritating to eyes, respiratory system and skin. R40 - Limited evidence of a carcinogenic effect. R42/43 - May cause sensitization by inhalation and skin contact. R48/20 - Harmful: danger of serious damage to health by prolonged 			
SAFETY PHRASES	exposure through inhalation. S23 - Do not breathe gas/fumes/vapour/spray. S24/25 - Avoid contact with skin and eyes. S37 - Wear suitable gloves. S45 - In case of accident or if you feel unwell, seek medical advice			
CONTAINS	Immediately (show the label where possible). Diphenylmethanediisocyanate, isomers and homologues			
SPECIAL PHRASES	Diphenylmethane-4,4'-diisocyanate Contains isocyanates. See information supplied by the manufacturer			
EINECS	All components of this product are listed in EINECS or ELINCS.			
16. OTHER INFORMATION				
REVISION REPLACES VERSION DATED ORIGINAL DOCUMENT DATED CHANGES SINCE PREVIOUS	19/07/2010 26/06/2009 05/03/2007 Section 1 Section 2 Section 3 Section 15 Section 16			
ISSUE FOR REGULATORY	Joseph Gabriel			
INFORMATION, CONTACT:	Technical Director			
	Purbond AG CH-6203 Sempach-Station			
	Switzerland			
	Tel. +41-(0)41-469-6863			
	Mail joseph.gabriel@purbond.com			
FULL TEXT OF RISK PHRASES FOR INGREDIENTS INDICATED IN SECTION 3:				
Diphenylmethanediisocyanate, isom	ers R20 - Harmful by inhalation. R26/7/28 - Initiating to group instant operations and altim			
and noniologues	R40 - Limited evidence of a carcinogenic effect.			
	R42/43 - May cause sensitization by inhalation and skin contact.			
	R48/20 - Harmful: danger of serious damage to health by prolonged			
Diphenvlmethane-4.4'-diisocvanate	R20 - Harmful by inhalation.			
F - S S - S S - S	R36/37/38 - Irritating to eyes, respiratory system and skin.			
	R40 - Limited evidence of a carcinogenic effect.			
	R42/43 - May cause sensitization by inhalation and skin contact. R48/20 - Harmful: danger of serious damage to health by prolonged exposure through inhalation.			

This information is provided for health and safety assessment by an industrial user. Reference should be made to any relevent local or national health, safety, and environmental legislation. This information does not constitute indication of suitability for specific uses.